**ASTRONOMY**

# **Determination of a Preliminary Orbit by the Cauchy–Kuryshev–Perov Geometric Method**

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**Abstract**—The determination of preliminary orbits of celestial bodies is of interest to observational astronomy in terms of discovering new bodies or identifying them with already known ones. The solution of this problem requires methods that are not limited both by the eccentricity of the orbit and by the time intervals between observations. This article considers the Cauchy–Kuryshev–Perov geometric method for determining a preliminary orbit. It is shown how to determine an orbit that does not lie in the observer's plane of motion within the two-body problem, based only on geometric constructions, and using five angular observations. This method makes it possible to reduce the problem of determining a preliminary orbit to the algebraic system of equations relative to two dimensionless variables with a finite number of solutions. The method is suitable for determining both elliptical and hyperbolic orbits. It has no restrictions on the length of the orbital arc of the observed body and is not limited by the number of complete revolutions around the attractive center between observations. All possible combinations of positions of the body in the orbit are divided into 64 variants and described by the corresponding systems of equations. This article presents an algorithm for finding solutions to the problem without having prior information about the desired orbit. The solutions are sought in a bounded region in which triangulations are performed with triangles ranked according to the search conditions, thus eliminating the consideration of most of them at the initial stage. The solutions of the system are found by the Nelder–Mead method through the search for minima of the target function. The obtained orbits are compared by means of a representation of observations, and the best one is selected. An example of determining the orbit of the comet 2I/Borisov is given.

**Keywords:** preliminary orbit determination, Cauchy–Kuryshev–Perov geometric method, Nelder– Mead method, algebraic equations, comet Borisov.

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## 1. INTRODUCTION

The history of the geometric method for determining the preliminary orbit from angular observations began in the 19th century [1]. In 1847, Cauchy proposed to determine the preliminary orbit through five points of intersection of five straight lines by a plane passing through the center of the Sun [2]. Since a conic section can be defined by these five points, the position of the plane of the orbit should be found such that the focus of the conic coincides with the Sun. There are no known earlier references to the geometric method, although they may have appeared after the works of Newton. In the early 20th century, Harzer addressed the problem, but he never obtained an acceptable solution [3]. According to Vil'ev, the problem was still waiting for researchers as of 1917 [4]. The modern history of the method should be retraced from 1982, when work [5] by Kuryshev and Perov was published, in which a system of equations was constructed and an iterative process for its solution was proposed. The author applied this method to determine the preliminary orbits of comets and asteroids [1]. The purpose of this work is to construct an effective algorithm for the numerical solution of the system of equations with finding all possible orbits. As an example, we consider the determination of the orbit of the interstellar comet 2I/Borisov.

## 2. CONSTRUCTION OF A SYSTEM OF EQUATIONS

Let us denote by  $\mathbf{r}_i$  ( $i = 1, 2, ..., 5$ ) the radius-vectors of the desired orbit at the corresponding times. The equations for heliocentric radius-vectors will then be as follows:

$$
\mathbf{r}_i = \mathbf{e}_i \mathbf{\rho}_i - \mathbf{R}_i,\tag{1}
$$

where  $e_i$  ( $i = 1, 2, ..., 5$ ) are unit vectors of the observed direction to the object, and  $\mathbf{R}_i$  ( $i = 1, 2, ..., 5$ ) are vectors of the Sun's position relative to the topocenter. In order to construct equations describing the desired orbit, let us refer to the formula for determining the orbit parameter by three radius-vectors [6]. If the three position vectors of the object are known, then we can express orbit parameter  $p$  through them [7]. By selecting the triples of vectors  $(\mathbf{r}_1, \mathbf{r}_2, \mathbf{r}_3)$ ,  $(\mathbf{r}_2, \mathbf{r}_3, \mathbf{r}_4)$ , and  $(\mathbf{r}_3, \mathbf{r}_4, \mathbf{r}_5)$ , we obtain three expressions for *p*:

$$
p_{123} = \frac{\pm r_1 |\mathbf{r}_2 \times \mathbf{r}_3| \mp r_2 |\mathbf{r}_1 \times \mathbf{r}_3| \pm r_3 |\mathbf{r}_1 \times \mathbf{r}_2|}{\pm |\mathbf{r}_2 \times \mathbf{r}_3| \mp |\mathbf{r}_1 \times \mathbf{r}_3| \pm |\mathbf{r}_1 \times \mathbf{r}_2|} \ge 0,
$$
  
\n
$$
p_{234} = \frac{\pm r_2 |\mathbf{r}_3 \times \mathbf{r}_4| \mp r_3 |\mathbf{r}_2 \times \mathbf{r}_4| \pm r_4 |\mathbf{r}_2 \times \mathbf{r}_3|}{\pm |\mathbf{r}_3 \times \mathbf{r}_4| \mp |\mathbf{r}_2 \times \mathbf{r}_4| \pm |\mathbf{r}_2 \times \mathbf{r}_3|} \ge 0,
$$
  
