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Numerical studies of laminar fow over two tandem elliptical cylinders using Ramanujan approximation

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Abstract

This paper studies a laminar fow over tandem elliptical cylinders at Re=200. While the aspect ratio (*AR*) of the upstream cylinder varies from $AR = a_1/b_1 = 0.25$ to 2.00, the aspect ratio of the downstream cylinder is kept constant at $AR = 1.00$ (e.g., $a_2/b_2 = 1$). This range of *AR* covers the most important practical cross sections of elliptical cylinders, including normal elliptic cylinder (with 90° incidence) and parallel elliptic cylinders (with 0° incidence). Although the spacing ratio between the centers of the cylinders is kept constant at $L^* = 4D_2$, the gap ratio $(G^* = G/D_2)$ between the surfaces of the cylinders is varied due to the alteration of *AR* of the upstream cylinder. Unlike the previously published studies, which estimated the hydraulic diameter of the elliptical cylinder, in this paper, the precise hydraulic diameter is evaluated and used to analyze the wake instabilities and the variation of the imposed pressure as well as forces coefficients on the cylinders. The results reveal that with the estimation of the hydraulic diameter of the elliptic cylinder, the maximum error of 178% has arisen, which significantly affects forces (lift and drag) coefficients. It was found that the phase lag between the sinusoidal lift coefficients of the cylinders varies and it reaches a minimum at $AR = 1.5$ and it slightly increases once $AR = 2.0$. Besides, the amplitude of the fluctuating drag coefficient is larger for the upstream cylinder as compared with that of the downstream cylinder. Besides, it is found that the minimum St=0.065 occurs at $AR = 0.25$, and it is regularly increased to the maximum value of $St = 0.211$ at $AR = 1.75$. A parabolic equation is deduced with high accuracy and a reasonable error of less than 1.8% to show the relationship between St and *AR*.

Keywords Tandem elliptic cylinders · Hydraulic diameter · Vortex shedding · Aspect ratio · Pressure distribution · Wake structure

List of Symbols

Latin alphabet

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1 Introduction

Laminar fow past tandem circular cylinders has been the subject of many studies due to the wide range of its applications. This typical arrangement can be widely applied in engineering problems, which are not limited to fow over buildings, heat exchangers and cooling systems, marine risers, pipelines lay on the sea-bed, ofshore structures, and even electronic chips on the mainboard.

Useful correlations between vortex frequency and aerodynamic instabilities are available in the literature for tandem circular cylinders. Nevertheless, a satisfactory study of the flow structure concerning the fluidic parameters of two tandem elliptical cylinders has not yet been reached in detail. In this study, numerical investigations are performed over tandem elliptical cylinders. The fow structure, the location of the separation on the upstream cylinder, the characteristics of the flow such as the drag and lift coefficients, and the Strouhal number are deeply investigated.

Notably, it is considered how the frequency of the vortices alters and afects the wake structure of the downstream cylinder. It has been shown that the fow around tandem cylinders highly depends on not only the spacing ratio between the centers of the cylinders [[1–](#page-14-0)[7](#page-14-1)] but also the cross section shape of the cylinder $[8-15]$ $[8-15]$. Zdravkovich [[3\]](#page-14-4) observed three regimes based on the spacing between the cylinders' centers ratio (here, for two circular cylinders, $L^* = L/D_1 = L/D_2$. The first regime or overshoot was identifed when the spacing between the centers of the cylinders was *L*[∗]<1.2. Because of a small spacing ratio, the generated positive and negative upstream vortices are stagnant between the cylinders. While L^* was set to $1.2 < L^* < 3.5$, the second regime known as the reattachment regime can

be effortlessly developed. In this condition, the generated upstream vortices have a chance to reattach the downstream cylinder. Finally, Zdravkovich [\[3](#page-14-4)] found by setting centers of the cylinder in 3.5<*L*[∗]<4, the co-shedding regime can be created, in which the upstream vortices are only rolled into the gap between the cylinders. Later, these regimes were identifed, confrmed, and even subcategorized in more regimes by other scholars [[4,](#page-14-5) [16](#page-14-6)[–22](#page-14-7)]. Numerical results of Alam [[22\]](#page-14-7) indicated that when the spacing ratio is set to 3.75 $\leq L^*$, the frequency of vortices increases and the Strouhal number (St) sharply jumps from 0.12 to 0.175. The spacing ratio L^* = 4 plays a significant role not only in the dynamic behavior of the upstream vortices but also in generating different flow regimes. Therefore, $L^* = 4$ is chosen in this study to investigate the fow structure around the tandem elliptical cylinders.

Considering the cross section of the cylinders, in general, the geometry of cylindrical structures vary, and they can be aerodynamically categorized into three primary geometry [[14,](#page-14-8) [23\]](#page-14-9): (1) circular and elliptical cylinders with a rounded surface, in which the fow separation may oscillate over a segment of the cylinder's surface, and it is not fxed; (2) sharp-edged surfaces with a fixed flow separation point; (3) combination of both geometries (1) and (2), such as D-shape cross-sectional structures. Though circular cylinders [\[24](#page-15-0)[–31\]](#page-15-1) and sharp-edged cylinders [[12](#page-14-10), [15](#page-14-3), [32](#page-15-2)[–40\]](#page-15-3) have received much attention in the literature, the studies of fow structure over elliptical [[11,](#page-14-11) [41](#page-15-4)[–47](#page-15-5)] and D-shape structures [[9,](#page-14-12) [48](#page-15-6), [49\]](#page-15-7) are limited to fewer examinations.

