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Fuzzy Sliding Mode Active Disturbance Rejection Control of an Autonomous Underwater Vehicle-Manipulator System

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Abstract In this paper, a fuzzy sliding mode active disturbance rejection control (FSMADRC) scheme is proposed for an autonomous underwater vehicle-manipulator system (AUVMS) with a two-link and three-joint manipulator. First, the AUVMS is separated into nine subsystems, and the combined effects of dynamic uncertainties, hydrodynamic force, unknown disturbances, and nonlinear coupling terms on each subsystem are lumped into a single total disturbance. Next, a linear extended state observer (LESO) is presented to estimate the total disturbance. Then, a sliding mode active disturbance rejection control (SMADRC) scheme is proposed to enhance the robustness of the control system. The stability of the SMADRC and the estimation errors of the LESO are analyzed. Because it is difficult to simultaneously adjust several parameters for a LESO-based SMADRC scheme, a fuzzy logic control (FLC) scheme is used to formulate the FSMADRC to determine the appropriate parameters adaptively for practical applications. Finally, two AUVMS tasks are illustrated to test the trajectory tracking performance of the closed-loop system and its ability to reject and attenuate the total disturbance. The simulation results show that the proposed FSMADRC scheme achieves better performance and consume less energy than conventional PID and FLC techniques.

Key words Fuzzy sliding mode active disturbance rejection controller (FSMADRC); linear extended state observer (LESO); autonomous underwater vehicle-manipulator system (AUVMS); total disturbance; fuzzy logic control

1 Introduction

More than 70% of the earth is covered by water and with the increase of the world's population and shortage of natural resources onshore, more countries have been searching for additional resources offshore. However, because of the complexity of the underwater environment and unknown hazards, most of the underwater world still remains unreachable for humans. To make underwater resources more accessible to humans, scientists and researchers have been examining ways and means to make deepsea explorations possible. Therefore, many underwater robotic vehicles, such as remotely operated vehicles (ROVs) and autonomous underwater vehicles (AUVs), have been developed to reach areas that are too hazardous for humans to directly explore. ROVs, however, are not suitable for nontrivial and precise tasks due to their high operation cost and high operator skills requirements (Marani *et al*., 2009; Mohan and Kim, 2015). AUVs provide solutions to the problems experienced with ROVs and create other challenges that researchers must deal with in underwater missions, such as maintenance of underwater pipelines, deepsea explorations, and military applications (Korkmaz *et al*., 2013; Yang *et al*., 2016; Wang *et al*., 2017; Li *et al*., 2018; Xu *et al*., 2019). The most important characteristic of AUVs is their ability to perform precise tasks by an appended manipulator system. The autonomous underwater vehiclemanipulator system (AUVMS) can perform delicate tasks, such as drilling, sampling, and coring, with precision, efficiency, and autonomy that will make it the tool of trade in the future (Cieslak *et al*., 2015; Zhang *et al*., 2015; Simetti and Casalino, 2016; Sugiyama and Toda, 2016).

To enhance the performance of the AUVMS, many control schemes have been proposed by researchers. For example, Wang *et al*. (2017) applied adaptive control methods to overcome the uncertainties of the kinematics and dynamics of the AUVMS, and Esfahani *et al*. (2015) presented a time-delay control for the AVUMS by fuzzytuning the terminal sliding mode control. Huang *et al*. (2016) analyzed the disturbance acting on the AUVMS through the kinematic and dynamic models of the system and proposed an observer-based coordinated control scheme for the AUVMS. Sagara *et al*. (2010) developed a digital-type disturbance compensation control scheme for the AUVMS system using a disturbance observer and re-

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solved-acceleration control scheme.

Most of the above studies focused on anti-disturbance control methods using an accurate model of the AUVMS and sophisticated optimization techniques (Klein and Huang, 1983; Chen *et al*., 2016). However, because of the irregularity of the AUVMS's shapes, measurement errors, hydrodynamic uncertainties, and time-varying dynamical characteristics, obtaining an accurate mathematical model is difficult. In addition, overcomplicated control schemes are impractical for real-life applications.

Active disturbance rejection control (ADRC) was initially proposed by (Han, 1998), and later many other researchers followed suit (*e.g*., Gao *et al*., 2002; Huang and Xue, 2014). The essential idea behind ADRC is the inclusion of the external disturbance and internal uncertainties as a 'generalized total disturbance' and the use of an extended state observer (ESO) to estimate and eliminate the disturbance in the nonlinear forward and feedback control loops (Fu and Tan, 2016). The ADRC design is independent of the system model and can explicitly mitigate the effects of the system's nonlinearities, unknown disturbances, and dynamic uncertainties. However, the complexity of the structure of the ADRC and the necessity for tuning numerous parameters is the Achilles heel of this scheme. To overcome this disadvantage, several researchers proposed a linear ADRC (LADRC) scheme with a simple structure and less parameters to tune (Gao, 2003; Ramírez *et al*., 2014; Morales *et al*., 2015). The LADRC scheme has been efficiently applied to many industrial systems (*e.g*., Du *et al*., 2016; Li *et al.*, 2016; Ran *et al*., 2016; Yang *et al*., 2018). Therefore, the use of the LADRC control technique to control the AUVMS is subject to multiple nonlinearities, highly coupled terms, uncertainties, and unknown disturbances is proposed here. Because the sliding mode control (SMC) scheme has better robustness properties than the PD control (Sun *et al*., 2018), SMC is combined here with ADRC to formulate the sliding-mode active disturbance rejection control (SMADRC) and address the control problem of the AUVMS.

As the AUVMS is a multi-input and multi-output system, many of the SMADRC's parameters must be tuned during implementation. To avoid the tedious process of parameter tuning, fuzzy logic control (FLC) is used here to achieve self-tuning. In 1973, Mamdani was one of the first researchers to successfully implement FLC to a laboratory steam engine plant and achieved good control results (Mamdani and Assilian, 1975). The same technique was effectively used to control a milling process, and its robust stability was analyzed through the circle criterion (Guerra *et al*., 2003). Therefore, such an FLC technique can be extended to fuzzify the LESO and SMC controllers in the LADRC scheme. Moreover, fuzzy ADRC has been studied by some researchers and successfully applied to real-life systems (*e.g*., Pan *et al*., 2010; Liu *et al*., 2011; Su *et al*., 2017). However, most of the aforementioned works mainly focused on the fuzzification of the PD controller of the LADRC, but not on the LESO itself. Motivated by the aforementioned works, the fuzzification of the SMC and LESO parameters both based on the

tracking error and its derivatives is proposed here to formulate the fuzzy sliding mode active disturbance rejection controller (FSMADRC) scheme.