\n
$$
p_{345} = \frac{\pm r_3 |\mathbf{r}_4 \times \mathbf{r}_5| \mp r_4 |\mathbf{r}_3 \times \mathbf{r}_5| \pm r_5 |\mathbf{r}_3 \times \mathbf{r}_4|}{\pm |\mathbf{r}_4 \times \mathbf{r}_5| \mp - |\mathbf{r}_3 \times \mathbf{r}_5| \pm |\mathbf{r}_3 \times \mathbf{r}_4|} \ge 0,
$$
\n(2)

where  $r_i = |\mathbf{r}_i|$  (*i* = 1, 2, …, 5), the upper and lower signs  $\mp$  and  $\pm$  in (2) correspond to the positive and negative signs of the sine of the angle between the vectors in the subsequent vector product. As can be easily seen, for Eqs. (2), the numerator and denominator cannot have different signs. If we equate the first and second Eqs. (2), i.e.,  $p_{123} = p_{234}$ , and then the second and third,  $p_{234} = p_{345}$ , we obtain two equations of orbit determination in the noncoplanar case. Therefore, the problem is reduced to solving a system of two algebraic equations with respect to five unknowns  $\rho_i$  ( $i = 1, 2, ..., 5$ ). However, the topocentric distances are related by three conditions from the mixed products of vectors (1). This makes it possible to express the other three through two of them and obtain an algebraic system with respect to two variables, e.g.,  $\rho_1$  and  $\rho_5$ :

$$
f_1 = (\pm r_1 |\mathbf{r}_2 \times \mathbf{r}_3| \mp r_2 |\mathbf{r}_1 \times \mathbf{r}_3| \pm r_3 |\mathbf{r}_1 \times \mathbf{r}_2|) (\pm |\mathbf{r}_3 \times \mathbf{r}_4| \mp |\mathbf{r}_2 \times \mathbf{r}_4| \pm |\mathbf{r}_2 \times \mathbf{r}_3|)
$$
  
\n
$$
-(\pm r_2 |\mathbf{r}_3 \times \mathbf{r}_4| \mp r_3 |\mathbf{r}_2 \times \mathbf{r}_4| \pm r_4 |\mathbf{r}_2 \times \mathbf{r}_3|) (\pm |\mathbf{r}_2 \times \mathbf{r}_3| \mp |\mathbf{r}_1 \times \mathbf{r}_3| \pm |\mathbf{r}_1 \times \mathbf{r}_2|) = 0,
$$
  
\n
$$
f_2 = (\pm r_2 |\mathbf{r}_3 \times \mathbf{r}_4| \mp r_3 |\mathbf{r}_2 \times \mathbf{r}_4| \pm r_4 |\mathbf{r}_2 \times \mathbf{r}_3|) (\pm |\mathbf{r}_4 \times \mathbf{r}_5| \mp |\mathbf{r}_3 \times \mathbf{r}_5| \pm |\mathbf{r}_3 \times \mathbf{r}_4|)
$$
  
\n
$$
-(\pm r_3 |\mathbf{r}_4 \times \mathbf{r}_5| \mp r_4 |\mathbf{r}_3 \times \mathbf{r}_5| \pm r_5 |\mathbf{r}_3 \times \mathbf{r}_4|) (\pm |\mathbf{r}_3 \times \mathbf{r}_4| \mp |\mathbf{r}_2 \times \mathbf{r}_4| \pm |\mathbf{r}_2 \times \mathbf{r}_3|) = 0.
$$
  
\n(3)

In this case, in view of the nonlinearity of system (3), several solutions are likely to appear. System (3) is not the only one possible for the geometric method [1], but it is one of the simplest from the point of view of derivation of equations.

Equations (3) satisfy both elliptic and hyperbolic motion. In this case, different combinations of upper and lower signs for seven angles between vectors correspond to 64 sets of equations for the ellipse and 32 for the hyperbola. A detailed description of the combinations of angles for triplets of vectors was presented in [6]. Moreover, the specified 64 sets of equations for elliptic motion include all cases with angles differing by values that are multiple of  $2\pi$  from the specified ones. Therefore, to find a solution on an arc with any number of revolutions around the attractive center, it is sufficient to consider 64 systems of equations (3).

