# Positioning and Obstacle Avoidance of Automatic Guided Vehicle in Partially Known Environment

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**Abstract:** This paper presents positioning and obstacle avoidance of Automatic Guidance Vehicle (AGV) in partially known environment. To do this task, the followings are done. Firstly, the system configuration of AGV is described. Secondly, mathematical kinematic modeling of the AGV is presented to understand its characteristics and behavior. Thirdly, the Simultaneous Localization and Mapping (SLAM) algorithm based on the laser measurement system and encoders is proposed. The encoders are used for detecting the motion state of the AGV. In a slippery environment and a high speed AGV condition, encoder positioning method generates big error. Therefore, Extended Kalman Filter (EKF) is used to get the best position estimation of AGV by combining the encoder positioning result and landmark positions obtained from the laser scanner. Fourthly, to achieve the desired coordinate, D\* Lite algorithm is used to generate a path from the start point to the goal point for AGV and to avoid unknown obstacles using information obtained from laser scanner. A backstepping controller based on Lyapunov stability is proposed for tracking the desired path generated by D\* Lite algorithm. Finally, the effectiveness of the proposed algorithms and controller are verified by using experiment. The experimental results show that the AGV successfully reaches the goal point with an acceptable small error.

Keywords: Automatic guided vehicle, differential drive, localization, obstacle avoidance, tracking.

# 1. INTRODUCTION

The use of AGV is the one of the most preferred means to reduce the operation costs by helping the factories to automate a manufacturing facility or warehouse. The common challenging problems related with AGV operation are positioning, path planning, obstacle avoidance and trajectory tracking.

The commonly used positioning methods in AGV were visual line follower using camera sensor [1], inductive guidance using electrical wire buried under the floor [2], semi-guided navigation method by using magnetic tapes [3], wall following algorithm [4] and laser navigation system [5]. However, these algorithms depend on predetermined paths or landmark positions. Therefore, they can only work in known environments.

To generate the optimal path from the start to goal position using a given map, rapidly-exploring random trees algorithm [6], potential function [7] and gradient method (GM) [8] were proposed. Those algorithms only work for known and static environments.

In factory environments, a sudden arrival of obstacle

could block the AGV path. To deal with this, several obstacle avoidance algorithms were proposed such as curvature velocity method [9], dynamic window approach [10], moving obstacle avoidance [11], and obstacle avoidance based on obstacle geometric [12]. However, as the calculated area are limited, the goal position reachability condition is not guaranteed.

After an optimal path is generated, a trajectory tracking control algorithm is need for the AGV to track the optimal path. Several control algorithms are proposed to accomplish this task such as Fuzzy-PID [1], sliding mode control theory [13], and time varying feedback control law [14]. Among these controllers, although the stability of the system is guaranteed, it might not be easy to find an appropriate control law.

To solve these problems, this paper proposes a new algorithm for positioning, path planning, obstacle avoidance and trajectory tracking in partially known environment. Positioning of AGV is obtained by using SLAM algorithm based on EKF. For path planning and obstacle avoidance, D\* Lite algorithm based on a given map and laser scanner data is proposed. To guarantee the tracking errors to be-

Manuscript received December 15, 2014; revised October 8, 2015; accepted December 29, 2015. Recommended by Associate Editor Dong-Joong Kang under the direction of Editor Fuchun Sun. This research was supported by a grant (11 Transportation System- Logistics 02) from Transportation System Efficiency Program funded by Ministry of Land, Infrastructure and Transport (MOLIT) of Korean government.

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bcome to zero, backstepping control method based on Lyapunov stability is introduced. Finally, the effectiveness of the proposed algorithms and controller are verified by experiments. The experimental results show that the AGV successfully reaches the goal point with an acceptable small error.

## 2. SYSTEM DESCRIPTION

The AGV used in this paper is shown in Fig. 1. The dimension of the AGV is 100 cm x 60 cm x 80 cm. This system uses a differential drive wheeled configuration. Two driving wheels are mounted on the left and right sides of AGV, and are driven by two BLDC motors. Two passive castor wheels are installed in front and back sides of AGV to support the AGV. The laser navigation system NAV-200 used for positioning sensor with an accuracy  $\pm 25$  mm is mounted on the top of AGV. Industrial PC as the main controller is placed inside the AGV platform and touch screen monitor as the input and display is located on back side of AGV. The batteries for power supply are mounted in the middle of AGV.

The electrical design schematics diagram of AGV is shown in Fig. 2. AGV uses 2 encoders and one laser scanner LMS151 for SLAM. LMS151 is also used for both landmark detection and obstacle detection. A laser navigation system NAV200 is used for measuring the position in absolute coordinate.



Fig. 1. The mechanical design of AGV.



Fig. 2. Electrical design schematics diagram.

