**ORIGINAL PAPER**



# **Study on Torsional Vibration of RV Reducer Based on Time‑Varying Stifness**

 $\mathsf{Song Wang}^1 \cdot \mathsf{Jing Tan}^1 \cdot \mathsf{Jingjun Gu}^1 \cdot \mathsf{Dishan Huang}^1$ 

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## **Abstract**

**Purpose** The torsional vibration of RV reducer is the key factor affecting the positioning accuracy and trajectory accuracy of industrial robots. It is of great signifcance to study the torsional stifness periodic fuctuation and parametric torsional vibration to improve the torsional vibration performance and transmission accuracy of RV reducer.

**Methods** Based on the torsional stifness test device of RV reducer, the torsional stifness is measured and the torsional stifness curve with periodic fuctuation is obtained. The trigonometric function ftting and mathematical description of torsional stifness curve are carried out, and the torsional vibration equation of RV reducer parameters is established. The parametric torsional vibration response of RV reducer is simulated by Runge–Kutta method, and the combined frequency characteristics and system stability are analyzed. The torsional vibration test device of RV reducer is developed to study the torsional stifness fuctuation and torsional vibration fault.

**Results** The torsional stifness fuctuation of RV reducer can be divided into a short period and long period, corresponding to diferent manufacturing errors, and the mechanism of torsional stifness fuctuation is found out. The frequency spectrum of torsional vibration under working condition is highly consistent with the simulated response spectrum, and the harmonic resonance interval obtained by frequency sweeping is close to the peak of the torsional vibration–speed curve, so the correctness of the established torsional vibration model is verifed. Using the mechanism of RV reducer stifness fuctuation to realize rapid diagnosis of defective parts of RV reducer has a strong engineering application value.

**Conclusion** The time-varying torsional stifness of RV reducer will cause the phenomenon of parametric torsional vibration. The vibration response characteristic frequency is a linear combination of internal reference variation frequency and external excitation frequency. The parametric torsional vibration will cause the harmonic resonance of RV reducer and aggravate the torsional vibration. Therefore, it is of great engineering value to study the time-varying torsional stifness and torsional vibration of RV reducer.

**Keywords** RV reducer · Time-varying torsional stifness · Parametric torsional vibration · Vibration spectrum · Property of torsional vibration–speed

# **Introduction**

Industrial robots have been widely used in many felds such as the national defense industry, equipment manufacturing industry, aerospace industry, electronics and electrical industry [[1\]](#page-10-0). As the application of industrial robots more precision, more strict requirements are put forward on the robot's stability. Therefore, the present research task focus

 $\boxtimes$  Dishan Huang hdishan@shu.edu.cn on improving the robot dynamic characteristic of the core driving part namely the RV reducer, to ensure the reliability of products, quality stability, and reduce the impact and torsional vibration in the operation and improve the end of the robot running track and position precision [\[2](#page-10-1), [3](#page-10-2)].

RV reducer is developed on the basis of cycloid-pin planetary reducer. It is a closed two-stage transmission device composed of planetary gear drive and cycloidal-pin drive. RV reducer has a complex and compact structure and high matching accuracy. Compared with the general planetary transmission, it has many advantages such as large deceleration ratio, high transmission efficiency, gentle movement, small return diference and large bearing capacity [[4\]](#page-10-3).



 $1$  Mechanical Engineering and Automation, Shanghai University, Shanghai 200444, China