## 3. TRANSITION TO NEW VARIABLES

Topocentric distances can only take positive values. Accordingly, the region of the desired solutions is an open quadrant, which should be restricted to maximum values. To do this, it was proposed in [6] to switch to normalized dimensionless variables. Consider unit vector  $N = (N_x, N_y, N_z)$ , which is the vector of the normal to the plane of the desired orbit. For convenience of relating **N** with the orbital elements, let us select an ecliptic coordinate system. Let us express  $\rho_i$  ( $i = 1, 2, ..., 5$ ) through **N**. If Eqs. (1) are scalarly multiplied by  $N$ , we obtain expressions for  $\rho_i$  [8]:

$$
\rho_i = \frac{\mathbf{N} \mathbf{R}_i}{\mathbf{N} \mathbf{e}_i}.
$$
 (4)

The singularity condition of **N** can be defined as follows:

$$
N_x^2 + N_y^2 + N_z^2 = 1.
$$
 (5)

We can now substitute (4) into all equations for noncoplanar orbits. We obtain (3) with constraint (5) and dimensionless unknowns  $N_x$ ,  $N_y$ , and  $N_z$ . The solution domain is bounded by the surface of a unit sphere. Intersection points  $f_1$  and  $f_2$  are possible directions of **N**. Since curves (3) are symmetric with

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respect to the ecliptic plane, one can select one of the hemispheres with prograde or retrograde orbital motion. Condition (5) can then be written as

$$
N_z^2 = \sqrt{1 - N_x^2 - N_y^2}
$$
 (6)

and the solution domain of the unit sphere can be mapped onto the unit region of the ecliptic circle. Therefore, we are back to the 2D case with unknowns  $N_r$  and  $N_v$ . An additional advantage of the new variables is that they are not directly related to specific observations and their number, and this simplifies the work on creating the target function. Regions with negative topocentric distances and orbital parameter *p* should be excluded from the unit circle. Introducing such constraints makes it possible to narrow the search for a solution from the entire unit circle to its individual regions.

We reduced the problem to finding a solution in the unit circle. However, the circle is not very convenient for partitioning and dense coverage by simple geometric shapes, such as triangles. For triangulation, a square that describes the given circle, i.e., with a unit half-side, is more suitable. The mapping of a circle of unit radius to the corresponding square is carried out by the following formulas [9]:

$$
N_x = N_{xs}\sqrt{1 - 0.5N_{ys}^2}, \quad N_y = N_{ys}\sqrt{1 - 0.5N_{xs}^2},
$$
\n(7)

where  $N_{xx}$  and  $N_{yy}$  are the coordinates corresponding to the square. Hereinafter, index *s* indicates these coordinates everywhere.

## 4. ORDER OF OBJECT POSITIONS IN THE ORBIT

In geometric solutions, the order of the points of intersection of lines of sight (observations) with the orbit is usually not taken into account. Hence, there can be solutions for which the order of orbit points does not correspond to the chronological one. To control them on orbital arcs  $\Delta\theta_{51}$  <  $2\pi$  for calculations with 16 significant figures, the following conditions can be used:

$$
0.999999 \le \left| \frac{\Delta \theta_{51}}{\Delta \theta_{21} + \Delta \theta_{32} + \Delta \theta_{43} + \Delta \theta_{54}} \right| \le 1.000001, \tag{8}
$$

where  $\Delta\theta_{ij}$  are the values of the angles between  $\bf{r}_i$  and  $\bf{r}_j$  and a relative error of  $10^{-7}$  is not exceeded in the chronological order of the points and is knowingly exceeded if it is violated. The latter condition implies not using very close observations corresponding to similarly small arcs.

## 5. SINGULAR POINTS IN THE SEARCH DOMAIN

On the resulting square, there exist singular points in which all  $\rho_i$  from (4) become indeterminate. Depending on whether the numerator or denominator turns to zero, the value of the corresponding topocentric distance can become either infinitesimal or infinitely large. In addition, the positivity of the distances requires the same sign for the numerator and denominator. The ten equations derived from the numerators and denominators of (4) define a grid of ten curves the pairwise intersection of which defines the boundaries of the domains where the desired solution can be found. Knowing the coordinates of the intersection points of the curves determined by the numerators and denominators of (4) makes it possible to determine the smallest scale of triangulation. The equations in "square" coordinates will be as follows  $(i, j = 1, 2, ..., 5)$ :

$$
X_i N_{xs} \sqrt{1 - 0.5 N_{ys}^2} + Y_i N_{ys} \sqrt{1 - 0.5 N_{xs}^2} + Z_i N_{ys} \sqrt{1 - N_{xs}^2 - N_{ys}^2 + N_{xs}^2 N_{ys}^2} = 0,
$$
  
\n
$$
e_{jx} N_{xs} \sqrt{1 - 0.5 N_{ys}^2} + e_{jy} N_{ys} \sqrt{1 - 0.5 N_{xs}^2} + e_{jz} N_{ys} \sqrt{1 - N_{xs}^2 - N_{ys}^2 + N_{xs}^2 N_{ys}^2} = 0,
$$
\n(9)

where  $\mathbf{R}_i = (X_i, Y_i, Z_i)$  are the ecliptic vectors of the Sun's position relative to the topocenter and  $\mathbf{e}_j = (e_{jx}, z_{jy})$ *ejy*, *ejz*) are the unit ecliptic vectors of the observed direction to the object.

