

# Assembly precision prediction for planar closed-loop mechanism in view of joint clearance and redundant constraint†

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#### **Abstract**

The assembly process for planar closed-loop mechanisms is full of complexity and uncertainty due to joint clearance, link coupling and probable redundant constraint. In order to ensure assembly precision, an algorithm of predicting accuracy for planar closed-loop mechanisms in view of joint clearance and redundant constraint is proposed. Firstly by analyzing the assembly process of a planar fivebar closed-loop mechanism, three components of single-fixed, two-connected and redundant-inserted links are proposed to describe the assembly process of arbitrary planar closed-loop mechanisms which is regarded as successive stacking of those components. Then error models of those components are established based on the linear kinematics and principle of virtual work. Subsequently, an algorithm of precision prediction for planar closed-loop mechanisms is constructed by combining those error models. Finally the extendible support structure of the SAR antenna is used as the numerical example to verify the validity and generality of the proposed algorithm.

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*Keywords*: Planar closed-loop mechanism; Assembly component; Joint clearance; Redundant constraint; Assembly precision prediction

#### **1. Introduction**

In high precision tasks, accuracy is always of the utmost importance. As for planar closed-loop mechanisms, manufacturing tolerance, constraint deviation and joint clearance inevitably lead to position errors of the terminal vertexes of links during the assembly process, resulting in degeneration of the final accuracy of the mechanism [1]. Therefore from the perspective of improving the geometrical precision and implementing accurate assembling adjustment, an effective method of predicting the assembly accuracy is especially critical for planar closed-loop mechanisms to guarantee the assembly quality and performance.

However, planar closed-loop mechanisms possess such complex constraints and interactive couplings that accuracy modeling is more difficult than open-loop mechanisms. And inherent randomness of the joint clearance brings about assembly precision has uncertainty [2-5]. In addition there are two kinds of planar closed-loop mechanisms which are determinate and indeterminate, respectively. Indeterminate closedloop mechanisms may produce deformation if geometric error exists which leads to the coordinate problem between deformation and geometric error [6]. Hence there are three significant questions to necessarily answer before conducting accuracy prediction for planar closed-loop mechanisms: (1) How to reveal the impact of the complicated coupling among loops on the assembly accuracy; (2) how to analyze the error uncertainty caused by the joint clearance; (3) how to deal with the deformation-error coordinating problem resulted from the geometric errors in the indeterminate closed-loop mechanisms.

Many researchers have been devoted to investigating the effect of manufacturing tolerance and joint clearance on the accuracy of closed-loop mechanisms. Tsai et al. [7] explained why the accuracy of multi-loop linkages is difficult to analyze and they used screw theory to solve the problem. Similarly, Kumaraswamy et al. [8] successfully applied the screw theory to the position error analysis of planar mechanism with considering the joint clearance and link length imperfection. Li et al. [9] studied the angular errors of a multi-loop structure systematically, establishing the explicit solutions of the angular errors with joint clearance and simplifying the extreme error to be an optimization problems based on the invariant rotatability of single-loop linkages. Chebbi et al. [10] developed an analytical predictive model of the pose error for a 3-UPU parallel robot due to joint clearance based on the kinetostatic analysis, and this method can obtain a deterministic result as long as the external force is known. Briot et al. [11] presented the local maximum orientation and position errors of a 3-DOF planar parallel robots considering the input bias according to

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Fig. 1. Assembly process of the planar single-loop five-bar mechanism: (a) Two constraints near the base AB; (b) two constraints away from the base AB; (c) one constraint near the base while the other away from the base; (d) inserting a redundant bar KM.

the interval analysis method. Chen [12] proposed a general method of accuracy analysis for planar parallel mechanisms subjected to errors from the input uncertainties and joints clearance. Besides, plenty of researchers also have studied the kinematic accuracy affected by the joint clearance [13-15] and link imperfection [16, 17]. And a lot of methods are used to deal with the precision problems by the interval approach [18], the matrix method [19], Lie group and Lie algebra method [20], the direct linearization method [21], the level set method [22] and the method based on the generalized kinematic mapping of constrained plane motions [23]. But in contrast fewer literatures [24, 25] involve in the accuracy analysis for closedloop mechanisms with considering redundant constraints.

It can be seen from above literatures that the previous researches with respect to the accuracy analysis of closed-loop mechanisms focus mostly on the influence of manufacturing errors, joint clearance or input uncertainty on the pose deviation of the end-manipulator or the platform, which still belongs to the field of kinematic accuracy rather than assembly precision. And the common deficiency of methods reviewed above is that they wholly didn't take account of the manufacturing deviation, joint clearance and redundant constraints simultaneously and meanwhile neglect the indeterminate mechanisms. Many researchers didn't perform a deep analysis on the deformation-error coordinating problem derived from the redundant constraints so that most methods were only applied to the statically determinate mechanism. As a result many methods of accuracy analysis mentioned above lack generality and are only available for specific mechanisms. Most importantly, few researchers systematically and clearly answered those three key questions about assembly precision prediction for planar closed-loop mechanism.

Therefore, this paper proposes a method of assembly precision prediction for planar closed-loop mechanisms in view of the joint clearance and redundant constraints. Compared with the conventional methods, this method not only takes the joint clearance and redundant constraint into consideration simulta neously but also solves the complex problem of deformation compatibility, which is usually neglected for simplicity in past, by a comparatively simple and intuitive way based on the kinetostatic analysis. And importantly, this method is appropriate for the error modeling of arbitrary planar closed-loop mechanisms regardless of whether the mechanism is indeter minate rather than being only applied to a given closed-loop mechanism.

The contents of this paper are arranged as follows. In Sec. 2, three assembly components are introduced and accordingly their error models are established according to the linear kinematics and principle of virtual work. In Sec. 3, the algorithm of precision prediction for arbitrary planar closed-loop mechanisms is constructed by combining those three error models. In Sec. 4, the proposed method and algorithm are verified by a numerical example of the extendible support structure (ESS) of the SAR antenna which is a three-loop ten bar mechanism. Finally conclusions and future works are presented in Sec. 5.

# **2. Error modeling**

#### *2.1 Description of assembly process*

Assembling planar closed-loop mechanisms is a process of jointing the links, substantially "generating new vertexes". Fig. 1 illustrates the assembly process of a simply but typical sin gle-loop five-bar mechanism whose DOF is 2, needing two constraints to keep statically determinate. It can be seen from Figs. 1(a)-(c) that the assembly process mainly contains two operations after the reference bar AB is ensured: 1) Create a new vertex by fixing a single bar; 2) generate a new vertex by



Fig. 2. Single-link assembly component (without considering assembly Fig. 2. Single-link assembly component (without considering assembly  $\frac{Fig. 3}{\text{Fig. 3}}$ . Single-link assembly component (considering assembly error).

connecting two bars. Those two operations is just subjected to the determinate mechanism. Those two operation are available for all planar closed-loop mechanisms. Besides, the determinate configuration will become indeterminate if adding more constraints or redundant bars to the mechanism as shown in Fig. 1(d). Therefore the conclusion can be drawn that assembly process of arbitrary planar closed-loop mechanisms is realized by fixing single bar, connecting two bars and inserting redundant bar. As a result, the error models of those three assembly operations must be obtained before establishing the assembly prediction algorithm.

## *2.2 Single-link fixed component*

#### *2.2.1 Without assembly deviation*

Fig. 2 presents the model of single-link fixed component without considering the assembly deviation. In order to obtain the pose errors of bar BC, the first work is to compute the position error of points B and C. It is readily known that the position error of B results from the base A and the joint clearance between A and B. Therefore the position error of point B Fig. 1(d). Therefore the conclusion can be drawn that assem-<br>
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realized by fixing single bar, connecting two bars and insert-<br>
ing redundant bar. As a result, the e Fig. 4. Schematic algorithm.<br> **Example 11.1** Fig. 4. Schematic sketch for calculation algorithm.<br> **Example 11.1** Fig. 4. Schematic sketch for calculation<br>
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model of single **Fig.** 4. Schematic sketch for calculation<br>
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smallering the assembly deviation. In order to obtain<br> **AB** *AB AB**AB**x* **<b>***AB <i>x absention AB x absention in the second of single-link fixed component*  $\delta_c^r = \delta_a^r + \Gamma_a^r + \Delta_{1a} \Phi_v^r$ .<br>
AB A Schematic sketch for case and the model of single-link fixed compon **and** *x x* (*x*) *x x* (*x*) *x x* (*x*) *x* (*x*) *x x y* embly prediction algorithm.<br> **Single-link fixed component**<br> **Eig. 4.** Schematic sketch for<br> **Civ**<br> **Civ**<br> **Civ**<br> **Civ**<br> **Eig. 2** presents the model of single-link fixed component<br> **Fig. 4.** Schematic sketch for<br>
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$$
\delta_B^T = \delta_A^T + \Gamma_{AB}^T \tag{1}
$$

where  $\delta_B$  and  $\delta_A$  are respectively the position error of B  $\epsilon$ and A,  $\Gamma_{AB}$  represents the joint clearance vector [7] and

