Adaptive Fault Diagnosis and Active Tolerant Control for Wind Energy Conversion System

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Abstract: The fault mathematic model of the transmission part of wind energy conversion system (WECS) is established, and adaptive fault observer is constructed in the presence of unknown disturbance, it can detect the faults of the system, and estimate these faults. Then, based on fault observer, an active tolerant controller is designed to ensure the stability of the transmission part of WECS with fault .The simulation results of different type faults of generator show the effectiveness and feasibility of adaptive fault diagnosis methods.

Keywords: Adaptive observer, fault detection and diagnosis, fault tolerant control, wind energy conversion system.

1. INTRODUCTION

In the last few years, with global warming and energy crisis, the development and utilization of new energy have been explored in many countries. Wind energy is a renewable energy sources, and does not produce harmful substances to the environment in the utilizing. In addition, there are abundant wind energy resources on Earth. Therefore, it has become a new power generation sources in many countries around the world.

During wind energy conversion system operating, the transmission system has high possibility of fault. Transmission system is mainly components of the wind turbine, the transmission equipment and the mechanical part of the generator, whose function is transmitting torque to the generator. Now most research works of wind farm only focus on oil temperature detection for transmission system, it is difficult to meet the requirement of the fault diagnosis of WECS based on model.

Currently, the fault diagnosis of wind power system based on the model has attracted many scholars' attention [1]. Manuel *et al.* completed sensor fault detection and isolation of wind energy conversion system (WECS) by the generalized observer [2]. In [3], Liu proposed a local mean decompose method for wind turbine, through iteration demodulating the amplitude and frequency of fault signal so as to control fault system. Anurat adopted states monitoring to realize fault detection and isolation

for wind energy conversion system [4]. In [5], the method of support vector machine has been used, and the characteristic parameters of various states of wind generator as learning samples. In the SVM training, in order to achieve fault diagnosis of wind energy conversion system, the mapping relations between different fault types and characteristic parameter vector are established. Many intelligent control algorithms have been studied and applied to fault diagnosis of WECS. In [6], BP neural network based on particle swarm optimization algorithm is used to diagnosis the fault of gearbox of WECS, through the particle swarm algorithm to make up for the deficiency of BP neural network and to improve the robustness of system. In [7], fuzzy control is adopted for fault diagnosis, through establishing the fault tree of system to solve the uncertainty problem of pitch control system, the rapidity and accuracy of the control system is improved. In [8], Expert system for the fault diagnosis of the gearbox of wind energy conversion system is proposed. Furthermore, artificial neural networks method for the fault diagnosis of WECS is proposed, it combined various neural networks to diagnosis the short circuit fault of generator [9]. The above methods have their own advantages, however, the problems of unknown disturbance have not considered. Because the running environment is harsh and many unknown disturbances exist [10], the method of fault diagnosis of WECS should have robustness; when there is an external disturbance, the system can run normally.

In this paper, a robust adaptive observer is designed for wind energy conversion system. It can estimate the fault accurately when there are unknown disturbances in the system. So it can make the output error not sensitive to unknown disturbance. Then based on fault observer, an active tolerant controller is designed to ensure the stability of WECS with fault. The simulation results show the effectiveness and feasibility of adaptive fault diagnosis method and active tolerant controller.

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2. FAULT MATHEMATIC MODEL OF WIND **ENERGY CONVERSION SYSTEM**

The mathematic model of wind energy conversion system with variable-speed and fixed-paddle can be written as follows (see [11]):