System (9) is nonlinear, and, in order to find its roots, it is convenient to apply the method of continuation of the solution by the parameter with the best parametrization [10]. The problem can be reduced to a system of three ordinary differential equations  $(i, j = 1, 2, ..., 5)$ :

$$
\frac{dN_{xs}}{d\mu} = (\mathbf{R}_i \times \mathbf{e}_j)_x \sqrt{1 - 0.5N_{xs}^2} + \frac{0.5(\mathbf{R}_i \times \mathbf{e}_j)_y N_{xs} N_{ys}}{\sqrt{1 - 0.5N_{ys}^2}},
$$

$$
\frac{dN_{ys}}{d\mu} = (\mathbf{R}_i \times \mathbf{e}_j)_y \sqrt{1 - 0.5N_{ys}^2} + \frac{0.5(\mathbf{R}_i \times \mathbf{e}_j)_x N_{xs} N_{ys}}{\sqrt{1 - 0.5N_{xs}^2}},
$$
\n
$$
\frac{dV}{d\mu} = \frac{(1 - 0.5(N_{xs}^2 + N_{ys}^2))[(\mathbf{R}_i \times \mathbf{e}_j)\mathbf{N}]}{\sqrt{(1 - 0.5N_{xs}^2)(1 - 0.5N_{ys}^2)(N_{xs}^2 - 1)(N_{ys}^2 - 1)}},
$$
\n(10)

where the third equation in square brackets represents the mixed product of vectors in the ecliptic coordinate system, while the lower indices of the vector product denote the corresponding components in the first two equations. We should integrate system (10) in the direction of increase of parameter  $\mu$  until  $v =$ 0 is reached. The solution of system (10) gives us 25 points the minimum and maximum coordinates of which will make it possible to constrain our square to a rectangular region with the finest structure of the solution domain and the densest possible arrangement of the solutions of system (3).

## 6. TRIANGULATION OF THE SOLUTION SEARCH DOMAIN

It is proposed to search for solutions of system (3) in the form of a minimum of the target function

$$
f_{\text{goal}}(\mathbf{N}) = f_1^2(\mathbf{N}) + f_2^2(\mathbf{N}).
$$
\n(11)

Moreover, the search is based on the use of triangulation, i.e., the partitioning of the domain of possible solutions into nonintersecting triangles [11]. These triangles are needed as an initial approximation for variable simplexes, by means of moving the triangles of which the minimum of the target function is searched using the Nelder–Mead method [12, 13].

The whole maximum possible solution domain is a square with a unit half-side. As a basic triangulation, we divide this square into 1024 small squares, each with a side length of 0.0625. We then divide these "small squares" into four equal triangles with a common vertex in the center. With these isosceles triangles with a base length of 0.0625, we fill the entire region for which at least one of the triangle vertices satisfies the positivity condition of five equations (4). The exception is the "region of singular points" (described in the previous section). It increases in all directions until it coincides with the boundaries of the nearest "small square" and takes the form of a rectangle with a side length that is a multiple of 0.0625. Next, we divide this region into squares with a side length of 0.03125. We then divide the obtained "small squares" into triangles with a common vertex in the center. For these, we also check that, for at least one vertex, all  $\rho_i > 0$  (*i* = 1, 2, ..., 5) and (if necessary) condition (8).

Therefore, we fill the entire domain of possible solutions with triangles of two kinds. They are ordered in sequence, as they are filled line by line from the lower left corner of the "large square" to the upper right corner. The "large triangles" are followed in sequence by "small triangles" according to the same rule.

#### 7. DETERMINING TRIANGLE RANKS

From the perspective of searching for solutions, the resulting set of several hundred triangles can be divided into several classes. Samotokhin and Khutorovskii [11] introduced the notion of the rank of a triangle. The authors proposed a linear interpolation of a possible solution by the values of functions  $f_1$  and  $f_2$  at the vertices of the triangle. Let us denote them as  $f_{1(i)}$  and  $f_{2(i)}$  ( $i = 1, 2, 3$ ). Then the coordinates of the linear interpolation point  $(N_{xx(0)})$ ,  $N_{yx(0)}$  can be expressed as follows:

$$
N_{xs\langle 0 \rangle} = N_{xs\langle 1 \rangle} + \xi(N_{xs\langle 2 \rangle} - N_{xs\langle 1 \rangle}) + \zeta(N_{xs\langle 3 \rangle} - N_{xs\langle 1 \rangle}),
$$
  
