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Optimization of hot gas pressure forming process for titanium alloy component

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Abstract

In this paper, hot gas pressure forming (HGPF) of titanium alloy irregularly profled tubular component using laser-welded tube was studied by both simulation and experiment. Uniaxial tensile tests of base metal (BM) under diferent conditions were performed to determine the true stress–strain curves. The forming process was optimized by fnite element simulation and response surface method (RSM). Results show that the forming pressure increases with the decreasing temperature and increasing strain rate. Microstructures of BM are sensitive of forming temperature, strain and strain rate. Wrinkling and local thinning of the component can be avoided by a reasonable initial tube diameter during the forming. Ideal weld position should be determined to avoid the failure of the weld seam (WS). A qualifed TC2 titanium alloy component with both high dimensional accuracy and good post-form properties was successfully formed by HGPF using the optimized forming parameters. The total heating and forming time of the tube was less than 30 min. Both of the post-form properties and microstructures of the component were almost the same with the initial material.

Keywords Titanium alloy · Microstructure and properties · Response surface method · Finite element analysis · Hot gas pressure forming

Introduction

Thin-walled tubular components with irregular profiles made of titanium alloys not only have excellent properties, but also demonstrate the structure advantage of saving materials and reducing weight. However, titanium alloy has high deformation resistance, poor plasticity and large springback during room temperature forming [\[1](#page-11-0), [2\]](#page-11-1). Therefore, hot forming processes have been developed: such as hot pressing (HP) [[3\]](#page-11-2) and superplastic forming (SPF) [[4,](#page-11-3) [5\]](#page-11-4). The conventional HP and SPF processes have shortcomings of low efficiency, high cost and deteriorating properties $[3, 6]$ $[3, 6]$ $[3, 6]$ $[3, 6]$. It is also very difficult to form thin-walled tubular components by conventional HP and SPF processes.

To improve the manufacturing efficiency and ensure the post-form properties, the HGPF process for integral forming of titanium alloy component was developed based on tube hydroforming process [[7](#page-11-6)], which has a lower forming temperature but a higher forming speed than SPF. However, the forming condition of HGPF also brings some new challenges for titanium alloys because of the relatively high strength. Reasonable process parameters such as pressure value and forming temperature are the key to fabricate complexshaped titanium alloys components by HGPF. At present, the researches mainly focus on the efect of process parameters on the thinning rate, defect control $[8, 9]$ $[8, 9]$ $[8, 9]$ $[8, 9]$ $[8, 9]$ and forming limit $[10, 11]$ $[10, 11]$ $[10, 11]$. The pressure loading path affects the corner radius of varying diameter tube. The corner radii change linearly over time at the pressurization stage but exponentially over time at the constant pressure stage [\[12\]](#page-12-4). The average thinning ratio of the bulging area could be reduced signifcantly and the thickness distribution of the workpiece was more uniform with increasing of the axial feeding [[13](#page-12-5), [14](#page-12-6)]. In

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addition, Liu et al. [[15](#page-12-7)] studied the non-isothermal HGPF forming of TA18 titanium alloy parts with large section difference. Paul et al. [\[16\]](#page-12-8) performed a forming test of grade 2 titanium exhaust component, and the dimensional deviations between the formed part and the CAD reference range from -0.07 mm to 0.11 mm. At present, the method used to determine the process window are calculating the minimum bulging pressure and simulation in the HGPF [[7,](#page-11-6) [17\]](#page-12-9), which requires a lot of forming experiments [\[9](#page-12-1), [18](#page-12-10)]. However, the process window determined by this method may be not the best one. Therefore, it is of great signifcance to consider the infuence of process parameters on dimensional accuracy and thickness uniformity when determining the process window of HGPF. Response surface method (RSM) originally introduced by Box and Wilson [[19](#page-12-11)], which was widely used in metal forming to predict the output response variables such as binder force, fllet radius of punch, friction coefficient and so on $[20, 21]$ $[20, 21]$ $[20, 21]$ $[20, 21]$. However, the optimization of the process window of HGPF by RSM method has seldom been reported. What's more, the titanium alloys reported in the study of HGPF include Ti-3Al-2.5 V alloy [\[5](#page-11-4)], Ti-22Al-24Nb-0.5Mo alloy [[22](#page-12-14)], Ti-55 alloy [\[23\]](#page-12-15), TA15 titanium alloy [\[24](#page-12-16)] and CP-Ti alloy [[25\]](#page-12-17). However, the HGPF of TC2 titanium alloy has seldom been reported. Therefore, it is necessary to develop an optimization method to determine the best process window for HGPF of TC2 titanium alloy component.

