Fracture Assessment for Electron Beam Welded Damage Tolerant Ti-6Al-4V Alloy by the FITNET Procedure

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Abstract: Fracture assessment of the cracked structures is essential to avoiding fracture failure. A number of fracture assessment procedures have been proposed for various steel structures. However, the studies about the application of available procedures for titanium alloy structures are scarcely reported. Fracture assessment for the electron beam(EB) welded thick-walled damage tolerant Ti-6Al-4V(TC4-DT) alloy is performed by the fitness-for-service(FFS) FITNET procedure. Uniaxial tensile tests and fracture assessment tests of the base metal and weld metal are carried out to obtain the input information of assessment. The standard options and advanced options of FITNET FFS procedure are used to the fracture assessment of the present material. Moreover, the predicted maximum loads of compact tensile specimen using FITNET FFS procedure are verified with the experimental data of fracture assessment tests. As a result, it is shown that the mechanical properties of weld metal are inhomogeneous along the weld depth. The mismatch ratio *M* is less than 10% at the weld top and middle, whereas more than 10% at the weld bottom. Failure assessment of standard options with error less than 6%. Furthermore, there are no potential advantages of applying higher options or mismatch options. Thus, the present welded joints can be treated as homogeneous material during the fracture assessment, and standard option 1 can be used to achieve accurate enough results. This research provides the engineering treatment methods for the fracture assessment of titanium alloy and its EB welds.

Key words: FITNET FFS procedure, fracture assessment analysis, electron beam welding, thick-walled plates, titanium alloy

1 Introduction

Thick Ti-6Al-4V titanium alloy has been widely employed in the load-bearing structures of aircraft, submarine shell for deep underwater detection, and gas bottles of satellite launch vehicles due to the superior combination properties, such as high specific strength, low specific gravity, and high corrosive resistance^[1–2]. Driven by the increasing damage tolerant requirements for the significant structures, damage tolerant Ti-6Al-4V(TC4-DT) alloy is developed on the basis of conventional Ti-6Al-4V alloy by decreasing the interstitial element content. The alloy exhibits well damage tolerant in terms of high strength, fracture toughness and fatigue resistance against crack growth, thus the titanium alloy with high thickness begins to apply in some significant structures^[3–4].

Welding technique is widely used to manufacture titanium alloy structures in modern industry. Reliable shielding of weld zone is essential during the welding considering the fact that titanium is easily reactive towards atmospheric gas at the temperature above 350 $^{\circ}C^{[5]}$. Electron

beam welding(EBW) is considered as the most suitable welding method to join thick titanium alloy due to its advantages, such as high power density, low heat input, and high vacuum environment^[6]. These advantages are helpful to achieve large penetration, low distortion, light weight and better quality of welded joint.

Welded joint usually exhibits microstructure and mechanical heterogeneity due to the different thermal cycle experienced during welding. The heterogeneity may give an influence on the mechanical properties and deformation performance of the structures. Moreover, welding residual stresses and weld defects easily exist in the welds. The weld defects usually become the crack nucleation due to the stress concentration, and then the crack instability extension and failure occurs under larger loading or the inadequate fracture toughness. Therefore, integrity assessment of welded structures is crucial for investors, designers, and supervising production and exploitation engineers^[7].

Several defect assessment methods specific to the strength mismatching welded structures are proposed up to now, such as ETM-MM^[8], R6^[9] and SINTAP^[10]. The fitness-for-service(FFS) FITNET procedure completed in 2006 is widely accepted as a unified assessment procedure to combine the strengths of the methods available^[11]. This

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procedure contains four major analysis modules including fracture, fatigue, creep and corrosion. The fracture module is mainly based on the SINTAP and R6 procedure, which is suitable for the assessment of metallic structures and component with welds when the failure mechanism considered is fracture. The use of fracture module of the FITNET procedure is important specifically to assess the structural significance of welding induced defects(pores, flaws or cracks), particularly crack-like flaws that need to be assessed to prevent failure of the welded component during service.

