**TECHNICAL PAPER**



# **Premixed fame heat release‑based optimum global single‑step chemistry for H2, CH4, and C3H8 mixtures with air**

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## **Abstract**

Despite the signifcant development of comprehensive detailed chemical kinetics mechanisms for combustion simulation in the past decades, reduced descriptions of the chemical process still enable engineering and direct numerical simulations. This work proposes a new formulation for the heat release rate of laminar premixed fame which extends the classical onestep Arrhenius global kinetics using a kinetically controlled assumption for the scalar governing the fnal reaction steps. The proposed methodology considers that the heat release rate as a function of temperature is known and obtained via a freely propagating laminar premixed fame model together with various detailed chemical kinetics mechanisms. The resulting formulation involves three free parameters, one of which is the Zel'dovich number, whereas the new ones characterize the final heat release stages. The proposed formulation is characterized for  $H_2$ , CH<sub>4</sub> and C<sub>3</sub>H<sub>8</sub> mixtures with air and compared to several detailed chemical mechanisms, exhibiting errors between 6% and 15% for the frst of these and smaller than 6% for the hydrocarbons. The three free parameters are determined for a range of equivalence ratios and seem to be mixture properties. In particular, the Zel'dovich number exhibits a minimum around stoichiometry. The hydrocarbons fuels are also characterized when diluted by  $CO<sub>2</sub>$ , which is motivated by its presence in Brazilian pre-salt oil wells. The new free parameters are not infuenced by dilution, whereas the Zel'dovich number is found to increase.

**Keywords** Single-step chemistry · Premixed combustion · Laminar fames · Zel'dovich number · Heat release rate

**Mathematics Subject Classifcation** 80A25 · 80A32

# **1 Introduction**

Despite the decade-long quest for comprehensive detailed chemical kinetics mechanisms for gas-phase combus-tion [[11](#page-9-0), [18](#page-10-0)], affordable engineering predictions and detailed computational fuid dynamics direct simulations still resort to simplifed descriptions. In recent years, this has fostered, for instance, the development of virtual kinetics mechanisms [[5](#page-9-1)] that rely on an optimization of the

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thermodynamic, transport and reaction properties. Optimized single-step chemistry models have also been developed to reproduce the burned gases temperature, the laminar fame speed and the laminar fame thickness based on the temperature gradient for the simulation of turbulent premixed fame propagation [[10\]](#page-9-2). This has also been applied for studies of auto-ignition of heterogeneous mixtures, where the objective is to determine the ignition delay [[6\]](#page-9-3). These studies also provide an ample discussion on the diferent requirements that simplifed chemical models should satisfy. In the case of premixed fames, these requirements may be summarized as the model ability to reproduce the burned gas temperature, the fame speed and thickness [[10\]](#page-9-2) in a wide range of compositions, pressures and temperatures.

In the present study, an original single-step chemistry approximation method is presented, which involves a heat release rate expression from a global chemical reaction controlled by a progress variable based on temperature, augmented by a Arrhenius type relaxation towards equilibrium.

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This method enables the Zel'dovich number determination, regardless the mixture equivalence ratio, also. The basic underlying assumption is that the combustion heat release rate is a function of temperature, obtained via detailed chemical kinetic computations. Here, such computations involve the freely propagating one-dimensional laminar flame, which results provide, as input to the proposed single-step model, the heat release rate as a function of temperature. The new method efectiveness is demonstrated for hydrogen, methane and propane mixtures with air on the basis of a priori analysis of the temperature-based reactive scalar reaction rate, but the use in more complex combustion simulations of interest to the authors [\[7](#page-9-4), [9](#page-9-5), [16](#page-10-1)] is deferred to future works. Furthermore, CO<sub>2</sub> diluted hydrocarbons are also considered due to the interest in assessing the behavior of such mixtures for the sake of industrial applications.

