



Output Feedback Robust Fault-Tolerant Control of Interval Type-2 Fuzzy Fractional Order Systems With Actuator Faults

Kaijing Jin¹ · Jin-Xi Zhang² · Xuefeng Zhang³

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Abstract This paper is concerned with the stabilization problem for a class of uncertain nonlinear fractional order systems described by an interval type-2 fuzzy model, under actuator faults. To solve the problem, a robust fault-tolerant control (FTC) scheme composed mainly of an augmented non-fragile observer and a new-type H_∞ controller is developed. The resulting control system is with the following advantages. On the one hand, the system stability domain is extended significantly, attributed to introducing the concept of D-stability to the control design and stability analysis, instead of the conventional indirect Lyapunov theory. Theoretically, the stability domain can be expanded from the original half-plane to nearly the overall plane. On the other hand, the control system is robust against the measurement noise and external disturbances which however are not taken into account in the related works. This is achieved by adopting the H_∞ control method in a novel way, in which a new technical lemma is presented to solve real linear matrix inequalities (LMIs). In this way, the robust FTC design is also simplified, wherein the number

of the required decision variables is reduced from four to two. Besides, the control design is less restrictive: three common requirements for the system matrix or the control gain selection are eliminated. Finally, the simulation results on an electrical circuit system and a numeral example both illustrate the above theoretical findings.

Keywords Fractional order systems · Interval type-2 fuzzy systems · Fault-tolerant control · H_∞ performance index

1 Introduction

Fractional order systems (FOSs) are general with respect to integer order systems (IOSs) which are a special case of FOSs [1–3]. On the other hand, many physical systems or processes can be described better by fractional derivatives and integrals such as the viscoelasticity of polymer materials [4], the fractional order electrical circuit systems [5], and signal processing [6]. Therefore, in the past several decades, FOSs have attracted great attention from the control field. The stability analysis of FOSs was first investigated [7–12]. A stability criterion for the FOSs was provided [7] by examining the locations of the system matrix eigenvalues. Some stability conditions were further given [8] by solving the LMIs with complex variables, and the approaches to transforming the complex variables to the real variables were developed [9, 10] for ease of solution. To this end, some brief theorems without LMIs were further established [11, 12] that simplify the stability analysis significantly. On the other hand, the observer and controller designs were carried out [13–20]. A non-fragile observer was constructed [13]; some output-feedback controllers based on the iterative algorithms were designed [14, 15]. To simplify the design and computational

✉ Jin-Xi Zhang
zhangjx@mail.neu.edu.cn

Kaijing Jin
jinkaijing@stumail.neu.edu.cn

Xuefeng Zhang
zhangxuefeng@mail.neu.edu.cn

¹ The College of Information Science and Engineering, Northeastern University, Shenyang 110819, Liaoning, China

² The State Key Laboratory of Synthetical Automation for Process Industries, Northeastern University, Shenyang 110819, China

³ The College of Sciences, Northeastern University, Shenyang 110819, Liaoning, China

complexity, the LMI tool [16]–[20] or the singular value decomposition technique [16] was skillfully adopted. For nonlinear FOSs, the Takagi-Sugeno (T-S) fuzzy model was employed in the control design [17–19]. And in the presence of model uncertainties, the sliding mode control method was applied to the T-S fuzzy singular FOSs [20]. To further enhance the robustness of the control systems, the H_∞ control method was applied to the FOSs successfully [21–23].

It should be noted that the above-mentioned results are established in the fault-free case. With the fast development of modern industry and information technology, many control systems are increasingly complex. As a result, the system is more likely to meet faults. Faults lead to unfavorable effects on the performance and/or stability of the control system, and even cause disastrous consequences. Therefore, the control system with tolerance to unexpected faults, especially to the actuator faults that change the control action straightforward, is of great significance. For this purpose, a variety of fault-tolerant control (FTC) approaches were proposed in the literature. In general, an observer is designed to estimate the faults for compensation. For example, under the assumption of known fault bounds, a sliding mode observer was constructed [24]. To avoid the difficulty in solving the bilinear matrix inequalities, the fault observer and the controller were designed separately [25, 26]. However, the comparative result between the integrated and separate designs [27] reveals that the former achieves better estimation and control performance. On the other hand, when the full system state is available, some FTC schemes were developed [28, 29] with respect to the energy-bounded faults. Further, an output-feedback FTC strategy was proposed [30]. To the FTC problem for FOSs, some solutions were also reported in recent years [31–34]. The model uncertainty was taken into consideration [31, 32], yielding some robust FTC laws. Under the assumption that the actuator faults are n -order differentiable, an augmented fault observer was developed [33]. The H_∞ control method and the dynamic output-feedback control design method were combined to deal with the actuator faults [34].

Motivated by the above observation, this paper presents a novel output-feedback FTC strategy for a class of FOSs with actuator faults. Its advantages are as follows.

- (1) The stability domain of the faulty FOS is extended by our approach. In the existing literature on FTC of FOSs [32], the control design and analysis are based mainly on the indirect Lyapunov theory, that is, the designers follow the FTC method for IOSs. This yields the system stability domain covering only the left half of the plane. Instead of the indirect Lyapunov theory, the D-stability-based analysis

approach in which the nonconvex property of the FOS is taken into account fully [8, 10], is adopted in this paper. In this way, the system stability domain is extended significantly.

- (2) The proposed approach of fault estimation and FTC is robust against the measurement noise and external disturbances which however are not considered in the existing FTC designs for FOSs [31, 32]. The robustness is achieved by adopting the H_∞ control method. Moreover, we evade the calculation of complex matrix inequalities [21]–[23], and the controller parameters are determined by solving real LMIs. This reduces the design complexity of the fault observer and the controller.
- (3) In addition, the proposed output-feedback control design for FOSs is less restrictive than the existing ones. The output matrix is required to be of full row rank or same [13, 16] and [28]; certain matrices in the intermediate steps of the control design need to be in the block diagonal form [13]. However, this paper is without the above requirements.

Notations: In this paper, \mathbb{R} , \mathbb{R}^n , $\mathbb{R}^{n \times m}$ and $\mathbb{C}^{n \times m}$ denote the real number field, n -dimensional Euclidean space, the set of all $n \times m$ real matrices, and the set of all $n \times m$ complex matrices, respectively. $\text{Re}(P)$ and $\text{Im}(P)$ mean the real and image parts of complex matrix P , respectively. j is an imaginary unit. $X > 0$ (< 0) demonstrates that X is a positive (negative) definite matrix. X^T and X^{-1} stand for the transpose and the inverse of X , respectively. The symbols $\text{sym}\{Y\}$ and \star represent $Y + Y^T$ and the transpose in the symmetric positions of a matrix, respectively. I_n stands for an identity matrix with n dimensions. For brevity, denote $a = \sin(\frac{\alpha\pi}{2})$ and $b = \cos(\frac{\alpha\pi}{2})$ in the sequel. A matrix is assumed to have appropriate dimensions to be compatible for algebraic calculus without specially statement.

2 System Description and Preliminaries

2.1 System Description

Different from the classical T-S fuzzy model [17–20, 35, 36], the type-2 fuzzy model is an effective way to describe the system with model uncertainties [37–40]. In an integer order type-2 fuzzy model, its i th fuzzy rule is:

Plant rule i : IF $h_1(x(t))$ is H_{i1} and \dots and $h_p(x(t))$ is H_{ip} , THEN

$$\begin{aligned} \dot{x}(t) &= A_i x(t) + B_i u(t) + \tilde{G}_i \omega_1(t), \\ y(t) &= C_i x(t) + \tilde{D}_i \omega_2(t), \end{aligned} \tag{1}$$

for $i = 1, 2, \dots, r$ with r the number of the IF-THEN rules, where H_{ig} and $h_g(x(t))$ stand for the fuzzy set and the premise variable, respectively, $g = 1, 2, \dots, p$; $x(t) \in \mathbb{R}^n$, $u(t) \in \mathbb{R}^m$, $\omega_1(t) \in \mathbb{R}^n$ and $\omega_2(t) \in \mathbb{R}^s$ are the state, control input, external disturbance and measurement uncertainty, respectively; $y(t) \in \mathbb{R}^s$ is the measured output; $A_i, B_i, \tilde{G}_i, C_i$ and \tilde{D}_i are the system matrices. Accordingly, the overall fuzzy system is

$$\begin{aligned} \dot{x}(t) &= \sum_{i=1}^r \theta_i(x(t))(A_i x(t) + B_i u(t) + \tilde{G}_i \omega_1(t)), \\ y(t) &= \sum_{i=1}^r \theta_i(x(t))(C_i x(t) + \tilde{D}_i \omega_2(t)), \end{aligned} \tag{2}$$

where $\theta_i(x(t))$ are the grades of membership:

$$\theta_i(x(t)) = \underline{\lambda}_i(x(t)) \underline{\theta}_i(x(t)) + \bar{\lambda}_i(x(t)) \bar{\theta}_i(x(t)), \tag{3}$$

where $\underline{\lambda}_i(x(t))$ and $\bar{\lambda}_i(x(t))$ are adjustable nonlinear functions according to the change of uncertain parameters, which meet

$$\begin{aligned} 0 \leq \underline{\lambda}_i(x(t)) \leq 1, \\ 0 \leq \bar{\lambda}_i(x(t)) \leq 1, \end{aligned}$$

$$\underline{\lambda}_i(x(t)) + \bar{\lambda}_i(x(t)) = 1, \quad i = 1, \dots, r.$$

And $\underline{\theta}_i(x(t))$ and $\bar{\theta}_i(x(t))$ are the lower and upper bounds of the membership functions of the i th rule, respectively, with

$$\begin{aligned} \underline{\theta}_i(x(t)) &= \prod_{g=1}^p \underline{\mu}_{H_{ig}}(h_g(x(t))), \\ \bar{\theta}_i(x(t)) &= \prod_{g=1}^p \bar{\mu}_{H_{ig}}(h_g(x(t))), \end{aligned}$$

where $\underline{\mu}_{H_{ig}}(h_g(x(t)))$ and $\bar{\mu}_{H_{ig}}(h_g(x(t)))$ denote the lower and upper membership of $h_g(x(t))$ in H_{ig} , respectively, meeting

$$\begin{aligned} \bar{\mu}_{H_{ig}}(h_g(x(t))) \geq \underline{\mu}_{H_{ig}}(h_g(x(t))) \geq 0, \quad i = 1, 2, \dots, r, \\ \bar{\theta}_i(x(t)) \geq \underline{\theta}_i(x(t)) \geq 0. \end{aligned}$$

For simplification of notation, let $\theta_i = \theta_i(x(t))$ and $\eta_i = \eta_i(x(t))$. Define

$$\begin{aligned} A(\theta) &= \sum_{i=1}^r \theta_i A_i, \quad B(\theta) = \sum_{i=1}^r \theta_i B_i, \quad \tilde{G}(\theta) = \sum_{i=1}^r \theta_i \tilde{G}_i, \\ C(\theta) &= \sum_{i=1}^r \theta_i C_i, \quad \tilde{D}(\theta) = \sum_{i=1}^r \theta_i \tilde{D}_i. \end{aligned}$$

Then, (2) becomes

$$\begin{aligned} \dot{x}(t) &= A(\theta)x(t) + B(\theta)u(t) + \tilde{G}(\theta)\omega_1(t), \\ y(t) &= C(\theta)x(t) + \tilde{D}(\theta)\omega_2(t). \end{aligned} \tag{4}$$

Now we consider a fractional order type-2 fuzzy model. Usually, the Caputo definition is adopted to describe the FOSs.