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\delta_{B}^{T} = \delta_{A}^{T} + \Gamma_{AB}^{T}
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\text{are} \quad \delta_{B} \quad \text{and} \quad \delta_{A} \quad \text{are respectively, the position error of Bh A,  $\Gamma_{AB}$  represents the joint clearance vector [7] and\n
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\begin{cases}\n\delta_{B} = [\Delta x_{A}, \Delta y_{A}] \\
\delta_{A} = [\Delta x_{A}, \Delta y_{AB}] \\
\Gamma_{AB} = [\Delta x_{AB}, \Delta y_{AB}]\n\end{cases}
$$
\n(2)

\nHere, the module of  $\Gamma$  is not more than the value of the
$$

where the module of  $\Gamma_{AB}$  is not more than the value of the where the first term of  $\Phi_{B/2+\varphi}$ ,  $\pm \sin(\vartheta/2+\varphi)$ , is positive joint clearance. In addition the position misalignment of C is derived from B and the manufacturing error of BC, formulated ance between A and B. Therefore the position error of point B<br>
can be given by<br>  $\delta'_a = \delta'_a + \Gamma'_{.a}$ <br>  $\delta'_b = \delta'_{.a} + \Gamma'_{.a}$ <br>
where  $\delta_a$  and  $\delta_a$  are respectively the position error of B<br>
expressed as<br>
and A,  $\Gamma_{.a}$  re

$$
\delta_C^T = \delta_B^T + \Delta l_{BC} \Phi_{\varphi}^T
$$
 (3)

and  $\varphi$  is the nominal assembly angle. Substituting Eq. (1) into Eq. (3) gets





two-link assembly component.

$$
\delta_C^T = \delta_A^T + \Gamma_{AB}^T + \Delta l_{BC} \Phi_{\varphi}^T.
$$
 (4)

#### *2.2.2 With assembly deviation*

Fixing a bar can be divided into two steps when considering the assembly error as depicted in Fig. 3: Firstly assembly BC at the ideal constrained angle  $\varphi$  to create point C; then rotate BC by  $\theta$  ( $\theta$  is the assembly angle error and very small) to produce the final point  $C_1$ . Note that the position error of  $C_1$ rather than B is affected by the assembly deviation which is expressed as  $\overrightarrow{Ox}_A$ <br>
Schematic sketch for calculating  $\partial x_x / \partial x_x$  and  $\partial y_x / \partial x_x$  in<br>
assembly component.<br>  $\delta_A^T + \Gamma_{AB}^T + \Delta I_{BC} \Phi_x^T$ . (4)<br>
Tith assembly deviation<br>  $\delta_A^T + \Gamma_{AB}^T + \Delta I_{BC} \Phi_x^T$ . (4)<br>
Tith assembly deviation<br>
embly c sketch for calculating  $\hat{\alpha}_{\kappa}/\hat{\alpha}_{\kappa}$  and  $\hat{\sigma}_{\gamma_{\kappa}}/\hat{\alpha}_{\kappa}$  in<br>y component.<br><br><br><br><br><br><br>**anbly deviation**<br><br>can be divided into two steps when considering<br>rror as depicted in Fig. 3: Firstly assembly BC<br>ststrained **The Expansion of the Control of**  $\partial x_A$ <br> **T**  $\partial x_A$ <br> **T** Schematic sketch for calculating  $\partial x_k / \partial x_A$  and  $\partial y_k / \partial x_A$  in the assembly component.<br> **The BC**  $\partial x_A + \nabla x_B + \Delta I_B C \Phi_{\varphi}^T$ .<br> **With assembly deviation**<br> **The BC** al constrained angle  $\varphi$  to create point C; then rotate<br>
( $\vartheta$  is the assembly angle error and very small) to<br>
the final point C<sub>1</sub>. Note that the position error of C<sub>1</sub><br>
in B is affected by the assembly deviation whic  $\partial x_A$ <br>
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pronent.<br>  $M_{\varepsilon c} \Phi_{\varphi}^T$ . (4)<br>  $y$  deviation<br>  $\omega$  deviation<br>  $\omega$  deviation<br>  $\omega$  deviation<br>  $\omega$  as depicted in Fig. 3: Firstly assembly BC<br>
read an 4.  $\Delta t = \frac{1}{\alpha X_A}$ <br>
4. Schematic sketch for calculating  $\alpha_X / \alpha_X$  and  $\partial_{Y_E} / \alpha_X$  in<br>  $\delta_C = \delta_A^T + \Gamma_{AB}^T + \Delta I_{BC} \Phi_{\varphi}^T$ . (4)<br>
1.2 With assembly deviation<br>  $\sum_{i=1}^N \sum_{i=1}^N \sum_{j=1}^N \Delta I_{ij} \Delta E_{\varphi}$ . (4)<br>  $\sum_{i=1}^N \sum$  $\partial x_A$ <br>
4. A Schematic sketch for calculating  $\partial x_k / \partial x_x$  and  $\partial y_k / \partial x_x$  in<br>  $\delta_C^T = \delta_A^T + \Gamma_{AB}^T + \Delta I_{BC} \Phi_\phi^T$ .<br>
(4)<br>
1.2 With assembly deviation<br>
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1.2 With assembly deviation<br>
1.2 With assem . 4. Schematic sketch for calculating  $\partial x_{\varepsilon}/\partial x_{\varepsilon}$  and  $\partial y_{\varepsilon}/\partial x_{\varepsilon}$  in  $-\lim_{\varepsilon} k \text{ assembly component.}$ <br>  $\delta_c^{\tau} = \delta_{\varepsilon}^{\tau} + \Gamma_{\varepsilon\theta}^{\tau} + \Delta l_{\varepsilon\epsilon} \Phi_{\varphi}^{\tau}$ . (4)<br>
1.2 With assembly deviation<br>  $\Gamma$  ixing a bar **δ δ Γ Φ Φ** (4)<br>
to two steps when considering<br>
n Fig. 3: Firstly assembly BC<br>
to create point C; then rotate<br>
angle error and very small) to<br>  $e^{\pm}$  that the position error of C<sub>1</sub><br>
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rather than B is affected by the assembly deviation which is expressed as\n
$$
\delta_c^T = \delta_A^T + \Gamma_{AB}^T + \Delta I_{BC} \Phi_{\varphi+\vartheta}^T + I_{BC} \partial \Phi_{g/2+\varphi}^T
$$
\n
$$
\Phi_{g/2+\varphi} = [\pm \sin(\frac{1}{2}\vartheta + \varphi), \cos(\frac{1}{2}\vartheta + \varphi)] \qquad (5)
$$
\nwhere the first term of  $\Phi_{g/2+\varphi}$ ,  $\pm \sin(\vartheta/2 + \varphi)$ , is positive when  $\vartheta > 0$  and otherwise is negative.  
\n2.3 Two-link connected component

when  $9 > 0$  and otherwise is negative.