Indeed, in a recent survey by ANP (The Brazilian Oil, Gas and Biofuels National Agency) [\[3](#page-9-6)], it is estimated that the so called pre-salt production will dominate the national oil and natural gas scenario. The composition of combustible gases from those pre-salt wells is considerably diferent from existing ones, mainly due to the presence of carbon dioxide  $(CO<sub>2</sub>)$  in its composition. Typically, the percentage of  $CO<sub>2</sub>$ in the hydrocarbon gas produced ranges from 0% to 80% [\[3](#page-9-6)], which justifes the choice of fuel diluent composition used here. Note that  $CO<sub>2</sub>$  may also be found in gases produced by mature wells, since it is injected to help displace the oils from the bedrock. To represent the fuel in such wells, methane and propane and their mixtures with air are considered.

The present manuscript is organized as follows, the proposed methodology is described in the next section (Sect. [2](#page-1-0)). In Sect. [3](#page-2-0), the results of the heat release expression are presented and discussed for the three considered fuels and different detailed chemistry mechanisms. It includes two subsections regarding to undiluted fuel/air mixtures and  $CO<sub>2</sub>$ diluted fuel/air mixtures, respectively. Finally, in Sect. [4](#page-8-0), the main observations are reported as a conclusion.

# <span id="page-1-0"></span>**2 Methodology**

A new formulation for a single scalar-based reaction rate is now presented. The proposed methodology considers that the heat release rate as a function of temperature is known,  $\dot{q}_{num}$ . In this work,  $\dot{q}_{num}$  is obtained using CHEKMIN's freely propagating laminar premixed fame code, together with various detailed chemical kinetics mechanisms, to be specifed further on. Therefore, for the sake of convenience,  $\dot{q}_{num}$  is expressed as a function of a progress variable,  $\theta$ , and a parameter,  $\gamma$ , associated with the chemical reaction heat release:  $\theta = (T - T_u)/T_b - T_u$  and  $\gamma = (T_b - T_u)/T_u$ , where *T* is the temperature,  $T<sub>b</sub>$  is the adiabatic combustion temperature

and  $T_{\mu}$  is the unburned gases temperature. A chemical equilibrium calculation code is used to determine  $T_b$ .

Under the hypotheses that the chemical process is governed by a single-step reaction controlled by the defcient reactant, the heat release rate may classically be written as [\[8](#page-9-7)]

<span id="page-1-1"></span>
$$
\dot{q}(\theta, Y) \propto Y^n \exp\left\{ \frac{\beta(\theta - 1)}{(1 + \gamma \theta)/(1 + \gamma)} \right\}.
$$
 (1)

In this equation, *Y* is the normalized mass fraction of the deficient reagent—the fuel in lean mixtures and the oxidizer in rich ones—*n* is usually identifed as the reaction order,  $\beta = E_a(T_b - T_u)/RT_b^2$  is the Zel'dovich number and, thus, the exponential term corresponds to the Arrhenius law for a global reaction. The hypotheses used to derive this equation are not valid close to stoichiometry, in particular because most combustion chemical reactions may not be assumed to be controlled by a single reactant. In the case where the Lewis number of this controlling reagent is one, it is possible to write that  $Y = 1 - \theta$ , and thus  $\dot{q}(\theta, Y) \rightarrow \dot{q}(\theta)$  only.

It is worth emphasizing that the hypotheses that lead to the derivation of Eq.  $(1)$  $(1)$  $(1)$  are not valid in the stoichiometry region. To illustrate this limitation, Fig. [1](#page-1-2) shows the comparison between  $\dot{q}_{num}$  and  $\dot{q}$  for a stoichiometric methane/air mixture, in which  $n = 1$  is presumed and  $\beta$  was obtained by an optimization process that aims to minimize the Euclidean norm  $||\dot{q}_{num} - \dot{q}||$ . Here,  $\dot{q}_{num}$  has been obtained from simula-tion results using the GRI 3.0 mechanism [[14\]](#page-10-2). In this figure, and throughout this work, the heat release rate is normalized using the corresponding integrals:

$$
I_q = \int_0^1 \dot{q} \, \mathrm{d}\theta,\tag{2}
$$

which is found to be a convenient way to compare the heat release rate for various fuels and equivalence ratios.