Definition 1 The α order Caputo derivative of the function $g(t)$ is defined as [2]

$$\mathcal{D}^\alpha g(t) = \frac{1}{\Gamma(m-\alpha)} \int_0^t \frac{g^{(m)}(\tau)}{(t-\tau)^{\alpha+1-m}} d\tau,$$

with $m-1 < \alpha < m$ for an integer m , and

$$\Gamma(s) = \int_0^\infty t^{s-1} e^{-t} dt.$$

Following (1), the type-2 fuzzy fractional order model is written as:

Plant rule i : IF $h_1(x(t))$ is H_{i1} and \dots and $h_p(x(t))$ is H_{ip} , THEN

$$\begin{aligned} \mathcal{D}^\alpha x(t) &= A_i x(t) + B_i u(t) + \tilde{G}_i \omega_1(t), \\ y(t) &= C_i x(t) + \tilde{D}_i \omega_2(t). \end{aligned} \tag{5}$$

As usual, we discuss the fixed order α with $0 < \alpha < 1$ [9–14]. The overall fuzzy FOS is

$$\begin{aligned} \mathcal{D}^\alpha x(t) &= \sum_{i=1}^r \theta_i(x(t))(A_i x(t) + B_i u(t) + \tilde{G}_i \omega_1(t)), \\ y(t) &= \sum_{i=1}^r \theta_i(x(t))(C_i x(t) + \tilde{D}_i \omega_2(t)). \end{aligned} \tag{6}$$

The possible actuator faults are taken into account: $u_f(t) = u(t) + f(t)$. The faulty system is thus:

$$\begin{aligned} \mathcal{D}^\alpha x(t) &= \sum_{i=1}^r \theta_i(x(t))(A_i x(t) + B_i u_f(t) + \tilde{G}_i \omega_1(t)), \\ y(t) &= \sum_{i=1}^r \theta_i(x(t))(C_i x(t) + \tilde{D}_i \omega_2(t)). \end{aligned} \tag{7}$$

Assumption 1 $f(t)$ belongs to $L_2[0, \infty)$ [30].

Remark 1 Besides Assumption 1, it is usually assumed in the related works on fault estimation that $f(t) \in L_2[0, \infty)$ [41] or that the fault bound is known [24]. In this paper, these requirements are not involved.

For ease of exposition, rewrite (7) in the following form:

$$\mathcal{D}^\alpha x(t) = \sum_{i=1}^r \theta_i(x(t))(A_i x(t) + B_i u_f(t) + G_i \omega(t)), \tag{8}$$

$$y(t) = \sum_{i=1}^r \theta_i(x(t))(C_i x(t) + D_i \omega(t)),$$

where $\omega(t) = \begin{bmatrix} \omega_1(t) \\ \omega_2(t) \end{bmatrix}$, $G_i = \tilde{G}_i [I \ 0]$ and $\tilde{D}_i [0 \ I]$.

The control objective for the system in (8) is stabilization, even in the presence of actuator faults.

2.2 Preliminaries

Consider the following FOS with $0 < \alpha < 1$,

$$\begin{aligned} \mathcal{D}^\alpha x(t) &= Ax(t) + G\omega(t), \\ y(t) &= Cx(t) + D\omega(t). \end{aligned} \tag{9}$$

In the classical H_∞ control designs for (9) [21, 22], the following lemma is usually applied to solving the linear complex matrix inequalities.

Lemma 1 *The FOS in (9) meets $\|y(t)\|_2 < \gamma \|\omega(t)\|_2$ if there exists a complex matrix P such that*

$$P = X + Yj > 0, \tag{10}$$

$$\begin{bmatrix} \text{sym}\{A(aX - bY)\} & \star & \star \\ C(aX - bY) & -\gamma I_p & \star \\ G^T & D^T & -\gamma I_q \end{bmatrix} < 0. \tag{11}$$

Lemma 1 yields complex solutions. In general, it is nontrivial to solve complex matrix inequalities; for example, the complex matrix inequality is unsolvable by the widely used mathematical tool MATLAB. This is because the complex matrix should be mapped to a real matrix for calculation. Motivated by this, a new lemma is established in this paper, presented as follows.

Lemma 2 *The FOS in (9) meets $\|y(t)\|_2 < \gamma \|\omega(t)\|_2$ if there exist a symmetric matrix X and a skew-symmetric matrix Y such that*

$$\begin{bmatrix} X & Y \\ -Y & X \end{bmatrix} > 0, \tag{12}$$

$$\begin{bmatrix} \text{sym}\{A^T(aX - bY)\} & \star & \star \\ G^T(aX - bY) & -\gamma I_p & \star \\ C & D & -\gamma I_q \end{bmatrix} < 0. \tag{13}$$

Proof Note that $\|G_{wz}(s)\| = \|C(s^\alpha I - A)^{-1}G + D\| < \gamma$ is equivalent to $\|G_{wz}^T(s)\| = \|G^T(s^\alpha I - A^T)^{-1}C^T + D^T\| < \gamma$. This yields (13), straightforward. Next, we show that $P = X + Yj > 0$ if

$$\begin{bmatrix} \text{Re}(P) & \text{Im}(P) \\ -\text{Im}(P) & \text{Re}(P) \end{bmatrix} > 0. \tag{14}$$

Under (14), there is $\text{Im}(P)^T = -\text{Im}(P)$. For any $x, y \in \mathbb{R}^n$, there thus hold $x^T \text{Im}(P)x = 0$ and $y^T \text{Im}(P)y = 0$. Calculating the real part of

$$\Sigma = (x^T - y^T j)(\text{Re}(P) + \text{Im}(P)j)(x + yj),$$

one has

$$\begin{aligned} &\text{Re}((x^T - y^T j)(\text{Re}(P) + \text{Im}(P)j)(x + yj)) \\ &= x^T \text{Re}(P)x + y^T \text{Re}(P)y - x^T \text{Im}(P)y + y^T \text{Im}(P)x \\ &= \begin{bmatrix} x^T & y^T \end{bmatrix} \begin{bmatrix} \text{Re}(P) & -\text{Im}(P) \\ \text{Im}(P) & \text{Re}(P) \end{bmatrix} \begin{bmatrix} x \\ y \end{bmatrix}. \end{aligned}$$

From (14), we have $\text{Re}(\Sigma) > 0$. Calculating the imaginary part of Σ , we have

$$\begin{aligned} &\text{Im}((x^T - y^T j)(\text{Re}(P) + \text{Im}(P)j)(x + yj)) \\ &= x^T \text{Re}(P)y - y^T \text{Re}(P)x + x^T \text{Im}(P)x + y^T \text{Im}(P)y = 0. \end{aligned}$$

Due to $\Sigma = \text{Re}(\Sigma) + \text{Im}(\Sigma)j = \text{Re}(\Sigma) > 0$, $P > 0$ holds. This completes the proof. \square

As seen, Lemma 2 gives a straightforward way to solve real matrix inequalities, yielding real solutions. Therefore, it significantly reduces the computation complexity, simplifying the H_∞ control design for FOSs.

In addition, two lemmas [42] are given as follows.

Lemma 3 *For any constant matrices S_1, S_2 and S_3 with $S_1^T = S_1$ and $S_3^T = S_3 > 0$, there is $S_1 + S_2 S_3^{-1} S_2^T < 0$ if and only if*

$$\begin{bmatrix} S_1 & S_2 \\ S_2^T & -S_3 \end{bmatrix} < 0.$$

Lemma 4 *For any T, H, E and $F^T(\sigma)F(\sigma) \leq I$, there holds*

$$T + HF(\sigma)E + E^T F^T(\sigma)H^T < 0,$$

if and only if there exists an $\varepsilon > 0$ such that

$$T + \varepsilon HH^T + \varepsilon^{-1} E^T E < 0,$$

where $\sigma \in \Theta \subset \mathbb{R}$.

3 Control Design

To achieve the control objective, an output-feedback FTC scheme based on a non-fragile observer is developed in this section.

3.1 Observer Design

To detect and estimate the actuator faults, the following non-fragile fault observer is constructed.