Based on Hertz formula and Ishikawa formula, Zhang [\[7\]](#page-10-6) established a 5-DOF torsional dynamic model of RV-6AII reducer under load. The concept that torsional stifness changes periodically with time was put forward earlier, which provided a new idea for torsional stifness modeling of RV reducer, but no explicit torsional stifness model was established. Park [\[8\]](#page-10-7) established the semi-analytical torsional stifness model according to the topology characteristics of RV reducer. On the basis of Park's model, Kim [[9\]](#page-11-0) considered the bearing elasticity at the hinge of cycloid gears and crank shafts in the reducer and equivalent the bearing to a fexible ring with a certain stifness. The research results showed that the bearing stiffness is the main factor affecting the torsional stifness of the whole machine. Yang [[10\]](#page-11-1) established the fnite element model of RV reducer including the stifness of bearing, the meshing stifness of gears and the elasticity of each component, and analyzed the infuence of the nonlinear characteristics of each component on the torsional stifness in RV reducer. The literature above mentioned that the torsional stifness changes periodically with time, but all research did not involve the test of periodic torsional stifness on RV reducer, and establishment of a complete theory of time-varying torsional stifness.

The literature above mentioned that the torsional stifness changes periodically with time, but all research did not involve the test of periodic torsional stifness on RV reducer, and establishment of a complete theory of time-varying torsional stifness.

To improve the performance of RV reducer transmission, the investigation of the periodic time-varying stifness of RV reducer and its application will be carried out in this study, and the contents of this manuscript are arranged as follows: (1) the stifness fuctuation of RV reducer based on experimental data; (2) the modeling of parametric vibration with single-parameter and double-parameter; (3) observation of the dynamic characteristics of RV reducer through torsional vibration test; (4) application of torsional stifness fuctuation mechanism of RV reducer.

## **Variable Torsional Stifness**

## **Torsional Stifness Test**

Torsional stifness is expressed by the ratio of the load torque and the corresponding torsion angle of a RV reducer:

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<span id="page-1-1"></span>
$$
K = \frac{\mathrm{d}T}{\mathrm{d}\phi},\tag{1}
$$

where *K-*torsional stifness (unit: N·m/arcmin), *T-*torque (unit: N·m), *𝜙*-torsion angle (unit: arcmin).

Figure [1](#page-1-0) shows the torsional stifness test platform for a RV reducer. The pin shell of the RV reducer 2 is solidly connected to the frame base, and the circumferential motion of the input shaft is restrained by a set screw. The output end is fxedly connected with the torque arm fange 1, and the torque arm fange is fxedly connected with the blocking plate 4. The micrometer gauge 3 is placed in vertical contact with the blocking plate. The pneumatic cylinder 7 is used to apply a tension force perpendicular to the torque arm 5 to generate torque. The tension force is read by the spring tension meter 6.

Three different torque loads (100 N·m, 200 N·m, 300 N·m) are applied to the output end. The torsional angles of the output end corresponding to loads are measured, respectively. Thus, the torsional stifness is obtained using Eq. ([1\)](#page-1-1). Through loosening the set screw, and repeat the above step to obtain a curve of torsional stifness as shown in Fig. [2.](#page-2-0)

RV reducer A, B and C are measured, and Fig. [2a](#page-2-0), b and c are corresponding records of torsional stifness. The ratio of the fuctuation value to average torsional stifness is evaluated as the torsional stifness fuctuation index. The above curves are corresponding to 9.17%, 11.59% and 4.16%, respectively.

<span id="page-1-0"></span>

**Fig. 1** Static torsional stifness test device of RV reducers. 1: Torque arm fange, 2: RV reducer, 3: microm gauge, 4: blocking plate, 5: torque arm, 6: spring tension meter, 7: pneumatic cylinder



<span id="page-2-0"></span>**Fig. 2** Torsional stifness curves measured from three samples of RV reducer

## **Variable Torsional Stifness Description**

The experimental data of the torsional stifness in Fig. [2a](#page-2-0) can be approximately equivalent to the linear superposition of two harmonic functions. As shown in Fig. [3](#page-2-1), the low-frequency part is 1.99 periodic wave, namely long-period fuctuation, and the high-frequency part is 40.58 periodic wave, namely shortperiod fuctuation. The long-period fuctuations also exist in Fig. [2b](#page-2-0) and c, but this phenomenon is not obvious, while the short-period torsional stifness fuctuations are all obvious.

