

Thermo-mechanical vibration analysis of a single-walled carbon nanotube embedded in an elastic medium based on higher-order shear deformation beam theory†

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Abstract

In this study, the thermal effect on the free vibration characteristics of embedded Single-walled carbon nanotubes (SWCNTs) based on the size-dependent Reddy higher order shear deformation beam theory subjected to in-plane thermal loading is investigated by presenting a Navier-type solution and employing a semi-analytical Differential transform method (DTM) for the first time. In addition, the exact nonlocal Reddy beam theory solution presented here should be useful to engineers designing nanoelectromechanical devices. The smallscale effect is considered based on nonlocal elasticity theory of Eringen. The nonlocal equations of motion are derived through Hamilton's principle, and they are solved by applying DTM. Numerical results reveal that the proposed modeling and semi-analytical approach can provide more accurate frequency results of the SWCNTs compared to analytical results and some cases in the literature. The detailed mathematical derivations are presented, and numerical investigations are performed, whereas emphasis is placed on investigating the effect of several parameters such as small-scale effects, boundary conditions, mode number, thickness ratio, temperature change, and Winkler spring modulus on the natural frequencies of the SWCNTs in detail. The vibration behavior of SWCNTs is significantly influenced by these effects. Results indicate that the inclusion of size effect results in a decrease in nanobeam stiffness and leads to a decrease in natural frequency. Numerical results are presented to serve as benchmarks for future analyses of SWCNTs.

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Keywords: Differential transformation method; Nonlocal elasticity; Reddy beam theory; SWCNT; Thermo-mechanical vibration

1. Introduction

Nanoscale engineering materials have attracted great interest in modern science and technology after the invention of Carbon nanotubes (CNTs) by Iijima [1]. They are characterized by significant mechanical, thermal, and electrical performances that are superior to conventional structural materials. In recent years, nanobeams and CNTs hold a wide variety of potential applications [2, 3] such as sensors, actuators, transistors, probes, and resonators in Nano-electro-mechanical system (NEMS). Continuum mechanics offers an easy and useful tool for the analysis of CNTs because conducting experiments at the nanoscale is a daunting task and atomistic modeling is restricted to small-scale systems owing to computer resource limitations. However, classical continuum models need to be extended to consider the nanoscale effects, and this extension can be achieved by nonlocal elasticity theory proposed by Eringen [4], which considers the sizedependent effect.

Furthermore, with the development of material technology,

Single-walled carbon nanotubes (SWCNTs) have been employed in Micro/Nano-electro-mechanical systems (MEMS/ NEMS) [5]. Understanding their mechanical properties and vibration behavior is significant in their design and manufacture because of the high sensitivity of these systems to external stimulations. Thus, establishing an accurate model of nanobeams is a key issue for successful NEMS design. Peddieson et al. [6] proposed a version of nonlocal elasticity theory employed to develop a nonlocal Euler beam model. The thermo-mechanical vibration of short CNTs embedded in Pasternak foundation was presented by Amirian et al. [7] based on nonlocal elasticity theory. They also concluded that the high temperature change and nonlocal parameter play important roles in the vibrational response of the CNTs. Murmu and Pradhan [8] used Euler-Bernoulli beam theory (EBT) to investigate thermo-mechanical vibration analysis of SWCNTs embedded in an elastic medium. They noticed that small-scale effects are significant in the vibration response of a SWCNT. Zhang et al. [9] studied the thermal effect on the vibration of double-walled CNTs based on thermal elasticity mechanics and nonlocal elasticity theory. Wang et al. [10] exploited the thermal effect on the vibration and instability of conveying fluid SWCNTs based on the Euler beam. Through

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the Galerkin method, the transverse vibration of embedded SWCNTs with light waviness based on nonlocal Euler- Bernoulli and Timoshenko beam models is studied by Soltani et al. [11]. Most of the recent studies on SWCNTs have been conducted based on EBT and Timoshenko beam theory (TBT). However, the effects of rotary inertia and shear deformation are neglected in EBT, so EBT always overestimates buckling load and natural frequency of free vibration and underestimates deflection. Moreover, in TBT, a shear correction factor is required to compensate for the difference between the actual stress state and the constant stress state. In avoiding the use of shear correction factor and obtaining a better prediction of the response of a deep beam, many higher-order shear deformation theories have been developed, such as the third-order where $t_{ij}(x')$ is the classical macroscopic stress tensor at presents a more realistic model of the beam in the determination of higher modes of vibration. To the author's best knowledge, no work has reported on the application of the Differential transform method (DTM) on the thermo-mechanical vibration analysis of embedded SWCNTs within the framework of Reddy higher order beam theory (RHOBT). In this study, the non-classical third-order shear deformation beam model is developed, and governing equations and boundary conditions tive Eqs. (1) and (2) are complicated: for the free vibration of a nonlocal embedded SWCNT have been derived through Hamilton's principle. A new semi analytical method called DTM is employed for the vibration analysis of size-dependent SWCNTs with four combinations of boundary conditions for the first time. The superiority of where ∇^2 is the Laplacian operator. The scale length $e_0 a$ the DTM is found in its simplicity and good precision and depends on the Taylor series expansion, and it takes less time The nonlocal parameter $e_0 a$ is experimentally obtained for to solve polynomial series. With this method, obtaining highly accurate results or exact solutions for differential equations is possible. The detailed mathematical derivations are presented, and numerical investigations are performed, whereas empha sis is placed on investigating the effects of several parameters such temperature change, mode number, thickness ratios, boundary conditions, Winkler spring modulus, and small-scale parameter on the vibration characteristics of SWCNTs. Com parisons of the results with those from the existing literature are provided, and the good agreement between the two sets of results validated the presented approach.

2. Theory and formulation

2.1. Nonlocal elasticity model for SWCNT

In the classic elastic continuum theory, the stress field at a point *x* only depends on the strain field at the same point. However, according to Eringen's nonlocal elasticity theory [13], the stress field at a point is dependent on the strains at all other points in the body. Therefore, the nonlocal stress ten sor σ_i at point *x* is defined by the following: the temperature change, mode unuber, thickness ratios,

independent on the vibration characteristics of SWCNTs. Com-

intersection the existing literature

provided, and the good agreement between the two sets of

provide

$$
\sigma_{ij}(x) = \int_{\Omega} \alpha(|x'-x|, \tau) t_{ij}(x') d\Omega(x')
$$
 (1)

Fig. 1. Schematic of embedded SWCNT.