#### *2.3 Two-link connected component*

 $\vec{a}$ ,  $\vec{a}$ ,  $\vec{b}$ ,  $\vec{a}$  is the center of BC,  $\vec{b}$  is altered by the sesembly deviation error of C<sub>1</sub><br>
where  $\vec{a}$ , and  $\vec{a}$ , represents the joint clearance vector [7] and<br>
and A,  $\Gamma_m$  represents the As shown in Fig. 4,  $A_1$  and  $B_1$  are the bases of AC and BD. Suppose C and D are ideally coincident after connecting AC and BD by initially neglecting the joint clearance between C and D, and the coincident point is denoted as E. Firstly, according to Eq. (1) the position error of A and B can be derived readily as

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\n*Q. Zhao et al. / Journal of Mechanical Science an*  
\n
$$
\begin{cases}\n\mathbf{\delta}_A^T = \mathbf{\delta}_{A_1}^T + \mathbf{\Gamma}_{A_1}^T \\
\mathbf{\delta}_B^T = \mathbf{\delta}_{B_1}^T + \mathbf{\Gamma}_{BB_1}^T.\n\end{cases}
$$
\n(6)

Then the position deviation of E comes from the base im perfection of A and B and link length errors of AC and BD in

$$
\delta_E^T = \lambda_1 \Delta_I^T + \lambda_2 \delta_A^T + \lambda_3 \delta_B^T \tag{7}
$$

matrixes defined as

Then the position deviation of E comes from the base im-  
\nflection of A and B and link length errors of AC and BD in  
\nms of Fig. 4, interpreted by  
\n
$$
\delta_E^T = \lambda_1 \Delta_i^T + \lambda_2 \delta_A^T + \lambda_3 \delta_B^T
$$
\n(7)  
\nhere  $\Delta_i = [\Delta I_{AC}, \Delta I_{BD}]$  and  $\lambda_1, \lambda_2, \lambda_3$  are the Jacobian  
\ntrixes defined as  
\n
$$
\lambda_1 = \begin{bmatrix} \frac{\partial x_E}{\partial I_{AC}} & \frac{\partial x_E}{\partial I_{BD}} \\ \frac{\partial y_E}{\partial I_{AB}} & \frac{\partial y_E}{\partial I_{BD}} \end{bmatrix}
$$
\n(8) tri

Here 
$$
\Delta_i = [\Delta l_{AC}, \Delta l_{BD}]
$$
 and  $\lambda_1$ ,  $\lambda_2$ ,  $\lambda_3$  are the Jacobian  
\ntrixes defined as  
\n
$$
\lambda_1 = \begin{bmatrix}\n\frac{\partial x_E}{\partial l_{AC}} & \frac{\partial x_E}{\partial l_{BD}} \\
\frac{\partial y_E}{\partial l_{AB}} & \frac{\partial y_E}{\partial l_{BD}}\n\end{bmatrix}
$$
\n(8) tri  
\n
$$
\lambda_2 = \begin{bmatrix}\n\frac{\partial x_E}{\partial x_A} & \frac{\partial x_E}{\partial y_A} \\
\frac{\partial y_E}{\partial x_A} & \frac{\partial y_E}{\partial y_A}\n\end{bmatrix}
$$
\n(9)

Then the position deviation of E comes from the base im-  
\nflection of A and B and link length errors of AC and BD in  
\nms of Fig. 4, interpreted by  
\n
$$
\delta_{E}^{r} = \lambda_{1}\Delta_{i}^{T} + \lambda_{2}\delta_{A}^{T} + \lambda_{3}\delta_{B}^{T}
$$
\n(7)  
\nare  $\Delta_{i} = [\Delta_{A,C}, \Delta_{BD}]$  and  $\lambda_{1}, \lambda_{2}, \lambda_{3}$  are the Jacobian  
\ntrives defined as  
\n
$$
\lambda_{1} = \begin{bmatrix} \frac{\partial x_{E}}{\partial I_{AC}} & \frac{\partial x_{E}}{\partial I_{BD}} \\ \frac{\partial y_{E}}{\partial I_{AB}} & \frac{\partial y_{E}}{\partial I_{BD}} \end{bmatrix}
$$
\n(8)  
\n
$$
\lambda_{2} = \begin{bmatrix} \frac{\partial x_{E}}{\partial I_{AC}} & \frac{\partial x_{E}}{\partial I_{BD}} \\ \frac{\partial y_{E}}{\partial I_{AC}} & \frac{\partial y_{E}}{\partial I_{CD}} \end{bmatrix}
$$
\n(9)  
\n
$$
\lambda_{3} = \begin{bmatrix} \frac{\partial x_{E}}{\partial I_{AC}} & \frac{\partial x_{E}}{\partial I_{BC}} \\ \frac{\partial y_{E}}{\partial I_{AC}} & \frac{\partial y_{E}}{\partial I_{BC}} \end{bmatrix}
$$
\n(10)  
\n
$$
\lambda_{4} = \begin{bmatrix} \frac{\sin(\beta - \varepsilon)}{\sin \alpha} & -\frac{\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)}{\cos(\gamma + \varepsilon)} & \frac{\cos(\gamma + \varepsilon)}{\cos(\gamma + \varepsilon)} \\ \frac{\partial y_{E}}{\partial I_{BC}} & \frac{\partial y_{E}}{\partial I_{BC}} \end{bmatrix}
$$
\n(10)  
\nTherefore  $\lambda_{1}, \lambda_{2}, \lambda_{3}$  should be evaluated before solving  
\n
$$
\mu_{1} = \begin{bmatrix} \frac{\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} & \frac{\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} \end{
$$

Therefore  $\lambda_1$ ,  $\lambda_2$ ,  $\lambda_3$  should be evaluated before solving Eq. (7). As presented in Fig. 4, point E will rotate around point B by a radius of BD if  $x_4$  produces a perturbation  $\partial x_4$ while the rest variables keep invariant. Under this circum stances, treat the perturbation  $\partial x_i$  as a velocity variable and then we have

$$
\lambda_{2} = \begin{bmatrix} \frac{\partial x_{E}}{\partial x_{A}} & \frac{\partial x_{E}}{\partial y_{A}} \\ \frac{\partial y_{E}}{\partial x_{A}} & \frac{\partial y_{E}}{\partial y_{B}} \end{bmatrix}
$$
\n
$$
\lambda_{3} = \begin{bmatrix} \frac{\partial x_{E}}{\partial x_{A}} & \frac{\partial x_{E}}{\partial y_{A}} \\ \frac{\partial y_{E}}{\partial x_{A}} & \frac{\partial y_{E}}{\partial y_{B}} \end{bmatrix}
$$
\n
$$
\lambda_{4} = \begin{bmatrix} \frac{\sin(\beta - \varepsilon)}{\cos(\beta - \varepsilon)} & \frac{\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)}{\sin \alpha} & \frac{\cos(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} & \frac{\cos(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)\sin(\gamma
$$

where  $\alpha$ ,  $\beta$  and  $\gamma$  are interior angles of the triangle formed by the two bars and the base structure, detailedly displayed in Fig. 4, and  $\varepsilon$  is originally the actual angle between and bearing,  $\Gamma_{CE}$  and  $\Gamma_{DE}$  must satisfy the positive direction of axis *x* and the connecting line of A and B but for simplicity, here  $\varepsilon$  is the included angle of positive direction of axis x and the base structure of  $A_1$  and  $B_1$ . Let on the introduction of  $\lambda_1$ ,  $\lambda_2$  and  $\lambda_3$  in the method of  $\lambda_1$ ,  $\lambda_2$  and  $\lambda_3$  in the method of  $\lambda_1$ ,  $\lambda_2$  and  $\lambda_3$  in the method of  $\lambda_1$ ,  $\lambda_2$  and  $\lambda_3$  in the state of print time we have<br>th ve  $\sin(\beta - \beta)\cos(\gamma + \varepsilon)$ <br>  $\sin(\beta - \varepsilon)\cos(\gamma + \varepsilon)$ <br>  $\sin(\beta - \varepsilon)\cos(\gamma + \varepsilon)$ <br>  $\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)$ <br>  $\sin(\beta - \varepsilon)\cos(\gamma + \varepsilon)$ <br>  $\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)$ <br>  $\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)$ <br>  $\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)$ <br>  $\sin(\beta - \varepsilon)\cos(\gamma + \varepsilon)$ <br>  $\sin(\beta - \varepsilon)\cos(\$  $\frac{s(y + \varepsilon)}{s} = \frac{\partial x_{\varepsilon}}{\partial s(x + \varepsilon)}$ <br>  $\frac{s(y + \varepsilon)}{s} = \frac{\partial y_{\varepsilon}}{\partial s(x + \varepsilon)}$ <br>  $\frac{s(y + \varepsilon)}{s} = \frac{\partial y_{\varepsilon}}{\partial s(x + \varepsilon)}$ <br>
(11) and D are given by<br>  $\frac{s(y + \varepsilon)}{s} = \frac{\partial y_{\varepsilon}}{\partial s(x + \varepsilon)}$ <br>
(11) and D are given by<br>  $\frac{s(y + \varepsilon)}{s} =$ *x*<sub>x</sub> cos( $y + \varepsilon$ ) =  $\frac{\partial y}{\partial s(y + \beta - \frac{\pi}{2})} = \frac{\partial y}{\partial \text{cos}(\frac{\pi}{2} - (\beta - \varepsilon))}$ <br>  $\frac{\partial x}{\partial s(y + \beta - \frac{\pi}{2})} = \frac{\partial y}{\partial \text{sin}(\frac{\pi}{2} - (\beta - \varepsilon))}$ <br>  $\therefore \alpha, \beta \text{ and } \gamma \text{ are interior angles of the triangle. If } \alpha \in \mathbb{R} \setminus \{1\} \text{ and } \alpha \in \mathbb{R} \setminus \{1\} \text{ and } \alpha \in \mathbb{R}$  $\frac{dx}{dt} = \frac{dx}{dt}$ <br>  $\frac{dx}{dt} = \frac{dy}{dt}$ <br>  $\frac{dx}{dt} = \frac{dy}{dt}$ <br>  $\frac{dy}{dt} = \frac{dy}{dt}$ <br>  $\frac{dy}{dt} = \frac{dy}{dt}$ <br>  $\frac{dy}{dt} = \frac{dy}{dt}$ <br>
(11) and D are given by<br>  $\delta_C^T = \delta_L^T + \Gamma_{cZ}^T$ <br>
and y are interior angles of the triangle<br>  $\delta_C^T = \delta_L^T + \Gamma$ n we have<br>  $\left.\begin{array}{lll}\n\frac{\partial x}{\partial x} \cos(y + \varepsilon) & \frac{\partial x}{\partial x} & \frac{\partial x}{\partial x} \cos(y + \varepsilon) & \frac{\partial y}{\partial x} \cos(y + \varepsilon) & \frac{\partial y}{\$  $\frac{dx_x \cos(y + \epsilon)}{\cos(y + \beta - \frac{\pi}{2})} = \frac{\partial x_{\epsilon}}{\partial x_{\epsilon}}$ <br>  $\frac{dx_{\epsilon}}{\cos(y + \beta - \frac{\pi}{2})} = \frac{\partial y_{\epsilon}}{\partial x_{\epsilon}}$ <br>  $\frac{dy_{\epsilon}}{\cos(y + \beta - \frac{\pi}{2})} = \frac{\partial y_{\epsilon}}{\sin(\frac{\pi}{2} - (\beta - \epsilon))}$ <br>  $\frac{dx_{\epsilon}}{\sin(y + \beta - \frac{\pi}{2})} = \frac{\partial y_{\epsilon}}{\sin(\frac{\pi}{2} - (\beta - \epsilon))}$ <br>
(11) and D are given