## 3. KINEMATICS MODELING

Fig. 3 shows the system modeling of the differential drive AGV system.

The kinematic equation of nonholonomic differential drive type of AGV system shown in Fig. 3 can be expressed as follows:

$$\mathbf{x}_{\nu} = \int \dot{\mathbf{x}}_{\nu} dt, \tag{1}$$

$$\dot{\mathbf{x}}_{\nu} = \begin{bmatrix} X_A \\ \dot{Y}_A \\ \dot{\theta}_A \end{bmatrix} = \begin{bmatrix} \cos\theta_A & 0 \\ \sin\theta_A & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} V_A \\ \omega_A \end{bmatrix}, \quad (2)$$

$$\begin{bmatrix} V_A \\ \omega_A \end{bmatrix} = \frac{r}{2} \begin{bmatrix} 1 & 1 \\ \frac{1}{b} & -\frac{1}{b} \end{bmatrix} \begin{bmatrix} \dot{\phi}_R \\ \dot{\phi}_L \end{bmatrix}, \qquad (3)$$

or in discrete type

$$\mathbf{x}_{\nu k} = \mathbf{x}_{\nu k-1} + \Delta \mathbf{x}_{\nu},\tag{4}$$

$$\Delta \mathbf{x}_{\nu} = \begin{bmatrix} \Delta X_A \\ \Delta Y_A \\ \Delta \theta_A \end{bmatrix} = \begin{bmatrix} \cos(\theta_A + \Delta \theta) & 0 \\ \sin(\theta_A + \Delta \theta) & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} \Delta s \\ \Delta \theta \end{bmatrix},$$
(5)

$$\begin{bmatrix} \Delta s \\ \Delta \theta \end{bmatrix} = \frac{r}{2} \begin{bmatrix} 1 & 1 \\ \frac{1}{b} & -\frac{1}{b} \end{bmatrix} \begin{bmatrix} \Delta \phi_r \\ \Delta \phi_l \end{bmatrix}, \tag{6}$$

where  $\mathbf{x}_{v}$  is the posture vector of AGV,  $(X_A, Y_A)$  is AGV position in global coordinates,  $\theta_A$  is the AGV orientation which is taken counterclockwise from the *X* axis,  $V_A$  is the linear velocity and  $\omega_A$  is the angular velocity, *r* is the



Fig. 3. System modeling.

wheel radius, *b* is the distance between the wheels and center of AGV,  $\dot{\phi}_R$  and  $\dot{\phi}_L$  are the right and left wheel angular velocities,  $\Delta s$  is the linear displacement of AGV,  $\Delta \theta$  is the change of rotational angle of AGV, and  $\Delta \phi_R$  and  $\Delta \phi_L$  represent the change of right and left wheel rotation angle, respectively.

#### 4. POSITIONING

In this paper, positioning of AGV can be obtained using SLAM algorithm based on EKF since the system is nonlinear. To do this, the following are done. Firstly, the positions of the landmarks are obtained using laser scanner LMS-151. Secondly, the position of the AGV is predicted using EKF prediction step based on encoder data. Thirdly, the positions of AGV and landmarks are updated using EKF update step based on landmark positions.

## 4.1. Landmark detection

In this paper, spike landmark extraction is used to extract the features that can be easily re-observed and distinguished from the environment. This method uses extrema to get landmarks by finding values in the range of current laser beam that differs by more than a certain amount such as 0.5 meters from previous laser beam. The landmarks detected using spike algorithm are shown in Fig. 4.

In this paper, laser scanner LMS-151 is used to detect landmarks and obstacles. The resolution of laser scanner is 0.5° and the scanning frequency is 25 Hz. The detected object can be expressed either with polar coordinate as  $(d_i,\beta_i)$ , where  $d_i$  is the distance between sensor and object, and  $\beta_i$  is the scanning angle or with Cartesian coordinate



Fig. 4. Landmark detection.

in local coordinate frame of sensor as follows:

$$(x_{iL}, y_{iL}) = (d_i \cos \beta_i, d_i \sin \beta_i) \text{ for } i = 1, 2, \dots, n.$$
 (7)

Using coordinate transformation, the position of landmark  $y_i$  in the global coordinate is obtained as:

$$\begin{bmatrix} X_{i,k} \\ Y_{i,k} \end{bmatrix} = \begin{bmatrix} X_{A,k} \\ Y_{A,k} \end{bmatrix} + \begin{bmatrix} \sin \theta_A & \cos \theta_A \\ -\cos \theta_A & \sin \theta_A \end{bmatrix} \begin{bmatrix} x_{iL} \\ y_{iL} \end{bmatrix} + D \begin{bmatrix} \cos \theta_A \\ \sin \theta_A \end{bmatrix}, \quad (8)$$

where  $(X_A, Y_A)$  is the position of AGV in global coordinate,  $\theta_A$  is the orientation of AGV from *X* axis and *D* is the distance between the center of the robot and the sensor.

