

Mean-value first-order saddlepoint approximation based collaborative optimization for multidisciplinary problems under aleatory uncertainty†

Debiao Meng, Hong-Zhong Huang* , Zhonglai Wang, Ning-Cong Xiao and Xiao-Ling Zhang

School of Mechanical, Electronic, and Industrial Engineering, University of Electronic Science and Technology of China, No. 2006, Xiyuan Avenue, West Hi-Tech Zone, Chengdu, Sichuan, 611731, China

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Abstract

Reliability-based multidisciplinary design optimization (RBMDO) has received increasing attention in engineering design for achieving high reliability and safety in complex and coupling systems (e.g., multidisciplinary systems). Mean-value first-order saddlepoint approximation (MVFOSA) is introduced in this paper and is combined with the collaborative optimization (CO) method for reliability analysis under aleatory uncertainty in RBMDO. Similar to the mean-value first-order second moment (MVFOSM) method, MVFOSA approximated the performance function with the first-order Taylor expansion at the mean values of random variables. MVFOSA uses saddlepoint approximation rather than the first two moments of the random variables to estimate the probability density and cumulative distribution functions. MVFOSA-based CO (MVFOSA-CO) is also formulated and proposed. Two examples are provided to show the accuracy and efficiency of the MVFOSA-CO method.

Keywords: Reliability-based multidisciplinary design optimization; Mean-value first-order saddlepoint approximation; Aleatory uncertainty; Probability density function; Cumulative distribution function

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1. Introduction

Multidisciplinary design optimization (MDO) is traditionally formulated as a deterministic problem that assumes the nonexistence of uncertainty. However, the real design and manufacturing processes are often affected by unavoidable uncertainties that cause system performance and output variations [1]. These variations may cause design solutions to be infeasible or unreliable. For high reliability and safety in MDO problems, reliability-based multidisciplinary design optimization (RBMDO) is one research focus. Sues et al. [2] created response surface models at the system level to replace the computationally expensive simulation models and relieve RBMDO of its reliability analysis and computation burden. Sues and Cesare [3] proposed an RBMDO framework where reliability analysis was decoupled from the optimization loop. Reliability is initially computed before the first execution of the optimization loop and then updated iteratively after. During this process, approximate reliability constraints are used. In Ref [4], a multi-stage parallel implementation of probabilistic design optimization integrated the existing reliability analysis method into MDO frameworks. The concurrent subsys-

*Corresponding author. Tel.: +86 28 6183 0248, Fax.: +86 28 6183 0227

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tem optimization was widely used to search for the most probable point (MPP) [5**-**7]; the collaborative reliability analysis method was also broadly used [8, 9]. In Ref. [10], a sequential optimization and reliability assessment (SORA) method for RBMDO was proposed. The deterministic formulation of MDO in SORA was constructed by using the MPP from the previous iteration. Following each optimization loop, reliability analysis was conducted at the optimal solution of the deterministic MDO to check the probability constraint feasibilities. Huang et al. [11] proposed an enhanced SORA (ESORA) method to further improve computational efficiency for reliability analysis. ESORA addresses the constant and varying variances of random design inputs. Zhang et al. [12] introduced probability and possibility analyses into RBMDO and proposed an RBMDO with discrete and continuous variables of various uncertainties to simultaneously analyze aleatory and epistemic uncertainties. Xiao et al. [13] used interval variables to consider epistemic uncertainty and proposed the unified uncertainty analysis method to estimate the failure probabilities of complex systems with both epistemic and aleatory uncertainties.

A number of works have been conducted on RBMDO but some issues still require further exploration. In some cases, the performance function is expensive to evaluate but the meanvalue first-order Second Moment (MVFOSM) method is feasible. MVFOSM is highly efficient and easy to use. However,

E-mail address: hzhuang@uestc.edu.cn

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it has obvious accuracy deficiencies [14**-**16]. MVFOSM uses only the first two moments of random variables instead of the complete distribution information and assumes that the response is normally distributed. The mean-value first-order saddlepoint approximation (MVFOSA) is used for aleatory uncertainty analysis in the mentioned situations to improve analysis accuracy while retaining high efficiency. Compared utilizes complete distribution information [15]. Instead of and cumulative distribution function (CDF) of the performance function by saddlepoint approximation. The MVFOSA method is combined with the collaborative optimization (CO) method in the present paper to solve the RBMDO problem. CO is a bi-level MDO method for large scale and distributedanalysis engineering design problems. CO contains optimization problems at both system and discipline levels. The system-level optimization problem optimizes the system objective and the coordinate consistency between the coupling disciplines, whereas the discipline-level optimization problem minimizes the discrepancy between the design variables and their targets [17].

The rest of the paper is organized as follows. Section 2 introduces the RBMDO problem. Commonly used reliability analysis methods, including simulation and approximation, are also briefly reviewed. Sec. 3 provides the fundamental analysis of MVFOSA is provided. Sec. 4 explains the proposed method, namely, the MVFOSA-based CO (MVFOSA-CO), including the strategy, procedure, and formulas. Sec. 5 portrays two examples to demonstrate the accuracy and efficiency of the proposed method. Finally, Sec. 6 concludes.

2. RBMDO problems and simulation or approximation methods for reliability analysis

RBMDO is formulated as follows:

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quired reliability of the probabilistic constraints X_{pv} denotes the design variables; **d** is the deterministic design variables; **X** denotes the random discipline design variables; X_s denotes the random sharing design variables; a **Y** denotes the random coupling design variables,

md Technology 28 (10) (2014) 3925-3935
 Y = {**Y**_{*i*}, **Y**_{*i*}}, *i* = 1, 2,..., *n* ; **Y**_{*i*} is the input coupling variables to the *i* th discipline; **Y**_{*i*} is the output coupling variables from the *i* th disc $\mathbf{Y} = \{ \mathbf{Y}_{i}, \mathbf{Y}_{i}, \}$, $i = 1, 2, ..., n$; \mathbf{Y}_{i} is the input coupling variables to the *i* th discipline; Y_i is the output coupling variables from the *i* th discipline; μ denotes the mean value of random variables and random parameters; superscripts *L* and *U* denote the lower and upper bounds, respectively; *n* is the total number of disciplines.

with MVFOSM, MVFOSA is relatively accurate because it $X_R = \{X, X_s, Y\}$ and express the function relationship besimply using the first two moments of random variables, $G = g(d, X_R)$. The probability of the probabilistic constraints MVFOSA estimates the probability density function (PDF) $P[g_i(\bullet) \le 0]$ can be described as the CDF of $g_i(\bullet)$ and is We use X_R to denote the set of random variables mal Technology 28 (10) (2014) 3925-3935
 Y = {**Y**<sub>₁,**Y**_n}, *i* = 1, 2,..., *n* ; **Y**_n is the input coupling vari-

ables to the *i* th discipline; **Y**_n is the output coupling vari-

ables the *i* th discipline; </sub> tween performance function *G* and the design variables as md Technology 28 (10) (2014) 3925-3935
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 Y={**Y_{_{***i}}, Y_n[}], i = 1, 2,..., n; Y_n[{] is the input coupling variables to the <i>i* th discipline; **Y_n** is the output coupling variables from the *i* th discipline; **µ**</sub></sub> theoretically calculated by Eq. (2). 28 (10) (2014) 3925-3935
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= 1, 2,..., n; **Y**₁, is the input coupling vari-

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es to the *i* th discipline; \mathbf{Y}_n is the output coupling vari-

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 i = 1, 2,..., *n*; **Y**_{*k*} is the input coupling vari-

th discipline; **Y**_{*k*} is the output coupling vari-

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bles and random parameters; supersc and Technology 28 (10) (2014) 3925-3933
 $\mathbf{Y} = \begin{bmatrix} \mathbf{V}_{n_1} \mathbf{V}_{n_2} \end{bmatrix}$, $i = 1, 2, ..., n$; \mathbf{Y}_{n_1} is the input coupling vari-
 $\mathbf{Z} = \begin{bmatrix} \mathbf{V}_{n_1} \mathbf{V}_{n_2} \end{bmatrix}$, $i = 1, 2, ..., n$; \mathbf{Y}_{n_1} is the outp

$$
F_G = P\Big[g_i\big(\mathbf{d}, \mathbf{X}_{\mathbf{R}}\big) \le 0\Big] = \int\limits_{g_i(\mathbf{d}, \mathbf{X}_{\mathbf{R}}) \le 0} f_{\mathbf{X}_{\mathbf{R}}}(\mathbf{X}_{\mathbf{R}})\, \mathrm{d}\mathbf{X}_{\mathbf{R}}\;, \tag{2}
$$

 $f_{\mathbf{x}_{R}}(\mathbf{x}_{R})$ is the joint PDF of \mathbf{X}_{R} . However, arriving at the analytical solution by using Eq. (2) is difficult because of the high dimensionality of random variables and the fore, simulation or approximation methods are widely used.