\n
$$
N_{ys\langle 0 \rangle} = N_{ys\langle 1 \rangle} + \xi(N_{ys\langle 2 \rangle} - N_{ys\langle 1 \rangle}) + \zeta(N_{ys\langle 3 \rangle} - N_{ys\langle 1 \rangle}),
$$
\n(12)

where  $(N_{\rm x}(\delta), N_{\rm y}(\delta))$  (*i* = 1, 2, 3) are the coordinates of the corresponding vertices,

$$
\xi = \frac{f_{2\langle 1\rangle}f_{1\langle 3\rangle} - f_{1\langle 1\rangle}f_{2\langle 3\rangle}}{f_{1\langle 2\rangle}f_{2\langle 3\rangle} - f_{1\langle 1\rangle}f_{2\langle 3\rangle} - f_{1\langle 2\rangle}f_{2\langle 1\rangle} - f_{1\langle 3\rangle}f_{2\langle 2\rangle} + f_{1\langle 3\rangle}f_{2\langle 1\rangle} + f_{1\langle 1\rangle}f_{2\langle 2\rangle}},
$$
\n
$$
\zeta = \frac{f_{1\langle 1\rangle}f_{2\langle 2\rangle} - f_{2\langle 1\rangle}f_{1\langle 2\rangle}}{f_{1\langle 2\rangle}f_{2\langle 3\rangle} - f_{1\langle 2\rangle}f_{2\langle 1\rangle} - f_{1\langle 3\rangle}f_{2\langle 2\rangle} + f_{1\langle 3\rangle}f_{2\langle 1\rangle} + f_{1\langle 1\rangle}f_{2\langle 2\rangle}}.
$$
\n(13)

The point  $(N_{xx(0)}, N_{yx(0)})$  will be inside the triangle if the following conditions are satisfied:

(14) ξ > 0, ζ > 0, ξ + ζ < 1.

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This interpolation is a generalization of the method of chords for solving a nonlinear equation with one variable. However, it should be noted that such a linearization does not guarantee the existence of a real solution inside the triangle if (14) is satisfied. Here, conditions (3) correspond to two straight lines, the zero lines intersecting at  $(N_{xx(0)}), N_{yx(0)}$ .

Let us describe the possible ranks of the triangles: 0, none of the zero-level lines intersect any side of the triangle (the values of  $f_1$  and  $f_2$  for all vertices have the same sign); 1, only one of the zero-level lines intersects the sides of the triangle (the values of either  $f_1$  or  $f_2$  have different signs for different vertices); 2, both zero-level lines intersect the sides of the triangle, but the point  $(N_{xx(0)})$ ,  $N_{yx(0)}$  is outside the triangle (values of  $f_1$  and  $f_2$  have different signs for different vertices, but condition (14) is not met); 3, the point  $(N_{xx(0)}, N_{yx(0)})$  lies inside the triangle, but one of its sides does not intersect any of the zero-level lines (conditions (14) hold, but the sums of the signs of the products of  $f_1$  and  $f_2$  for the vertices of each side are nowhere zero); and 4, the point  $(N_{xx(0)}, N_{yx(0)})$  lies inside the triangle and all its three sides intersect the zero-level lines (conditions (14) are satisfied and there are sides for which the sums of signs of the products of  $f_1$  and  $f_2$  for their vertices are zero).

Such a ranking is convenient to reduce the amount of computation. If there are triangles with ranks 4 and 3, the search for a solution should begin with them. Then, if there is no result or to check its completeness, it is necessary to proceed to triangles with rank 2. Usually, this is sufficient to find all admissible solutions. Triangles with rank 1 should be used only if there are no solutions, in order to be absolutely sure of it.

## 8. THE NELDER–MEAD METHOD

After the triangles are ranked, we can proceed to the search for solutions. The Nelder–Mead method [12] is well suited for this purpose. This iterative method belongs to the gradient-free method; i.e., it does not use derivatives. It only evaluates the values of the target function in the vertices of the deformed triangle at each iteration. A description of the algorithm and the source code in the Fortran language were presented in [14, 15].

The isosceles triangle from Section 6 is considered as a zero approximation. After evaluating the target function in all vertices of the simplex, we find point  $\langle 1 \rangle$  with maximum  $f_{\text{goal}}(N\langle 1 \rangle)$  and point  $\langle 3 \rangle$  with minimum  $f_{\text{goal}}(N\langle 3 \rangle)$ . We then construct the "center of gravity," point  $\langle 0 \rangle$ , by the two remaining vertices  $\langle 2 \rangle$ and  $\langle 3 \rangle$ :

$$
\mathbf{N}_{\langle 0 \rangle} = \frac{1}{2} (\mathbf{N}_{\langle 2 \rangle} + \mathbf{N}_{\langle 3 \rangle}). \tag{15}
$$

The search for a new vertex occurs along the line  $(1)$ – $(0)$ . First, the "reflection" operation is performed: we find the reflection of point  $\langle 1 \rangle$  with respect to point  $\langle 0 \rangle$ :

$$
\mathbf{N}_{\langle 4 \rangle} = \mathbf{N}_{\langle 0 \rangle} + \alpha (\mathbf{N}_{\langle 0 \rangle} - \mathbf{N}_{\langle 1 \rangle}), \tag{16}
$$

where  $\alpha > 0$  is the reflection coefficient.