$$
\begin{cases}\n\frac{\partial x_E}{\partial x_A} = \frac{\sin(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} \\
\frac{\partial y_E}{\partial x_A} = \frac{\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha}.\n\end{cases}
$$
\n(12)



Fig. 5. Actual position of endpoints when connecting two bars.

By making use of the similar mechanisms like Fig. 4, the other independent variables with respect to the position error of E can be figured out as well. Accordingly Jacobian matrixes  $\lambda_1$ ,  $\lambda_2$ ,  $\lambda_3$  are identified as

$$
\lambda_{1} = \begin{bmatrix} \frac{\sin(\beta - \varepsilon)}{\sin \alpha} & -\frac{\sin(\gamma + \varepsilon)}{\sin \alpha} \\ \frac{\cos(\beta - \varepsilon)}{\sin \alpha} & \frac{\cos(\gamma + \varepsilon)}{\sin \alpha} \end{bmatrix} \tag{13}
$$

$$
\lambda_2 = \begin{vmatrix} \frac{\sin(\beta - \epsilon)\cos(\gamma + \epsilon)}{\sin \alpha} & \frac{\sin(\beta - \epsilon)\sin(\gamma + \epsilon)}{\sin \alpha} \\ \cos(\beta - \epsilon)\cos(\gamma + \epsilon) & \cos(\beta - \epsilon)\sin(\gamma + \epsilon) \end{vmatrix}
$$
 (14)

(a) Ideal statement  
\n(b) Actual statement  
\n(c) Actual statement  
\n(d) Ideal statement  
\n(a) Ideal position of endpoints when connecting two bars.  
\nBy making use of the similar mechanisms like Fig. 4, the  
\nher independent variables with respect to the position error  
\nE can be figured out as well. Accordingly Jacobian ma-  
\naxes 
$$
\lambda_1
$$
,  $\lambda_2$ ,  $\lambda_3$  are identified as  
\n
$$
\lambda_1 = \begin{bmatrix}\n\frac{\sin(\beta - \varepsilon)}{\sin \alpha} & -\frac{\sin(\gamma + \varepsilon)}{\sin \alpha} \\
\frac{\cos(\beta - \varepsilon)}{\sin \alpha} & \frac{\cos(\gamma + \varepsilon)}{\sin \alpha}\n\end{bmatrix}
$$
\n(13)  
\n
$$
\lambda_2 = \begin{bmatrix}\n\frac{\sin(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} \\
\frac{\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha}\n\end{bmatrix}
$$
\n(14)  
\n
$$
\lambda_3 = \begin{bmatrix}\n\frac{\cos(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\sin(\gamma + \varepsilon)}{\sin \alpha} \\
\frac{\cos(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha} & \frac{\sin(\beta - \varepsilon)\cos(\gamma + \varepsilon)}{\sin \alpha}\n\end{bmatrix}
$$
\n(15)  
\nThen the introduction of  $\lambda_1$ ,  $\lambda_2$  and  $\lambda_3$  into Eq. (7) can  
\neld the position error of the concurrent point E.  
\nNevertheless in fact, C and D maybe not hold together with  
\neat probability due to the existence of joint clearance and  
\ney will deviate from the concurrent E as shown in Fig. 5.  
\nSince uniting Eqs. (1) and (7), the actual position errors of C  
\nd D are given by  
\n
$$
\begin{cases}\n\delta_c^r = \delta_c^r + \Gamma_{DE}^r \\
\delta_b^r = \delta_c^r + \Gamma_{DE}^r\n\end{cases}
$$
\n(16)  
\ntherefore E. Because of the mating constraint between the shaft  
\nd bearing,  $\Gamma_{CE}$  and  $\Gamma_{DE}$  must satisfy  
\n
$$
\Gamma_{DE} = \begin{bmatrix}\
$$

Then the introduction of  $\lambda_1$ ,  $\lambda_2$  and  $\lambda_3$  into Eq. (7) can yield the position error of the concurrent point E.

Nevertheless in fact, C and D maybe not hold together with great probability due to the existence of joint clearance and they will deviate from the concurrent E as shown in Fig. 5. Hence uniting Eqs. (1) and (7), the actual position errors of C Then the introduction of  $\lambda_1$ ,  $\lambda_2$  and  $\lambda_3$  into Eq. (7)<br>yield the position error of the concurrent point E.<br>Nevertheless in fact, C and D maybe not hold together<br>great probability due to the existence of joint cle **Example 12 F FC** and **F FC** and **F FC** and **F FC** and **F F F C C** *C* **F C** *C C C C C C C C C C C C C C C*

$$
\begin{cases}\n\boldsymbol{\delta}_c^T = \boldsymbol{\delta}_E^T + \boldsymbol{\Gamma}_{CE}^T \\
\boldsymbol{\delta}_D^T = \boldsymbol{\delta}_E^T + \boldsymbol{\Gamma}_{DE}^T\n\end{cases}
$$
\n(16)

where  $\Gamma_{CE}$  and  $\Gamma_{DE}$  are the offset vectors of C and D relative to E. Because of the mating constraint between the shaft

$$
|\Gamma_{CE} - \Gamma_{DE}| \le r_c \tag{17}
$$

where  $r_c$  is the magnitude of joint clearance and  $|\cdot|$  represents the vector norm.

#### *2.4 Redundant-link inserted component*

In general, two cases exist in assembling redundant link. On the one hand, the length of redundant link equals the distance



Fig. 6. Actual sketch of inserting a redundant bar.

between two bases and the mechanism keeps stable without deformation; on the other hand, the redundant link has deviation comparing with the two connected bases, unfortunately resulting in deformation inside the bars after inserting it. In this paper the latter case illustrated in Fig. 6 is taken into account and discussed sequentially.