Observer rule *i*: IF $h_1(x(t))$ is H_{i1} and \dots and $h_p(x(t))$ is H_{ip} , THEN

$$\begin{aligned} \mathcal{D}^z \hat{x}(t) &= A_i \hat{x}(t) + B_i(u(t) + \hat{f}(t)) + (L_i + \Delta L_i)(y(t) - \hat{y}(t)), \\ \hat{y}(t) &= C_i \hat{x}(t), \\ \mathcal{D}^z \hat{f}(t) &= (F_i + \Delta F_i)(y(t) - \hat{y}(t)), \end{aligned} \tag{15}$$

where $\hat{x}(t)$, $\hat{y}(t)$ and $\hat{f}(t)$ are the estimates of $x(t)$, $y(t)$ and $f(t)$, respectively; L_i , F_i , ΔL_i and ΔF_i constitute the observer gains, with

$$[\Delta L_i \ \Delta F_i] = [U_{Li}F(\sigma)V_{Li} \ U_{Fi}F(\sigma)V_{Fi}], \tag{16}$$

where $L_i, F_i, U_{Li}, V_{Li}, U_{Fi}$ and V_{Fi} are fixed, and $F(\sigma)$ is variable but should meet

$$F^T(\sigma)F(\sigma) < I. \tag{17}$$

Then the overall fault observer is

$$\begin{aligned} \mathcal{D}^z \hat{x}(t) &= A(\theta)\hat{x}(t) + B(\theta)(u(t) + \hat{f}(t)) \\ &\quad + (L(\theta) + \Delta L(\theta))(y(t) - \hat{y}(t)), \\ \hat{y}(t) &= C(\theta)\hat{x}(t), \\ \mathcal{D}^z \hat{f}(t) &= (F(\theta) + \Delta F(\theta))(y(t) - \hat{y}(t)), \end{aligned} \tag{18}$$

where

$$\begin{aligned} L(\theta) &= \sum_{i=1}^r \theta_i L_i, \quad F(\theta) = \sum_{i=1}^r \theta_i F_i, \\ \Delta L(\theta) &= \sum_{i=1}^r \theta_i \Delta L_i = \sum_{i=1}^r \theta_i U_{Li}F(\sigma)V_{Li}, \\ \Delta F(\theta) &= \sum_{i=1}^r \theta_i \Delta F_i = \sum_{i=1}^r \theta_i U_{Fi}F(\sigma)V_{Fi}. \end{aligned}$$

Remark 2 The non-fragile observer is used to estimate the state and the actuator faults simultaneously. The observer gains are comprised of the fixed and variable gains such that the non-fragile observer is still effective when tiny bias or disturbances occur in the observer gains.

Next, we determine the observer parameters L_i and F_i in (15). It begins with analyzing the error systems. Let $e_x(t) = x(t) - \hat{x}(t)$, $e_f(t) = f(t) - \hat{f}(t)$, $e_y(t) = y(t) - \hat{y}(t)$. From (8) and (18), the error systems are described by

$$\begin{aligned} \mathcal{D}^z e_x(t) &= (A(\theta) - (L(\theta) + \Delta L(\theta))C(\theta))e_x(t) + B(\theta)e_f(t) \\ &\quad + (F(\theta) - (L(\theta) + \Delta L(\theta))D(\theta))\omega(t), \\ \mathcal{D}^z e_f(t) &= -(F(\theta) + \Delta F(\theta))C(\theta)e_x(t) - (F(\theta) \\ &\quad + \Delta F(\theta))D(\theta)\omega(t) + \mathcal{D}^z f(t), \\ e_y(t) &= C(\theta)e_x(t) + D(\theta)\omega(t). \end{aligned} \tag{19}$$

By Definition 1, Assumption 1 implies that $\mathcal{D}^z f(t)$ belongs to $L_2[0, \infty)$. Let

$$\begin{aligned} e(t) &= \begin{bmatrix} e_x(t) \\ e_f(t) \end{bmatrix}, \quad \xi(t) = \begin{bmatrix} \omega(t) \\ \mathcal{D}^z f(t) \end{bmatrix}, \\ \hat{A}(\theta) &= \begin{bmatrix} A(\theta) & B(\theta) \\ 0 & 0 \end{bmatrix}, \quad \hat{L}(\theta) = \begin{bmatrix} L(\theta) \\ F(\theta) \end{bmatrix}, \\ \hat{C}(\theta) &= \begin{bmatrix} C(\theta) & 0 \end{bmatrix}, \quad \hat{D}(\theta) = \begin{bmatrix} D(\theta) \\ 0 \end{bmatrix}, \\ \hat{G}(\theta) &= \begin{bmatrix} G(\theta) & 0 \\ 0 & I \end{bmatrix}, \quad \Delta \hat{L}(\theta) = \begin{bmatrix} \Delta L(\theta) \\ \Delta F(\theta) \end{bmatrix} = \sum_{i=1}^r \theta_i \Delta \hat{L}_i \\ &= \sum_{i=1}^r \theta_i \hat{U}_i \hat{F}(\sigma) \hat{V}_i, \quad \Delta \hat{L}_i = \hat{U}_i \hat{F}(\sigma) \hat{V}_i, \quad \hat{U}_i = \begin{bmatrix} U_{Li} & 0 \\ 0 & U_{Fi} \end{bmatrix}, \\ \hat{F}(\sigma) &= \begin{bmatrix} F(\sigma) & 0 \\ 0 & F(\sigma) \end{bmatrix}, \quad \hat{V}_i = \begin{bmatrix} V_{Li} \\ V_{Fi} \end{bmatrix}. \end{aligned}$$

The overall error system is

$$\begin{aligned} \mathcal{D}^z e(t) &= (\hat{A}(\theta) - (\hat{L}(\theta) + \Delta \hat{L}(\theta))\hat{C}(\theta))e(t) \\ &\quad + (\hat{G}(\theta) - (\hat{L}(\theta) + \Delta \hat{L}(\theta))\hat{D}(\theta))\xi(t), \\ e_y(t) &= \hat{C}(\theta)e(t) + \hat{D}(\theta)\xi(t). \end{aligned} \tag{20}$$

Choose a symmetric matrix X_o , a skew-symmetric matrix Y_o , and a bank of matrices with the appropriate dimensions, $W_k, k = 1, 2, \dots, r$, together with a set of positive scalars $\varepsilon_{ik}, i, k = 1, 2, \dots, r$, such that

$$\begin{bmatrix} X_o & Y_o \\ -Y_o & X_o \end{bmatrix} > 0, \tag{21}$$

$$Q_{ii} < 0, \tag{22}$$

$$Q_{ik} + Q_{ki} < 0, \quad i < k, \tag{23}$$

where

$$\begin{aligned} Q_{ik} &= \begin{bmatrix} \Pi_{11ik} & \star & \star & \star \\ \Pi_{21ik} & \Pi_{22ik} & \star & \star \\ \hat{f} & 0 & -\gamma_o I & \star \\ \hat{U}_k^T(aX_o - bY_o) & 0 & 0 & -\varepsilon_{ik} I \end{bmatrix}, \\ \Pi_{11ik} &= \text{sym}\{\hat{A}_i^T(aX_o - bY_o) - \hat{C}_i^T W_k\} + \varepsilon_{ik} \hat{C}_i^T \hat{V}_k^T \hat{V}_k \hat{C}_i, \\ \Pi_{21ik} &= \hat{G}_i^T(aX_o - bY_o) - \hat{D}_i^T W_k + \varepsilon_{ik} \hat{D}_i^T \hat{V}_k^T \hat{V}_k \hat{C}_i, \\ \Pi_{22ik} &= -\gamma_o I + \varepsilon_{ik} \hat{D}_i^T \hat{V}_k^T \hat{V}_k \hat{D}_i \\ \hat{f} &= [0 \quad I_s]. \end{aligned}$$

Accordingly, the observer gain is set to be

$$\hat{L}_i = (aX_o - bY_o)^{-T} W_i^T, i = 1, 2, \dots, r. \tag{24}$$

3.2 Controller Design

With the fault estimates, we design a dynamic output feedback fault-tolerant controller to stabilize the system in (8).

Controller rule k : IF $f_1(x(t))$ is F_{k1} and \dots and $f_p(x(t))$ is F_{kp} , THEN

$$u(t) = C_{ck}\psi(t) - \hat{f}(t), \tag{25}$$

$$\mathcal{D}^\alpha \psi(t) = A_{ck}\psi(t) + B_{ck}y(t),$$

where F_{kg} stands for the k th fuzzy set of the function $f_g(x(t))$, $k = 1, 2, \dots, r$, $g = 1, 2, \dots, p$, and p is the number of the premise variables. A_{ck} , B_{ck} , and C_{ck} are the controller gains to be determined. Let

$$A_{ck} = \sum_{i=1}^r \theta_i A_{cik},$$

and

$$A_c(\theta, \eta) = \sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k A_{cik}, \quad B_c(\eta) = \sum_{k=1}^r \eta_k B_{ck},$$

$$C_c(\eta) = \sum_{k=1}^r \eta_k C_{ck}.$$

The overall output feedback controller is

$$u(t) = C_c(\eta)\psi(t) - \hat{f}(t), \tag{26}$$

$$\mathcal{D}^\alpha \psi(t) = A_c(\theta, \eta)\psi(t) + B_c(\eta)y(t),$$

where

$$\eta_k(x(t)) = \underline{m}_k(x(t))\underline{\eta}_k(x(t)) + \bar{m}_k(x(t))\bar{\eta}_k(x(t)) \geq 0, \tag{27}$$

and $\underline{m}_k(x(t))$ and $\bar{m}_k(x(t))$ are the predefined functions that meet

$$0 \leq \underline{m}_k(x(t)) \leq 1,$$

$$0 \leq \bar{m}_k(x(t)) \leq 1,$$

$$\underline{m}_k(x(t)) + \bar{m}_k(x(t)) = 1.$$

In (27), $\underline{\eta}_k(x(t))$ and $\bar{\eta}_k(x(t))$ are the lower and upper bounds of the membership functions of the k th rule, respectively, with

$$\underline{\eta}_k(x(t)) = \prod_{g=1}^p \underline{\mu}_{F_{kg}}(f_g(x(t))),$$

$$\bar{\eta}_k(x(t)) = \prod_{g=1}^p \bar{\mu}_{F_{kg}}(f_g(x(t))),$$

where $\underline{\mu}_{F_{kg}}(f_g(x(t)))$ and $\bar{\mu}_{F_{kg}}(f_g(x(t)))$ denote the lower and upper memberships of $f_g(x(t))$ in F_{kg} , respectively, meeting $\bar{\mu}_{F_{kg}}(f_s(x(t))) \geq \underline{\mu}_{F_{kg}}(f_s(x(t))) \geq 0, i = 1, 2, \dots, r.$

$$\bar{\eta}_k(x(t)) \geq \underline{\eta}_k(x(t)) \geq 0.$$

Substituting (26) into (8) gives the dynamic equation of the closed-loop system:

$$\mathcal{D}^\alpha \varphi(t) = \bar{A}(\theta, \eta)\varphi(t) + \bar{G}(\theta, \eta)v(t), \tag{28}$$

$$y(t) = \hat{C}(\theta)\varphi(t) + \hat{D}(\theta)v(t),$$

where

$$\varphi(t) = \begin{bmatrix} x(t) \\ \psi(t) \end{bmatrix}, \quad v(t) = \begin{bmatrix} \omega(t) \\ e_f(t) \end{bmatrix},$$

$$\bar{A}(\theta, \eta) = \begin{bmatrix} A(\theta) & B(\theta)C_c(\eta) \\ B_c(\eta)C(\theta) & A_c(\theta, \eta) \end{bmatrix},$$

$$\bar{G}(\theta, \eta) = \begin{bmatrix} G(\theta) & B(\theta) \\ B_c(\eta)D(\theta) & 0 \end{bmatrix}.$$