Till now, a number of studies have been carried out to validate the application of the FITNET FFS procedure for the various steels and some aluminum alloy welded structures with various configurations^[12–15]. It is found that the fracture module of the FITNET FFS procedure provides an accuracy estimate of critical load or crack size below which failure will not occur. However, few validation studies have been reported for titanium alloy and its welded structures, especially the welded structures by advanced welding techniques. Considering the specific features of titanium alloy and the EB welded joints(narrow weld width, strength mismatch, etc), some basic equations of FITNET FFS procedure may be not suitable for the fracture assessment of titanium alloy and its EB welded joints. Thus, there is an increasing need of fitness-for-service fracture assessment for the EB welded titanium structures using FITNET FFS procedure with the increasing application of the material and technology in important load-bearing structures. In previous studies, the analysis of the mechanical properties including tensile properties and fracture toughness of EB welded thick TC4-DT joints have been made to achieve basic input data of fracture assessment^[16]. The aim of the present work mostly focus attention on the fracture assessment of these joints, and to provide a validation for the application of the FITNET FFS procedure to the EB welded thick titanium alloy.

2 Fracture Module of the FITNET FFS Procedure

The fracture module of FITNET FFS procedure^[11] is based on the principle that failure is deemed to occur when the crack driving force exceeds the material's ability to resist the extension of the crack (fracture toughness of material). Two approaches can be adopted to assess the integrity of cracked structures in the FITNET FFS procedure: the failure assessment diagram(FAD) approach and the crack driving force(CDF) approach. The two approaches provide identical results as long as the same input data are used. In this paper, the FAD approach is used.

In the FAD route, a failure assessment line(FAL) is constructed in terms of two parameters, K_r and L_r by

$$K_{\rm r} = f(L_{\rm r}), \qquad (1)$$

where K_r is the crack driving force in terms of stress intensity factor K_I normalized by the materials fracture toughness K_{mat} , and L_r is the ratio of the total applied load F giving rise to the primary stresses to plastic limit load F_Y .

It is necessary to plot an assessment point or a set of assessment points of coordinates(L_r , K_r) calculated under the loading conditions applicable to use the FAD approach. The assessment of the component is then based on the relative location of the failure assessment point and the FAL, as shown in Fig. 1. The assessment points lying on or within the envelope indicates that the structure is acceptable against this critical condition. The point lying outside the envelope indicates that the structure had failed to meet this critical condition.

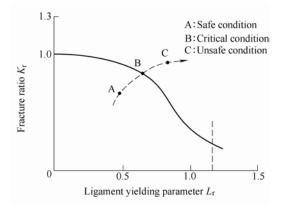


Fig. 1. Failure assessment diagram for fracture initiation

Different analysis options have been provided to plot the FAL in the FITNET FFS Fracture Module depended on the quality and detail of the material's property data available. There are one basic option, three standardized options and two advanced options. The higher the option of analysis, the higher is the quality required of the input data, and the more complex are the analysis routines. Standard option 1 requires values of the material's yield and tensile strength. The option analysis is developed for homogeneous components or the weldments with the mismatch ratio M(the yield strength of weld metal normalized by that of base metal) less than 10%. Standard option 2 is the mismatch option, which is the extension of option 1 to weldments by incorporating the strength mismatch effect when the mismatch ratio is more than 10%. The method requires yield strength and tensile strength of both the base and weld metal. Standard option 3 requires full stress strain data and can be applied either to homogeneous component or to the mismatch component. Advanced option 4 requires results of elastic-plastic finite element(FE) analysis of the defective component, while option 5 invokes constraint treatment and also requires results of detailed elastic-plastic analysis of flawed component.

In this paper, only some options of FITNET FFS

procedure will be used. The required information for the fracture assessment is schematically illustrated in Fig. 2.

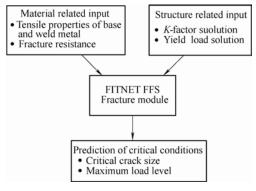


Fig. 2. Required input information of the FITNET FFS procedure

3 Experimental Procedure and Material Properties

3.1 Materials and electron beam welding process

The materials used in this study were 50 mm thick TC4-DT plates in duplex annealing condition. Nominal compositions (wt. %) of the alloy were 5.6–6.35 Al, 3.6–4.4 V, ≤ 0.25 Fe, ≤ 0.05 C, ≤ 0.03 N, ≤ 0.0125 H, ≤ 0.13 O and balance Ti. EB welds (butt-joints) were produced along the transverse direction of the plate in vacuum chamber of a high voltage ZD 150-15A EBW machine using the heat input of 24.75 kJ/m, focus current of 2191 mA and velocity of 800 mm/s.