$$\begin{bmatrix} \dot{\omega}_{\mathrm{r}} \\ \dot{\omega}_{\mathrm{g}} \\ \dot{\theta}_{\Delta} \end{bmatrix} = \begin{bmatrix} -\frac{\left(B_{\mathrm{dt}} + B_{\mathrm{r}}\right)}{J_{\mathrm{r}}} & \frac{B_{\mathrm{dt}}}{N_{\mathrm{g}}J_{\mathrm{r}}} & -\frac{K_{\mathrm{dt}}}{J_{\mathrm{r}}} \\ \frac{\eta_{\mathrm{dt}}B_{\mathrm{dt}}}{N_{\mathrm{g}}J_{\mathrm{g}}} & \frac{-\frac{\eta_{\mathrm{dt}}B_{\mathrm{dt}}}{N_{\mathrm{g}}^{2}} - B_{\mathrm{g}}}{J_{\mathrm{g}}} & \frac{\eta_{\mathrm{dt}}K_{\mathrm{dt}}}{N_{\mathrm{g}}J_{\mathrm{g}}} \\ 1 & -\frac{1}{N_{\mathrm{g}}} & 0 \end{bmatrix} \begin{bmatrix} \omega_{\mathrm{r}} \\ \omega_{\mathrm{g}} \\ \theta_{\Delta} \end{bmatrix} \\ + \begin{bmatrix} \frac{T_{\mathrm{r}}}{J_{\mathrm{r}}} \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} 0 \\ -\frac{T_{\mathrm{g}}}{J_{\mathrm{g}}} \\ 0 \end{bmatrix}, \quad y = \begin{bmatrix} 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} \omega_{\mathrm{r}} \\ \omega_{\mathrm{g}} \\ \theta_{\Delta} \end{bmatrix}, \quad (1)$$

where ω_{γ} is rotor speed of wind turbine, ω_{g} is rotor speed of generator, θ_{Λ} is torsion angle of transmission system, $J_{\rm r}$ and J_{g} is the inertia of the rotor and generator respectively. K_{dt} is torsion stiffness, B_{dt} , B_g and B_r is the inherent damping of torque (torsion), generator and turbine respectively, N_g is the speed increasing ratio of the gearbox. η_{dt} is the efficiency of transmission system, T_r and T_g is the torque of the rotor and generator respectively.

Supposing the generator is constant magnetic flux and the torque is exactly below $T_{g \max}$, non-linear generator torque can be approximate linearized into:

$$T_{\rm g} = B_{\rm g}(\omega_{\rm g} - \omega_{\rm z})$$

where ω_z is rotor speed of zero torque.

In the process of system operation, there will be various uncertain factors such as disturbance and system fault, the fault model of system can be written as follows;

$$\dot{\mathbf{x}}(t) = A\mathbf{x}(t) + B\mathbf{u}(t) + E\mathbf{f}(t) + \mathbf{r}(t) + \eta_1(t)$$

$$y(t) = C\mathbf{x}(t),$$
(2)

where
$$\boldsymbol{B} = \begin{bmatrix} 0 & \frac{B_{g}}{J_{g}} & 0 \end{bmatrix}^{\mathrm{T}}$$
, $\boldsymbol{C} = \begin{bmatrix} 0 & 1 & 0 \end{bmatrix}$,
$$\boldsymbol{A} = \begin{bmatrix} -\frac{(B_{\mathrm{dt}} + B_{\mathrm{r}})}{J_{\mathrm{r}}} & \frac{B_{\mathrm{dt}}}{N_{\mathrm{g}}J_{\mathrm{r}}} & -\frac{K_{\mathrm{dt}}}{N_{\mathrm{g}}} \end{bmatrix}$$
$$\frac{\eta_{\mathrm{dt}}B_{\mathrm{dt}}}{N_{\mathrm{g}}J_{\mathrm{g}}} & \frac{-\frac{\eta_{\mathrm{dt}}B_{\mathrm{dt}}}{N_{\mathrm{g}}^{2}} - 2B_{\mathrm{g}}}{J_{\mathrm{g}}} & \frac{\eta_{\mathrm{dt}}K_{\mathrm{dt}}}{N_{\mathrm{g}}J_{\mathrm{g}}} \end{bmatrix}$$
,

$$\boldsymbol{E} \in \boldsymbol{R}^{3\times3}, \quad \boldsymbol{x}(t) = \begin{bmatrix} \omega_{\mathrm{r}} & \omega_{\mathrm{g}} & \theta_{\Delta} \end{bmatrix}^{T}, \quad \boldsymbol{u}(t) = \omega_{\mathrm{z}};$$
$$\boldsymbol{r}(t) = \begin{bmatrix} \frac{T_{\mathrm{r}}}{J_{\mathrm{r}}} & 0 & 0 \end{bmatrix}^{\mathrm{T}}, \quad \boldsymbol{y} = \omega_{\mathrm{g}}, \quad \boldsymbol{f}(t) \in \boldsymbol{R}^{3}$$