In this paper, a new method for determining the process window for HGPF is proposed. Diferent from traditional methods, this method could optimize process parameters by considering dimensional accuracy and thickness uniformity simultaneously. The uniaxial tensile behavior and microstructure evolution of TC2 titanium alloy at high temperature under diferent conditions were studied. Then, the forming process was optimized by fnite element analysis and RSM. In addition, the selection of weld position was also analyzed. Finally, a TC2 titanium alloy tubular component with irregular profles was formed to verify the process window. Combining with the characterization of properties and microstructure evolution, the best process window for HGPF of titanium alloys could be obtained to ensure dimensional accuracy, thickness uniformity and post-form properties of the formed part efficiently.

Materials and methods

Materials

The as-received material was a TC2 titanium alloy (Ti-4Al-1.5Mn) sheet and the average thickness is 1.8 mm. Figure [1a](#page-1-0) shows the XRD pattern of the initial sheet. As can be seen from the peak value, the original sheet is composed of α and β phases. The initial microstructure is shown in Fig. [1b.](#page-1-0) The dark phase is the β phase, and the bright phase is the α phase. The β phase are mainly distributed in the grain boundaries, and minority are in α grains. Both of the average size and volume fraction of the β phase are much smaller than those of the α phase. The TC2 titanium alloy sheet was cut according to the designed size, and then it was rolled into a tube at 650 °C, next welded by LBW. The laser power and the welding speed were 1.2 kW and 1.2 m/min, respectively.

Figure [2a](#page-2-0) is the microstructure of TC2 titanium alloy after laser welding. There are three regions: the base metal (BM), the fusion zone (FZ) and heat-afected zone (HAZ). Figure [2b](#page-2-0) shows the microstructure in FZ, which consists of coarse columnar grains with fine α' martensitic structure in the matrix. The microstructure of HAZ contains a

Fig. 1 XRD pattern and microstructure of the initial sheet. (**a**) XRD patterns of the TC2 alloy; (**b**) initial microstructure of the TC2 alloy sheet

Fig. 2 Weld microstructure and microhardness. (**a**) Microstructure of LBW joint; (**b**) Microstructure of the FZ; (**c**) Microstructure of the HAZ; (**d**) Distribution of Vickers microhardness

mixture of α', primary α phase, $β$ phase and transformed $β$ phase, as shown in Fig. [2c](#page-2-0). A Vickers microhardness test machine was used to measure the hardness with a load of 0.5 kg and duration of 15 s. The microhardness is shown in Fig. [2d.](#page-2-0) It can be seen that the hardness values in the weld center zone are 328.5~339.7 HV. The hardness values in the HAZ and BM are 291.1 ~328.5 HV and 272.3 ~291.1 HV, respectively. The strength of the weld is higher than that of the base metal.

Hot uniaxial tensile and microstructure tests

Uniaxial tensile tests under diferent conditions were implemented to study the relationship between fow stress and strain of TC2 titanium alloy. The temperature range is from 750 to 900 °C, and strain rate range is from 0.001 to 0.1 s^{-1} . The gauge length and the width of the sample are 15 mm and 5 mm, respectively. All the tests were repeated three times. The microstructures were investigated by a Zeiss Supra55 scanning electron microscope (SEM) and optical microscope (OM). The samples were prepared by electro-polishing with a solution of 6% per-chloric acid, 34% butanol, and 60%

methanol (vol $\%$) at -40 °C with a potential of 25 V and current of 0.7 A with the time of 52 s. The polished samples were then etched using a mixture of 13% nitric acid, 7% hydrofuoric acid and 80% water (vol %) with the time of 10 s.

Finite element model of HGPF

A TC2 alloy tubular component with irregular profles was formed. Figure [3](#page-2-1) is the detailed dimensions of the component. The total length of component is 555.5 mm. The section length and width are about 140 mm and 52 mm, respectively. The local fllet radius is 7 mm.