<span id="page-1-3"></span>

<span id="page-1-2"></span>**Fig. 1** Normalized heat release rate as a function of the progress variable for a stoichiometric methane/air mixture. Result of simplex opti-mization of Eq. ([1](#page-1-1)) concerning a variable  $(\beta)$  with  $n = 1$ . The solid curve represents the simulation with the GRI 3.0 mechanism [\[14\]](#page-10-2) and the dashed curve represents the optimization results

Figure [1](#page-1-2) clearly shows that the simplifed global reaction rate does not correctly describe *q̇ num*.

Indeed, the initial slope of  $\dot{q}_{num}(\theta)$  is smaller than that of  $\dot{q}(\theta)$ , indicating that the numerical value of  $\beta$  is smaller than that obtained using Eq. ([1\)](#page-1-1). Note that the Zel'dovich number,  $\beta$ , controls the increase of  $\dot{q}(\theta)$  in the lower temperature range, i.e., for  $\dot{q} \leq \dot{q}_{max}$ , where  $\dot{q}_{max}$  is the maximum heat release rate. Furthermore, both the maximum heat release rate and the relaxation toward the equilibrium when  $\theta \rightarrow 1$  are not well described by the simple functional dependency. This is not surprising, given that the assumptions involved in the derivation of Eq. [\(1](#page-1-1)) at the vicinity of the stoichiometry do not hold.

To better describe the variation of  $\dot{q}_{num}$  when  $\theta \rightarrow 1$ , in this paper, the following expression is proposed:

$$
Y = 1 - \exp[\beta'(1 - 1/\theta)].
$$
\n(3)

This expression considers that the limiting normalized reactant scalar, *Y*, is controlled by an Arrhenius type dependency:

$$
\frac{d\ln(Y-1)}{d(1/\theta)} = -\beta',\tag{4}
$$

which introduces a first relaxation parameter,  $\beta'$ , that aims to approximate the decrease of  $\dot{q}_{num}$  with  $\theta$  from its maximum value. Under the hypothesis that equilibrium is reached, i.e., that  $Y(\theta = 1) = 0$ , the integration of Eq. ([4\)](#page-2-1) leads to Eq. [\(3](#page-2-2)). Note, also that  $Y(\theta \to 0) \to 1$ . As a consequence, the initial stages of the chemical reaction are still controlled by the classical Arrhenius dependency. Thus, *Y* can be interpreted as being representative of the chemical species that govern the tendency to equilibrium in the fnal stages of reaction. This  $\beta'$  parameter is then associated to the heat release during the fnal stages of the global reaction.

Furthermore, in the present formulation, *n* is not interpreted as the reaction order, but as a free parameter loosely related to the complexity of the tendency of the chemical process towards equilibrium. More precisely, when *n* increases the reaction rate derivative with respect to  $\theta$  becomes smaller, and thus the tendency towards equilibrium is slowed. Note that this formulation guarantees that  $Y(\theta = 1) = 0$ , i.e., the global reaction rate is zero at equilibrium.

Therefore, to approximate the behavior of  $\dot{q}_{num}(\theta)$ , the following expression is proposed:

$$
\dot{q}(\theta) \propto \{1 - \exp[\beta'(1 - 1/\theta)]\}^n \exp\left\{\frac{\beta(\theta - 1)}{(1 + \gamma \theta)/(1 + \gamma)}\right\},\tag{5}
$$

where  $\beta$ ,  $\beta'$  and *n* are parameters to be determined. The new parameters,  $\beta'$  and  $n$ , control, respectively, the decrease of  $\dot{q}$  for values of  $\theta$  immediately greater than that corresponding to  $\dot{q}_{max}$  and the behavior of  $\dot{q}(\theta)$  close to  $\theta \to 1$ . Since these two parameters are characteristic of the fnal stages of combustion, it is expected that, within the limits of very

lean or very rich mixtures, they depend less on the nature of the fuel and mainly on the mixture C/H/O ratio. To determine  $\beta$ ,  $\beta'$  and *n* an optimization, aiming to minimize the norm  $||\dot{q}_{num} - \dot{q}||$  uses the basic simplex method [[4,](#page-9-8) [15](#page-10-3)]. In all cases the results are verifed to ensure that a global minimum error is attained.

