#### **TECHNICAL ARTICLE**



# **Metallography, Microstructure, and Wear Analysis of AA 6063/TiC Composites for Augmented Dry Sliding Property at Room Temperature**

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#### **Abstract**

This paper summarizes the experimental investigations for augmented dry sliding property  $\sim$  AA 6063/TiC composite. An in situ metal matrix composite has been prepared with AA 6063 alloy for the matrix and TiC for reinforcement particulates. The fabrication of the composites is done by using potassium titanium fluoride ( $\kappa$ T<sub>IF6</sub>) and powder of graphite as a mixture for reaction along with molten aluminum alloy at 950 °C. The weight for tion of  $\heartsuit$  particulates has been varied in the composites  $C_0$ ,  $C_1$ ,  $C_2$ , and  $C_3$  as 0, 3, 6, and 9%, respectively. Various characterization techniques and tests were conducted to observe the morphological properties of the composites. The composites were categorized into three sets. The first set of the composites is as-cast condition. The composites were heat-treated to form two more sets, namely solutionized and solutionized-aged condition sets. The "pin-on-disk" apparatus sused for conducting the tests, wear analysis, and comparative analysis of the wear for the dry sliding condition. The study indicated a significant enhancement of the resistance to wear by composites with the higher fractions of titanium carbide (TiC) particulates and further enhancement with the heat treatment. **Example 18:** S. S. S. N. Noorthy<sup>1</sup>. G. Naveen Kumar<sup>2</sup>. V. Srinkvar<sup>2</sup>. M. Ashock Kumar<sup>4</sup>.<br>
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**Keywords** AA 6063/TiC Composites · Wear · In itu MMC  $K_2$ TiF<sub>6</sub> · Characterization · Morphology

## **Introduction**

Aluminum though extensively  $\iota$ , in aerospace applications, owing to their less resistance to wear and lesser strength, limits their potentiality to substitute for the applications in engineering mate.  $\frac{1}{2}$  resources the composites with discontinuous einforce particulates were emerged to fit not only for the  $\lambda$  in aerospace applications with lower

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weight and more performance but also in the automobile felds with augmented mechanical properties and resistance to wear  $[1]$ .

The behavior to wear of particle-reinforced composites with metal matrix has been studied with various ceramic reinforcements as  $B_4C$  [2], TiB<sub>2</sub> [3, 4], Al<sub>2</sub>O<sub>3</sub> [5, 6], SiC  $[7, 8]$ , TiC  $[9]$ , and WC  $[10]$ . Among these reinforcements, titanium carbide (TiC) has been commonly used in composites of reinforcement with particulates in metal matrix due to its increased hardness, enhanced heat stability, improved electrical and thermal conductivities, good wettability, and resistance to wear. TiC usage in various metal matrix composites to enhance their properties such as aluminum and iron  $[11-15]$  is being reported. Various synthesizing techniques [\[16,](#page-10-12) [17](#page-10-13)] have been used in fabricating composite materials with TiC reinforcement, such as powder metallurgy route and in situ melt reaction route. However, in powder metallurgy route, the difficulty of contamination at the surface interfaces has been observed.

It was reported by Peijie et al. [[12](#page-10-14)] that if the composites are prepared with "in situ" technique, then the interatomic bonding between reinforcement and matrix tends to be stronger. Another major advantage of in situ generation is the low raw material cost and low processing costs. Aluminum and titanium carbide composites [[15](#page-10-11)] have been synthesized using in situ casting route with potassium titanium fluoride  $(K_2TiF_6)$  and powder of graphite. However, the synthesis of TiC in aluminum melt has been made by a spontaneous reaction of  $K_2TiF_6$  and graphite salts. Tyagi [\[9](#page-10-8)] has synthesized Al–TiC composite through reaction of SiC and granules of titanium in aluminum melt and studied the infuence of TiC reinforcement in aluminum composites on wear resistance. A comparison of wear among various discontinuous reinforced metal matrix composites has been made [18–23].