In this work, a new FSMADRC technique is proposed to address the control problem of the AUVMS. The main contributions of this work are as follows. First, the proposed FSMADRC can achieve good trajectory tracking by the AUVMS where complex disturbances, modeling uncertainties, and coupling terms are lumped into a total disturbance and compensated by the SMADRC. Second, the fuzzy control scheme perform parameter online selftuning, which is very valuable for the practical implementation of the proposed control technique on real systems. Finally, it provides a framework for extending the proposed fuzzy rules to other SMADRC techniques.

This paper is organized as follows: The dynamic model of an AUVMS is presented in Section 2. The FSMADRC design method is given and the convergence and estimation errors of LESO are analyzed in Section 3. The setup of the simulation is described in Section 4, and the simulation results are illustrated in Section 5. Finally, some conclusions are drawn in Section 6.

2 Dynamic Model of an AUVMS

An AUVMS with a two-link and three-joint manipulator is considered here, and its structure is shown in Fig.1. First, I_e (*X*, *Y*, *Z*) is defined as the earth-fixed coordinate system (inertial coordinate system), I_b (x_b , y_b , z_b) is defined as the body-fixed coordinate system of the vehicle, and $I_t = [x_t, y_t, z_t]^T$ is defined as the task space (or manipulator effector) coordinate system. The standard form of the proposed 9-degree-of-freedom (DOF) AUVMS dynamic model can be written as follows (Fossen, 1994):

$$
M(q)\ddot{q} + C(q,q)\dot{q} + D(q,\dot{q})\dot{q} + G(q) + F(q,\dot{q}) = \tau_c + \tau_d,
$$
\n(1)

where $q = [q_v^T q_m^T]^T \in \mathbb{R}^9$; $q_v = [XYZ \phi \theta \psi]^T \in \mathbb{R}^6$ represents the kinematic states of the vehicle in the inertial coordinate system, corresponding to the surge, sway, heave, roll, pitch, and yaw angles, respectively; and the elements in $q_m = [\theta_1 \theta_2 \theta_3]^T \in \mathbb{R}^3$ show the joint angles of the manipulator, as shown in Fig.1.

Fig.1 Structure of an autonomous underwater vehicle with a two-link and three-joint manipulator.

$$
M(q) = \begin{bmatrix} M_v(q_v) & H^{\mathrm{T}}(q_m) \\ H(q_m) & M_m(q_m) \end{bmatrix},
$$

\n
$$
C(q, \dot{q}) = \begin{bmatrix} C_v(q_v, \dot{q}_v) & 0 \\ 0 & C_m(q_m, \dot{q}_m) \end{bmatrix},
$$

\n
$$
D(q, \dot{q}) = \begin{bmatrix} D_v(q_v, \dot{q}_v) & 0 \\ 0 & D_m(q_m, \dot{q}_m) \end{bmatrix},
$$

\n
$$
G(q) = \begin{bmatrix} G_v(q_v) \\ G_m(q_m) \end{bmatrix}, \quad F(q, \dot{q}) = \begin{bmatrix} F_v(q, \dot{q}) \\ F_m(q, \dot{q}) \end{bmatrix},
$$

\n
$$
\tau_c = \begin{bmatrix} \tau_v \\ \tau_m \end{bmatrix}, \quad \tau_d = \begin{bmatrix} \tau_{dv} \\ \tau_{dm} \end{bmatrix}.
$$

Then, $M(q)\ddot{q} \in \mathbb{R}^9$ denotes the total inertial forces and movements (including the inertia terms and added mass), and $M_{\nu}(q_{\nu})q_{\nu} \in \mathbb{R}^{6}$ and $M_{m}(q_{m})q_{m} \in \mathbb{R}^{3}$ represent the inertial forces and movements of the vehicle and manipulator, respectively. $C(q, \dot{q})\dot{q} \in \mathbb{R}^9$ is the total Coriolis and centrifugal terms (including the added mass terms), and $C(q_v, \dot{q}_v) \dot{q}_v \in \mathbb{R}^6$ and $C(q_w, \dot{q}_w) \dot{q}_w \in \mathbb{R}^3$ are the Coriolis and centrifugal forces of the vehicle and manipulator, respectively. The detailed features of the Coriolis and centrifugal terms can be found in Mulero *et al*. (2007).

The vector of damping effects, including friction and hydrodynamic forces, is represented by $D(q, \dot{q})\dot{q} \in \mathbb{R}^9$. The restoring force moments acting on the vehicle and manipulator are represented by vectors $G_v(q)$ and $G_m(q)$. The vector of the coupling dynamic effects on the vehicle from the motion of the manipulator is represented by $F_v(q, \dot{q})$. $F_m(q, \dot{q})$ denotes the interaction effects on the manipulator generated by the relative movement between the vehicle and manipulator. τ_c is the vector of input forces; $\tau_d = \tau_{ed} + \tau_{id}$ is the disturbance vector, where τ_{id} is made up of internal disturbances due to measurement errors, parameter uncertainties, and modeling inaccuracies; and τ_{ed} is the external disturbance caused by ocean current and seawater density sudden variation.

Of note, only the rotation of the manipulator around the *z*-axis in the body-fixed coordinate system is considered in this study. In addition, the vehicle and manipulator are considered rigid bodies, and the manipulator is assumed to be made up of cylindrical elements.

The AUVMS consists of a 6-DOF vehicle and 3-DOF manipulator. Therefore, the system has more DOFs than the task space, making it kinematically redundant. The transformation relationship from the task space coordinates to system states (configuration space) is given by Ismail and Dunnigan, (2011):

$$
\dot{\boldsymbol{I}}_t = \boldsymbol{J}(\boldsymbol{q})\dot{\boldsymbol{q}}\,,\tag{2}
$$

where $I_t = [x_t, y_t, z_t]^T$ is the position vector of the end effector in the task space coordinate system, and $J(q)$ is the Jacobian matrix, by which the velocities in the body-fixed coordinate system are mapped to the configuration space velocities.

To accurately perform underwater operations with the AUVMS, the complex underwater environment should be modeled and analyzed. A cylindrical element moving in deep water with density *ρ* will have hydrodynamic forces acting on it, which will be discussed in the next section.

