ORIGINAL ARTICLE



# **Isothermal High‑Temperature Oxidation Behaviour of Nickel‑Based Superalloy IN740H**

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**Abstract** The isothermal oxidation behaviour of IN740H alloy was studied by conducting experiments in a thermogravimetric analyser in the presence of air and synthetic fue gas in the temperature range of 760–900 °C. SEM/XRD/ EPMA/Raman spectroscopy was used for analysing surfaces/cross sections of oxidized/corroded specimens. The oxide layer consists of Ti-doped  $Cr_2O_3$  and NiCr<sub>2</sub>O<sub>4</sub> in airoxidized specimens, whereas additional NiO and NiS form in the presence of fue gas at 900 °C. Alumina forms at the oxide/matrix interface and at 900 °C; it was followed by *γ*′ denuded zone with dispersed (Ti, Nb)C. The kinetics of oxidation in a steady state are found to be sub-parabolic for all the conditions.

**Keywords** IN 740H · High-temperature oxidation · Advanced ultra-supercritical

# **1 Introduction**

Nickel-based superalloy IN740H is a candidate material for manufacturing superheater and reheater tubes for advanced ultra-supercritical (A-USC) steam boilers in coal-fred power plants [\[1](#page-10-0)[–6](#page-10-1)]. IN740H is considered to be a prospective alloy for manufacturing tubes for A-USC steam boilers in coalfred power plants. Superheater and reheater tube material should have adequate oxidation resistance  $(=2 \text{ mm cross--})$ section loss in  $2 \times 10^5$  h) in addition to superior mechanical

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properties to withstand the stringent operating conditions in A-USC boilers [[6\]](#page-10-1). The improvement in efficiency over ~ 50% can be brought by operating the boilers at the steam temperature of 760 °C and pressures of 35 MPa [\[1](#page-10-0)]. During operations, the metal surface temperature of superheater and reheater tubes may reach 800 °C or even more [[7,](#page-10-2) [8](#page-10-3)]. The damage mode in the superheater/reheater tubes is reported to be a combination of creep, oxidation, and hot corrosion  $[7-11]$  $[7-11]$  $[7-11]$ . However, the possibility of running the plants with frequently reduced load introduces thermal transient leading to thermal fatigue [[1](#page-10-0), [2](#page-10-5), [12](#page-10-6)]. Therefore, the deformation and damage result from the combined efect of creep and fatigue. The fnal damage is ascribed to the combined action of creep, fatigue, oxidation, and hot corrosion. It is known that the surface condition has a major role, as thermal fatigue cracks mostly initiate at the surface [[12](#page-10-6)[–14\]](#page-10-7). For example, Sahu et al. [[12](#page-10-6)]*.* in their study on fatigue and hot corrosion interaction in alloy Nimonic 263 have shown that fatigue cracks were initiated at the sites where the protective chromium oxide layer was destroyed by the hot corrosive media. Moreover, creep voids and cavitations ultimately result from the grain boundary attack and deterioration at the surface as shown by Bagui et al. [[14\]](#page-10-7). It is also known that the composition has a considerable efect on the oxidation and hot corrosion behaviour of these alloys. For example, it has been reported by Cruchley et al. [[15\]](#page-10-8) that the oxidation kinetics of the outer chromia layer in the temperature range of 600–900 °C was faster for Ti-doped chromia in comparison with Ti-free chromia in nickel-based superalloy RR1000. Wang et al. [\[16](#page-10-9)] have shown that the Cr difusion was enhanced in the presence of Ni in Co-17Re-23Cr alloy at high temperature and the protective chromia layer growth was promoted.