A principle for the selection of the controller gains is given as follows. Choose two symmetric matrices X_{c1} and X_{c2} , a skew-symmetric matrix Y_{c1} , and two banks of matrices with appropriate dimensions, Φ_i and $\Psi_i, i = 1, 2, \dots, r$ such that

$$\begin{bmatrix} X_{c1} & Y_{c1} & I & 0 \\ -Y_{c1} & X_{c1} & 0 & I \\ I & 0 & X_{c2} & 0 \\ 0 & I & 0 & X_{c2} \end{bmatrix} > 0, \tag{29}$$

$$\Xi_{ik} - \Delta_i < 0, \tag{30}$$

$$q_i \Xi_{ii} + (1 - q_i)\Delta_i < 0, \tag{31}$$

$$q_k \Xi_{ik} + (1 - q_k)\Delta_i + q_i \Xi_{ki} + (1 - q_i)\Delta_k < 0, i < k, \tag{32}$$

where

$$\Xi_{ik} = \begin{bmatrix} \Gamma_{11ik} & * & * & * & * \\ 0 & \Gamma_{22ik} & * & * & * \\ \Gamma_{31ik} & \Gamma_{32i} & -\gamma_c I & * & * \\ \Gamma_{41i} & \Gamma_{42i} & 0 & -\gamma_c I & * \\ C_i & C_i X_{c2} & D_i & 0 & -\gamma_c I \end{bmatrix},$$

$$\Gamma_{11ik} = \text{sym}\{A_i^T(aX_{c1} - bY_{c1}) + \Phi_k C_i\},$$

$$\Gamma_{22ik} = \text{sym}\{aA_i X_{c2} + aB_i \Psi_k\},$$

$$\Gamma_{31ik} = G_i^T(aX_{c1} - bY_{c1}) + D_i^T \Phi_k^T,$$

$$\Gamma_{32i} = aG_i^T,$$

$$\Gamma_{41i} = B_i^T(aX_{c1} - bY_{c1}),$$

$$\Gamma_{42i} = aB_i^T.$$

Accordingly, the controller gain matrices are set to be

$$\begin{aligned}
 A_{cik} &= -(a(X_{c2}^{-1} - X_{c1}) + bY_{c1})^{-T}(aA_i^T X_{c2}^{-1} \\
 &\quad + (aX_{c1} - bY_{c1})^T(B_i\Psi_k X_{c2}^{-1} + A_i) + \Phi_k C_i), \\
 B_{ci} &= (a(X_{c2}^{-1} - X_{c1}) + bY_{c1})^{-T}\Phi_i, \\
 C_{ci} &= \Psi_i X_{c2}^{-1}.
 \end{aligned}
 \tag{33}$$

4 Performance Analysis

We first show the effectiveness of the designed observer by the following theorem.

Theorem 1 Under Assumption 1, applying the observer in (18) with (24) to the system in (20) ensures $\|e_f(t)\|_2 < \gamma_o \|\xi(t)\|_2$.

Proof Substituting (24) into (20) gives

$$\begin{aligned}
 \mathcal{D}^z e(t) &= (\hat{A}(\theta) - ((aX_o - bY_o)^{-T} W^T(\theta) + \Delta\hat{L}(\theta))\hat{C}(\theta))e(t) \\
 &\quad + (\hat{G}(\theta) - ((aX_o - bY_o)^{-T} W^T(\theta) \\
 &\quad + \Delta\hat{L}(\theta))\hat{D}(\theta))\xi(t),
 \end{aligned}
 \tag{34}$$

where $W^T(\theta) = \sum_{i=1}^r \theta_i W_i^T$. Note that

$$e_f(t) = \hat{I}e(t), \tag{35}$$

with $\hat{I} = [0 \ I_s]$. Applying Lemma 3, it is obtained that (22) is equivalent to

$$\begin{aligned}
 &\begin{bmatrix} \Pi_{11ii} & \star & \star \\ \Pi_{21ii} & -\gamma_o I + \varepsilon_{ii} \hat{D}_i^T \hat{V}_i^T \hat{V}_i \hat{D}_i & \star \\ \hat{I} & 0 & -\gamma_o I \end{bmatrix} \\
 &+ \frac{1}{\varepsilon_{ii}} \begin{bmatrix} (aX_o - bY_o)^T \hat{U}_i \\ 0 \\ 0 \end{bmatrix} \begin{bmatrix} (aX_o - bY_o)^T \hat{U}_i \\ 0 \\ 0 \end{bmatrix}^T \\
 &= \begin{bmatrix} \text{sym}\{\hat{A}_i^T(aX - bY) - \hat{C}_i^T W_i\} & \star & \star \\ \hat{G}_i^T(aX - bY) - \hat{D}_i^T W_i & -\gamma_o I & \star \\ \hat{I} & 0 & -\gamma_o I \end{bmatrix} \\
 &+ \varepsilon_{ii} \begin{bmatrix} -\hat{C}_i^T \hat{V}_i^T \\ -\hat{D}_i^T \hat{V}_i^T \\ 0 \end{bmatrix} \begin{bmatrix} -\hat{C}_i^T \hat{V}_i^T \\ -\hat{D}_i^T \hat{V}_i^T \\ 0 \end{bmatrix}^T \\
 &+ \frac{1}{\varepsilon_{ii}} \begin{bmatrix} (aX - bY)^T \hat{U}_i \\ 0 \\ 0 \end{bmatrix} \begin{bmatrix} (aX - bY)^T \hat{U}_i \\ 0 \\ 0 \end{bmatrix}^T < 0.
 \end{aligned}
 \tag{36}$$

From (17), we have

$$\begin{aligned}
 \hat{F}^T(\sigma)\hat{F}(\sigma) &= \begin{bmatrix} F(\sigma) & 0 \\ 0 & F(\sigma) \end{bmatrix}^T \begin{bmatrix} F(\sigma) & 0 \\ 0 & F(\sigma) \end{bmatrix} \\
 &= \begin{bmatrix} F^T(\sigma)F(\sigma) & 0 \\ 0 & F^T(\sigma)F(\sigma) \end{bmatrix} < I.
 \end{aligned}
 \tag{37}$$

By Lemma 4, combining (36) and (37) yields

$$\begin{aligned}
 &\begin{bmatrix} \text{sym}\{\hat{A}_i^T(aX_o - bY_o) - \hat{C}_i^T W_i\} & \star & \star \\ \hat{G}_i^T(aX_o - bY_o) - \hat{D}_i^T W_i & -\gamma_o I & \star \\ \hat{I} & 0 & -\gamma_o I \end{bmatrix} \\
 &+ \text{sym}\left\{ \begin{bmatrix} -\hat{C}_i^T \hat{V}_i^T \\ -\hat{D}_i^T \hat{V}_i^T \\ 0 \end{bmatrix} \hat{F}^T(\sigma) \begin{bmatrix} (aX_o - bY_o)^T \hat{U}_i \\ 0 \\ 0 \end{bmatrix}^T \right\} < 0.
 \end{aligned}
 \tag{38}$$

Noting that

$$\Delta\hat{L}_i = \hat{U}_i \hat{F}(\sigma) \hat{V}_i,$$

(38) is rewritten as

$$\begin{aligned}
 &\begin{bmatrix} \text{sym}\{\hat{A}_i^T(aX_o - bY_o) - \hat{C}_i^T W_i\} & \star & \star \\ \hat{G}_i^T(aX_o - bY_o) - \hat{D}_i^T W_i & -\gamma_o I & \star \\ \hat{I} & 0 & -\gamma_o I \end{bmatrix} \\
 &+ \text{sym}\left\{ \begin{bmatrix} -\hat{C}_i^T \\ -\hat{D}_i^T \\ 0 \end{bmatrix} \Delta\hat{L}_i^T \begin{bmatrix} (aX_o - bY_o)^T \\ 0 \\ 0 \end{bmatrix}^T \right\} < 0.
 \end{aligned}
 \tag{39}$$

From (39), it follows that

$$T_{ii} < 0. \tag{40}$$

Similarly, from (23), we have

$$T_{ik} + T_{ki} < 0, \quad i < k, \tag{41}$$

where

$$T_{ik} = \begin{bmatrix} \text{sym}\{(\hat{A}_i - \Delta\hat{L}_k \hat{C}_i)^T(aX_o - bY_o) - \hat{C}_i^T W_i\} & \star & \star \\ (\hat{G}_i - \Delta\hat{L}_k \hat{D}_i)^T(aX_o - bY_o) - \hat{D}_i^T W_i & -\gamma_o I & \star \\ \hat{I} & 0 & -\gamma_o I \end{bmatrix}.
 \tag{42}$$

According to

$$\sum_{i=1}^r \sum_{k=1}^r \theta_i \theta_k T_{ik} = \sum_{i=1}^r \theta_i^2 T_{ii} + \sum_{i=1}^{r-1} \sum_{k=i+1}^r \theta_i \theta_k (T_{ik} + T_{ki}),$$

it is straightforward to see by the combination of (40) and (41) that

$$\begin{bmatrix} \Pi_{11} & \star & \star \\ \Pi_{21} & -\gamma_o I & \star \\ \hat{f} & 0 & -\gamma_o I \end{bmatrix} < 0, \tag{43}$$

where

$$\begin{aligned} \Pi_{11} &= \text{sym}\{(\hat{A}(\theta) - \Delta \hat{L}(\theta) \hat{C}(\theta))^T (aX_o - bY_o) - \hat{C}^T(\theta)W(\theta)\}, \\ \Pi_{21} &= (\hat{G}(\theta) - \Delta \hat{L}(\theta) \hat{D}(\theta))^T (aX_o - bY_o) - \hat{D}^T(\theta)W(\theta). \end{aligned}$$

Following Lemma 1, we have $\|e_f(t)\|_2 < \gamma_o \|\zeta(t)\|_2$. This ends the proof. \square

It is noted that the minimum H_∞ attenuation level of Theorem 1 can be obtained by solving the programming problem: minimize γ_o subject to (21)–(23).

Next, we show the effectiveness of the designed controller by the following theorem.