The flow past tandem cylinders with sharp edges such as square have been studied in great detail due to their relevance to a wide range of engineering applications [[50–](#page-15-8)[52\]](#page-15-9) and less attention has been received for tandem rectangular cylinders [[53](#page-15-10), [54\]](#page-15-11). It has been shown by Derakhshandeh and Alam [\[14\]](#page-14-8) and Chatterjee and Amiroudine [\[52\]](#page-15-9) that the vortex structure in the wake of the cylinders with sharp edges behaves diferently as compared with tandem circular cylinders. One of the main reasons is related to the fact that the cylinders with sharp edges have a fxed stagnation point causing diferences in critical regimes. The separation mechanism in tandem square cylinders alters the vortex frequency at a range of $Re = 55-60$ due to the change of flow pattern in the wake of the tandem cylinders. The numerical studies performed by Sohankar and Etminan [[51](#page-15-12)] showed that by increasing Re, the drag coefficient of the upstream square cylinder remains constant while that of the downstream cylinder gradually increases due to the altering the fow structure in spacing between the cylinders.

Using elliptical structures, on the other hand, is becoming more common these days due to their structural strength [[45](#page-15-13)]. Since the fow over the elliptical cylinders can be afected by geometrical parameters such as Aspect Ratio $(AR = a₁/b₁$ $(AR = a₁/b₁$ $(AR = a₁/b₁$ see Fig. 1) and the angle of attack of the elliptical Fig. 1 Schematic of the flow past a tandem elliptical cylinders, including the computational domain and the boundary conditions

cylinder, the wake structure of vortices become more complex as compared with tandem circular cylinders. Here, a_1 and b_1 are defined as the parallel and normal radiuses of the cross section of the elliptic cylinder relative to the freestream velocity.

Flow over a normal and parallel elliptic cylinder to the flow was studied by Johnson et al. [\[41\]](#page-15-4) and Raman et al. [[46](#page-15-14)], respectively. The selected *AR* in both studies was 0<*AR*<1.0. While Johnson et al. [\[41\]](#page-15-4) selected a smaller range of Reynolds numbers, e.g., 30 < *Re* < 200 in their studies, Raman et al. [\[46](#page-15-14)] focused on a relatively wider range of Re (50 < Re < 500). Here, $Re = \rho U d_h / \mu$, where ρ and *U* are the density and velocity of the fluid, d_h stands for the hydraulic diameter and μ represents the dynamic viscosity of the fuid. Johnson et al. [\[41\]](#page-15-4) in their numerical studies found that for the *Normal Elliptic Cylinder* (*NEC* or with 90° incidence) by decreasing *AR*, the vortex shedding in the wake of the elliptical cylinder occurs differently as compared with von Karman Street (VKS) of a single circular cylinder. Two distinct regions behind the elliptic cylinder were identifed. The frst region appeared with two expanded rows of vortices, where the vortices rolling up from each side of the cylinder's surface. The vortices were placed around the second region with fairly dead dynamic vortices in between when $AR = 0.5$ at $Re = 40$. It was also observed that the drag coefficient increases by increasing the Re for all aspect ratios except $AR = 1$ [\[11](#page-14-11)]. Since the viscous drag acts parallel to the bluff body, they showed that when Re increases, the pressure drag also increases up to 25% depends on *AR*. The results revealed that for all *AR* of test cases, the viscous drag is smaller than that of a circular cylinder since it acts normal to the surface. In the meantime, it was found that the pressure drag is more signifcant due to its action, which is normal to the surface of the NEC. Even though impressive results were found in this study, the hydraulic diameter of the elliptical cylinder was approximated by the vertical diameter of the ellipse $(2 b_1)$, which was not precise enough.

On the other hand, the fow over a parallel elliptic cylinder (*PEC* or with 0° incidence) was investigated by Raman et al. [\[46\]](#page-15-14). The results showed that by increasing *AR*, the drag coefficient increases while it decreases with increasing Re. The vortex frequency values also decrease with increasing AR; however, these vortex shedding frequencies can increase from 0.165 to 0.23 when the Re number grows from 100 to 500. Though the Re was relatively higher than the selected range of Re by Johnson et al. [\[11\]](#page-14-11), a 2D numerical in-house code was used to investigate the fow features. As the 3D vortex structure can be generated for Re>200, using a 2D code might correlate with some errors [\[33](#page-15-15), [55\]](#page-15-16). In addition, the details of the evaluation of the length scale of the elliptical cylinder have been not provided in their study [[46](#page-15-14)]. The dynamic response of the elastically mounted elliptical cylinder was studied at diferent angles of attack by Leontini et al. [[56\]](#page-15-17). Though diferent fow regimes were observed by authors once the angle of attack changed, the hydraulic diameter of the elliptical cylinder was simply chosen as the length scale for evaluating Re [[56\]](#page-15-17).

The effect of *AR* on the flow past an elliptical cylinder was also examined by Faruquee et al. [[42](#page-15-18)] at zero incidences, where the major axis of the elliptic cylinder was parallel to the flow stream. A series of test cases were chosen and examined with $0.3 < AR < 1$ at Re <40. The numerical results showed that at a very low range of $Re(e.g., 5 < Re < 10)$, a pair of steady vortices generates in the wake of the cylinder when the critical aspect ratio was *AR*=0.34. For smaller *AR*, no signifcant vortices were shed in the wake of the cylinder. They also observed that when $AR < 0.6$, the viscous forces become more dominant, while for *AR*>0.6, the pressure forces become more infuential.