With such complex interacting damage modes [[13](#page-10-10), [14\]](#page-10-7) and with the fact that altering the alloy composition has a significant effect on high-temperature oxidation resistance [\[15,](#page-10-8) [16](#page-10-9)], it is essential to ascertain the type of oxide scale formed in the newly developed nickel-based superalloy IN740H. Moreover, research on the oxidation behaviour of IN740H is very scarce, though few researchers reported the high-temperature oxidation and hot corrosion behaviour of Inconel 740 [\[7](#page-10-2)[–11\]](#page-10-4). Zhao et al. [\[7](#page-10-2)] have carried out the oxidation of Inconel 740 in air between 850 and 1000 °C to investigate the oxidation behaviour. The formation of  $Cr_2O_3$ ,  $TiO<sub>2</sub>$ ,  $Al<sub>2</sub>O<sub>3</sub>$ , and (Ni, Co)  $Cr<sub>2</sub>O<sub>4</sub>$  has been reported [\[7](#page-10-2)[–11](#page-10-4)]. Aung et al.  $[10]$  $[10]$ . have studied the effect of  $SO<sub>2</sub>$  in flue gas at 750 °C by high-temperature electrochemical sensor and found that sulphidation attack was more pronounced in the flue gas atmosphere with  $SO<sub>2</sub>$ . The present study highlights the isothermal oxidation behaviour of alloy IN740H in air and synthetic fue gas atmosphere at 760 °C, 800 °C, and 900 °C. Isothermal oxidation studies have been carried out in a thermogravimetric analyser (TGA) to study the oxidation kinetics and oxide formation when exposed to the above-mentioned temperature range in air and synthetic fue gas atmosphere. The surface oxides have been extensively characterized by scanning electron microscopy (SEM), X-ray difraction (XRD) analysis, and Raman spectroscopy. The cross section of the oxide layer was analysed by electron probe microanalysis (EPMA), and line scans were done to generate the profle of elements distribution present across the oxide layer.

#### **2 Experimental Procedure**

The material was received in the form plate of 25.4 mm thickness in hot-rolled, solution-annealed, and age-hardened condition. The as-received plates were solutionized at 1120 °C for 1 h aged at 800 °C for 4 h and then air-cooled to generate the typical microstructure in which it is put into service. As per the Manufacturer's data [[6\]](#page-10-1), the recommended temperature range for hot-forming operations such as forging or hot-rolling is between 870 and 1190 °C. Annealing practices are described in ASME code case 2702, which specifies a temperature of 1100 °C minimum, for 1 h per one inch (25.4 mm) of thickness but not less than 30 min. The annealing range may extend as high as 1160 °C. Specifc annealing conditions will be dependent upon the product form and intended application. Ageing shall be performed at a temperature between 760 °C and 815 °C, for a minimum of 4 h. The minimum ageing time shall be increased for thicknesses above 2 inches (50.8 mm) at a rate of  $\frac{1}{2}$  hour per inch of additional thickness. Ageing shall be followed by air cooling. The alloy has a two-phase microstructure consisting of *γ* (nominally Ni, Co) matrix and approximately an 18% volume fraction of *γ'* precipitates (nominally Ni<sub>3</sub>[Al, Ti]). The chemical composition of the as-received material is shown in Table [1](#page-1-0).

Specimens of size  $10 \times 10 \times 4$  mm were cut by electrical discharge machining (EDM) for oxidation and hot corrosion studies from the as-received plate material. Wire-cut EDM setup (Make: Steer Engineering Pvt. Ltd.; Model: DK7712) was used for sample cutting, having axis travel (X-120 mm, Y-160 mm, Z-100 mm), and power supply (2 KW). The specimens were ground to 1200 grit size emery papers followed by polishing with alumina powder of 1 μm. After polishing, specimens were ultrasonically cleaned in ethanol. A digital weighing balance (Metler, PGS07) was used to record the weight of the specimens before and after high-temperature oxidation experiments. The specimen was then placed in the ceramic bucket inside the TGA furnace to study the isothermal oxidation kinetics of the alloy. The high-temperature oxidation was performed in the temperature range of 760–900 °C for 100 h in the presence of laboratory-grade air and synthetic fue gas in TherMax 700 TGA (sensitivity of balance of  $\pm$  1 µg and temperature stability of  $\pm 1$  °C). The synthetic flue gas consisted of  $15\%CO_2 + 3.5\%O_2 + 0.25\%SO_2$  and balance N<sub>2</sub>. During oxidation experiments, the fow rate of the pre-mixed fue gas and air was maintained at 80 ml min<sup>-1</sup>.