## <span id="page-2-0"></span>**3 Results and discussion**

<span id="page-2-2"></span>The determination of the reaction rate parameters is carried out for H<sub>2</sub>/air, CH<sub>4</sub>/air/CO<sub>2</sub> and C<sub>3</sub>H<sub>8</sub>/air/CO<sub>2</sub> mixtures. For each mixture and undiluted fuel are considered at least two diferent kinetics mechanism: Keromnes Mech [[12](#page-9-9)] and Alekseev Mech [[2](#page-9-10)] for hydrogen mixtures, GRI 3.0 [[14](#page-10-2)], San Diego Mech [[1\]](#page-9-11) and USC II Mech [[17](#page-10-4)] for methane mixtures and San Diego Mech and USC II Mech for propane mixtures. Subsequently, for  $CO<sub>2</sub>$  diluted fuel mixtures, the GRI 3.0 is considered for methane, and San Diego Mech for propane. These mechanisms are chosen as representative of the combustion chemical kinetics of these diferent fuels [[11](#page-9-0), [13\]](#page-10-5). For the sake of simplicity, a pressure of 1 bar and an unburned gases temperature of 300 K are considered, only.

<span id="page-2-1"></span>To provide a frst assessment of the proposed formulation behavior for methane/air mixtures, Fig. [2a](#page-3-0)–c, in which are shown the heat release rates,  $\dot{q}_{num}$  and  $\dot{q}$ , normalized by their integrals, illustrates one of the worst and best cases of optimization obtained with the new formulation, which have 4 %, 3 % and 0.9 % of error, respectively.

In this fgure, the solid curve represents the numerical result obtained with kinetic mechanisms,  $\dot{q}_{num}$ , and the dashed curve represents the optimization resulting from the use of Eq.  $(5)$  $(5)$ . The analysis of this figure underscores the excellent agreement between  $\dot{q}_{num}$  and  $\dot{q}$  during the initial phase of the chemical reaction, i.e.,  $\dot{q} \leq \dot{q}_{max}$ . Furthermore, the Zel'dovich number,  $\beta$ , that describes the initial phases of the chemical heat release, is larger for both lean and rich mixtures than for stoichiometric mixtures. Indeed, the latter is characterized by a more gradual and earlier heat release in comparison with the lean and rich mixtures. As will be demonstrated by the analysis below, minimum values of the Zel'dovich number arise around stoichiometry. In addition, the decrease of  $\dot{q}$  after  $\dot{q}_{max}$  is also very well described by the formulation proposed here. It is also observed that the most significant discrepancies occur for  $\theta \rightarrow 1$ , that is, in the region close to the chemical equilibrium. In the proposed formulation, this behavior is controlled by the parameter *n*. These findings suggest that  $\beta'$  and  $n$  are in fact properties of the chemical reaction, also.

<span id="page-2-3"></span>A systematic analysis for three diferent fuel/air mixtures is now performed to further explore this formulation results.

<span id="page-3-0"></span>**Fig. 2** Normalized heat release rate as a function of the progress variable for methane/ air mixtures. The solid curve represents the simulation with GRI 3.0 mechanism [\[14\]](#page-10-2) and the dashed curve represents the Eq. ([5\)](#page-2-3) optimization results



(a) Lean mixture,  $\phi = 0.6$ , parameters:  $\beta = 5.2$ , (b) Stoichiometric mixture,  $\phi = 1.0$ , parame- $\beta' = 16.6$  and  $n = 6.2$ . ters:  $\beta = 2.3$ ,  $\beta' = 17.7$  and  $n = 105$ .