2.1 Added Mass

When an object accelerates through a liquid, the body will also accelerate the water surrounding it, therefore, there is an added mass force in the opposite direction that affects the body motion. The effect of the added mass can be described by a matrix $I_A \in \mathbb{R}^{6 \times 6}$. Force R_A and moment *TA* acting on the body can be represented as

$$
\begin{bmatrix} \mathbf{R}_A \\ \mathbf{T}_A \end{bmatrix} = -\mathbf{I}_A \begin{bmatrix} \dot{\mathbf{v}}_r \\ \dot{\boldsymbol{\omega}}_r \end{bmatrix} - \begin{bmatrix} \tilde{\boldsymbol{\omega}}_r & 0 \\ \tilde{\mathbf{v}}_r & \tilde{\boldsymbol{\omega}}_r \end{bmatrix} \mathbf{I}_A \begin{bmatrix} \dot{\mathbf{v}}_r \\ \dot{\boldsymbol{\omega}}_r \end{bmatrix},
$$
(3)

where v_r and ω_r represent the relative angular and linear velocities with respect to the ocean current velocity in the body-fixed coordinate system and \tilde{v}_r and $\tilde{\omega}_r$ represent the operators $(\nu_r \times)$ and $(\omega_r \times)$, respectively.

2.2 Buoyancy

The buoyancy force vector is given by

$$
\begin{bmatrix} R_{GB} \\ T_{GB} \end{bmatrix} = \begin{bmatrix} f_G + f_B \\ r_G \times f_G + r_B \times f_B \end{bmatrix}, \tag{4}
$$

where \mathbf{R}_{GB} and \mathbf{T}_{GB} are the resultant forces and moment of buoyancy and gravity; f_B and f_G denote the buoyancy and gravity force vectors; and r_B and r_G represent the position vectors of the buoyancy and mass centers, respectively.

2.3 Lift and Drag Forces

The lift and drag forces refer to the forces acting on the body due to vortex shedding, pressure, and viscous forces. When a rigid body moves in the fluid, the lift forces are orthogonal to the fluid velocity and drag forces are parallel to it.

First, a first-order Gauss-Markov process is employed to model the kinematics of the ocean currents (Kim *et al*., 2014).

$$
\begin{cases}\n\dot{v}_c = -\varepsilon_v v_c + \omega_v \\
\dot{\alpha}_c = -\varepsilon_\alpha \alpha_c + \omega_\alpha , \\
\dot{\beta}_c = -\varepsilon_\beta \beta_c + \omega_\beta\n\end{cases}
$$
\n(5)

where v_c is the ocean current velocity; α_c and β_c represent the orientation of the ocean current; ε_c , ε_a , and ε_β are positive constants; and ω_v , ω_a , and ω_β are the Gaussian white noises.

The dynamic model of drag force F_D can be presented as follows (Fossen, 1994; Avila and Adamowski, 2011):

$$
F_D = D_S v_r + \frac{1}{2} \rho C_D(\mathbf{R}_e, \alpha) A(\alpha) v_r^2 \text{sign} |v_r| + o(v_r^3)
$$
, (6)

where D_S is the linear skin-friction coefficient and F_S = D_{S_v} is the linear skin-friction forces. $A(\alpha)$ is a frontal projected area of the body and α is the attack angle. $C_D(R_e)$ *α*) is the drag coefficient. The term $O(v_r^3)$ is the sum of the third and higher-order terms that are usually neglected.

In underwater environments, the lift forces acting on a rigid body are only caused by vortex shedding and can be expressed as follows (Shah *et al.*, 2017; Shah and Hong, 2018):

$$
F_L = \frac{1}{2} \rho C_L(\mathbf{R}_e, \alpha) A(\alpha) v_r^2 \text{sign} |v_r|.
$$
 (7)

Similarly, both the lift and drag coefficients, C_L and C_D , are determined by the Reynolds number R_e , angle of attack *α*, and Keulegan-Carpenter number (KC number) (Baykal *et al*., 2014). The range of values of these coefficients can be empirically determined, as shown in Table 1.

The lift and drag forces are conveniently defined along the relative velocity axis. The frictional force vector can be written as:

$$
\boldsymbol{F}_L = \boldsymbol{v}_r^{\mathrm{T}} \boldsymbol{D}_L \boldsymbol{v}_r, \quad \boldsymbol{F}_D = \boldsymbol{v}_r^{\mathrm{T}} \boldsymbol{D}_D \boldsymbol{v}_r, \tag{8}
$$

where D_L and D_D are diagonal matrices containing the lift and drag coefficients. Thus, the external hydrodynamic disturbance in this paper can be represented as

$$
\boldsymbol{\tau}_d = \boldsymbol{F}_D + \boldsymbol{F}_L + \boldsymbol{\tau}_o, \tag{9}
$$

where F_D and F_L are the drag force and lift forces due to vortex shedding, respectively, and τ ^{*o*} is other external unknown and unmeasurable disturbances, such as current loads and diffractions forces.

Parameter	Ke.			
Subcritical flow	$R_{\rm e}$ < 2×10 ⁵	1.0	[3, 0.6]	0.2
Critical flow	2×10^5 < R _e < 5 × 10 ⁵	[1.0, 0.4]	0.6	0.2
Transcritical flow	5×10^5 < R _e < 3 $\times 10^6$	04	16	0.28

Table 1 Lift and drag coefficients of a cylinder

To obtain a relatively accurate model of the AUVMS, the following properties are introduced.

Property 1. The maximum angle of the manipulator and the maximal velocity of the vehicle are considered to be finite, which means that vector q is bounded:

$$
\boldsymbol{q} \in \boldsymbol{D}_q = \left\{ \boldsymbol{q} \in \mathbb{R}^9 : ||\boldsymbol{q}|| \le ||\boldsymbol{q}||_{\text{max}} \right\}.
$$
 (10)

Property 2. The inertia matrix *M*(*q*) is symmetric and the positive definite function of q . The 2-norm of $M(q)$ also is bounded:

$$
M(q) = MT(q) > 0, \qquad (11)
$$

$$
\forall q \in D_q, \delta_1(q) \le \|M(q)\| \le \delta_2(q) ,\qquad (12)
$$

where $\delta_1(q)$ and $\delta_2(q)$ are the position vector scalar-valued functions.

Property 3. For the AUVMS system, there are some relationships between matrices $\mathbf{M}(q)$ and $\mathbf{C}(q,\dot{q})$ (Mohammadi *et al*., 2013):

$$
\begin{cases} \xi^{\mathrm{T}}[\dot{M}(q) - 2C(q, \dot{q})] \xi = 0, \forall \xi \in \mathbb{R}^{9}, \forall q \in \mathbb{R}^{9} \\ [\dot{M}(q) - 2C(q, \dot{q})]^{T} = [\dot{M}(q) - 2C(q, \dot{q})] \end{cases} (13)
$$

Property 4. There is an upper bound for the Coriolis matrix:

$$
\begin{cases}\n\forall q \in \mathcal{D}_q, \|C(q, \dot{q})\| \leq \sigma \|\dot{q}\|_{\max}^2, \\
\sigma = \sup_{q \in \mathcal{D}_q} \{C_b(q)\}\n\end{cases}
$$
\n(14)

where $C_b(q)$ is a scalar-valued function of the position vector *q*.