Before solving the problem of inserting redundant bar, the following assumption is given: All bars except the redundant one don't generate deformation when installing it, namely  $-\Delta \mathbf{f}_2$ ). implying the base structure is thought of as a rigid body. Firstly, because the offset between the length of the inserted bar and fixed bases is so small that deformation of the redundant bar can't produce great force on the base structure; on the other hand, the stiffness of the base structure is much bigger in contrast with the redundant bar, indicating the base structure exerted by small force from the redundant bar hardly appear where  $f = \Delta f_2$  and  $s_i$  is a 2×2 matrix whose elements only deformation. Therefore the above hypothesis can be truly accessible for practical engineering. In an discussed sequentially.<br>
Electro solving the problem of inserting redundant bar, the estapona of inserting redundant force  $\Delta f_z$  (if  $-\Delta f_z$  is<br>  $\Delta f_z$ ), the example of the state of the state of the computation when bar and fixed bases is so small that deformation of the redun-<br>
contact force in the *j*th joint is<br>
dant bar can't produce great force on the base structure; on the<br>
other hand, the stiffness of the base structure is muc *x* and fixed bases is so small that deformation of the redun-<br> *x* and fixed bases is so small that deformation of the redun-<br> *x* that can't produce great force on the base structure; on the<br> *x* rata  $x$  in that the in

In term of the hypothesis, inserting redundant bar with deviation  $\Delta l$ , can be identified with implementing forces  $\Delta f$ , and  $-\Delta f$ , on the base structure as shown in Fig. 7. The magnitude of  $\Delta f$ , is

$$
|\Delta \mathbf{f}_2| = k \Delta l_r^* \tag{18}
$$

which can be rewritten by a vector form for sequent discus-

$$
\Delta \mathbf{f}_2 = [\Delta f_{2x}, \Delta f_{2y}]^T. \tag{19}
$$

It should be noted that  $\Delta l_r^*$  doesn't equal  $\Delta l_r$  since the position of the base points will change under the force. Equally important, the imposing orders of  $\Delta f$ , and  $-\Delta f$ , distinguishes in assembly process. As the one end of the redundant link is jointed to the one fixed base, the force  $\Delta \mathbf{f}_2$  is<br>exerted on this base point. And then **cf** and **-cf** the con-<br> $(\mathbf{cf}_i \delta \mathbf{cp}_{ij} - \mathbf{cf}_i \delta \mathbf{cp}_{ij}) +$ exerted on this base point. And then  $cf_i$  and  $(cf_i$ , the contact forces of the shaft and bearing of the joint as shown in Fig. 8, appear under the action of  $\Delta f$ , and accordingly the



Fig. 7. Equivalent sketch of inserting a redundant bar.

mechanism maintains static equilibrium by those forces. Subsequently  $-\Delta f$ , acts on the other base point and the whole base structure still keeps original equilibrium configuration. In terms of the above analysis, determining the position errors of the endpoints of links when inserting the redundant bar is in essence to obtain the displacement bias under the external force  $\Delta f$ , (if  $-\Delta f$ , is applied first, here the external force is  $-\Delta \mathbf{f}_2$ ). Fig. 7. Equivalent sketch of inserting a redundant bar.<br>
Fig. 7. Equivalent sketch of inserting a redundant bar.<br>
mechanism maintains static equilibrium by those forces. Sub-<br>
sequently  $-\Delta f$ , acts on the other base point Fig. 7. Equivalent sketch of inserting a redundant bar.<br>
Fig. 7. Equivalent sketch of inserting a redundant bar.<br>
executently  $-\Delta f_2$  acts on the other base point and the whole<br>
base structure still keeps original equilib Hency  $-\Delta x_1$  aas on the onter lose folmt and the whole<br>is estructure still keeps original equilibrium configuration. In<br>ms of the above analysis, determining the position errors of<br>endpoints of links when inserting the r

The actual joint generally has clearance but here let the joint be clearance-free. According to the static equilibrium, the

$$
cf_j = s_j \Delta f_2 \tag{20}
$$

depend on the mechanism configuration and  $\Delta f$ <sub>2</sub>. Next based on the Principle of Virtual work the mechanism of restoring equilibrium satisfies the following equation *g* and on the mechanism contiguration and  $\Delta t_2$ . Next based<br> *j* Principle of Virtual work the mechanism of restoring<br>
brium satisfies the following equation<br> **f**  $\cdot \delta \mathbf{r} = 0$ . (21)<br>
cordingly applying the principle *x*  $x_1$ ,  $(x - 2x_2)$  *x* as applied *x* as, there are external rote *x*  $x_1$ , the actual joint generally has clearance but here let the joint clearance-free. According to the static equilibrium, the tact force in the *j* Let  $\alpha_1$  the  $\alpha_2$  to approach the station where let the joint<br>actual joint generally has clearance but here let the joint<br>arance-free. According to the static equilibrium, the<br>force in the *j*th joint is<br> $\mathbf{s} \cdot \Delta \mathbf$  $\frac{1}{1}$  -  $\frac{1}{1}$  -  $\frac{1}{2}$  -  $\frac{1}{2}$  -  $\frac{1}{2}$  -  $\frac{1}{2}$  -  $\frac{1}{2}$  -  $\frac{1}{2}$  and s  $\frac{1}{2}$  to the static equilibrium, the in the *j*th joint is<br>
(20)<br>  $f_2$  and s is a 2×2 matrix whose elements only<br>
e

$$
\sum \mathbf{f} \cdot \delta \mathbf{r} = 0. \tag{21}
$$

Accordingly applying the principle to the base structure free of joint clearance in Fig. 7 can obtain

$$
(cfμδcpqix + cfjpδcpqiy - cfjkδcphix-cfjpδcphjy) + Δf2xδjk + Δf2yδjp = 0
$$
\n(22)

**Example 10** and interactive the static equilibrium, the electrance-free. According to the static equilibrium, the act force in the *j*th joint is  $f_j = s_j \Delta f_2$  (20)<br> **cf**  $f = \Delta f$ , and **s**, is a 2×2 matrix whose elements o notual joint generally has clearance but here let the joint<br>rance-free. According to the static equilibrium, the<br>force in the *j*th joint is<br> $s_j\Delta f$ , (20)<br> $f = \Delta f_2$  and  $s_j$  is a 2×2 matrix whose elements only<br>on the mech The actual joint generally has clearance but here let the joint<br>clearance-free. According to the static equilibrium, the<br>tact force in the *j*th joint is<br> $cf$ ,  $= s_j \Delta f_2$  (20)<br>ere  $f = \Delta f_2$  and  $s_j$  is a 2×2 matrix whose e where  $cf_{jx}$  and  $cf_{jy}$  are the contact forces of the *j*th joint,  $\delta cp_{\text{qjx}}$ ,  $\delta cp_{\text{qjx}}$ ,  $\delta cp_{\text{bjx}}$  and  $\delta cp_{\text{bjx}}$  are the virtual displacements of the two contact points a and b as presented in Fig. 8 along the directions of axis *x* and *y*, respectively. Rewrite Eq. (22) by the vector format as the Principle of Virtual work the inectralism of restoring<br>ilibrium satisfies the following equation<br> $\sum \mathbf{f} \cdot \delta \mathbf{r} = 0$ . (21)<br>Accordingly applying the principle to the base structure free<br>iont clearance in Fig. 7 can

$$
(\mathbf{cf}_{i}\delta\mathbf{cp}_{ai} - \mathbf{cf}_{i}\delta\mathbf{cp}_{bi}) + \Delta\mathbf{f}_{i}\delta\mathbf{r}_{f} = 0.
$$
 (23)

It's readily known from Fig. 8 that

$$
\delta \mathbf{cp}_i = \delta \mathbf{cp}_{ai} - \delta \mathbf{cp}_{bi} \,. \tag{2}
$$

Then substituting Eqs. (20) and (24) into Eq. (23) gets

$$
\Delta \mathbf{f}_2 \cdot \delta \mathbf{\Gamma} + \mathbf{s}_j \Delta \mathbf{f}_2 \cdot \delta \mathbf{c} \mathbf{p}_j = 0. \tag{25}
$$

$$
\delta \Gamma = -s_j \cdot \delta cp_j \tag{26}
$$

where

$$
\delta \mathbf{cp}_j = \delta c p_j \frac{\mathbf{cf}_j}{|\mathbf{cf}_j|}.
$$
 (27)

It is seen from Eq. (27) that the influence of joint clearance on the position error of the acting point is not relevant with the magnitude of the force. Given the base structure owns more than one joint, the position error of the acting point  $\frac{cf_j}{|cf_j|}$ . (27) that the influence of joint clearance<br>
n error of the acting point is not relevant with the<br>
the force. Given the base structure owns more<br>
the position error of the acting point<br>  $\cdot \delta cp_j$  (28<br>
e joint

$$
\delta \Gamma = \sum_{j=1}^{n} -\mathbf{s}_{j} \cdot \delta \mathbf{c} \mathbf{p}_{j}
$$
 (28)

where *n* is the joint number. Substitute Eq. (27) into Eq. (28) and we obtain

in one joint, the position error of the acting point  
\n
$$
\delta \mathbf{\Gamma} = \sum_{j=1}^{n} -\mathbf{s}_{j} \cdot \delta \mathbf{c} \mathbf{p}_{j}
$$
\n(28)  
\nhere *n* is the joint number. Substitute Eq. (27) into Eq. (28)  
\n1 we obtain\n
$$
\delta \mathbf{\Gamma} = \sum_{j=1}^{n} -\delta c p_{j} \mathbf{s}_{j} \cdot \frac{\mathbf{cf}_{j}}{|\mathbf{cf}_{j}|}.
$$
\n(29)  
\nGenerally speaking, the joint clearance can be treated as in-