is time-varying unknown fault, and suppose f(t) and its differential is norm bounded $\|\boldsymbol{f}(t)\| \leq f_0$, $\|\dot{\boldsymbol{f}}(t)\| \leq f_1$. Where $f_0 \ge 0$, $f_1 \ge 0$; $\eta_1(t)$ is the modeling error or external disturbance vector, and $\|\boldsymbol{\eta}_1(t)\| \leq g_1$.

3. DESIGN OF ADAPTIVE FAULT OBSERVER

3.1. The model of adaptive fault observer The model of adaptive fault observer is:

$$\begin{cases} \dot{\hat{x}}_{\rm m} = A\hat{x}_{\rm m} + B\boldsymbol{u} + E\hat{\boldsymbol{f}} + \boldsymbol{G}(\hat{y}_{\rm m} - y) + \boldsymbol{r}(t) \\ \hat{y}_{\rm m} = C\hat{x}_{\rm m}, \end{cases}$$
(3)

where $\hat{x}_{m}(t) \in \mathbf{R}^{3}$ is the state vector of observer. $\hat{y}_{m} \in \mathbf{R}$ is output of observer. \hat{f} is the estimate of f. $G \in R^3$ is observer gain matrix to be designed, and satisfies (A+GC) is stable.

When the system without fault and uncertainty factors, Equation (3) can be rewritten as

$$\begin{cases} \dot{\hat{x}} = A\hat{x} + Bu + G(\hat{y} - y) + r(t), \\ \dot{y} = C\hat{x}. \end{cases}$$
(4)

Define $e(t) = \hat{x}(t) - x(t)$, $\varepsilon_0(t) = \hat{y}(t) - y(t)$.

By (2) and (4) the observation error and output error equation are obtained

$$\dot{\boldsymbol{e}}(t) = (\boldsymbol{A} + \boldsymbol{G}\boldsymbol{C})\boldsymbol{e}(t) - \boldsymbol{E}\boldsymbol{f}(t) - \boldsymbol{\eta}_{1}(t),$$

$$\varepsilon_{0}(t) = \boldsymbol{C}\boldsymbol{e}(t).$$
(5)

By (5) the threshold of the system output error is defined

$$\left\|\varepsilon_{0}\right\| \leq \max_{\omega \geq 0} \left\|\boldsymbol{C}(j\omega\boldsymbol{I} - \boldsymbol{A} - \boldsymbol{G}\boldsymbol{C})^{-1}\right\| g_{1} \triangleq \lambda.$$
(6)

Fault detection can be executed as follows:

$$\begin{cases} \|\varepsilon_0\| \le \lambda, \text{ no fault,} \\ \|\varepsilon_0\| > \lambda, \text{ fault.} \end{cases}$$
(7)

3.2. The determine of feedback gain matrix of observer

Assumption 1: Suppose that there exist symmetric positive definite matrix Q, $P = P^T$, satisfying the following matrix inequality

$$(\boldsymbol{A} - \boldsymbol{E}\boldsymbol{C})^{\mathrm{T}}\boldsymbol{P} + \boldsymbol{P}(\boldsymbol{A} - \boldsymbol{E}\boldsymbol{C}) \leq -\boldsymbol{Q} - 2\boldsymbol{C}^{\mathrm{T}}\boldsymbol{C}.$$
 (8)

According to Assumption 1 the feedback gain matrix G can be deduced, and meets (A+GC) being stable. The deriving process as follows:

By (8) gives $(A - EC)^{\mathrm{T}} P + P(A - EC) + 2C^{\mathrm{T}} C \leq -Q$. Transforming the left of the inequality as:

$$(\boldsymbol{A} - \boldsymbol{E}\boldsymbol{C})^{\mathrm{T}}\boldsymbol{P} + \boldsymbol{P}(\boldsymbol{A} - \boldsymbol{E}\boldsymbol{C}) + 2\boldsymbol{C}^{\mathrm{T}}\boldsymbol{C}$$
$$= [\boldsymbol{A} + (\boldsymbol{P}^{-1}\boldsymbol{C}^{\mathrm{T}} - \boldsymbol{E})\boldsymbol{C}]^{\mathrm{T}}\boldsymbol{P}$$

$$+ P[A + (P^{-1}C^{T} - E)C]$$

$$\leq -Q.$$
If $G = P^{-1}C^{T} - E$, yield to
 $(A + GC)^{T}P + P(A + GC) \leq -Q.$
(9)

According to Lyapunov stable theory, if there exists symmetric positive definite matrix P, Q, satisfy (9), then (A+GC) is stable. So the feedback gain matrix of observer is $G = P^{-1}C^{T} - E$.

3.3. The adaptive estimation of fault

For the estimation of fault, the following assumption is needed.

Assumption 2: For any positive scalar c > 0 and positive semidefinite matrix *S*, the following inequality holds [12,13]

$$2\boldsymbol{w}^{\mathrm{T}}\boldsymbol{S}\boldsymbol{v} \leq (1/c)\boldsymbol{w}^{\mathrm{T}}\boldsymbol{S}\boldsymbol{w} + c\boldsymbol{v}^{\mathrm{T}}\boldsymbol{S}\boldsymbol{v}, \qquad (10)$$

where $w \in \mathbf{R}^n$, $v \in \mathbf{R}^n$.

Define $\boldsymbol{e}_{\mathrm{m}} = \hat{\boldsymbol{x}}_{\mathrm{m}} - \boldsymbol{x}, \ \tilde{\boldsymbol{f}} = \hat{\boldsymbol{f}} - \boldsymbol{f}(t).$

By (2) and (3) the dynamic equations of the estimation error and output error are described by

$$\dot{\boldsymbol{e}}_{\mathrm{m}} = (\boldsymbol{A} + \boldsymbol{G}\boldsymbol{C})\boldsymbol{e}_{\mathrm{m}} + \boldsymbol{E}\boldsymbol{f} - \boldsymbol{\eta}_{\mathrm{l}}(t),$$

$$\boldsymbol{\varepsilon} = \hat{\boldsymbol{y}}_{\mathrm{m}} - \boldsymbol{y} = \boldsymbol{C}\boldsymbol{e}_{\mathrm{m}}.$$
(11)

Using σ correction method in adaptive control (see [14]), the adaptive estimation of fault is

$$\frac{d\hat{f}(t)}{dt} = -\mathbf{K}\boldsymbol{\Phi}\varepsilon(t) - \boldsymbol{\sigma}\mathbf{K}\hat{f}(t), \qquad (12)$$

where $\sigma > 0$ is a scalar, $\mathbf{K} \in \mathbf{R}^{3\times 3}$ and $\mathbf{K} = \mathbf{K}^{T} > 0$ $\boldsymbol{\Phi} \in \mathbf{R}^{3}$, the learning rate of (12) is defined.

If the design of the adaptive diagnosis algorithm can ensure the system described by (11) and (12) asymptotically stable, then the system state and fault estimation error is bounded, and the accurate estimation of the state and fault can be got.

Proof: Choosing a possible Lyapunov function

$$V(t) = e_{\rm m}^{\rm T} \boldsymbol{P} e_{\rm m} + \tilde{\boldsymbol{f}}^{\rm T} \boldsymbol{K}^{-1} \tilde{\boldsymbol{f}}.$$
 (13)

Then

$$\dot{V}(t) = \dot{\boldsymbol{e}}_{m}^{T} \boldsymbol{P} \boldsymbol{e}_{m} + \boldsymbol{e}_{m}^{T} \boldsymbol{P} \dot{\boldsymbol{e}}_{m} + \dot{\boldsymbol{f}}^{T} \boldsymbol{K}^{-1} \boldsymbol{\tilde{f}} + \boldsymbol{\tilde{f}}^{T} \boldsymbol{K}^{-1} \dot{\boldsymbol{\tilde{f}}}$$