Various synthesis methods and applications in nanomaterials have been published in detail [24–26] which may also be considered for the synthesis of the materials in the composites. These applications of composites for various felds have been experimentally studied and published in detail [\[27–](#page-11-3)30]. Further, in recent past few years various composites are experimentally studied and researchers are publishing evident research such as efect of various processing parameters and corrosion behaviors [31–33], effect of ultrasonic and high temperatures on microstructures [34–36], effect of oxidation resistance and laser irradiation [37–43], the erosion and corrosion behavior of in situ composites of TiC with aluminum alloys other than AA 6063  $[44-\sqrt{9}]$ , and microstructural studies of TiC and AA  $6063$  separately [\[42,](#page-11-13) 47–49]. The total literature is restricted to  $t'$  e studies of either of the materials further if some studies are done with the mentioned combination of the composites but are at higher temperatures. Hence, this study if attempted to conclude exact slide wear behavior of the proposed composite. **PE[R](#page-11-16)IOR D[E](#page-11-12)SCR[I](#page-11-17)PED AND THE SETTENT SETTEMBER IS 1792 PROPORTION INTERFERING THE CARD TRIPACT AND THE SET CAN ART AND THE S** 

It is generally accepted that  $t \cdot \text{follown}$ , mechanisms contribute to the strengthening  $(f_A \cdot MCS: (1)$  grain refinement of the metal  $\mathbf{m}$ -trix, (2) load-bearing effect from the hard reinforcement base  $\sqrt{\alpha}$  an interfacial shear stress, (3) Orowan diversion is strengthening from the presence of nanometric vinforcement particles or other possible dispersoids in the matrix, and (4) dislocations originating from geometrical const aints, i.e., geometrical necessary dislocations  $C<sub>N</sub>DB$ , mismatch in coefficients of thermal expansion (CTE) between the matrix and the reinforcement phase (thermal contraction), and plastic deformation during proces  $\lceil \frac{50}{50} \rceil$ .

Many  $\overline{J}$  the problems encountered in solidification processing of particulate-reinforced aluminum composites such as more number of manufacturing steps and cost of processing are overcome by an alternative approach which relies upon the in situ formation of TiC particles in aluminum melts. Since TiC particles so generated are thermodynamically stable, the matrix–particle interfaces tend to be clean, free from adsorbed gases, oxide flms, and detrimental reaction products and are thus much stronger [[15](#page-10-11)].

### **Fabrication Techniques for Metal Matrix Composites (MMC) of Aluminum**

Any material prepared from either two or more diferent materials having considerably distinct chemical and/or physical properties is termed as composite material. Any composite material contains three components, viz. reinforcement, matrix, and interface between them. I general, reinforcement is a material which has higher strength and distributed in a certain predetermined orientation in continuous phase of another distinct material called matrix which holds the reinforcements [51].

Moreover, the interfaces between reinforcement and matrix play a vital role in getti. the enhanced properties of composite materials. The final properties of any composite always depend on the mechanical and chemical interactions between the reinforcing and matrix materials. Strong interfacial bonding  $\sigma$  is necessary for MMCs to achieve more load-bearing capacity and stiffness. This bonding nature is strongly and by various reasons such as wettability, mechanical, chemical, structural, and thermal properties of materials used in the manufacturing of composites. Further, the  $s$  faces of the reinforced materials in the composite and their  $\mu$  urity play a very important role in the final property  $e_n$  be accement of the composites [52].

Metal matrix composites can be manufactured broadly in three categories of methods. Various methods that are majorly used are mentioned below [51]:

- 1. MMC preparation using solid-state methods
	- a. Blending of powders with consolidation
	- b. Bonding with difusion
	- c. Physical vapor disposition (PVD)
- 2. MMC preparation using liquid-state methods
	- a. Casing using stirring method
	- b. Casting using squeezing method
	- c. Process of infltration
	- d. Deposition using spray method
- 3. MMC preparation using in situ methods

The selection of the method or technique for the preparation of the composites would be based on various factors including the materials used for the composite and several properties that are required to be enhanced for a particular application. Each technique has its own distinct advantages and disadvantages. Brief comparison is presented in Table [1](#page-2-0)