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 Y = $\left\{ \mathbf{Y}_{a}, \mathbf{Y}_{a}\right\}$, $i = 1, 2, ..., n$; **Y**_n is the input coupling vari-

ables to the *i*th discipline; **Y_n** is the output coupling vari-

ables to the *i*th discipline Simulation or approximation methods have three types: (1) sampling-based methods, (2) moment matching methods, and (3) MPP-based methods [15]. Sampling-based methods are easy to apply and can provide accurate probability estimations with sufficient simulations [19**-**22]. However, sampling-based methods are inefficient for many engineering problems with high reliability and computationally expensive performance functions [16]. Moment-matching methods ease computational difficulties by approximating the distribution of performance functions and by fitting the first few moments [23**-** 26]. MVFOSM is one of the commonly used moment matching methods. It uses the first two statistical moments and employs first-order Taylor expansion at the mean values of random variables. The moment matching method is highly efficient. However, its accuracy is generally lower than that of sampling-based methods [15]. MPP-based methods have a good balance between efficiency and accuracy. The first-order reliability method (FORM) and second-order reliability method (SORM) [27, 28] approximate the performance function with Taylor expansion at the MPP. However, the MPP location is also an optimization problem and needs more function evaluations than MVFOSM, thus making FORM and SORM computationally expensive for complex and coupling systems [15]. The original random variables in **X -**space have to be transformed into standard normal variables for FORM and SORM [29], thus increasing the nonlinearity of the performance function [30, 31]. First-order saddlepoint approximation (FOSA) is proposed for reliability analysis to avoid random variable transformation [31]. FOSA linearizes the performance function in the original random **X** -space without any random variable transformation. The expansion point, that is, the most likelihood point (MLP), has the highest probmations "too" women-maticuring "attentons" exact computer
tional difficulties by approximating the distribution of performance functions and by fitting the first few moments [23-26]. MVFOSM is one of the commonly used mome ing the MLP in FOSA also requires an optimization process.

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	Table 1. CGFs of some common distributions.		
Distribution	PDF	CGF	
Normal	$f(x) = \left(\frac{1}{\sqrt{2\pi}\sigma}\right) \exp\left(\left(x-\mu\right)^2/2\sigma^2\right)$	$K(t) = \mu t + \frac{1}{2} \sigma^2 t^2$	
Uniform	$f(x) = 1/(b-a)$	$K(t) = \ln(e^{bt} - e^{at}) - \ln(b - a) - \ln(t)$	
Exponential	$f(x)=\alpha \exp(-\alpha x)$	$K(t) = -\ln(1-t/\alpha)$	
Gumbel	$f(x) = (\frac{1}{\sigma}\exp[(x-\mu)/\sigma]$ $\exp\left\{-\exp\left[\left(x-\mu\right)/\sigma\right]\right\}$	$K(t) = \mu t + \ln \Gamma(1 - \sigma t)$	
Gamma	$f(x)=\beta^{\alpha}/\Gamma(\alpha)x^{\alpha-1}e^{-\beta x}$	$K(t) = \alpha \left[\ln \beta - n(\beta - t) \right]$	
χ^2	$f(x) = \left[1/\Gamma(n/2)2^{n/2}\right]x^{n/2-1}e\left(-\frac{1}{2}x\right)$	$K(t) = -\frac{1}{2} n \ln (1-2t)$	
	3. Mean-value first-order saddlepoint approximation	$K_{Y}(t) = -(1/2) n \ln (1-2at) + bt$. On the basis of the above two properties, the CGF of \hat{G} is	
	MVFOSA was introduced in reliability analysis by Huang and Du [15] not only for its similar efficiency and robustness	given by Eq. (4) . The saddlepoint t_c can be determined by solving Eq. (5)	
	to MVFOSM but also for its relatively high accuracy.	[15, 34], where $K'_{\sigma}(t)$ is the first-order derivative of CGF.	
	MVFOSA uses saddlepoint approximation to evaluate the CDF and PDF of the performance function. MVFOSM has	Once both the CGF of \hat{G} and saddlepoint t_{s} are obtained, saddlepoint approximation can be applied for PDF and CDF	
reliability analysis [33].	been used in reliability sensitivity analysis [32] and structural	estimations.	
	In MVFOSA, performance function G is linearized in the	$K_{\hat{G}}(t) = \left(g\left(\mathbf{d}^*, \mathbf{\mu}_{\mathbf{x}_R}\right) + \sum_{i=1}^n \frac{\partial G}{\partial \mathbf{d}_i} \middle _{\mathbf{d}_i} \left(\mathbf{d}_i - \mathbf{d}_i^*\right) - \sum_{i=1}^n \frac{\partial G}{\partial \mathbf{X}_{Ri}} \middle _{\mu_{\mathbf{x}_R}} \mu_{\mathbf{x}_{Ri}}\right)t$	
	original random space by using first-order Taylor expansion. The expansion point is at values of deterministic variables \mathbf{d}_i^*		
	and mean values of random variables $\mu_{\mathbf{x}_{\mathbf{p}}}$. The first-order approximation function $G \approx \hat{G} = \hat{g}(\mathbf{d}, \mathbf{X}_{R})$ can be expressed	$+\sum_{i=1}^n K_{\mathbf{x}_{Ri}}\left(\frac{\partial G}{\partial \mathbf{X}_{Ri}}\Bigg _{t_{Ri}}\right)$	
as follows:			(4)
	$\hat{G} = \hat{g}(\mathbf{d}, \mathbf{X}_{R}) = g(\mathbf{d}^*, \mathbf{\mu}_{\mathbf{X}_{R}}) + \sum_{i=1}^{n} \frac{\partial G}{\partial \mathbf{d}_{i}} \left(\mathbf{d}_{i} - \mathbf{d}_{i}^* \right)$	$K'_{\hat{G}}(t) = \left(g\left(\mathbf{d}^*, \mathbf{\mu}_{\mathbf{x}_R} \right) + \sum_{i=1}^n \frac{\partial G}{\partial \mathbf{d}_i} \middle _{\mathbf{d}_i} \left(\mathbf{d}_i - \mathbf{d}_i^* \right) - \sum_{i=1}^n \frac{\partial G}{\partial \mathbf{X}_{\mathbf{x}_i}} \middle _{\mathbf{d}_{\mathbf{x}_i}} \mu_{\mathbf{x}_{\mathbf{x}_i}} \right)$	
	(3) $+\sum_{i=1}^n\frac{\partial G}{\partial \mathbf{X}_{p_i}}\left(\mathbf{X}_{p_i}-\mu_{\mathbf{X}_{p_i}}\right)$	$+\sum_{i=1}^n\frac{\partial G}{\partial \mathbf{X}_{\kappa i}}\Bigg _{\kappa_{\kappa}}K'_{\mathbf{X}_{\kappa i}}\left(\frac{\partial G}{\partial \mathbf{X}_{\kappa i}}\Bigg _{\kappa_{\kappa}}t\right) = 0$	
	The cumulant generating function (CGF) of XR was de- noted as $K_{x_n}(t)$. Table 1 lists the CGFs of some commonly used distributions [31]. Two useful properties of CGF are	A simple formula for computing the PDF of \hat{G} is given as follows $[15, 35]$:	(5)
given as follows $[15]$:	Property I. If $X_R = (X_{R1}, X_{R2},, X_{Rn})$ are independent random variables and their corresponding CGFs are	$f_{\hat{G}} \approx \left \frac{1}{2\pi K^{\prime\prime}_{\hat{G}}\!\left(t_{s}\right)}\right ^{1/2} e^{\left[K_{\hat{G}}\!\left(t_{s}\right)\right]} \,,$	(6)

Table 1. CGFs of some common distributions.