#### 4.2. Prediction

The estimated position obtained from prediction step  $\hat{\mathbf{x}}_{k|k-1}$  and the covariance matrix obtained from prediction step  $\mathbf{P}_{k|k-1}$  at current sampling time *k* based on encoder data are written as follows:

$$\hat{\mathbf{x}}_{k|k-1} = \hat{\mathbf{x}}_{k-1|k-1} + \mathbf{F}_x^T f(\hat{\mathbf{x}}_{\nu,k-1}, \mathbf{u}_{k-1}),$$
(9)

$$\mathbf{P}_{k|k-1} = \mathbf{A}_k \mathbf{P}_{k-1|k-1} \mathbf{A}_k^T + \mathbf{W}_k \mathbf{Q}_{k-1} \mathbf{W}_k^T, \qquad (10)$$

where  $\mathbf{x} = \begin{bmatrix} \mathbf{x}_{v} & \mathbf{y}_{1} & \cdots & \mathbf{y}_{n} \end{bmatrix}^{T}$  is the state vector that consists of AGV posture vector  $\mathbf{x}_{v} = \begin{bmatrix} X_{A} & Y_{A} & \theta_{A} \end{bmatrix}^{T}$ and the landmark position vector  $\mathbf{y}_{i} = \begin{bmatrix} X_{i} & Y_{i} \end{bmatrix}^{T}$ ,  $\hat{\mathbf{x}}_{k-1|k-1}$  is the estimated position obtained from the previous update,  $\mathbf{F}_{x} = \begin{bmatrix} \mathbf{I}^{(3)} & \mathbf{O}^{(3)\times(2n)} \end{bmatrix}$  is used to make sure that only the AGV estimation position is updated,  $f(\hat{\mathbf{x}}_{v,k-1}, \mathbf{u}_{k-1})$  is the mathematic model of AGV,  $\hat{\mathbf{x}}_{v,k-1}$  is the AGV estimated posture vector obtained from the previous update,  $\mathbf{u}_{k-1}$  is the control input,  $\mathbf{A}_{k}$  is the Jacobian of prediction model,  $\mathbf{P}_{k-1|k-1}$  is the covariance matrix obtained from the previous update,  $\mathbf{W}_{k}$  is the process noise,  $\mathbf{Q}_{k-1}$  is the process noise covariance. This prediction is based on the input  $\mathbf{u}_{k-1}$  that consists of changes of right encoder  $\Delta \phi_r$  and left encoder  $\Delta \phi_l$  respectively. AGV mathematic modeling in discrete type in (5) and (6) can be reduced as follows:

$$f(\hat{\mathbf{x}}_{\nu,k-1},\mathbf{u}_{k-1}) \equiv \Delta \mathbf{x}_{\nu}$$

$$= \begin{bmatrix} \cos(\hat{\theta}_{A,k-1} + \Delta \theta) & 0\\ \sin(\hat{\theta}_{A,k-1} + \Delta \theta) & 0\\ 0 & 1 \end{bmatrix} \begin{bmatrix} \Delta s\\ \Delta \theta \end{bmatrix}, \quad (11)$$

$$\begin{bmatrix} \Delta s\\ \Delta \theta \end{bmatrix} = \frac{r}{2} \begin{bmatrix} 1 & 1\\ \frac{1}{b} & -\frac{1}{b} \end{bmatrix} \begin{bmatrix} \Delta \phi_r\\ \Delta \phi_l \end{bmatrix},$$

$$\mathbf{u}_{k-1} = \begin{bmatrix} \Delta \phi_r & \Delta \phi_l \end{bmatrix}, \quad (12)$$

where *b* is the distance between the left and right wheels,  $\Delta s$  is the linear displacement of AGV, and  $\Delta \theta$  is the rotational angle of AGV.

The covariance matrix  $\mathbf{P}_{k|k-1}$  and Jacobian of prediction model  $\mathbf{A}_k$  are defined as:

$$\mathbf{P}_{k|k-1} = \begin{bmatrix} P_{xx} & P_{xy_1} & \cdots & P_{xy_n} \\ P_{y_1x} & P_{y_1y_1} & \cdots & P_{y_1y_n} \\ \vdots & \vdots & \ddots & \\ P_{xx} & P_{xx_1} & P_{xx_2} & P_{xx_2} \end{bmatrix}, \quad (13)$$

$$\begin{bmatrix} P_{y_1x} & P_{y_ny_1} & P_{y_ny_n} \end{bmatrix}$$
$$\mathbf{A}_k = \mathbf{I}^{(3+2n)} + \mathbf{F}_x^T \mathbf{A}_{x,k} \mathbf{F}_x, \tag{14}$$

where

$$\mathbf{A}_{x,k} = \frac{\partial f}{\partial \mathbf{x}_{\nu}} = \begin{bmatrix} 0 & 0 & -\Delta s \sin(\theta_A + \Delta \theta/2) \\ 0 & 0 & \Delta s \cos(\theta_A + \Delta \theta/2) \\ 0 & 0 & 0 \end{bmatrix}.$$
(15)

Equation (14) is used to make sure that only the covariance matrix of AGV is updated.