The formability of the welded TC2 alloy tube is afected by process control, such as forming temperature, strain and pressurizing rate. However, the thickness accuracy of the formed part was mainly controlled by the initial tube diameter and weld position. To control the local thinning, wrinkling and fashing defects, the commercial software Abaqus was used for numerical simulation. The tube used in this paper is 555.5 mm in length, and there is no axial feeding in the HGPF process. Therefore, the tube was assumed as a plane strain state, and a 2D plane strain model as shown

Fig. 3 The geometry of TC2 alloy component (unit: mm)

in Fig. [4](#page-3-0) was used in the simulation. By choosing a reasonable weld position, large deformation of the weld could be avoided. In addition, the weld area occupies a very small proportion compared with the base material area. Therefore, the weld was not modeled separately during the simulation. The selection of weld position was discussed in the later section. The parameters in fnite element analysis are presented in Table [1](#page-3-1), such as analysis steps, mesh size and boundary conditions. The Mises yield criterion was adopted in this paper. True stress–strain curves of the TC2 titanium alloy obtained by hot uniaxial tensile tests were used in the simulation to describe the material fow behavior.

Hot gas pressure forming

The forming equipment and related systems are shown in Fig. [5.](#page-4-0) HGPF systems include gas pressurization system, gas pressure control system, water cooling system, heating and

Table 1 Setup and paramet in finite element model

ples were used to detect the temperature of the tube. The tube was heated to the target temperature after holding for 10 min, and the temperature diference of the three thermocouples was less than 5 ℃, which proved the uniformity of the tube temperature. The punches were fed to the target positions to seal the tube, and then the argon was pumped into the tube according to the specifc loading route. The component was taken out of the dies and air cooled to the room temperature.

temperature monitoring system. In addition, it also includes forming dies, seal punches, thermocouples, gas inlet and induction coils. During the forming, the forming dies were

Results and discussion

Hot uniaxial tensile test and microstructure evolution

Hot uniaxial tensile tests under diferent conditions were performed to obtain the material properties. The true stress–strain curves of BM are shown in Fig. [6a-c.](#page-4-1) Since the maximum true strain of the forming part is less than 0.3, the hot uniaxial tensile test was stopped as the true strain reached 0.6 of BM. The stress increased with the decrease of temperature and the increase of strain rate. At low temperature and high strain rate, the hardening behavior is mainly caused by the increase of strain and strain rate, and the main reason of softening is necking behavior. At high temperature and low strain rate, the grain growth leads to hardening. The main reason of softening are dynamic recovery, dynamic recrystallization and phase transformation. When **Fig.** 4 Finite element model of HGPF than 750 °C, the elongation of

Fig. 5 Apparatuses for HGPF of TC2 alloy, where 1 is forming die, 2 is left punch, 3 is right punch, 4 is thermocouple, 5 is the induction coil, 6 is gas inlet, and 7 is heating equipment

Fig. 6 True stress–strain curves of BM. (**a**) 0.1 s^{-1} ; (**b**) 0.01 s^{-1} ; (**c**) 0.001 s^{-1}

BM is higher than 0.6 , which has a sufficient elongation for HGPF. During the process of HGPF, the gas pressure required to form the minimum corner of components is the largest, which is calculated according to the formula in the reference [\[17\]](#page-12-9). At 800 °C/0.001 s⁻¹, the forming pressure is 15 MPa. As the 800 $^{\circ}$ C/0.01 s⁻¹, the forming pressure is higher than 30 MPa. Efficiency can be improved via a high strain rate, but the requirements for equipment also increase.

Figure [7](#page-5-0) is the microstructures of TC2 titanium alloy after hot uniaxial tensile tests. When the strain rate is 0.01 s^{-1} , the grain size increased frst and then decreased with the temperature from 800 to 900 °C as shown in Fig. $7a$ ~c. This is because dynamic recrystallization (DRX) occurred. When the temperature is $850 \degree C$, the grain size decreased with strain rate from 0.001 to 0.1 s⁻¹ as shown in Fig. [7d~f](#page-5-0). The grain size growth due to prolonged high temperature. Voids appeared at 850 °C/0.1 s⁻¹. The secondary α phase precipitated at 900 °C, which leads to high strength but reduced plasticity. The grain size decreased with the increasing strain

due to the occurrence of DRX as shown in Fig. $7g$ ~i. It can be seen that DRX and phase transformation are the two main factors altering the microstructure evolution. In order to maintain the initial microstructure of the material after forming, the forming temperature should be controlled at 800-850 °C.