The post-oxidation characterization of the oxide surface as well as a cross section of the oxide scales was done using FEI Nova nano SEM-430 equipped with an energy-dispersive X-ray microanalysis (EDX). For cross-sectional analysis of the oxide layer, one side of the specimens was ground to 1200 grit size emery papers followed by polishing with 0.5 μm alumina powder. The cross section was examined by highresolution SEM and EPMA (JXA-8230, JEOL) to measure the oxide thickness and study the cross-section morphology of the oxide layer. A line scan was performed across the oxide layer by EPMA to know the elemental distribution across the thickness of the oxide layer. Prior to SEM and EPMA analysis, X-ray difraction (XRD) analysis and Raman spectroscopy of the surface oxides were performed. XRD was done on a D8 Discover, Bruker XRD system using Cu K*α* radiation, indexed between 2θ values 20–100°. Raman spectroscopy was also done to confrm the oxides formed on the surface of oxidized specimens. The Raman spectroscopic studies were done using Alpha 300R Confocal Raman Microscope, WiTech GmbH, having a wavelength of 532 nm, and a charged coupled device



<span id="page-1-0"></span>**Tabl** of as (CCD) detector with grating size (600 g/mm), and spectral resolution down to 0.1 cm−1. All the spectra recorded each spot on the surface of the samples for 10 s.

## **3 Results and Discussion**

### **3.1 Gravimetric Oxidation Kinetics**

The change in the specifc weight of specimens oxidized in air and fue gas in the temperature range of 760 and 800 °C for 100 h is shown in Fig. [1](#page-2-0)a, b. It is observed that there is a rapid mass gain in the initial 2–3 h of the transient stage of oxidation in all the oxidizing conditions. The graphs also reveal the higher oxidation rate during the transient period of oxidation. The thickness of the oxide layer or the depth of oxidation depends on temperature and the duration of the initial transient state of oxidation. After the transient period, a uniform layer of oxides is formed and thus the oxidation rate slows down. During steady-state conditions, the rate of oxidation is slow. Figure [2](#page-2-1) shows the change in the specific weight of specimens oxidized in air and fue gas at 900 °C. At 900 °C, the initial gain in mass is almost fve times higher than the mass gain at 760 and 800 °C and it also shows two slopes in the stable region during oxidation in fue gas. The mass gain in the case of synthetic fue gas specimens is higher in comparison with specimens oxidized in the air at all the tested temperatures. The oxidation kinetics can be described by examining the growth-time constant, *n* which is the exponent in Eq. [\(1](#page-2-2))

$$
\left(\frac{\Delta w}{A}\right) = k_p t^n + C \tag{1}
$$

where  $(\Delta w/A)$  is the oxidation mass per unit surface area, *t* is oxidation time in s, and *C* is constant.

The value of *n* is obtained from the slope of the log-log plot and is indicative of the mechanism controlling the



<span id="page-2-1"></span>**Fig. 2** Weight change per unit area of the specimen oxidized in air and fue gas atmosphere at 900 °C

<span id="page-2-2"></span>oxidation rate. The exponent *n* values obtained for all oxidizing conditions at each temperature are collated in Table [2.](#page-3-0) If the value of  $n < 0.5$ , it is sub-parabolic, and if  $n = 0.5$  then it is parabolic growth kinetics. It is said to be linear if the value of  $n = 1$ . The exponent *n* value was also verified by nonlinear curve ftting of the oxidation weight gain curves. It was found to follow the power function  $y = ax^b$  ( $a > 0$ ,  $0 < b < 1$ ). The exponent *b* values are depicted in Table [2](#page-3-0). The parameter *b* decides the ability of curves to be stable with time. The power function becomes parabolic if exponent *b* is 0.5 [[17\]](#page-10-12). The exponent *n* values of the steady state and *b* values obtained are the same as evident from Table [2](#page-3-0). In the present investigation, both exponent values are less than 0.5, and thus, the oxidation kinetics is sub-parabolic in all the oxidizing conditions at all the investigated temperature ranges in the steady-state condition except the specimen oxidized



<span id="page-2-0"></span>**Fig. 1** Weight change per unit area of the specimens oxidized in air and fue gas atmosphere at **a** 760 °C and **b** 800 °C

<span id="page-3-0"></span>**Table 2** The parabolic rate constant and the exponent values in steady-state region



in the fue gas atmosphere at 900 °C, which has the *n* value closer to 0.5 and seems to follow parabolic kinetics.