3.2 Tensile properties

As mentioned in the previous section, it is found that tensile properties of base metal and weld metal are the important input parameters for the fracture assessment of weldments. In this work, micro tensile specimens were used to determine the tensile properties of base and weld metal. As the welded joints are thick, the tensile properties of weld metal along different weld depth were measured. Fig. 3 shows the sampling position and tensile specimen dimension defined based on the small size geometry of the ASTM E8M-04 specification^[18]. Tensile tests were performed at room temperature using a MTS testing system with a displacement extensometer attached in the gage length of 15 mm, and the full stress strain data were recorded.

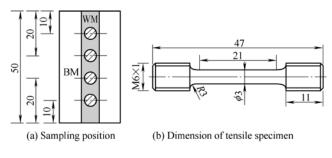


Fig. 3. Schematic illustration of specimen

The stress strain curve of both the base and weld metal is continuous and do not exhibit a yield plateau, as shown in Fig. 4. The tensile test results are given in Table 1.

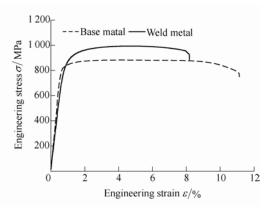


Fig. 4. Engineering stress strain curves

Material		Yield strength $\sigma_{\rm Y}$ / MPa	Ultimate tensile Strength $\sigma_{\rm UTS}$ / MPa	Elongation $\delta_5 / \%$	Mismatch ratio M	Strain hardening exponent n	Elastic modulus E/GPa
Base metal		815	880	11.3	—	25	125
	Position 1	820	940	8.75	1.01		
Weld	Position 2	862	982	8.5	1.06	17	115
metal	Position 3	892	1004	10.2	1.09	1 /	115
	Position 4	944	1053	5.5	1.16		

It is found that the yield strength of weld metal is larger than that of base metal, which means that the EB welded joints are weld strength overmatching. The reason for this is the formation of acicular martensite in the fusion zone after the EBW in comparison with primary α and lamellar ($\alpha + \beta$) bimodal structure of base metal(Fig. 5). Moreover, the weld strength mismatch ratios along different weld depth are different. The mismatch ratio increases with the weld depth increasing. At the position from 1 to 3, the *M* is smaller than 10%, while at position 4, the *M* increases to 1.16. In the previous section, it is known that when the strength mismatch ratio is smaller than 10 %, the options for homogeneous materials should be used to assess the welded joint. For the present thick-walled welds, both the conditions of M > 10% and M < 10% are found. Therefore, in the present work, the weld metal is still supposed as homogeneous material, and both the homogeneous options and mismatch options were used for the fracture assessment. The maximum strength mismatch ratio of 1.16 was used for the case of mismatch options.

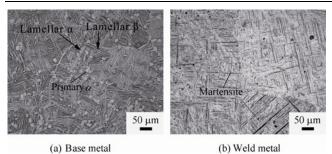
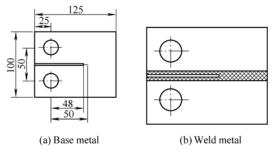
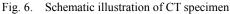


Fig. 5. Microstructure of EB welded TC4-DT joints

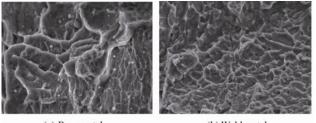
3.3 Fracture properties

Fracture toughness K_{IC} of the base metal and EB welded joints was tested using CT specimen based on ASTM E399-09 standard^[16]. The specimen depth is the depth of primary plates. The geometry and dimension of CT specimens are shown in Fig. 6. Starter notch and fatigue precracking were carried out to develop an initial sharp crack. The notch was spark machined at the center of the weld metal for the welds and in the L-T direction for the base metal. The final crack length over specimen width a/W was 0.5. The fracture toughness tests were performed using a MTS testing system at room temperature.





Both the base metal and welded joints exhibit ductile dimpled fracture features (Fig. 7). The fracture toughness values is shown in Table 2. The toughness of base metal is higher than that of weld metal. The toughness reduction of the weld metal is mainly related to the needle martensite formed in the weld metal.