To achieve high mechanical, physical, and thermal properties, the aluminum composites are majorly formed in "discontinuously reinforced Al alloy composites (DRACs)" due to the good combination of ceramic and metallic nature of

	S. no. Technique	Suitability	Important features including advantages, disadvantages, and economical aspects
$\mathbf{1}$	Powder mixtures	Particulate reinforcements.	Both materials should be in powder form before blending Used to form small components Moderate stiffness and strength up to 300 °C Low density Moderate expensive
2	Diffusion bonding	Reinforcing materials are like filaments, and matrix materials are in the form of sheets and foils	Major problems of oxidation at high temperatur High expensive
3	<b>PVD</b>	Thin films, medical, or surgical dies	Most of the times harder More corrosion resistive when compared and electroplat- ing coatings High expensive
4	Stir casting	Particulate reinforcement of aluminum metal compos- ites	Process parameters $nd$ $\bf{h}$ 'erial properties affect the method Lesser therm co. vetivity and density Very poor surface pu. Less expendence
5	Squeeze casting	Mass production of any type of reinforcement	Lesser $r$ auctivity and density Very poor rface purity derately spensive
6		Infiltration process Reinforcement for usually filament type	$\frac{1}{2}$ oc $\frac{1}{2}$ and stiffness up to 200 °C L w thermal and dense properties Ve $\hat{y}$ poor surface purity Expensive
7	Spray deposition	Reinforcement of particulates	Low density and thermal expansion coefficient Very poor surface purity Moderately expensive
8	In situ process	Reactive processing method	Best reinforcement Best compatibility with the matrix Very good homogeneity of the distribution of the rein- forced particles Very good strength, ductility, toughness, and fatigue resistant
			Best purity of interfacial surfaces leading to strong bonds Economically favorable due to simple method which leads to less equipment
the aluminum alloy. The dividing the DRACs are fabricated using various meth. <sup>1</sup> s and tec <i>Aques</i> which are ex situ such temperatures leading to weak bonds. as powder blending it hniques, casting, spray deposition,			6. Chemical reaction at the interfaces when used at high 7. Cost of the DRACs using conventional techniques is
infiltration, and vortex. Jut these techniques have the fol- very high due to the equipment for the preparation of lowing sign. Sount disadvantages [52]: small-sized reinforced particles, surface treatment and			
skilled control, and monitoring of the stepwise pro- chi ving the optimal mechanical properties is very dif- cesses. fic 't.			
2.		As the reinforced particles are prepared separate, their	Hence, in situ process for the fabrication can be adopted

<span id="page-2-0"></span>**Table 1** Comparison of various techniques for composites' fabrication [\[51,](#page-11-17) [53\]](#page-11-19)

- 1.  $\Delta$  chi ving the optimal mechanical properties is very diffic 't.
- 2. As  $t_1$   $\epsilon$  reinforced particles are prepared separate, their surfaces are contaminated during the process of manufacturing.
- 3. The interfacial incompatibility due to the impure surfaces of the reinforcements.
- 4. Consequently weak interface bonding and failure during the service.
- 5. Thermodynamic instability of reinforced particles with the metal matrix.
- 6. Chemical reaction at the interfaces when used at high temperatures leading to weak bonds.
- 7. Cost of the DRACs using conventional techniques is very high due to the equipment for the preparation of small-sized reinforced particles, surface treatment and skilled control, and monitoring of the stepwise processes.

Hence, in situ process for the fabrication can be adopted for the following advantages [[52\]](#page-11-18)

- 1. Due to the economical favorability as the process is simple and requires very less cost equipment.
- 2. Fabrication through chemical reactions.
- 3. Enhanced properties.
- 4. Thermodynamically stable reinforcement in matrix.
- 5. Non-occurrence of harmful interfacial reactions.
- 6. Surface purity, strong bonding, and better compatibility of the reinforced particles with metal matrix.
- 7. Enhanced mechanical properties due to homogeneous distribution and fne size of reinforced particles.

