RESEARCH ARTICLE-MECHANICAL ENGINEERING

Characterization of Fully Developed Pressure‑Driven, Shear‑Driven and Combined Pressure and Shear Driven Flow of Sisko Fluids Through Rectangular Channels

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Abstract

Pressure-driven, shear-driven and combined pressure and shear driven fow of a non-Newtonian Sisko fuid through rectangular channels is investigated. Inclusion of the aspect ratio in the formulation yields a highly nonlinear partial diferential equation, which is not reported in the existing literature. Thus, neither analytical nor numerical solution to this equation is available in the open literature. In the present study, the partial diferential equation, describing the fow, is solved employing the fnite diference method. Explicit method is adopted, and the solution for the non-dimensional velocity and wall shear stress is obtained. An exact solution for the fow of a Sisko fuid, for a special case (for non-Newtonian index 2), through large parallel plates (aspect ratio to be zero) is obtained. Expression for the friction factor, including the efect of the aspect ratio, is given. The efects of the aspect ratio, Sisko fuid parameter, non-Newtonian index on the non-dimensional velocity distribution and shear-stress distribution are analyzed both for shear-thinning and shear-thickening fuids. The results indicate that for pressure-driven fow, the efect of the aspect ratio on the velocity is negligible when it is less than 0.1. In case of shear-driven fow and combined pressure and shear driven fow also, the characteristics of fow through large parallel plates exist in nearly 50% of the channel for the aspect ratio of 0.1 or less, which means that for up to 50% of the channel, near the core, the parallel plates assumption will generate reasonably accurate results.

Keywords Sisko fuid · Aspect ratio · Shear-thickening fuid · Shear-thinning fuid · Pressure-driven fow · Shear-driven flow

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List of Symbols

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Greek Symbols

1 Introduction

Flow and heat transfer aspects of various fuids through rectangular channels find wide applications in different industrial and engineering processes. Numerous researchers considered fow through rectangular channel with low aspect ratio (ratio of depth and width of the channel) which reduced the problem to that of flow through large parallel plates, for which analytical solution is admissable. During the last decade, technological shift toward miniaturization attracted the research community to consider fow and heat transfer aspects of fuids through micro-channel which fnd wide spectrum of applications in the felds of micro-electromechanical systems (MEMS), bio-technology, small-scale heat exchangers etc. Some of the important studies are discussed here. Shashikumar et al. [[1\]](#page-21-0) investigated efects of diferent alloy nanoparticles of aluminum and titanium on micro-channel fow with partial slip and convective boundary condition. Menni et al. [[2](#page-21-1)] carried out analysis of turbulent flow and heat transfer in forced convection of pure water, pure ethylene glycol as base fuids, with dispersion of Al_2O_3 nanosized particles. Effects of different base fluids and shape of the nanoparticle on the fow and heat transfer were analyzed. Chamkha [\[3](#page-21-2)] studied laminar flow and heat transfer of a suspension in an electrically conducting fuid through channel and circular pipes, and the effect of the magnetic field, viscosity ratio on the friction coefficient, Nusselt number, particle-phase volume fow rate and liquid-phase fow rate was reported. Mixed convection of pulsating ferrofluid flow over a backward facing step has been studied by Selimefendigil et al. [[4\]](#page-21-3) employing fnite element method. Unsteady two-fuid fow and heat transfer of two-viscous fuids in a rectangular channel was studied by Umavathi et al. [\[5](#page-21-4)]. Analytical solution has been obtained, and the effect of viscosity ratio, conductivity ratio, Prandtl (Pr) number and the frequency parameter on the velocity and temperature has

been analyzed. Chamkha [[6\]](#page-21-5) studied mixed convection in a vertical channel with symmetric and asymetric wall heating conditions. Analytical solutions for velocity and temperature have been obtained, and the effects of pertinent parameters on velocity and temperature have been obtained. A comprehensive review on nanofuids applications in micro-channel has been carried out by Chamkha et al. [\[7](#page-21-6)]. Oscillatory flow and heat transfer in unsteady condition in a horizontal composite porous medium channel has been examined by Umavathi et al. [\[8](#page-21-7)]. Another interesting aspect of channel fow could be the fow in diferent geometries with fexible walls which have adjustable elasticity, thin size and desirable thermal properties. Some important and interesting studies on flow through cavities with fexible walls can be found in Refs. $[9-12]$ $[9-12]$.

The preceding discussion reports the studies on fow of Newtonian fuids through channel, micro-channel and the possibilities of exploring the fow through channels with fexible walls. However, it is well known that several fuids used in the industry do not follow the linear stress–strain rate relation of the Newtonian fuid model and exhibit nonlinear stress–strain rate behavior. For capturing the complex response of such fuids to shear force, several non-Newtonian fuid models such as power law fuid, third grade fuid, viscoelastic fuid, Casson fuid and Sisko fuid have been proposed by the researchers. Viscoelastic model possesses the characteristics of both solid and fuid, and this model has drawn the attention of numerous researchers [\[13](#page-21-10)[–15](#page-21-11)]. Narain et al. [[16\]](#page-21-12) reported how even Newtonian fuids can display the characteristics of viscoelastic fuids. The power law fuid model is another widely adopted non-Newtonian fluid model. Tso et al. [[17\]](#page-21-13) investigated flow and heat transfer of a power law fuid through large parallel plates. Exact solutions for the velocity and temperature distribution are obtained, and the efects of the non-Newtonian index and the non-Newtonian parameter on the temperature and heat transfer have been discussed. Electro-magnetohydrodynamic (EMHD) flow and heat transfer of a third-grade fuid through large parallel plates have been studied by Wang et al. [[18](#page-21-14)], and the ordinary diferential equations (ODEs) describing the fow and heat transfer have been solved by the perturbation method and spectral method. Exact analytical solutions of the velocity of a third-grade fuid, for Poiseuille and Couette-Poiseuille flow, flowing between parallel plates have been reported by Danish et al. [\[19\]](#page-21-15). The effect of the third-grade fuid parameter on the non-dimensional velocity has been analyzed and an expression for the fow rate is obtained. Akbarzadeh [[20\]](#page-21-16) investigated pulsatile magnetohydro-dynamic blood fow through the porous blood vessel, modeling blood as a third-grade fuid. The equations governing the fow are solved by the perturbation method and also by numerical technique. The effects of the pertinent nodimensional parameters on the velocity, fow rate and the wall shear stress have been discussed.

The above discussion emphasizes the importance of applications of the non-Newtonian fuids in diferent felds. One such fuid is the Sisko model which can describe the flow of several fluids in the high shear rate region. Different polymers, polymer melts, rubber melts and slurries are observed to follow the Sisko model. Sisko fuid model is introduced by A. W. Sisko [\[21](#page-21-17)], which is a three-parameter model. This model can be adopted to describe the fow of the fuids, mentioned above, for a wider range of shear rate. Some important investigations on Sisko fuid are discussed here. Khan et al. [[22\]](#page-22-0) investigated flow and heat transfer of a Sisko fuid in an annular pipe. The equations governing the fow and heat transfer have been solved by employing the homotopy analysis method (HAM) [\[23](#page-22-1)]. Chaudhuri and Das [[24](#page-22-2)] investigated forced convection of Sisko fuid in a pipe and semi-analytical solutions, adopting the least square method (LSM) [\[25,](#page-22-3) [26\]](#page-22-4) for the velocity and temperature have been obtained. The effects of the Sisko fluid parameter, non-Newtonian index on the velocity, temperature and heat transfer have been analyzed. Taylor's scrapping problem of a Sisko fuid has been studied by Siddiqui et al. [[27](#page-22-5)]. In a recent study, Khan et al. [[28\]](#page-22-6) investigated entropy generation in flow of a Sisko nanofluid employing HAM and the effect of various parameters on the entropy generation rate has been examined. Wire coating analysis by withdrawal from a bath of Sisko fuid has been studied by Sajid and Hayat [\[29](#page-22-7)]. Peristaltic fows of a Sisko fuid through an endoscope and through an inclined tube have been investigated by the researchers $[30, 31]$ $[30, 31]$ $[30, 31]$. Peristaltic flow, with heat and mass transfer efect, in a undulating porous curved channel has been studied by Zeeshan et al. [\[32](#page-22-10)].Implicit fnite diference (FDM) scheme is employed to solve the highly nonlinear ordinary diferential equations. Saheen et al. [\[33](#page-22-11)] investigated peristaltic fow of Sisko fuids, under the action of a magnetic feld, including the efect of viscous dissipation. The efect of variable thermal conductivity has been included. Flow of Sisko nanofuid over a curved surface has been numerically studied by Ali et al. [[34](#page-22-12)]. The effect of chemical process, radiation, thermophoresis and Lorentz force is considered.