Determination of the initial tube diameter

To determine the reasonable initial tube diameter, the fnite element analysis was carried out for the initial tube with expansion ratio of 0%, 3%, 4%, 5% and 6%, respectively. When the expansion ratio of the tube is 0%, as shown in Fig. [8a.](#page-6-0) The tube collapsed seriously during die closing, and there are dead wrinkles in the HGPF because the material has no fow space. A similar risk exists for tube with a 3% expansion ratio as shown in Fig. [8b](#page-6-0). Therefore, in the die closing process, a tube with expansion ratio of $4 \sim 6\%$ is recommended. To further determine the expansion ratio of

Temperature increasing from 800 \degree C to 900 \degree C (strain rate 0.01 s⁻¹)

Fig. 7 SEM of the BM samples after tensile deformation, diferent temperature (**a**)-(**c**), diferent strain rate (**d**)-(**f**), diferent true strain (**g**)-(**i**)

the initial tube, the prediction of the thickness of the component with diferent expansion ratios will be discussed later.

The thinning rate increased with the increase of expansion ratio. The thickness distribution of diferent expansion ratios is shown in Fig. [9.](#page-6-1) The measuring points of thickness are shown in Fig. [9a.](#page-6-1) The 28 points were evenly selected along the perimeter. The minimum thickness of the component with diferent expansion rates is shown in Fig. [9b.](#page-6-1) Although the thinning rate of the tube with an expansion ratio of 3% is the minimum, it is prone to serious defects of wrinkling and fashing. The thickness is reduced seriously with an expansion ratio of 6%. The maximum thinning rate reached 16.11%. Therefore, it is recommended to use an initial tube with an expansion ratio between 4 and 5%. When the expansion ratio was 5%, the minimum thickness was 1.562 mm and the maximum thinning rate was

10um

13.22%, which has met demand. Therefore, initial tube with an expansion ratio of 5% was used in this paper. The experimental result of the component with 5% expansion ratio was provided in Fig. [9c.](#page-6-1) In the preforming process (die closing), thanks to the small increase in perimeter, the thickness was distributed uniformly. In the HGPF process, the two corners were thinned due to large expansion according to Fig. [8d.](#page-6-0)

10um

Optimization of HGPF process window

To further obtain the reasonable forming process window, including forming temperature and gas pressure, optimization was performed by the combination of fnite element analysis and RSM. During the optimization, the temperature ranged from 800 to 900 ℃ and the gas pressure ranged from 5 to 20 MPa. The tube was soaked for 10 min in the

Fig. 8 Simulation results of tube with diferent expansion ratio. (**a**) 0%; (**b**) 3%; (**c**) 4%; (**d**) 5%; (**e**) 6%

Fig. 9 Efect of expansion ratio on thickness distribution. (**a**) measuring points; (**b**) minimum thickness with diferent expansion ratio; (**c**) thickness distribution

heated dies before forming to stabilize the temperature, and the temperature was also constant during the forming. Therefore, the temperature was treated as a constant during the RSM. The forming process included four stages such as die closing, pressurization, forming and calibration (Fig. [13](#page-9-0)). During the calibration stage, the pressure value was the maximal one, which was also the dominant factor affecting the dimensional accuracy. Therefore, only the maximum gas pressure value in the calibration stage was optimized, and it was also treated as a constant during each simulation. To express the thickness distribution of the component more intuitively, the inhomogeneity of thickness Y is defined, which can be calculated by Eq. (1) (1) (1) :

$$
Y = 100\% \times \left(T_{max} - T_{min}\right) / T_0 \tag{1}
$$

where T_{max} and T_{min} are the maximum and minimum thickness, respectively. T_0 is the thickness of initial sheet. A parameter related to dimensional accuracy *Dist* is defned, which is equal to the average distance between the corresponding points of the source point cloud and the target point cloud. The flow chart of process window optimization is shown in Fig. [10.](#page-7-0)

The Central Composite module in response surface analysis software was used to optimize the process parameters. The forming quality evaluation index *Z* value is defned, which is the function of inhomogeneity of thickness *Y* and dimensional accuracy deviation *Dist*. The *Z* value is an evaluation index of the degree of deviation, and the smaller the value of *Z*, the better the quality of the formed component. The calculation formula is shown in Eq. ([2](#page-7-1)).

$$
Z = 0.8 \times Dist + 0.2 \times Y/100
$$
 (2)

The *Z* values were calculated via Design Expert 10.0 commercial software, and the response model after data analysis was shown in Eq. (3) (3) .