If  $f(N_{(4)}) \leq f(N_{(3)})$ , the vector  $N_{(4)}-N_{(0)}$  is extended according to the following relation:

$$
\mathbf{N}_{\langle 5 \rangle} = \mathbf{N}_{\langle 0 \rangle} + \gamma (\mathbf{N}_{\langle 4 \rangle} - \mathbf{N}_{\langle 0 \rangle}), \tag{17}
$$

where  $\gamma > 1$  is the extension coefficient. If  $f(N_{(5)}) \le f(N_{(3)})$ , then  $N_{(1)}$  is replaced by  $N_{(5)}$ ; otherwise,  $N_{(1)}$  is replaced by  $N_{(4)}$ . Then, we start the next iteration with the "reflection" operation.

If  $f(N_{(4)}) > f(N_{(i)})$  for all  $i \neq 1$ , then the vector  $N_{(1)}-N_{(0)}$  is extended according to the following ratio:

$$
\mathbf{N}_{\langle 6 \rangle} = \mathbf{N}_{\langle 0 \rangle} + \beta (\mathbf{N}_{\langle 1 \rangle} - \mathbf{N}_{\langle 0 \rangle}),\tag{18}
$$

where  $0 \le \beta \le 1$  is the compression ratio. We replace  $N_{(1)}$  with  $N_{(6)}$ ; then, we start the next iteration with the "reflection" operation.

If  $f(\mathbf{N}_{\langle 4\rangle}) > f(\mathbf{N}_{\langle i\rangle})$ , then, for  $i = 1, 2$ , vectors  $\mathbf{N}_{\langle i\rangle} - \mathbf{N}_{\langle 3\rangle}$  are reduced by a factor of 2 according to the following formula:

$$
N_{\langle i \rangle} = N_{\langle 3 \rangle} + 0.5(N_{\langle i \rangle} - N_{\langle 3 \rangle}), \quad i = 1, 2. \tag{19}
$$

$t$ (UT) (year, month, day)	$\alpha(2000)$ (h, min, s)	$\delta(2000)$ (deg, min, s)	Observatory
2019 09 08.630642	08 44 37,105	$+30,57,54.54$	Mauna Kea
2019 09 28.234820	09 21 15.673	$+241506.14$	OGS, Tenerife
2019 10 18.14757	09 57 58.12	$+152508.0$	Piszkesteto Stn. (Konkoly)
2019 11 07.19364	10 34 06.83	$+040359.1$	<b>Buchloe</b>
2019 11 27.22933	11 09 09.82	$-094449.4$	Tacande, La Palma

Table 1. Observations of the comet 2I/Borisov (Minor Planet Center)

Here, *t* is the world time (year, month, day);  $\alpha$  is the right ascension (hours, minutes, seconds), and  $\delta$  is the declination of the comet (degrees, minutes, seconds) represented in the equatorial coordinate system referenced to the equator for the J2000.0 epoch [16].

Then we start the next iteration with the "reflection" operation. The criterion for ending the iteration process is to check the condition

$$
\sqrt{\frac{1}{3}\sum_{i=1}^{3} \left[f(\mathbf{N}_{\langle i\rangle}) - f(\mathbf{N}_{\langle 0\rangle})\right]^2} \le \epsilon,
$$
\n(20)

where  $\epsilon$  is an arbitrary small number.

## 9. FINDING SOLUTIONS AND REPRESENTING OBSERVATIONS

As described in Section 6, the solutions are sought by all triangles with ranks 4 and 3, if there are any, and with rank 2, if necessary. The resulting solutions should be pairwise compared and multiples should be discarded. The multiplicity condition will be the distance between points smaller than a given small number. We then convert squared coordinates **N** to coordinates on circle (7), and we move from ecliptic

coordinates to equatorial coordinates:  $N^{eqv} = \{N_x^{eqv}, N_y^{eqv}, N_z^{eqv}\}\$ . By substituting  $N^{eqv}$  into (4), we obtain values of five topocentric distances  $\rho_i$  ( $i = 1, 2, ..., 5$ ). After substituting them into (1), we determine the heliocentric position vectors for the five times. From the vectors for the first and fifth observations, we can calculate the orbits for all the found solutions. The representation by the orbits of the central observations (the second, third, and fourth) will allow us to select the desired solution in the best possible way (O–C values).  $N_x^{\text{eqv}}, N_y^{\text{eqv}}, N_z^{\text{eqv}}$ 

## 10. EXAMPLE

As a numerical example, let us consider determining the orbit of the comet 2I/Borisov on a short arc (only the upper signs are taken in Eqs. (3)). This comet was discovered on August 30, 2019, by



Fig. 1. Graphs of Eqs. (3). General view.