Having the above-mentioned advantages in situ method of preparing or fabricating the composites is very much suited to prepare the proposed composite with AA6063 with TiC as the cost of the materials is obviously high, but the in situ process is simple and requires less equipment and thereby economical when compared with the cost incurred in the traditional processes due to their complexity and equipment needed [54].

In the present work, AA 6063 metal matrix composite has been synthesized with TiC as reinforcement through in situ casting route. The synthesis of TiC has been done through a spontaneous reaction between  $K_2TiF_6$  and graphite powders in aluminum alloy melt. The wear analysis of three variations, one as-cast set and other two heat-treated composites, namely solutionized and solutionized-aged condition sets, has been carried out using the "pin-on-disk" experimentation for resistance to wear at dry sliding condition.

## **Experimental Details**

## **Casting and Heat treatment**

The AA 6063 alloy that inherits good machinability, formability, weldabilty, and medium streng  $\hbar$  emerged  $\hbar$  as the best candidature to form the matrix of  $t \geq \text{in}$  situ composite comprising titanium carbide  $(T_iC)$  as  $\Box$  forcement.

<span id="page-3-0"></span>



The nominal composition of AA 6063 alloy is provided in Table [2.](#page-3-0) TiC particulates are formed using alkaline salt  $K_2$ Ti $F_6$  and graphite powder. The TiC particulate fractions have been changed by varying the mass and added in the aluminum alloy in molten state. The mass of chemicals to be added in the melt had been calculated as per the stoichiometric ratio to form composites  $C_0$  (Al60<sup>6</sup>3-0 wt.%) TiC),  $C_1$  (Al6063-3 wt.% TiC),  $C_2$  (Al6063-6 we TiC), and  $C_3$  (Al6063-9 wt.% TiC). The weight of chemicals added in different composites is shown in Table 3.

In order to prepare the above- $r_{\text{at}}$  composites, the chemicals, namely  $K_2TiF_6$  and graphite, were carefully weighed as calculated earlier and  $\ddot{\phantom{a}}$  wed thoroughly. Then, the weighed AA 6063 allow was loaded in a silica crucible to heat in the furnace as shown in Fig. 1.

Thus, the prepared  $mix$  re of the chemicals was added in the AA  $6063$  and  $\gamma$  melt maintained at the temperature of 950 °C for  $\mathcal{A}_p$  m inside the furnace for the in situ development of  $\Lambda$   $\tilde{\Lambda}$  particulates. Birol [15] reported the importance of temperature range from 900 to 1100  $\degree$ C in order to synthesize TiC particulates from  $K_2$ TiF<sub>6</sub> and graphite powders. A vigorous exothermic reaction was started due to the addition of chemicals into the melt of  $AA\ 6\ 63$ . The mixture was stirred in the crucible after  $t_{\text{interval}}$  of 10 min. The progress of TiC formation is as follows [15];



<span id="page-3-2"></span>TiC composites

<span id="page-3-1"></span>**Table 3** The weights as per the stoichiometric ratio for the preparation of composites



$$
4\text{Al} + 3\text{K}_2\text{TiF}_6 \xrightarrow{\text{yields}} \text{K}_3\text{AlF}_6 + 3\text{KAlF}_4 + 3\text{Ti}.\tag{1}
$$

$$
3\text{Al} + \text{Ti} \xrightarrow{\text{yields}} \text{Al}_3 \text{Ti}.\tag{2}
$$

$$
Ti + C \xrightarrow{yields} Ti C \tag{3}
$$

Finally, the molten state mixture is carefully taken into a preheated cast iron mold to the temperature of 300 °C.

After preparing the composites with this casting process, viz. AA  $6063-wTiC$  (where  $w =$  the wt.% of 0, 3, 6 & 9), the cast blocks were subjected to two diferent heat treatments. Thus, three sets of composites were prepared on the basis of their condition. The frst set of composites was not heattreated and remained in the condition of "as-cast set." The solutionized composites at 530 °C for 1 h quenched in water are taken as the second set. The same solutionized composite with an aging at 175 °C for 8 h is taken as the third set. The X-ray difraction analyses (XRD) for composites with varying TiC contents in as-cast condition were carried out.