Later, Vijay et al. [\[57](#page-15-19)] numerically studied the effect of AR on flow-induced vibration of the elliptical cylinder with a constant mass ratio of 10 when Re=100. The authors showed

that a synchronization regime is directly correlated with aspect ratio, and found that lock in response occurs for small aspect ratios and the magnitude of oscillation of elliptical cylinder with $AR = 0.1$ is twice that of a circular cylinder Vijay et al. [\[57](#page-15-19)]. While 2 single vortices (2S mode) were observed in this study for all examined aspect ratios, the Reynolds number was set based on the major axis with the elliptical cylinder.

2D numerical simulations were performed by Nair et al. [\[58](#page-15-20)] to study the influence of a wall on the wake structure of circular and elliptical cylinders. Nair et al. [[58](#page-15-20)] examined different gaps between cylinder and wall, which was set between 0.1 and 1.0 with $Re = 40-1000$. In this study, the diameters of circular and elliptical cylinders were considered for evaluating Re. Thus, the results of circular and elliptical cylinders did not show signifcant changes due to the geometrical parameters, when an identical gap and diameter were used in simulations [\[58](#page-15-20)].

The literature review given above demonstrates that both the spacing between the centers of the tandem cylinders and the cross section area of the cylinders are two critical parameters that remarkably afect the Karman vortex structure and imposed forces on the cylinders. To the authors' knowledge, there has been no prior study on the fow over tandem elliptical cylinders to examine the efect of the aspect ratio of the upstream cylinder on the wake structure. Besides, in the majority of the previous studies, the Reynolds number length scale was only identifiable with $D_1 = 2b_1$ (see Fig. [1\)](#page-2-0). The previously published studies neither considered the wet perimeter of the elliptical cylinder nor provided the details of explanation for evaluating accurate hydraulic diameter.

According to the above gap in the literature, in this study, we attempt to evaluate and use the precise hydraulic diameter of the elliptical cylinder using the Ramanujan approximation theorem, which has not been used before in literature. Based on this theorem and to better understand the fow instabilities in the wake and around the cylinders, we numerically investigate the vortex shedding frequency, forces (lift and drag) coeffcients of the cylinders once the aspect ratio of the upstream cylinder alters from *VEC* (with 90° incidence) to *PEC* (with 0° incidence).

2 Mathematical equations and numerical model

The Navier–Stokes (*NS*) equations in two dimensions are used for incompressible laminar viscous flow. The conservation of mass can be given as:

$$
\frac{\partial u_i}{\partial x_j} = 0.
$$
 (1)

The conservation of the momentum can be written as:

$$
\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} = -\frac{1}{\rho} \frac{P}{x_i} + \frac{\partial}{x_j} \left(\vartheta \frac{\partial u_i}{\partial x_j} \right).
$$
 (2)

In the above equation, $u(x, y, t)$ and $P(x, y, t)$ are the velocity and pressure felds of the fuid in "*i"* and "*j"* directions. Considering the Eulerian solution of the *NS* equations, the fnite volume method (FVM) was used in ANSYS Fluent software to simulate the test cases. The governing equations of fuid fow, thus, were integrated and solved iteratively based on the conservation of mass and momentum on each control volume. Since FVM can be used for the structured and unstructured meshes, the structured mesh was created for all simulations. The numerical discretization was based on the second-order upwind diferencing scheme to increase the accuracy of the solution. The computational domain around the cylinders is shown in Fig. [1.](#page-2-0) Water properties as a Newtonian fuid were chosen and set in the ANSYS Fluent database with a density of ρ =998.2 (kg/m3) and the dynamic viscosity of μ =0.001003 (Pa. s). In the simulations, the fow on the surfaces of the cylinders was set with zero velocity and no-slip.

The aspect ratio of the upstream cylinder varies from $AR = 0.25$ to 2.0. Therefore, the maximum and minimum gap ratio of $G^* = G/D_2 = 3.375$ and 2.5 were produced, respectively, and accordingly, eight confgurations were modeled and studied as shown in Fig. [2](#page-4-0). The advantage of this selection is examining both NEC and PEC in a tandem arrangement of the cylinders. Figure [2](#page-4-0) highlights the variation of *AR* of the upstream cylinder in pink dash-lines. The thicker pink dash-lines highlight the minimum and maximum aspect ratios, whereas the blue dash-line shows the elliptic cylinder with $AR = 1.0$ or a simple circular cylinder. The hydraulic diameter of the upstream cylinder is considered for evaluating the Re. By using diferent *AR*, the hydraulic diameter of the upstream cylinder changes due to the alteration of area and perimeter. Consequently, it is important to note that *Re* should be evaluated and set at Re=200 separately for each model based on the new hydraulic diameter (d_h) of the upstream cylinder. For each test case, d_h is calculated using the following formula:

$$
d_{\rm h} = \frac{4A}{P_{\rm w}}.\tag{3}
$$

Here *A* represents the cross-sectional area of the elliptical upstream cylinder and P_w stands for the wetted perimeter of the cylinder. Though the area of the elliptical cylinder can be calculated accurately based on the major and minor axes of the ellipse, e.g., $A = \pi a_1 b_1$, the perimeter of the elliptical cylinder is challenging to be accurately calculated [[59](#page-15-21)]. There are many theorems and formulas to estimate the cross-sectional perimeter of the elliptic cylinder, such as the Ramanujan approximation theorem and Maclaurin