Theorem 2 *Apply the controller in (26) with (33) to the system in (8). Under Assumption 1 and $\eta_k - \varrho_k \theta_k \geq 0, 0 < \varrho_k < 1, i, k = 1, 2, \dots, r$, the resulting control system in (28) is robustly stable and meets $\|y(t)\|_2 < \gamma_c \|v(t)\|_2$.*

Proof Eq. (29) shows

$$\begin{bmatrix} X_{c1} & Y_{c1} \\ -Y_{c1} & X_{c1} \end{bmatrix} > 0, \quad \begin{bmatrix} X_{c2} & 0 \\ 0 & X_{c2} \end{bmatrix} > 0, \tag{44}$$

and

$$\begin{bmatrix} X_{c1} & Y_{c1} \\ -Y_{c1} & X_{c1} \end{bmatrix} - \begin{bmatrix} X_{c2} & 0 \\ 0 & X_{c2} \end{bmatrix}^{-1} = \begin{bmatrix} X_{c1} - X_{c2}^{-1} & Y_{c1} \\ -Y_{c1} & X_{c1} - X_{c2}^{-1} \end{bmatrix} > 0. \tag{45}$$

Let

$$X_c = \begin{bmatrix} X_{c1} & X_{c2}^{-1} - X_{c1} \\ X_{c2}^{-1} - X_{c1} & -X_{c2}^{-1} + X_{c1} \end{bmatrix}, \tag{46}$$

$$Y_c = \begin{bmatrix} Y_{c1} & -Y_{c1} \\ -Y_{c1} & Y_{c1} \end{bmatrix}, \tag{47}$$

and then construct the following matrix in the form of (12):

$$\begin{bmatrix} X_c & Y_c \\ -Y_c & X_c \end{bmatrix} = \left[\begin{array}{cc|cc} X_{c1} & X_{c2}^{-1} - X_{c1} & Y_{c1} & -Y_{c1} \\ X_{c2}^{-1} - X_{c1} & -X_{c2}^{-1} + X_{c1} & -Y_{c1} & Y_{c1} \\ \hline -Y_{c1} & Y_{c1} & X_{c1} & X_{c2}^{-1} - X_{c1} \\ Y_{c1} & -Y_{c1} & X_{c2}^{-1} - X_{c1} & -X_{c2}^{-1} + X_{c1} \end{array} \right]. \tag{48}$$

Let $Z = \begin{bmatrix} I & 0 & 0 & 0 \\ 0 & 0 & I & 0 \\ 0 & I & 0 & 0 \\ 0 & 0 & 0 & I \end{bmatrix}$. Then, there is

$$\begin{aligned} & Z \begin{bmatrix} X_c & Y_c \\ -Y_c & X_c \end{bmatrix} Z^T \\ &= \begin{bmatrix} X_{c1} & Y_{c1} & X_{c2}^{-1} - X_{c1} & -Y_{c1} \\ -Y_{c1} & X_{c1} & Y_{c1} & X_{c2}^{-1} - X_{c1} \\ X_{c2}^{-1} - X_{c1} & -Y_{c1} & -X_{c2}^{-1} + X_{c1} & Y_{c1} \\ Y_{c1} & X_{c2}^{-1} - X_{c1} & -Y_{c1} & -X_{c2}^{-1} + X_{c1} \end{bmatrix}. \end{aligned} \tag{49}$$

From (44), we have

$$\begin{aligned} & \begin{bmatrix} X_{c1} & Y_{c1} \\ -Y_{c1} & X_{c1} \end{bmatrix} - \begin{bmatrix} X_{c2}^{-1} - X_{c1} & -Y_{c1} \\ Y_{c1} & X_{c2}^{-1} - X_{c1} \end{bmatrix} \\ & \times \begin{bmatrix} -X_{c2}^{-1} + X_{c1} & Y_{c1} \\ -Y_{c1} & -X_{c2}^{-1} + X_{c1} \end{bmatrix}^{-1} \begin{bmatrix} X_{c2}^{-1} - X_{c1} & -Y_{c1} \\ Y_{c1} & X_{c2}^{-1} - X_{c1} \end{bmatrix} \\ & \begin{bmatrix} X_{c2} & 0 \\ 0 & X_{c2} \end{bmatrix}^{-1} > 0, \end{aligned} \tag{50}$$

which meets the condition of Lemma 3. Therefore, by Lemma 3, we further have

$$Z \begin{bmatrix} X_c & Y_c \\ -Y_c & X_c \end{bmatrix} Z^T > 0, \tag{51}$$

which in turn means that

$$\begin{bmatrix} X_c & Y_c \\ -Y_c & X_c \end{bmatrix} > 0. \tag{52}$$

Combining (46) and (47) yields

$$aX_c - bY_c = \begin{bmatrix} aX_{c1} - bY_{c1} & a(X_{c2}^{-1} - X_{c1}) + bY_{c1} \\ a(X_{c2}^{-1} - X_{c1}) + bY_{c1} & -a(X_{c2}^{-1} - X_{c1}) - bY_{c1} \end{bmatrix}. \tag{53}$$

Let

$$\Xi = \sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k \Xi_{ik} = \sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k \left[\begin{array}{cc|cc|cc} \Gamma_{11ik} & \star & \star & \star & \star & \\ 0 & \Gamma_{22ik} & \star & \star & \star & \\ \hline \Gamma_{31ik} & \Gamma_{32i} & -\gamma_c I & \star & \star & \\ \Gamma_{41i} & \Gamma_{42i} & 0 & -\gamma_c I & \star & \\ \hline C_i & C_i X_{c2} & D_i & 0 & -\gamma_c I & \end{array} \right]. \tag{54}$$

Due to $\sum_{i=1}^r \sum_{k=1}^r \theta_i (\theta_k - \eta_k) \Delta_i = 0$, where $\Delta_i = \Delta_i^T$ is introduced only for analysis, (54) becomes

$$\begin{aligned}
 \Xi &= \sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k \Xi_{ik} \\
 &= \sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k \Xi_{ik} + \sum_{i=1}^r \sum_{k=1}^r \theta_i (\theta_k - \eta_k) \Delta_i \\
 &= \sum_{i=1}^r \sum_{k=1}^r \theta_i (\eta_k + \varrho_k \theta_k - \varrho_k \theta_k) \Xi_{ik} \\
 &\quad + \sum_{i=1}^r \sum_{k=1}^r \theta_i (\theta_k - \eta_k + \varrho_k \theta_k - \varrho_k \theta_k) \Delta_i \\
 &= \sum_{i=1}^r \sum_{k=1}^r \theta_i \theta_k (\varrho_k \Xi_{ik} + (1 - \varrho_k) \Delta_i) \\
 &\quad + \sum_{i=1}^r \sum_{k=1}^r \theta_i (\eta_k - \varrho_k \theta_k) (\Xi_{ik} - \Delta_i).
 \end{aligned}$$

Under $\eta_k - \varrho_k \theta_k \geq 0$, from (30), we further have

$$\begin{aligned}
 \Xi &< \sum_{i=1}^r \sum_{k=1}^r \theta_i \theta_k (\varrho_k \Xi_{ik} + (1 - \varrho_k) \Delta_i) \\
 &= \sum_{i=1}^r \theta_i^2 (\varrho_i \Xi_{ii} + (1 - \varrho_i) \Delta_i) \\
 &\quad + \sum_{k=1}^r \sum_{i < k} \theta_i \theta_k (\varrho_k \Xi_{ik} + (1 - \varrho_k) \Delta_i) \\
 &\quad + \varrho_i \Xi_{ki} + (1 - \varrho_i) \Delta_k.
 \end{aligned} \tag{56}$$

Note that $\sum_{i=1}^r \sum_{k=1}^r \theta_i \theta_k = \sum_{i=1}^r \theta_i = 1$, and recall (30)–(32).

Then, (56) further meets

$$\Xi < 0. \tag{57}$$

By Lemma 3, (57) is equivalent to

$$\begin{aligned}
 &\sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k \begin{bmatrix} \Gamma_{11ik} & \star \\ 0 & \Gamma_{22ik} \end{bmatrix} \\
 &+ \frac{1}{\gamma_c} \left(\sum_{i=1}^r \theta_i \begin{bmatrix} C_i^T \\ X_{c2} C_i^T \end{bmatrix} \right) \left(\sum_{i=1}^r \theta_i \begin{bmatrix} C_i^T \\ X_{c2} C_i^T \end{bmatrix} \right)^T \\
 &+ \left(\sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k \begin{bmatrix} \Gamma_{31ik} & \Gamma_{32i} \\ \Gamma_{41i} & \Gamma_{42i} \end{bmatrix} \right)^T \\
 &+ \frac{1}{\gamma_c} \left(\sum_{i=1}^r \theta_i \begin{bmatrix} C_i^T \\ X_{c2} C_i^T \end{bmatrix} \right) \left(\sum_{i=1}^r \theta_i \begin{bmatrix} D_i^T \\ 0 \end{bmatrix} \right)^T \\
 &\left(\gamma_c I + \left(\sum_{i=1}^r \theta_i \begin{bmatrix} D_i^T \\ 0 \end{bmatrix} \right) \left(\sum_{i=1}^r \theta_i \begin{bmatrix} D_i^T \\ 0 \end{bmatrix} \right)^T \right)^{-1} \\
 &\left(\sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k \begin{bmatrix} \Gamma_{31ik} & \Gamma_{32i} \\ \Gamma_{41i} & \Gamma_{42i} \end{bmatrix} \right. \\
 &\left. + \frac{1}{\gamma_c} \left(\sum_{i=1}^r \theta_i \begin{bmatrix} C_i^T \\ X_{c2} C_i^T \end{bmatrix} \right) \left(\sum_{i=1}^r \theta_i \begin{bmatrix} D_i^T \\ 0 \end{bmatrix} \right)^T \right)^T < 0.
 \end{aligned} \tag{58}$$