TiC particles form in increasing numbers with increasing reaction temperatures when Ti is introduced into molten aluminum in the form of a halide salt, together with graphite. Al<sub>3</sub>Ti forms as soon as the K<sub>2</sub>TiF<sub>6</sub>, and graphite mixture is added to the melt.  $AI<sub>3</sub>Ti$  particles which precipitate out dissolve in the melt with increasing temperatures and with  $\iota$ . The reaction between the graphite particles and the solution Ti is promoted once the graphite particles are incorporated into the melt. The solute Ti reacts with  $c_{\text{at}}$  to produce TiC particles owing to the improved wet ing of the graphite particles. While the dissolution of the  $\frac{1}{2}$  minides is also of help in this regard, the improved incorporation of the graphite particles into the melt is largely linked with  $KAIF_4$ and  $K_3AIF_6$ . These fluxing  $\alpha_k$  nts generated by reaction (1) clean the particle surfaces and remove the oxide layer from the surface of the mean. The reaction between the graphite particles and the solute Ti is promoted once the former are incorporated into the metal  $[15]$ . **Fig. <sup>2</sup>**"Pin-on-disk" apparatus for conduction wear tests **RE[T](#page-10-11)RACTED ARTICLE**

## **Sliding Wear Test**

In the method of "pin-on-disk" apparatus, a disk was rotated and a setionary pin was kept in contact with disk using counterweight on pin causing sliding motion of the disk relative to the pin. In the present work, three varieties of pins with 10 mm and 25 mm diameter and length, respectively, were manufactured from the composite sets, namely as-cast set, solutionized set, and solutionized-aged set, and each set of composites with varying volume fractions of the TiC particulates and named as  $C_0$ ,  $C_1$ ,  $C_2$ , and  $C_3$ . One set was solutionized, and another set was aged after solutionizing to identify the aging efect on resistance to wear of the

<span id="page-4-1"></span>**Table 4** The parameters of fxed sliding wear

<span id="page-4-0"></span>



<span id="page-4-2"></span>composites at room temperature. High-carbon–high-chromium steel  $(D_2$  steel) rotating disk is used which has 82, 12 mm, Ra  $0.17 \mu$ m, and  $230 H<sub>V</sub>$  diameter, thickness, value of surface roughness and the hardness, respectively. The pin and disk surfaces were ground, and acetone was used for cleaning to achieve proper fat contact. The experimental investigations were carried out by considering the parameters of fxed sliding wear as indicated in Table 4.

The experiments were carried out at constant speed of sliding by fxing the angular speed and track diameter of the disk. To calculate the loss due to wear of the composites, the pins were weighed before and after the conduction of the experimental tests with digital balance 1 mg accuracy. The relative motion of the disk while rotation causes application of a frictional torque on the pin that is hold by a movable arm which was rested against the apparatus frame. A load cell mounted on fxed frame which is being touched the movable arm measures the frictional torque as shown in Fig. [2.](#page-4-2) With the help of continuous data of frictional torque generated during the experiment, the frictional coefficient was calculated. The loss of volume for each set of pins of the composites is calculated using the measured data related to loss of weight. The Archimedes principle is used to determine the corresponding densities for the conversion, and they are 2700 kg/m<sup>3</sup> for  $C_0$  AA 6063, 2767 kg/m<sup>3</sup> for  $C_1$ 

<span id="page-5-0"></span>**Fig. 3** Images for solutionized set of AA 6063-TiC using optical microscopy with varying weight %s (**a**)  $C_0$ , (**b**)  $C_1$ , (**c**)  $C_2$ , and (**d**)  $C_3$ 





in situ composite, 2834 kg/m<sup>3</sup> for  $C_2$  in situ composite, and 2900 kg/m<sup>3</sup> for  $C_3$  in situ composite.

## **Experimental Results**

## **Optical Microscopy Analysis**

The optical microscopy is majorly amed to ensure the TiC particulates as per the a ned weight and volume fractions. Figure 3 shows the micrographs of the prepared composites with varied weight fractions  $C_1$ ,  $C_2$ , and  $C_3$  of solutionized set.