The researches, cited above, indicate that the fow of Sisko fuids in various engineering systems is an important area of study. The fow of Sisko fuid through a rectangular channel, considering the efect of the aspect ratio, has never been investigated, though it fnds wide applications in industry involving fow of polymer melt, rubber melt. Most of the investigations on non-Newtonian fuids for rectangular geometry considered the fow through large parallel plates which neglected the effect of the aspect ratio. In most of the real-life situations, however, plates are not very large in the lateral direction and efect of the aspect ratio plays a signifcant role in the fow and thermal characteristics. In this regard, the investigation by Chaudhuri and Sahoo

[[35\]](#page-22-13) can be mentioned, which considered the effect of the aspect ratio on the magneto-hydrodynamic (MHD) fow of a third-grade fuid through a rectangular channel. In the present study, the efect of the aspect ratio has been considered for hydro-dynamically fully developed fow of a Sisko fluid through a rectangular channel. Inclusion of the effect of the aspect ratio in the governing equation yields a highly nonlinear partial diferential equation (PDE), required to describe the fow. On the contrary, if the fow is considered through large parallel plates (neglecting the efect of the aspect ratio), the equation describing the fow is a nonlinear ordinary diferential equation (ODE), relatively simple to obtain the solution. The PDE describing the fow of Sisko fuid through a rectangular channel is reported for the frst time in the present study, and numerical solution has been generated adopting the fnite diference scheme. The results of numerical solution are compared with that of the exact solution of Sisko fuid fowing through large parallel plates (aspect ratio zero) for a special case of the non-Newtonian index. The comparison of the results exhibits an excellent agreement which validates the results of the present numerical solution. Moreover, an expression of the friction factor has been provided. Non-dimensional velocity distributions and wall shear-stress variation at the lower wall, for three diferent cases, namely pressure-driven fow, shear-driven flow and combined pressure and shear driven flow, have been presented. The effects of the pertinent non-dimensional parameters such as aspect ratio, Sisko fuid parameter, non-Newtonian index and the non-dimensional pressure gradient on velocity and wall shear are discussed.

2 Mathematical Formulation

Hydro-dynamically fully developed fow of a Sisko fuid through a rectangular channel is considered. The physical problem addressed here is pictorially sketched in Fig. [1.](#page-3-0) The reference frame is fxed at the center of the channel as depicted in the fgure. The *x* axis is along the direction of the fow, *y* axis is chosen along the direction perpendicular to the fow and *z* axis is chosen along the lateral direction.

The flow is assumed to be steady, laminar and incompressible. The equations governing the fow are derived from the mass and momentum conservation principle which are the basic equations of fuid mechanics [\[22](#page-22-0)] and for the sake of brevity, not presented here. The stress tensor for non-Newtonian fuid difers from that of the Newtonian fuid.

The stress tensor equation $[22]$ $[22]$, in general, is given as:

$$
\tilde{T} = -p\tilde{I} + \tilde{S}
$$
 (1)

where $\tilde{T}, p, \tilde{S}, \tilde{l}$ are Cauchy stress tensor, thermodynamic pressure, extra stress tensor and unit stress tensor,

Fig. 1 Problem geometry

respectively. Depending on the behavior of the non-Newtonian fuids, the extra stress tensor introduces various forms of nonlinearity into the equation. The equation of extra stress tensor for the Sisko fuid [[22](#page-22-0)] is given as:

$$
\tilde{S} = \left[a + b \left(\sqrt{0.5 \text{tr}(A_1^2)} \right)^{n-1} \right] A_1 \tag{2}
$$

$$
\tilde{A}_1 = \tilde{L} + \tilde{L}^{\mathrm{T}} \tag{3}
$$

$$
\tilde{L} = \text{grad}\,\vec{V} \tag{4}
$$

 A_1 is the first Rivlin–Erickson tensor [\[22](#page-22-0)], *L* is the velocity gradient matrix [\[22](#page-22-0)], *a*, *b* and *n* are constants which vary for diferent Sisko fuids. For hydro-dynamically fully developed flow condition, only the axial component of velocity *u* exists and *u* is a function of *y*, *z* and independent of *x*. We seek a solution of the following form:

$$
\vec{V} = [u(y, z), 0, 0] \tag{5}
$$

where V is the velocity vector, u is the velocity of the fluid in the axial direction x . y and z are the coordinates along the vertical and lateral directions, respectively. The inertia term from the momentum equation vanishes as the flow considered is hydro-dynamically fully developed.

Using Eqs. (5) (5) , (1) (1) and (2) (2) , the momentum conservation equations in the *x*, *y* and *z* directions are obtained as follows:

From the momentum conservation equations in the other directions, it can be concluded that the pressure gradient along the axial direction is only function of x ; the LHS of Eq. ([6\)](#page-3-3) indicates that it is a function of *y* and *z* only, while the RHS is a function of *x* only. These two conditions can be satisfed only if

$$
\frac{\partial p}{\partial x} = C \tag{7}
$$

For shear-driven fow, the pressure gradient term is set to be zero, which then describes the PDE for shear-driven fow condition. If *u* is assumed to be independent of *z* (large parallel plate case, $A=0$), then Eq. [\(6\)](#page-3-3) [[22](#page-22-0)] reduces to the following:

$$
a\frac{d^2u}{dy^2} + nb\frac{d^2u}{dy^2}\left|\frac{du}{dy}\right|^{n-1} = \frac{dp}{dx}
$$
 (8)

2.1 Pressure‑Driven Flow

$$
At y = \pm L_1, \quad u = 0 \tag{9.1}
$$

$$
At z = \pm L_2, \quad u = 0 \tag{9.2}
$$

2.2 Shear‑Driven Flow

At
$$
y = -L_1
$$
, $u = 0$, at $y = L_1$, $u = U_p$ (10.1)

$$
At z = \pm L_2, u = 0 \tag{10.2}
$$

$$
a\left[\frac{\partial^2 u}{\partial y^2} + \frac{\partial^2 u}{\partial z^2}\right] + b\frac{\partial^2 u}{\partial y^2}\left[\left(\frac{\partial u}{\partial y}\right)^2 + \left(\frac{\partial u}{\partial z}\right)^2\right]^{\frac{n-1}{2}} + 2\left(\frac{n-1}{2}\right)b\frac{\partial u}{\partial y}\left[\frac{\partial u}{\partial y}\frac{\partial^2 u}{\partial y^2} + \frac{\partial u}{\partial z}\frac{\partial^2 u}{\partial y\partial z}\right]\left[\left(\frac{\partial u}{\partial y}\right)^2 + \left(\frac{\partial u}{\partial z}\right)^2\right]^{\frac{n-3}{2}} + b\frac{\partial^2 u}{\partial z^2}\left[\left(\frac{\partial u}{\partial y}\right)^2 + \left(\frac{\partial u}{\partial z}\right)^2\right]^{\frac{n-1}{2}} + 2\left(\frac{n-1}{2}\right)b\frac{\partial u}{\partial z}\left[\frac{\partial u}{\partial y}\frac{\partial^2 u}{\partial y\partial z} + \frac{\partial u}{\partial z}\frac{\partial^2 u}{\partial z^2}\right]\left[\left(\frac{\partial u}{\partial y}\right)^2 + \left(\frac{\partial u}{\partial z}\right)^2\right]^{\frac{n-3}{2}} = \frac{\partial p}{\partial x}
$$
(6)

 $2L_1$ and $2L_2$ are the depth and width of the rectangular channel, respectively, and U_p is the velocity of the upper plate in shear-driven and combined pressure and shear driven fow conditions. For combined pressure and shear driven fow, the boundary conditions are same as the boundary conditions of shear-driven flow.