Generally speaking, the joint clearance can be treated as infinitesimal because it is much smaller relative to the nominal length of bars, consequently  $=\sum_{j=1}^{n} -\delta cp_j s_j \cdot \frac{cf_j}{|cf_j|}$ . (2<br> *j* erally speaking, the joint clearance can be treated as i<br> *j* and because it is much smaller relative to the nominor<br> *j*  $= \Delta cp_j = r_{cf} \frac{cf_j}{|cf_j|}$  (3<br>  $r_{cf}$  is the value of the *j* 

$$
\delta \mathbf{cp}_j = \Delta \mathbf{cp}_j = r_{cj} \frac{\mathbf{cf}_j}{|\mathbf{cf}_j|}
$$
 (30)

where  $r_{ci}$  is the value of the *j*th joint clearance. Then the introduction of Eq. (30) into Eq. (29) leads to the equation

$$
δcpj = Δcpj = rσ  $\frac{cf_j}{|cf_j|}$  (30)  
\nhere  $rσ$  is the value of the *j*th joint clearance. Then the  
\nroduction of Eq. (30) into Eq. (29) leads to the equation  
\n
$$
δΓ = \sum_{j=1}^{n} -rσsj \cdot \frac{cf_j}{|cf_j|}.
$$
 (31)  
\nIn the meantime considering the two forces respectively ap-
$$

In the meantime considering the two forces respectively applied on the shaft and bearing are reaction, the virtual displacements of their centers equal but possess opposite directions, which signifies

$$
\delta \mathbf{cp}_{ai} = -\delta \mathbf{cp}_{bi} \tag{32}
$$

Uniting Eqs. (24), (30) and (32) generates



Fig. 8. Sketch of virtual displacement of the *j*th joint clearance as the mechanism applied by  $\Delta f$ , .



Fig. 9. Open-chain generalized link.



Fig. 10. Algorithm of assembly precision prediction for planar closedloop mechanisms.

$$
\Delta \mathbf{cp}_{aj} = -\Delta \mathbf{cp}_{bj} = \frac{1}{2} \Delta \mathbf{cp}_{ab} = \frac{1}{2} r_{cj} \frac{\mathbf{cf}_{j}}{|\mathbf{cf}_{j}|}.
$$
 (33)

Accordingly the position errors of the endpoints, i.e. the

centers of the shaft and bearing, can be calculated by Eqs. (33) and (31) when assembling a redundant link. Then the actual assembly errors of links are able to be obtained by superposition with the original error of the base structure before inserting the redundant bar. But it should be noted that the mentioned 'original error' above should be evaluated without tak-

Table 1. Structure parameters of the ESS.

<b>Bars</b>	Nominal value (mm)	Deviation (mm)
AF	212	1.10
DF	93	$-0.75$
ED	218.71	0.94
FI	210	$-1.26$
DI	229.67	$-0.86$
AB	125	0.94
BD	214.4	1.03
AE	14	$-0.35$

Table 2. Errors of 90- and 180-degree locked joints.





Fig. 11. The ESS of the planar SAR antenna (planar three-loop mechanism with ten bars).

ing account of the joint clearance.

#### *2.5 Generalized link*

Three abovementioned assembly components all involve in 'single link'. This 'single link' is defined as a generalized link which represents not only the actual single bar but also multibar subassembly. Thus the generalized link probably possesses 'length deviation' due to the manufacturing imperfection, joint clearance and assembly error as well. 'single link'. This 'single link' is defined as a generalize<br>which represents not only the actual single bar but also<br>bar subassembly. Thus the generalized link probably<br>sesses 'length deviation' due to the manufacturing *Cehnology 32 (7) (2018) 3395-3405* 3401<br>
3401<br>
account of the joint clearance.<br> **Ceneralized link**<br>
have abovementioned assembly components all involve in<br>
gle link'. This 'single link' is defined as a generalized link<br> *rechnology 32 (7) (2018) 3395-3405* 3401<br> *account of the joint clearance.*<br> *account of the joint clearance.***<br>
<b>***Pherenology 32 (7) (2018) 3395-3405*  $\blacksquare$ <br> **Pherenology areas all involve in all involve in algel li** 

Fig. 9 shows a simply generalized bar consisted of two bars. Without considering the joint clearance, the length of equiva-

$$
l_{e0} = \sqrt{l_1^2 + l_2^2 - 2l_1l_2\cos(\theta - \Delta\theta)}
$$
 (34)

where  $l_1$  and  $l_2$  is the actual length,  $\Delta\theta$  is the constraint error between two bars.

The clearance vector can rotate by 360 degrees. Therefore translate the clearance vector to the end of equivalent link according to the parallelogram rule and as a result the actual where  $l_1$  and  $l_2$  is the actual length,  $\Delta\theta$  is the construent of the equivalent link of the equivalent link with clearance is expressed as 2  $l_e = \sqrt{l_{e0}^2 + r_e^2 - 2l_{e0}r_e \cos \zeta}$  (where  $\zeta$  is the angle of clearanc subassembly. Thus the generalized link probably pos-<br>ses 'length deviation' due to the manufacturing imperfec-<br>in, joint clearance and assembly error as well.<br>Tig. 9 shows a simply generalized bar consisted of two bars.<br>t  $e_{\theta} = \sqrt{l_i^2 + l_2^2 - 2l_i l_2 \cos(\theta - \Delta \theta)}$  (34)<br>  $e^2 = \sqrt{l_i^2 + l_2^2 - 2l_i l_2 \cos(\theta - \Delta \theta)}$  (34)<br>  $e^2 = \sqrt{l_i^2 + l_2^2 - 2l_i l_2 \cos(\theta - \Delta \theta)}$ <br>  $e^2 = \sqrt{l_i^2 + l_2^2 - 2l_i l_2 \cos(\theta - \Delta \theta)}$ <br>  $e^2 = \sqrt{l_i^2 + l_2^2 - 2l_i l_2 \cos(\theta - \Delta \theta)}$  (35)<br>  $e^2 = \sqrt{l_i^2 + l_$ **a**  $V_1$  *l*  $V_2$  **l**  $V_2$  **l**  $V_3$  **l** is the actual length,  $\Delta\theta$  is the constraint between two bars.<br>
the clearance vector can rotate by 360 degrees. Therefore slate the clearance vector to the end of equivalent l  $l_{e0} = \sqrt{l_1^2 + l_2^2 - 2l_1l_2 \cos(\theta - \Delta \theta)}$  (34)<br>
ere  $l_1$  and  $l_2$  is the actual length,  $\Delta \theta$  is the constraint<br>
or between two bars.<br>
The clearance vector can rotate by 360 degrees. Therefore<br>
sushte the clearance vect

$$
l_e = \sqrt{l_{e0}^2 + r_c^2 - 2l_{e0}r_c \cos\varsigma}
$$
 (35)

where  $\zeta$  is the angle of clearance vector in the local frame and  $r<sub>c</sub>$  is the joint clearance. Thus the minimum and maximum value respectively are  $=\sqrt{l_{e0}^2 + r_c^2 - 2l_{e0}r_c \cos \zeta}$ <br>  $= \sqrt{l_{e0}^2 + r_c^2 - 2l_{e0}r_c \cos \zeta}$ <br>  $\therefore$   $\zeta$  is the angle of clearance vector in the local f<br>  $r_c$  is the joint clearance. Thus the minimum and if<br>
value respectively are<br>  $\zeta$ <sub>emin</sub>

$$
\begin{cases}\n l_{\text{emin}} = l_e - r_c \\
 l_{\text{emu}} = l_e + r_c\n\end{cases} \tag{36}
$$

By parity of reasoning, The minimum and maximum equivalent length of a n-bar open-chain generalized bar can be

$$
l_e = \sqrt{l_{eo}^2 + r_c^2 - 2l_{eo}r_c \cos \varsigma}
$$
 (35)  
where  $\varsigma$  is the angle of clearance vector in the local frame  
and  $r_c$  is the joint clearance. Thus the minimum and maxi-  
mum value respectively are  

$$
\begin{cases} l_{\text{min}} = l_e - r_c \\ l_{\text{max}} = l_e + r_c \end{cases}
$$
 (36)  
By parity of reasoning, The minimum and maximum  
equivalent length of a n-bar open-chain generalized bar can be  
derived from Eq. (36) as  

$$
\begin{cases} l_{\text{min}} = |\vec{l}_1 + \vec{l}_2 + \dots + \vec{l}_n| - (n-1)r_c \\ l_{\text{max}} = |\vec{l}_1 + \vec{l}_2 + \dots + \vec{l}_n| + (n-1)r_c \end{cases}
$$
 (37)  

$$
\begin{cases} l_{\text{max}} = |\vec{l}_1 + \vec{l}_2 + \dots + \vec{l}_n| + (n-1)r_c \end{cases}
$$



Fig. 12. Assembly process of over-constrained ESS.