Fig. 2 Schematic of tandem elliptical cylinders showing the variation of the aspect ratio of the upstream cylinder in dash lines. The blue dash-line highlights the $AR=1$ (simple circular cylinder). The spacing ratio between the centers of the cylinders is *L*[∗], while *G*[∗] is the gap ratio between the cylinders; the *AR* of the downstream cylinder

was kept constant in all test cases. The thicker pink dash-lines of the upstream cylinder highlights the minimum and maximum *AR* corresponding to the normal and parallel elliptic cylinders to the fow, respectively

expansion [[59,](#page-15-21) [60](#page-15-22)]. Ramanujan approximation theorem with high accuracy is one of the acceptable theorem, which can be used with an excellent satisfactory error (*e*< 10−2%) to estimate the perimeter of the elliptic cylinder. The Ramanujan approximation is given by:

$$
P_{\rm w} = \pi \left[(a_1 + b_1) + \frac{3(a_1 - b_1)^2}{10(a_1 + b_1) + \sqrt{a_1^2 + 14a_1b_1 + b_1^2}} + e \right]
$$
\n(4)

Table [1](#page-4-1) summarizes the geometrical parameters of the upstream elliptical cylinder, comprising the parallel and normal radiuses, and the area and perimeter and the hydraulic diameter of the upstream cylinder. In 2D modeling, the length of the cylinder is set to one in all simulations. It is observed that by changing the aspect ratio, both the cross section area and wetted perimeter of the upstream cylinder change. These two aspects directly affect the hydraulic diameter of the cylinder. Hence, the last column of the table shows the hydraulic diameter of the cylinder, which is used for evaluating Re.

Table 1 The geometrical parameters of the upstream elliptic cylinder to evaluate the hydraulic diameter of the elliptical cylinder

Table 2 Estimation of the error by assuming that the hydraulic diameter of the elliptic cylinder is identical with the diameter of a circular cylinder

It is worth estimating the percentage of the discrepancy (or error) for the hydraulic diameter if the Reynolds number length scale was identical with a diameter of a circular cylinder. Considering the hydraulic diameters of a circle and an ellipse, (e.g., $D_{h,circle}$ and $d_{h,ellipse}$, respectively), the accurate amount of Re for the circular cylinder (Re_c) can be estimated according to the following equation:

$$
Re_c = Re_e \times \frac{D_{h,circle}}{d_{h,ellipse}}
$$
 (5)

Table [2](#page-4-2) comprises two Reynolds numbers that can be calculated based on the precise hydraulic diameter of the elliptic cylinder using the Ramanujan formula or an estimated diameter of a simple circular cylinder. The last column of the table displays the percentage of the error between the accurate value and estimation one. The results show that nonlinear error is obtained for calculating Re resulting in a larger error for the NEC as compared with the PEC. To provide a clearer prospect, the amount of error versus *AR* is shown in Fig. [3,](#page-5-0) which is associated with Table [2.](#page-4-2) It is found that even for a relatively low Re, the maximum error for evaluating Re, approximately reaches 178% and 25% for NEC and PEC, respectively. The amount of error is calculated based on the following equation:

$$
Error = \frac{(D_{h,circle} - d_{h,ellipse})}{d_{h,ellipse}} \times 100\%
$$
 (6)

Therefore, to avoid these errors and study the flow structure of elliptical cylinders with high accuracy and reliability,

Fig. 3 Variation of the percentage of error as a function of *AR* at Re=200 when the hydraulic diameter of the elliptic cylinder is evaluated by the diameter of a circular cylinder. The data are associated with Table [2](#page-4-2)

Fig. 4 Effects of blockage on the mean-drag coefficient, the fluctuating lift coefficient and the Strouhal number of the upstream cylinder arranged in a tandem elliptical cylinders with *AR*=1 (circular cylinder) at $Re = 200$

here the Ramanujan approximation is used to evaluate the perimeter of the elliptical cylinder, and consequently, hydraulic diameter of the cylinder.

In numerical simulations, to avoid the blockage effects, four computational widths were selected and set at $H/D_1 = 12$, 15, 18, and 24 for a typical tandem circular cylinders (when $AR = 1$ for both cylinders) at $Re = 200$. Figure [4](#page-5-1) shows the effect of blockage on the mean drag coefficient as well as fluctuating lift coefficient and Strouhal number of the upstream cylinder. Though it is seen that there is about 2% discrepancy between $H/D_1 = 15$ and 18, the results reveal that by increasing the width to 18, the trends of \overline{C}_{D} , C_{lf} , and St remain constant. Thus, the computational width with $H/D = 18$ is chosen for the rest of the simulations.

The upstream and downstream lengths of $10D_1$ and $30D_1$ are also selected for the total length of the domain. These distances are long enough to guarantee that the inlet and outlet effects are eliminated $[61–64]$ $[61–64]$. Here, D_1 is the vertical length of the upstream cylinder and is kept constant and equal to $D_1 = 2b_1 = 2b_2 = D_2$ in all test cases. Consequently, the *AR* was varied only by altering the horizontal axis of the cylinder (e.g., a_1).

3 Results and discussion

In this section, the numerical simulation is initially validated. Then, the flow structure in the wake of the cylinders and fluidic parameters such as pressure coefficients, and the lift and drag coefficients are analyzed in detail. Finally, the

Table 3 Mesh resolution independence results at Re=200, including the mean-time drag coefficients and the fluctuating lift coefficients of the cylinders and the Strouhal number of the upstream cylinder at *AR*=1

Fluidic parameters	Number of grids				
	$MR_1 = 34,822$	$MR_2 = 46,256$	$MR_3 = 61,366$		
	1.189	1.440	1.460		
$\frac{\overline{C}_{\mathrm{D}(1)}}{\overline{C}_{\mathrm{D}(2)}}$	0.31	0.35	0.36		
$C_{\text{lf}(1)}$	0.544	0.553	0.553		
	1.105	1.118	1.119		
$\begin{array}{c} C_{\text{lf}(2)} \\ \text{St}_{(1)} \end{array}$	0.171	0.174	0.175		

efect of aspect ratio on the Strouhal number is studied, and a relationship between *AR* and St is derived.