## **Mechanism of Variable Torsional Stifness**

## **RV Reducer Transmission Structure**

Take the RV-40E as the example, it is a two-stage closed planetary gear train composed of the frst stage planetary



<span id="page-2-1"></span>**Fig. 3** Torsional stifness ftting curve of RV reducer

transmission and the second stage cycloid transmission. The high-speed class is a K-H diferential gear train consisting of the sun gear 1, two symmetrical planet gears 2 and the tie bar; The low-speed class is a K-H-V planetary gear train consisting of two crankshafts 3, two cycloid gears 4, pin teeth 5 and output shaft 6. The tie bar and the output shaft are fxed as the same part (Fig. [4](#page-3-0)).

The parameters of the RV reducer: the number of sun gear teeth  $Z_1$  is 12; the number of planet gear teeth  $Z_2$  36; the number of cycloid gear teeth  $Z_4$  39; and the number of pin teeth  $Z_5$  40.

First-order transmission ratio:

$$
i_{12} = \frac{n_1}{n_2} = -\frac{Z_1 + Z_5 Z_2}{Z_1 Z_4} \tag{2}
$$

Second order transmission ratio:





#### <span id="page-3-0"></span>**Fig. 4** RV reducer structure



(a) Structure sketch of RV reducer

$$
i_{45}^3 = \frac{n_4 - n_3}{n_5 - n_3} = 1 - \frac{n_4}{n_3} = \frac{Z_5}{Z_4}
$$
 (3)

The total transmission ratio of RV reducer:

$$
i = \frac{n_1}{n_6} = 1 + \frac{Z_2 Z_5}{Z_1} \tag{4}
$$

According to the structure of the RV reducer, the conditions of  $n_2 = n_3$  and  $n_4 = n_6$  are obtained. The transmission ratio could be calculated as:  $i_{12} = \frac{121}{39}$ ,  $i_{34} = 39$ .

### **Short‑Period Fluctuation**

In the practice, the larger the magnitude of torsional stifness fuctuation is, the worse the dynamic performance of RV reducer will be. For example, the torsional stifness fuctuation reduces transmission accuracy, and induces torsional vibration and so on. Therefore, it is necessary to consider what causes the torsional stifness fuctuation to increase, so as to reduce the torsional stifness fuctuation by optimizing the design and manufacturing process.

#### *(a) Cycloid transmission uniformity*

The fuctuation of the torsional stifness is in the same period as the fuctuation of the meshing stifness in the gear system. Therefore, the theoretical frequency of short-period torsional stifness fuctuation is the cycloid transmission ratio 39, which is close to fuctuation frequency 40.58 in the ftted curve in Fig. [3a](#page-2-1).

The working tooth profle of cycloid gears meshes with the pin teeth to produce the torque to bear the load. The uniformity of cycloid transmission is mainly related to the tooth profle deviation of cycloid gears and the diameter error of pin teeth. ZEISS CMM is used to measure and calculate the dimensional error of parts in RV reducer A and C.

The accuracy of the left and right tooth profles of the cycloid gears is measured, the tooth profle deviation is obtained, as shown in Fig. [5](#page-3-1).



(b) 3D model of RV reducer structure



**(a)** Left tooth profile deviation



**(b)** Right tooth profile deviation

<span id="page-3-1"></span>**Fig. 5** Tooth profle deviation distribution of cycloid gears

The deviation fuctuation can be obtained by calculating the diference between all tooth profle deviation and the one that is closest to the design datum, which can refect the uniformity of the tooth profle error of the cycloid gears.

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Through measurement and statistics, it is concluded that the mean value of deviation fuctuation is 0.872 μm at the cycloid gear's left tooth profle in the RV reducer A, and the mean value of deviation fuctuation is 1.433 μm at the right tooth profle; The mean value of deviation fuctuation is 0.692 μm at the cycloid gear's left tooth profle in the RV reducer C, and the mean value of deviation fuctuation is 0.682 μm at the right tooth profle. The uniformity of tooth profle of the cycloid gear in RV reducer C is obviously better than that in RV reducer A. Better tooth profle uniformity can improve the smoothness of cycloid gear meshing process and reduce torsional stifness fuctuation.