(c) Rich mixture,  $\phi = 1.4$ , parameters:  $\beta = 3.9$ ,  $\beta' = 11.8$  and  $n = 1.8$ .

#### **3.1 Undiluted fuel/air mixtures**

Figure [3](#page-4-0)a–c presents the heat release rate comparison between  $\dot{q}$  calculated with Eq. ([5](#page-2-3)) and that by considering the different detailed chemical kinetics mechanisms for  $H_2$ ,  $CH_4$  and  $C_3H_8$ , respectively.

The heat release rate is depicted in situations where the worst agreement between  $\dot{q}$  and  $\dot{q}$ <sub>num</sub> is observed, i.e.,  $\phi = 3.0, \phi = 1.2$  and  $\phi = 0.9$  for H<sub>2</sub>, CH<sub>4</sub> and C<sub>3</sub>H<sub>8</sub> mixtures with air, respectively. For the three diferent undiluted fuel/air mixtures, the qualitative agreement between *q̇* and  $\dot{q}_{num}$  is remarkable. The comparison between these results indicate that, for  $H_2$ /air, some discrepancies exist between  $\dot{q}$  and  $\dot{q}_{num}$  for the different chemical kinetic mechanisms, whereas the results obtained for  $CH_4$  and  $C_3H_8$  are nearly identical. This good agreement is seen even for the rich  $H_2$ /air mixtures, which exhibits a peak of  $\dot{q}$  for  $\theta \approx 0.3$ , i.e., signifcantly departed from the burned gases.

To quantify the errors associated with the present formulation throughout an equivalence ratio range, the Euclidean norm of the heat release rate error between the numerical simulation with detailed kinetic mechanisms and the proposed methodology is presented in Fig. [4a](#page-4-1)–c.

For H<sub>2</sub>/air mixtures, the error varies from 6  $\%$  to 16  $\%$  and both chemical mechanisms exhibit the same error bias with equivalence ratio, albeit with diferent values. A remarkable feature concerning  $CH<sub>4</sub>/air$  mixtures is the nearly identical behavior and value of the error with equivalence ratio, which is found to occur for rich propane mixtures also, but not for lean ones. Indeed, for CH<sub>4</sub>/air mixtures and  $C_3H_8$  mixtures the error is smaller than for the  $H_2/air$  mixtures with values varying from 0.5 % to 4 % and 2 % to 6.5 %, respectively.

To further characterize the proposed reaction rate formation, Figs. [5](#page-5-0)a, [6](#page-5-1)a and [7a](#page-6-0) present the results of the Zel'dovich number,  $\beta$ , as a function of the equivalence ratio,  $\phi$ , for H<sub>2</sub>/ air, CH<sub>4</sub>/air and C<sub>3</sub>H<sub>8</sub>/air mixtures. The Zel'dovich number is seen to exhibit a non-monotonic behavior with a minimum value at the stoichiometric region. In all the cases, there are rather small diferences among the kinetics mechanisms results, with the most signifcant lying at the very rich equivalence ratio for methane mixtures. This agreement between chemical mechanisms is an indication that  $\beta$  is a property for the considered fuel/air mixtures. The discrepancies between the computed  $\beta$  values for methane/air mixtures at  $\phi = 1.6$  is associated to a heat release,  $\dot{q}_{num}$ , that occurs slightly earlier when the GRI 3.0 is compared to the San Diego and the USC II mechanisms. For the sake of brevity, these results are not reported here.