Then, after algebraic manipulations, we get

$$\begin{aligned}
 &A^T \text{sym} \{ \bar{A}^T(\theta, \eta) (aX_c - bY_c) \} A \\
 &= \sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k A^T \text{sym} \left\{ \begin{bmatrix} A_i & B_i C_{ck} \\ B_{ck} C_i & A_{cik} \end{bmatrix} \right\}^T \\
 &\quad \left[\begin{array}{cc} aX_{c1} - bY_{c1} & a(X_{c2}^{-1} - X_{c1}) + bY_{c1} \\ a(X_{c2}^{-1} - X_{c1}) + bY_{c1} & -a(X_{c2}^{-1} - X_{c1}) - bY_{c1} \end{array} \right] A \\
 &= \sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k \left(\begin{bmatrix} \Gamma_{11ik} & \star \\ 0 & \Gamma_{22ik} \end{bmatrix} \right),
 \end{aligned} \tag{59}$$

where

$$A = \begin{bmatrix} I & X_{c2} \\ 0 & X_{c2} \end{bmatrix},$$

and

$$\begin{aligned}
 \Phi_k &= (a(X_{c2}^{-1} - X_{c1}) + bY_{c1})^T B_{ck}, \\
 \Psi_k &= C_{ck} X_{c2}.
 \end{aligned}$$

Accordingly, there hold

$$\begin{aligned}
 &A^T ((aX_c - bY_c)^T \bar{G}(\theta, \eta) + \frac{1}{\gamma_c} \hat{C}^T(\theta) \hat{D}(\theta)) \\
 &\quad \left(\gamma_c I - \frac{1}{\gamma_c} \hat{D}^T(\theta) \hat{D}(\theta) \right)^{-1} \\
 &\quad \times \left(\bar{G}^T(\theta, \eta) (aX_c - bY_c) + \frac{1}{\gamma_c} \hat{D}^T(\theta) \hat{C}(\theta) \right) A \\
 &= \left[\begin{array}{cc} \Gamma_{31}(\theta, \eta) & \Gamma_{32}(\theta) \\ \Gamma_{41}(\theta) & \Gamma_{42}(\theta) \end{array} \right]^T \left[\begin{array}{cc} \gamma_c I - \frac{1}{\gamma_c} \hat{D}^T(\theta) \hat{D}(\theta) & 0 \\ 0 & \gamma_c I \end{array} \right]^{-1} \\
 &\quad \left[\begin{array}{cc} \Gamma_{31}(\theta, \eta) & \Gamma_{32}(\theta) \\ \Gamma_{41}(\theta) & \Gamma_{42}(\theta) \end{array} \right] \\
 &= \left(\sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k \begin{bmatrix} \Gamma_{31ik} & \Gamma_{32i} \\ \Gamma_{41i} & \Gamma_{42i} \end{bmatrix} \right)^T + \frac{1}{\gamma_c} \left(\sum_{i=1}^r \theta_i \begin{bmatrix} C_i^T \\ X_{c2} C_i^T \end{bmatrix} \right) \\
 &\quad \left(\sum_{i=1}^r \theta_i \begin{bmatrix} D_i^T \\ 0 \end{bmatrix} \right)^T \\
 &\quad \left(\gamma_c I + \left(\sum_{i=1}^r \theta_i \begin{bmatrix} D_i^T \\ 0 \end{bmatrix} \right) \left(\sum_{i=1}^r \theta_i \begin{bmatrix} D_i^T \\ 0 \end{bmatrix} \right)^T \right)^{-1} \\
 &\quad \left(\sum_{i=1}^r \sum_{k=1}^r \theta_i \eta_k \begin{bmatrix} \Gamma_{31ik} & \Gamma_{32i} \\ \Gamma_{41i} & \Gamma_{42i} \end{bmatrix} \right. \\
 &\quad \left. + \frac{1}{\gamma_c} \left(\sum_{i=1}^r \theta_i \begin{bmatrix} C_i^T \\ X_{c2} C_i^T \end{bmatrix} \right) \left(\sum_{i=1}^r \theta_i \begin{bmatrix} D_i^T \\ 0 \end{bmatrix} \right)^T \right)^T,
 \end{aligned} \tag{60}$$

and

$$\begin{aligned}
& \frac{1}{\gamma_c} A^T \hat{C}^T(\theta) \hat{C}(\theta) A \\
&= \frac{1}{\gamma_c} [C(\theta) \quad C(\theta)X_{c2}]^T [C(\theta) \quad C(\theta)X_{c2}] \\
&= \frac{1}{\gamma_c} \left(\sum_{i=1}^r \theta_i \begin{bmatrix} C_i^T \\ X_{c2} C_i^T \end{bmatrix} \right) \left(\sum_{i=1}^r \theta_i \begin{bmatrix} C_i^T \\ X_{c2} C_i^T \end{bmatrix} \right)^T.
\end{aligned} \tag{61}$$

Now, we combine (58)–(61), and obtain

$$\begin{aligned}
& A^T (\text{sym}\{A^T(\theta, \eta)(aX_c - bY_c)\} + \frac{1}{\gamma_c} \hat{C}^T(\theta) \hat{C}(\theta) \\
&+ ((aX_c - bY_c)^T \bar{G}(\theta, \eta) + \frac{1}{\gamma_c} \hat{C}^T(\theta) \hat{D}(\theta)) (\gamma_c I - \frac{1}{\gamma_c} \hat{D}^T(\theta) \hat{D}(\theta))^{-1} \\
&\times (\bar{G}^T(\theta, \eta)(aX_c - bY_c) + \frac{1}{\gamma_c} \hat{D}^T(\theta) \hat{C}(\theta))) A < 0,
\end{aligned} \tag{62}$$

which means

$$\begin{aligned}
& \text{sym}\{A^T(\theta, \eta)(aX_c - bY_c)\} + \frac{1}{\gamma_c} \hat{C}^T(\theta) \hat{C}(\theta) \\
&+ ((aX_c - bY_c)^T \bar{G}(\theta, \eta) + \frac{1}{\gamma_c} \hat{C}^T(\theta) \hat{D}(\theta)) (\gamma_c I - \frac{1}{\gamma_c} \hat{D}^T(\theta) \hat{D}(\theta))^{-1} \\
&\times (\bar{G}^T(\theta, \eta)(aX_c - bY_c) + \frac{1}{\gamma_c} \hat{D}^T(\theta) \hat{C}(\theta)) < 0.
\end{aligned} \tag{63}$$

According to Lemma 3, (63) is equivalent to

$$\begin{bmatrix} \text{sym}\{A^T(\theta, \eta)(aX_c - bY_c)\} & \star & \star \\ \bar{G}^T(\theta, \eta)(aX_c - bY_c) & -\gamma_c I & \star \\ \hat{C}(\theta) & \hat{D}(\theta) & -\gamma_c I \end{bmatrix} < 0. \tag{64}$$

This means that the closed-loop system in (28) is robustly stable and meets the H_∞ performance index $\|y(t)\|_2 < \gamma \|v(t)\|_2$. This completes the proof. \square

It is also noted that the minimum H_∞ attenuation level of Theorem 2 can be obtained by solving the programming problem: minimize γ_c subject to (29)–(32).

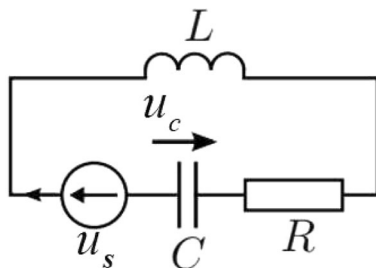


Fig. 1 Electrical circuit system

5 Simulation Examples

In this section, two simulation examples are given to illustrate the effectiveness of the control strategy proposed in this paper and the performance of the resulting control system.

5.1 Electrical Circuit System

Consider an electrical circuit system shown in Fig. 1 with the inductor L , the capacitance C , the resistance R , and the source voltage u_s . Let i_L and i_C denote the currents passing through the inductor L , and the capacitance C , respectively; let u_L and u_C denote the voltage on the inductor L and on the capacitance C , respectively. As described [3],

$$\begin{aligned}
u_L(t) &= L \frac{d^x i_L(t)}{dt^x}, \\
i_C(t) &= C \frac{d^x u_C(t)}{dt^x}, \quad 0 < \alpha < 1.
\end{aligned}$$

Further, according to the Kirchhoffs law, we have

$$\begin{aligned}
u_s &= R i_L + L \frac{d^x i_L}{dt^x} + u_C, \\
i_L &= C \frac{d^x u_C}{dt^x}.
\end{aligned}$$

Therefore, the electrical circuit system is described by

$$\frac{d^x}{dt^x} \begin{bmatrix} u_C \\ i_L \end{bmatrix} = \begin{bmatrix} 0 & \frac{1}{C} \\ -\frac{1}{L} & -\frac{R}{L} \end{bmatrix} \begin{bmatrix} u_C \\ i_L \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{L} \end{bmatrix} u_s.$$

Let $u_f(t) = u(t) + f(t)$, where

$$f(t) = \begin{cases} 0, & 0 < t < 30, \\ 5(1 - e^{30-t}), & \text{else,} \end{cases}$$

simulates the actuator fault. Moreover, to test the robustness of our observer, referring to Ref. [3], consider the measurement output as

$$y(t) = \sum_{i=1}^r \theta_i(x(t))(C_i x(t) + \tilde{D}_i \omega_2(t)),$$

with

$$\omega_2(t) = 1/(t + 10),$$

and take the external disturbance into account:

$$\omega_1(t) = 0.1 \sin(5t).$$

In the simulation, let $\alpha = 0.9$, $L = 2$, $C = 1$, $R = 4 + 2\sin(t)$ and $u_s = 1$. The matrices in the fuzzy system in (8) are selected as

$$A_1 = \begin{bmatrix} 0 & 1 \\ -0.5 & \max\{-\frac{R}{L}\} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -0.5 & -1 \end{bmatrix},$$

$$A_2 = \begin{bmatrix} 0 & 1 \\ -0.5 & \min\{-\frac{R}{L}\} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -0.5 & -3 \end{bmatrix},$$

$$B = \begin{bmatrix} 0 \\ 0.5 \end{bmatrix}, \tilde{G} = \begin{bmatrix} 0.1 \\ 0.2 \end{bmatrix}, C_1 = [1 \quad -3],$$

$$\tilde{D}_1 = 0.1, C_2 = [1 \quad -2], \tilde{D}_2 = -0.2.$$

The corresponding membership functions are chosen as $\theta_1 = (1 - 0.5\sin(t))/2$ and $\theta_2 = 1 - \theta_1$. Note that the above system is input-to-state stable. Thus, it is mainly used to verify the effectiveness of our fault observer. Following Theorem 1, we design a non-fragile observer for the above system. The variable gains in the observer in (18) are determined by

$$U_{L1} = \begin{bmatrix} -0.1 \\ 0.2 \end{bmatrix}, U_{L2} = \begin{bmatrix} -0.1 \\ -0.2 \end{bmatrix}, V_{L1} = 0.2, V_{L2} = 0.3,$$

$$U_{F1} = 0.1, V_{F1} = -0.2, U_{F2} = 0.5, V_{F2} = 0.2, F(\sigma) = \cos(0.2\sigma).$$

Set the H_∞ performance index to be $\gamma_o = 0.5$. Then, the constant observer gains are obtained by solving the LMIs in (21)–(23):

$$\hat{L}_1 = \begin{bmatrix} 32.5392 \\ 8.0845 \\ 51.7311 \end{bmatrix}, \hat{L}_2 = \begin{bmatrix} 42.1103 \\ 11.5032 \\ 68.6017 \end{bmatrix}.$$

Applying the designed fault observer to the electrical circuit system, the simulation results are exhibited in Figs. 2, 3 and 4. It is observed that the fault estimation is achieved by our observer, despite the presence of measurement uncertainties and external disturbances. This thus illustrates the effectiveness of the proposed approach for the observer design.