The process noise  $W_k$  can be calculated as:

$$\mathbf{W}_k = \mathbf{F}_x^T \mathbf{W}_{u,k},\tag{16}$$

where process noise for AGV  $\mathbf{W}_{u,k}$  and its covariance  $\mathbf{Q}_{k-1}$  with error constants  $k_r$  and  $k_l$  are defined as:

$$W_{u,k} = \frac{\partial f}{\partial u}$$
  
=  $\frac{1}{2} \begin{bmatrix} \cos(\theta_k) - \frac{\Delta s}{b} \sin(\theta_k) & \cos(\theta_k) + \frac{\Delta s}{b} \sin(\theta_k) \\ \sin(\theta_k) + \frac{\Delta s}{b} \cos(\theta_k) & \sin(\theta_k) - \frac{\Delta s}{b} \cos(\theta_k) \\ \frac{2}{b} & -\frac{2}{b} \end{bmatrix}$ (17)

with

$$\boldsymbol{\theta}_{k} = \boldsymbol{\theta}_{A} + \frac{\Delta \boldsymbol{\theta}}{2}, \ \boldsymbol{Q}_{k-1} = \begin{bmatrix} k_{r} |\Delta \boldsymbol{\phi}_{r}| & 0\\ 0 & k_{l} |\Delta \boldsymbol{\phi}_{l}| \end{bmatrix}.$$
(18)

## 4.3. Update

The estimated position obtained from update step  $\hat{\mathbf{x}}_{k|k}$ and the covariance matrix obtained from update step  $\mathbf{P}_{k|k}$  at sampling time *k* based on landmarks data are written as follows:

$$\hat{\mathbf{x}}_{k|k} = \hat{\mathbf{x}}_{k|k-1} + \mathbf{K}_k v_k \text{ for } \mathbf{K}_k = \mathbf{P}_{k|k-1} \mathbf{H}_k^T \mathbf{S}_k^{-1}, \qquad (19)$$

$$\mathbf{P}_{k|k} = (\mathbf{I} - \mathbf{K}_k \mathbf{H}_k) \mathbf{P}_{k|k-1},$$
(20)

where  $\mathbf{K}_k$  is Kalman gain,  $\mathbf{v}_k$  is the innovation as difference between landmark positions obtained from measurement and prediction. If the number of landmark is i = 1, 2, ..., *n*, the innovation of each landmark  $\mathbf{v}_{i,k}$  is calculated as follows:

$$v_{i,k} = z_{i,k} - h(\hat{x}_{v,k|k-1}, \hat{y}_{i,k|k-1}),$$
(21)

$$h(\hat{\mathbf{x}}_{\nu,k|k-1}, \hat{\mathbf{y}}_{i,k|k-1}) = \begin{bmatrix} \hat{P}_{i,k|k-1} \\ \hat{\alpha}_{i,k|k-1} \end{bmatrix} \\ = \begin{bmatrix} \sqrt{(\hat{X}_{i,k|k-1} - \hat{X}_{A,k|k-1})^2 + (\hat{Y}_{i,k|k-1} - \hat{Y}_{A,k|k-1})^2} \\ \tan^{-1}(\frac{\hat{Y}_{i,k|k-1} - \hat{Y}_{A,k|k-1}}{\hat{X}_{i,k|k-1} - \hat{X}_{A,k|k-1}}) - \hat{\theta}_{A,k|k-1} \end{bmatrix},$$
(22)

$$z_{i} = \begin{bmatrix} \hat{\rho}_{i,k} \\ \hat{\alpha}_{i,k} \end{bmatrix}$$
$$= \begin{bmatrix} \sqrt{(X_{i,k} - \hat{X}_{A,k|k-1})^{2} + (Y_{i,k} - \hat{Y}_{A,k|k-1})^{2}} \\ \tan^{-1}(\frac{Y_{i,k} - \hat{Y}_{A,k|k-1}}{X_{i,k} - \hat{X}_{A,k|k-1}}) - \hat{\theta}_{A,k|k-1} \end{bmatrix}, \quad (23)$$

where range  $\hat{\rho}_i$  is the estimated distance between AGV and current positions of landmarks, and  $\hat{\alpha}_i$  is an estimated angle between AGV and landmark current position,  $(\hat{X}_{A,k|k-1}, \hat{Y}_{A,k|k-1}, \hat{\theta}_{A,k|k-1})$  is the AGV posture obtained from prediction step,  $(\hat{X}_{i,k|k-1}, \hat{Y}_{i,k|k-1})$  is the landmark position obtained from prediction step.  $\mathbf{z}_i$  is the measurement value vector of landmark obtained using landmark position  $\mathbf{y}_i$  in global coordinate from (8).