It should be emphasized that the proposed formulation does not exhibit singular behavior in the stoichiometry region, enabling  $\beta$  to be determined across the mixtures fammability range. For hydrogen/air mixtures, the

<span id="page-4-1"></span><span id="page-4-0"></span>

<span id="page-5-0"></span>**Fig. 5** Results for  $H_2$ /air mixtures with two diferent kinetics mechanism: Keromnes Mech (◇) and Alekseev Mech (×). Behavior as a function of the equivalence ratio of **a** Zel'dovich number, **b**  $\beta'$ parameter, **c** *n* parameter, and **d** heat release rate normalization integral, *Iq*



<span id="page-5-1"></span>**Fig. 6** Results for  $CH<sub>4</sub>/air mix$ tures with three diferent kinetics mechanisms: GRI 3.0 (hexagram), San Diego Mech (+) and USC Mech II (∗). Behavior as a function of the equivalence ratio of **a** Zel'dovich number,  $\beta$ , **b**  $\beta'$ parameter, **c** *n* parameter, and **d** heat release rate normalization integral, *Iq*



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<span id="page-6-0"></span>**Fig. 7** Results for  $C_3H_8/air$ mixtures with two diferent kinetics mechanism: San Diego Mech (+) and USC Mech II (∗). Behavior as a function of the equivalence ratio of **a** Zel'dovich number,  $\beta$ , **b**  $\beta'$ parameter, **c** *n* parameter, and **d** Heat release rate normalization integral,  $I_a$ 



Zel'dovich number varies between 0.8 and 3, whereas it is possible to observe that for methane/air mixtures, the Zel'dovich number varies between 2 and 5.2. For propane mixtures, the Zel'dovich number varies between 2 and 11. The Zel'dovich number for  $\phi = 1$  is  $\beta = 1.0$ , 2.3 and 2.47 for mixtures of hydrogen, methane and propane, respectively. It is interesting to note that, except for very lean or very rich mixtures, the computed Zel'dovich number cannot be considered "large".

Figures  $5b$ ,  $6b$  and  $7b$  show the behavior of the  $\beta'$  parameter, which is defned in Eq. ([4\)](#page-2-1), as a function of the equivalence ratio, for the diferent fuels and baseline detailed chemical kinetic mechanisms.

For each of the fuels considered,  $\beta'$  exhibits a quite different behavior. In the case of hydrogen/air mixtures,  $\beta'$  varies between 0.1 and 1. For these mixtures,  $\beta'$  exhibits a behavior analogous to the Zel'dovich number, i.e., with a minimum around the stoichiometry region. For methane/air mixtures  $\beta'$  varies between 2 and 30 being nearly constant for  $\phi \leq 1$ , but increasing, then decreasing for rich mixtures. Indeed, for very lean mixtures, the values vary between 15 and 20 depending on the choice of chemical kinetic mechanism. The signifcant increase occurs in the region with equivalence ratio between 1 and 1.2. Then, from  $\phi = 1.2$ ,  $\beta'$  decreases to 1.6. Nevertheless, for hydrogen/air and methane/air mixtures, the qualitative variation of  $\beta'$  with  $\phi$  also suggests this could be a property of the mixtures. However, a similar

agreement between the diferent chemical mechanism is not seen in Fig. [7](#page-6-0)c for the propane/air mixtures, except for  $\phi \geq 1.2$ . Indeed, for mixtures of propane/air, the values of  $\beta'$  vary between 2 and 20, for the USC II Mech only and exhibits a non-monotonic behavior between equivalence ratio 0.6 and 1.2 and, subsequently, a decreasing behavior from equivalence ratios 1.3 up to 1.6. Regarding the different kinetics mechanisms used, the main diferences are observed for the hydrocarbon mixtures, i.e., for the methane mixtures around  $\phi = 1.2$  and propane mixtures for  $\phi \leq 1.2$ . Such a qualitative agreement between the diferent chemical kinetic mechanisms results suggest that  $\beta'$  is a property of the studied mixtures also.

The behavior of the parameter *n* with the mixture equivalence ratio is shown in Figs. [5c](#page-5-0), [6c](#page-5-1) and [7](#page-6-0)c for the diferent fuels and kinetic mechanism studied.