$$
t_{ij} = C_{ijkl} \varepsilon_{kl} \tag{2}
$$

shear deformation theory proposed by Reddy [12], which point x, the kernel function $\alpha(|x'-x|, \tau)$ is the nonlocal point *x* , the kernel function a the same point $x \cdot x + x \cdot y = x + y + z + z$
 point $\sum_{k=1}^{n} x + y \cdot z = 1$
 $\sum_{k=1}^{n} x + y \cdot z = 1$
 $\sum_{k=1}^{n} x + y \cdot z = 1$
 $\sum_{k=1}^{n} x + z \cdot z = 1$
 $\sum_{k=1}^{n} x + z \cdot z = 1$
 $\sum_{k=1}^{n} x + z \cdot z = 1$
 \sum_{k modulus, *x x* ¢ - is the distance, and t is a material constant that depends on type of material. t_{ij} is the macroscopic stress tensor at a point *x* in a Hookean solid, which is dependent on the strain ε at the same point according to generalized Hooke's law. *C* is the fourth-order elasticity tensor. A simplified equation of differential form is used as a basis of nonlocal constitutive formulation because solving the integral constituwhere $t_{ij}(x')$ is the classical macroscopic stress tensor at
point x, the kernel function $\alpha(|x'-x|, \tau)$ is the nonlocal
modulus, $|x'-x|$ is the distance, and τ is a material constant
that depends on type of material. t *x* $V_{\text{unkler medium}}$
 $t_{ij} = C_{ijkl} \varepsilon_{kl}$ (2)
 $\text{here } t_{ij}(x')$ is the classical macroscopic stress tensor at
 x , the kernel function $\alpha(\vert x' - x \vert, \tau)$ is the nonlocal

dudulus, $\vert x' - x \vert$ is the distance, and τ is a material $t_{ij} = C_{ijkl} \varepsilon_{kl}$ (2)

where $t_{ij}(x')$ is the classical macroscopic stress tensor at

probut x, the kernel function $\alpha(|x'-x|, \varepsilon)$ is the nonlocal

modulus, $|x'-x|$ is the distance, and τ is a material constant

that d ϵ_{ij} $\epsilon_{ijk} \epsilon_{kl}$
where $t_{ij}(x')$ is the classical macroscopic stress tensor at
point x, the kernel function $\alpha(|x'-x|, \tau)$ is the nonlocal
modulus, $|x'-x|$ is the distance, and r is a material constant
then depends on ty (scopic stress tensor at $x|x|$, τ) is the nonlocal τ is a material constant
the macroscopic stress
which is dependent on
cording to generalized
sticity tensor. A simpli-
d as a basis of nonlocal
ng the integral cons

$$
(1 - (e_0 a)^2 \nabla^2) \sigma_{kl} = t_{kl}
$$
 (3)

considers the size effect on the response of nanostructures. various materials. Thus, in the present study, a conservative estimate of the nonlocal parameter $(e_0a)^2$ for SWCNT is considered in the range of 0-4 (nm)^2 [8]. For an elastic material, the nonlocal constitutive relations may be simplified as the following: *C* is the fourth-order elasticity tensor. A simpli-

of differential form is used as a basis of nonlocal

rmulation because solving the integral constitu-
 (2) are complicated:
 $\frac{2}{\sqrt{K}} = t_{kl}$ (3)

(3)

are Laplacia *xx* **ke's** law. *C* is the fourth-order elasticity tensor. A simpli-
equation of differential form is used as a basis of nonlocal
titutive formulation because solving the integral constitu-
etgs. (1) and (2) are complica oke's law. *C* is the fourth-order elasticity tensor. A simpli-
dequation of differential form is used as a basis of nonlocal
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Eqs. (1) and (2) are complicated:
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s. (1) and (2) are complicated:
 $(e_0 a)^2 \nabla^$ Institutive formulation because solving the integral constitu-

Eqs. (1) and (2) are complicated:

(1 – $(e_0a)^2 \nabla^2$) $\sigma_{kl} = t_{kl}$ (3)

(3)

ere ∇^2 is the Laplacian operator. The scale length $e_0 a$

is isolers th tutive formulation because solving the integral constitu-

[s, (1) and (2) are complicated:
 $(e_0a)^2 \nabla^2$) $\sigma_{kl} = t_{kl}$ (3)
 ∇^2 is the Laplacian operator. The scale length $e_0 a$
 ∇^2 is the Laplacian operator. (3)
 $\mathbf{v}_{\theta}(\mathbf{a}) \times \mathbf{b}\sigma_{kl} = t_{kl}$ (3)
 $\mathbf{v}_{\theta}^2(\mathbf{a}) \times \mathbf{b}\sigma_{kl}$ is experimentally obtained for

materials. Thus, in the present study, a conservative

of the nonlocal parameter $e_0 a$ is experimentally obtain where ∇^2 is the Laplacian operator. The scale length $e_0 a$
considers the size effect on the response of nanostructures.
The nonlocal parameter $e_0 a$ is experimentally obtained for
various materials. Thus, in the pr

$$
\sigma_{xx} - \mu \frac{\partial^2 \sigma_{xx}}{\partial x^2} = E(\varepsilon_{xx} - \alpha_x T) \tag{4}
$$

$$
\sigma_{xz} - \mu \frac{\partial^2 \sigma_{xz}}{\partial x^2} = G \gamma_{xz}
$$
 (5)

where σ and ε are the nonlocal stress and strain, respectively, $\mu = (e_0 a)^2$ is the nonlocal parameter, E is the elasticis Poisson's ratio. α_x and *T* are the thermal expansion and temperature change, respectively.

2.2 Kinematic relations

 (1) axis is considered in the width direction. The SWCNT vi-As shown in Fig. 1, a SWCNT is modeled as a higher order beam with length *L* and circular cross-section of radius *^R*, where the *^x* axis is taken along the central axis, and the *^z* brates only in the *x-z* plane. Based on RHOBT, the displacement of an arbitrary point in the beam along the *x*- and *z*-axes, *F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 3797~3803*

ment of an arbitrary point in the beam along the x- and z-axes,

denoted by $\tilde{U}(x, z, t)$ and $\tilde{W}(x, z, t)$, respective lowing [14]: *F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 3797-3803*

2 (of an arbitrary point in the beam along the x- and z-axes,
 $w = 0$

12 (et by $\overline{U}(x, z, t)$ and $\overline{W}(x, z, t)$, respe *F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 3797-3803*
 z and *W U*(*x,z,t*) and *W*(*x,z,t*), respectively, is the fol-
 $\frac{4}{3h^2}P = 0$ or $\frac{\partial w}{\partial x} = 0$.
 $\frac{d}{dx}I(2x,$ *F. Ebrahimi and E. Salari* / Journal of Mechanical Science and Technology 29 (9) (2015) 379.

arbitrary point in the beam along the x- and z-axes,
 $\vec{U}(x, z, t)$ and $\vec{W}(x, z, t)$, respectively, is the fol-
 $= z\varphi(x, t) - \$

$$
\tilde{U}(x, z, t) = z\varphi(x, t) - \frac{4z^3}{3h^2} \left(\varphi(x, t) + \frac{\partial w(x, t)}{\partial x} \right)
$$
\nThe integral of the integral of the integral:

\n(6) The integral of the integral of the integral:

$$
\tilde{W}(x, z, t) = w(x, t) \tag{7}
$$

F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 3:

and first point in the beam along the x- and z-axes,
 $w = 0$

onoted by $\bar{U}(x, z, t)$ and $\bar{W}(x, z, t)$, respectively, is the fo where t is time, *w* and φ are the transverse displacement and angular displacement of the cross-sections at any point on the neutral axis of the beam, respectively. Based on Eqs. (6) and (7), the nonzero strains of the Reddy beam theory can be obtained as the following: *xx z w ^z x x h x F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 3797*

arbitrary point in the beam along the x- and z-axes,
 $V = 0$
 $\frac{V}{3h^2}P = 0$ or $\frac{\partial w}{\partial x} = 0$.
 \therefore
 $= z\varphi(x,t) - \frac{4z^3}{3$ *F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 3797-3803*

and for an arbitrary point in the beam along the x- and z-axes,

w = 0

or details by $U(x, z, t)$ and $\hat{W}(x, z, t)$, respec *F. Ebrahimi* and *E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 3797-3803*

f an arbitrary point in the beam along the x- and z-axes, $w = 0$

d by $\overline{U}(x, z, t)$ and $\overline{W}(x, z, t)$, respectively, to f an arbitrary point in the beam along the x- and z-axes,

y = 0

ded by $U(x, z, t)$ and $\hat{W}(x, z, t)$, respectively, is the fol-
 $\frac{4}{3h^2}P = 0$ or $\frac{\partial w}{\partial x} = 0$.
 $\frac{4}{3h^2}(x, z, t) = z\varphi(x, t) - \frac{4z^3}{\partial t^2}(\varphi(x, t) + \frac$ ent of an arbitrary point in the beam along the x- and z-axes,

where the by $\vec{U}(x, z, t)$ and $\vec{W}(x, z, t)$, respectively, is the fol-
 $\vec{U}(x, z, t) = z\varphi(x, t) - \frac{4z^3}{3h^2} \left(\varphi(x, t) + \frac{\partial w(x, t)}{\partial x} \right)$

(6) The integration a anomathaly point in the beam along the x- and z-axes,
 $x_1x_2y_1 = x_0(x, t)$ and $\overline{W}(x, z, t)$, respectively, is the fol-
 $\frac{4}{3h^2}P = 0$ or $\frac{\partial w}{\partial x} = 0$.
 $\frac{4}{3h^2}(14)$:
 $\frac{4}{3h^2}(14)$:
 $\frac{4}{3h^2}(14)$:
 $\frac{4$

$$
\varepsilon_{xx} = z \frac{\partial \varphi}{\partial x} - \frac{4z^3}{3h^2} \left(\frac{\partial \varphi}{\partial x} + \frac{\partial^2 w}{\partial x^2} \right) \tag{8}
$$

$$
\gamma_{xz} = (\varphi + \frac{\partial w}{\partial x})(1 - \frac{4z^2}{h^2}).
$$
\n(9)

In defining the nonlocal stress resultants, *M*, *P*, *Q*, and *R* are defined as the following:

$$
M = \int_{A} z \sigma_{xx} dA, \quad P = \int_{A} z^{3} \sigma_{xx} dA
$$

\n
$$
Q = \int_{A} \sigma_{xx} dA, \quad R = \int_{A} z^{2} \sigma_{xx} dA
$$

\n(10)
$$
+ \frac{8GA}{15}
$$

\n16*a*

where σ_{xx} and σ_{xz} are the normal and shear stresses, respectively. Through Eqs. (8)-(10), based on Hamilton's principle [15] and nonlocal elasticity theory, the governing equations are obtained as the following:

$$
Q = \int_{A} \sigma_{xx} dA, \qquad R = \int_{A} z^{2} \sigma_{xx} dA
$$
\n
$$
= \int_{A} \sigma_{xx} dA, \qquad R = \int_{A} z^{2} \sigma_{xx} dA
$$
\n
$$
= \int_{105} \frac{1}{\theta^{2}} \sigma_{xx} dA
$$
\n
$$
= \int
$$

where K_w is the Winkler modulus and N_{Temp} is the thermally $\frac{3}{2}$ induced axial force derived from the constitutive relationship between the thermal strain and thermal stress as follows [15]:

$$
N_{Temp} = -EA\alpha_x T \tag{13}
$$

Furthermore, the following boundary conditions at the edges of the nanobeam ($x = 0, L$) are obtained as follows:

$$
\frac{4}{3h^2}P - M = 0 \quad \text{or} \quad \varphi = 0 \tag{14}
$$
\n
$$
\frac{-4}{3h^2} \frac{\partial P}{\partial x} + \frac{4}{h^2}R - Q - \frac{\rho I}{21} \frac{\partial^3 w}{\partial t^2 \partial x} + \frac{16\rho I}{105} \frac{\partial^2 \varphi}{\partial t^2} = 0 \quad \text{or}
$$

$$
w = 0 \tag{15}
$$

$$
\frac{4}{3h^2}P = 0 \quad \text{or} \quad \frac{\partial w}{\partial x} = 0 \tag{16}
$$

 $\frac{\partial \varphi(x,t) - \frac{\partial^2 f}{\partial x^2} \varphi(x,t) + \frac{\partial^2 f}{\partial x}}{n}$ (b) nonlocal elasticity for a nanobeam may be written as the fol-The integration of Eqs. (4) and (5) constitutive equations of lowing: 29 (9) (2015) 3797-3803 3799
 $\frac{\partial w}{\partial x} = 0$. (15)

of Eqs. (4) and (5) constitutive equations of

for a nanobeam may be written as the fol-
 $\frac{\partial L}{\partial x}$ (4 $\frac{\partial \varphi}{\partial x} - \frac{\partial^2 w}{\partial x^2}$) (17a)
 $\frac{4h^4}{80}$ (16 $\frac{\partial \$ *xe* and Technology 29 (9) (2015) 3797-3803 3799
 $w = 0$
 $\frac{4}{3h^2}P = 0$ or $\frac{\partial w}{\partial x} = 0$. (16)

The integration of Eqs. (4) and (5) constitutive equations of

alocal elasticity for a nanobeam may be written as the fo and Technology 29 (9) (2015) 3797-3803

= 0 (15)
 $\frac{1}{2}P = 0$ or $\frac{\partial w}{\partial x} = 0$. (16)

integration of Eqs. (4) and (5) constitutive equations of

cal elasticity for a nanobeam may be written as the fol-
 $-\mu \frac{\partial^2 M}{\partial x$ gy 29 (9) (2015) 3797~3803 3799

(15)
 $\frac{\partial w}{\partial x} = 0$ (16)

on of Eqs. (4) and (5) constitutive equations of

tity for a nanobeam may be written as the fol-
 $= \frac{EI}{5} (4 \frac{\partial \varphi}{\partial x} - \frac{\partial^2 w}{\partial x^2})$ (17a)
 $= \frac{E A h^4}{1680$ *P EAh ^w* and Technology 29 (9) (2015) 3797-3803 3799

= 0 (15)
 $\frac{1}{t^2}P = 0$ or $\frac{\partial w}{\partial x} = 0$. (16)

e integration of Eqs. (4) and (5) constitutive equations of

cold elasticity for a nanobeam may be written as the fol-
 $-\mu \$ 29 (9) (2015) 3797-3803 3799