The innovation covariance  $S_k$  can be defined as:

$$\mathbf{S}_k = \mathbf{H}_k \mathbf{P}_{k|k-1} \mathbf{H}_k^T + \mathbf{R}, \tag{24}$$

$$\mathbf{R} = \begin{bmatrix} \sigma_{\rho\rho} & 0\\ 0 & \sigma_{\alpha\alpha} \end{bmatrix},\tag{25}$$

where **R** is the noise covariance of sensor with measurement accuracy  $\sigma_{\rho\rho}$  and  $\sigma_{\alpha\alpha}$  of the sensor.

The detected landmarks are then associated with the previous known landmarks based on Mahalanobis distance ( $\chi_i$ ) that  $\chi_i^2 = \mathbf{v}_i^T \mathbf{S}_i^{-1} \mathbf{v}_i$ . The detected landmark belongs to the previous known landmark if  $\chi_i^2 < \gamma$  where  $\gamma$  is threshold value [16].

Jacobian of measurement model  $\mathbf{H}_k$  is defined as:

$$\mathbf{H}_{k} = \begin{bmatrix} \mathbf{H}_{x} & \mathbf{H}_{y} \end{bmatrix} \begin{bmatrix} \mathbf{F}_{x} \\ \mathbf{F}_{y} \end{bmatrix}, \qquad (26)$$

where

$$\mathbf{H}_{x} = \frac{\partial h}{\partial \mathbf{x}_{v}} = \begin{bmatrix} -\frac{X_{i} - X_{A}}{\rho_{i}} & -\frac{Y_{i} - Y_{A}}{\rho_{i}} & 0\\ \frac{Y_{i} - Y_{A}}{\rho_{i}^{2}} & -\frac{X_{i} - X_{A}}{\rho_{i}^{2}} & -1 \end{bmatrix}, \quad (27)$$

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$$\mathbf{H}_{y} = \frac{\partial h}{\partial \mathbf{y}_{i}} = \begin{bmatrix} \frac{X_{i} - X_{A}}{\rho_{i}} & \frac{Y_{i} - Y_{A}}{\rho_{i}} \\ -\frac{Y_{i} - Y_{A}}{\rho_{i}^{2}} & \frac{X_{i} - X_{A}}{\rho_{i}^{2}} \end{bmatrix}, \quad (28)$$
$$\mathbf{F}_{y} = \begin{bmatrix} \mathbf{I}^{(2) \times (3)} & \mathbf{O}^{(2) \times (2(i-1))} \\ \mathbf{I}^{(2) \times (2)} & \mathbf{O}^{(2) \times (2(n-i+1))} \end{bmatrix},$$

where  $\mathbf{H}_x$  is a Jacobian related with AGV position and  $\mathbf{H}_y$  is a Jacobian related with landmark position.

If new landmarks are detected, the new landmarks are included in the state vector  $\hat{\mathbf{x}}_{k|k}$  as follows:

$$\hat{\mathbf{x}}_{k|k} = \begin{bmatrix} \hat{\mathbf{x}}_{k|k-1} \\ y(\hat{\mathbf{x}}_{v,k|k-1}, \mathbf{z}) \end{bmatrix},$$
(29)

where

$$y(\mathbf{x}_{\nu}, \mathbf{z}) = \begin{bmatrix} X_A + \rho \cos(\alpha + \theta_A) \\ Y_A + \rho \sin(\alpha + \theta_A) \end{bmatrix}.$$
 (30)

The covariance matrix  $\mathbf{P}_{k|k}$  can be obtained as follows:

$$\mathbf{P}_{k|k} = \begin{bmatrix} \mathbf{P}_{k|k-1} & \mathbf{P}_{x(x \sim y_n)}^T \mathbf{Y}_x^T \\ \mathbf{Y}_x \mathbf{P}_{x(x \sim y_n)} & \mathbf{Y}_x \mathbf{P}_{xx} \mathbf{Y}_x^T + \mathbf{Y}_z \mathbf{R} \mathbf{Y}_z^T \end{bmatrix}, \quad (31)$$

where

$$\mathbf{P}_{x(x \sim y_n)} = \begin{bmatrix} \mathbf{P}_{xx} & \mathbf{P}_{xy_1} & \cdots & \mathbf{P}_{xy_n} \end{bmatrix}, \quad (32)$$

$$\mathbf{Y}_{x} = \frac{\partial y}{\partial \mathbf{x}_{v}} = \begin{bmatrix} 1 & 0 & -\rho \sin(\alpha + \theta_{A}) \\ 0 & 1 & \rho \cos(\alpha + \theta_{A}) \end{bmatrix}, \quad (33)$$

$$\mathbf{Y}_{z} = \frac{\partial y}{\partial \mathbf{z}} = \begin{bmatrix} \cos(\alpha + \theta_{A}) & -\rho \sin(\alpha + \theta_{A}) \\ \sin(\alpha + \theta_{A}) & \rho \cos(\alpha + \theta_{A}) \end{bmatrix}.$$
(34)