$$= \boldsymbol{e}_{m}^{T} \Big[(\boldsymbol{A} + \boldsymbol{G} \boldsymbol{C})^{T} \boldsymbol{P} + \boldsymbol{P} (\boldsymbol{A} + \boldsymbol{G} \boldsymbol{C}) \Big] \boldsymbol{e}_{m} + 2\boldsymbol{e}_{m}^{T} \boldsymbol{P} \boldsymbol{E} \boldsymbol{\tilde{f}}$$

$$- 2\boldsymbol{e}_{m}^{T} \boldsymbol{P} \boldsymbol{\eta}_{1} - 2 \boldsymbol{\tilde{f}}^{T} \boldsymbol{\Phi}^{T} \boldsymbol{\varepsilon}(t)$$

$$- 2\boldsymbol{\sigma} \boldsymbol{\tilde{f}}^{T} (\boldsymbol{\tilde{f}} + \boldsymbol{f}) - 2 \boldsymbol{\tilde{f}}^{T} \boldsymbol{K}^{-1} \boldsymbol{\dot{f}}.$$

According to Assumptions 1 and 2, we get

$$\dot{V} \leq -a_1 \left\| \boldsymbol{e}_{\mathrm{m}} \right\|^2 - a_2 \left\| \tilde{\boldsymbol{f}} \right\|^2 + b_1,$$
 (14)

where

$$a_1 = \lambda_{\min}(\boldsymbol{Q}) - h\lambda_{\max}(\boldsymbol{P}) > 0$$

$$a_{2} = \sigma + 2\lambda_{\min}(\boldsymbol{\Phi}\boldsymbol{\Phi}^{\mathrm{T}}) - \lambda_{\min}(\boldsymbol{K}^{-1}) > 0,$$

$$b_{1} = \frac{\lambda_{\max}(\boldsymbol{P})}{h} \|g_{1}\|^{2} + (\sigma f_{0}^{2} + \lambda_{\min}(\boldsymbol{K}^{-1}) f_{1}^{2}) > 0, \ h > 0$$

is a scalar. From (13)

$$V \le \lambda_{\max}(\boldsymbol{P}) \|\boldsymbol{e}_m\|^2 + \lambda_{\max}(\boldsymbol{K}^{-1}) \|\tilde{\boldsymbol{f}}\|^2.$$

Then $\dot{\boldsymbol{V}} \le -\tau \boldsymbol{V} + \boldsymbol{h}$

where
$$\tau = \frac{\min(a_1, a_2)}{\max[\lambda_{\max}(\boldsymbol{P}), \lambda_{\max}(\boldsymbol{K}^{-1})]},$$

 $V \ge \frac{\lambda_{\min}(\boldsymbol{P})}{\|\boldsymbol{C}\|} \|\boldsymbol{\varepsilon}\|^2 + \lambda_{\min}(\boldsymbol{K}^{-1})[\|\hat{\boldsymbol{f}}\|^2 - f_0^2].$
If $(\boldsymbol{\varepsilon}, \hat{\boldsymbol{f}}) \in N, N = \left\{\boldsymbol{\varepsilon}(t), \hat{\boldsymbol{f}}(t) \middle| \frac{\lambda_{\min}(\boldsymbol{P})}{\|\boldsymbol{C}\|} \|\boldsymbol{\varepsilon}\|^2 + \lambda_{\min}(\boldsymbol{K}^{-1}) \|\hat{\boldsymbol{f}}\|^2$
 $> \lambda_{\min}(\boldsymbol{K}^{-1})f_0^2 + \frac{b_1}{\tau} \right\},$ we can get $V \ge \frac{b_1}{\tau}$, and then

 $\dot{V} < 0.$

The error system is asymptotically stable. So the designed of adaptive fault diagnosis algorithm can obtain accurate estimation of the state and fault.

4. DESIGN OF FAULT-TOLERANT CONTROLLER BY STATE FEEDBACK

To design the fault-tolerant controller by state feedback, a useful assumption is given.