3. Mean-value first-order saddlepoint approximation

$$
\hat{G} = \hat{g}(\mathbf{d}, \mathbf{X}_{R}) = g(\mathbf{d}^{*}, \mathbf{\mu}_{\mathbf{X}_{R}}) + \sum_{i=1}^{n} \frac{\partial G}{\partial \mathbf{d}_{i}} \begin{bmatrix} (\mathbf{d}_{i} - \mathbf{d}_{i}^{*}) & K_{\hat{G}}'(t) = \begin{bmatrix} g(\mathbf{d}^{*}, \mathbf{\mu}_{\mathbf{X}_{R}}) + \sum_{i=1}^{n} \frac{\partial G}{\partial \mathbf{d}_{i}} \end{bmatrix}_{\mathbf{d}_{i}} (\mathbf{d}_{i} - \mathbf{d}_{i}^{*}) \\ + \sum_{i=1}^{n} \frac{\partial G}{\partial \mathbf{X}_{R}} \begin{bmatrix} (\mathbf{X}_{Ri} - \mathbf{\mu}_{\mathbf{X}_{Ri}}) & \cdots & (\mathbf{G}) & \mathbf{G} \\ \end{bmatrix}_{\mathbf{d}_{i}} \qquad (3)
$$

random variables and their corresponding CGFs are $=\sum_{i=1}^{n} X_{\text{R}i}$ is where $K''(\bullet)$ is the second *n i* $\dot{Y}_{Y}\!\left(t\right) = \sum_{i=1}^{n} K_{X_{\text{\tiny{R}i}}}\!\left(t\right).$ *i* **f** $\hat{G} = \hat{g}(\mathbf{d}, \mathbf{X}_R) = g(\mathbf{d}^*, \mathbf{H}_{\mathbf{X}_R}) + \sum_{i=1}^n \frac{\partial G}{\partial \mathbf{d}_i} \begin{bmatrix} (\mathbf{d}_i - \mathbf{d}_i^*) & \mathbf{K}'_0(t) = \begin{bmatrix} g(\mathbf{d}^*, \mathbf{H}_{\mathbf{X}_R}) + \sum_{i=1}^n \frac{\partial G}{\partial \mathbf{X}_{R_i}} \end{bmatrix} \begin{bmatrix} (\mathbf{X}_R - \mathbf{H}_{\mathbf{X}_R}) & \mathbf{K}'_0(t) =$

Property II. If X_R is a random variable and its CGF is $K_{X_p}(t)$, then the CGF of $Y = aX_p + b$ (a and b are con-

For example, if X_R follows χ^2 distribution with CGF

^ˆ *G G g* » = ^ˆ **d X**, can be expressed () ^ˆ ^ˆ , , *i i* = = + å **dX d X d μ d d** noted as () ^R *K t* **^X** Property I. If **X**R R1 R 2 R ⁼ (*X X X* , , , ^K *ⁿ*) are independent () *^X*R*ⁱ K t* (*i n* ⁼1, 2, , ^K) , then the CGF of ^R *ⁱ Y X* () () ^R () *^X*^R *K t* , then the CGF of *Y aX b* = + ^R stants)is () () *^Y ^X*^R *K t K at bt* = + . For example, if *^X* ^R () () () ^R 1 2 ln 1 2 *K t n t ^X* = - - , then the CGF of *Y* is R R R * * ˆ 1 1 R 1 R , *i i i i i n n ^G i i i i i i n i ⁱ ^G ^G K t g ^t G ^K ^t* ^m ^m ^m = = ⁼ ^æ ^ö ¶ ¶ ⁼ ^ç ⁺ - - [÷] ¶ ¶ è ø æ ö ¶ ⁺ ç ÷ ¶ è ø å **X ^X ^X ^X d ^X d μ d d d X X** , (4) () () () ^R ^R R R R R * * ˆ 1 1 R =1 R R , 0 *i i i i i i n n ^G i i i i i i n i ⁱ ⁱ ^G ^G K t g G G ^K ^t* ^m ^m ^m ^m ⁼ ⁼ ^æ ^ö ¶ ¶ ^ç [÷] ¢ ⁼ ⁺ - - ^ç ¶ ¶ [÷] è ø æ ö ¶ ¶ ç ÷ ⁺ ¢ ⁼ ¶ ¶ è ø å å å **X ^X ^X ^X ^X d ^X d μ d d d X X X** . (5) () ^ˆ () 1/ 2 *K t ^G ^s ^G ^G ^s ^f ^e* ^p *K t* é ù ë û é ù » ê ú ¢¢ ë û where ^ˆ () *^G ^K*¢¢ · is the second-order derivative of CGF. Two ^ˆ () () () ^R 1 1 ^ˆ 0 + *^G F P g w w w v* ^f æ ö = £ » F é ù - ë û ç ÷ è ø **^X** . (7)

$$
f_{\hat{G}} \approx \left[\frac{1}{2\pi K_{\hat{G}}''(t_s)}\right]^{1/2} e^{\left[K_{\hat{G}}(t_s)\right]},
$$
\n(6)

concise formulas are proposed for calculating the CDF of \hat{G} [36, 37]:

$$
F_{\hat{G}} = P\Big[\hat{g}\big(\mathbf{X}_{R}\big) \leq 0\Big] \approx \Phi\big(w\big) + \phi\big(w\big)\bigg(\frac{1}{w} - \frac{1}{v}\bigg). \tag{7}
$$

or

^G ^v F P g ^w dard normal distribution, respectively.

$$
w = \operatorname{sign}(t_s) \left\{ 2 \left[-K_{\hat{G}}(t_s) \right] \right\}^{1/2},\tag{9}
$$

and

$$
v = t_s \left[K''_{\hat{G}}(t_s) \right]^{1/2}, \qquad (10)
$$

where

sign(
$$
t_s
$$
)= $\begin{cases} 1, & \text{if } t_s > 0 \\ 0, & \text{if } t_s = 0 \\ -1, & \text{if } t_s < 0 \end{cases}$
MVFOSA uses full distribution information. Thus,

By $\mu = \frac{1}{2} \left[\hat{g}(\mathbf{X}_k) \le 0 \right] \approx \Phi \left(w + \frac{1}{w} \log \frac{v}{w} \right)$.
 $F_{\hat{o}} = P \left[\hat{g}(\mathbf{X}_k) \le 0 \right] \approx \Phi \left(w + \frac{1}{w} \log \frac{v}{w} \right)$.

(8) variables, respectively. Some that $\text{Step 2: Solve the sys}$

sign $\mathcal{E}(\mathbf{X}_k) = 0$ and $\phi(\bullet$ MVFOSA is generally more accurate than MVFOSM [15]. MVFOSA requires only a process of finding one saddlepoint without any integration or optimization. Thus, MVFOSA is employed and combined with CO in this paper to solve RBMDO problems under the aleatory uncertainty.

4. Collaborative optimization under aleatory uncertainty

4.1 SORA method

CO was introduced and developed to maintain multidisciplinary engineering characteristics [38**-**42]. CO decomposes the design problem at the system and discipline levels. At the discipline level, local constraints are satisfied while the discrepancies between the design variables and their target values are minimized. At the system level, target values are determined for design variables and the system objective is optimized.

CO has many advantages [43]. First, multidisciplinary feasibility can be maintained by using compatibility constraints. Moreover, CO enjoys discipline autonomy. Thus, discipline analysis is easy and can be processed in parallel. Finally, CO can keep disciplinary feasibility at the optimization process. In engineering applications, CO was employed for the design of launch vehicles [40, 44] and aircraft configurations [42]. CO has also been widely used in decision-making [45] and conceptual design [46]. CO has many advantages [43]. First, multidisciplinary fea-

sibility can be maintained by using compatibility constraints.

Moreover, CO enjoys discipline autonomy. Thus, discipline

analysis is easy and can be processed ibility at the optimization process. In

CO was employed for the design of

and aircraft configurations [42]. CO

d in decision-making [45] and consistent

s.t

cedure is presented as follows:

values for system-level des Example in the maintained by using compatibility constraints.

Moreover, CO enjoys discipline automory. Thus, discipline $\mathbf{d}^{\text{max}(k)}$, $\mathbf{p}^{\text{max}(k)}_{\text{x}}$, $\mathbf{p}^{\text{max}(k)}_{\text{x}}$, $\mathbf{p}^{\text{max}(k)}_{\text{x}}$, $\mathbf{p}^{\text{$ Moreover, CO enjoys discipline autonomy. Thus, discipline \mathbf{d}_{i}^{max}
analysis is easy and can be processed in parallel. Finally, CO Tl
can keep disciplinary feasibility at the optimization process. In
engineering appli O enjoys discipline autonomy. Thus, discipline $\mathbf{d}_{i}^{(x_0(x_i)}, \mathbf{\mu}_{i}^{(x_0(x_i))}$, gy and can be processed in parallel. Finally, CO The flowchart of MVF iplinary feasibility at the optimization process. In explored for

4.2 MVFOSA-CO

The MVFOSA-CO procedure is presented as follows:

Step 1: Set the initial values for system-level design variables as $\mathbf{d}_{i}^{\text{sys}(0)}$, $\mathbf{\mu}_{\mathbf{x}_{i}}^{\text{sys}(0)}$, $\mathbf{\mu}_{\mathbf{x}_{s}}^{\text{sys}(0)}$, $\mathbf{\mu}_{\mathbf{x}_{s}}^{\text{sys}(0)}$, $\mathbf{\mu}_{\mathbf{x}_{i}}^{\text{sys}(0)}$, and $\mathbf{\mu}_{\mathbf{y}_{i}}^{\text{sys}(0)}$ and vehicles [40, 44] and aircraft configurations [
been widely used in decision-making [45] a
design [46].
FOSA-CO
IVFOSA-CO procedure is presented as follow
1: Set the initial values for system-level desi
is \mathbf{d}_i^{\text sys and dis denote system level and discipline-level design and

 $\hat{g}(\mathbf{X}_{R}) \leq 0$ $\propto \Phi \left(w + \frac{1}{w} \log \frac{v}{w} \right)$, (8) $k = 1, k$ variables, respectively. Subscript *i* denotes discipline *i* . At $k = 1$, *k* denotes the *k* th cycle.