The optical microscopic images emphasize the reinforcement of Tic particles and increase in the size and volume of the particles as the weight percentage increases.

## **Morphological Investigations Using Scanning Electron Microscopy**

As morphology investigations play an important role in identifying the reinforcement of the composites, investigations were carried out using "scanning electron microscope (SEM)." The samples were prepared using Ringer's solution as etchant. Figures [4,](#page-5-1) [5](#page-5-2), and [6](#page-6-0) indicate clearly the reinforcement of TiC particles but with irregular shapes in the composites  $C_1$ ,  $C_2$ , and  $C_3$ , respectively. Even though some islands of TiC clusters were observed in AA 6063



**Fig. 4** Composite with 3% TiC

<span id="page-5-1"></span>

<span id="page-5-2"></span>**Fig. 5** Composite with 6% TiC

alloy matrix, the dispersion of TiC seems to be uniform throughout the matrix. EDAX analysis (Fig. [7](#page-6-1)) has shown the presence of titanium and carbon of TiC particles. The diference in density between the TiC particle and the aluminum matrix plays critical part during solidifcation. The in situ method produces particles of very fne size. The fne size is a result of high nucleation rate of in situ particles. The fne size particles remain in suspension for a longer duration in the aluminum melt without settling at the bottom of the crucible. The sinking rate is insignifcant. As TiC particles are formed by the in situ reaction, the molten aluminum spreads on the surface of the TiC particle in order to wet it. This wetting action retards the free movement of TiC particle within the melt. Thus, a homogeneous distribution is achieved. The familiar defects such as porosity, shrinkages, or slag inclusion are absent in the micrographs in Figs. 4, 5, and 6 which demonstrates the quality and hence assumed to be homogeneous. Density of the samples was determined by rule of mixtures. The density of the composites enhances with the increase in reinforcement content in the composites. Three sets of 10 measurements each were carried out. On

average, the density of  $3.178 \pm 0.002$  g/cm<sup>3</sup> was achieved in the calculations.

Images were processed using Scion Image version Beta 4.0.2 (Scion Corporation, Frederick, MD), and particle sizes were measured. The mean grain size of the grains which are equiaxed is observed to be 90 µm. The distribution of size of TiC particles is shown in Figs.  $8/9$ , and 10. The increase in the size of the mean particle  $w_1$  increasing volume fraction of TiC particles in the composities is evident from Fig. 11 which is due to higher kinetics of the reaction nearer to TiC nuclei while roumation.

#### **XRD Analysis**

The cross section of  $t_1$  compositions weight fractions mentioned as  $C_0$ ,  $C_1$ ,  $C_2$ , and  $C_3$  is analyzed with the "X-ray diffraction" (XRD), and the corresponding patterns are presented in Fig. 12. All the peaks are identified as arising from the phases, namely, Al-based phase, TiC, and  $AI<sub>3</sub>T<sub>1</sub>$  Further, it was clearly evident that the increase in the weight fractions of TiC enhances the formations of TiC and  $Al<sub>3</sub>Ti$  as observed by the highest corresponding peaks.



<span id="page-6-2"></span>**Fig. 8** Reinforcement particle size distribution in AA 6063-3 wt.%

<span id="page-6-0"></span>

<span id="page-6-1"></span>**Fig. 7** (**a**) HRSEM image of the composite with 9% TiC. (**b**) Corresponding EDAX result



<span id="page-7-0"></span>**Fig. 9** Reinforcement particle size distribution in AA 6063-6 wt.% TiC in situ composite



<span id="page-7-1"></span>**Fig. 10** Reinforcement particle size distribution in  $A \setminus 60$ , 9 wt. TiC in situ composite



<span id="page-7-2"></span>**Fig. 11** M an particle size of TiC in AA 6063-TiC in situ composite

From the XRD peak profle analysis, the area under the peak was measured and the relative fractions of reinforced TiC particulates were determined for composites  $C_1$ ,  $C_2$ , and *C*3. The corresponding volume fractions were calculated and found to be 2.85, 4.20, and 7.25%. These reduced fractions are attributed to the melting losses, as against the targeted values.