In the case of hydrogen/air mixtures, *n* varies from 1.2 to 1.8, peaking at the vicinity of stoichiometry. Even if the qualitative behavior is similar, quantitative diferences can be seen for the two chosen mechanisms values for  $\phi \geq 1$ . Furthermore, for  $H_2/air$ , *n* assumes values significantly smaller than those for the methane and propane mixtures. Indeed, it is possible to observe that, for methane/air mixtures, *n* non-monotonically varies approximately between 1 and 120, with the highest values close to stoichiometry. For  $\phi \geq 1.3$ , the value is almost constant, with  $n \approx 1$ . For the propane/air mixtures, when the San Diego Mechanism is used,

the behavior of *n* is constant for  $0.6 < \phi < 1.1$ . Beyond this range, *n* shows a decreasing behavior up to equivalence ratio 1.2. As with methane mixtures, for  $\phi \geq 1.3$ , it presents a constant behavior with values of  $n \approx 1$ .

For the sake of completeness, the heat release rate integral, Eq. ([2\)](#page-1-3), is given in Figs. [5](#page-5-0)d, [6d](#page-5-1) and [7](#page-6-0)d, where the results for the mixtures of hydrogen/air, methane/air and propane/air are represented, respectively. The adequate description of  $I_a$  is indeed crucial for practical flame computations, as is the functional description of the thermodynamic and transport properties, which may be efected using polynomial fts as well, but are not object of this study. As it could have been expected, in these fgures, it is possible to observe that the integral of the heat release rate has a behavior similar to the laminar fame front speed (not given for the sake of brevity), with the maximum in the stoichiometric region.

It is worth mentioning that fits to the above studied parameters are given in the supplementary material.

## **3.2 CO2 diluted hydrocarbon/air mixtures**

To further assess the variation of Eq. [\(5\)](#page-2-3) parameters with the mixture studied,  $CO<sub>2</sub>$  diluted methane or propane are now considered. The chemical kinetic mechanisms of GRI 3.0 Mech, USC II Mech and San Diego Mech are considered. The specifc choice of dilution percentage is motivated by the relevance of such diluted fuels to the gas mixtures

produced by mature oil and gas wells, for instance. Therefore, the  $CO<sub>2</sub>$  percentages used here refers to the fuel, not the mixture, dilution.

Figures [8](#page-7-0)a and [9](#page-8-1)a present the results found for the Zel'dovich number in mixtures of methane/air/diluent and propane/air/diluent with the fuel diluent varying between 0 % and 75 %.

For both fuel/air mixtures used,  $\beta$  increases with the fuel dilution percentage. For instance, for the methane mixtures, the minimum ( $\phi = 1$ ) Zel'dovich number is 2.3, 2.63 and 3.4 for  $0\%$ ,  $25\%$  and  $50\%$  CO<sub>2</sub> dilution, respectively. In addition, in the propane mixtures, the values for 0 %, 25%, 50% and 75% are 2.47, 3.03, 3.42 and 4.64, respectively.

It is interesting to note that both parameters  $\beta'$  and  $n$ , in Figs. [8](#page-7-0)b and c and [9b](#page-8-1) and c, are slightly dependent on the dilution of the fuel by  $CO<sub>2</sub>$  in both methane mixtures and propane mixtures.

Furthermore, it is notable that for  $\phi \ll 1$  or  $\phi \gg 1$ , the values of  $\beta'$  are very similar for both fuel mixtures. Unsurprisingly, the heat release rate integral decreases with the fuel dilution percentage by  $CO<sub>2</sub>$  throughout the equivalence ratio range, as is evidenced in Figs. [8](#page-7-0)d and [9d](#page-8-1).