(15)
 $\frac{\partial w}{\partial x} = 0$. (16)

of Eqs. (4) and (5) constitutive equations of

for a nanobeam may be written as the fol-
 $\frac{EI}{5}(4\frac{\partial \varphi}{\partial x} - \frac{\partial^2 w}{\partial x^2})$ (17a)
 $\frac{4h^4}{5}(16\frac{\partial \varphi}{\partial x}$ *ge and Technology 29 (9) (2015) 3797-3803* 3799
 $w = 0$ (15)
 $\frac{4}{3h^2}P = 0$ or $\frac{\partial w}{\partial x} = 0$. (16)

The integration of Eqs. (4) and (5) constitutive equations of

alocal elasticity for a nanobeam may be written as t and Technology 29 (9) (2015) 3797-3803 3799
 $= 0$ (15)
 $\frac{1}{t^2}P = 0$ or $\frac{\partial w}{\partial x} = 0$. (16)

e integration of Eqs. (4) and (5) constitutive equations of

ceal elasticity for a nanobeam may be written as the fol-
 $-\$

$$
M - \mu \frac{\partial^2 M}{\partial x^2} = \frac{EI}{5} (4 \frac{\partial \varphi}{\partial x} - \frac{\partial^2 w}{\partial x^2})
$$
 (17a)

$$
P - \mu \frac{\partial^2 P}{\partial x^2} = \frac{EAh^4}{1680} (16 \frac{\partial \varphi}{\partial x} - 5 \frac{\partial^2 w}{\partial x^2})
$$
 (17b)

$$
Q - \mu \frac{\partial^2 Q}{\partial x^2} = \frac{2GA}{3} (\varphi + \frac{\partial w}{\partial x})
$$
 (17c)

$$
R - \mu \frac{\partial^2 R}{\partial x^2} = \frac{2GI}{5} (\varphi + \frac{\partial w}{\partial x})
$$
 (17d)

(15)
 $\frac{\partial w}{\partial x} = 0$ (16)

(16)
 $\frac{\partial w}{\partial x} = 0$ (16)

(16)

(16)

(16)

(16)

for a nanobeam may be written as the fol-
 $\frac{EI}{5}(4\frac{\partial \varphi}{\partial x} - \frac{\partial^2 w}{\partial x^2})$ (17a)

(17a)

(17a)

(17b)

(17b)

(30)

(17b)

(4)

(4)

(4 *Re and Technology 29 (9) (2015) 3797-3803* 3799
 W = 0 (15)
 $\frac{4}{3h^2}P = 0$ or $\frac{\partial w}{\partial x} = 0$. (16)

The integration of Eqs. (4) and (5) constitutive equations of

flocal elasticity for a nanobeam may be written as and Technology 29 (9) (2015) 3797-3803

= 0 (15)
 $\frac{1}{r^2}P = 0$ or $\frac{\partial w}{\partial x} = 0$. (16)

e integration of Eqs. (4) and (5) constitutive equations of

coll elasticity for a nanobeam may be written as the fol-

ig:
 $-\mu \$ As a final point, the nonlocal governing equations in terms of the displacement can be written by substituting *M*, *P*, *^Q*, and *^R* from Eq. (17) into Eqs. (11) and (12) as follows:

*E. Erichoint and E. Salart / Aamard of Methodanical Science and Techorology 29 (9) (2015) 3797-3803
\nnot of an arbitrary point in the beam along the x- and z-axes,
\n
$$
W^2(x,z,t) = \exp(x,t) - \frac{4z^3}{3h^2} \left(\varphi(x,t) + \frac{\partial w(x,t)}{\partial x} \right)
$$
\n
$$
U(x,z,t) = \exp(w,t) - \frac{4z^3}{3h^2} \left(\varphi(x,t) + \frac{\partial w(x,t)}{\partial x} \right)
$$
\n
$$
U(x,z,t) = w(x,t)
$$
\n
$$
W(x,z,t) = w(x,t)
$$
\n
$$
W(x,z) = w(x,t)
$$
\n
$$
W(x) = \frac{w^2}{6x^2} - \frac{2G}{3} \left(\frac{d^2w}{6x^2} - \frac{G^2w}{6x^2} \right)
$$
\n
$$
V(x,z,t) = w(x,t)
$$
\n
$$
V(x,z) = \frac{w^2}{6x^2} - \frac{3G}{5} \left(\frac{d^2w}{6x^2} - \frac{G^2w}{6x^2} \right)
$$
\n
$$
V(x,z,t) = w(x,t)
$$
\n
$$
W(x,z) = \frac{w^2}{6x^2} - \frac{3G}{5} \left(\frac{d^2w}{6x^2} - \frac{G^2w}{6x^2} \right)
$$
\n
$$
W(x,z) = \frac{w
$$*

3. Solution method

3.1 Analytical solution

 $\frac{3h^2}{3h^2} \frac{\partial x^4}{\partial x^2} + \frac{8x}{6h} \frac{\partial x}{\partial x} + N_{Temp} \frac{\partial x^2}{\partial x^2}$ (12) $+ \rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{21} \frac{\partial^3 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{105} \frac{\partial^2 \phi}{\partial t^2 \partial x}$
 $- K_w w = \rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{21} \frac{\partial^3 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{10$ $\frac{\partial^2 P}{\partial x^2} + \frac{\partial Q}{\partial x} - \frac{4}{h^2} \frac{\partial R}{\partial x} + N_{r\omega\omega} \frac{\partial^2 w}{\partial x^2}$
 $\int = \rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{2l} \frac{\partial^2 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{2l} \frac{\partial^2 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{2l} \frac{\partial^2 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{2l} \frac{\partial^2 w}{\partial t^2 \partial x^2}$
 $\frac{h^2}{h^2} \frac{\partial x^3}{\partial x^4} + \frac{x}{\partial x} - \frac{h^2}{h^2} \frac{\partial x}{\partial x} + N_{Temp} \frac{\partial x}{\partial x^2}$
 $K_w w = \rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{2I} \frac{\partial^4 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{105} \frac{\partial^2 \phi}{\partial t^2 \partial x^2}$
 $K_w w = \rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{2I} \frac{\partial^4 w}{\partial t^2 \partial x^2} + \frac{1$ $\frac{4}{3h^2} \frac{\partial^2 P}{\partial x^2} + \frac{\partial Q}{\partial x} - \frac{4}{h^2} \frac{\partial R}{\partial x} + N_{Temp} \frac{\partial^2 w}{\partial x^2}$
 $-\frac{1}{2h} \frac{\partial^2 w}{\partial x^2} + \frac{16\rho I}{105} \frac{\partial^2 w}{\partial t^2} - \frac{16\rho V}{21} \frac{\partial^2 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{105} \frac{\partial^2 w}{\partial t^2} + \frac{16\rho I}{105} \frac{\partial^2 w}{\partial t^2 \partial x}$ ²+ $\frac{\partial Q}{\partial x} = \frac{4}{h^2} \frac{\partial R}{\partial x} + N_{F\theta\theta} \frac{\partial^2 w}{\partial x^2}$ (12) $+ \rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{21} \frac{\partial^2 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{105} \frac{\partial^3 \phi}{\partial t^2 \partial x^2}$
 $\rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{21} \frac{\partial^4 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{105} \frac{\partial^3 \phi}{$ $\frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial x} - \frac{\rho I}{h^2} \frac{\partial^2 w}{\partial x^2} + \frac{Nr_{low}}{16\rho I} \frac{\partial^2 \phi}{\partial t^2 \partial x^2}$
 $w = \rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{21} \frac{\partial^2 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{105} \frac{\partial^2 \phi}{\partial t^2 \partial x^2}$
 $w = \rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{21} \frac{\partial^2 w}{\partial t^2$ Here, based on the Navier-type solution method for the free vibration of a simply supported SWCNT, the displacement functions are expressed as the product of undetermined coefficients and known trigonometric functions to satisfy the gov erning equations and the conditions at $x = 0$, *L*. The following displacement fields are assumed to be of the form: $\mu \left(\frac{\rho I}{21} \frac{\partial^6 w}{\partial t^2 \partial x^4} - \rho A \frac{\partial^4 w}{\partial t^2 \partial x^2} - \frac{16\rho I}{105} \frac{\partial^5 \phi}{\partial t^2 \partial x^3} \right)$ (19)
 $\rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{21} \frac{\partial^4 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{105} \frac{\partial^3 \phi}{\partial t^2 \partial x}$