From the above EKF calculation, the estimated position of the AGV  $\hat{\mathbf{x}}_{v} = \begin{bmatrix} X_{A} & Y_{A} & \theta_{A} \end{bmatrix}$  can be obtained. These values are used as the feedback values for tracking controller in the next section.

## 5. PATH PLANNING

#### 5.1. Grid map representation

In this paper, a grid map technique is used to represent the environment. Fig. 5 illustrates how LMS-151 scans the environment with resolution of  $0.5^{\circ}$ laser signal. The initial value sign to each grids is  $\infty$ . When an obstacle is detected by LMS-151, the obstacle position is calculated by (8). Then, the node of this position is considered as an object in the grid map. The value of the grid with obstacle is 1. The value of empty grid is 0. By incorporating this node in the grid map, it will be considered in path planning or replanning algorithm process.

# 5.2. D\* Lite algorithm

The pseudo code of  $D^*$  Lite algorithm is shown in Fig. 6.  $D^*$  Lite algorithm utilizes a heuristic and a priority queue to perform its search and to order its cost updates.



Fig. 5. Obstacle detection.

## key(s) 01. return $\left[\min(g(s), rhs(s)) + h(s_{start}, s); \min(g(s), rhs(s))\right]$ ;

UpdateState(s) 02. if s was not visited before 03.  $g(s) = \infty$ ; 04. if  $(s \neq s_{onal})$   $rhs(s) = \min_{c' \in Surc} (c) (c(s, s') + g(s'))$ 05. if  $(s \in OPEN)$  remove s from OPEN; 06. if  $(g(s) \neq rhs(s))$  insert s into OPEN with key(s); ComputeShortestPath() 07. while  $\begin{pmatrix} \min_{z \in OPEN} (key(s)) < key(s_{ztart}) \\ OR rhs(s_{ztart}) \neq g(s_{ztart}) \end{pmatrix}$ 

09. if (g(s) > rhs(s))

10. 
$$g(s) = rhs(s)$$

11. for all 
$$s' \in \operatorname{Pred}(s)$$
 UpdateState(s');

12. els

13. 
$$g(s) = \infty;$$

14. for all  $s' \in \operatorname{Pred}(s) \cup \{s\}$  UpdateState(s');

#### Main()

- 15.  $g(s_{start}) = rhs(s_{start}) = \infty; g(s_{goal}) = \infty;$
- 16.  $rhs(s_{poal}) = 0; OPEN = \emptyset;$
- 17. insert sgoal into OPEN with key(sgoal);
- 18. forever
- 19. ComputeShortestPath();
- 20. Wait for changes in edge costs;
- 21. for all directed edges (u, v) with changed edge costs
- 22. Update the edge costs c(u, v);
- UpdateState(u);

Fig. 6. D\* Lite algorithm [15].

It calculates the path from the goal to the start position. In  $D^*$  Lite algorithm, the heuristic value is the distance from the each grid to the goal. So if the state has 4 nodes away



Fig. 7. Illustration of D\* Lite algorithm path planning.

from the goal, the heuristic value is also 4.

The illustration of D\* Lite algorithm is shown in Fig. 7. Fig. 7(a) shows the calculation result of D\* Lite algorithm. Fig. 7(b) shows the number description of the node, whereas Fig. 7(c) and Fig. 7(d) show the first and second calculation steps of D\* Lite algorithm, respectively. According to Koenig and Likhachev [15], D\* Lite algorithm can be two times more efficient than A\* algorithm in replanning a new path when the AGV encountered the obstacle. A\* algorithm has to replan the path from the beginning whereas D\* Lite algorithm does not replan the path as it already has information of the surrounding from the first search.

The output of D\* Lite algorithm is a consecutive coordinate  $\begin{bmatrix} s_1 & s_2 & \cdots & s_j \end{bmatrix}$  and  $s_i = (s_{x,i}, s_{y,i})$  connecting start node  $s_{start} = (X_{start}, Y_{start})$  and goal node  $s_{goal} = (X_{goal}, Y_{goal})$ . The distance between two coordinates depends on the size of each grid in the grid map.