For the sake of completeness, the Euclidean norm of the heat release rate error between the numerical simulation with detailed kinetic mechanisms and the proposed methodology is presented in Fig. [10](#page-8-2)a and b. For methane/air/ diluent mixtures, the error varies between 1 % and 5 %. For

<span id="page-7-0"></span>**Fig. 8** Results for  $CH<sub>4</sub>/CO<sub>2</sub>/$ air mixtures as a function of the equivalence ratio. **a** Zel'dovich number,  $\beta$ , **b**  $\beta'$  parameter, **c** *n* parameter, and **d** heat release rate normalization integral,  $I_a$ 



<span id="page-8-1"></span>**Fig. 9** Results for  $C_3H_8/CO_2/$ air mixtures as a function of the equivalence ratio. a Zel'dovich number,  $\beta$ , b  $\beta'$  parameter, c *n* parameter, and d heat release rate normalization integral,  $I_a$ 



<span id="page-8-2"></span>

propane/air/diluent mixtures, it varies between 2 % and 7 %. A remarkable feature is that the error value is nearly independent of the fuel dilution percentage.

# <span id="page-8-0"></span>**4 Conclusion**

A new formulation for a single-step reaction rate is proposed that relies on a single scalar, a normalized temperature, only. This is achieved by augmenting the classical Arrhenius reaction rate with a generalized scalar that describes the tendency to equilibrium, thus avoiding the classical hypothesis of a limiting reactant. This formulation is controlled by three non-dimensional parameters: (1) the Zel'dovich number, (2) one describing the heat release rate initial decay beyond the maximum value and (3) another controlling the fnal stages of the heat release process. The equilibrium obtained temperature increase, which is a thermodynamic property, is also accounted for in the formulation. The three free parameters are obtained via an optimization that minimizes the error with respect to the heat release rate of a laminar premixed fame expressed in terms of a temperature-based reactive scalar. This rate is here determined numerically using detailed chemical kinetics mechanisms.

The proposed formulation was shown to qualitatively describe the heat release rate as a function of this scalar for three different fuels,  $H_2$ , CH<sub>4</sub> and C<sub>3</sub>H<sub>8</sub>, and for several baseline detailed chemistry mechanisms. Furthermore, the

Euclidean norm of the heat release rate error between the numerical simulation with detailed kinetic mechanisms and the proposed formulation was determined for equivalence ratios ranging between 0.6 and 1.6. For methane mixtures, this error varies between 1 % and 5 %. For propane mixtures, it varies between 2 % and 7 %. The  $H_2$  mixtures exhibits errors between 6 % and 15 %. Such error values are similar, in magnitude, with those with which laminar fame speeds are measured.

For all the mixtures studied, the heat release rate integral behaves similarly to the laminar fame speed, with the maximum in the stoichiometric region decreasing with dilution addition. The computed Zel'dovich number exhibits with a minimum at the stoichiometric region and increases with dilution  $CO<sub>2</sub>$  for all fuels studied.

The first new parameter,  $\beta'$ , presents a different behavior for the fuels studied. Indeed, for  $H_2$ , it is similar to that of the Zel'dovich number, but the same was not seen for the hydrocarbons/air mixtures. For the latter, this parameter is independent of  $CO<sub>2</sub>$  dilution. The second new parameter,  $n$ , also shows a diferent behavior for all fuels, but is not significantly influenced by  $CO<sub>2</sub>$  dilution. This parameter also shows values signifcantly greater for hydrocarbon mixtures than for hydrogen mixtures, i.e., for methane, it presents a maximum of the order of 110 and for hydrogen, it presents a maximum around 1.8. The obtained results suggests that the new formulation parameters could be considered as properties of the reactive mixtures and not numerical artifacts.

The proposed formulation is still to be applied to solving combustion problems involving the studied gas mixtures. Interesting questions for further studies are the extension to diferent pressure ranges, to fuels consisting of a mixture of species, such as  $H<sub>2</sub>/CO$ , and to larger molecular weight hydrocarbons, which may exhibit a negative temperature coefficient dependency. Furthermore, a posteriori testing of the developed reaction rate formulation is still to be performed, for instance, by determining the laminar fame speed and thickness of the studied mixtures. The ability of the proposed single-step formulation to reproduce those crucial fame properties requires the development correlations that describe the mixtures' thermal properties, which is currently underway.

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**Code availability** Not applicable.

#### **Declarations**

**Conflict of interest** Not applicable.

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