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ressed as the product of undetermined coeffi-

i** trigonometric functions to satisfy the gov-

and the conditions a $=\mu \left(\frac{\rho I}{21} \frac{\partial^6 w}{\partial t^2 \partial x^4} - \rho A \frac{\partial^4 w}{\partial t^2 \partial x^2} - \frac{16\rho I}{105} \frac{\partial^5 \varphi}{\partial t^2 \partial x^3} \right)$ (19)
 $+\rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{21} \frac{\partial^4 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{105} \frac{\partial^3 \varphi}{\partial t^2 \partial x}$
 Solution method
 Analytical solu $\rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{21} \frac{\partial^4 w}{\partial t^2 \partial x^2} + \frac{16 \rho I}{105} \frac{\partial^3 \varphi}{\partial t^2 \partial x}$
 olution method
 **olution are blue to the Navier-type solution method for the free

that the shape of a simply supported SWCNT, the displ** *intrimal*
 *i the Navier-type solution method for the free imply supported SWCNT, the displacement pressed as the product of undetermined coeffi-

<i>n* trigonometric functions to satisfy the governant triends at $x = 0$ $A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{21} \frac{\partial^4 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{105} \frac{\partial^3 \varphi}{\partial t^2 \partial x}$
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ons are exp $+\rho A \frac{\partial^2 w}{\partial t^2} - \frac{\rho I}{21} \frac{\partial^4 w}{\partial t^2 \partial x^2} + \frac{16\rho I}{105} \frac{\partial^3 \varphi}{\partial t^2 \partial x}$
 Solution method
 Solution the Navier-type solution method for the free
 Solution Here, based on the Navier-type solution method for t

$$
w(x,t) = \sum_{n=1}^{\infty} W_n \sin(\frac{n\pi}{L}x) e^{i\omega_n t}
$$
 (20a)

$$
\rho(x,t) = \sum_{n=1}^{\infty} \phi_n \cos(\frac{n\pi}{L}x) e^{i\omega_n t}.
$$
 (20b)

where (W_n, ϕ_n) are the unknown Fourier coefficients to be determined for each *n* value. By substituting Eq. (20) into Eqs. (18) and (19) and setting the determinant of the coefficient matrix, the analytical solutions can be obtained from the following equation: F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 3797-38

rec (W_n , ϕ_n) are the unknown Fourier coefficients to be

mined for each *n* value. By substituting Eq. (20) into Eqs.

a *F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9)*
 C (W_n , ϕ_n) are the unknown Fourier coefficients to be
 Table 1. Some of the transform indefined for each *n* value. By substituti *F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 379*
 (W_n, ϕ_n) are the unknown Fourier coefficients to be

and for each *n* value. By substituting Eq. (20) into Eqs.
 $\left\{\begin{array}{l}\text{$ *F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 379*

are the unknown Fourier coefficients to be

<sup>Table 1. Some of the transformation r

ch *n* value. By substituting Eq. (20) into</sup>

$$
\left\{ [K] - \omega^2 [M] \right\} \begin{Bmatrix} W_n \\ \phi_n \end{Bmatrix} = 0
$$
 (21)

where $[K]$ and $[M]$ are the stiffness matrix and mass matrix, respectively. By setting the determinant of the coefficient matrix, one obtains a quadratic polynomial for ω_n^2 , and then by setting this polynomial to zero, one can find ω_n .

3.2 Implementation of DTM

Some of the common numerical methods used to solve the initial- and boundary-value problems occurring in engineering are homotopy perturbation method, variational iteration method, parameter-expanding methods, and others [16]. Moreover, the principle of DTM is to transform the differential equations of governing and boundary equations into a set of algebraic equations using Taylor series expansion with transformation rules. The details of DTM have been described elsewhere [17]. Therefore, the elaborate information on this aspect is not attempted here. Based on the basic transformation operations introduced in Table 1, the transformed form of the governing Eqs. (18) and (19) may be obtained as the following: Implementation of DTM

Some of the common numerical methods used to solve the

al- and boundary-value problems occurring in engineering

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methods, and others [16].

bo and boundary-value problems occurring in engineering

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d, parameter-expanding methods, and others [16] boundary conditions, one a

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By using Eqs. (22) and (23) toger

method, variational iteration

IM is to transform the differen-

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lis of DTM have omotopy perturbation method, variational iteration

by using Eqs.

1, parameter-expanding methods, and others [16]. boundary condit

ver, the principle of DTM is to transform the differen-

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the principle of DTM **ation of DTM**

common numerical methods used to solve the

mdary-value problems occurring in engineering

perturbation method, variational iteration

perturbation method, variational iteration

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bolems occurring in engineering

method, variational iteration

EV is to transform the differen-

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M is to transform the differen-

M is to transform the d and boundary-value problems occurring in engineering
 A parameter-expanding methods, and others [16]. Boundary conditions, one arrives

ver, the principle of DTM is to transform the differen-

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principle of DTM is to transform the differen-