## **3. Algorithm of assembly precision prediction**

In Sec. 2, error models for three assembly components of planar closed-loop mechanisms have been established. At present, we are ready to use those three models within a whole algorithm so as to calculate the position errors generated during the assembly process. This algorithm depicted in Fig. 10 are divided into four steps:

Step 1: Determine the base bar to be the reference of the whole mechanism in assembly process;

Step 2: Judge the operation type of this assembly sequence and then conduct the proper calculations of the position error in term of those three models;

Step 3: Implement the next-sequence assembling operation and continue to repeat the step 2;

Step 4: Recycle the steps 2 and 3 until the mechanism is assembly completely and all the position errors of the points are obtained.

# **4. Results and discussion**

The ESS of the SAR antenna (a planar three-loop mechanism with ten bars), whose structure is presented in Fig. 11, will be the numerical example to demonstrate the utility of the above algorithm. In general the ESS is a structure with redundant constraint in actual assembly process. However for lowering the analysis difficulty, the over-constraint ESS was eversimplified to a statically determinate structure [9]. Now without loss of generality, those two cases both are to be analyzed in this paper.

## *4.1 Over-constraint ESS*

In Fig. 11, joints C, G and H are all 180-degree locked

joints which will guarantee the angle between the two bars connected by them being 180 degrees after locked completely and accordingly there are three generalized links BD, DI and DF. Meanwhile joint A is a 90-degree locked joint which attaches the antenna panel AF to the load cabin and two antenna panels are associated with the 180-degree locked joint F. The whole ESS is over-constraint because the constraints are more than the DOFs ( the ESS has 5 constraints and 3 DOFs ). The concrete structure parameters are presented in Table 1. It should be noted that for simplicity the equivalent lengths and deviations of three generalized bars, which can be evaluated by the Sec. 2.5, are directly given in Table 1. And Table 2 gives the locked angle errors (assembly errors) and the rotation angles of clearance vectors of joints A and F in local framework (note that the rotation angle in local framework is a constant after the joint is locked). Besides all the joint clearances are 0.5 mm.

In the shop floor, the load cabin AB is the reference base and the practical assembly sequences of over-constraint ESS are shown in Fig. 12.

Sequence 1: Assembly the single bar  $A_1F$  (the inner panel) attached to the load cabin by a 90-degree locked joint;

Sequence 2: Connect the generalized bar  $B_1D$  (generated by  $B_1C$  and  $C_1D$ ) and the single bar  $E_1D_1$ ;

Sequence 3: Insert the redundant generalized bar  $F_1D_2$ (formed by  $F_1G_1$  and  $GD_2$ );

Sequence 4: Fix single bar  $F<sub>2</sub>I$  (the outer panel);

Sequence 5: Install the redundant generalized bar  $I_1D_3$ (formed by  $I_1H_1$  and  $H_1D_3$ ).

According to the precision prediction algorithm presented in Fig. 10, the assembling position errors of the endpoints of all links in every aforesaid assembly sequences can be readily evaluated. Note that there is a two-bar connected operation in Sequence 2 and based on the Sec. 2.3, the Jacobian matrixes are equence 1: Assembly the single bar A<sub>1</sub>F (the inner panel)<br>
heled to the load cabin by a 90-degree locked joint;<br>
equence 2: Connect the generalized bar B<sub>1</sub>D (generated by<br>
and C<sub>1</sub>D) and the single bar E<sub>1</sub>D<sub>1</sub>;<br>
equenc nce 1: Assembly the single bar A<sub>1</sub>F (the inner panel)<br>to the load cabin by a 90-degree locked joint;<br>noe 2: Connect the generalized bar B<sub>1</sub>D (generated by<br>C<sub>1</sub>D) and the single bar E<sub>1</sub>D<sub>1</sub>;<br>here 3: Insert the redundant med to the load calm by a 90-dagger ocked joint,<br>equence 2: Connect the generalized bar B<sub>1</sub>D (generated by<br>and C<sub>1</sub>D) and the single bar E<sub>1</sub>D<sub>1</sub>;<br>aquence 3: Insert the redundant generalized bar F<sub>1</sub>D<sub>2</sub><br>requence 3: Inse Let control the given the prediated on  $P_1D_2$  (giodinate of  $P_1D_2$ ) and the single bar  $E_1D_1$ ;<br>  $E_1D_1$  on the single bar  $E_2D_1$ ;<br>  $E_1D_2$  or  $E_1$  (the outer panel);<br>  $E_1D_1$  and  $G_1D_2$ );<br>  $E_1$  and  $G_2D_1$ equence 3: Insert the redundant generalized bar  $F_1D_2$ <br>
med by  $F_1G_1$  and GD<sub>2</sub>);<br>
equence 4: Fix single bar  $F_2I$  (the outer panel);<br>
equence 5: Install the redundant generalized bar  $I_1D_3$ <br>
equence 5: Install the y F<sub>1</sub>G<sub>1</sub> and GD<sub>2</sub>);<br>
ee 4: Fix single bar F<sub>2</sub>I (the outer panel);<br>
ee 5: Install the redundant generalized bar  $I_1D_3$ <br>
y I<sub>1</sub>H<sub>1</sub> and H<sub>1</sub>D<sub>3</sub>).<br>
ing to the precision prediction algorithm presented<br>
the assembling p

$$
\lambda_1 = \begin{bmatrix} 1.739 & -1.781 \\ 0.471 & 0.269 \end{bmatrix}
$$
 (38)

$$
\lambda_2 = \begin{bmatrix} 0.260 & 1.720 \\ -0.07 & 0.465 \end{bmatrix}
$$
 (39)

$$
\lambda_3 = \begin{bmatrix} 0.465 & -1.720 \\ -0.07 & 0.259 \end{bmatrix} . \tag{40}
$$



Fig. 14. Assembly process of statically determinate ESS.





Table 3. Assembly position errors of over-constrained ESS.

Endpoints	Position error (mm)	connect the generalized bar $D_2F$
A	(0,0)	and $D_2E$ , generating a closed-cha
A <sub>1</sub>	(0.433, 0.250)	Sequence 2: Joint generalized
B	(0,1.100)	$B_1C$ and $C_1D$ ); Sequence 3: Associate genera
$\mathbf{B}_1$	$(-0.989, 1.249)$	and $H_1I_1$ ) and the outer panel $F_1I_2$ .
D	$(-0.658, 1.930)$	It's obviously seen that the a
$D_1$	$(-0.388, 3.083)$	determine ESS differs from the
E	$(-0.631, 0.456)$	involves in the equivalence of
$E_1$	$(-1.536,-0.033)$	and two-bar connecting in Sec.
$\overline{F}$	$(-4.019,-0.173)$	errors can be calculated accordi
F <sub>1</sub>	$(-3.769, 0.606)$	algorithm. And because how to
I	(0.790, 0.730)	has been demonstrated entirely in calculating for statically determi
inserted and Fig. 13 gives the equivalent schematic diagram. According to Sec. 2.4, the contact-force matrixes of clearance-		5. Conclusions
	In Sequences 3 and 5, two redundant generalized bars are	
free joints B, E and D applied by $f_1$ are		Assembly precision predict
		mechanisms in view of joint c
		straints is investigated systemati
		assembly process of arbitrary pl
		is described by successive stack
$\begin{cases} \mathbf{s}_{_{1B}} = \begin{bmatrix} -1.611 & 0 \\ 0 & 0.243 \end{bmatrix} \\ \mathbf{s}_{_{1E}} = \begin{bmatrix} 1.611 & 0 \\ 0 & 0.757 \end{bmatrix} \end{cases}$	(41)	components, which are single-li
		and redundant-link inserted com
$\mathbf{s}_{1D} = \begin{bmatrix} 0 & 0 \\ 0 & 1 \end{bmatrix}.$		els of those three components
		among them, the most difficult sion with redundant constraints
And at the same time the contact-force matrixes exerted by		ple of virtual work. Subsequentl
f <sub>2</sub> are calculated as		the assembly precision is derived

$$
\begin{cases}\n\mathbf{s}_{1B} = \begin{bmatrix}\n-1.611 & 0 \\
0 & 0.243\n\end{bmatrix} \\
\mathbf{s}_{1E} = \begin{bmatrix}\n1.611 & 0 \\
0 & 0.757\n\end{bmatrix} \\
\mathbf{s}_{1D} = \begin{bmatrix}\n0 & 0 \\
0 & 1\n\end{bmatrix}.\n\end{cases}
$$
\n(41)