**Fig. 2.** Neighborhoods of intersection points of curves (9).

Gennadii Borisov at the MARGO Crimean observatory (in the town of Nauchny). This is the first interstellar comet moving along a hyperbolic trajectory with eccentricity  $e = 3.36$  (see [16] and Table 1).

The plots of Eqs. (3) in "square" coordinates are created on the *Nxs* and *Nys* axes in Figs. 1 and 2. The regions with negative topocentric distances and orbit parameter *p* are colored in gray; where condition (8) is not satisfied, correspondingly, the regions of possible solutions are colored in white.

In Fig. 1, we can see a trapezoidal area of white on the right and a triangular area on the left formed by the intersection of curves (9) and conditions (8). In Fig. 2, two more smaller triangular areas can be seen in the center. The points of contact of these areas correspond to the simultaneous equality of the numerator and denominator (4) to zero. Equations (4) are not defined at these points; consequently, system (8) is not defined either. However, in their close neighborhoods, functions  $f_1$  and  $f_2$  are close to each other. Next, we find the coordinates of intersection points (9) from the solution of system (10). All 25 points are located in the rectangle  $[-0.293, 0.221] \times [-0.894, 0.757]$ . Let us expand its boundaries to multiples of 0.0625:  $[-0.313, 0.250] \times [-0.906, 0.781]$ , and we obtain a rectangle with a side ratio of 9 : 28. It is convenient to divide it into  $18 \times 56$  "small" squares and 4032 "small" triangles. The fraction of "large" squares will be 772, and the fraction of triangles will be 3088. After checking whether domains of possible solutions are covered, 1302 triangles will remain: 1081 "large" and 221 "small" triangles.

The triangles' ranking yields 5 triangles of rank 4, 6 triangles of rank 3, 37 triangles of rank 2, 258 triangles of rank 1, and 996 triangles of rank 0. The search for minima of target function (11) was performed by the Nelder–Mead method with the following given parameters:  $\alpha = 1$ ,  $\beta = 0.5$ ,  $\gamma = 2$ , and  $\epsilon = 10^{-16}$ . Five triangles of rank 4 yielded three solutions. Six triangles of rank 3 gave five solutions. Of the 37 triangles of rank 2, solutions were found for 31. Obviously, the number of real solutions is less than 39, and it is necessary to discard all duplicates and leave only the one in each case, with the smallest value of the target function. The value of  $10^{-3}$  was taken as the minimum distance between different solutions. After checking it, eight sets of solutions were obtained. The target function takes the minimum values in the solutions presented in Table 2.

After proceeding to ecliptic coordinates on circle (7) and from them to equatorial coordinates, the values of topocentric distances were obtained from (4). The first solution contained a negative value of topo-

No.	$N_{\rm xs}$	$N_{\rm{vs}}$	$J_{\text{goal}}$	No.	$N_{\rm xs}$	$N_{\rm{vs}}$	$J_{\text{goal}}$
	0.39143	0.06996	$2.76 \times 10^{-17}$		0.99008	0.00045	$6.48 \times 10^{-17}$
	$-1.00000$	$-0.05046$	$2.81 \times 10^{-12}$		$-1.00000$	$-0.04986$	$2.81 \times 10^{-12}$
	$-0.17961$	0.11626	$8.26 \times 10^{-18}$		0.62035	$-0.16045$	$1.05 \times 10^{-16}$
	$-0.57975$	$-0.47088$	$6.98 \times 10^{-17}$		1.00000	0.05041	$2.81 \times 10^{-12}$

**Table 2.** Solutions for the comet 2I/Borisov

Here, No. is the set number,  $N_{xx}$  and  $N_{ys}$  are the "square" coordinates of the solution point, and  $f_{goal}$  is the value of the target function at that point.

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Orbit	$T_0/M_0$ , deg	$a$ , au	e	$i$ , deg	$\omega$ , deg	$\Omega$ , deg
2	2019 11 11.56	$-0.711$	2.217	90.000	218.008	272.045
3	108.666	0.782	0.253	12.294	50.101	237.209
4	2019 12 08:52	$-0.851$	3.360	44.044	209.110	308.155
6	2019 11 11.55	$-0.714$	2.213	90.000	217.963	272.021
8	2019 11 11.56	$-0.711$	2.217	90.000	218,005	272.043
<b>MPC</b>	2019 12 08:55	$-0.851$	3.357	44.053	209.127	308.149

**Table 3.** Elements of solutions of the orbits of the comet 2I/Borisov

Here,  $T_0$  is the time of perihelion passage and  $M_0$  is the value of the mean anomaly at the time of the third observation. The second, sixth, and eighth solutions are close hyperbolic orbits perpendicular to the ecliptic plane. The third solution corresponds to an elliptical orbit. The elements of the fourth solution coincide well with the elements of the reference orbit. The O–C values in Table 4 confirm this.