3.1 Verifcation

To test the grid independence, three structured mesh resolution namely: MR_1 , MR_2 and MR_3 were examined for the flow over tandem circular cylinders. The results of each test are summarized and compared in Table [3,](#page-6-0) comprising the meantime drag coefficient of the cylinders $(C_{D(1)}$ and $C_{D(2)})$ the fluctuating lift coefficient of the cylinders $(C_{lf(1)}$, and $C_{Diff(2)}$ and St. A typically generated mesh in the fow domain and around tandem elliptical cylinders are shown in Fig. [5,](#page-6-1) when the aspect ratio of the upstream and downstream cylinders are set at *AR*=0.25 and 1.0, respectively.

It is worth noting that similar to the previously published data [\[22\]](#page-14-7), the Strouhal number was identical for both circular cylinders, and therefore, one value is only presented in Table [3](#page-6-0). The minimum mesh size in the finest mesh resolution (e.g., MR_3) was approximately 0.02. Based on this size, the time step of $\Delta t = 0.025$ (s) was chosen, which was adequately small to satisfy the Courant–Friedrichs–Lewy number (CFL= $U\Delta t/\Delta x$ <1). The results of Table [3](#page-6-0) highlight that $MR₂$ is an acceptable mesh resolution as compared with MR₃. While the maximum error of $e_{C_{\text{D}(1)}} \approx 4\%$ was obtained for the drag coefficient of the upstream cylinder, other parameters show less than 2% errors. Therefore, MR₂ was selected for the rest of the numerical simulations.

Further validations were also conducted for flow over tandem circular cylinders at Re=200. Table [4](#page-6-2) compares the obtained numerical results including $C_{D(1)}$, $C_{D(2)}$, $C_{lf(1)}$, $C_{\text{DIf}(2)}$, and St with those in literature. The subscripts '1' and '2' in Table [3](#page-6-0) refer to the upstream and downstream cylinders, respectively. It is seen that the present results are in good agreement with published data. While the percentage

Table 4 Comparison between the fuidic parameters comprising drag coefficients, fluctuating lift coefficients, and the Strouhal number of tandem circular cylinders with $AR = 1$ at $Re = 200$

Scholars	$C_{D(1)}$	$C_{D(2)}$	$C_{\text{lf}(1)}$	$C_{\text{lf}(2)}$	St
Present study	1.332	0.591	0.559	1.385	0.189
Alam $[22]$	1.255	0.36	0.551	1.117	0.175
Koda and Lien $[65]$	1.285	0.440	0.558	1.229	0.170
Mahir and Altac [37]	1.340	0.558	0.569	1.407	0.181
Dehkordi et al. [66]	1.160	0.520	0.575	1.059	0.179

of deviation between the results is acceptable, the best match is seen between the present study and the data ofered by Mahir and Altac [[37](#page-15-26)] for all parameters with less than 5%.

3.2 Separation points and reattachment locations

Highlighting the details of fow separation can gain more insight into the fow structure in the wake of the upstream elliptical cylinder. The separation points also directly can afect the reattachment of vortices on the downstream cylinder. Both of these locations can infuence the imposed pressure on the cylinders. Samples of instantaneous vorticity contours are shown in Fig. $6a-c$ at $Re = 200$ for three aspect ratios, minimum $(AR = 0.25)$, identical $(AR = 1.00)$, and maximum $(AR = 2.00)$. The figure shows the separation points, separation angles, and the locations of reattachments. As is well known, the boundary layer equation is a valid approximation for a wide range of Re until separation begins to intervene. In a laminar flow, such a feature corresponds to disappearing wall shear stress when the fow is broken away from the surface of the bluff body. Accordingly, the exact separation angle (e.g., θ_s) can be extracted from numerical results, where the wall shear stress vanishes on the surface of the upstream elliptical cylinder.

Figure [6](#page-7-0) highlights that by increasing *AR*, the separation points shift to the trailing surface of the upstream cylinder and the separation angle gradually increases. On the other hand, due to the increment of separation angle at larger *AR*, the spacing ratio between cylinders reduces; thus, the reattachment of the vortices on the downstream cylinder also takes place at larger angles as compared to the smaller *AR*.

The position of separation points can be understood by choosing the peak of the distribution of the fuctuating pressure coefficient around the cylinder $[67, 68]$ $[67, 68]$ $[67, 68]$ $[67, 68]$. Accordingly, the RMS fluctuation pressure coefficients of the upstream and downstream cylinders are plotted in Fig. [7](#page-8-0) for all *AR*.

Figure [7](#page-8-0) shows that the peak of $C_{\text{Pf(1)}}$ occurs at about 90 \degree for $AR = 0.25$. The sinusoidal trend, then, is seen after a peak of $C_{Pf(1)}$. By increasing AR, the magnitude of $C_{\text{Pf(1)}}$ gradually decreases. It is also observed that the peaks shift to larger angles (larger than 90°) and relocates on the trailing surface of the upstream cylinder, which again confrms the results of vorticity contours in Fig. [6.](#page-7-0) It is observed that the sinusoidal trend of $C_{\text{Pf(1)}}$, gradually alters to one oscillation for 0.25 < *AR* < 1.25 and for *AR* \geq 1.25, the curves change to a relatively flat trend. On the other hand and in general, the maximum fuctuating pressure of the downstream cylinder is $C_{\text{Pf(2)}} < 1$ for all *AR*. The peaks of $C_{\text{Pf(2)}}$ of $AR = 0.25$ and 0.5 occur at 68° and 60°, respectively, which are larger than that one with higher aspect ratios except *AR*=2.0. By increasing *AR* to 0.75, the peak of $C_{\text{Pf(2)}}$ approximately 30% increases as compared to that of $AR = 0.5$; however, the angle of reattachment at $AR = 0.75$ shifts to a smaller angle of 46°. For