<span id="page-7-3"></span>

### **Microhardness Analysis**

"Vickers hardness tester (Zwick)" is used to fnd the microhardness of the composites under a load of 0.5 kg and is tabulated in Table 5.

The following observations are made from Table 5.

- 1. Higher the wt.% of TiC, higher the hardness value. This is due to the improvement in the hard phase volume fraction.
- 2. The solutionized set of composites exhibited a better hardness value to that of as-cast set. The reason may be associated with homogenization of the entire microstructure during solutionizing heat treatment.
- 3. The solutionized-aged set of composites exhibited an improved and higher hardness to the other two sets. This can be attributed to the formation of fine  $Mg_2Si$  precipitates from AA 6063 matrix.

#### **Wear Analysis**

The decrease in the volume because of the wear at a fxed load per unit distance of sliding is the rate of the wear. Calculations from the data of volume loss are used for fnding the rate of the wear and plotted against varying the wt.% of reinforcement as presented in Fig. [13.](#page-8-1)

The plot envisages that the increase of wt.% of the TiC reinforcement in AA 6063 alloy matrix decreases the rate of wear for the composite. The wear rate decreased sharply during the initial stage for the case of as-cast condition <span id="page-8-0"></span>**Table 5** Microhardness values  $(Hv_{0.5})$  of the three sets of composites with all the varying wt.% of TiC





<span id="page-8-1"></span>Fig. 13 Wear rate variation with the wt.% of TiC particle reinfor ment

compared with that of solutionized condition.  $F_{\text{L}}$  then, the decrease in the rate of wear slowed down in the case of solutionized and aged condition compared with other two conditions, namely as-cast and solutic ized. The initial increase in wear rate in 0% TiC composite *particularly* during the as-cast condition is due to inhomogeneity in ascast structure exhibited by the alloy of AA 6063. This is due to the oxide layer formation  $\alpha$  ring the test which is in line with the observation of  $\{qq_1\}_{1}$ . From here, the linear decrease in the rate of wear due to the increase in wt.% of TiC can be observed and this decrease in wear rate is not only because of the increase in fraction of TiC particles but also the high hardness of TiC particles and better bonding at the surface interfaces formed during in situ synthesis. Also, the ear properties were found to improve with aging after solutionizing heat treatment as shown in Fig.  $13$ . In the case of solutic lized and aged condition, the lowest wear rate was observed due to precipitation hardening. **Fig. 15**  $\frac{9}{25}$ <br> **15 15**<br> **Example 2 Example 2 Examp** 

Figure [14](#page-8-2) shows an increase in wear resistance due to increasing reinforcement wt.% of the composites. Especially, sharp increase in resistance to wear of the composite  $C_3$  in solutionized and aged condition was observed due to the formation of precipitation hardening and the reinforcement of TiC particles to AA 6063 alloy matrix. Thus, the composite  $C_3$  exhibited more load-bearing capacity and the frictional load was distributed to a greater extent on TiC particles.

<span id="page-8-2"></span>**Fig. 14** Variation of wear resistance with respect to the wt.% of the TiC particle reinforcement



<span id="page-8-3"></span>the TiC particle reinforcement

This distributed load could have resulted in less wear of the composite.

The fraction of applied load to frictional load is the coefficient of friction. The applied load was made constant for all the experiments as 9.8 N. Figure [15](#page-8-3) shows the plot between coefficient of friction variations with





**Fig. 16** SEM image showing the worn-out surface of AA 6063 (0%



<span id="page-9-1"></span>**Fig. 17** Worn-out surface of AA  $6.3-3$  wteraction composite of solutionized condition



<span id="page-9-2"></span>**Fig. 18** Worn-out surface of AA 6063-6 wt.% TiC composite of solutionized condition