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 $\int_{\gamma} = P\left[\hat{g}(\mathbf{X}_R) \le 0\right] \approx \Phi\left(w + \frac{1}{w} \log \frac{v}{w}\right)$, (8) variables, respectively. Subscript *i* denotes discipline *i*
 $k =$ *w* eng *et al. / Journal of Mechanical Science and Technology 28 (10) (2014) 3925-3935
* $\frac{1}{w}$ *log* $\frac{v}{w}$ *, (8) variables, respectively. Subscript <i>i* denotes disc
 $k = 1$, *k* denotes the *k* th cycle.

Step 2: Solv *D. Meng et al. / Journal of Mechanical Science and Technology 28 (10) (2014) 3925~3935*
 $= P\left[\hat{g}(\mathbf{X}_R) \le 0\right] \approx \Phi\left(w + \frac{1}{w} \log \frac{v}{w}\right)$, (8) variables, respectively. Subscript *i* denotes the *k* th cycle.
 $\Phi(\bullet)$ 3928
 *B. Meng et al. / Journal of Mechanical Science and Technology 28 (10) (2014) 3925-3935
* $F_{\hat{\sigma}} = P\Big[\hat{g}(\mathbf{X}_k) \le 0\Big] \approx \Phi\Big(\frac{w + \frac{1}{w} \log \frac{v}{w}}{w}\Big),$ *

(8)* $k = 1$ *, <i>k* denotes the stan-level continization proble *D. Meng et al. / Journal of Mechanical Science and Technology 28 (10) (2014) 3925-3935*

(8) variables, respectively. Subscript *i*
 $k = 1$, *k* denotes the *k* th cycle.

Step 2: Solve the system-level of

(4) are the Step 2: Solve the system-level optimization problem (Eq. (11)). Compatibility constraints J_i should be less than or equal to ε , which is a small positive number. Eq. (11) aims to optimize the system objective and obtain the solutions of and Technology 28 (10) (2014) 3925-3935

variables, respectively. Subscript *i* denotes discipline *i*. At
 $k = 1$, *k* denotes the *k* th cycle.

Step 2: Solve the system-level optimization problem (Eq.

11)). Compatibil $\mu_{X_i}^{\text{sys}(k)}$, $\mu_{X_s}^{\text{sys}(k)}$, $\mu_{Y_{\bullet i}}^{\text{sys}(k)}$, and $\mu_{Y_{\bullet}}^{\text{sys}(k)}$. Thereafter, send 8 (10) (2014) 3925-3935
ectively. Subscript *i* denotes discipline *i*. At
otes the *k* th cycle.
we the system-level optimization problem (Eq.
tibility constraints J_i should be less than or
which is a small positive nu (4) 3925-3935

. Subscript *i* denotes discipline *i*. At
 k th cycle.

system-level optimization problem (Eq.

constraints J_i should be less than or

a small positive number. Eq. (11) aims

n objective and obtain the *3925~3935*

Subscript *i* denotes discipline *i*. At i th cycle.

tem-level optimization problem (Eq.

nstraints J_i should be less than or

small positive number. Eq. (11) aims

objective and obtain the solutions of
 these variables to discipline i at the discipline level as design parameters. logy 28 (10) (2014) 3925-3935

respectively. Subscript *i* denotes discipline *i*. At

denotes the *k* th cycle.

: Solve the system-level optimization problem (Eq.

mapatibility constraints J_i should be less than or
 denotes the *k* th cycle.
Solve the system-level optimization problem (Eq.
patibility constraints J_i should be less than or
, which is a small positive number. Eq. (11) aims
the system objective and obtain the solutions atibility constraints J_i should be less than or
which is a small positive number. Eq. (11) aims
the system objective and obtain the solutions of
 $\mu_{X_s}^{s,s,(k)}$, $\mu_{Y_{s_i}}^{s,s,(k)}$, and $\mu_{Y_{s_i}}^{s,s,(k)}$. Thereafter, send
e notes the *k* th cycle.

we the system-level optimization problem (Eq.

tibility constraints J_i should be less than or

which is a small positive number. Eq. (11) aims

e system objective and obtain the solutions of
 $\$ the system-level optimization problem (Eq.

lity constraints J_i should be less than or

th is a small positive number. Eq. (11) aims

ystem objective and obtain the solutions of
 $\mathbf{x}_s^{ss(k)}$, $\mathbf{\mu}_{\mathbf{Y}_{k_i}}^{ss(s)}$, and

28
\n28
\n29.
$$
E_{\tilde{G}} = P\left[\hat{g}(\mathbf{X}_{k}) \le 0\right] \approx \Phi\left(w + \frac{1}{w} \log \frac{v}{w}\right),
$$
\n20.
$$
M_{\text{eng}} = \Phi\left[\hat{g}(\mathbf{X}_{k}) \le 0\right] \approx \Phi\left(w + \frac{1}{w} \log \frac{v}{w}\right),
$$
\n30.
$$
E_{\tilde{G}} = \begin{cases} 0 & \text{for } w \ne 0 \\ 0 & \text{for } w \ne 0 \end{cases}
$$
\n31.
$$
W = \text{sign}(t, \left[\left(2\left[-K_{\tilde{G}}(t)\right]\right]^{1/2}, \left(\frac{1}{w} \log \frac{1}{w}\right)\right],
$$
\n31.
$$
W = \text{sign}(t, \left[\left(2\left[-K_{\tilde{G}}(t)\right]\right]^{1/2}, \left(\frac{1}{w} \log \frac{1}{w}\right)\right],
$$
\n32.
$$
W = \text{sign}(t, \left[\left(2\left[-K_{\tilde{G}}(t)\right]\right]^{1/2}, \left(\frac{1}{w} \log \frac{1}{w}\right), \left(\frac{1}{w} \log \frac{1}{w}\right), \left(\frac{1}{w}\log \frac{1}{w}\right),
$$
\n41.
$$
W = \text{sign}(t, \left[\left(\frac{k}{\tilde{G}}(t)\right]\right]^{1/2}, \left(\frac{1}{w}\log \frac{1}{w}\right),
$$
\n42.
$$
W = \text{sign}(t, \left[\frac{k}{\tilde{G}}(t, \left)\right]\right]^{1/2}, \left(\frac{1}{w}\log \frac{1}{w}\right),
$$
\n43.
$$
W = \text{sign}(t, \left[\frac{k}{\tilde{G}}(t, \left)\right]\right]^{1/2},
$$
\n5

Step 3: Solve the discipline-level optimization problems (Eq. (12)), where discipline optimization problem tasks minimize J_i , which denotes the compatibility constraints at the system level. In the discipline optimization process, systemlevel design variables are treated as design parameters, and discipline optimization and reliability analysis are nested. Reliability analysis consists of three steps: (1) linearize the Fig. 2. Solve the system-level optimization problem (Eq. (11)). Compatibility constraints J_x should be less than or equal to *ε*, which is a small positive number. Eq. (11) aims to optimize the system objective and bo $\left(\mu_{Y_n}^{s s(k)}, \mu_{Y_n}^{s s(k)}, \mu_{Y_n}^{s s(k)}\right)^2$
 $\left(\mu_{Y_n}^{s s(k)} - \mu_{Y_n}^{dis(k-1)}\right)^2$ (11)
 $\left(\mu_{Y_n}^{s s(k)} - \mu_{Y_n}^{dis(k-1)}\right)^2$ (11)
 $\left(\mu_{Y_n}^{s s(k)} - \mu_{Y_n}^{dis(k-1)}\right)^2$ (11)
 $\left(\mu_{Y_n}^{s s(k)} - \mu_{Y_n}^{dis(k-1)}\right)^2 \leq \varepsilon, i = 1, 2, ..., n$

scriptine $+(\mathbf{\mu}_{x_k}^{\text{sys}(k)} - \mathbf{\mu}_{x_k}^{\text{dis}(k-1)})^2 + (\mathbf{\mu}_{x_{\ell}}^{\text{sys}(k)} - \mathbf{\mu}_{x_{\ell}}^{\text{dis}(k-1)})^2$
 $+(\mathbf{\mu}_{y_{\ell}}^{\text{sys}(k)} - \mathbf{\mu}_{y_{\ell}}^{\text{dis}(k-1)})^2 \leq \varepsilon, i = 1, 2, ..., n$