Assumption 3: The boundary functions

$$\left\| \boldsymbol{x}^{\mathrm{T}} \boldsymbol{H} \boldsymbol{\eta}(t) \right\| \leq \Gamma(\boldsymbol{x}, t).$$

Function $\Gamma(\mathbf{x},t)/[||\mathbf{x}^{T}\mathbf{H}\mathbf{B}||]$ is continuous and locally bounded on \mathbf{x} , where $\mathbf{H} = \mathbf{H}^{T} > 0$ and satisfy the following Riccati equation

$$\boldsymbol{A}^{\mathrm{T}}\boldsymbol{H} + \boldsymbol{H}\boldsymbol{A} - 2\boldsymbol{H}\boldsymbol{B}\boldsymbol{B}^{\mathrm{T}}\boldsymbol{H} + \boldsymbol{W} = 0, \qquad (15)$$

where $\boldsymbol{W} = \boldsymbol{W}^{\mathrm{T}} > 0$.

According to the state of fault system being estimated by the observer, the fault-tolerant controller of fault system need to be designed to ensure the closed-loop stability. The feedback control law is designed as follows

$$\boldsymbol{u} = \boldsymbol{u}_1 + \boldsymbol{u}_2, \tag{16}$$

where

$$\boldsymbol{u}_{1} = -\boldsymbol{B}^{\mathrm{T}}\boldsymbol{H}\hat{\boldsymbol{x}}_{\mathrm{m}} - \boldsymbol{E}\hat{\boldsymbol{f}}, \qquad (17)$$

$$\boldsymbol{u}_{2} = -\frac{\boldsymbol{\phi}(\boldsymbol{x},t)}{\left\|\boldsymbol{\phi}(\boldsymbol{x},t)\right\| + d/2} \frac{\boldsymbol{\Gamma}(\boldsymbol{x},t)}{\left\|\boldsymbol{x}^{\mathrm{T}}\boldsymbol{H}\boldsymbol{B}\right\|}.$$
 (18)

With $\phi(\mathbf{x},t) = \frac{\mathbf{B}^{\mathrm{T}} \mathbf{H} \mathbf{x}}{\|\mathbf{x}^{\mathrm{T}} \mathbf{H} \mathbf{B}\|} \Gamma(\mathbf{x},t)$, *d* is a positive scalar

and $E = B\overline{E}$.

Substituting control law (16) into transmission system model (2), the following closed-loop state equations of fault system is got

$$\dot{\mathbf{x}} = (\mathbf{A} - \mathbf{B}\mathbf{B}^{\mathrm{T}}\mathbf{H})\mathbf{x} - \mathbf{B}\mathbf{B}^{\mathrm{T}}\mathbf{H}\mathbf{e}_{\mathrm{m}} -\mathbf{E}\tilde{\mathbf{f}} + \mathbf{r}(t) + \mathbf{B}\mathbf{u}_{2} + \boldsymbol{\eta}_{1}(t).$$
(19)

If the design of the feedback control law can ensure the closed-loop system (19) stability, the active fault tolerant controller designed can guarantee the normal running of the fault system.

Proof: Choosing a possible Lyapunov function

$$\boldsymbol{V}_{1} = \boldsymbol{e}_{\mathrm{m}}^{\mathrm{T}} \boldsymbol{P} \boldsymbol{e}_{\mathrm{m}} + \tilde{\boldsymbol{f}}^{\mathrm{T}} \boldsymbol{K}^{-1} \tilde{\boldsymbol{f}} + \beta \boldsymbol{x}^{\mathrm{T}} \boldsymbol{H} \boldsymbol{x},$$

where $\beta > 0$ is a scalar.