Fig. 6 Separation and reattachment points of the upstream and downstream cylinders for test cases with **a** minimum aspect ratio, *AR*=0.25, **b** identical aspect ratio, $AR = 1.00$ and **c** maximum aspect ratio, $AR = 2.00$ at $Re = 200$

Fig. 7 Root-mean-square pressure fuctuations of **a** upstream and **b** downstream cylinders for tested cases of *AR*=0.25 to 2.00 at Re=200

 $AR = 1.0$ till 1.75, the peaks of $C_{\text{Pf(2)}}$ are found to be about 43°, 40°, 39.5°, and 39°, respectively. For test cases with $AR > 1$, it is observed that the magnitude of $C_{\text{Pf(2)}}$ gradually decreases. This means the infuence of vortices becomes smaller on the downstream cylinder as compared to those with $AR < 1$. The pressure fluctuations become very low or minimum around the upstream and downstream cylinders when $AR = 2.0$. Therefore, it is expected the imposed forces on both cylinders with $AR = 2.0$ become minimum.

To gain a better understanding and insight about the pressure distribution of tandem cylinders, frst, the timeaveraged pressure distribution (\overline{C}_{P}) , and RMS fluctuating pressure distribution (C'_{P}) along with the single circular cylinder at $Re = 200$ are both shown in Fig. [8](#page-8-1)a, b. Figure [8](#page-8-1)c also depicts the time-averaged contour plots of the pressure coefficient of the cylinder. It is seen that the maximum pressure on the single cylinder occurs at around 82°, which is consistent with the literature [[68\]](#page-16-0).

Fig. 8 Pressure coefficient distribution over the surface of a single cylinder at $Re = 200$, **a** time-averaged pressure coefficient distribution (\overline{C}_P), **b** RMS fluctuating pressure coefficient $(C'_{\rm p})$, ${\bf c}$ time-averaged contour plot of pressure coefficient

Fig. 9 Time-averaged pressure coefficient distribution along the surface (left and middle columns) and in the wake of the tandem elliptical cylinders (right column) with spacing $L^*=4$ at Re=200 as a function of aspect ratio of the upstream cylinder, **a** *AR*=0.25, **b** $AR = 0.50$, **c** $AR = 0.75$, **d** $AR = 1.00$, **e** $AR = 1.25$, **f** *AR*=1.50, **g** *AR*=1.75, **h** *AR*=2.00

In Fig. [9,](#page-9-0) the distribution of the time-averaged pressure coefficients (C_P) along the surfaces of the upstream (left column) and downstream (middle column) cylinders are presented, respectively. The right column of Fig. [9](#page-9-0) also shows the time-averaged contour plots of the pressure coefficient in the wake area of the cylinders. On comparing Figs. [8](#page-8-1) and [9](#page-9-0), it is observed that except for two test cases, $AR = 0.25$ and 0.5, the trend of $C_{P(1)}$ for the rest of the models is very similar to that around a single circular cylinder. It is seen that the pressure distribution on the upstream cylinder includes positive and negative values and alters as a function of *AR*; however, the downstream cylinder is always in a negative region apart from the aspect ratio of the cylinders.

The positive $C_{P(1)}$ on the upstream elliptical cylinder occurs at $\theta \le 79^\circ$ for $AR = 0.25$. By increasing AR, the angle of positive values of $C_{P(1)}$ on the leading edge of the elliptical cylinder gradually decreases, and therefore, $\overline{C}_{P(1)}=0$ steadily shifts to smaller angles as shown in Fig. [9](#page-9-0). Comparison of $C_{P(1)}$ reveals that the maximum negative magnitude of pressure coefficients of the upstream cylinder gradually decreases by increasing *AR*. This decrement is very sensible when *AR* alters from 0.25 to 0.50, in which $C_{P(1)}$ shows 25% less pressure (e.g., $C_{P(1)}$ changes from -2.4 at $\theta = 180^\circ$) to − 1.8 at θ = 86°). Since, $C_{P(1)}$ at the rear surface of the upstream cylinder for $AR = 0.25$ and 0.50 is much larger as compared with other test cases, it is expected to generate a reduced drag force on the upstream cylinder for these two test cases. This concept will be explained in the next section.

On the other hand, the results of \overline{C}_{P2} show that the timeaveraged pressure distribution along the downstream cylinder is not similar to that around a single nor upstream one. The values of $C_{P(2)}$, in general, are negative and smaller than that of the upstream cylinder (Fig. [9](#page-9-0)) or the single cylinder (Fig. [8a](#page-8-1)). It is found that, for *AR*>0.75, in which the gap

between cylinders reduces, the maximum negative magnitude of $C_{P(2)}$ occurs after $\theta = 90^\circ$ as it was expected earlier due to the reattachment. The contour plots in the wake and around the cylinders confrm that at *AR*=0.25 and 0.5, the pressure distribution corresponds to larger negative values of $C_{\rm P}$ on the lower and upper sides of both cylinders and in the wake of the upstream cylinder.