The oxidation kinetics of Ni–Cr–Al-based superalloys usually follow the parabolic rate law as given in Eq. [\(1](#page-2-2))  $[11-13]$  $[11-13]$ . The parabolic rate constant is calculated using Eq. ([2\)](#page-3-1).

$$
\left(\frac{\Delta w}{A}\right)^2 = k_p t \tag{2}
$$

where  $k_p$  represents the parabolic rate constant and  $(\Delta w/A)$ is the specifc weight gain of the specimens for exposure time, *t*. The calculated  $k_p$  values are listed in Table [2](#page-3-0) for all the oxidizing conditions. Parabolic oxidation kinetics signify that a thermal difusion process is rate determining. Such a process may include a uniform difusion of one or more than one reactant through the growing compact scale. The parabolic rate constant which can be represented as a temperature dependent function by using an Arrhenius equation is shown in Eq.  $(3)$  $(3)$ .

$$
k = Ae^{-E_a/RT}
$$
 (3)

where  $k$  is the rate constant,  $E_a$  is activation energy (kJ) mole<sup> $-1$ </sup>), *T* is the temperature (K), *R* is ideal gas constant  $(8.3145 \text{ J/K}^{-1} \text{ mol}^{-1})$ , and *A* is the frequency or pre-exponential factor.

Both *A* and  $E_a$  are specific to a given reaction. The plot of  $k_p$  versus the inverse of temperature results in a straight line the slope of which gives the activation energy of the reaction. The linear expression is obtained by taking the log on both sides. The term activation energy is not rigorously correct as it involves more than one element. However, it is useful as it describes the temperature dependence reaction kinetics.

The slope of the plot of ln *k* versus the inverse of temperature (*1/T*) gives the activation energy values as shown in Fig. [3.](#page-3-3) It is observed that the reaction with higher activation energy,  $E_a$ , has a steeper slope. So, the reaction rate is very sensitive to temperature change. In contrast, in the reaction with lower activation energy,  $E_a$  is less sensitive to

<span id="page-3-1"></span>

<span id="page-3-3"></span>**Fig. 3** Arrhenius plots corresponding to oxidation of alloy IN740H in air and fue gas atmosphere at temperatures 760, 800, and 900 °C

<span id="page-3-4"></span><span id="page-3-2"></span>**Table 3** Activation energy,  $E_a$  in steady-state region

Oxidation conditions	$E_{\rm a}$ (kJ/mol)
Oxidized in air	289
Oxidized in synthetic flue gas	394

temperature change. The obtained activation energy values under diferent oxidizing conditions are given in Table [3](#page-3-4). It is observed that the activation energy values for specimens oxidized in flue gas are 394 kJ mol<sup>-1</sup> and in air 289 kJ mol<sup>-1</sup>, respectively. The value of the specimen oxidized in fue gas is higher than in the air atmosphere, indicating that the oxidation in fue gas is more sensitive to the temperature.

#### **3.2 Characterization of Oxides**

#### *3.2.1 Air Atmosphere*

Figure [4](#page-4-0)a shows the XRD patterns of the oxide scale formed on the specimens subjected to isothermal exposure in the temperature range of 760–900 °C in air. From the XRD analysis, it is evident that the Ti-doped  $Cr_2O_3$  and  $Nicr_2O_4$  spinel are present in the oxide scales formed during isothermal oxidation. The specimen surface oxidized in the air atmosphere at all the investigated temperatures was also identifed and analysed by Raman spectroscopy. The oxides found by Raman spectroscopy are listed in Table [4.](#page-4-1) A representative Raman analysis confirming the presence of  $Cr_2O_3$ , NiCr<sub>2</sub>O<sub>4</sub> and  $TiO<sub>2</sub>$  on a specimen oxidized in air is shown in Fig. [5](#page-5-0)a.

It is observed that surface oxide morphology changes with oxidizing temperature. SEM images revealing the surface oxide morphology of IN740H specimens oxidized in air at the temperature of 760–900 °C are shown in Fig. [6](#page-6-0). The EDS analysis of the oxide scale of specimens oxidized in air shows Cr-, Ti-, and Ni-rich oxides on the surface.