At the same time, the mean value of the diameter fuctuation is 2.1 μm at the pin teeth in RV reducer A, and that in RV reducer C is  $0.25 \mu m$ . The diameter fluctuation of pin teeth in RV reducer A is about  $8 \sim 10$  times of that in reducer C, which will seriously afect the meshing smoothness of cycloid transmission.

*(b) The gap between cycloid gears and pin teeth*



<span id="page-4-0"></span>**Fig. 6** Torsional stifness fuctuation related with diferent pin teeth diameters

<span id="page-4-1"></span>**Fig. 7** Eccentricity of the sungear



B

(a) Initial state

In the study, by changing the diameter of the pin teeth to change the ft gap between cycloid gears and pin teeth, the fuctuation magnitude of torsional stifness can be improved. The torsional stifness curve under diferent diameter tolerances of the pin teeth are shown in Fig. [6.](#page-4-0) Thus, it can be concluded that the gap between cycloid gears and pin teeth decreases, and the fuctuation of torsional stifness decreases.

However, the diameter of the pin teeth should not be increased too much, otherwise the clearance between the cycloid gears and the pin teeth is too small, it results increment of meshing wear and reduction of transmission efficiency, and even lead to the whole machine jammed. Therefore, to improve the torsional stifness performance of RV reducer using this method, it is necessary to ensure that the transmission efficiency is not affected.

#### **Long‑Period Fluctuation**

As shown in Fig. [3](#page-2-1)b, The period of torsional stifness fuctuation is approximately 2. The direct reason of long-period fuctuation is the eccentricity of the cycloid gears, which causes the loose-tight meshing with the pin teeth. There are many reasons leading to the eccentricity of the cycloid gears, which can be summarized as follows:

The eccentricity of the sun gear produces extrusion pressure  $F$  on the planet gear, as shown in Fig.  $7$ , which causes the center of the double crankshafts to deviate from the original rotating center. The eccentricity of the double crankshafts is shown in Fig. [8](#page-5-0). The double crankshafts drive the cycloid gears directly, and its eccentricity causes the cycloid gears the loose-tight meshing with the pin teeth. As shown in Fig. [9](#page-5-1)a, the initial state is that a cycloid gear is closely engaged with the pin teeth, and the dotted line is used to represent the over-tight meshing. When the cycloid gear rotates for half a cycle, Fig. [9](#page-5-1)b shows that another cycloid gear is tightly engaged with the pin teeth. Thus, the torsional stifness fuctuates with a period of 2.



(b) Planet gears running half a circle





<span id="page-5-0"></span>**Fig. 8** Eccentricity of the double crankshafts



(a) Cycloid gears running half a circle

<span id="page-5-1"></span>**Fig. 9** Eccentricity of the double cycloid gears

# **Parametric Vibration Caused by Time‑Varying Torsional Stifness**

## **Parametric Vibration**

When the RV reducer works, the torsional stifness with position periodicity is converted to periodic time-varying torsional stifness, forming the system parametric torsional vibration. Parametric vibration is a forced vibration response generated by external excitation due to the periodic change of parameters in the system [\[11,](#page-11-2) [12\]](#page-11-3).

Taking the test result of torsional stifness of RV reducer A as an example, the time-varying torsional stifness function is mathematically described as:

$$
K(t) = K_0 \left( 1 + \beta_1 \cos \omega_{o1} t + \beta_2 \cos \omega_{o2} t \right)
$$
 (5)

where  $K_0$  is the average torsional stiffness,  $\omega_{01}$  the frequency of short-period fluctuation,  $ω<sub>02</sub>$  the frequency of long-period fluctuation,  $\beta_1$  the short-period torsional



stiffness fluctuation index and  $\beta_2$  the long-period the torsional stifness fuctuation index.