For reducing Eq. ([6\)](#page-3-3) into its non-dimensional form, following non-dimensional variables and parameters are introduced:

$$
y^* = \frac{y}{L_1}, z^* = \frac{z}{L_2}, u^* = \frac{u}{u_0}, A = \frac{L_1}{L_2}, N = \frac{dp}{dx}\frac{L_1^2}{au_0}, b^* = \frac{b}{a}\left(\frac{u_0}{L_1}\right)^{n-1}
$$
(11)

where y^* , z^* , u^* , u_0 , A , N are the non-dimensional coordinates in the vertical and lateral directions, non-dimensional velocity in the axial direction, reference velocity, aspect ratio of the channel and non-dimensional pressure gradient. u_0 for shear-driven fow and combined pressure and shear driven

average velocity. This leads to an increase in the factor *N*. Therefore, *N* can be considered as a measure of the pressure gradient. *N* could be considered as unity also as assumed in the study of Danish et al. [[19\]](#page-21-15). But in that case, the efect of the higher pressure gradient cannot be included in the study. Average velocity, however, has to be calculated numerically (analytical solution, even if possible, will be valid for a very small range of parameters, only) which requires the values of the material constants *a*, *b* and *n*, which are not reported in the literature. For this reason, here, *N* is arbitrarily chosen, for which u_0 can be considered as a multiplication of the average velocity and some factor. This factor may be less than unity or greater than unity or even unity (unit value represents the average velocity).

Using the non-dimensional variables and parameters given by Eqs. (11) (11) (11) and (12) and omitting the asterisks for convenience, the non-dimensional forms of the governing equation and the boundary conditions are given below as:

$$
\frac{\partial^2 u}{\partial y^2} + A^2 \frac{\partial^2 u}{\partial z^2} + b \left[\left(\frac{\partial u}{\partial y} \right)^2 + A^2 \left(\frac{\partial u}{\partial z} \right)^2 \right]_{\frac{n-3}{2}}^{\frac{n-1}{2}} \frac{\partial^2 u}{\partial y^2} \n+ (n-1)b \frac{\partial u}{\partial y} \left[\left(\frac{\partial u}{\partial y} \right)^2 + A^2 \left(\frac{\partial u}{\partial z} \right)^2 \right]_{\frac{n-3}{2}}^{\frac{n-3}{2}} \left(\frac{\partial u}{\partial y} \frac{\partial^2 u}{\partial y^2} + A^2 \frac{\partial u}{\partial z} \frac{\partial^2 u}{\partial y \partial z} \right) \n+ A^2 b \left[\left(\frac{\partial u}{\partial y} \right)^2 + A^2 \left(\frac{\partial u}{\partial z} \right)^2 \right]_{\frac{n-1}{2}}^{\frac{n-1}{2}} \frac{\partial^2 u}{\partial z^2} \n+ A^2 (n-1)b \frac{\partial u}{\partial z} \left[\left(\frac{\partial u}{\partial y} \right)^2 + A^2 \left(\frac{\partial u}{\partial z} \right)^2 \right]_{\frac{n-3}{2}}^{\frac{n-3}{2}} \left(\frac{\partial u}{\partial y} \frac{\partial^2 u}{\partial y \partial z} + A^2 \frac{\partial u}{\partial z} \frac{\partial^2 u}{\partial z^2} \right) = N
$$
\n(13)

flow can be considered as the velocity of the upper plate. For pressure-driven flow condition, u_0 is considered as follows:

$$
u_0 = \frac{1}{N} \left(\frac{\mathrm{d}p}{\mathrm{d}x}\right) \frac{L_1^2}{a} \tag{12}
$$

It is important to mention that u_0 , defined in Eq. ([12](#page-4-0)), is not the average velocity through the channel. For a specific value of *N* only, u_0 from Eq. ([12](#page-4-0)) represents the average velocity. In case of Newtonian fuid, this *N* has got a specifc value for average velocity; however, in case of non-Newtonian fuid, the average velocity is not directly proportional to the pressure gradient, rather it is a nonlinear function of the pressure gradient. But to express the average velocity for non-Newtonian fuid in the same form as that of Newtonian fuid case, a factor *N* can be introduced and u_0 can be expressed as given in Eq. ([12](#page-4-0)). The value of *N* can be fixed once the average velocity is calculated. Higher pressure gradient results in a higher

Boundary conditions in the non-dimensional form for three diferent cases are given as follows:

2.3 Pressure‑Driven Flow

At
$$
y = \pm 1
$$
, $u = 0$ (14.1)

And, at
$$
z = \pm 1
$$
, $u = 0$ (14.2)

2.4 Shear‑Driven Flow

For shear-driven fow and combined pressure and shear driven fow, the velocity is made non-dimensional by taking the velocity of the upper plate U_p as the reference velocity. The boundary conditions for shear-driven flow are given below as:

At $y = -1$, $u = 0$ and at $y = 1$, $u = 1$ (15.1)

$$
At z = \pm 1, u = 0 \tag{15.2}
$$

Boundary conditions for combined pressure and shear driven flow are same as the boundary conditions of sheardriven flow condition.

2.5 Shear Stress

The expression of the shear stress, in the axial direction, for Sisko fuid is given as follows:

$$
\tau_{xy} = a \frac{\partial u}{\partial y} + b \left[\left(\frac{\partial u}{\partial y} \right)^2 + \left(\frac{\partial u}{\partial z} \right)^2 \right] \frac{\frac{n-1}{2}}{\partial y} \frac{\partial u}{\partial y}
$$
(16)

The shear-stress equation in the lateral direction is as given below:

$$
\tau_{xz} = a \frac{\partial u}{\partial z} + b \left[\left(\frac{\partial u}{\partial y} \right)^2 + \left(\frac{\partial u}{\partial z} \right)^2 \right]^{-\frac{n-1}{2}} \frac{\partial u}{\partial z}
$$
(17)

Upon substitution of the non-dimensional variables from Eq. (11) (11) into Eqs. (16) (16) and (17) , and omitting the asterisks, the non-dimensional forms of the shear-stress equations are as given below:

$$
\tau_{xy} = \frac{\partial u}{\partial y} + b \left[\left(\frac{\partial u}{\partial y} \right)^2 + A^2 \left(\frac{\partial u}{\partial z} \right)^2 \right] \frac{d^2 u}{dy} \tag{18}
$$

$$
\tau_{xz} = A \frac{\partial u}{\partial z} + b \left[\left(\frac{\partial u}{\partial y} \right)^2 + A^2 \left(\frac{\partial u}{\partial z} \right)^2 \right]^\frac{n-1}{2} A \frac{\partial u}{\partial z} \tag{19}
$$

To obtain the wall shear stress in the axial direction at the lower wall, $y=-1$ is substituted in Eq. [\(18](#page-5-1)). Shear stress along the lateral direction in the wall is zero as u is zero at the wall (no slip) rendering the gradient of u with respect to z is zero at all points. Therefore, from Eq. (16) (16) (16) we obtain that $\tau_{\rm xz}$ =0. For flow through large parallel plates, *A* is negligible and considered to be zero. Substituting $A = 0$ in Eq. ([18\)](#page-5-1), the expression of the shear stress in the axial direction reduces to the following:

$$
\tau_{xy} = \frac{\mathrm{d}u}{\mathrm{d}y} + b \left(\frac{\mathrm{d}u}{\mathrm{d}y}\right)^n \tag{20}
$$

3 Solution

The non-dimensional governing diferential equation (i.e., Eq. [13](#page-4-2)) is a second-order highly nonlinear PDE involving cross-derivatives which does not have any exact analytical solution. Moreover, getting a solution employing any of the approximate analytical method, too, is a formidable task. Hence, the nonlinear PDE has been solved numerically using fnite diference method. The nonlinear momentum equation is frst discretized using second-order accurate central diference scheme in order to keep the numerical error within the acceptable limit. The resulting system of algebraic equations is written in an explicit fashion, i.e., the right-hand side of the algebraic equation is calculated based on the known values of the variables from the previous iteration. The iterative solver is executed till the diference between the values of the unknown (here velocity) obtained from two successive iterations attains the required tolerance level. In order to make the results independent of the mesh size, mesh sensitivity test is carried out. The number of grid points both in y and z direction are varied from 10 to 300, and the local velocity as well as the wall shear stress are plotted as a function of number of grids. The grid size is fixed 200×200 as no appreciable variation in the local velocity and wall shear stress is observed by reducing grid size further. Uniform and equal grid spacing is considered. The convergence criterion is fxed at error less than 10−6. The error is defned as follows:

error =
$$
\sqrt{\sum_{j=1}^{M} \sum_{i=1}^{M} (u_{ij}^{k+1} - u_{ij}^k)^2}
$$
 (21.1)

where u_{ij}^k is the velocity in the *x* direction at *ij* node for the *k*th iteration. For the calculation of wall shear stress using Eq. [20](#page-5-2), the 1st velocity gradient at the wall is calculated by using one-sided three-point second-order diference scheme to keep the result second-order accurate.