Figure [7](#page-7-0) shows the EPMA mapping of a cross section of the specimen oxidized in air at 760 °C which reveals that Tidoped chromia formation on the outer oxide layer followed by alumina at the interface of matrix and oxide layer. A very thin layer of  $\sim$  1 µm of mixed oxides of Ti-doped Cr<sub>2</sub>O<sub>3</sub> is formed on the specimen. At 800  $\degree$ C,  $\sim$  5-µm-thick oxide layers consisting of Ti-doped  $Cr_2O_3$  and NiCr<sub>2</sub>O<sub>4</sub> sandwiched between the outer oxide layer and the matrix is observed. Figure [9a](#page-9-0) shows the cross section of the specimen oxidized in air at 900 °C. The thickness of the oxide scale is  $\sim$  10 µm consisting of Ti-doped  $Cr_2O_3$  formed on the outer layer and  $NiCr<sub>2</sub>O<sub>4</sub>$  as the intermediate layer, followed by  $Al<sub>2</sub>O<sub>3</sub>$  at the interface of oxide/matrix. (Ti, Nb)C and oxide protuberance is observed in the adjacent matrix region of *γ*′ denuded zone.

The average oxide thickness is collated in Table [5.](#page-7-1) Due to the high Cr fraction in the alloy,  $Cr_2O_3$  was formed during the initial stage of oxidation. From the EPMA line analysis shown in Fig. [10](#page-9-1)a, it is observed that complex oxides are formed on specimens at diferent temperatures. The difusivity of elements like Ti and Al increases with an increase in temperature. The formation of oxides depends on the alloying elements (such as Cr and Al) and the difusion process. Due to the high affinity of aluminium towards oxygen,



<span id="page-4-0"></span>**Fig. 4** XRD plots showing diferent oxides formed during oxidation of alloy IN740H at 760, 800, and 900 °C: **a** in air and **b** fue gas atmosphere

900 °C
311
195, 239, 444
566, 655, 703

<span id="page-4-1"></span>**Table 4** Oxides confrmed by Raman analysis



<span id="page-5-0"></span>**Fig. 5** Raman peaks found on specimen oxidized at 800 °C: **a** in air and **b** in fue gas atmosphere

aluminium from the  $\gamma'$  diffuses out and thus forms  $Al_2O_3$  at the interface [\[17](#page-10-12)].

#### *3.2.2 Flue Gas Atmosphere*

The XRD analysis of specimens exposed to the fue gas atmosphere in the temperature range of 760–900 °C shows the presence of Ti-doped  $Cr_2O_3$ , Ni $Cr_2O_4$ , and NiO phases along with nickel sulphide(NiS) phase as shown in Fig. [4b](#page-4-0). The existence of NiO and NiS phases is observed only on the specimen oxidized at 900 °C. The Raman analysis also confrms the presence of NiS on specimens oxidized in fue gas and is shown in Fig. [5](#page-5-0)b.

The surface morphology observed under SEM of the specimen oxidized at 760–900 °C is shown in Fig. [8.](#page-8-0) The oxide morphology of the specimen oxidized in fue gas shows a dense chromia oxide when compared to the specimen oxidized in air. Sulphur was also traced in the EDS analysis on the surface layer of the specimen. In the fue gas atmosphere at 760  $\degree$ C,  $\sim$  1-µm thin layer of Ti-doped chromia is observed on the specimen. The  $Cr_2O_3$  layer thickness increases with an increase in temperature. At 800 °C, also a thin layer of  $\sim$  1.5 µm thickness of similar oxides was observed on the specimen. The oxide layer thickness at 900  $\degree$ C is ~ 15 µm, and oxide thickness at all the temperatures is compiled in Table [5.](#page-7-1) It clearly indicates that the thickness of the oxide layer formed at 760 and 800 °C is less, while there is a drastic change in oxide thickness at 900 °C. Figure [9](#page-9-0)b and [10](#page-9-1)b shows the cross-section oxide morphology and line scans of the specimen oxidized in fue gas at 900 °C, respectively. At 900 °C (Ni, Co),  $Cr_2O_4$  spinel is formed as an outer layer and  $\text{Al}_2\text{O}_3$  at the oxide/matrix interface followed by *γ*′ denuded zone.