Property 5. In the dynamic model of the AUVMS, the damping matrix is positive definite (Santhakumar

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and Kim, 2011):

$$
\boldsymbol{D}(\boldsymbol{q},\dot{\boldsymbol{q}}) > 0, \forall \boldsymbol{q} \in \mathbb{R}^9, \forall \dot{\boldsymbol{q}} \in \mathbb{R}^9. \tag{15}
$$

Using the above properties, the upper bound of $\|M(q)\|$ can be obtained. To design the control law, the following is defined:

$$
\begin{cases}\nM(q) = \hat{M}(q) + \Delta M(q) \\
C(q, \dot{q}) = \hat{C}(q, \dot{q}) + \Delta C(q)\n\end{cases}
$$
\n(16)

where $\hat{M}(q)$ and $\hat{C}(q,\dot{q})$ are the estimates of the matrices $M(q)$ and $C(q, q)$ and $\Delta M(q)$ and $\Delta C(q)$ are their corresponding estimate errors. The characteristics of $\hat{M}(q)$ and $\hat{C}(q, \dot{q})$ are not necessarily the same as $M(q)$ and $C(q, \dot{q})$ due to the resulting modeling errors, and the effects of $\Delta M(q)$ and $\Delta C(q)$ are estimated by the LESO and then canceled by the ADRC, as shown in the next sections.

Using Eqs. (1) and (16), the dynamic model of the AUVMS can be rewritten as:

$$
\hat{M}(q)\ddot{q}+\hat{C}(q,q)\dot{q}+D(q,q)\dot{q}+G(q)+F'(q,\dot{q})=\tau_c+\tau_d,
$$
\n(17)

where

$$
F'(q,\dot{q}) = F(q,\dot{q}) + \Delta M(q)\ddot{q} + \Delta C(q,\dot{q})q . \quad (18)
$$

3 FSMADRC Design for the AUVMS

In this section, the design problem of SMADRC is studied for the AUVMS, which is separated into 9 SISO subsystems for constructing LESO. Then, the estimation error analyses of the LESO and the closed-loop stability of the SMADRC are discussed. Then, an FSMADRC method is developed to self-tune the parameters of the proposed LESO and SMC.

3.1 SMADRC Design

The traditional ADRC is made up of a tracking differentiator, nonlinear state feedback control law, and a nonlinear ESO, which can deal with unknown disturbances, coupling effects, and uncertainties. However, the nonlinear functions in the ADRC can be replaced by linear ones to obtain a simpler structure. Hence, the LADRC is adopted here, where the PD controller in the LADRC is replaced by an SMC. For simplicity and no loss of generality, the AUVMS is decomposed into nine subsystems, and an SMADRC is designed for each subsystem in a

> a_{12} a_{12} a_{19} | y_1 | c_{11} c_{12} a_{13} | y_1 | a_1 | y_1 21 u_{22} $\left[\sqrt{y_2}\right]$ $\left[\frac{v_{21}}{21}$ $\frac{c_{22}}{22}$ $\left[\sqrt{y_2}\right]$ $\left[\frac{u_{21}}{21}$ $\frac{1}{22}$ 91 a_{99} y_9 c_{91} c_{99} y_9 u_{9} u_{9} u_{9} a_{11} a_{12} \cdots a_{19} $\begin{bmatrix} \ddot{y}_1 \end{bmatrix}$ $\begin{bmatrix} c_{11} & c_{12} & \cdots & c_{19} \end{bmatrix}$ $\begin{bmatrix} \dot{y}_1 \end{bmatrix}$ $\begin{bmatrix} u_1 \end{bmatrix}$ $\begin{bmatrix} f \end{bmatrix}$ a_{21} a_{22} $|\n\vec{y}_2|$ $|c_{21}$ c_{22} $|\n\vec{y}_2|$ $|u_2|$ $|f_1|$ a_{91} a_{99} $|\n\begin{array}{c|c|c|c|c} \hline \end{array}$ c_{91} $|y_{9}$ $|y_{9}$ $|y_{1}$ $|a_{11} \quad a_{12} \quad \cdots \quad a_{19} | \quad |\ddot{y}_1| \quad |c_{11} \quad c_{12} \quad \cdots \quad c_{19} | \quad |\dot{y}_1| \quad |u_1| \quad |f_1|$ $\begin{bmatrix} a & a \\ a & a \end{bmatrix}$ $\begin{bmatrix} a & b \\ c & d \end{bmatrix}$ $\begin{bmatrix} a & b \\ c & d \end{bmatrix}$ $\begin{bmatrix} a & b \\ c & d \end{bmatrix}$ $|\mathsf{x}| \left| \begin{array}{cc} 1 \\ 1 \end{array} \right| + \left| \begin{array}{cc} 21 & 22 \\ 1 & 1 \end{array} \right|$ $\begin{bmatrix} a_{91} & a_{99} \end{bmatrix} \begin{bmatrix} y_9 \end{bmatrix} \begin{bmatrix} c_{91} & c_{99} \end{bmatrix} \begin{bmatrix} y_9 \end{bmatrix} \begin{bmatrix} u_9 \end{bmatrix} \begin{bmatrix} f_9 \end{bmatrix}$ \cdots a_{19} $\begin{bmatrix} \ddot{y}_1 \end{bmatrix}$ $\begin{bmatrix} c_{11} & c_{12} & \cdots & c_{19} \end{bmatrix}$ $\begin{bmatrix} \dot{y}_1 & \dot{y}_2 & \cdots & \dot{y}_{19} \end{bmatrix}$ \ddot{v}_2 | c_{21} c_{22} | \ddot{v} E SA MENTE A MENTENE \ddot{v}_{9} | c_{91} c_{99} | \ddot{v} $, \t(21)$

in which $\bar{\tau} = [f_1, f_2, \cdots, f_9]^T$, $\tau_c = [u_1, u_2, \cdots, u_9]^T$ and $q = [y_1, y_2, \cdots, y_9]^T$ (y_2, \dots, y_9) ^T. The two coefficient matrices of *y* and *y* in Eq. (21) are equal to $M(q)$ and $C(q, \dot{q})$, respectively.

Some traditional methods can deal with highly coupled systems (20) *via* feed-forward decoupling algorithms and diagonal matrix decoupling (*e.g*., Lin and Mon, 2005; Yang *et al*., 2012; Yang *et al*., 2013; Barbalata *et al*., 2018). However, these methods are too complex to apply to an AUVMS case with complex uncertainties and disturbances. In this study, a novel technique is developed to lump the coupling terms from the other subsystems into a total disturbance to be estimated by the LESO.