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 urion of DTM

common numerical methods used to solve the

mary-value problems occurring in engineering

perturbation method, variational iteration

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thod, parameter-expanding methods, and others [16]. boundary correcover, the principle of DTM is to transform the differen-

equations of governing and boundary equat

are homotopy perturbation method, variational iteration
\nMethod, variational iteration
\nMoreover, the principle of DTM is to transform the difference
\nMoreover, the principle of DTM is to transform the difference
\ntransformation rules. The details of DTM have been described
\ntransformation rules. The detection is the basic transform with
\ntransformation rules. The detection of the basic transformations
\nthis is the first information on this
\ntransformation rules. The detection table 1, the transformed form of
\nthe governing Eqs. (18) and (19) may be obtained as the fol-
\n
$$
F = \sqrt{\frac{8GM}{105} - \mu \frac{16\rho I \omega^2}{105}} \frac{k+3}{k+1}W[k+1]
$$
\n
$$
= \sqrt{\frac{8GM}{105} - \frac{6R\rho I \omega^2}{105}} \frac{(k+2)!}{k!}W[k+2]
$$
\n
$$
= \sqrt{\frac{8GM}{105} - EA\alpha_s T + iK_s - \mu \rho A\omega^2 - \frac{\rho I \omega^2}{21}} \frac{(k+4)!}{k!}W[k+4]
$$
\n
$$
= \sqrt{\frac{8GM}{105} - \frac{6R\rho I \omega^2}{105}} \frac{(k+1)!}{k!}W[k+1]
$$
\n
$$
= \sqrt{\frac{8GM}{105} - \frac{6R\rho I \omega^2}{105}} \frac{(k+2)!}{k!}W[k+2]
$$
\n
$$
= \sqrt{\frac{8GM}{105} - \frac{6R\rho I \omega^2}{105}} \frac{(k+2)!}{k!}W[k+4]
$$
\n
$$
= \sqrt{\frac{8GM}{105} - \frac{6R\rho I \omega^2}{105}} \frac{(k+2)!}{k!}W[k+4]
$$
\n
$$
= \sqrt{\frac{8GM}{105} - \frac{6R\rho I \omega^2}{105}} \frac{(k+4)!}{k!}W[k+4]
$$
\n
$$
= \sqrt{\frac{8GM}{105} - EA\alpha_s T + iK_s - \mu \rho A\omega^2 - \frac{\rho I \omega^2}{21}} \frac{(k+4)!}{k!}W[k+4]
$$
\n
$$
= \sqrt{\frac{8GM}{105} - EA\alpha_s T + iK_s - \mu \rho A\omega^2 - \frac{\rho I \omega^2}{21}} \frac{(k+4)!}{k!}W[k+4]
$$
\n
$$

$$

and *w* , respectively. By employing DTM theorems into Eqs.

Table 1. Some of the transformation rules of DTM.

cience and Technology 29 (9) (2015) 3797~3803	
Table 1. Some of the transformation rules of DTM.	
Original function	Transformed function
$f(x) = g(x) \pm h(x)$	$F(K) = G(K) \pm H(K)$
$f(x) = \lambda g(x)$	$F(K) = \lambda G(K)$
$f(x) = \frac{d^n g(x)}{dx^n}$	$F(K) = \frac{(k+n)!}{k!} G(K+n)$
can be given as the following:	$(14)-(16)$, the transformed boundary conditions are obtained. For instance, the transformed simply supported edge condition
$W[0] = 0, \frac{68}{105}\phi[1] - \frac{16}{105}W[2] = 0$	(24)
	$\sum_{k=0}^{\infty} W[k] = 0, \frac{68}{105} \sum_{k=0}^{\infty} k \phi[k] - \frac{16}{105} \sum_{k=0}^{\infty} k(k-1) W[k] = 0.$
problem:	By using Eqs. (22) and (23) together with the transformed boundary conditions, one arrives at the following eigenvalue
$\begin{bmatrix} M_{11}(\omega) & M_{12}(\omega) & M_{13}(\omega) \end{bmatrix}$ $\begin{bmatrix} M_{21}(\omega) & M_{22}(\omega) & M_{23}(\omega) \\ M_{31}(\omega) & M_{32}(\omega) & M_{33}(\omega) \end{bmatrix} [C] = 0$	(25)

$$
f(x) = \frac{d^n g(x)}{dx^n}
$$

\n
$$
F(K) = \frac{(k+n)!}{k!} G(K+n)
$$

\n(14)-(16), the transformed boundary conditions are obtained.
\nFor instance, the transformed simply supported edge condition
\ncan be given as the following:
\n
$$
W[0] = 0, \frac{68}{105} \phi[1] - \frac{16}{105} W[2] = 0
$$
\n(24)
\n
$$
\sum_{k=0}^{\infty} W[k] = 0, \frac{68}{105} \sum_{k=0}^{\infty} k\phi[k] - \frac{16}{105} \sum_{k=0}^{\infty} k(k-1) W[k] = 0.
$$

\nBy using Eqs. (22) and (23) together with the transformed
\nboundary conditions, one arrives at the following eigenvalue
\nproblem:
\n
$$
\begin{bmatrix} M_{11}(\omega) & M_{12}(\omega) & M_{13}(\omega) \\ M_{21}(\omega) & M_{22}(\omega) & M_{23}(\omega) \\ M_{31}(\omega) & M_{32}(\omega) & M_{33}(\omega) \end{bmatrix} [C] = 0
$$
\n(25)
\nwhere [C] corresponds to the missing boundary conditions at
\n $x = 0$. For the non-trivial solutions of Eq. (25), the determinant
\nof the coefficient matrix should be equal to zero. In solving
\nEq. (25), the ith estimated eigenvalue for nth iteration

$$
\begin{bmatrix} M_{11}(\omega) & M_{12}(\omega) & M_{13}(\omega) \\ M_{21}(\omega) & M_{22}(\omega) & M_{23}(\omega) \\ M_{31}(\omega) & M_{32}(\omega) & M_{33}(\omega) \end{bmatrix} [C] = 0
$$
 (25)

 $(\omega = \omega_i^{(n)})$ may be obtained, and the total number of iterations x = 0. For the non-trivial solutions of Eq. (25), the determ
nant of the coefficient matrix should be equal to zero. In so
ing Eq. (25), the ith estimated eigenvalue for nth iterati
($\omega = \omega_i^{(n)}$) may be obtained, and th where $\lceil c \rceil$ corresponds to the missing boundary conditions at $x = 0$. For the non-trivial solutions of Eq. (25), the determinant of the coefficient matrix should be equal to zero. In solving Eq. (25), the ith estimated eigenvalue for nth iteration ² 105

² $\sum_{k=0}^{\infty} W[k] = 0, \frac{68}{105} \sum_{k=0}^{\infty} k\phi[k] - \frac{16}{105} \sum_{k=0}^{\infty} k(k-1)W[k] = 0$.

By using Eqs. (22) and (23) together with the transformed

boundary conditions, one arrives at the following eigenvalue
 is related to the accuracy of the calculations, which can be determined by the following equation: (*y* using Eqs. (22) and (23) together with the transformed
ndary conditions, one arrives at the following eigenvalue
blem:
 $M_{11}(\omega) M_{12}(\omega) M_{13}(\omega)$
 $M_{21}(\omega) M_{22}(\omega) M_{23}(\omega)$
 $m_{31}(\omega) M_{32}(\omega) M_{33}(\omega)$
 $\left[C \right] = 0$ (25

$$
\left| \omega_i^{(n)} - \omega_i^{(n-1)} \right| < \varepsilon \,. \tag{26}
$$

In this study, $\varepsilon = 0.0001$ is considered in the process of determining eigenvalues, resulting in a four-digit precision in the estimated eigenvalues.