Step 3: Solve the discipline-level optimization problems

(Eq. (12)), + $(\mathbf{\mu}_{x_k}^{\text{w}_k(k)} - \mathbf{\mu}_{x_k}^{\text{w}_k(k-1)})$ + $(\mathbf{\mu}_{y_k}^{\text{w}_k(k-1)})^2 \leq \varepsilon$, $i = 1, 2, ..., n$
 $+(\mathbf{\mu}_{y_k}^{\text{w}_k(k)})^2 \leq \varepsilon$, $i = 1, 2, ..., n$

Step 3: Solve the discipline-level optimization problems

(Eq. (12)), where discipl $\mu_{\mathbf{x}_{\mathbb{R}}}^{\text{sys},(k)}$ by using firstorder Taylor expansion; (2) calculate the CGF of the performance function \hat{G} (Eq. (4)); (3) estimate the CDF and PDF of the performance function \hat{G} (Eqs. (6)-(8)). Step 3: Solve the discipline-level optimization problems
q. (12)), where discipline optimization problem tasks mini-
ze J_i , which denotes the compatibility constraints at the
stem level. In the discipline optimization p Step 1. Solve the distipline primidiation problem tasks mini-
ize J_i , which denotes the compatibility constraints at the
lize J_i , which denotes the compatibility constraints at the
system level. In the discipline opti 12), where discipline potimization problem tasks mini-
 J_i , which denotes the compatibility constraines a mini-
 J_i , which denotes the compatibility constraints at the

level. In the discipline optimization process, s Free discipline optimization problems
there discipline optimization problems have
the denotes the compatibility constraints at the
In the discipline optimization process, system-
variables are treated as design parameters **i** discripting optimization problem takes mini-
denotes the compatibility constraints at the
discipline optimization process, system-
the discipline optimization process, system-
iables are treated as design parameters, Step 3: Solve the discipline-level optimization problems
(Eq. (12)), where discipline optimization problem tasks mini-
mize J , which denotes the compatibility constraints at the
system level. In the discipline optimizat **Eq.** (12), where discipline-level optimization problem has mini-
Eq. (12), where discipline optimization problem tasks mini-
Fq. (12), where discipline optimization problem tasks mini-
nize J_i , which denotes the compat containty dimits of the consumed of the system (1, phence the system) and the system of discipline-level random variables $\mathbf{u}_{x_k}^{s s_k(k)}$ (Eq. (3)) at the system of discipline-level random variables $\mathbf{u}_{x_k}^{s s_k(k)}$ by diplomation $G \approx \hat{G} = \hat{g}(\mathbf{d}, \mathbf{X}_R)$ (Eq. (3)) at the function $G \approx \hat{G} = \hat{g}(\mathbf{d}, \mathbf{X}_R)$ (Eq. (3)) at the ministic variables $\mathbf{d}^{\text{sys}(k)}$ and the mean values even function variables $\mathbf{d}^{\text{sys}(k)}$ by usin Solved by the set of the set of $G \approx \hat{G} = \hat{g}(\mathbf{d}, \mathbf{X}_R)$ (Eq. (3)) at the variables $\mathbf{d}^{\text{sys}(k)}$ and the mean values
ndom variables $\mathbf{\mu}_{\mathbf{x}_N}^{\text{sys}(k)}$ by using first-
n; (2) calculate the CGF of the perform *i* $G \approx \hat{G} = \hat{g}(\mathbf{d}, \mathbf{X}_{R})$ (Eq. (3)) at the variables $\mathbf{d}^{\text{v}_S(k)}$ and the mean values
om variables $\mathbf{d}^{\text{v}_S(k)}$ and the mean values
om variables $\mathbf{\mu}_{X_{Ri}}^{\text{vs}(k)}$ by using first-
(2) calculate the C liability analysis consists of three steps: (1) linearize the
formance function $G \approx \hat{G} = \hat{g}(\mathbf{d}, \mathbf{X}_k)$ (Eq. (3)) at the
uss of deterministic variables $\mathbf{d}^{w_k(i)}$ and the mean values
discipline-level random vari **Example 11 Y Y Y C Example 1 Y Y Y Eq.** (1) linearize the function $G \approx \hat{G} = \hat{g}(\mathbf{d}, \mathbf{X}_R)$ (Eq. (3)) at the terministic variables $\mathbf{d}^{\text{sys}(k)}$ by using first-level random variables \mathbf{d}^{\text we the disciplime-level optimization problems
rec discipline optimization problem tasks mini-
denotes the compatibility constrains at the
heloces the compatibility constrains at the
nication are reacted as design paramete

, $\boldsymbol{\mu}_{\mathbf{X}_i}^{\text{dis},(k)}$, $\boldsymbol{\mu}_{\mathbf{X}_s}^{\text{dis},(k)}$, $\boldsymbol{\mu}_{\mathbf{Y}_{\bullet i}}^{\text{dis},(k)}$, and $\boldsymbol{\mu}_{\mathbf{Y}_k}^{\text{dis}}$ $\mu_{\mathbf{x}_s}^{\text{dis},(k)}$, $\mu_{\mathbf{x}_i}^{\text{dis},(k)}$, and $\mu_{\mathbf{x}_i}^{\text{dis},(k)}$ to the system level and calculate compatibility constraints. If all compatibility consystem objective function is stable, go to Step 5; otherwise, set r expansion; (2) calculate the CGF of the perform-

on \hat{G} (Eq. (4)); (3) estimate the CDF and PDF of

ance function \hat{G} (Eqs. (6)-(8)).

Theck the convergence. Send the values of $\mathbf{d}_{i}^{dis,(k)}$,
 (\hat{k}) , $\mathbf{\mu}_{$ alculate the CGF of the perform-

3) estimate the CDF and PDF of

(Eqs. (6)-(8)).

ence. Send the values of $\mathbf{d}_{i}^{dis,(k)}$,
 $\mathbf{u}_{\mathbf{v}_{i}}^{dis,(k)}$ to the system level and

raints. If all compatibility con-

2,..., *n* an of the perform-

TDF and PDF of

alues of $\mathbf{d}_i^{\text{dis}(k)}$,

stem level and

mpatibility con-

e value of the

5; otherwise, set

rt final solutions

.

Fig. 1.
 $\left(\mathbf{\mu}_{x_k}^{\text{sys}(k)} - \mathbf{\mu}_{x_k}^{\text{dis}(k)}\right)^2$
 $\begin{bmatrix} \geq [R_$, $\mathbf{\mu}_{x_i}^{\text{dis},(k)}$, and $\mathbf{\mu}_{x_i}^{\text{dis},(k)}$ to the system level and
patibility constraints. If all compatibility con-

y $J_i \leq \varepsilon$, $i = 1, 2, ..., n$ and the value of the

ive function is stable, go to Step 5; otherwise μ_{x_k} , μ_{x_k} , μ_{x_k} , and μ_{x_k} to the system level is

compatibility constraints. If all compatibility c

satisfy $J_i \le \varepsilon$, $i = 1, 2, ..., n$ and the value of

bjective function is stable, go to Step 5; otherwise

Step 5: Stop the optimization process. Export final solutions $\mu_{X_i}^{\text{sys}(k)}$, $\mu_{X_i}^{\text{sys}(k)}$, $\mu_{Y_{i}}^{\text{sys}(k)}$, $\mu_{Y_{i}}^{\text{sys}(k)}$, and f .
The flowchart of MVFOSA-CO is shown in Fig. 1.