Taking the first subsystem as an example and from Eq. (21), the dynamics of the first subsystem can be written as:

$$
a_{11}\ddot{y}_1 + a_{12}\ddot{y}_2 + \cdots + a_{19}\ddot{y}_9 + c_{11}\dot{y}_1 + c_{12}\dot{y}_2 + \cdots + c_{19}\dot{y}_9 = u_1 + f_1.
$$
\n(22)

Using the total disturbance concept, Eq. (22) can be rewritten as:

$$
\ddot{y}_1(t) = b_1 u_1(t) + \xi_1, \qquad (23)
$$

where

$$
\begin{cases} \xi_1 = \frac{f_1 - (a_{12}\ddot{y}_2 + \cdots a_{19}\ddot{y}_9 + c_{11}\dot{y}_1 + \cdots c_{19}\dot{y}_2)}{a_{11}}, \\ b_1 = \frac{1}{a_{11}}, \end{cases} (24)
$$

where ξ_1 is the total disturbance of the first subsystem, including hydrodynamic forces, coupling terms, Coriolis and centrifugal terms, and other unknown disturbances, and b_1 is the input parameter.

Assumption 1. The derivative of the total disturbance term *ξ*1 exists and it is bounded (Shao and Gao, 2016).

The lumped disturbance is defined as an extended state variable $x_3 = \xi_1$ and its time derivative $\dot{x}_3 = w_1$. Set $x_1 = y_1$ and $\dot{x}_1 = \dot{y}_1$. Then, from Eq. (23), one can obtain the following state-space equation:

similar manner to the remaining eight subsystems until an overall SMADRC is obtained for the AUVMS.

The dynamic Eq. (17) of the AUVMS can be rewritten as follows:

$$
\hat{M}(q)\ddot{q} + \hat{C}(q,\dot{q})\dot{q} = \tau_c + \overline{\tau}_d , \qquad (19)
$$

where $\bar{\tau}_d$ is the total disturbance given by:

$$
\overline{\tau}_d = \tau_d - (D(q, \dot{q})\dot{q} + F'(q, \dot{q}) + G(q)). \qquad (20)
$$

Then, the AUVMS Eq. (18) can be written in an unfolded form as:

$$
\begin{array}{c}\na_{12} & \cdots & a_{19} \\
a_{22} & \ddots & \ddots \\
a_{99}\n\end{array}\n\begin{array}{c}\n\ddot{y}_1 \\
\ddot{y}_2 \\
\vdots \\
\ddot{y}_9\n\end{array}\n+\n\begin{bmatrix}\nc_{11} & c_{12} & \cdots & c_{19} \\
c_{21} & c_{22} & \cdots & c_{19} \\
\vdots & \ddots & \ddots & \ddots \\
c_{99}\n\end{bmatrix}\n\times\n\begin{bmatrix}\n\dot{y}_1 \\
\dot{y}_2 \\
\vdots \\
\dot{y}_9\n\end{bmatrix}\n=\n\begin{bmatrix}\nu_1 \\
u_2 \\
u_2 \\
\vdots \\
u_9\n\end{bmatrix}\n+\n\begin{bmatrix}\nf_1 \\
f_2 \\
\vdots \\
f_9\n\end{bmatrix},
$$
\n(21)

$$
\begin{cases}\n\dot{x}_1 = x_2 \\
\dot{x}_2 = b_1 u_1 + \xi_1 \\
\dot{x}_3 = w_1 \\
y_1 = x_1\n\end{cases}
$$
\n(25)

Using Eq. (24), one can propose a LESO as follows:

$$
\begin{cases} \n\dot{z}_1 = z_2 - l_1(z_1 - y_1) \\ \n\dot{z}_2 = z_3 - l_2(z_1 - y_1) + b_1 u_1 \,, \\ \n\dot{z}_3 = -l_3(z_1 - y_1) \n\end{cases} \tag{26}
$$

where l_1 , l_2 , and l_3 are the LESO parameters to be determined later. By choosing the parameters in Eq. (26) appropriately, the outputs of the LESO will converge to x_1, x_2 , and x_3 (Shao and Gao, 2016).

The observer gains can be given as:

$$
\begin{bmatrix} l_1 & l_2 & l_3 \end{bmatrix} = [\beta_1 \omega_{o1} \quad \beta_2 \omega_{o1}^2 \quad \beta_3 \omega_{o1}^3], \tag{27}
$$

where ω_{o1} is the observer bandwidth of the LESO and β_i , and $i=1, 2, 3$ are selected to ensure that $s^3 + l_1s^2 + l_2s + l_3$ is a Hurwitz ploynomial. For example, one can let $s^3 + l_1 s^2$ $+ l_2 s + l_3 = (s + \omega_{o1})$, and then the binomial coefficients can be determined by

$$
\beta_i = \frac{(n+1)!}{i!(n+1-i)!}, 1 \le i \le n+1, \tag{28}
$$

where *n*=2 is the order of the subsystem, *i.e*.,

$$
\beta_1 = 3, \beta_2 = 3, \beta_3 = 1 . \tag{29}
$$

Next, before designing the SMADRC scheme for the subsystem Eq. (23), we recall the following lemma.

Lemma 1 (Ioannou and Sun, 1995): For $V(t)$: [0, ∞] \in *R*, if

$$
\dot{V}(t) \le -\alpha V(t) + \gamma(t), \ \forall t \ge t_0 \ge 0 \ , \tag{30}
$$

then $V(t)$ satisfies

$$
V(t) \le e^{-\alpha(t-t_0)} V(t_0) + \int_{t_0}^t e^{-\alpha(t-\tau)} \gamma(\tau) d\tau.
$$
 (31)

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The sliding surface is defined as

$$
s = c_1 \hat{e}_1 + \dot{\hat{e}}_1 , \t\t(32)
$$

where $c_1 > 0$ is a given parameter, $\hat{e}_1 = \hat{y}_1 - y_{1d} = z_1 - y_{1d}$, \hat{y}_1 is the output of the LESO, and y_{1d} is the bounded desired output. Then, an SMC scheme is presented as follows:

$$
u = \frac{-k_1 s + \ddot{y}_{1d} - c\dot{\hat{e}}_1 - \hat{\xi}_1}{b_1},
$$
 (33)

where $\hat{\xi}_1 = z_3$ is the estimate of the total disturbance of the LESO Eq. (26).