External excitation torque includes drive torque excitation and load torque excitation [\[13\]](#page-11-4), which can be described as:

$$
M(t) = A \cos \omega_{p1} t + B \cos \omega_{p2} t \tag{6}
$$

Thus, the parametric torsional vibration equation is:

$$
J \cdot \ddot{\theta}(t) + C \cdot \dot{\theta}(t) + K(t) \cdot \theta(t) = M(t)
$$
\n(7)

where *J* is the moment of inertia of the system, *C* is the damping,  $\theta(t)$  is the angular displacement, and  $M(t)$  is the torque.

## **Parametric Vibration Response and Spectrum Analysis**

The torsional parametric vibration equation under the harmonic excitation:

$$
\frac{d^2\theta}{dt^2} + 2\zeta\omega_n \frac{d\theta}{dt} + \omega_n^2 \theta (1 + \beta_1 \cos \omega_{o1} t + \beta_2 \cos \omega_{o2} t) = A \cos \omega_{p1} t + B \cos \omega_{p2} t
$$
\n(8)

where  $\zeta$  is the damping ratio,  $\omega_n$  is the torsional natural frequency. Using frequency response measurement, the frst natural frequency of system, with a symmetric moment of inertia of 56.65 kg·m<sup>2</sup>, is estimated as  $19\pi$ , and the damping ratio is 0.025. When the driving speed is 1400 rpm, the driving frequency is  $\omega_{p1}$  = 46.66 $\pi$ , and the load torque excitation frequency is  $\omega_{p2}$  = 0.386 $\pi$ ; The amplitude of driving excitation A is 1000, and the amplitude of load excitation B is 1000; Meanwhile, the indexes and coefficients in the parametric equations are determined as,  $\omega_{01} = 15.04\pi$ ,  $\omega_{02}$ =0.772 $\pi$ ,  $\beta_1$ =0.1172 and  $\beta_2$ =0.0764.

The response of the parametric torsional vibration is solved by Runge–Kutta method, and Fig. [10](#page-6-0) shows the time history of steady-state response.

The time-history of the simulation response is obtained at the speed of 1400 rpm, and the corresponding spectrum is obtained, as shown in Fig. [11](#page-6-1). Its characteristics are as follows: there is a complex band around the primary frequency [[14\]](#page-11-5), which is caused by the torsional stiffness fluctuation. Therefore, the interval between the primary frequency and the band frequency is basically the same, and corresponds to the harmonic combination of  $\omega_{01} = 15.04\pi$  (7.02 Hz) and  $\omega_{02}$ =0.772 $\pi$  (0.386 Hz).

#### **Stability Analysis**

The response of the parametric system is sometimes weak, but sometimes stronger resonance phenomenon may occur, which depends on the stability of the parametric system [[15](#page-11-6)].



<span id="page-6-0"></span>**Fig. 10** Time-history of vibration response



6000

4000

Driving speed/rpm

<span id="page-6-2"></span>**Fig. 12** Instability area under single-parameter excitation  $\omega_{\text{o}1}$ 

2000

 $0.5$ 

New Sole<br>
Manufacture<br>
Manufacture

 $\mathbf 0$  $\sigma$ 



<span id="page-6-3"></span>**Fig. 13** Instability area under single-parameter excitation  $\omega_{02}$ 



<span id="page-6-1"></span>**Fig. 11** Spectrum of vibration response

<span id="page-6-4"></span>**Fig. 14** Instability area under double-parameter excitation

The critical frequency of instability is solved by Fourier series method, and the instability area of torsional vibration is obtained.

The instability diagrams of the parametric system with the excitation parameters  $\omega_{01}$  or  $\omega_{02}$  is given, as shown in Fig. [12](#page-6-2) and Fig. [13.](#page-6-3) where the *x*-coordinate represents the driving speed of the motor. It can well refect the instability area of torsional vibration of the RV reducer system.