On under aleatory uncer- Reliability analysis consists of three steps: (1) linearize the performance function
$$
G \approx \hat{G} = \hat{g}(\mathbf{d}, \mathbf{x}_k)
$$
 (Eq. (3)) at the values of deterministic variables $\mathbf{d}^{m(k)}$ and the mean values of discipline-level random variables $\mathbf{u}^{m(k)}$ by using first-
lsloped to maintain multidisci- order Taylor expansion; (2) calculate the GFG the performance function \hat{G} (Eq. (4)); (3) estimate the CDF and PDF of a machine nucleus. At the performance function \hat{G} (Eqs. (6)-(8)).
as are satisfied while the distance of random variables and their target values are deter- calculate $\mathbf{u}^{m(k)}$, $\mathbf{u}^{m(k)}$, $\mathbf{u}^{m(k)}$, and $\mathbf{u}^{m(k)}$ to the system level and level, target values are deter- calculate complexity constants. If all compatibility constraints. If all compatibility constraints is a single, $J, \leq \varepsilon$, $i = 1, 2, ..., n$ and the value of the system objective function is stable, go to Step 5; otherwise, set i .
First, multidisciplinary fea $k = k + 1$ and go to Step 2.
using compatibility constraints. Step 5: Stop the optimization process. Export final solutions a two-
equation process. In particular, $\mathbf{u}^{m(k)}$, $\mathbf{u}^{m(k)}$, $\mathbf{u}^{m(k)}$, $\mathbf{u}^{m(k)}$, $\mathbf{u}^{m(k)}$, and f .
3. The optimization process. In a standard form $J_{\tau} = (\mathbf{d}^{m(k)}_{\tau} - \mathbf{d}^{m(k)})^2 + (\mathbf{u}^{m(k)}_{\tau} - \mathbf{u}^{m(k)})^2 + (\mathbf{u}^{m(k)}_{\tau} - \mathbf{u}^{m(k)})^2$ (therefore $J = \frac{1}{2} \left[\mathbf{d} \mathbf$

.

(12)

5. Examples

Two examples are used to show the accuracy and efficiency of the proposed method. MVFOSA-CO, MVFOSM-based CO

Fig. 1. Flowchart of MVFOSA-CO.

(MVFOSM-CO), and MCS-based CO (MCS-CO) are compared. The results obtained by MCS-CO are used as reference for accuracy comparison.

5.1 Mathematical example

The mathematical problem is provided as a simple test problem. The integrated framework of the design optimization problem is given as follows:

Output	System level optimization											
End	$\mu_{x_i^{m_1}}, \mu_{x_i^{m_2}}, \mu_{y_i^{m_3}}, \mu_{y_i^{m_4}}, \mu_{y_i^{m_4}}$	J_1	J_2	J_2	J_3							
IVFOSM-CO), and MCS-based CO (MCS-CO) are com- red. The results obtained by MCS-CO are used as reference	$\mu_{x_i^{m_i}}, \mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}$	$\mu_{x_i^{m_i}}$
1	Mathematical example	$\mu_{x_i^{m_i}}, \mu_{x_i^{m_i}}, \mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}$	$\mu_{x_i^{m_i}}$	$\mu_{x_i^{m_i}}$	$\mu_{x_i^{m_i}}$	$\mu_{x_i^{m_i}}$					
1	Mathematical example	$\mu_{x_i^{m_i}}, \mu_{x_i^{m_i}}, \mu_{x_i^{m_i}}, \mu_{y_i^{m_i}}$	$\mu_{x_i^{m_i}}$	$\mu_{x_i^{m_i}}$	$\mu_{x_i^{m_i$							

where *f* is the system objective, and x_1 , x_2 , x_3 , y_{12} and y_{21} are the design variables.

The problem is modified into a MDO problem including two disciplines given in Fig. 2. In the modified problem, coupling variables y_{12} and y_{21} affect each discipline. The distribution information of all random design variables is given in Table 2. We assume two distribution types: the normal and Gumbel distribution. The formulations of the MVFOSA-CO optimization problem including two disciplines are provided in Eqs. (14)-(16).

The MVFOSA-CO approach analysis for the mathematical example is provided in Fig. 3.

Table 2. Distribution information of random variables in the mathematical problem.

		and Technology 28 (10) (2014) 3925~3935	Table 2. Distribution information of random variables in the mathe-	3929
matical problem.				
Variables	Mean	Standard deviation	Distribution 1	Distribution 2
x_{1}	$\mu_{\rm r_1}$	$0.001\mu_{x_1}$	Normal	Gumbel
x_{2}	μ_{x_2}	$0.001\mu_{x_2}$	Normal	Gumbel
x_{3}	$\mu_{\rm x_3}$	$0.001\mu_{\rm x_3}$	Normal	Gumbel
y_{12}	μ_{y_1}	$0.001\mu_{y_{12}}$	Normal	Gumbel
y_{21}	$\mu_{y_{21}}$	$0.001\mu_{y_{21}}$	Normal	Gumbel

Fig. 3. MVFOSA-CO approach analysis for the mathematical example.

(1) System optimization problem

$$
\begin{array}{|c|c|c|}\n\hline\n\text{Discipline analysis 1} & \text{Discipline analysis 2} \\
\hline g. 3. MVFOSA-CO approach analysis for the mathematical example. \\
\hline\n\end{array}
$$
\n(1) System optimization problem

\nFind $\mu_{x_1^{p_3}}, \mu_{x_2^{p_3}}, \mu_{x_3^{p_3}}, \mu_{y_1^{p_3}}, \mu_{y_2^{p_3}}$

\nmin $f = \left(\mu_{y_1^{p_3}} - 1\right)^2 + \left(\mu_{x_1^{p_3}}\right)^2 + \left(\mu_{x_2^{p_3}}\right)^2$

\ns.t. $J_1 \leq 0.01$, $J_2 \leq 0.01$

\n(2) Discipline 1 optimization problem

\nFind $\mu_{x_1^{dis1}}, \mu_{x_2^{dis1}}, \mu_{y_1^{dis1}}, \mu_{y_2^{dis1}})$

\nmin $J_1 = \left(\mu_{x_1^{p_3}} - \mu_{x_1^{dis1}}\right)^2 + \left(\mu_{x_2^{p_3}} - \mu_{x_2^{dis1}}\right)^2 + \left(\mu_{y_2^{p_3}} - \mu_{y_2^{dis1}}\right)^2 + \left(\mu_{y_1^{p_3}} - \mu_{y_2^{dis1}}\right)^2$

\nfor $\mu_{x_1^{p_3}} = \mu_{y_1^{p_3}} \text{ and } \mu_{y_1^{p_3}} = \mu_{y_2^{p_3}} \text{ and } \mu_{y_1^{p_3}} = \mu_{y_2^{p_3}} \text{ and } \mu_{y_1^{p_3}} = \mu_{y_1^{p_3}} \text{ and } \mu_{y_1^{p_3}} = \mu_{y_2^{p_3}} \text{ and } \mu_{y_1^{p_3}} = \mu_{y_1^{p_3}} \text{ and } \mu_{y_1^{p_3}} = \mu_{y$

(2) Discipline 1 optimization problem

$$
\mu_{x_1^{(n_1)}}, \mu_{x_2^{(n_2)}}, \mu_{y_1^{(n_2)}}, \mu_{y_1^{(n_1)}} \n\hline\nF_{x_1^{(n_1)}}, \mu_{y_1^{(n_1)}}, \mu_{y
$$

Table 3. Optimization results of the mathematical example (Case 1: Normal distribution).

	μ_{x}^{sys}	$\mu_{\rm xgs}$	μ x_2 ^{sys}	$\mu_{y_{12}^{\rm sys}}$	$y_{21}^{\rm sys}$	
MVFOSA-CO	-0.3177	0.3142	0.9056	0.4133	0.5251	3.6312
MVFOSM-CO	-0.3177	0.3142	0.9056	0.4133	0.5251	3.6312
MCS-CO	-0.3177	0.3142	0.9057	0.4134	0.5252	3.6314

Table 4. Reliability of the probabilistic constraints and calculation efficiency of the mathematical example (Case 1: Normal distribution).

			n.	n_{γ}		Calculation time
MVFOSM-CO	0.9739	0.9864	5924	4444	50	$5 \text{ min } 50 \text{ s}$
MVFOSA-CO	0.9739	0.9864	5924	4444	50	$5 \text{ min } 50 \text{ s}$
MCS-CO	0.9831	0.9907	5930	4408	50	

Table 5. Optimization results of the mathematical example (Case 2: Gumbel distribution).

	$\mu_{\rm x^{sys}}$	$\mu_{\rm x_s^{\rm sys}}$	$\mu_{\chi^{\rm sys}}$	$v_{y_1^{sys}}$	$\mu_{\substack{sys \\ y_{21}^{\rm sys}}}$	
MVFOSM-CO	-0.3106	0.3051	0.9113	0.4231	0.5098	3.6609
MVFOSA-CO	-0.3177	0.3142	0.9057	0.4133	0.5251	3.6313
MCS-CO	-0.3177	0.3142	0.9056	0.4134	0.5252	3.6315

Table 6. Reliability of probabilistic constraints and calculation efficiency of the mathematical example (Case 2: Gumbel distribution).