3.1 Exact Solution for the Sisko Fluid, for *n***=2, Flowing Through Large Parallel Plates (***A***=0)**

Equation [\(8\)](#page-3-4), in the non-dimensional form, after omitting the asterisks, is given below as:

$$
\frac{\mathrm{d}^2 u}{\mathrm{d}y^2} + nb \frac{\mathrm{d}^2 u}{\mathrm{d}y^2} \left| \frac{\mathrm{d}u}{\mathrm{d}y} \right|^{n-1} = N \tag{22}
$$

Substituting $n=2$ in Eq. ([2\)](#page-3-2), we get the following form:

$$
\frac{\mathrm{d}^2 u}{\mathrm{d}y^2} + 2b \frac{\mathrm{d}^2 u}{\mathrm{d}y^2} \frac{\mathrm{d}u}{\mathrm{d}y} = N \tag{23}
$$

In the ODE, a modulus sign is present in Eq. (23) (23) . To get rid of this, Eq. (23) (23) (23) has been solved in the domain of *y* = −1 to *y*=0. In this domain, u decreases with a decrease in *y* (at the plate $u = 0$ and u increase toward the center). Beyond the center $(y=0)$, u decreases with an increase in *y*. Therefore, in the domain $y=0$ to $y=-1$, modulus of the velocity gradient is positive. In the domain $y=0$ to $y=1$, the velocity will be symmetric. Once the velocity in the domain *y*=−1 to $y=0$ is obtained, the velocity from $y=0$ to $y=1$ can be reproduced using the condition of symmetry. Equation ([23\)](#page-5-3) is a second-order, nonlinear ODE requiring two boundary conditions for obtaining the solution. The boundary conditions are given below as:

At
$$
y = -1
$$
, $u = 0$ (24.1)

$$
y = 0, \frac{\mathrm{d}u}{\mathrm{d}y} = 0 \tag{24.2}
$$

The solution of Eq. (23) (23) , using the boundary conditions by Eqs. (24.1) (24.1) – (24.2) (24.2) , is given below as:

$$
u = -\frac{1}{2b}(1+y) + \frac{1}{12b^2N} \left[(1+4bNy)^{1.5} - (1-4bN)^{1.5} \right]
$$
\n(25)

3.2 Exact Solution for Combined Pressure and Shear Driven Flow of a Sisko Fluid, with *n***=2, Through Large Parallel Plates (***A***=0)**

To obtain the solution for combined pressure and shear driven fow, governing diferential equations remain same as given by Eq. (23) (23) , but the boundary conditions are different from the previous case. It is a known fact that for combined pressure and shear driven fows, the maximum velocity does not occur always at the upper moving plate. Rather, the maximum velocity is observed to occur at a point away from the plate depending on the magnitude of the pressure gradient. It is important to note that the modulus sign present in the governing equation will produce positive values of the velocity gradient term (d*u*/d*y*) *n*−1 for odd values of *n*. However, for even *n*, modulus of (d*u*/ d*y*) *n*−1 will be negative beyond the point where the velocity is maximum. For $n=2$, an exact solution can be easily obtained. For combined pressure and shear driven fow, the location of the point where the peak velocity occurs, is unknown a priori. To locate the point, we will assume the distance from the center is being in the positive direction of the coordinate system as shown in Fig. [1.](#page-3-0) Then the domain is divided in two region $-1 ≤ y ≤ r$ and the other region is $r \leq y \leq 1$.

The governing diferential equations are given below as:

$$
\frac{d^2 u_1}{dy^2} + 2b \frac{d^2 u_1}{dy^2} \frac{du_1}{dy} = N \quad -1 \le y \le r \tag{26}
$$

$$
\frac{d^2 u_2}{dy^2} - 2b \frac{d^2 u_2}{dy^2} \frac{du_2}{dy} = N \quad r \le y \le 1
$$
 (27)

The boundary conditions are as follows:

$$
y = -1, \quad u_1 = 0 \tag{28.1}
$$

$$
y = r, \quad u_1 = u_2 \tag{28.2}
$$

$$
y = r, \quad \frac{\mathrm{d}u_1}{\mathrm{d}y} = \frac{\mathrm{d}u_2}{\mathrm{d}y} \tag{28.3}
$$

$$
y = 1, u_2 = 1 \tag{28.4}
$$

Solutions of Eqs. (26) (26) and (27) (27) (27) are given below as:

$$
u_1 = \frac{1}{2b} \left[-y + \frac{\left\{ 1 + 4b(Ny + k_1) \right\}^{1.5}}{6bN} \right] + k_2 \tag{29}
$$

$$
u_2 = -\frac{1}{2b} \left[-y - \frac{\left\{1 - 4b(Ny + k_3)\right\}^{1.5}}{6bN} \right] + k_4 \tag{30}
$$

where k_1 , k_2 , k_3 and k_4 are constants. Utilizing the boundary conditions given by Eqs. [\(28.1](#page-6-4))–[\(28.4](#page-6-5)), four algebraic equations are obtained; the solution of which provides the values of unknowns k_1 , k_2 , k_3 , and k_4 . The equations are as given below:

$$
\frac{1}{2b} \left[1 - \frac{\left\{ 1 + 4b(-N + k_1) \right\}^{1.5}}{6bN} \right] + k_2 = 0 \tag{31.1}
$$

$$
\frac{1}{2b} \left[-r - \frac{\left\{1 + 4b\left(Nr + k_1\right)\right\}^{1.5}}{6bN} \right] + k_2
$$
\n
$$
= \frac{-1}{2b} \left[-r - \frac{\left\{1 - 4b\left(Nr + k_3\right)\right\}^{1.5}}{6bN} \right] + k_4
$$
\n(31.2)

$$
\frac{1}{2b} \left[-1 + \sqrt{1 + 4b(Nr + k_1)} \right] = -\frac{1}{2b} \left[-1 - \sqrt{1 - 4b(Nr + k_3)} \right]
$$
\n(31.3)

$$
\frac{-1}{2b} \left[-1 - \frac{\left\{ 1 - 4b(N + k_3) \right\}^{1.5}}{6bN} \right] + k_4 = 1 \tag{31.4}
$$

Equations (31.1) – (31.4) (31.4) (31.4) are coupled, nonlinear algebraic equations. For a fixed value of *r*, the unknowns k_1 , k_2 , k_3 and k_4 can be determined by solving the coupled equations. As the equations are nonlinear, multiple roots are possible. Physically realistic values are chosen such that velocity feld is positive. The value of *r*, where the non-dimensional velocity attains the peak value, depends on the factors like the non-dimensional pressure gradient *N* and the Sisko fuid parameter *b*. Hence, *r* is an unknown which needs to be fixed. In order to determine the value of r , we again fix the coordinate system at the bottom surface. The governing equations remain the same as given by Eqs. (26) (26) (26) and (27) (27) (27) , but the boundary conditions get modifed as follows:

$$
y = -1, u_1 = 0 \tag{32.1}
$$

$$
y = 0.5 + r, u_1 = u_2 \tag{32.2}
$$

$$
y = 0.5 + r, \frac{du_1}{dy} = \frac{du_2}{dy}
$$
 (32.3)

$$
y = 1, u_2 = 1 \tag{32.4}
$$

The no-slip boundary conditions remain same as the previous case. But the point, where the peak velocity occurs, now has the coordinate of $0.5+r$. Therefore, in Eq. ([32.2\)](#page-7-0) and in Eq. (32.3) (32.3) (32.3) *r* will be replaced by $0.5 + r$ and the two sets of equation will be solved. The value of *r* has to be guessed and the solutions of the two sets are obtained. The velocity profles will coincide for the correct value of the r. Therefore, by hit and trial process, that *r* has to be fxed for which the velocity profles, obtained from the solution of Eqs. (26) and (27) (27) using the boundary conditions given by Eqs. ([31.1\)](#page-6-6)–[\(31.4](#page-6-7)) and Eqs. [\(32.1](#page-7-2))–([32.4\)](#page-7-3), are the same.