Theorem 1: System Eq. (23) with the SMADRC scheme Eq. (33) and LESO Eq. (26) is considered. Choosing the sliding mode parameter $k_1 > 1/2$ in the SMADRC leads to a bounded closed-loop system tracking error and bounded estimation error of the LESO. Moreover, if parameter k_1 and parameter ω_{o1} in LESO Eq. (26) are large enough, then the tracking and estimation errors can converge asymptotically to zero. The convergence speed is dependent on parameter ω_{o1} in the LESO and *k* in the SMC.

Proof. From Eqs. (25) and (26), the error equation of the LESO can be obtained as:

$$
\begin{cases}\ne_1 = z_1 - y_1 \\
\dot{e}_1 = e_2 - l_1 e_1 \\
\dot{e}_2 = e_3 - l_2 e_1 \\
\dot{e}_3 = w_1 - l_3 e_1\n\end{cases} (34)
$$

Then, Eq. (34) can be rewritten as follows:

$$
\dot{e} = (A - LC)e + Dw_1, \qquad (35)
$$

where $\boldsymbol{L} = [l_1, l_2, l_3]^T$, and *A*, *C*, and *D* are given as follows:

$$
A = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{bmatrix}, C = \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix}^T, D = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}.
$$

For convenience, the state transformation of *e* can be defined as follows (Shao and Gao, 2016):

$$
\begin{cases}\ne_1 = \omega_{01}^{-2} \eta_1 \\
e_2 = \omega_{01}^{-1} \eta_2, \\
e_3 = \eta_3\n\end{cases}
$$
\n(36)

which can be rewritten as

$$
e = A\eta \tag{37}
$$

where $e = [e_1 \ e_2 \ e_3]^T$, $\eta = [\eta_1 \ \eta_2 \ \eta_3]^T$, and $A = \text{diag}\{\omega_{o1}^2, \ \omega_{o1}^3, \ \omega_{o2}^4\}$ 1}. The combination of Eqs. (35) and (37) obtained

$$
\Lambda \dot{\eta} = (A - LC) \Lambda \eta + E w_1. \tag{38}
$$

Referring to the definition of Shao and Gao (2016), the error dynamics of LESO can be represented as

$$
\frac{1}{\omega_{o1}}\dot{\eta} = A_e \eta + \frac{1}{\omega_{o1}} E w_1, \qquad (39)
$$

where

$$
A_e = \begin{bmatrix} -\beta_1 & 1 & 0 \\ -\beta_2 & 0 & 1 \\ -\beta_3 & 0 & 0 \end{bmatrix} = \begin{bmatrix} -3 & 1 & 0 \\ -3 & 0 & 1 \\ -1 & 0 & 0 \end{bmatrix}.
$$
 (40)

Because A_e is a Hurwitz polynomial, for positive matrix *Q* there always exists a positive matrix *P* satisfying the following Lyapunov equation:

$$
A_e^{\mathrm{T}} \boldsymbol{P} + \boldsymbol{P}^{\mathrm{T}} A_e + \boldsymbol{Q} = 0 \,. \tag{41}
$$

Then, the following Layapunov function is considered:

$$
V_o = \frac{1}{\omega_{o1}} \eta^{\mathrm{T}} P \eta \,. \tag{42}
$$

The time differential of V_o can be obtained as

$$
\dot{V}_o = \frac{1}{\omega_{o1}} \dot{\eta}^{\mathrm{T}} P \eta + \frac{1}{\omega_{o1}} \eta^{\mathrm{T}} P \dot{\eta}
$$
\n
$$
= (A_e \eta + \frac{1}{\omega_{o1}} D \dot{\xi}_1)^{\mathrm{T}} P \eta + \eta^{\mathrm{T}} P (A_e \eta + \frac{1}{\omega_{o1}} D \dot{\xi}_1)
$$
\n
$$
= \eta^{\mathrm{T}} A_e^{\mathrm{T}} P \dot{\eta} + \frac{1}{\omega_{o1}} (D \dot{\xi}_1)^{\mathrm{T}} P \eta + \eta^{\mathrm{T}} P A_e \eta + \frac{1}{\omega_{o1}} \eta^{\mathrm{T}} P D \dot{\xi}_1
$$
\n
$$
= \eta^{\mathrm{T}} (A_e^{\mathrm{T}} P + P A_e) \eta + \frac{2}{\omega_{o1}} \eta^{\mathrm{T}} P D \dot{\xi}_1
$$
\n
$$
\leq -\eta^{\mathrm{T}} Q \eta + \frac{2}{\omega_{o1}} \| P D \| \cdot \| \eta \| \cdot |\dot{\xi}_1|
$$
\n
$$
\leq -\lambda_{\min} (Q) \eta^2 + \frac{2 \| P D \| \cdot \| \eta \| \cdot |\dot{\xi}_1|}{\omega_{o1}} \tag{43}
$$

Because $\zeta_1 = z_1$ is bounded, if parameter ω_{o1} is sufficiently large, then this choice will lead to $\dot{V}_o \le 0$, which means that the error estimates are bounded and will converge asymptotically to zero.

Furthermore, the following Lyapunov function is considered:

$$
V_s(t) = \frac{1}{2}s^2.
$$
 (44)

Then, from Eq. (32), the differential of *s* can be written as

$$
\dot{s} = c_1 \dot{\hat{e}} + \ddot{\hat{e}} = c_1 \dot{\hat{e}} + \ddot{\hat{y}}_1 - \ddot{y}_{1d} = c_1 \dot{\hat{e}} + b_1 u + \xi_1 - \ddot{y}_{1d} + \dot{\hat{e}}_2, (45)
$$

where $\hat{e}_2 = \hat{y}_1 - \hat{y}_1$ is the error of the LESO resulting from tracking the actual output. Therefore, one can obtain the time derivative of *V* as

$$
\dot{V}_s(t) = s(c_1\dot{\hat{e}} + \ddot{y}_{1d} - \ddot{y}_{1d} - k_1\hat{s} - c_1\dot{\hat{e}} + \xi_1 - \dot{\xi}_1 + \dot{\hat{e}}_2), \quad (46)
$$

which can be rewritten as

$$
\dot{V}_s(t) = -k_1 \cdot s^2 + s(\xi_1 - \hat{\xi}_1 + \hat{e}_2) \,. \tag{47}
$$

By defining $|\xi_1 - \hat{\xi}_1 + \hat{e}_2| < \Delta_{\text{max}}$, the following can be obtained:

$$
\dot{V}_s(t) \le -k_1 \cdot s^2 + \frac{1}{2} s^2 + \frac{1}{2} \Delta_{\text{max}}^2 = -(2k_1 - 1)V_s + \frac{1}{2} \Delta_{\text{max}}^2 \cdot (48)
$$