5.2 Numerical Example

Now, we adopt a general example to show the effectiveness of our control approach. Consider an FOS with $\alpha = 0.8$ and with three state variables and one input,

$$D^{0.8}x(t) = \begin{bmatrix} 2 & 2 & -1 \\ 2 & -2 & 1 \\ 4 & 3 & 2z(t) \end{bmatrix} x(t) + \begin{bmatrix} 1 \\ 2 \\ -z(t) \end{bmatrix} u(t) + \tilde{G}\omega_1(t),$$

$$y(t) = Cx(t) + \tilde{D}\omega_2(t), \tag{65}$$

where $z(t) = \sin(x_1(t)) + M$, and $M \in [-1, 0]$ is a uncertain parameter. It is described by (8) with the following rules.

Plant rule 1: IF $x_1(t)$ is $\theta_1(x_1(t))$, THEN

$$A_1 = \begin{bmatrix} 2 & 2 & -1 \\ 2 & -2 & 1 \\ 4 & 3 & 2 \end{bmatrix}, B_1 = \begin{bmatrix} 1 \\ 2 \\ -1 \end{bmatrix},$$

$$\tilde{G}_1 = \begin{bmatrix} 0.01 \\ 0 \\ 0.01 \end{bmatrix}, C_1 = \begin{bmatrix} 1 & -1 & 3 \\ 2 & 2 & 0 \end{bmatrix}, \tilde{D}_1 = \begin{bmatrix} 0.001 \\ 0.002 \end{bmatrix},$$

Plant rule 2: IF $x_1(t)$ is $\theta_2(x_1(t))$, THEN

$$A_2 = \begin{bmatrix} 2 & 2 & -1 \\ 2 & -2 & 1 \\ 4 & 3 & 2 \end{bmatrix}, B_2 = \begin{bmatrix} 1 \\ 2 \\ 1 \end{bmatrix}, \tilde{G}_2 = \begin{bmatrix} 0.01 \\ 0 \\ 0.01 \end{bmatrix},$$

$$C_2 = \begin{bmatrix} 1 & -1 & 3 \\ 2 & 2 & 0 \end{bmatrix}, \tilde{D}_2 = \begin{bmatrix} 0.001 \\ 0.002 \end{bmatrix}.$$

To test the fault tolerance and robustness of our approach, the following actuator fault is taken into account in the simulation:

$$f(t) = \begin{cases} 0, & 0 < t < 10, \\ 10\sin(t - 10), & \text{else.} \end{cases}$$

The external disturbance and measurement uncertainty are the same as those in last simulation example.

Following Theorems 1 and 2, we design an FTC scheme for the above system. We first construct a non-fragile observer for fault estimation. The variable gains in the observer in (18) are determined by

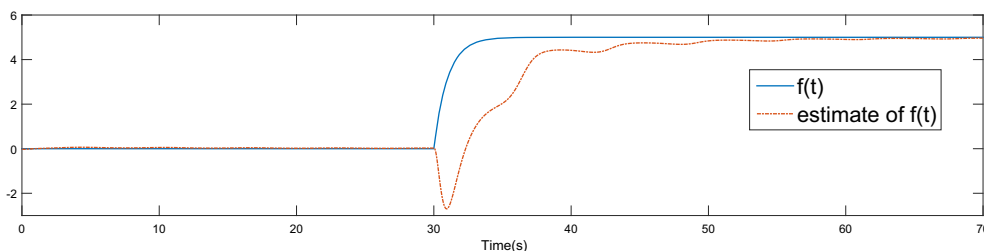


Fig. 2 The actuator fault and its estimate

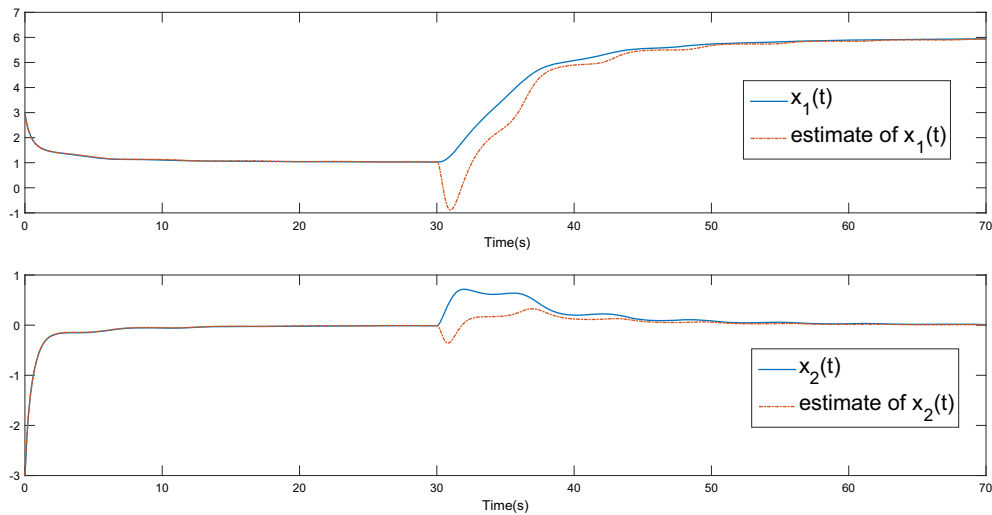


Fig. 3 The system state and its estimate

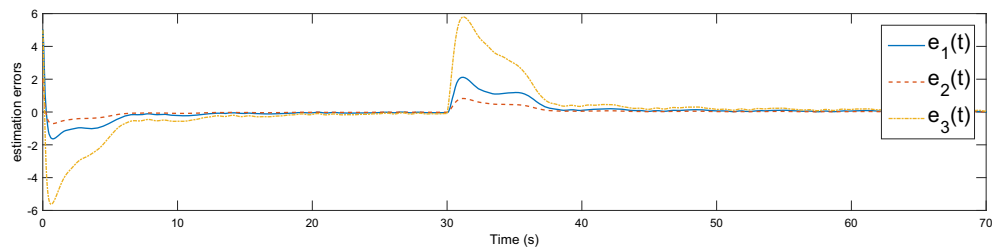


Fig. 4 The estimation errors

Table 1 Membership functions of the system

Lower membership functions	Upper membership functions
$\underline{\theta}_1(x_1) = \frac{\sin(x_1)+1}{3}$	$\bar{\theta}_1(x_1) = \frac{\sin(x_1)+2}{3}$
$\underline{\theta}_2(x_1) = \frac{2-\sin(x_1)}{3}$	$\bar{\theta}_2(x_1) = \frac{1-\sin(x_1)}{3}$

Table 2 Membership functions of the controller

Lower membership functions	Upper membership functions
$\underline{\eta}_1(x_1) = 0.2 - \frac{0.2}{1+e^{\frac{x_1-2.5}{4}}}$	$\bar{\eta}_1(x_1) = 0.2 - \frac{0.1}{1+e^{\frac{x_1-4.5}{4}}}$
$\underline{\eta}_3(x_1) = 1 - \bar{\eta}_1(x_1)$	$\bar{\eta}_3(x_1) = 1 - \underline{\eta}_1(x_1)$

$$U_{L1} = \begin{bmatrix} -0.1 \\ 0.2 \\ 0.3 \end{bmatrix}, \quad U_{L2} = \begin{bmatrix} -0.1 \\ -0.2 \\ 0.1 \end{bmatrix},$$

$$V_{L1} = [0.2 \quad 0.2], \quad V_{L2} = [0.3 \quad 0.1],$$

$$U_{F1} = 0.1, \quad U_{F2} = 0.5, \quad F(\sigma) = \sin(0.2\sigma)$$

$$V_{F1} = [-0.2 \quad 0.1], \quad V_{F2} = [0.2 \quad 0.2].$$

The lower and upper membership functions of the type-2 fuzzy model are listed in Table 1. Then, set the corresponding weights as $\bar{\lambda}_i = \underline{\lambda}_i = 0.5, i = 1, 2$. By (3), the membership functions of the type-2 fuzzy model can be obtained. The lower and upper membership functions of the type-2 fuzzy controller are listed in Table 2. Then, set the corresponding weights as $\bar{m}_k = \underline{m}_k = 0.5, k = 1, 2$. By (27), the membership functions of the type-2 controller can be obtained.