3.3 Shear‑Driven Flow

For shear-driven flow, the governing equation (with $A=0$) is given below as:

$$
\frac{\mathrm{d}^2 u}{\mathrm{d}y^2} + nb \frac{\mathrm{d}^2 u}{\mathrm{d}y^2} \left| \frac{\mathrm{d}u}{\mathrm{d}y} \right|^{n-1} = 0 \tag{33}
$$

The solution, using the boundary conditions of Eqs. (15.1) (15.1) (15.1) and (15.2) (15.2) , is given below as:

$$
u = \frac{1+y}{2} \tag{34}
$$

3.4 Evaluation of Friction Factor

Friction factor is an important parameter in determining the pressure drop in the channel for pressure-driven and

$$
\begin{pmatrix} 1 \end{pmatrix} \cdot
$$

combined pressure and shear driven fow conditions. The expression of the friction factors for fow of Sisko fuid can be yielded by modifying the available relation of the friction factor in case of Newtonian fluid. For flow of a Newtonian fuid through a circular pipe, the friction factor is given by the well-known relation:

For flow of non-Newtonian fluids, a modified Reynolds number is defned which is based on the apparent viscosity or efective viscosity of the fuids. In the present study, for the flow of a Sisko fluid, the apparent viscosity has to be identified first. For flow of a Sisko fluid through large parallel plates, the expression of the shear stress, in the axial direction, can be obtained from Eq. [\(16\)](#page-5-0) upon substitution which gives the required equation as follows:

$$
\frac{\partial u}{\partial z} = 0 \quad \tau_{xy} = a \frac{du}{dy} + b \left(\frac{du}{dy} \right)^{n-1} \tag{35}
$$

The effective viscosity can be obtained from Eq. (35) (35) as follows:

$$
\mu_{\rm e} = a + b \left(\frac{\mathrm{d}u}{\mathrm{d}y} \right)^{n-1} \tag{36}
$$

Equation (36) (36) can be modified to the following as mentioned by Gupta [[36\]](#page-22-14):

$$
\mu_{\rm e} = a + b \left(\frac{8u_0}{D_e} \right)^{n-1} \tag{37}
$$

where D_e is the hydraulic diameter [\[36](#page-22-14)] of the rectangular channel, as given below:

$$
D_e = \frac{4L_1L_2}{L_1 + L_2} \tag{38}
$$

$$
\text{Re}_{\text{m}} = \frac{1}{\frac{1}{\text{Re}_{\text{N}}\left(\frac{1}{1+A}\right)} + \frac{1}{\text{Re}_{\text{p}}\left(\frac{1}{(1+A)^n}\right)}}\tag{39}
$$

$$
Re_N = \frac{\rho u_0 4L_1}{a} \quad Re_p = \frac{\rho u_0^{2-n} (4L_1)^n}{8^{n-1} b} \tag{40}
$$

 Re_N and Re_n can be considered as the Reynolds numbers of the Newtonian (*a* represents the dynamic viscosity of the fluid when $b=0$) and power law fluid (when $a=0$, *b* is used to defne the modifed Reynolds number for power law fuid), respectively. Therefore, the friction factor for the pressuredriven fow of a Sisko fuid through a rectangular channel is given below as:

$$
f = \frac{16}{\text{Re}_{\text{m}}} \tag{41}
$$

Fig. 2 Comparison of the velocity distributions (pressure-driven flow) obtained from the numerical solution and the exact solution **a** *A*=0 and *n*=2 for *N*=−6; **b** *A*=0, *n*=2 and *N*=−8; **c** velocity

distributions (combined pressure and shear driven flow) when $A = 0$, *n*=2, *N*=−6; **d** *A*=0, *n*=2, *N*=−8

4 Results and Discussion

4.1 Validation

Before proceeding further with the study of the efect of parametric variation in the velocity distribution, frst, a comparison of the results obtained from the numerical solution for the limiting case of $A=0$, and the exact solution obtained for large parallel plate flow problem $(A=0)$ for $n=2$, has been made in Fig. [2a](#page-8-0)–d for pressure-driven and combined pressure and shear driven fow conditions. Figure [2](#page-8-0)a and c presents the comparison for *N*=−6, and for diferent values of the Sisko fuid parameter b for pressure-driven fow and combined pressure and shear driven fow. In Fig. [2b](#page-8-0) and d, comparison between the velocity distributions is made for *N*=−8 and diferent values of b for pressure-driven and combined pressure and shear driven fow conditions. In both the cases, the results of the numerical solution are observed to overlap with the exact solution which demonstrates the validity of the present numerical scheme. Then, the physically permissible values of the non-dimensional parameters have to be fxed, before carrying out the study of parametric variation. Sisko fuid parameter *b*, in the previous studies [\[10](#page-21-18)], has been varied in the range of 0.1 to 0.9. Therefore, in the present study, *b* has been varied in the same range. The value of *N* is considered in between −6 and −8. *n* has been varied in the range of 1.1 to 2.5 for shear-thickening fuids and for shear-thinning fuids, n has been varied in between 0.7 and 0.9.

Fig. 3 **a** Non-dimensional velocity distribution at $z=0$, for different *A*, when *N* = −6, *b* = 0.5 and *n* = 1.5, **b** Non-dimensional velocity distribution at $z=0$, for different *A*, when $N=-6$, $b=0.5$ and $n=0.8$,

4.2 Pressure‑Driven Flow

Efect of the aspect ratio on non-dimensional velocity distribution, for shear-thickening and shear-thinning fuids, is displayed in Fig. [3a](#page-9-0) and b, respectively. It is evident from the fgures, that both for shear-thickening and shear-thinning fuids, the non-dimensional velocity decreases with an increase in the aspect ratio. With an increase in the aspect ratio, the resistance offered by the side walls on the flow

c Non-dimensional wall shear-stress variation with *z* for diferent *A*, when $N = -6$, $b = 0.5$, $n = 1.5$, **d** Non-dimensional wall shear-stress variation with *z* for different *A* when $N = −6$, $b = 0.5$ and $n = 0.8$

1.00

increases. Therefore, the effect of wall friction on the velocity increases resulting in a decrease in the velocity. Figure [3](#page-9-0)c and d depicts efect of the aspect ratio on the wall shear-stress variation along the lateral direction, both for shear-thickening and shear-thinning fuids. The results indicate that the wall shear stress increases with a decrease in the aspect ratio. A decrease in the aspect ratio indicates an increase in the distance between the side walls. Therefore, the resistance offered by the side walls decreases which,

fnally, leads to the higher velocity resulting in an increase in the wall shear stress. Wall shear stress reaches maximum at the central plane, as the velocity is maximum and then it gradually reduces toward the side walls. It is interesting to note that for $A = 0.1$, Fig. [3](#page-9-0)c displays a different trend compared to other graphs. The reason is discussed as follows.

Lower values of the aspect ratio *A* indicates a higher length along the lateral direction of the channel. This means when *A* reaches a certain value, the effect of the side walls on the central part of the channel is insignifcant indicating fow through parallel plates. It can be mentioned here that for fow through parallel plates, it is assumed that the results are independent of the lateral coordinate (*z*). Practically, it means that the results of parallel plates case are valid only in the central portion of the channel. In Fig. [3](#page-9-0)c, the graph for *A*=0.1 displays a fatter portion near the central portion of the channel, which is in sharp contrast to the other curves for $A = 0.75$, 0.5 and 0.25. The flat portion means that the shear stress is invariant along the lateral direction (or independent of *z*). The graph for $A = 0.1$ shows that when aspect ratio is 0.1 or lower, the shear-stress distribution is independent of *z* for a signifcant portion along the central region indicating resemblance with fow through parallel plates. Figure [3](#page-9-0)d indicates a different when $A = 0.1$.

Figure [3c](#page-9-0) displays the result for $n = 1.5$ (shear-thickening fluids, $n > 1$), and Fig. [3d](#page-9-0) displays the results for $n = 0.8$ (shear-thinning fluids, $n < 1$). With a decrease in the aspect ratio *A*, the efect of the wall on the results diminishes and when *A* reaches a certain value, the variation with *z* is insignifcant for a large portion of the channel. Figure [3](#page-9-0) d displays that when aspect ratio *A* reaches 0.1, for $n = 0.8$ the results are independent of *z* resembling the fow through the parallel plate case.