(a) Base metal (b) Weld metal Fig. 7. Fractography of CT specimens

 Table 2
 Fracture properties of base metal and weld metal

Material	Fracture toughness $K_{\rm IC}/({\rm MPa}\cdot{\rm m}^{-1/2})$	Characteristic value of fracture toughness $K_{\text{mat}}/(\text{MPa} \cdot \text{m}^{-1/2})$	
Base metal	104, 118 (K ₀), 115 (K ₀)	104	
Weld metal	72.3, 70.1, 73.2	70.1	

The FITNET FFS procedure indicates that the characteristic value of the fracture toughness K_{mat} may be based on the minimum value obtained in a set of three tests. In this work, the failure mechanism of base metal and welded joints is ductile, thus, K_{mat} may be based on the minimum value. According to the Table 2, K_{mat} is chosen to be 104 MPa • m^{-1/2} for the base metal, and 70.1 Mpa • m^{-1/2} for the welded joints.

4 Fracture assessment for EB welded thick-walled TC4-DT alloy

FALs of standard options are mainly based on some engineering equations. The application of these equations has been validated in many researches for the various steels and some aluminum alloy welded structures. However, there is a need for the validation of these equations for titanium alloy and its welded structure. As the FAL of option 4 is proposed based on the detailed elastic-plastic FE analysis, it was used to validate the suitability of FAL equations of standard options. Moreover, the standard options were used to predict the maximum load level of CT specimen, and the application of these options was validated by experiment results.

4.1 Standard options for homogeneous component

Option 1 and option 3 can be used for homogeneous component, which are decribed as follows:

Option 1

The FAL is computed by using

$$f(L_{\rm r}) = \left[E\varepsilon_{\rm r}/\sigma_{\rm r} + 0.5L_{\rm r}^2/(E\varepsilon_{\rm r}/\sigma_{\rm r}) \right]^{-1/2},$$

for $0 \le L_{\rm r} \le L_{\rm r}^{\rm max},$ (2)

$$f(L_{\rm r}) = f(L_{\rm r} = 1)L_{\rm r}^{(N-1)/2N}$$
, for $1 \le L_{\rm r} \le L_{\rm r}^{\rm max}$, (3)

where N is strain hardening coefficient,

$$\mu = \min[0.001(E/\sigma_Y); 0.6], \qquad (4)$$

$$L_{\rm r}^{\rm max} = 0.5 \left(1 + \sigma_{\rm UTS} / \sigma_{\rm Y} \right). \tag{5}$$

Option 3

The FAL for homogeneous component is given by

$$f(L_{\rm r}) = \left[E\varepsilon_{\rm r}/\sigma_{\rm r} + 0.5L_{\rm r}^2/(E\varepsilon_{\rm r}/\sigma_{\rm r})\right]^{-1/2}, \quad (6)$$

for $0 \leq L_{\rm r} \leq L_{\rm r}^{\rm max}$,

where $\varepsilon_{\rm r}$ is the true strain obtained from the uniaxial stress-strain curve at a true stress $\sigma_{\rm r}$ of $L_{\rm r}\sigma_{\rm Y}$.

4.2 Standard options for mismatch component

Option 2 and option 3 can be used for mismatch

component, which are decribed as follows:

Option 2

The FAL of option 2 is given as follows:

$$f(L_{\rm r}) = \left[1 + 0.5L_{\rm r}^2\right]^{-1/2} \times \left[0.3 + 0.7\exp\left(-\mu_{\rm M}L_{\rm r}^6\right)\right], \quad (7)$$

for $0 \le L_{\rm r} \le 1$,

$$f(L_{\rm r}) = f(L_{\rm r} = 1)L_{\rm r}^{(N_{\rm M}-1)/2N_{\rm M}}$$
, for $1 \le L_{\rm r} \le L_{\rm r}^{\rm max}$, (8)

where

$$\mu_{\rm M} = \min\left\{\frac{M-1}{(F_{\rm YM}/F_{\rm YB}-1)/\mu_{\rm W} + (M-F_{\rm YM}/F_{\rm YB})/\mu_{\rm B}}; \ 0.6\right\},\tag{9}$$

$$\mu_{\rm W} = \min[0.001(E_{\rm W}/\sigma_{\rm YW}); 0.6], \qquad (10)$$

$$\mu_{\rm B} = \min[0.001(E_{\rm B}/\sigma_{\rm YB}); 0.6], \qquad (11)$$

and the plastic collapse limit L_r^{max} (SINTAP procedure):

$$L_{\rm r}^{\rm max} = 0.5 \left(1 + \sigma_{\rm UTS} / \sigma_{\rm Y}\right). \tag{12}$$