$$
Z = 1.316^{-5}T^2 + 1.379^{-4}P^2 + 1.712^{-5}TP - 0.023T - 0.021P + 10.223
$$
 (3)

where *T* and *P* are the forming temperature and gas pressure, respectively. Through regression analysis of the equation, the prediction result of response surface analysis is shown in Fig. [11](#page-8-0). The Z value decreased frst and then increased with the increasing temperature. This is because when the forming temperature was relatively low, the corner area of the component could not be fully formed due to the requirement of very high forming pressure. However, when the temperature was relatively high, the strength of forming dies decreased and the deformation of the forming dies increased accordingly, which may lead to the increase of the dimensional accuracy deviation. The Z value decreased with the

Fig. 11 The prediction chart by response surface analysis

increasing pressure. But it is more sensitive of forming temperature than pressure. The Z value decreased slightly from 0.3781 to 0.3712 with pressure increasing from 5 to 20 MPa at temperature of 850 ℃. The predicted optimal forming temperature and gas pressure were 853 ℃ and 13.6 MPa respectively. However, the length of the formed component was more than 550 mm, which needs large volume high pressure gas. Therefore, a gas pressure of 6 MPa was choose to perform the actual forming. The dimensional accuracy deviation of the component formed at 850 ℃ with gas pressure of 6 MPa was predicted to be 0.42 mm. This value is slightly higher than the actual one. The possible reason is that stress relaxation of titanium alloys at high temperatures reduces spring back. However, it is not considered in the current optimization process.

Determination of weld position during HGPF

The selection of weld position has a significant effect on the HGPF and the improper selection of weld position leads to weld cracks in the deformation process. To determine a reasonable weld position, a simulation of the 2D model with the tube of 5% expansion ratio was used. The equivalent stress in specifc positions were calculated as shown in Fig. [12.](#page-8-1) A rectangular coordinate system (blue dotted line) is established in the horizontal and vertical directions, and the origin of the coordinate is placed in the center of the section (Fig. [12\)](#page-8-1). In position I, the equivalent stress is the largest in the initial stage of the die closing, which may cause the weld cracking. In position II, the stress state of the weld is biaxial tension, and it is easy to cause the crack of the weld during the late HGPF. The weld position in the concave area (the

Fig. 12 Characteristic points and equivalent stress of characteristic points. (**a**) the selection region of characteristic points; (**b**) equivalent stress

green rectangular area in Fig. [12a](#page-8-1) also has the risk of cracking. In position III ($7 \sim 28^\circ$), the weld is contacted with the die during most of the forming time. The equivalent stress is the smallest in three areas. Therefore, the weld in position III is the reasonable position. As reported in the reference [[26\]](#page-12-18), the uniform bulging ratio of the laser-welded titanium alloy tube was close to 43.8% considering shape deviation and uniformity of wall thickness. The maximum strain of the component in this paper is 3% in the weld area, which is much smaller than 43.8%. Therefore, the effect of weld seam on the deformation was neglected.

HGPF of the TC2 alloy irregular profle component

A TC2 component was formed at 850 °C based on the above optimization results, and the specifc loading route is shown in Fig. [13](#page-9-0). The geometries at different stages are given in the figure.

When the die temperature was elevated to 850 °C, the tube was placed into the die, and then the die was closed for preforming. In the forming process, the gas pressure was increased from 0 to 3 MPa firstly and hold for 15 min, most of the tube area was formed to the ideal position. To improve the forming accuracy of the corner area, the gas pressure was raised from 3 to 6 MPa and hold for 10 min.

Fig. 13 Specifc loading path of the HGPF

Figure [14](#page-9-1) shows the TC2 thin-walled irregular profiled component, which was fully formed.

Evaluation of Formed component

To measure the microstructures and properties of the fnal formed component, the positions of the selected samples are shown in Fig. [15a](#page-10-0). Sample (b) and sample (c) are BM and sample (d) is WS. The equivalent strain of the sample

Fig. 14 The fnal formed TC2 alloy component by HGPF

(b) was 0.15 and the sample (c) was 0.26. Compared with the initial sheet microstructure in Fig. 0.1 (b), the microstructures of sample (b) and sample (c) varied slightly. Few dynamic recrystallized grains were observed in sample (c). The microstructure of the WS changed a little as shown in Fig. [15d](#page-10-0), which was the martensitic decomposing into lath α and β phase due to its unstable at elevated temperature. However, the strain of the WS was relatively small during the forming, so few globalizations was observed after the forming. It could be summarized that the overall microstructures of the fnal formed component were similar with the initial sheet because of the short forming time during the HGPF.