The contour plots of Fig. [9](#page-9-0) also show that the minimum time-averaged pressure coefficient over the surface of the cylinder occurs at smallest $AR = 0.25$ with the magnitude of C_p =3.7. Further increasing in AR =0.5, the magnitude of $\overline{C_{\rm P}}$ shows a sudden drop to 1.92 on the cylinder surface and C_P regularly decreases to the minimum magnitude of $C_{\rm P}$ =0.82 for the largest *AR* = 2.00. To better understand the discrepancy of the \overline{C}_{P} in the wake of tandem cylinders, the variation of $C_{\text{P,min}}$ versus to the *AR* is plotted in Fig. [10.](#page-10-0)

3.3 Lift and drag coefficients

Figure 11 shows the time history of the lift coefficients of the upstream elliptical cylinder and downstream circular cylinder at different *AR* (left column). The figure also represents the Fast Fourier transform (FFT) function corresponding to the peaks of the lift coefficient of the upstream cylinder over the total convergence time (right column).

The time history of the lift coefficients can be considered in terms of the amplitude of the forces and phase lags between the forces. As expected in the previous section, the magnitude of the lift coefficient shows the maximum value when $AR = 0.25$, and it reaches a minimum value at $AR = 2.0$. The results also reveal that the phase lag between the sinusoidal lift coefficients of the cylinders varies by changing *AR* and it gradually reduces. The phase lag reaches

Fig. 10 a Different angles of upstream elliptical cylinder where the magnitude of $\overline{C}_{\rm p}$ becomes maximum and zero, **b** variation of $\overline{C}_{\rm p,min}$ as a function of *AR* for the upstream cylinder extracted from contour plots of Fig. [9](#page-9-0) (a_3 – h_3)

Fig. 11 Time history of the lift coefficients of the upstream elliptic cylinder and down stream circular cylinder with different AR (left column) and the magnitude of FFT of the lift coefficient showing the Strouhal number (right column)

a minimum at $AR = 1.5$ and it increases again by an increment of *AR* to 2.0. When comparing the FFT figures for each *AR* it becomes readily apparent that the frequency of vortices also varies. The minimum *St* occurs at *AR*=0.25, and it is regularly increased to the maximum value at *AR*=1.75. The results show that *St* becomes close to the vortex frequency formation of a single cylinder when $1.25 \le AR \le$ 1.75 (e.g., St \approx 0.2). At this range of AR (e.g., $1.25 \le AR \le$ 1.75) the phase lags between lift forces are 14.08°, 9.98° and 27.9°, respectively. When *AR* increases to 2.0, vortex shedding frequency with 15% drops reaches to St = 0.17. The reason is due to the minimum gap ratio between the cylinders, which generates the overshoot regime, and hence, the vortices cannot be produced in the wake of the upstream cylinder and accelerated to increase *St*. It is emphasized that the overshot regime generates once the G^* < 1.2, where the separated shear layers of the upstream cylinder overshoot the downstream one and rolling in the wake of the downstream

Table 5 Lift and drag coefficients of tandem elliptic cylinders at *Re*=200

Test case	AR	$C_{\text{lf}(1)}$	$C_{\text{lf}(2)}$	$C_{\mathrm{D}(1)}$	$C_{D(2)}$
1	0.25	1.4367	2.7623	6.8831	-1.3455
2	0.50	0.9373	1.1538	2.7991	-0.3003
3	0.75	0.6202	1.0855	1.6592	0.1288
$\overline{4}$	1.00	0.5598	1.3857	1.3322	0.5915
5	1.25	0.3898	1.0076	1.0487	0.7668
6	1.50	0.1874	0.5413	0.8508	0.3478
7	1.75	0.1403	0.4123	0.7277	0.2655
8	2.00	0.0077	0.0929	0.6555	0.0840

cylinder without reattaching [[31\]](#page-15-1). The FFT plots of the lift coefficients also demonstrate that the magnitude of the lift forces is diferent and it varies by *AR*. Considering the magnitude of the FFT plots, it is confrmed that the maximum and minimum magnitude of the lift forces occur at *AR*=0.25 and 2.0, respectively.

The exact values of fluctuating lift coefficients (C_{lf}) and the time-averaged drag coefficients (C_D) are shown in Table [5](#page-12-0) for each test case. To better understand the trend of variation of the imposed forces on the cylinder, the fuc-tuating lift, and drag coefficients are plotted in Fig. [12.](#page-12-1) The results at $AR = 0.75$ and 1 are well matched with those measured and reported by Alam et al. [\[16\]](#page-14-6). In general, the results of the lift coefficients reveal that the upstream cylinder undergoes a smaller C_{lf} as compared with the downstream cylinder. The fluctuating lift coefficient of the cylinders follows similar trends as the spacing is increased. The fluctuations are remarkably large at $AR = 0.25$. The fluctuations of elliptical upstream cylinder remain extremely high, when $G^* = 3.375$ (e.g., $AR = 0.25$). As this gap ratio, $C_{\text{lf(1)}}$ and $C_{lf(2)}$ are 2.15 and 1.11 ($C_{lf(1)}$ 30 times higher than that of minimum *G** (test case 8), respectively (see Table [5](#page-12-0)). Although the fluctuations of the lift coefficient of both cylinders become low as the gap ratio decreases, the trend of the reduction is not similar. The variation of the $C_{lf(1)}$ is sinusoidal with a signifcant reduction, in particular, when *AR* increases to 0.5. *AR*=1.0 can be nominated as a critical aspect ratio (AR_C) , in which the fluctuating lift and drag coefficients of the downstream cylinder shows a peak. The results confirm that both C_{lf} downstream and C_{Df} upstream cylinders are very sensitive when *AR*<1.