In the FITNET procedure, Eq. (12) is replaced by

$$L_{\rm r}^{\rm max} = \frac{F_{\rm YM}}{F_{\rm YB}} \min \begin{cases} 0.5 \left(1 + \sigma_{\rm UTS,W} / \sigma_{\rm YW}\right), \\ 0.5 \left(1 + \sigma_{\rm UTS,B} / \sigma_{\rm YB}\right), \end{cases}$$
(13)

where $F_{\rm YM}$ and $F_{\rm YB}$ are the yield load for the mismatch and base material component, respectively.

Option 3

The FAL for mismatch component is given by

$$f(L_{\rm r}) = \left[E \varepsilon_{\rm r,M} / \sigma_{\rm r,M} + 0.5 L_{\rm r}^2 / (E \varepsilon_{\rm r,M} / \sigma_{\rm r,M}) \right]^{-1/2}, \quad (14)$$

for $0 \leq L_{\rm r} \leq L_{\rm r}^{\rm max},$

where $\varepsilon_{r, M}$ is the equivalent true strain obtained from the uniaxial stress strain curve at a equivalent true stress σ_r of $L_r\sigma_{Y}$.

The equivalent stress-strain curve of mismatch material is defined as follows:

$$\sigma_{\rm M}\left(\varepsilon_{\rm p}\right) = \frac{\left(F_{\rm YM}/F_{\rm YB}-1\right)\sigma_{\rm W}\left(\varepsilon_{\rm p}\right)}{M-1} + \frac{\left(M-F_{\rm YM}/F_{\rm YB}\right)\sigma_{\rm p}\left(\varepsilon_{\rm p}\right)}{M-1},$$
(15)

where $F_{\rm YM}/F_{\rm YB}$ is defined for $M(\varepsilon_{\rm p}) = \sigma_{\rm W}(\varepsilon_{\rm p})/\sigma_{\rm B}(\varepsilon_{\rm p})$ as a function of the plastic strain $\varepsilon_{\rm p}$ of the plastic branch of the stress strain curve.

The plastic collapse limit L_r^{max} is defined using the equivalent yield stress σ_{YM} and equivalent flow stress σ_{fM} :

$$\sigma_{\rm YM} = \left(F_{\rm YM} / F_{\rm YB} \right) \sigma_{\rm YB},\tag{16}$$

$$\sigma_{\rm fM} = \left[F_{\rm YM} \left(\overline{\varepsilon_{\rm p}} \right) \middle/ F_{\rm YB} \left(\overline{\varepsilon_{\rm p}} \right) \right], \tag{17}$$

with $\overline{\varepsilon_p}$ being the lower of the plastic strains at the flow stresses of base metal and weld metal.

The equivalent true stress-strain curves for the mismatch components are dependent on the properties of both the base and weld metal, as well as the geometry of the component. The curve for the present joints with a center crack in tension(CCT) is shown in Fig. 8. The stress-strain curve of weld metal in weld position 4 is used to construct the curveas M at the position is more than 10%.

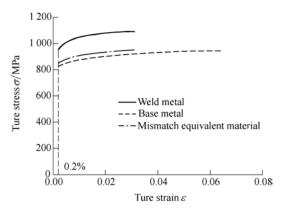


Fig. 8. True stress-strain curve for equivalent material

4.3 Finite element validation of standard options

Option 4 was used to validate the applicability of standard options FALs in this section.

Option 4

When the elastic-plastic J-integral for the mismatch component of interest is available, the FAL can be determined from

$$f(Lr) = (J/J_e)^{-1/2},$$
 (18)

where J_e and J are the values of the J-integral from the elastic and elastic-plastic analyses, respectively.

4.3.1 Finite element analysis

J-integral may be available from a FE analysis. In the FE analysis, the actual weld configurations were simplified to be composed of only base metal and weld metal. The heat affected zone was included in the weld metal due to its

similar microhardness with the weld metal^[16]. Moreover, the tensile mode is dominant in the loading of most structures, and wide plate test is widely accepted to be a more realistic method to assess the fracture behavior of the welded structure. Therefore, in the present work, the 2D CCT specimen with W=50 mm, 2H=8 mm, a/W=0.5 is considered, as shown in Fig. 9(a).