#### **3.3 Discussion**

The formation of oxides depends on the alloying elements (such as Cr and Al) and the difusion process. Oxidation generally starts with the nucleation of NiO and  $Cr_2O_3$  and by the end of the transient state a continuous layer of  $Cr_2O_3$ forms [[29](#page-10-22)].  $Cr_2O_3$  is stoichiometric oxide, so diffusion is low.  $Cr_2O_3$  is a protective layer. The anti-exfoliating ability of scale is improved by the presence of  $Cr_2O_3$ , and the oxidation resistance of an alloy is strengthened by the presence of  $Al_2O_3$  and  $SiO_2$  [[17\]](#page-10-12). Ti-rich oxides are known to form on the outer surface of the chromia layer. The ability of chromia to dissolve titanium in the oxidized state has been known for some time. It is likely that titanium is incorporated as the  $Ti<sup>4+</sup>$  ion since this has minimal effect on the chromia lattice spacing [[15\]](#page-10-8). The difusivity of elements like Ti and Al increases with an increase in temperature. Due to the high afnity of aluminium towards oxygen, aluminium from the *γ*' gets dissolved in the matrix and difuses towards the outer layer. The Al<sup>3+</sup> reacts with O<sup>2−</sup> forms Al<sub>2</sub>O<sub>3</sub> at the interface [\[17](#page-10-12)], thus forming *γ*′ denuded zone. A1 oxidizes extensively internally, permitting Cr to difuse to the alloy–oxide interface and maintain the  $Cr_2O_3$  layer; rather, it oxidizes internally. The Cr<sub>2</sub>O<sub>3</sub> supplies O<sub>2</sub> for the formation of  $Al_2O_3$ internally. However, the dissociation pressure of  $Cr_2O_3$  is low and is sufficient to cause the formation of  $A<sub>1</sub>O<sub>3</sub>$  beneath the  $Cr_2O_3$  layer, leading to a substantially slow oxidation rate in the later stage [\[30](#page-10-23)].

In the presence of flue gas, high  $SO<sub>3</sub>$  partial pressure is required to stabilize the low-melting-point solution of sulphates to sustain the hot corrosion during the propagation stage [[8\]](#page-10-3). The depletion of Cr and Al oxide takes place due to higher ion activity as it forms basic fux and thereby



<span id="page-6-0"></span>**Fig. 6** SEM images showing the surface oxide morphology on the specimens oxidized in laboratory air at **a** 760 °C; **b** 800 °C; **c** 900 °C and the corresponding bulk EDS plots

<span id="page-7-0"></span>**Fig. 7** EPMA EDS elemental mapping of specimen oxidized in air at 760 °C



<span id="page-7-1"></span>**Table 5** Average thickness, µm of oxide layer



increases the oxidation rate than in normal oxidation. During the steady state, the oxidation rate slows down as thick  $Cr_2O_3$  formed on the surface prevents fluxing with sulphates, and as a result, there is a meagre presence of sulphur in the oxide layer. Thus, the depletion of  $Cr_2O_3$  also slows down and reduces the inward difusion of sulphur in the forms of S,  $S^{2-}$ , SO<sub>2</sub> etc. The specimens up to 800 °C show slow oxidation kinetics, whereas at 900 °C the sulphur accelerates the oxidation rate. The infuence of diferent elements on the high-temperature oxidation process will depend on the particular mechanism of accelerated oxidation that is operative.

# **4 Conclusions**

- (i) The oxidation rate was higher in the specimens oxidized in the fue gas atmosphere due to the difusion of sulphur.
- (ii) From the obtained *n* values, it is found that the rate of oxidation is almost sub-parabolic in the steady state.



<span id="page-8-0"></span>**Fig. 8** SEM images showing the surface oxide morphology on the specimens oxidized in fue gas at **a** 760 °C; **b** 800 °C; **c** 900 °C and the corresponding bulk EDS plots



<span id="page-9-0"></span>**Fig. 9** EPMA images of cross section of specimens oxidized under air and fue gas at 900 °C indicating the possible phases formed: **a** specimen oxidized in the air; **b** specimen oxidized in fue gas



<span id="page-9-1"></span>**Fig. 10** EPMA line analysis of a cross section of specimens oxidized under air and fue gas at 900 °C **a** specimen oxidized in air **b** specimen oxidized in fue gas

- (iii) The thickness of the oxide layer formed mainly depends on the temperature and the duration and rate of oxidation in the transient period.
- (iv) The oxide thickness formed at 900  $\degree$ C was five times that of the thickness of oxide formed at 760 °C and 800 °C.
- (v) Oxide protuberance and MC type carbides in *γ*′ denuded zone were prominent at 900 °C in specimen oxidized in air, while in fue gas, only *γ*′ denuded zone formation was observed.

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# **Declarations**

**Confict of interest** The authors whose names are listed immediately below title of manuscript certify that they have NO affiliations with or involvement in any organization or entity with any fnancial interest or non-fnancial interest in the subject matter or materials discussed in this manuscript.

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