Efect of Sisko fuid parameter on the non-dimensional velocity, for shear-thickening and shear-thinning fuids, is presented in Fig. [4](#page-11-0)a and b, respectively. It is evident from the fgures that the velocity decreases with an increase in the Sisko fuid parameter both for shear-thickening and shear-thinning fuids. An increase in b implies an increase in the viscosity efect, which indicates higher resistance for pressure-driven fow. Therefore, both for shear-thinning and shear-thickening fuids, velocity decreases. Similar trends have been reported for the flow of a Sisko fluid through annular pipe studied by Khan et al. [[10](#page-21-18)]. It is interesting to note, in Fig. [4](#page-11-0)a, that an increase in b from 0.1 to 0.3 results in a drastic reduction in the velocity. Whereas the decrease in velocity caused by a change in b from 0.3 to 0.5 is marginal. In this case, the value of the non-Newtonian index *n* is 1.8; $n=1$ signifies the case for Newtonian fluid; $n=1.8$ indicates a strong deviation from a Newtonian fuid. Therefore, when *b* changes from 0.1 to 0.3, the effect is drastic. On the contrary, when $n = 0.8$, the decrease in the velocity caused by an increase in *b* from 0.1 to 0.3 is relatively less.

For $n = 0.8$, the non-Newtonian effect is not so strong; as a result a change in b for 0.1 to 0.3 causes less efect in the velocity. The velocity in case of shear-thinning fuid is higher compared to the velocity in shear-thickening fuids in all the cases.

Figure [4c](#page-11-0) and d presents the efect of *b* on the wall shearstress variation along the lateral direction of shear-thickening and shear-thinning fuids. It is evident from the fgures that wall shear stress is higher for lower values of *b* as the velocity is higher resulting in a higher velocity gradient. As expected, the wall shear stress is a maximum at the center plane as the velocity is a maximum.

Non-dimensional velocity distribution at $z=0$, considering n as a parameter, for both shear-thickening and shearthinning fuids is presented in Fig. [5](#page-12-0)a and b, respectively. The results clearly indicate that an increase in the non-Newtonian index *n* causes a decrease in the velocity both for shear-thickening and shear-thinning fuids. However, the rate of decrease in velocity due to increase in *n* is lower in case of shear-thinning fuid compared to shear-thickening fluid. It is evident from Fig. [5a](#page-12-0) and b that, for other parameters remaining the same, for $n < 1$ (shear-thinning fluids), the velocity is higher than the velocity of the fluid for $n > 1$ (shear-thickening fuids). Figure [5](#page-12-0)a reveals that the velocity profile is almost uniform for $n=2$ and for higher values of *n* for shear-thickening fuids. It is important to note that for $n = 2.5$ $n = 2.5$, a different trend is displayed in Fig. 5a. With increase in the non-Newtonian index n, the resistance toward the flow increases. As a result, velocity decreases. For *n*=2.5, velocity remains unchanged along *y* meaning that velocity is nearly constant along the depth of the channel.

Non-dimensional wall shear-stress variation with *z* for diferent values of *n* is presented in Fig. [5c](#page-12-0) and d. It is evident from the fgure that an increase in *n* leads to a decrease in the wall shear stress for both shear-thickening and shearthinning fuids. An increase results in an increase in the viscosity efect, which results in a decrease in fow velocity. Consequently, the wall shear stress is reduced.

4.3 Shear‑Driven Flow

For shear-driven flow, the effect of various parameters on the non-dimensional velocity and wall shear stress is graphically presented in this section. Figure [6](#page-13-0)a and b depicts the efect of the aspect ratio on the non-dimensional velocity both for shear-thickening and shear-thinning fuids, respectively. The results indicate that with an increase in the aspect ratio *A*, the velocity decreases signifcantly. An increase in the aspect ratio (*A*) indicates higher resistance toward the flow, resulting in a decrease in the velocity. In Fig. [6](#page-13-0)a, the graph for $A = 0.1$ represents the case when efect of the aspect ratio is marginal and the velocity profle nearly follows a straight line starting from

Fig. 4 a Non-dimensional velocity distribution at $z=0$, for different values of *b*, when $A=0.8$, $N=-8$, $n=1.8$, **b** non-dimensional velocity distribution at $z=0$, for different values of *b*, when $A=0.8$, $N=-8$ and $n=0.8$, **c** non-dimensional wall shear-stress variation with *z*, for

zero at the lower plate and reaching the velocity of the upper plate linearly. Even the velocity profile for $A = 0.25$ also displays a linear pattern as depicted in Fig. [6a](#page-13-0). The efect of the aspect ratio is clearly indicated by the nonlinear nature of the velocity profile starting from $A = 0.5$. The linear profiles of the velocity up to $A = 0.25$ reveal that for values of the parameters $b=0.5$, $n=1.5$, $N=-8$, up to the aspect ratio 0.25, the flow can very well be considered as fow through large parallel plates and relatively simple ODE can be solved to obtain the velocity feld. Beyond

different *b*, when $A=0.8$, $N=-8$, $n=1.8$, **d** non-dimensional wall shear-stress variation with *z* for diferent *b*, when *A*=0.8, *N*=−8 and $n=0.8$

 $A = 0.5$, for given values of the other parameters, the effect of the aspect ratio is dominant and must be considered. For shear-thinning fuids also, Fig. [6b](#page-13-0) reveals that up to aspect ratio $A = 0.25$, the velocity profiles follow a linear pattern and the fow can very well be considered as fow through large parallel plates.

Efect of the aspect ratio on the non-dimensional wall shear-stress variation along *z* is depicted in Fig. [6](#page-13-0)c and d for shear-thickening and shear-thinning fuids, respectively. The results indicate that for $A = 0.1$, both for $n = 1.5$ and $n = 0.8$,

Fig. 5 a Non-dimensional velocity distribution at $z=0$ for different *n*, when *N* = −8, *A* = 0.8, *b* = 0.5, **b** non-dimensional velocity distribution at $z=0$ for different *n*, when $N=-8$, $A=0.8$, $b=0.5$, **c** non-

the wall shear stress is invariant along *z* in almost entire portion of the channel except near the walls. For $A = 0.25$, the deviation from a constant wall shear stress is large and the efect of the aspect ratio starts dominating.

Efect of the Sisko fuid parameter b on non-dimensional velocity, for shear-thickening fuids, is depicted in Fig. [7](#page-14-0)a. The results indicate that with an increase in *b*, the velocity decreases. When *b* changes from 0.1 to 0.3, the decrease in velocity is signifcant. For other values of *b*, the change in

dimensional wall shear-stress variation with *z* for diferent *n* when *A*=0.8, *N*=−8 and *b*=0.5, **d** non-dimensional wall shear-stress variation with *z* for different *n* when $A = 0.8$, $N = −8$ and $b = 0.5$

velocity is relatively marginal. For shear-thinning fuids, it is interesting to note that for $n=0.8$, velocity remains more or less unafected by any increase in *b*. Efect of the Sisko fuid parameter on variation of the wall shear stress is depicted in Fig. [7](#page-14-0)c and d both for shear-thickening and shear-thinning fuids respectively. It is evident from the fgure that with an increase in b , the wall shear stress decreases significantly. For $n = 0.8$, the wall shear stress is higher as displayed in the figures. For $n = 0.8$ ($n < 1$), the fluid exhibits shear-thinning

(b) Non-dimensional velocity distribution at $z = 0$ for different A, when $b=0.5$, $n=0.8$

Fig. 6 **a** Non-dimensional velocity distribution at $z = 0$ for different *A*, when $b=0.5$, $n=1.5$, **b** non-dimensional velocity distribution at $z=0$ for different *A*, when $b=0.5$, $n=0.8$, **c** non-dimensional wall shear-

behavior, resulting in a higher velocity compared to $n > 1$ case. Aspect ratio *A* in both the cases is chosen to be 0.8. Therefore, the velocity profle does not indicate any constant value of wall shear stress over any portion of the channel wall.