The room (25 °C) and service temperature (450 °C) properties of the fnal formed component were evaluated by uniaxial tensile tests as shown in Fig. [16](#page-10-1). The properties of BM are almost the same with the initial sheet at both temperatures, but the strength of the weld decreased by 7.05% at 25 °C and 9.54% at 450 °C due to the inevitable decomposition of martensitic at high temperatures. At 450 °C, the average yield stress and tensile stress of BM are 363.1 and 528.6 MPa, respectively. Regarding to the WS, the average yield stress and tensile stress are 423.8 and 538.0 MPa, respectively. The above studies show that the microstructures and properties of the component are basically consistent with those of the initial sheet.

The stress of titanium alloy and steel is dropped signifcantly after stress relaxation for a certain time at a high temperature [[27,](#page-12-19) [28\]](#page-12-20). Therefore, the dimensional deviations are less than calculated value obtained above. A 3D Power Scan tester was used to obtain the outer surface of the fnal formed part. The imaging data was compared with the ideal component by a commercial software of Geomagic Qualify. The dimensional deviation distributions of three specifc crosssections are shown in Fig. [17](#page-11-7). The maximum dimensional deviation was about 0.19 mm, indicating the high precision of the part formed by HGPF.

The basic design of HGPF process should consider the forming windows, dimensional accuracy and post-form mechanical properties. According to the hot uniaxial tension tests and microstructure evolutions, the ranges of forming temperature and pressure could be obtained for HGPF. Then the fnite element analysis is needed to further optimize the forming process by revealing the effects of forming parameters on the forming defects such as wrinkles, cracks and local serious thinning. The WS has a higher strength than the BM, which could cause non-uniform deformation by localizing deformation in the BM zone. Moreover, the WS has a smaller elongation than the BM, which could cause early fracture failure in the WS during the forming. One should carefully compensate the two possible consequences to design the forming process.

Factors affecting the dimensional accuracy includes the dimension of the forming tools and spring back. The

Fig. 16 Mechanical properties of the TC2 alloy before and after HGPF. (**a**) 25 °C and (**b**) 450 °C

materials' thermal expansion coefficients for titanium alloy and forming tools are diferent, therefore the dimension of the forming tools should be calculated carefully according to the thermal expansion coefficients $[29]$ $[29]$ $[29]$. The accurate value of thermal expansion coefficient is very important. The spring back could be predicted according to young's modulus and stress relaxation curves. Generally, the spring back could be reduced efectively by increasing the forming temperature. The post-form mechanical properties are mainly controlled by the initial microstructure and deformation parameters including temperature, strain rate, strain and holding time, which should be designed according to the service requirement.

In this paper, both of the microstructure and properties of the BM are close to that of the initial material before and after HGPF. Because of the relatively high forming temperature and long cycle time during the conventional superplastic forming, the post-form properties could decrease compared with the initial material [\[30](#page-12-22)]. However, the total heating and forming time of the tube was less than 30 min during the HGPF, therefore almost no strength reduction was observed. Regarding to the dimensional

accuracy, components formed by both processes could satisfy the requirements of the service. Therefore, the HGPF process demonstrates the advantages of high efficiency, low energy consumption and excellent mechanical properties compared with the conventional superplastic forming.

Conclusions

In this paper, HGPF process window of a laser-welded TC2 titanium alloy component with irregular profles was optimized. Hot uniaxial tensile tests, microstructure characterization, fnite element simulation and HGPF experiments were studied. The main conclusions are as follows:

- 1. The microstructure evolution of TC2 titanium alloy is mainly DRX and phase transformation. During the hot deformation of BM with temperature ranging from 750 to 850 °C, DRX is the main factor altering the microstructure evolution, but more phase transformation occurred at 900 °C leading to the formation of secondary α after deformation.
- 2. The initial size of the tube should be designed carefully to avoid possible wrinkling and severe local thinning. The weld is suggested to be placed in section with the smallest equivalent stress during the HGPF to avoid possible crack.
- 3. A new method for establishing HGPF window of titanium alloy was proposed. The method of optimizing the process parameters by considering the dimensional accuracy and thickness uniformity was used in the HGPF by RSM. A qualifed TC2 titanium alloy component was successfully formed to verify the optimized process window. The mechanical properties of the formed component are similar to that of the initial

material and the dimensional deviation of the component only about 0.19 mm could be achieved.

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Declarations

Conflict of interest The authors declare no competing interests.

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