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$$
V_s(t) \le e^{-\alpha(t-t_0)} V_s(t_0) + \int_{t_0}^t e^{-\alpha(t-\tau)} \gamma(\tau) d\tau
$$

= $e^{-(2k-1)(t-t_0)} V_s(t_0) + \frac{1}{2} \int_{t_0}^t e^{-(2k-1)(t-\tau)} d\tau$. (49)
= $e^{-(2k-1)(t-t_0)} V_s(t_0) + \frac{1}{2(2k-1)} \zeta_1^2 (1 - e^{-(2k-1)(t-t_0)})$

When $k_1 > 1/2$,

$$
\lim_{t \to \infty} V_s(t) \le \frac{1}{2(2k_1 - 1)} \Delta_{\text{max}}^2 \,. \tag{50}
$$

From Eqs. (42) and (44), the following close-loop Lyapunov function is considered:

$$
V = V_s + V_e \tag{51}
$$

Using Eqs. (43) , (48) and (50) leads to

$$
\dot{V} = \dot{V}_s + \dot{V}_e \le -(2k_1 - 1)V_s - \lambda_{\min}(\mathbf{Q})\eta^2
$$

+ $\frac{1}{2}\Delta_{\max}^2 + \frac{2||\mathbf{PD}|| \cdot ||\eta|| \cdot |\dot{\xi}_1|}{\omega_{o1}}$. (52)

From Eq. (52), the tracking error of the closed-loop system converges to a bounded neighborhood of zeros, and the convergence rate of the system is determined by sliding mode parameter k_1 and observer parameter ω_{o1} . The controllers and their stability for the other subsystems can be established and verified in a similar manner. Eq. (33) clearly shows that only two parameters need to be tuned for the first subsystem. In a similar manner, one can design an SMADRC scheme for each subsystem culminating in the development of an overall SMADRC scheme for the AUVMS with 18 parameters to be tuned, which is less than the number of parameters to be tuned when PID is used instead SMC.

3.2 FSMADRC Design

FLC has been effectively used in the control of complex nonlinear systems, which is the main reason why FLC is proposed here for parameter self-tuning. Self-tuning is achieved by using relationships between the tracking errors and parameters of the SMC and LESO. The FLC procedure is demonstrated using the first subsystem of the AUVMS to show how to develop its FSMADRC while considering that the other subsystems are treated in a similar manner. The inputs of the FLC are the error (e_1) and its derivative (e_1) , in which $e_1 = r_1 - y_1$, where r_1 is the desired output and y_1 is the actual system output. The outputs of the FLC are k_1 and ω_{o1} . In this work, Mamdani's fuzzy inference rule is adopted. There are seven fuzzy sets of e_1 and \dot{e}_1 , which consist of negative big (NB), negative medium (NM), negative small (NS), zero (Z), positive small (PS), positive medium (PM), and positive big (PB). There are also three fuzzy sets of k_1 and ω _o, consisting of small (S), medium (M), and big (B). Here,

the FLC is designed based on the principle that large errors correspond to low gains and small errors correspond to high gains. Then, the relationship between e_1 and e_1 and parameters k_1 and ω ^o is described in the fuzzy language and rules shown in Tables 2 and 3. The fuzzy membership functions $\mu(e_1)$, $\mu(e_1)$, $\mu(\omega_{o_1})$ and $\mu(k_1)$ are chosen as triangular distributions, as shown in Figs.2 and 3, respectively.

Table 2 Fuzzy rules of *k*¹

k_1, e_1, e_1	$_{\rm NB}$	NΜ	NS	Z	PS	PM	РB
NΒ	S	S	М	М	M	S	S
NM	S	S	М	В	М	S	S
NS	S	М	М	В	M	М	S
Z	S	M	B	B	в	М	S
PS	S	М	М	В	М	М	S
PM	S	S	М	B	M	S	S
PВ	S	S	М	M	М	S	S

Table 3 Fuzzy rules of *ωo*¹

Fig.3 Membership functions of ω_{o1} and k_1 .

The fuzzy sets of *e* and *ė* are chosen as [−0.3, 0.3] and [−3, 3], and the fuzzy sets of *ωo*1 and *k*1 are chosen as [400, 1200] and [1, 15], respectively. The structure of the FSMA-DRC is shown in Fig.4. Large values for parameter ω_{o1} improve the effectiveness of the LESO. However, the bandwidth of the LESO can be too large in real-life applications (Gao, 2003), so a balance must be considered between the bandwidth values and their suitability for implementation.

After the design completion of the first FSMADRC, the FSMADRCs for self-tuning the parameters of the remaining subsystems of the AVUMS can be developed in a similar manner.

Fig.4 Structure of the FSMADRC.

4 Description of the Tasks

The AUVMS is comprised of a 6-DOF vehicle and 3- DOF manipulator, whose parameters are chosen as Table 4. First, an open-loop analysis is conducted on the AUVMS to analyze the ocean current effects on the vehicle. Then, two typical tasks are carried out to test the control performance of the proposed FSMADRC.

4.1 Task I

In underwater operations, the capability of the AUVMS to keep the arm balanced while moving a grasped object is important. In Task I, the AUVMS is required to move to a specified location to pick weights, move the weights to another location, and then return to the starting point. Therefore, the AUVMS will move from position [0.012 $-4.71 - 2.885$ ^T at time $t=0$ s in the earth-fixed coordinate system to the pick-up point $[5.78 -4.69 -0.13]^T$ at time *t*= 5 s along the first desired trajectory to pick up an object with 3.25kg weight. When the AUVMS arrives at the dropoff point at $t = 10$ s following the second trajectory, the object will be dropped off there. Then, following a third desired trajectory, the vehicle reverses the course to the stop position at $[3.945 - 3.412 - 5.03]$ ^T and reaches it time $t=15$ s. The joint angles of the manipulator are kept at $[\theta_1]$ $(\theta_2 \ \theta_3)^T = [60^\circ \ 30^\circ \ 20^\circ]^T$.