The effect of non-Newtonian index n on the non-dimensional velocity distribution at *z*=0, both for shear-thickening and shear-thinning fuids is depicted in Fig. [8](#page-15-0)a and b respectively. With an increase in *n*, the velocity is observed to decrease. A higher value of *n* indicates an increase in

stress variation with *z* for different *A* when $b=0.5$, $n=1.5$, **d** nondimensional wall shear-stress variation with *z* for diferent *A*, when $b=0.5$, $n=0.8$

the resistance toward the fow, thus causing a decrease in the velocity. The variation of wall shear stress with *n* is presented in Fig. [8](#page-15-0)c and d for shear-thickening and shearthinning fuids, respectively. Wall shear stress decreases with an increase in n both of shear-thickening and shear-thinning fuids. However, in case of shear-thinning fuids, the efect is marginal.

Fig. 7 a Non-dimensional velocity distribution at $z = 0$ for different *b*, when $n = 1.8$, $A = 0.8$, **b** non-dimensional velocity distribution at $z = 0$ for different *b*, when $n=0.8$, $A=0.8$, **c** non-dimensional wall shear-

4.4 Combined Pressure and Shear Driven Flow

Figure [9a](#page-16-0) and b presents effect of the aspect ratio on nondimensional velocity distribution at *z*=0, both for shearthickening and shear-thinning fuids, respectively, for combined pressure and shear driven fow conditions. It is evident from the fgures that efect of the aspect ratio *A* is signifcant on the velocity distribution both for shear-thickening and shear-thinning fuids. It is important to note that for a change in *A* from 0.1 to 0.25, decrease in velocity is not very high, but a change in *A* from 0.25 to 0.5 causes a drastic reduction

(b) Non-dimensional velocity distribution at $z = 0$ for different b, when n=0.8, A=0.8

stress variation with *z* for different *b* when $n=1.8$, $A=0.8$, **d** nondimensional wall shear-stress variation with *z* for diferent *b*, when $n=0.8$, $A=0.8$

in velocity both for shear-thickening and shear-thinning fuids. For shear-thinning fuid, the velocity is higher than the velocity of shear-thickening fuid for other parameters remaining the same.

Efect of the aspect ratio *A*, on the non-dimensional wall shear-stress variation on the lower wall, is depicted in Fig. [9c](#page-16-0) and d, for shear-thickening and shear-thinning fuid, respectively. For $n = 1.8$, it is observed from Fig. [9c](#page-16-0) that wall shear stress exhibits a fat portion, indicating an invariant shear stress along the lateral direction. This invariance along the lateral direction implies that for nearly 75% of the channel in

the central core, the fow is dominated by the characteristics of fow through large parallel plates for the chosen values of the parameters.

Efect of *b* on the non-dimensional velocity distribution at $z=0$, for shear-thickening and shear-thinning fluid, is presented in Fig. [10a](#page-17-0) and b, respectively. The results indicate that velocity decreases drastically with an increase in *b*, when *b* is small. Figure [10](#page-17-0)a depicts that for $b = 0.1$, the velocity profle exhibits mainly the characteristics of pressure-driven fow. But as *b* increases from 0.1 to 0.3, the characteristic of pressure-driven fow is suppressed and the

ferent *n* (shear thickening), when $b = 0.5$, $A = 0.8$, **d** non-dimensional wall shear-stress variation with z for different n (shear thinning), when $b = 0.5$, $A = 0.8$

thining), when $b=0.5$, $A=0.8$

 0.00

z

(d) Non-dimensional wall shear stress variation with z for different n (shear

 $0.25 \quad 0.50$

 0.75 1.00

 -0.1

 $-1.00 -0.75 -0.50 -0.25$

characteristics of shear-driven fow dominate. With a further increase in *, the velocity remains significantly lower than* the velocity of the moving plate, up to a major portion of the channel. Velocity again increases and fnally reaches the velocity of the moving plate only after reaching region very close to the plate.

Figure [10](#page-17-0)a displays the velocity distribution for combined pressure and shear driven flow. It is evident that for $b = 0.1$, the nature of the graph is diferent from the other graph. The non-Newtonian parameter *b* is a measure of the viscosity of the fuid. For lower values of *b*, the resistance toward

 $z=0$ for different A, when N=-6, b=0.5 and

Fig. 9 a Non-dimensional velocity distribution at $z=0$ for different *A*, when *N* = −6, *b* = 0.5 and *n* = 1.5, **b** non-dimensional velocity distribution at $z=0$ for different *A*, when $N=-6$, $b=0.5$ and $n=0.8$, **c** non-dimensional wall shear-stress variation with *z* considering *A* as

the fow is less (contribution of the pressure-driven fow is present) and fuid velocity increases in nearly all parts of the channel. But near the upper plate as the plate itself moves, the fuid velocity reaches that of the plate (no slip condition). Thus, velocity shows an increasing pattern up to a certain depth and then it decreases the plate velocity. But with increase in b, the resistance toward the flow increases and velocity in the entire section decreases. Near the upper plate, due to the movement of the plate, velocity increases

a parameter, when $N=-6$, $b=0.5$ and $n=1.5$, **d** non-dimensional wall shear-stress variation with z considering A as a parameter, when *N*=−6, *b*=0.5 and *n*=0.8

and reach that of the plate. Therefore, for higher values of *b*, velocity in lower and central part is less and then near the upper wall it shows an increasing trend ad reaches the plate velocity. Therefore, the graph for $b = 0.1$ shows an opposite trend.

For shear-thinning fuids, Fig. [10](#page-17-0)b indicates an increase in the velocity for all the corresponding values of the parameters. It is evident from the fgure that when b changes from 0.7 to 0.9, velocity decreases drastically and the decrease

Fig. 10 a Non-dimensional velocity distribution at $z=0$ for different values of *b* when $N = -6$, $A = 0.8$ and $n = 1.8$, **b** non-dimensional velocity distribution at $z=0$ for different values of *b* when $N=-6$, $A=0.8$ and $n=0.8$, **c** non-dimensional wall shear stress at $z=0$ for

in velocity, when *b* changes from 0.1 to 0.7, is gradual. For higher values of the non-Newtonian parameter *b*, the resistance toward the fow increases. Therefore, with increase in *b*, velocity decreases. For $b = 0.9$, velocity decreases significantly indicating that beyond this value of *b* (for the other values of the parameters as given) velocity is signifcantly lower in the entire channel.

at $z=0$ for different values of b when N=-6,

A= 0.8 and n= 0.8

different values of *b* when $N = -6$, $A = 0.8$ and $n = 1.8$, **d** non-dimensional wall shear stress at $z=0$ for different values of *b* when $N=-6$, *A*=0.8 and *n*=0.8

Wall shear stress, at the lower wall, along *z* for diferent values of *b* is presented in Fig. [10c](#page-17-0) and d. Figure [10](#page-17-0)d displays nearly unafected behavior with a change in *b*. With an increase in *b*, velocity decreases due to increase in fow resistance. However, this decrease in velocity is marginal when the velocity is already lower for higher values of *b*.

Figure [11](#page-18-0)a and b depicts the efect of *n* on the non-dimensional velocity distribution both for shear-thickening and

when N=-6, A=0.8 and $b=0.5$

(d) Non-dimensional wall shear-stress variation in lower wall along z for different n when $N=-6$, $A=0.8$ and $b=0.5$

Fig. 11 a Non-dimensional velocity distribution at $z=0$, for different *n* when $N = -6$, $A = 0.8$ and $b = 0.5$, **b** non-dimensional velocity distribution at $z=0$, for different *n* when $N=-6$, $A=0.8$ and $b=0.5$, **c** non-dimensional wall shear-stress variation in lower wall along *z*,

shear-thinning fuids. An increase in n causes a signifcant decrease in the velocity both for shear-thickening and shearthinning fuids. However, for shear-thickening fuid, velocity is higher as expected. For $n=2$, in case of shear-thickening fuid, the velocity decreases at a very lower value and then again it increases near the plate. It is interesting to note that velocity profle displays a diferent trend for with changes in *n*. Higher values of the non-Newtonian index *n* result in for different *n* when $N = -6$, $A = 0.8$ and $b = 0.5$, **d** non-dimensional wall shear-stress variation in lower wall along *z* for diferent *n* when *N*=−6, *A* = 0.8 and *b* = 0.5

higher values of apparent viscosity of the fuids at any shear rate. Therefore, with an increase in *n*, for combined pressure and shear driven flow, for a given N (non-dimensional pressure gradient) the resistance toward the fow increases which results in a decrease in the velocity. From the figure, it is clear that for *N*=−6 and *n*=1.1 the maximum velocity achieved is higher than that of the plate velocity. That means the shearing of the plate is retarding fow induced by the