Set the H_∞ performance index as $\gamma_o = 0.1$, and then solve the LMIs in (21)–(23). The resulting observer gain matrices are

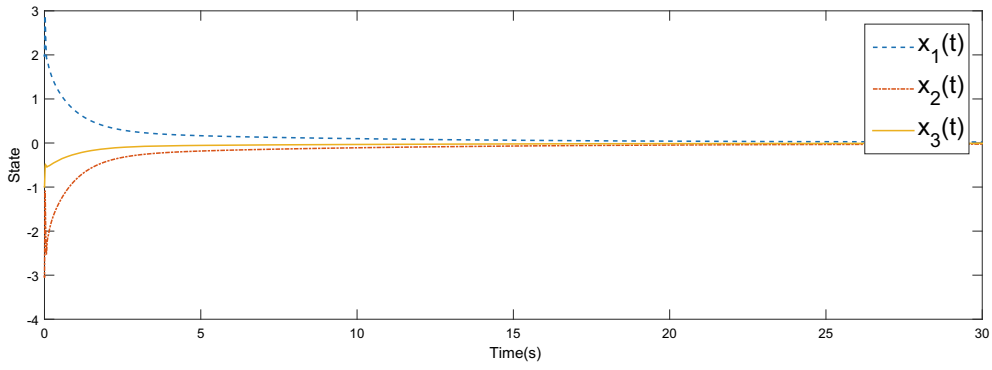


Fig. 5 The system state

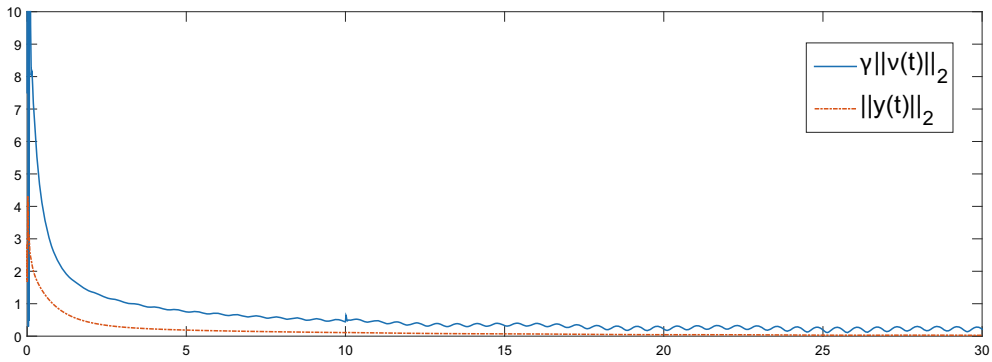


Fig. 6 The H_∞ performance with $\gamma = 2.5$.

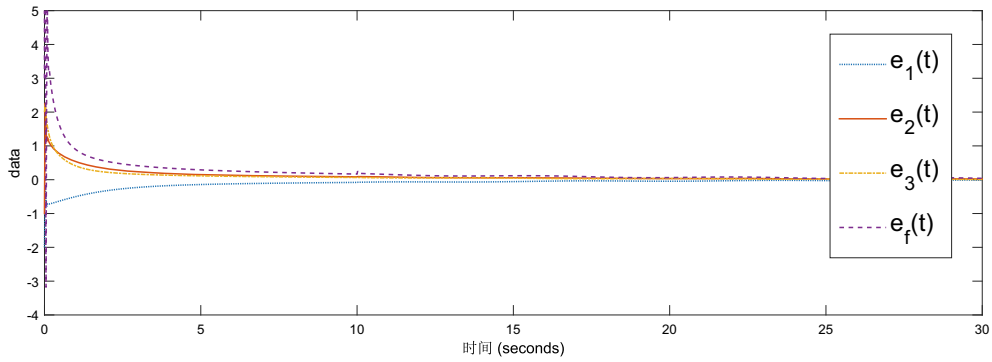


Fig. 7 The estimation errors

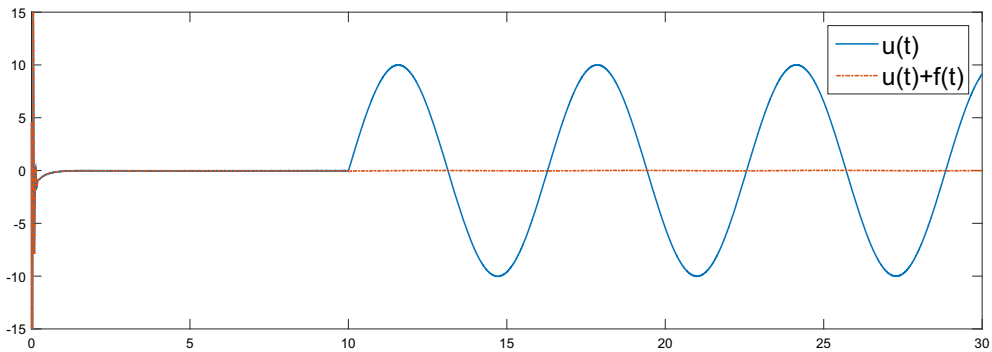


Fig. 8 The control signal and actuator output

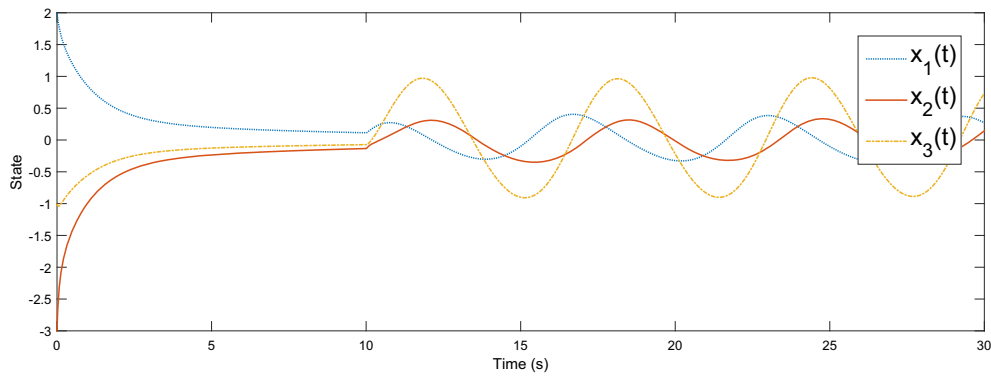


Fig. 9 The state response obtained by the comparative approach [17]

$$L_1 = \begin{bmatrix} -0.0298 & 1.3845 \\ 0.7548 & 1.3437 \\ 1.1579 & 2.1611 \end{bmatrix},$$

$$E_1 = [-158.3659 \quad 359.6916],$$

Next, we design an output feedback controller in the form of (26). Set $\gamma_c = 2.5$, $q_1 = q_2 = 0.2$, $q_3 = 0.3$, and calculate the LMIs in (29)–(32). The resulting controller gain matrices are

$$A_{c11} = \begin{bmatrix} -299.9633 & -188.8278 & -172.5259 \\ -556.1612 & -348.2801 & -326.0795 \\ -176.6531 & -51.5066 & -188.6263 \end{bmatrix},$$

$$A_{c12} = \begin{bmatrix} -302.6856 & -187.4891 & -178.6757 \\ -561.1038 & -345.9107 & -337.1585 \\ -178.3161 & -50.7541 & -192.2446 \end{bmatrix},$$

$$A_{c21} = \begin{bmatrix} -364.4329 & -215.8316 & -242.3319 \\ -680.7257 & -400.2992 & -461.2425 \\ -290.0705 & -98.8483 & -311.7352 \end{bmatrix},$$

$$A_{c22} = \begin{bmatrix} -367.3040 & -214.5241 & -248.7640 \\ -685.9513 & -397.9893 & -472.8587 \\ -291.9908 & -98.1497 & -315.8415 \end{bmatrix},$$

$$B_{c1} = \begin{bmatrix} 50.1776 & 117.5344 \\ 95.2037 & 214.6950 \\ 67.2869 & 65.0342 \end{bmatrix}, \quad B_{c2} = \begin{bmatrix} 52.1758 & 117.8556 \\ 98.7979 & 215.2911 \\ 68.4974 & 65.2640 \end{bmatrix},$$

$$C_{c1} = [-16.3667 \quad -6.8745 \quad -15.6860],$$

$$C_{c2} = [-16.4050 \quad -6.8826 \quad -15.7587].$$

Applying the above controller to the system under consideration, the simulation results are displayed in Figs. 5, 6, 7 and 8. Figure 5 shows that the stabilization of the FOS is achieved, in spite of the measurement uncertainty, the persistent disturbance, and the persistent fault. Specifically, Fig. 6 clearly presents that the system output meets the prescribed H_∞ performance index. Besides, Figs. 7 and 8 indicate the effectiveness of the designed observer and the boundedness of the computed control signal and the

actuator output, respectively. Therefore, the simulation results clarify and verify the theoretical findings established above.

For comparison, a non-fragile control design approach [17] for fractional order fuzzy systems is adopted. Apply it to the system in (65) under the same simulation condition. The result is displayed in Fig. 9, which shows that the state convergence is lost when $t \geq 10$ s, due to the actuator faults. This in turn illustrates the superiority of our control approach in enhancing the fault tolerance of the control system.

6 Conclusion

This paper presents an output feedback robust fault-tolerant control strategy for a class of interval type-2 fuzzy fractional order systems subject to the possible actuator faults. Its superiority over the existing approaches lies in three aspects. First, the stability domain of the faulty FOS is extended significantly. This is attributed to the introduction of the concept of D-stability to the control design and stability analysis, instead of the conventional indirect Lyapunov theory. Second, the resulting control system is robust against the measurement noise and external disturbances that, however, are not taken into account in the existing FTC designs for FOSs. This is achieved by adopting the H_∞ control method in a new way, in which the controller parameters are determined by solving real LMIs rather than complex matrix inequalities. As a byproduct, the FTC design is simplified. Third, our design approach is less restrictive than the existing ones: certain requirements for the system output matrix or the structure of the control gain matrices are eliminated. The simulation results on an electrical circuit system and a numeral example both illustrate the effectiveness of the proposed approach.

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University, Shenyang, China. His research interests include nonlinear control, marine vehicle control, fault diagnosis and fault-tolerant control, etc.



Xuefeng Zhang received the B.Sc. degree in applied mathematics, the M.S. degree in control theory and control engineering, and the Ph.D. degree in control theory and control engineering from Northeastern University, Shenyang, China in 1989, 2004, and 2008, respectively, where he is currently with the College of Sciences. He has published more than 100 journal



Kaijing Jin received the B.S. degree in information and computing science from Bohai University, Jinzhou, China, in 2018, and the M.S. degree in operations research and cybernetics from Northeastern University, Shenyang, China, in 2021, where she is currently pursuing the Ph.D. degree in control science and engineering. Her research interests include fractional order systems, Takagi-Sugeno fuzzy systems and singular systems.



Jin-Xi Zhang received the B.S. degree in automation and the Ph.D. degree in control theory and control engineering from Northeastern University, Shenyang, China, in 2014 and 2020, respectively. From 2019 to 2020, he was a Research Fellow with the Institute for Intelligent Systems, Faculty of Engineering and the Built Environment, University of Johannesburg, South Africa. He is currently a Distinguished Associate Professor with the State Key Laboratory of Synthetical Automation for Process Industries, Northeastern

University, Shenyang, China. His research interests include nonlinear control, marine vehicle control, fault diagnosis and fault-tolerant control, etc.