**Table 4.** Representation of the observations of the comet 2I/Borisov by orbit 4

$\Delta \alpha$ , arcsec	$-0.3$	$-1.8$	$-3.3$	
$\Delta \sigma$ , arcsec	70 <i>' . /</i>	0.0	$J \cdot I$	

centric distance  $\rho_1$ , and the fifth and seventh solutions had negative values of orbital parameter *p* and were, therefore, immediately discarded. For the remaining five solutions, the orbital elements are presented in Table 3 and the orbital elements of the comet 2I/Borisov for the epoch 2020 05 31.0 obtained taking into account all perturbations on the MPC website [16] are given as a reference.

The extreme observations at times  $t_1$  and  $t_5$  accurately represent all orbits. However, only the fourth solution represents well the mean observations at times  $t_2$ ,  $t_3$ , and  $t_4$ ; therefore, other solutions can be discarded.

## ADDITIONAL INFORMATION

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## REFERENCES

- 1. V. B. Kuznetsov, "Using the Kuryshev–Perov geometrical method for determination of asteroid and comet orbits," in *Near-Earth Astronomy — 2015: Proc. Int. Conf., Terskol, Russia, Aug. 31 – Sept. 5, 2015* (Moscow, 2015), pp. 51–57.
- 2. A. Cauchy, "Mémoire sur deux formules générales dont chacune permet de calculer rapidement des valeurs très approchées des éléments de l'orbite d'une planète ou d'une comète," C. R. **25**, 953 (1847). http://sites.mathdoc.fr/cgi-bin/oetoc?id=OE\_CAUCHY\_1\_10
- 3. P. Harzer, "Über eine geometrische Methode zur Bahnbestimmung der Bahnen von Himmelskorper aus fünf Beobachtungen," A. N. **184** (1910); Publ. Sterwar., Kiel **12** (1910).
- 4. M. A. Vil'ev, "Studies on the number of solutions to the main problem of theoretical astronomy in connection with its general position at the present time," Tr. AO LGU, Ser. Mat. Nauk (Astron.) **5** (27), 81–256 (1938).
- 5. V. I. Kuryshev and N. I. Perov, "A nontraditional means of determining the orbital elements of unknown cosmic objects from the data of computer processing of survey photographs," Sov. Astron. **26**, 727–729 (1982).
- 6. V. B. Kuznetsov, "Revisiting the determination of a preliminary orbit for a celestial body," Sol. Syst. Res. **53**, 462–472 (2019).

https://doi.org/10.1134/S0038094619060054

- 7. K. Stumpff, *Himmelsmechanik* (Deutscher Verlag, Berlin, 1959), Vol. 1.
- 8. A. J. Sarnecki, "A projective approach to orbit determination from three sight-lines," Celestial Mech. Dyn. Astron. **66**, 425–451 (1996). https://doi.org/10.1007/BF00049380

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- 9. C. Fong, "Analytical methods for squaring the disc," in *Proc. Seoul ICM 2014 Satellite Conference on Imaging, Multi-Scale and High-Contrast Partial Differential Equations, Daejeon, Korea, Aug. 7–9, 2014* (American Mathematical Society, Providence, R.I., 2014).
- 10. V. I. Shalashilin and E. B. Kuznetsov, *Parametric Continuation and Optimal Parametrization in Applied Mathematics and Mechanics* (URSS, Moscow, 2003; Springer, Dordrecht, 2003).
- 11. A. S. Samotokhin and Z. N. Khutorovsky, "Method for initial determination of parameters of near-Earth orbits based on three angular observations," Preprint No. 44, IPM RAN (Keldysh Inst. of Applied Mathematics, Moscow, 2014).
- 12. J. A. Nelder and R. Mead, "Simplex method for function minimization," Comput. J. **7**, 308–313 (1965). https://doi.org/10.1093/comjnl/7.4.308
- 13. D. M. Himmelblau, *Applied Nonlinear Programming* (McGraw-Hill, New York, 1972; Mir, Moscow, 1975).
- 14. V. Potemkin, "Programming language Fortran" (2005). http://fortran90.pvbk.spb.ru/min.html#FM28. Accessed September 10, 2021.
- 15. O. B. Arushanyan, "Scientific and educational Internet resource of the Research Computing Center of Moscow State University on numerical analysis" (2006). http://numanal.srcc.msu.ru/lib\_na/cat/mn/mnb6r.htm. Accessed September 10, 2021.
- 16. International Astronomical Union, "Minor Planet Center 2I/Borisov = C/2019 Q4 (Borisov)" (2020). https://minorplanetcenter.net/db\_search/show\_object?utf8=%E2%9C%93&object\_id=2I. Accessed September 10, 2021.

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