The AGV moves at constant linear velocity. Therefore, the reference linear velocity  $V_r(k)$  is predefined and has constant value. The reference value can be obtained from following calculations:

while 
$$(X_r(k) \neq X_{goal} \text{ and } Y_r(k) \neq Y_{goal})$$
  
 $\theta_r(k+1) = \operatorname{atan2}((s_{y,i+1} - s_{y,i}), (s_{x,i+1} - s_{x,i}))$   
 $X_r(k+1) = X_r(k) + (V_r\Delta t) \cos(\theta_r(k))$   
 $Y_r(k+1) = Y_r(k) + (V_r\Delta t) \sin(\theta_r(k))$   
 $\omega_r(k+1) = (\theta_r(k+1) - \theta_r(k))/\Delta t$   
if  $(X_r(k+1) = s_{x,i+1})$  and  $(Y_r(k+1) = s_{y,i+1})$ 

$$i = i + 1$$

end

end.

where  $\Delta t$  is the sampling time.

# 6. TRACKING CONTROLLER

The purpose of this section is to design a trajectory tracking controller for AGV to track the reference position  $(X_r(t), Y_r(t))$  and reference orientation  $\theta_r(t)$  with reference linear velocity  $V_r(t)$  and angular velocity  $\omega_r(t)$  obtained from D\* Lite algorithm. As shown in Fig. 3, the tracking error vector and its time derivative are defined as follows:

$$e(t) = \begin{bmatrix} e_1 \\ e_2 \\ e_3 \end{bmatrix}$$

$$= \begin{bmatrix} \cos \theta_A & \sin \theta_A & 0 \\ -\sin \theta_A & \cos \theta_A & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} X_r - X_A \\ Y_r - Y_A \\ \theta_r - \theta_A \end{bmatrix}, \quad (36)$$

$$\dot{e}(t) = \begin{bmatrix} \dot{e}_1 \\ \dot{e}_2 \\ \dot{e}_3 \end{bmatrix}$$

$$= \begin{bmatrix} \cos e_3 & 0 \\ \sin e_3 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} V_r \\ \omega_r \end{bmatrix} + \begin{bmatrix} -1 & e_2 \\ 0 & -e_1 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} V_A \\ \omega_A \end{bmatrix}$$

$$(37)$$

To guarantee the stability of the system, Lyapunov function can be chosen as:

$$V_0 = \frac{1}{2}e_1^2 + \frac{1}{2}e_2^2 + \frac{1}{k_2}(1 - \cos e_3) \text{ for } k_2 > 0, \qquad (38)$$

and its derivatives becomes

$$\dot{V}_0 = e_1 \dot{e}_1 + e_2 \dot{e}_2 + \frac{1}{k_2} (\sin e_3) \dot{e}_3$$
  
=  $e_1 (-V_A + V_r \cos e_3) + \frac{1}{k_2} (\sin e_3) (\omega_r - \omega_A + k_2 e_2 V_r).$   
(39)

To achieve  $\dot{V}_0 \leq 0$ , a control law vector **U** is chosen as follows:

$$\mathbf{U} = \begin{bmatrix} V_A \\ \boldsymbol{\omega}_A \end{bmatrix} = \begin{bmatrix} V_r \cos e_3 + k_1 e_1 \\ \boldsymbol{\omega}_r + k_2 V_r e_2 + k_3 \sin e_3 \end{bmatrix}.$$
 (40)

The velocities of left and right wheels can be written from (3) as follows:

$$\begin{bmatrix} \dot{\phi}_L \\ \dot{\phi}_R \end{bmatrix} = \begin{bmatrix} \frac{1}{r} & -\frac{b}{2r} \\ \frac{1}{r} & \frac{b}{2r} \end{bmatrix} \begin{bmatrix} V_A \\ \boldsymbol{\omega}_A \end{bmatrix}.$$
 (41)

The above introduced total control loop for the AGV can be described as shown in Fig. 8.

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Fig. 8. Total control loop.

Table 1. Parameters and initial values.

Parameter	Values	Parameter	Values
r	0.09 m	b	0.3 m
$k_1$	0.7	$X_A(0)$	1.65 m
k <sub>2</sub>	6	$Y_A(0)$	-2 m
<i>k</i> <sub>3</sub>	0.5	$\theta_A(0)$	0 rad
$V_A(0)$	0 m/s	$\sigma_{lpha lpha}$	0.001
$\omega_A(0)$	0 rad/s	$\sigma_{ ho ho}$	0.002
<b>P</b> <sub>0</sub>	<b>O</b> <sup>(3×3)</sup>		



Fig. 9. Experimental environment.

# 7. EXPERIMENTAL RESULTS

The purpose of this paper is to generate a trajectory from start to goal position and to track the generated trajectory in partially known environment. In this section, the unknown environment is represented by obstacles that are not graphed in the given map.

To verify the effectiveness of the proposed algorithm, experiments are carried out. The parameters and initial values for experiment are shown in Table 1.