 (b) Non-dimensional velocity distribution at $z=0$, for different n when N=-6, A= 0.8 and $b=0.5$

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(a) Variation of maximum nondimensional velocity with A, for pressure driven flow, considering b as a parameter when $N = -8$ and $n = 1.8$

Fig. 12 a Variation of maximum non-dimensional velocity with *A*, for pressure-driven fow, considering *b* as a parameter when *N*=−8 and $n = 1.8$, **b** variation of maximum non-dimensional velocity with *A*, for pressure-driven flow, considering *b* as a parameter when

N=−8 and *n*=0.8, **c** variation of maximum non-dimensional veloc-

 $N=-8$ and $n=0.8$

pressure and shear driven flow, considering b

as parameter when $N=-8$ and $n=0.8$

ity with *A*, for pressure- and shear-driven flow, considering *b* as parameter when *N*=−8 and *n*=1.8, **d** variation of maximum nondimensional velocity with *A*, for combined pressure and shear driven flow, considering *b* as parameter when $N = -8$ and $n = 0.8$

pressure gradient. For *n*=1.5, velocity decreases compared to the case of $n=1.1$ as shown in Fig. [11](#page-18-0)a. With further increase in n, for $n=2$, there is a significant decrease in the velocity in the central zone below that of the plate velocity. This may be attributed to the adverse effect of increased viscosity on the driving force due to the pressure gradient. However, near the upper plate (near $y=1$), velocity shows an increasing trend due to the infuence of the moving upper plate. At $y = 1$ (in the upper plate), velocity reaches the same velocity as of the plate. $n=1$ implies the case of Newtonian

fluid; therefore, $n = 1.1$ indicates that the deviation from the Newtonian fuid behavior is marginal. For this case, velocity is relatively higher and then near the upper plate, it reaches the plate velocity. The graph for this case $(n=1.1)$ frst shows an increasing trend with the depth followed by a decrease near the upper plate. In contrast, the velocity for $n=2$ displays in increase with depth, but rate of increase is marginal. Near the plate, there is a signifcant increase in the velocity due to the moving plate. Therefore, the nature of these graphs difers in their pattern.

An increase in *n*, results in a signifcant variation in the lower wall shear stress as depicted in Fig. [11c](#page-18-0) and d.

Figure [12a](#page-19-0) and b presents the variation of non-dimensional velocity with *A* for diferent values of *b*, for shearthickening and shear-thinning fuids, respectively, in case of pressure-driven fow. It is important to note that even for *A* in the range of 0.1 to 0.2, the change in the maximum velocity, for $n = 1.8$, is higher in case of pressure-driven flow condition, indicating a significant effect of the aspect ratio on the velocity. It is evident from Fig. [12](#page-19-0)a that the maximum velocity decreases with an increase in *A*. This rate of decrease of u_{max} with *A* is lower for higher values of *b* when *A* is greater than 0.5. For higher values of *A*, the velocity is lower. Any further increase in *b*, causes a marginal decrease in the velocity as revealed by Fig. [12](#page-19-0)a. Figure [12b](#page-19-0) reveals that for shear-thinning fluid when $n = 0.8$, the infuence of aspect ratio *A* on the maximum velocity is insignificant when *A* lies in the range 0.1 to 0.2. For $b = 0.9$, the effect of *A* on u_{max} is less even for $A = 0.4$. For higher values of *b*, the velocity is lower; with any increase in *A*, then, the rate of decrease in velocity is lower rendering u_{max} insensitive toward *A* even for *A*=0.4. For combined pressure and shear driven flow, the effect of A on u_{max} is depicted in Fig. [12c](#page-19-0) and d for shear-thickening and shear-thinning fluid, respectively. It is interesting to note from Fig. [12c](#page-19-0) that u_{max} is unaffected by *A* for $A > 0.5$, when $b = 0.9$. When $b=0.7$, u_{max} is unaffected when *A* lies in the range 0.8 to 1. The results indicate that max is nearly unafected by *A* on the higher side, for higher values of *b*. Higher values of *b* imply higher flow resistance, nullifying the effect of pressure gradient in case of combined pressure and shear driven fow, rendering the maximum non-dimensional velocity to be unity (same as upper plate velocity). It is interesting to note that effect of change in the parameter b diminishes as the parameter *A* increases. Increase in b implies increase in the viscosity and thus higher resistance to the fow causing lower velocity. Similarly, higher values of the aspect ratio *A* means higher resistance offered by the wall friction resulting in lower velocity. It is evident from the fgure that for $A = 0.75$ and above and for $b = 0.3$ and above, velocity is low. For this range of *A*, with an increase in *b*, the rate of decrease in the velocity is marginal. As the velocity is relatively low, any further increase in *b* affects the velocity

marginally. However, for $b = 0.1$ to $b = 0.3$, the decrease in velocity with a decrease in *b* is signifcant even for higher values of $A(A=0.75$ and above) as depicted by Fig. [12c](#page-19-0).

Due to the combined efect of increase in *b* and *A*, velocity decreases signifcantly and any further change in *b* or even in *A* cannot affect the velocity significantly which is displayed by the graph for $b = 0.7$.

For $n = 0.8$, as presented in Fig. [12d](#page-19-0), it is observed that the effect of *A* on u_{max} is less for higher values of *b* even for *A* up to 0.5. Beyond $A = 0.5$, the effect of *A* on u_{max} is signifcant for all values of *b*.

5 Conclusion

The pressure-driven, shear-driven and combined pressure and shear driven fow of Sisko fuid through rectangular channels has been investigated. Efect of the aspect ratio for fow of Sisko fuid has never been considered, and PDE, governing the flow, is reported for the first time. The governing PDE is strongly nonlinear in nature, and obtaining a closed-form solution or approximate analytical or semianalytical solution is a formidable task. Explicit fnite difference scheme with an iterative solver is used to generate the numerical solution of the equation, and the variation of velocity and wall shear stress has been studied. The important conclusions made from the study are given below:

- For pressure-driven flow, for *A* up to 0.1, effect of the aspect ratio in a rectangular channel can be neglected.
- The wall shear-stress variation along the lateral direction remains nearly unaltered for around 60% of the channel indicating the characteristics of fow through large parallel plates.
- In case of shear-driven flow and combined pressure and shear driven flow also, the characteristics of flow through large parallel plates are observed in nearly 50% of the channel for $A = 0.1$ or less. This implies for up to 50% of the channel, near the core, the l assumption of large parallel plate generates reasonably accurate results, beyond which the effect of the aspect ratio is significant on several occasions.
- An increase in the aspect ratio in case of both sheardriven and combined pressure and shear driven fow, from 0.1 to 1, drastically decreases the velocity in a large portion of the channel. Whereas the linear velocity profles, obtained in case of large parallel plate assumptions, for shear-driven flow, are observed for up to $A = 0.25$ and less, for both shear-thickening and shear-thinning fuids.
- The velocity of shear-thinning fuid is higher than the velocity of shear-thickening fuid in all cases.

- The effect of n on velocity and wall shear stress is very strong both for shear-thickening and shear-thinning fuids.
- With an increase in n , velocity and shear stress decreases significantly.
- The effect of *b* in presence of the aspect ratio effect is interesting. An increase in *b* from 0.1 to 0.3, when deviation of *n* from 1 is large, results in a drastic decrease in the velocity for pressure-driven, shear-driven and combined pressure and shear driven fows. This feature is depicted for the shear-thickening fuid case shown for $n=1.8$. However, when *n* is close to unity, the effect of *b* on the velocity is gradual over a relatively wide range for all the cases.

In the present study, only the momentum conservation equation is solved and the results may be useful for study of forced convection heat transfer of Sisko fuid through rectangular channels, where the velocity feld is required to solve the energy conservation equation. For further investigation, flow and heat transfer aspect through channel with flexible walls can be considered; the advantages of fexible walls are already discussed. Forced convection and mixed convection can be considered with temperature-dependent properties, which will produce a coupled set of momentum and energy conservation equation, requiring more computational effort. It is important to mention that no experimental results on the flow of Sisko fluids through channel for pressure-driven, shear-driven and combined pressure and shear driven fow are reported in the open literature. Therefore, the results of the present study cannot be compared with any experimental data. In future, experimental study can also be conducted and the results of the present study can be compared.

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