(3) Discipline 2 optimization problem

Find
$$
\mu_{x_3^{dis2}}, \mu_{y_{12}^{dis2}}, \mu_{y_{21}^{dis2}})
$$

\nmin $J_2 = (\mu_{x_3^{dis2}}, -\mu_{x_3^{dis2}})^2 + (\mu_{y_{12}^{dis2}}, -\mu_{y_{12}^{dis2}})^2$
\n $+ (\mu_{y_{21}^{dis2}}, -\mu_{y_{21}^{dis2}})^2$
\n
\ns.t. $P_2 [(x_3^{dis2})^2 + (y_{12}^{dis2})^2 + y_{21}^{dis2} - 5 \le 0] \ge 0.96,$
\n $0 \le \mu_{x_3^{dis2}} \le 5, \quad 0 \le \mu_{y_{12}^{dis2}} \le 10,$
\n $0 \le \mu_{y_{21}^{dis2}} \le 10, \quad \mu_{y_{12}^{dis2}} = y_{21} (\mu_{x_3^{dis2}}, \mu_{y_{12}^{dis2}}).$ (16)

The target reliability of each probabilistic constraint is 0.96. System optimization is solved by using sequential quadratic programming (SQP), and the discipline optimization problems are solved by using a genetic algorithm (GA). The results from different methods are given in Tables 3-6. A sufficiently large number of simulations $(10⁶)$ is used; thus, the MCS-CO results are considered accurate references. R_1 and R_2 are the reliabilities of probabilistic constraints 1 and 2, respectively; n_s is the number of iterations in system optimization problem; n_1 and n_2 are the numbers of iterations in discipline optimization 1 and 2, respectively.

In Tables 3 and 4, for the special case where all random variables are normally distributed, MVFOSA-CO and MVFOSM-CO produce similar results and have the same values for system and discipline analyses because MVFORM is a special case of MVFOSA [15]. In Tables 5 and 6, MVFOSA-CO generates more accurate results than MVFOSM-CO. However, both methods have almost the same efficiency because MVFOSA-CO uses the complete distribution information of random variables and has good performance in the tail regions.

5.2 Speed reducer design

The second problem is derived from NASA MDO evaluation examples [47]. This problem represents the design of a speed reducer and is posed as an artificial multidisciplinary problem comprising three subsystems: subsystem 1 (Bearing group 1 and Shaft 1), subsystem 2 (Bearing group 2 and Shaft 2), and subsystem 3 (Gear 1 and Gear 2) (Fig. 4).

The problem has three sharing design variables. We assume that aleatory uncertainty is associated with some input variables. All random variables are described by Gumbel distribution. The details of the design variables are given in Table 7.

The system objective f is to minimize the speed reducer volume. The system objective and constraints are as follows:

Fig. 4. Speed reducer design.

$$
\min f = 0.7854x_1x_2^2 \left(3.333x_3^2 + 14.933x_3 - 43.0934 \right) - 1.508x_1 \left(x_6^2 + x_7^2 \right) + 7.477 \left(x_6^3 + x_7^3 \right) + 0.7854 \left(x_4 x_6^2 + x_5 x_7^2 \right).
$$

s.t. $g_1 = 27 / (x_1 x_2^2 x_3) - 1 \le 0$: Upper bound on the bending stress of the gear tooth.

 $g_2 = 397.5 / (x_1 x_2^2 x_3^2) - 1 \le 0$: Upper bound on the contact stress of the gear tooth.

 $g_3 = 1.93x_4^3 / (x_2x_3x_6^4) - 1 \le 0$, $g_4 = 1.93x_5^3 / (x_2x_3x_7^4) - 1 \le 0$: Upper bounds on the transverse deflection of the shaft.

 $g_5 = A_1 / B_1 - 1100 \le 0$, $g_6 = A_2 / B_2 - 850 \le 0$: Upper bounds on the stresses of the shaft,

$$
A_{1} = \left[\left(\frac{745x_{4}}{x_{2}x_{3}} \right)^{2} + 16.9 \times 10^{6} \right]^{0.5},
$$

\n
$$
A_{2} = \left[\left(\frac{745x_{5}}{x_{2}x_{3}} \right)^{2} + 157.5 \times 10^{6} \right]^{0.5}, \quad B_{1} = 0.1x_{6}^{3}, \quad B_{2} = 0.1x_{7}^{3}.
$$

\n
$$
g_{7} = x_{1}x_{2} - 40 \le 0, \quad g_{8} = x_{1}/x_{7} - 12 \le 0,
$$

 $g_9 = -x_1 / x_2 + 5 \le 0$: Dimensional restrictions based on space.

 $g_{10} \sim g_{23}$: Dimensional restrictions of design variables (Table 7).

 $g_{24} = (1.5x_6 + 1.9) / x_4 - 1 \le 0$, $g_{25} = (1.1x_7 + 1.9) / x_5 - 1 \le 0$: Design condition for the shaft based on experience.

The MVFOSA-CO analysis for the speed reducer MDO problem is shown in Fig. 5.

MVFOSA-CO optimization problem formulations, including three disciplines are provided in Eqs. (17)-(20).

(1) System optimization problem

Find
$$
x_1^{\text{sys}}, x_2^{\text{sys}}, x_3^{\text{sys}}, \mu_{x_4^{\text{sys}}}, \mu_{x_5^{\text{sys}}}, \mu_{x_6^{\text{sys}}}, \mu_{x_7^{\text{sys}}}
$$

\n
$$
\min f = 0.7854x_1^{\text{sys}} \left(x_2^{\text{sys}} \right)^2 \left(3.333 \left(x_3^{\text{sys}} \right)^2 + 14.933x_3^{\text{sys}} - 43.0934 \right) \n-1.508x_1^{\text{sys}} \left(\left(\mu_{x_6^{\text{sys}}} \right)^2 + \left(\mu_{x_7^{\text{sys}}} \right)^2 \right) + 7.477 \left(\left(\mu_{x_6^{\text{sys}}} \right)^3 + \left(\mu_{x_7^{\text{sys}}} \right)^3 \right) \n+ 0.7854 \left(\mu_{x_6^{\text{sys}}} \left(\mu_{x_6^{\text{sys}}} \right)^2 + \mu_{x_7^{\text{sys}}} \left(\mu_{x_7^{\text{sys}}} \right)^2 \right)
$$
\ns.t. $J_1 < \varepsilon$, $J_2 < \varepsilon$, $J_3 < \varepsilon$

 (17)

(2) Optimization problem for discipline 1

Find
$$
x_1^{\text{disl}}
$$
, x_2^{disl} , x_3^{disl} , $\mu_{x_4^{\text{disl}}}$ and $\mu_{x_5^{\text{disl}}}$
\n
$$
\min J_1 = (x_1^{\text{sys}} - x_1^{\text{disl}})^2 + (x_2^{\text{sys}} - x_2^{\text{disl}})^2 + (x_3^{\text{sys}} - x_3^{\text{disl}})^2 + (\mu_{x_5^{\text{sys}}} - \mu_{x_5^{\text{disl}}})^2
$$
\n
$$
+ (\mu_{x_5^{\text{sys}}} - \mu_{x_4^{\text{disl}}})^2 + (\mu_{x_5^{\text{sys}}} - \mu_{x_5^{\text{disl}}})^2
$$
\n
$$
\text{s.t. } P_1 \left[g_3 \left(x_2^{\text{disl}} , x_3^{\text{disl}} \right) \le 0 \right] \ge 0.95, \ P_2 \left[g_5 \left(x_2^{\text{disl}} , x_3^{\text{disl}} \right) \le 0 \right] \ge 0.95,
$$
\n
$$
P_3 \left[g_{24} \left(x_4^{\text{disl}} , x_6^{\text{disl}} \right) \le 0 \right] \ge 0.95,
$$
\n
$$
g_1 \left(x_1^{\text{disl}} , x_2^{\text{disl}} \right) \le 0, \ g_2 \left(x_1^{\text{disl}} , x_2^{\text{disl}} \right) \le 0, \ g_5 \left(x_2^{\text{disl}} , x_3^{\text{disl}} \right) \le 0, \ g_6 \left(x_1^{\text{disl}} , x_2^{\text{disl}} \right) \le 0 \tag{18}
$$