$$\dot{V}_{1} = \dot{\boldsymbol{e}}_{m}^{T} \boldsymbol{P} \boldsymbol{e}_{m} + \boldsymbol{e}_{m}^{T} \boldsymbol{P} \dot{\boldsymbol{e}}_{m} + \dot{\boldsymbol{f}}^{T} \boldsymbol{K}^{-1} \boldsymbol{\tilde{f}} + \boldsymbol{\tilde{f}}^{T} \boldsymbol{K}^{-1} \dot{\boldsymbol{\tilde{f}}} + \beta \{ \boldsymbol{x}^{T} [\boldsymbol{H} (\boldsymbol{A} - \boldsymbol{B} \boldsymbol{B}^{T} \boldsymbol{H}) + (\boldsymbol{A} - \boldsymbol{B} \boldsymbol{B}^{T} \boldsymbol{H})^{T} \boldsymbol{H}] \boldsymbol{x} - 2 \boldsymbol{x}^{T} \boldsymbol{H} \boldsymbol{B} \boldsymbol{B}^{T} \boldsymbol{H} \boldsymbol{e}_{m} + 2 \boldsymbol{x}^{T} \boldsymbol{H} [\boldsymbol{r}(t) + \boldsymbol{B} \boldsymbol{u}_{2} + \boldsymbol{\eta}_{1}(t)] - 2 \boldsymbol{x}^{T} \boldsymbol{H} \boldsymbol{E} \boldsymbol{\tilde{f}} \}.$$

By substitution of (18) yields

$$2\beta \mathbf{x}^{\mathrm{T}} \boldsymbol{H}[\boldsymbol{r}(t) + \boldsymbol{B}\boldsymbol{u}_{2} + \boldsymbol{\eta}_{1}(t)]$$

$$\leq 2\beta \left[\|\boldsymbol{\phi}(\boldsymbol{x}, t)\| - \frac{\|\boldsymbol{\phi}(\boldsymbol{x}, t)\|^{2}}{\|\boldsymbol{\phi}(\boldsymbol{x}, t)\| + d/2} \right] \leq \beta d.$$

By using the condition (10) and (15), we get $V_1 \leq -m_1 ||\mathbf{x}||^2 - m_2 ||\mathbf{e}_m||^2 - m_3 ||\tilde{f}||^2 + \beta d$, we can get the appropriate *d* to meet

$$\dot{V}_{1} \leq -m_{1} \left\| \boldsymbol{x} \right\|^{2} - m_{2} \left\| \boldsymbol{e}_{m} \right\|^{2} - m_{3} \left\| \tilde{\boldsymbol{f}} \right\|^{2} + \beta d \leq 0, \quad (20)$$

where

$$m_{1} = \lambda_{\min}(S_{1}),$$

$$S_{1} = \beta W - \beta H \left(BB^{T} / c_{1} + EE^{T} / c_{2} \right) H,$$

$$m_{2} = \lambda_{\min}(S_{2}), S_{2} = Q - \beta c_{1} HBB^{T} H, m_{3} = 2 - \beta c_{2}.$$

The parameters c_1 , c_2 and β are chosen to satisfy the following inequalities

$$S_1 > 0, S_2 > 0, m_3 > 0.$$

According to (20), it is known that the feedback control law (16) can make the closed-loop fault system stable, and ensures the normal running of system when fault occurred.

5. SIMULATION STUDIES

In the simulation, the parameters are as follows:

$$B_{\rm dt} = 9.45 \times 10^5$$
, $B_{\gamma} = 2.78 \times 10^4$, $B_{\rm g} = 3.034$,
 $N_{\rm g} = 95$, $K_{\rm dt} = 2.7 \times 10^9$, $\eta_{\rm dt} = 0.92$, $J_{\rm g} = 390$,

$$J_{\gamma} = 55 \times 10^6, \ T_{\rm r} = 15.84 \times 10^7, \ \boldsymbol{\eta}_{\rm l}(t) = \begin{bmatrix} 0 \\ rand(0.0015) \\ 0 \end{bmatrix}.$$

Then

$$\boldsymbol{A} = \begin{bmatrix} -1.7687 \times 10^{-2} & 1.8096 \times 10^{-4} & -49.091 \\ 23.466 & -0.2626 & 6.7045 \times 10^{4} \\ 1 & -0.0105 & 0 \end{bmatrix},$$
$$\boldsymbol{B} = \begin{bmatrix} 0 \\ 0.0078 \\ 0 \end{bmatrix}.$$

In simulation, only generator fault is considered.