Fig. 9 shows the environment for the experiment of the proposed system. The environment consists of landmarks, known obstacles and unknown obstacles. Experimental environment in Fig. 9 can be represented in grid map as shown in Fig. 10. Grid size used in this paper is  $40 \times$ 



Fig. 10. Grid map of environment.



Fig. 11. Experimental result.

40 cm.

The positions of the AGV during the experiment are shown in Fig. 11. Fig. 11(1) shows an initial position of the AGV. In this position, since the obstacle in front of AGV is unknown, the AGV generates the trajectory as shown in Fig. 12(a). Fig. 11(2) shows the AGV detects an unknown obstacle and then replans the trajectory as shown in Fig. 12(b). Fig. 11(3-5) show that the AGV tracked the new generated trajectory and finally reaches the goal position of Fig. 11(6).

## 7.1. D\* path planning and obstacle avoidance

The generated path obtained by D\* Lite algorithm is shown in Fig. 12. As shown in Fig. 12(a), since at the start



Fig. 12. D\* Lite path planning.



Fig. 13. Trajectory tracking.

state the robot does not know that there is an unknown obstacle in the surroundings, the robot plans the shortest path to the goal state that passes the unknown obstacle. After the path is built, the robot tracks the generated path until the robot meets the unknown obstacle and replans the path when it meets a n unknown obstacle.

## 7.2. Trajectory tracking

The trajectory tracking result is shown in Fig. 13. In this figure, different positioning method such as encoder, SLAM and NAV are compared. Fig. 13 shows that the controller successfully makes the AGV track the generated path with an acceptable small error. The position obtained by SLAM method is almost similar to absolute position measured by NAV.

The tracking error of the proposed controller is shown in Fig. 14. In this paper, the SLAM positioning method is



Fig. 14. Tracking error  $e_1$ ,  $e_2$ ,  $e_3$  obtained from each sensors.

used as a feedback of trajectory tracking. This figure shows that the tracking errors  $e_1$ ,  $e_2$ ,  $e_3$  obtained from SLAM are approaching to zero.

The error  $e_1$  obtained from SLAM is increased at t = 10 s, t = 20 s, t = 40 s, t = 60 s since the direction of trajectory is changed. The error  $e_2$  obtained from SLAM at t=0s is large since the initial position is different with the initial reference position. However, the controller succesfully reduces the error. Therefore, the error  $e_2$  is approaching zero after 20 s. The error  $e_3$  obtained from SLAM is increased at t = 10 s, t = 20 s, t = 40 s, t = 60 s since the reference direction of the AGV is changed.

## 7.3. Positioning

To verify the effectiveness of SLAM positioning method, the positions obtained from encoders and SLAM are compared with the position obtained from NAV. NAV is a common positioning sensor used in industrial applications. The comparison of three different positioning methods is shown in Table 2 as follows:

The experimental results regarding the difference among SLAM, encoder and NAV are shown in Fig. 15. Fig. 15 shows that the position difference between SLAM and NAV in X and Y axis is less than 20 mm. The angle difference between SLAM and NAV is less than  $10^{\circ}$ . On the other hand, the position difference between en-

Items	NAV	Encoder	SLAM
Sensor	NAV-200	2 encoders	2 encoders and LMS-151
Positioning method	Triangulation method using mapped reflectors positioned around the environment	Based on the accumu- lation of wheel rotation	Combaining encoder positioning with landmark position using EKF
Positioning measure- ment error	Acorrding to datasheet $X = \pm (4 \sim$ 20) mm, $Y = \pm (4 \sim$ 20) mm, $\theta = \pm 0.1^{\circ}$	Compared to NAV $X = \pm 80$ mm, $Y = \pm 200$ mm, $\theta = \pm 20^{\circ}$	Compared to NAV $X = \pm 20$ mm, $Y = \pm 20$ mm, $\theta = \pm 10^{\circ}$

Table 2. Comparison of NAV, encoder and SLAM.



(a) Difference in X axis.



(c) Difference in angle.Fig. 15. Difference between SLAM, encoder and NAV

positioning.

coder positioning and NAV is less than 80 mm in X axis, and more than 200mm in Y axis. Moreover, the angle difference between encoder positioning and NAV is less than 20°. The proposed SLAM algorithm obtains better positioning result compared to encoder positioning.

## 8. CONCLUSION

Positioning and obstacle avoidance of AGV in partially known environment was proposed. SLAM algorithm based on EKF was proposed to obtain the position of AGV. For path planning and obstacle avoidance, D\* algorithm was proposed based on a given map and laser scanner data. To guarantee that the tracking errors approach to zero, a backstepping control method based on Lyapunov stability was proposed. The effectiveness of the proposed algorithms and controller was verified using experiment. The experimental results showed that the AGV successfully reached the goal point when there was an unknown obstacle with an acceptable error.

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