4. Convergence and correctness of the method

($\omega = \omega_i^{(n)}$) may be obtained eigenvalue for finite

($\omega = \omega_i^{(n)}$) may be obtained, and the total number of iteration

is related to the accuracy of the calculations, which can b

determined by the following equation:
 21 $k!$ will be performed, and the effects of several parameters on the $\int 16EI$ $16\rho I\omega^2$ $(k+3)!$ *k*, a natural frequencies of the SWCNT will be identified. The $+\left(\frac{102h}{105} - \mu \frac{16p \ln 3}{105}\right)$ $\frac{(h+3)}{k!} \phi[k+3]$ effective properties of SWCNT are taken as that of Reddy and $A\omega^2 - \frac{\rho I\omega^2}{2} \left| \frac{(k+2)!}{k+2!} w[k+2] \right|$ Through this section, a numerical testing of the procedure (23) [19], the coefficients of thermal expansion for CNTs are nega-Pang [18]. The Young's modulus $E = 1000$ GPa, mass density $p = 2300$ Kg/m³, Poisson's ratio $v = 0.19$, and diameter $d = 1$ nm are considered in the analysis. As indicated by Jiang et al. tive at a lower temperature and become positive at a higher temperature; for the case of room or low temperature, we suppose a $\alpha_x = -1.6 \times 10^{-6}$ 1/K [20]. The following relation is andy, $\varepsilon = 0.0001$ is considered in the process of de-
igenvalues, resulting in a four-digit precision in the
genvalues.
ence and correctness of the method
this section, a numerical testing of the procedure
ormed, and

Table 2. Convergence study for the first three natural frequencies of S-

					F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 3797~
S SWCNT $(L/d = 50, \mu = 3 \text{ nm}^2, T = 30 \text{ K}$).				Table 2. Convergence study for the first three natural frequencies of S-	(varying from 0 to 4) and length-to presented by Reddy [12], which
Method	k	$\hat{\omega}_1$	$\hat{\omega}_2$	$\hat{\omega}_3$	analytical method. The parameter
	10	10.5517	\blacksquare	\blacksquare	following [12]: $L = 10, E = 30*1$
	12	10.7477			clearly notice that the non-dimensi
DTM	14	10.7280			parameters obtained in the presen
	16	10.7293	37.3927		lent agreement to the results pres-
					compared cases and validate the pr
	18	10.7292	40.0444	÷	
	20	10.7292	39.4259	$\overline{}$	5. Numerical results and discu
	22	10.7292	39.4904		After the approaches are validat
	24	10.7292	39.4841		are conducted in this section to exa
	26	10.7292	39.4846	84.2696	ous parameters such as nonlo
	28	10.7292	39.4846	85.2870	thickness ratios, temperature cha
	30	10.7292	39.4846	85.1365	tions on the natural frequencies of
	32	10.7292	39.4846	85.1528	combinations of edge conditions a
	34	10.7292	39.4846	85.1512	simply-supported/simply-supporte
	36	10.7292	39.4846	85.1513	supported (C-S), clamped/clampe
	38	10.7292	39.4846	85.1513	(C-F). In Table 4, the first three d
	40	10.7292	39.4846	85.1513	the S-S SWCNT are presented fo
	42	10.7292	39.4846	85.1513	aspect ratios ($L/d = 10, 50$) and
	44	10.7292	39.4846	85.1513	$(0,1,2,3,4)$ for low temperature ch both DTM and analytical Navier s
Analytical		10.7288	39.4725	85.0864	the nonlocal parameter vanishes,
S-S SWCNT $(L/d = 10)$.				Table 3. Comparison of non-dimensional fundamental frequencies of	beam theory is rendered. An inc rameter gives rise to a decremer sionless frequencies because the p
			Present DTM		fect tends to decrease the stiffnes
$\mu = (e_0 a)^2$		Analytical [12] 9.7454		9.7360	hence decreases the values of nature-
0 1		9.2974		9.2885	sionless natural frequencies predi agreement with those evaluated
\overline{c}		8.9060		8.8974	Moreover, the first three dimens
					increase by increasing aspect ratio
3		8.5602		8.5520	offocts of different houndary cond

Table 3. Comparison of non-dimensional fundamental frequencies of S-S SWCNT (*L/d =*10).

performed to calculate the non-dimensional natural frequen cies:

$$
\hat{\omega} = \omega L^2 \sqrt{\rho A / EI} \tag{27}
$$

where *I* is the moment of inertia of the cross section of the nanotube. It is found that in DTM after a certain number of iterations eigenvalues converged to a value with good precision, so the number of iterations is important in DTM conver gence. Table 2 tabulates the convergence study for the first three natural frequencies of the simply supported SWCNT. As seen in Table 2, the third natural frequency converged after 38 iterations with 4-digit precision, whereas the first and second ones converged after 20 and 28 iterations, respectively. To evaluate the accuracy of the natural frequencies predicted by the present method, Table 3 compares the semi-analytical results of the present study and the results obtained for the simply supported SWCNT with different nonlocal parameters (varying from 0 to 4) and length-to-thickness ratios $(L/d = 10)$ presented by Reddy [12], which has been obtained by the analytical method. The parameters used in this table are the Figure and Technology 29 (9) (2015) 3797~3803 3801
(varying from 0 to 4) and length-to-thickness ratios ($L/d = 10$)
presented by Reddy [12], which has been obtained by the
analytical method. The parameters used in this tab clearly notice that the non-dimensional fundamental frequency parameters obtained in the present investigation are in excellent agreement to the results presented by Reddy [12] for all compared cases and validate the proposed method of solution.

5. Numerical results and discussions

Analytical 10.7292 39.4846 85.1513

Analytical 10.7292 39.4846 85.1513 6.1613

both DTM analytical Na

both DTM analytical Na

both DTM analytical Na

both DTM analytical Carrier gives rise to a decrease the st

SENCNT After the approaches are validated, some parametric studies are conducted in this section to examine the influences of various parameters such as nonlocal parameters, length-tothickness ratios, temperature changes, and boundary conditions on the natural frequencies of the SWCNT. Four different combinations of edge conditions at *x* =0, *L* are considered as: simply-supported/simply-supported (S-S), clamped/simply supported (C-S), clamped/clamped (C-C), and clamped/free (C-F). In Table 4, the first three dimensionless frequencies of the S-S SWCNT are presented for the various values of the aspect ratios ($L/d = 10$, 50) and nonlocal parameters ($\mu = 0,1,2,3,4$) for low temperature change ($T = 30$ K) based on both DTM and analytical Navier solution method. First, when the nonlocal parameter vanishes, Reddy's classical isotropic beam theory is rendered. An increase in nonlocal scale parameter gives rise to a decrement in the first three dimen sionless frequencies because the presence of the nonlocal effect tends to decrease the stiffness of the nanostructures and hence decreases the values of natural frequencies. The dimen sionless natural frequencies predicted by DTM are in close agreement with those evaluated through analytical solution. Moreover, the first three dimensionless natural frequencies increase by increasing aspect ratio (*L/d*). In investigating the effects of different boundary conditions and nonlocal parameters on the vibration characteristics of SWCNT, the non dimensional frequencies of SWCNT under low temperature change ($T = 50$ K) with different edge conditions (i.e., C-C, C-S, and C-F) are tabulated in Table 5. This table shows that frequencies predicted by the nonlocal solution are smaller than those of the local frequency because of the small-scale effects. For this reason, the size effect plays a major role in the vibration behavior of CNTs. Moreover, for a C-F SWCNT, increasing the nonlocal parameter leads to an increase in fundamental frequency. Furthermore, a beam with stiffer edges, such as C- C and C-S, shows higher natural frequencies than those of other boundary conditions, as expected. The fundamental frequency parameter as a function of aspect ratio and temperature increase is presented in Fig. 2 for the S-S SWCNT with different nonlocal parameters. Similarly, the variation of the first dimensionless frequency with temperature change for different nonlocality is depicted in Fig. 3. Observing these two figures shows that increasing the nonlocality parameter yields the reduction in dimensionless frequencies for all temperature changes and aspect ratios, which highlights the significance of