(3) Optimization problem for discipline 2

Find
$$
x_1^{\text{dis2}} \cdot x_2^{\text{dis2}} \cdot x_3^{\text{dis2}}
$$

\n
$$
\min J_2 = (x_1^{\text{sys}} - x_1^{\text{dis2}})^2 + (x_2^{\text{sys}} - x_2^{\text{dis2}})^2 + (x_3^{\text{sys}} - x_3^{\text{dis2}})^2
$$
\n
$$
\text{s.t.} \quad g_1(x_1^{\text{dis2}}, x_2^{\text{dis2}}, x_3^{\text{dis2}}) \le 0, \quad g_2(x_1^{\text{dis2}}, x_2^{\text{dis2}}, x_3^{\text{dis2}}) \le 0,
$$
\n
$$
g_7(x_2^{\text{dis2}}, x_3^{\text{dis2}}) \le 0, \quad g_8(x_1^{\text{dis2}}, x_2^{\text{dis2}}) \le 0, \quad g_9(x_1^{\text{dis2}}, x_2^{\text{dis2}}) \le 0 \tag{19}
$$

(4) Optimization problem for discipline 3

Find
$$
x_1^{\text{dis3}}, x_2^{\text{dis3}}, x_3^{\text{dis3}}, \mu_{x_3^{\text{dis3}}}, \mu_{x_3^{\text{dis3}}})
$$

\n
$$
f(x_1^{\text{sys}} - x_1^{\text{dis3}})^2 + (x_2^{\text{sys}} - x_2^{\text{dis3}})^2 + (x_3^{\text{sys}} - x_3^{\text{dis3}})^2 + (\mu_{x_3^{\text{sys}}} - \mu_{x_3^{\text{dis3}}})^2
$$
\n
$$
+ (\mu_{x_3^{\text{sys}}} - \mu_{x_3^{\text{dis3}}})^2 + (\mu_{x_3^{\text{sys}}} - \mu_{x_3^{\text{dis3}}})^2
$$
\n
$$
f(x_1^{\text{dis3}}, x_3^{\text{dis3}}) \le 0 \ge 0.95, P_5 \left[g_6 \left(x_3^{\text{dis3}}, x_3^{\text{dis3}} \right) \le 0 \right] \ge 0.95, P_6 \left[g_{25} \left(x_3^{\text{dis3}}, x_3^{\text{dis3}} \right) \le 0 \right] \ge 0.95, P_6 \left[g_{35} \left(x_3^{\text{dis3}}, x_3^{\text{dis3}} \right) \le 0 \right] \ge 0.95, P_6 \left[g_{45} \left(x_3^{\text{dis3}}, x_3^{\text{dis3}} \right) \le 0, P_6 \left[x_3^{\text{dis3}}, x_3^{\text{dis3}} \right] \le 0, P_6 \left(x_3^{\text{dis3}}, x_3^{\text{dis3}} \right) \le 0, P_6 \left(x_3
$$

The required reliability for each probability constraint is 0.95, and the compatibility constraint accuracy ε is 0.001. System optimization is solved by using SQP, and discipline optimization problems are solved by using GA. The accuracy and efficiency of MVFOSA-CO are compared with MVFOSM-CO and MCS-CO in Tables 8 and 9. $R(i=1 \sim 6)$ denotes the reliabilities of probabilistic constraints. The objective function optimization histories are shown in Fig. 6 by using three different methods. The methods have similar optimization histories, as well as similar number of iterations n . in the system optimization problem and iterations n_1 , n_2 ,

Variables	Description	Mean	Standard deviation	Distribution	Lower bound	Upper bound
x_{1}	Gear face width				2.6	3.6
x_2	Teeth module		$\overline{}$		0.3	1.0
x_{3}	Number of teeth of pinion				17	28
$x_{\scriptscriptstyle 4}$	Distance between bearings 1	μ_{x_4}	$0.001\mu_{x}$	Gumbel	7.3	8.3
x_{ς}	Distance between bearings 2	$\mu_{\rm x}$	$0.001\mu_{\rm x}$	Gumbel	7.3	8.3
x_{6}	Diameter of shaft 1	$\mu_{\mathbf{x}_{6}}$	$0.001\mu_{\rm x}$	Gumbel	2.9	3.9
x_{7}	Diameter of shaft 2	μ_{x}	$0.001\mu_{x}$	Gumbel		5.5

Table 7. Details of the design variables in the speed reducer design problem.

Table 8. Optimization results of the reducer design.

	\sim Sys \mathcal{N}	\sim Sys λ_{2}	\sim Sys	$\mu_{\rm sys}$	$\mu_{\rm,sys}$	$\mu_{\tiny\rm sys}$	$\mu_{\rm sw}$	
MVFOSM-CO	3.4273	0.6504	18	7.3003	7.6884	3.3208	5.2642	2888.5820
MVFOSA-CO	3.4247	0.6457	18	7.3005	7.6865	3.3231	5.2635	2866.0874
MCS-CO	3.4236	0.6442	18	7.3005	7.6856	3.3235	5.2633	2859.1180

Table 9. Reliability of probabilistic constraints and calculation efficiency of the reducer design.

Fig. 5. MVFOSA-CO approach for the speed reducer MDO problem.

Fig. 6. Optimization histories of three different methods.

and n_1 in discipline optimization problems 1 to 3 (Table 9). This result is attributed to the use of the same MDO method(i.e., the CO method). Furthermore, the optimal solution 2866.0874 obtained by MVFOSA-CO is closer to the reference value 2859.1180 obtained by MCS-CO than the solution 2888.5820 obtained by MVFOSM-CO (Table 8). MVFOSM-CO has the more conservative solution than MVFOSA-CO. MVFOSA-CO uses full distribution information rather than the first two moments of the random variables; thus, MVFOSA-CO performs relatively better in the tail regions. The solution obtained by MVFOSA-CO is more accurate than MVFOSM-CO. Their calculation times are almost the same (Table 9). This example shows that MVFOSA-CO has the same efficiency as MVFOSM-CO but is highly accurate.

6. Conclusions

This work aims to improve reliability analysis accuracy in MDO problem performance functions that are expensive and only respond to traditional MVFOSM methods. MVFOSA-CO is proposed to address RBMDO problems under aleatory uncertainty. The proposed method introduces the first-order Taylor expansion of a performance function at the mean values of random variables and uses saddlepoint approximation to estimate CDF and PDF. MVFOSA-CO has several advantages. First, the bi-level analysis and coordination structure of MVFOSA-CO allows the application of different subspace optimizers among various analysis groups. Different disciplines are easily parallelized and well-suited for conventional discipline organizations. Second, MVFOSA-CO has high accuracy in tail regions while keeping the same efficiency as the MVFOSM. Finally, in MVFOSA-CO, non-normal random variables do not need to be transformed into normal random variables. However, the proposed method is only suitable for RBMDO problems that are under aleatory uncertainty and that have the analytical CGF of the random variable.

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Debiao Meng is a Ph.D. student at the University of Electronic Science and Technology of China. He received his B.S. degree in Mechanical Engineering from Northwest A&F University. His main research interests include reliability-based design and optimization and reliability-based multidisciplinary de-

sign and optimization.

Hong-Zhong Huang is the Dean of the School of Mechanical, Electronic, and Industrial Engineering at the University of Electronic Science and Technology of China. He received his Ph.D. in Reliability Engineering from Shanghai Jiaotong University, China. He has published 150 journal papers and 5 books

on reliability engineering. He has held visiting appointments at several universities in the United States, Canada, and Asia. He received the Golomski Award from the Institute of Industrial Engineers in 2006. His current research interests include system reliability analysis, warranty, maintenance planning and optimization, and computational intelligence in product design.

Zhonglai Wang received his Ph.D. in Mechatronics Engineering from the University of Electronic Science and Technology of China, where he is currently an associate professor. He was a visiting scholar in the Department of Mechanical and Aerospace Engineering of Missouri University of Science and

Technology from 2007 to 2008. His research interests include reliability-based design and robust design.

Ning-Cong Xiao received his Ph.D. in Mechatronics Engineering from the University of Electronic Science and Technology of China, where he is currently a lecturer. He was a visiting scholar in the Department of Industrial and Systems Engineering of Rutgers University from 2011 to 2012. His re-

search interests include reliability-based design and robust design.

Xiao-Ling Zhang received her Ph.D. in Mechatronics Engineering from the University of Electronic Science and Technology of China, where she is currently a lecturer. She was a visiting scholar in the Department of Mechanical and Aerospace Engineering of Rutgers University from 2009 to 2011. Her re-

search interests include reliability-based multidisciplinary design and optimization.