$$\boldsymbol{E} = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0.0078 & 0 \\ 0 & 0 & 0 \end{bmatrix}, \quad \boldsymbol{\Phi} = \begin{bmatrix} 1 & 1 & 1 \end{bmatrix}^{\mathrm{T}},$$

we get

$$P = \begin{bmatrix} 534.85 & 4.0122 \times 10^{-5} & 9.1878 \times 10^{-5} \\ 4.0122 \times 10^{-5} & 0.39002 & 6.3807 \times 10^{-6} \\ 9.1878 \times 10^{-5} & 6.3807 \times 10^{-6} & 26256 \end{bmatrix},$$
$$G = \begin{bmatrix} 0.0184 \\ -3.4 \\ 0.0057 \end{bmatrix},$$
$$H = \begin{bmatrix} 1.7245 \times 10^8 & -1.2159 \times 10^6 & -1.3005 \times 10^8 \\ -1.2159 \times 10^6 & 32734 & -6.2087 \times 10^6 \\ -1.3005 \times 10^8 & -6.2087 \times 10^6 & 1.0244 \times 10^{10} \end{bmatrix}.$$

The fault is as follows $f(t) = \begin{bmatrix} 0 & f_i(t) & 0 \end{bmatrix}^T$, i = 1, 2 where fault f_1 is supposed the actuator fault of the main transmission chain, and result in constant biases. The fault can be described as follows:

F1:
$$f_1(t) = \begin{cases} 0, & 0 < t < 20s \\ 80, & t \ge 20. \end{cases}$$

Fault f_2 is supposed the shaft fault of gear box, and result in double frequency vibrations, the vibrations frequency of shaft is 10Hz. Then, the fault can be described as follows:

F2:
$$f_2(t) = \begin{cases} 0, & 0 < t < 20s \\ 160 + 80\sin(125.6t), & t \ge 20s. \end{cases}$$

The simulation results are shown in Figs. 1~6:

Figs. 1 and 2 show the output error when fault $f_1(t)$ and $f_2(t)$ occurred, respectively. The system output error is quickly beyond the threshold when fault accrued; the simulation results show that the observer designed can detect the system fault accurately and rapidly.

Figs. 3 and 4 show the estimation of fault $f_1(t)$ and $f_2(t)$ by the observer designed. The simulation results show that the fault observer designed can estimate the fault accurately, whether it is a constant fault or time-varying fault.



Fig. 1. The detection of fault f_1 .



Fig. 2. The detection of fault f_2 .



Fig. 3. The fault f_1 and its estimate \hat{f}_1 .



Fig. 4. The fault f_2 and its estimate f_2 .



Fig. 5. The fault-tolerant controller output of constant fault occurred.



Fig. 6. The fault-tolerant controller output of time - varying fault occurred.

Figs. 5 and 6 show the system output after the active fault-tolerant controller is used when constant fault and time-varying fault occurred, respectively. When fault happened, the results show that fault-tolerant controller can make the system output wave in a smaller range. It is show that the control method has very strong robustness, no matter constant fault or time-varying fault.

In [15], the benchmark model is adopted, which includes hydraulic pitch model, converter model and drive train model, so it is comprehensive. Three kinds of fault have been considered, sensor fault (fixed value or gain factor), actuator fault (offset) and parameter abrupt. Only the parameter abrupt can be estimated by least-square method which needs last N samples of input and output. Sensor and actuator fault are got by measurement or the deference between measured and expected value, so it needs many sensors. The tolerant control corresponding needs auxiliary controllers (it needs to be designed in advance for redundant).

In this paper, only drive train model is considered, and is used to test the effectiveness of control scheme. The fault is estimated by an adaptive mechanism which depends on the model, it can approximate the fault gradually, not need least-square method which needs many samples of input and output, and the number of sensor can be reduced effectively. The fault can be estimated adaptively, auxiliary controllers are not needed. It is simple and suitable for appliance.

6. CONCLUSION

In this paper, a new adaptive fault observer is designed for the transmission part of the wind energy conversion system, in the case of unknown input and disturbance. It can accurately detect and estimate the fault, no matter constant fault or time-varying fault, respectively. Then a state feedback fault-tolerant controller is designed, it can maintain system normal running when a fault occurred. The simulation results show the feasibility and effectiveness of the adaptive fault observer and faulttolerant controller for wind energy conversion system.

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