Both elastic and elastic-plastic analyses were carried out to obtain J and J_e , respectively, using a commercial software ABAQUS. The required material properties are indicated in Table 1, including the strain hardening exponent and elastic modulus. A quarter of the specimen was analyzed due to the symmetry, and the crack tip singularity was designed using collapsed element. The finite element mesh is shown in Fig. 9(b). The model includes 384 elements and 1 247 nodes. Moreover, reduced integration eight-node plain strain elements (CPE8R) were used considering the high thickness of the specimen. The value of J and J_e can be extracted directly from ABAQUS.

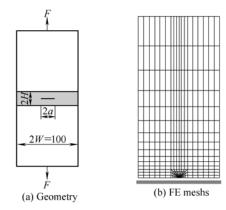


Fig. 9. Schematic illustration of CCT specimen

4.3.2 Component-related input parameters

The component-related input parameter for the FALs determination of FITNET FFS procedure is the mismatch yield load F_{YM} , the solutions of which has been provided in the Annex B of FITNET FFS procedure^[11]. The details are not given for the sake of space.

4.3.3 Comparison of options

Fig. 10 compares the FAL of option 4 with those of standard options. The comparison for the case of homogeneous material is shown in Fig. 10(a), whereas the comparison for the case of mismatch material is shown in Fig. 10(b). It is found that the FALs are in the correct order for the two cases. The higher option curve is associated with less conservation, while the lower option curve is

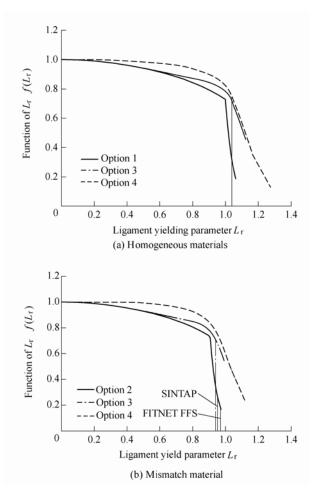


Fig. 10. Comparison of FALs from various options

For the homogeneous options, the FAL of option 1 is quite close to that of option 3, and the FALs of option 1 and option 3 are slightly conservatism comparing with that of option 4. This may be dependent on the inherent properties of TC4-DT alloy and its EB welds. From Table 1 and Fig. 5, it is found that the plastic branch of stress-strain curve is quite flat, and the strain hardening properties beyond yield is weak. This will make the cut-off of plastic collapse L_r^{max} be quite close to 1. Moreover, the benefit of a higher option is not obvious at the values of $L_r \leq 0.8^{[11]}$. Therefore, the FALs of option 1 and option 3 show excellent agreement. Due to the flat stress strain curve beyond yield, the assessment of option 1 and option 3 exhibit conservatism, which is similar with the results for the case where the material shows a yield plateau^[20].

For the mismatch options, the similar trends are found with that of homogeneous options. The FAL of option 2 is quite close to that of option 3, and the FALs of option 2 and 3 are slightly conservatism. However, it is noted that the L_r^{max} of option 3 is smaller than that of option 2, which may be questionable. It is known that the option 3 for mismatch component is based on the equivalent stress strain curve,

associated with more conservation.

which is derived from the full stress-strain curves of both base and weld metal. The equivalent curve is strongly affected by the true plastic strain of base and weld metal. In the present work, very obvious difference of true plastic strain exists between the weld metal and base metal (Fig. 8). The equivalent stress strain curve is restricted by the weld metal, and then gives an influence on the value of L_r^{\max} in option 3. Therefore, the application of option 3 to the defective assessment of mismatch component with plastic collapse dominated failure should be treated with caution, when there is obvious difference of plastic deformation between the base metal and weld metal. In addition, the L_r^{max} definition of SINTAP procedure is more close to the option 3 than that of FITNET FFS procedure, so the SINTAP procedure equation of L_r^{max} is suggested for the present materials.

The effect of strength mismatch on the FALs of standard options is also investigated and shown in Fig. 11. It is found that the effect of mismatch may be negligible due to the narrow weld width of EB welds.