<span id="page-9-3"></span>**Fig. 19** Worn-out surface of AA 6063 alloy 9% TiC composite of solutionized condition

track or direction of sliding. The grooves formed in unreinforced alloy seemed to be deeper and more in number as compared to the composites shown in Fig. [16](#page-9-0). Figure [17](#page-9-1) shows the groove formed in AA 6063–TiC with 3 wt.% reinforcement at high magnification level. The plastic

<span id="page-9-0"></span>corresponding weight percentages of TiC particles in the composite. It infers that the coefficient of friction is decreased with the increase in the wt.% of TiC reinforcement. It implies that a reduction in frictional force was observed with an increasing percentage of TiC reinforcement particles because of more amount of hard particles. This may be attributed to the presence of carbon due to graphite which adds to the lubrication phenomena and reduces the frictional force. Further, it is evident that  $f_{AC}$ tional force gets distributed because of the increased  $TiC$ reinforcement which causes less wear. Also, the solution ized and aged heat treatment increased the wear resistance compared with as-cast and solutionized composite. This can be due to the fact that the aging heat treatment leads to an increased hardness value due to the formation of intermetallic formed precipitates. Another ason may be due to the formation of strong and  $c_n$  interfacial bond between the AA 6063 alloy matrix and TiC particles. 13<br> **RETRACTION RETRACTED RETRACTED**

From Fig.  $16$ , which shows the worn-out surface of composite  $C_0$  with solution it was observed that there were  $gr_{\alpha}$  ves and  $\alpha$  laminated layers formed by the sliding wear on the surface. The prime mechanism of the wear producing such surface should be due to abrasive in nature.  $\mathbf{h}$ , this, the surface projections or asperities of two surfaces came in contact due to sliding. The sliding ctich and the high local pressure produced the heat gener<sub>a</sub> on and caused the weldment and the interlocking among the asperities. Further motion produced a plastic deformation and broke these bondings. These broken particles became the third body and caused three-body abrasion. This wear from the third body produced grooves and ridges in the unreinforced aluminum alloy deformed plastically without much resistance. Figures [17,](#page-9-1) [18](#page-9-2), and [19](#page-9-3) show the worn-out surface of  $C_1$ ,  $C_2$ , and  $C_3$  composites.

The investigations of the composites' worn-out surfaces at higher magnification level revealed the difference among the ridges and grooves formed inline with the wear deformation causing generation of cracks is quite clear on the edge of the groove generated. Also, the micropits can be seen. The SEM images in Figs. [16](#page-9-0), [17](#page-9-1) and [18](#page-9-2) show smooth wear of the unreinforced alloy, whereas in composites  $C_1$ ,  $C_2$ , and  $C_3$  there is more wear debris of transfer layer and pits and crater formed by severe delamination of the matrix metal. Also, the fracture of particles and transverse cracks is visible in Fig. 18 of composite  $C_3$ .

## **Conclusions**

The present morphological investigations and wear analysis lead to the conclusions presented below

- A very easy and simple melt reaction synthesis which is cost-efective can be used to generate TiC particulates  $(K_2TiF_6/C)$  as reinforcement in an AA 6063 alloy matrix.
- The optical micrographs, SEM, and microhardness testing results indicated the presence of TiC particles which have been proved by XRD and EDAX analyses and hence the feasibility of the composite.
- Sliding wear test results clearly showed the reduction in the rate of wear due to an increased reinforcement  $\tau$  articulate weight fraction in the composites. Also, the wear rate is the least for samples of solutionized-aged set  $\alpha$ to the formation of high hardness and  $prec'_{\rho}$  tion and the highest for as-cast condition. and the rest in the time term into the second method in the second method
- The impact of the reinforcement on resistance to wear of composite is highest for 9 wt.% of Ti $\overline{C}$  reinfor ement for solutionized and aged condition.
- Linear decrease in "coefficient  $\ln$  from" due to weight percentage increase in TiC rein orcement is observed, whereas the highest oefficient is exhibited for as-cast set of AA 6063/T<sup>o</sup> composite and minimum for solutionized-aged s
- The predominant war mechanism is due to sliding in which the two-body a .d the three-body abrasion together played a major role.

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