The parametric vibration with double-parameter excitation would superimpose the instability area of singleparameter excitation and generate the coupling instability area, as shown in Fig. [14](#page-6-4). Because of the action of nonlinear coupling, the instability area of the parametric vibration with double-parameter excitation is greater than the sum of the single-parameter excitation.

Numerical simulation of single-parameter and doubleparameter torsional vibration responses are obtained by frequency sweeping. The vibration response and its envelope are shown in Figs. [15](#page-7-0) and [16](#page-7-1). It should be noted that the amplitude of vibration response with double-parameter excitation is nearly four times that with single-parameter excitation, and the magnitude of vibration response caused



8000

<span id="page-7-0"></span>

by double-parameter excitation is much larger than singleparameter excitation.

# **Experiment**

<span id="page-7-1"></span>tion and its envelope

To investigate the time-varying stifness afecting torsional vibration characteristics under actual working conditions, a torsional vibration test device is built for RV reducer. The device is composed of a servo motor, a RV-40E reducer, a test frame, an inertial load, a wireless acceleration sensor and an electric control box, etc., as shown in Fig. [17](#page-7-2). When the servo motor is driving, the inertial load arm rotates with the output end of RV reducer. The wireless acceleration sensor is installed on the inertial load arm 550 mm away from the rotating center to pick up the circumferential vibration, which can be considered as the torsional vibration [[16\]](#page-11-7).



(a) Schematic diagram of test device



(b) Picture of test device

<span id="page-7-2"></span>**Fig. 17** Sketch of RV torsional vibration automatic test device





<span id="page-8-0"></span>**Fig. 18** Torsional vibration–speed characteristic of RV reducer (number A)



<span id="page-8-1"></span>**Fig. 19** Time record of torsional vibration

To verify the double-period stifness fuctuation characteristics of the parametric torsional vibration model, the test is carried out on RV reducer A. The torsional vibration-speed characteristic curves is shown as Fig. [18,](#page-8-0) it can be found the peak of the curve is very close to the simulation (Fig. [16b](#page-7-1)).

When the driving speed is 1400 rpm, the time record of torsional vibration (Fig. [19](#page-8-1)) is collected and the spectrum is analyzed. As shown in Fig. [20,](#page-8-2) there is an obvious band in the spectrum of torsional vibration, which is due to the combination of internal and external excitation frequencies of the torsional vibration, namely the linear combination of the torsional stifness fuctuation frequency 30

40

<span id="page-8-2"></span>**Fig. 20** Band in the spectrum of torsional vibration

 $10$ 

15.12Hz

7.753Hz

0.2079Hz

 $10<sup>C</sup>$ 

 $10^{-2}$ 

 $10$ 

 $\overline{0}$ 

Acceleration/G



20

Frequency/Hz

<span id="page-8-3"></span>**Fig. 21** Torsional vibration-speed curve of the RV reducer

 $(f<sub>o1</sub>=7.02hz, f<sub>o2</sub>=0.386hz)$  and the external excitation frequency  $(f_{p1}=23.33\text{hz}, f_{p2}=0.193\text{hz})$ . Compared with the frequency spectrum of simulation (Fig. [11\)](#page-6-1), there is good consistency between their features, and it can be concluded that the band frequencies in the spectrum are all linear combinations of the above excitation frequencies.

# **Application**

The torsional vibration performance curve, i.e. the property of torsional vibration-speed, of an RV reducer is shown in Fig. [21.](#page-8-3) When the driving speed is 2000 rpm, the torsional





<span id="page-9-0"></span>**Fig. 22** Time record of torsional vibration



<span id="page-9-1"></span>**Fig. 23** Frequency spectrum of torsional vibration

vibration displacement reaches the maximum value and exceeds the displacement standard value, which is 0.1 mm.