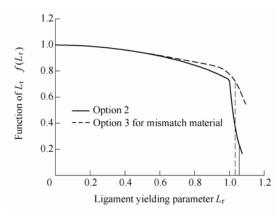


Fig. 11. Effect of strength mismatch on the FALs

4.4 Experimental validation of standard options

In this section, the CT specimen of the fracture toughness testing was used to validate the applicability of standard options of FITNET FFS procedure for the present EB welds by comparing the predicted maximum load level with the experimental results.

4.4.1 Component-related input data and calculation of K_r

The component-related input parameter for the prediction of critical load using FITNET FFS procedure is the mismatch yield load F_{YM} , and the stress intensity factor K_I . The solutions of F_{YM} have been provided in Ref. [21]. The details are not given for the sake of space.

The stress intensity factor K_1 for CT specimen is given in a closed form solution as

$$K_I = \frac{P}{BW^{1/2}} f(x), \tag{19}$$

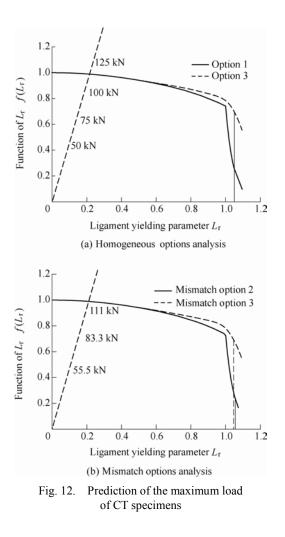
$$f(x) = \frac{(2+x)\left[0.886 + 4.64x - 13.32x^2 + 14.72x^3 - 5.6x^4\right]}{(1-x)^{3/2}},$$
(20)

where x = a/W, *F* is the applied load, *W* is the width of the specimen, *a* is the crack length and *B* is the specimen thickness. Since *K*-factor is a geometrical function, it is also valid for welded specimen.

 $K_{\rm r}$ can be evaluated from primary and secondary stresses. The secondary stress for weldments is usually the residual stress due to the welding process. In this paper, the post weld heat treatment was performed to reduce the as-welded residual stresses. Therefore, the effect of secondary stresses is ignored during the fracture assessment, and only primary stress was used to calculate the $K_{\rm r}$.

4.4.2 Prediction of maximum loads

The crack initiation analysis of standard options was carried out to predict the maximum loads of CT specimen (Fig. 12), which was then compared with the measured maximum loads in the experiment, as shown in Fig. 13.



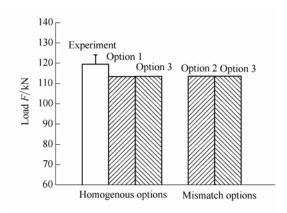


Fig. 13. Comparison of predicted maximum load by standard options with experimental result

It is found that for both the case of homogeneous and mismatch option, the predicted maximum loads of lower standard options(option 1 and option 2) are almost the same with that of option 3, which are all quite close to the experimental results. This means that the FITNET FFS procedure is applicable to the fracture assessment of thick-walled EB welded TC4-DT titanium alloy. In addition, the predicted maximum loads using mismatch options are similar with those using homogeneous options. The homogeneous options do not exhibit more conservative. All of this demonstrates that the potential advantage of applying higher options or mismatch options is not obvious for the fracture assessment of the thick-walled titanium alloy welds. Standard option 1 can be used to achieve accurate enough results. The experimental validation provides confidence of the utilization of the FITNET procedure to assess titanium alloy and its EB welds.

The results of the present validation work clearly shows that the suitability of the FITNET procedure for the thick-walled titanium alloy with narrow EB welds, when failure is fracture-dominated. Further studies are desirable for the fracture assessment of thin-walled titanium alloy and its welds covering plastic collapse-dominated failure.

5 Conclusions

(1) The FALs of standard options of FITNET procedure is suitable for the TC4-DT alloy and its EB welded joints. Nevertheless, the applicability of option 3 to mismatch material should be treated with caution for the plastic collapse analysis.

(2) Accurate estimation of the maximum loads can be achieved using both option 1 and option 3 for homogeneous material, and option 2 and option 3 for mismatch material.

(3) For fracture assessment of the present EB welded TC4-DT joints, there are no potential advantages of applying higher options or mismatch options. The welds

can be treated as homogeneous material, and standard option 1 can be used to achieve accurate enough results.

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