And at the same time the contact-force matrixes exerted by **f**, are calculated as

$$
\begin{cases}\n\mathbf{s}_{2B} = \begin{bmatrix}\n-1.44 & 0 \\
0 & 0.51\n\end{bmatrix} & \text{propto} \\
\mathbf{s}_{2E} = \begin{bmatrix}\n0.44 & 0 \\
0 & 0.49\n\end{bmatrix} & \text{or} \\
\mathbf{s}_{2D} = \begin{bmatrix}\n0.92 & 0 \\
0 & 0.39\n\end{bmatrix}.\n\end{cases}
$$
\n(42)

Substitute those Jacobian matrixes, contact-force matrixes and the related parameters into the error models and accordingly the precision can be predicted in assembly process so as to guide the workers to adjust the accuracy in assembly shop of the ESS. In addition the results of all position errors are presented in Table 3.

#### *4.2 Statically determinate ESS*

Different from the over-constrained ESS, the 90- and 180 degree locked joints of A and F of the statically determine ESS as shown in Fig. 14 become free for the purpose of sim plicity. The assembly sequences for the statically determine ESS shown in Fig. 14 are elaborated as following:

Sequence 1: Let the inner panel  $A_1F$  be the base bar and connect the generalized bar  $D_2F_2$  (formed by  $D_2G$  and  $G_1F_2$ ) and  $D_2E$ , generating a closed-chain generalized bar  $A_1D_2$ ;

Sequence 2: Joint generalized links  $A_1D_2$  and  $B_1D$  (fixed by  $B_1C$  and  $C_1D$ );

Sequence 3: Associate generalized bar  $D_3I_1$  (locked by  $D_3H$ and  $H_1I_1$ ) and the outer panel  $F_1I_1$ .

It's obviously seen that the assembly process of statically determine ESS differs from the over-constrained one and just involves in the equivalence of generalized links in Sec. 2.5 and two-bar connecting in Sec. 2.3. Similarly the assembling errors can be calculated according to the precision prediction algorithm. And because how to evaluate the position errors has been demonstrated entirely in Sec. 4.1, the concrete error calculating for statically determine ESS will not be displayed once again in case of repetition.

#### **5. Conclusions**

Dr. (a.  $\frac{1}{1}$  (b.  $\frac{1}{2}$  (a.  $\frac$ E<br>
E<br>  $\frac{F}{F_1}$  (-0.631,0.456)<br>  $\frac{F_1}{F_2}$  (-1.356,0.003)<br>  $\frac{F_2}{F_3}$  (-3.769,0.606)<br>  $\frac{F_1}{F_2}$  (-3.769,0.606)<br>  $\frac{F_2}{F_3}$  (-3.769,0.606)<br>  $\frac{F_1}{F_2}$  (-3.769,0.606)<br>  $\frac{F_1}{F_3}$  (-3.769,0.606)<br>  $\frac{F_1$ D<sub>1</sub> (-0.388,3.083)<br>
E<br>
E<br>
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Timolecular mode because however connecting in Sec<br>
E<br>
(-1.536,-0.033)<br>
Timolecular mo (41) components, which are single-link fixed, two-link connected Sequences 3 and 5, two redundant generalized bars are<br>
are again in case of repetition.<br>
ed and Fig. 13 gives the equivalent schematic diagram.<br>
Thing to Sec. 2.4, the contact-force matrixes of clearance<br>  $u = \begin{bmatrix} -1.611 &$ ces 3 and 5, two redundant generalized bars are<br>
Fig. 13 gives the equivalent schematic diagram.<br>
Sec. 2.4, the contact-force matrixes of clearance-<br>
E and D applied by  $f_1$  are<br>  $(611 \t 0)$ <br>  $(11 \t 0)$ <br>  $(12 \t 0)$ <br>  $(13$ d Fig. 13 gives the equivalent schematic diagram.<br>
0.5. **Conclusions**<br>
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1.611 9 Discontinue of the ESS of the SKR of the Haussian space of the ESS of the SKR of the decobian matrixes, contact-force matrixes contact of the Haussian space of and redundant-link inserted by two consisting and redundant  $\begin{bmatrix}\n1.611 & 0 \\
0 & 0.243\n\end{bmatrix}$ <br>  $\begin{bmatrix}\n1.611 & 0 \\
0 & 0.757\n\end{bmatrix}$ <br>  $\begin{bmatrix}\n0 & 0 \\
0 & 1\n\end{bmatrix}$ .<br>  $\begin{bmatrix}\n0 &$  $[1.611 \t 0] \t 1.611 \t 0 \t 0 \t 0.243]$ <br>  $[0 \t 0]$   $[0 \t 1]$   $[0 \t 0]$   $[0 \t 1]$   $[0 \t 0]$   $[0 \t 0]$  In Sequences 3 and 5, two redundant generalized bars are once again in case of repetition.<br>
Exterd and Fig. 13 gives the equivalent schematic diagram.<br>
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cording to Sec. 2.4, the contact-force matrixes of cleara In Sequences 3 and 5, two redundant generalized bars are<br>
erted and Fig. 13 gives the equivalent schematic diagram.<br>
colding to Sec. 2.4, the contact-force matrixes of clearance-<br>
be joints B, E and D applied by f, are<br> ered and Hig, 13 gives the equivalent schematic dagram.<br>
Scoredusions<br>
ejoints B, E and D applied by f, are<br>  $\begin{bmatrix}\ns_{10} = \begin{bmatrix}\n-1.611 & 0 \\
0 & 0.243\n\end{bmatrix}\n\end{bmatrix}$ <br>  $\begin{bmatrix}\ns_{11} = \begin{bmatrix}\n-1.611 & 0 \\
0 & 0.243\n\end{bmatrix}\n\end{bmatrix}$ <br>  $\$ Found 5 particle 1.611 0<br>  $\begin{bmatrix}\ns_{10} = \begin{bmatrix}\n1.611 & 0 \\
0 & 0.243\n\end{bmatrix}\n\end{bmatrix}$ <br>  $\begin{bmatrix}\ns_{10} = \begin{bmatrix}\n1.611 & 0 \\
0 & 0.757\n\end{bmatrix}\n\end{bmatrix}$ <br>  $\begin{bmatrix}\ns_{11} = \begin{bmatrix}\n1.611 & 0 \\
0 & 0.757\n\end{bmatrix}\n\end{bmatrix}$ <br>  $\begin{bmatrix}\ns_{12} = \begin{bmatrix}\n1.611 & 0 \\
0 & 0$ Assembly precision prediction for planar closed-loop mechanisms in view of joint clearance and redundant con straints is investigated systematically in this paper. Firstly the assembly process of arbitrary planar closed-loop mechanisms is described by successive stack of three proposed assembly and redundant-link inserted components. Then the error models of those three components are established and chief of among them, the most difficult problem of assembling precision with redundant constraints is solved based on the principle of virtual work. Subsequently the prediction algorithm of the assembly precision is derived from the combination of the abovementioned error models and moreover calculating the position errors of each assembling sequence is realized for planar closed-loop mechanisms by this algorithm. Finally the prediction algorithm is verified by evaluating the assembly error of the ESS of the SAR antenna.

Compared with the previous methods of accuracy analysis, the advantages of the algorithm proposed in this paper is not only considering all factors influencing the precision including manufacturing deviation, joint clearance and redundant con straints but also available to all planar closed-loop mecha nisms. Furthermore the modeling procedures and relevant methods can be expanded to accuracy analysis of spatial closed-loop mechanisms. Therefore our subsequent research will focus more on exploring the algorithm of precision prediction for spatial closed-loop mechanisms.

But meanwhile it should be noted that the assumption in Sec. 2.4 for solving the problem of inserting a redundant link is adequately true only when the components in the mecha nism possess good stiffness, indicating the obtained results of position errors based on the above assumption will become inaccurate if stiffness of the links is too small. Hence, for utmost exactitude and covering all closed-loop mechanisms with different flexibility, our follow-up work will also concentrate on the error-deformation coordination problem in view of deformation existing in all bars apart from extending the methods of this paper to the spatial mechanisms.

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