To analyse the dynamic characteristic of RV reducer, the vibration data is collected at the speed of 2000 rpm, as shown in Fig. [22.](#page-9-0) Spectrum analysis on vibration is carried out, as shown in Fig. [23.](#page-9-1) It is found that torsional vibration includes two parts. The low-frequency vibration is based on the frequency bands of 11.07 Hz, and a high-frequency vibration is a group of spectra whose center is 133.3 Hz.

The cycloid meshing error in RV reducer transmission is an exciting source of vibration, which will cause lowfrequency vibration corresponding to short-period stifness fuctuation. It is basically consistent with 11.07 Hz in the



<span id="page-9-2"></span>**Fig. 24** STFT analysis of torsional vibration



<span id="page-9-3"></span>**Fig. 25** Envelope spectrum of torsional vibration

spectrum. Because the short-period stifness fuctuation index is relative high, it leads to the parametric resonance in the torsional vibration of RV reducer at this speed.

The frequency of shock vibration response is relative high, and it is located the high-order modal frequency (the 3rd modal frequency) of RV reducer system. Its wave and spectrum can not be used to diagnose defects directly, so special signal processing is carried out on the torsional vibration.

STFT [[17](#page-11-8)] analysis of the torsional vibration can be used to directly observe the fuctuation in the vibration wave, as shown in Fig. [24](#page-9-2). However, no exact characteristic can be evaluated from the fgure.

To extract the defect information from the shock response, the envelope analysis is carried out on the vibration response after band fltering. The envelope spectrum is shown in Fig. [25](#page-9-3), where the loading torque excitation frequency of 0.279 Hz and short-period stifness fuctuation



<span id="page-10-8"></span>**Fig. 26** Torsional vibration–speed curve of the RV reducer (After replacing cycloid gears and crankshafts)

frequency of 11.07 Hz can be obtained. It is concluded that the high-frequency vibration is caused by the joint action of load torque excitation and short-period stifness fuctuation excitation, so the high-frequency vibration is also related to short-period stifness fuctuation.

On the above analysis, the dimensions of cycloid gears, crankshafts and pin teeth are measured by a ZEISS Calypso three coordinates measuring instrument. It is found that the tooth profle error in the cycloid gears is beyond 60% of the normal tolerance, the eccentricity of crankshafts exceeds 50% of the normal tolerance. After replacing the crankshafts and cycloid gears in the RV reducer, the torsional vibration test is repeated. Experimental result shows that the reassembly is successful. It can be seen that the displacement value and acceleration value of torsional vibration is signifcantly reduced to one half of the original one through comparing Fig. [26](#page-10-8) with Fig. [21.](#page-8-3)

## **Conclusion**

The time-varying torsional stifness of RV reducer is divided into short-period fuctuation and long-period fuctuation. The short-period torsional stifness fuctuation is caused by cycloid gears meshing. The long-period torsional stifness fuctuation is caused by the assembly error of the sun gear, crankshafts and cycloid gears.

According to the time-varying stifness of RV reducer, the parametric torsional vibration equation is established and the dynamic characteristics are analyzed. The correctness of the dynamic model and the mathematical description of time-varying torsional stifness are verifed by the

By applying the mechanism of stifness fuctuation of RV reducer, the defective parts of RV reducer can be quickly diagnosed on line. By replacing defective parts, the torsional vibration of RV reducer is reduced to one half of the original one and its performance is greatly improved. In addition, the correctness of characteristics of torsional stifness fuctuation is also proved.

There is a correlation between stifness fuctuation and torsional vibration characteristics in RV reducer transmission. The time-varying torsional stifness of the reducer will cause the phenomenon of parametric torsional vibration of the reducer, which will cause harmonic resonance and aggravate the torsional vibration of the RV reducer, so the research on time-varying torsional stifness has practical engineering valuable. In future research, more work will be done on modifying the profle of cycloid gears and solving the optimal clearance to reduce the torsional stifness fuctuation, which will have great practical signifcance.

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