Fig. 12 Variation of the fluctuating of lift (rms) and time-averaged drag coefficient as a function of the aspect ratio for the upstream and downstream cylinders at Re=200

Fig. 13 Variation of the mean-drag coefficient as a function of the aspect ratio for the upstream and downstream cylinders at Re=200

The results of the time-averaged drag coefficients of the cylinders are shown in Fig. [13.](#page-13-0) In general, the results reveal that the upstream elliptical cylinder undergoes a larger \overline{C}_{D} as compared with the downstream cylinder. It is found that the time-averaged drag coefficients of the upstream cylinder show a signifcant drop when *AR* increases. Besides, the amplitude of the fluctuating drag coefficient is larger for the upstream cylinder as compared with that of the downstream cylinder. For the downstream cylinder, it is seen that the time-averaged drag is negative at $AR = 0.25$ and 0.5 meaning that the downstream cylinder is pushed forwards by the fuid. The drag, then, gently increases to positive values by further increasing AR. Indeed, \overline{C}_{D} of the downstream cylinder neither approaches that of the upstream cylinder nor a single circular cylinder, even for maximum *G**.

Interestingly, the presence of the downstream cylinder causes a significant reduction of C_D of the upstream cylinder. The reason for this reduction is the increment of the pressure in the separated wake behind the upstream cylinder. This increment of the pressure occurs when the vortices of the upstream and downstream cylinders are in in-phase mode. This concept has been proved earlier for two and three tandem circular cylinders at $Re = 200$ $[22, 64, 69-72]$ $[22, 64, 69-72]$ $[22, 64, 69-72]$ $[22, 64, 69-72]$ $[22, 64, 69-72]$ $[22, 64, 69-72]$.

3.4 Relationship between St and *AR*

The variation of St as a function of *AR* is shown in Fig. [14.](#page-13-1) It is seen that *St* signifcantly increases by growing *AR* as expected. The variation of St starts from $St = 0.065$ at $AR = 0.25$ and continues to the maximum value of $St = 0.211$ at $AR = 1.75$. This trend can be formulated by showing the

Fig. 14 Variation of the Strouhal number as a function of aspect ratio, including a ftted curve for the obtained numerical results

relationship between St and *AR* to facilitate the estimation of St at other *AR*, which is arranged between the test cases studied in this paper. Therefore, a curve is ftted on the trend of Strouhal number versus the aspect ratio. As a result, an equation is driven and identifed as shown in Fig. [14.](#page-13-1) The equation can be written as follow:

$$
St = -0.11AR^2 + 0.3AR + 0.0053
$$
 (7)

Evaluating the residuals is very valuable to understand the accuracy of the ftted curve on the trend of St versus the *AR*. It is found that the residuals approximate positive and negative errors randomly with less than 1.5% diferentiation. Thus, Eq. [\(7\)](#page-13-2) fts the data very well and can be applied with a 1.8% error.

4 Conclusions

In the present paper, a 2D laminar fow over elliptical tandem cylinders was studied numerically. The previously published studies of fow over an elliptic cylinder mostly estimated the hydraulic diameter of the elliptic cylinder by a diameter of a circular cylinder. In the case of using the hydraulic diameter, in some studies, the details of the calculation are missing, and the details have not been provided. In this study, there was an attempt to evaluate the correct hydraulic diameter due to its importance on all the fuidic parameters, using the Ramanujan theorem.

The percentage of the errors was highlighted considering the precise hydraulic diameter of the elliptical cylinder. Once using NEC the maximum error of evaluating Re

approximately reaches 180%. It was also found that this error reaches 25% for PEC.

The results show that the peak of pressure fuctuation of the upstream cylinder $(C_{Pf(1)})$ occurs at about 90 $^{\circ}$ for $AR = 0.25$. The sinusoidal trend, then, is generated after a peak. The peaks of pressure fluctuation of the downstream cylinders with $AR = 0.25$ and 0.5 occur at 68 $^{\circ}$ and 60°, respectively. These peaks are larger than those with higher aspect ratios except *AR*=2.0. It was found that further increasing AR to 0.75, the peak of $C_{\text{Pf(2)}}$ approximately 30% rises as compared to pressure fuctuation of *AR*=0.5. The pressure fuctuations become very low around both cylinders for the test case with *AR*=2.0. Thus, the imposed forces on both cylinders with *AR*=2.0 become minimum.

Based on the imposed pressure on the cylinders, it was observed that by altering the aspect ratio, the phase lag between the sinusoidal lift coefficients of the cylinders varies and it steadily reduces. The phase lag reaches the minimum at $AR = 1.5$ and it slightly increases again by the increment of *AR* to 2.0.

The results reveal that the upstream elliptical cylinder undergoes a larger \overline{C}_{D} as compared with the downstream cylinder. The time-averaged drag coefficients of the upstream cylinder signifcantly drop by increasing *AR*. Besides, the amplitude of the fluctuating drag coefficient is larger for the upstream cylinder as compared with that of the downstream cylinder. For the downstream cylinder, it is seen that the time-averaged drag is negative at $AR = 0.25$ and 0.5. This means that the downstream cylinder is pushed forwards by the fuid. By further increasing *AR,* the drag slightly increases to positive values.

Based on the generated phase lag between forces and accordingly vortices, the Strouhal number varies. The minimum St=0.065 occur at $AR = 0.25$, and it is regularly increased to the maximum value of $St = 0.211$ at $AR = 1.75$. The *St* of tandem elliptic cylinders becomes close to the vortex frequency formation of a single cylinder when $1.25 \le$ $AR \leq 1.75$ (e.g., St ≈ 0.2). When *AR* increases to 2.0, vortex shedding frequency with 15% drops reaches St = 0.17 due to the minimum gap ratio between the cylinders. As a result, the relationship between the Strouhal number and *AR* was deduced and identifed with high accuracy and a reasonable error of 1.8%.

Declarations

Conflict of interest The authors declare that they have no confict of interest.

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