3802				F. Ebrahimi and E. Salari / Journal of Mechanical Science and Technology 29 (9) (2015) 3797~3803							
	Table 4. Effects of nonlocality parameters and aspect ratios on the first three dimensionless frequencies of S-S SWCNT ($T = 30K$).										
L/d	$\mu = (e_0 a)^2$		Present DTM	Present Analytical		Present DTM		Present DTM		Present Analytical	
			$\hat{\omega}_1$			$\hat{\omega}_2$		$\hat{\omega}_3$			
	$\boldsymbol{0}$		9.8147	9.7932		38.0899	37.7910	82.0930		80.9035	
	-1		9.3670	9.3466		32.2645		59.7655		58.9018	
10	$\mathbf{2}$		8.9761	8.9566		28.4951		49.3061		48.5961	
	$\mathfrak z$		8.6309	8.6121		25.8018	25.6012		42.9333	42.3175	
	$\overline{4}$		8.3231	8.3050		23.7542	23.5701	38.5332		37.9827	
	$\boldsymbol{0}$		10.7822	10.7818		40.3648	40.3522	89.4739		89.4050	
	$\mathbf{1}$		10.7644	10.7640		40.0644	40.0520	87.9585		87.8910	
50	\overline{c}		10.7468	10.7464		39.7711	39.7589	86.5197		86.4535	
	3		10.7292	10.7288		39.4846	39.4725	85.1513		85.0864	
	$\overline{4}$		10.7118	10.7114		39.2046	39.1926	83.8478		83.7840	
	Table 5. Effect of different boundary conditions on the first three dimensionless frequencies of SWCNT ($T = 50$ K).										
						Boundary conditions					
L/d	$\mu = (e_0 a)^2$		$C-C$			$\mbox{C-S}$				$\operatorname{C-F}$	
		$\hat{\omega}_1$	$\hat{\omega}_2$	$\hat{\omega}_3$	$\hat{\omega}_1$	$\hat{\omega}_2$	$\hat{\omega}_3$	$\hat{\omega}_1$	$\hat{\omega}_2$	$\hat{\omega}_3$	
	$\boldsymbol{0}$	22.1465	59.1398	110.8100	15.2966	48.0657	95.9430	3.5033	21.7941	59.1722	
	$\mathbf{1}$	20.8464	48.4212	77.1223	14.4800	40.0684	68.3604	3.5202	20.4085	48.5095	
10	$\overline{2}$	19.7498	41.9787	62.7322	13.7814	35.0894	56.0132	3.5376	19.2244	42.1439	

	$\mu = (e_0 a)^2$	Boundary conditions									
L/d		$C-C$			$C-S$			$C-F$			
		$\hat{\omega}_1$	$\hat{\omega}_2$	$\hat{\omega}_3$	$\hat{\omega}_1$	ω	$\hat{\omega}_3$	$\hat{\omega}_1$	$\hat{\omega}_2$	$\hat{\omega}_3$	
10	$\mathbf{0}$	22.1465	59.1398	110.8100	15.2966	48.0657	95.9430	3.5033	21.7941	59.1722	
		20.8464	48.4212	77.1223	14.4800	40.0684	68.3604	3.5202	20.4085	48.5095	
	2	19.7498	41.9787	62.7322	13.7814	35.0894	56.0132	3.5376	19.2244	42.1439	
	3	18.8092	37.5709	54.2844	13.1751	31.6129	48.6279	3.5553	18.1958	37.8274	
	4	17.9913	34.3152	48.5681	12.6428	29.0103	43.5794	3.5734	17.2902	34.6722	
50	$\mathbf{0}$	23.2232	62.7430	121.7400	16.5587	51.2340	105.2980	3.1556	22.9575	62.7638	
		23.1801	62.2102	119.4670	16.5300	50.8304	103.4330	3.1584	22.9112	62.2324	
	2	23.1373	61.6917	117.3210	16.5016	50.4371	101.6680	3.1613	22.8653	61.7153	
	3	23.0948	61.1868	115.2910	16.4733	50.0537	99.9940	3.1642	22.8197	61.2119	
	4	23.0527	60.6950	113.3670	16.4452	49.6798	98.4040	3.1670	22,7744	60.7217	

Fig. 2. Variation of the first dimensionless frequency of S-S SWCNT versus aspect ratio (L/d) for different low temperatures ($\mu = 2$).

the nonlocal effect. In addition, the fundamental frequency increases by increasing temperature changes, and temperature change has a significant effect on the fundamental frequency of the S-S SWCNT.

Finally, Fig. 4 displays the variations of the first dimen sionless natural frequency of the S-S SWCNT with respect to

Fig. 3. Variation of first dimensionless frequency of S-S SWCNT ver sus temperature change for different nonlocal parameters $(L/d = 20)$.

Winkler spring modulus for different values of nonlocal parameters at a constantly low temperature ($T = 40$ K) and aspect ratio $(L/d = 10)$. The figure shows that the fundamental frequency of SWCNT increases with the increase of Winkler spring modulus. In addition, the small-scale parameter has a softening effect on SWCNT.

Fig. 4. Variation of the first dimensionless frequency of S-S SWCNT versus Winkler spring modulus for different nonlocal parameters (*L/d* $= 10$, $T = 40$ K).

6. Conclusions

The vibrational behavior of embedded SWCNTs subjected to thermal loading with various boundary conditions is investigated based on the Reddy beam theory in conjunction with DTM. Eringen's theory of nonlocal elasticity together with higher order beam theory is used to model the CNT by the implementation of Hamilton's principle. The accuracy of the results is examined using available data in the literature. Finally, through some parametric studies and numerical exam ples, various factors such as nonlocal parameter, temperature change, boundary conditions, and Winkler spring modulus are found to play important roles in the dynamic behavior of SWCNTs. The nanobeam model produces a smaller natural frequency than the classical (local) beam model. Therefore, the small-scale effects should be considered in the analysis of the mechanical behavior of nanostructures. The dimensionless frequency decreases with the increase in temperature. Moreover, under an increase in temperature, an increase in the Winkler spring modulus results in an increase in frequency.

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