The desired motion trajectories of the vehicle are chosen as a cubic time polynomial given by

$$
\begin{cases}\nX: 0.012 - 2 \times 10^{-3} t^2 + 5 \times 10^{-2} t^3 \\
Y: -4.71 - 6 \times 10^{-3} t^2 - 1.35 \times 10^{-3} t^3 \\
Z: -2.885 - 4.63 \times 10^{-2} t^2 + 1.145 \times 10^{-1} \times t^3\n\end{cases}, 0 \le t \le 5s
$$
\n
$$
\begin{cases}\nX: 5.947 - 2 \times 10^{-2} t^2 - 2.5 \times 10^{-3} t^3 \\
Y: -4.456 - 1.86 \times 10^{-3} t^2 + 1.85 \times 10^{-3} t^3 \\
Z: -1.0028 - 1.23 \times 10^{-3} t^2 - 8.5 \times 10^{-4} \times t^3\n\end{cases}, 5s < t \le 10s
$$
\n
$$
\begin{cases}\nX: 8.5125 - 2.105 \times 10^{-2} t^2 + 5.1 \times 10^{-5} \times t^3 \\
Y: -2.78 - 4.5 \times 10^{-2} t^2 - 2.8125 \times 10^{-3} t^3 \\
Z: -1.8572 + 5.862 \times 10^{-2} t^2 - 4.85 \times 10^{-3} \times t^3\n\end{cases}, 10s < t \le 15s
$$

To reduce the power consumption of the vehicle when moving in deep waters, the frequent switching of the rudder angle should be avoided. Therefore, the amplitude of the variation of the path angles (roll, yaw, and pitch angles) should also be included in the closed-loop system.

4.2 Task II

The inertia coupling forces always act on the vehicle to force it to vibrate when the manipulator swings at a high frequency. The accuracy of the hovering operations of the AUVMS is important to study; therefore, in Task II, the AUVMS is required to hover at a fixed position. Task II is designed to test the capability of the FSMADRC in dealing with rapidly changing external forces and high coupling effects. In this task, only the angles of the manipulator joints are allowed to vary, and the vehicle is kept in still mode. The working position of the AUVMS is set as $[0\ 0\ 0]^T$ in the earth-fixed coordinate frame.

In addition, during the moving process, the roll, yaw, and pitch angles of the vehicle remain at zero, whereas the joints' angle trajectories of the manipulator are given by

$$
\begin{cases}\n\theta_1 = (-2.21 - 28.521t + 2.862t^2)\sin(3.024 - 0.2132t - 0.212t^2 - 0.035t^3) \\
\theta_2 = (3.015 - 10.2135t - 1.021t^2)\sin(0.6631 + 0.645t + 0.36t^2 - 0.0664t^3) \\
\theta_3 = (3.3232 - 10.046t - 1.0454t^2)\sin(0.2155 + 0.06446t - 0.1245t^2 - 0.00213t^3)\n\end{cases}
$$
\n(54)

5 Simulations and Discussion

In this section, the control performance of the FSMA-DRC will be studied, and compared with PID controllers and classic FLC (CFLC), which are chosen respectively as

$$
\begin{cases}\n\tau_{PID} = K_p + K_i \int e \, dt + K_d e \\
\tau_{\text{CFLC}} = K_c U_{\text{CFLC}}\n\end{cases}
$$
\n(55)

where K_p , K_i , and K_d are the parameters of the PID controller, K_c is the output scaling gain of the CFLC controller, and U_{CFLC} is its control output.

In accordance with the membership functions in Tasks I and II, the fuzzy sets for both tasks are chosen as [−0.3, 0.3] for the error, [−3, 3] for the derivative of the error, [1, 15] for the bandwidth of the controller, and [400, 1200] for the bandwidth of the observer. Here, all SMC parameters c_i , $i=1, \dots, 9$ of the subsystems are set to be 10.

5.1 Results of Task I

The simulation results of Task I are shown in Figs.5– 11. The 3D view of the moving trajectories of the center of gravity of the vehicle controlled by different controllers are depicted in Fig.5. The tracking errors are shown in Fig.6, which also shows that the FSMADRC can track the desired trajectories more accurately than the CFLC and PID, especially during picking up and dropping off the weights.

Fig.5 Trajectory of the AUV in Task I in a 3D view: (green) desired trajectory, (dashed-black) using FSADRC, (blue) using CFLC, (red) using PID control.

The LESO can estimate the disturbance resulting from the additional load weight picked up by the AUVMS and cancel its effect in the closed-loop system. Thus, the proposed method can ensure that the FSMADRC spends less

Fig.6 Trajectories of the tracking error of the AUV in Task I: (black) using FSMADRC, (blue) using CFLC, and (red) using PID control.

Fig.7 shows that the proposed FSMADRC can keep the manipulator stable when picking up and dropping off the load weight and spends less time restoring the original angles. However, the other two controllers were not able to achieve this objective. Next, as shown in Fig.8, the roll, yaw, and pitch angle trajectories resulting from using the FSMADRC switch much more smoothly than those resulting from the PID and CFLC schemes. Moreover, the FSMADRC achieves better control performance with less energy consumed than the CFLC and PID, as shown in Fig.9. The disturbance estimation performance of the LESOs is shown in Fig.10, in which $e_1 - e_9$ correspond to the total disturbance estimation errors of the LESO in each of the nine subsystems. Clearly, the estimation errors converge to a small neighborhood of zero, which ensures an effective rejection of the total disturbance.

5.2 Results of Task II

The simulation results of Task II are presented in Figs.11–

Fig.7 Trajectories of joint angles of the AUVMS in Task I: (black) using FSMADRC, (blue) using CFLC, and (red) using PID control.

Fig.9 Comparison of the power consumption in Task I.

14. Figs.11 and 12 show that the proposed method can track the swaying trajectories and keep the vehicle stable in a fixed position relatively better than the PID and CFL control schemes. In addition, the yaw angle of the AUV-MS under the FSMADRC exhibits better stability performance than that under the other two controllers. Fig.13 shows that the FSMADRC demonstrates better performance in trajectory tracking by the end effector than that under the PID or CFL control scheme. Moreover, the FS-MADRC consumes less energy than the other two controllers while performing Task II, as shown in Fig.14.

6 Conclusions

Underwater robotics has attracted a lot of attention from the research community for applications previously deemed impossible to be performed by humans. Applications, such as surveys, security, explorations, subsea en-

Fig.8 Path angle of the AUV in Task I: (black) using FS-MADRC, (blue) using CFLC, and (red) using PID control.

Fig.10 Total disturbance estimation error of each LESO in Task I.

Fig.11 Trajectory tracking errors of the joint angles of the AUVMS in Task II: (black) using FSMADRC, (blue) using CFLC, and (red) using PID control.

Fig.13 Trajectory tracking errors of the end effector of the AUVMS in Task II: (black) using FSMADRC, (blue) using CFLC, and (red) using PID control.

Fig.14 Power consumption comparison for Task II.

Fig.12 Position response and yaw angle of the AUV in Task II: (black) using FSMADRC, (blue) using CFLC, and (red) using PID control.

gineering, and oil/gas explorations, require spot discoverries and manipulation. Oftentimes, the aforementioned tasks/ missions are executed using ROVs or divers, which is costly and often puts human life at risk. Underwater vehicles equipped with manipulator(s) have proven to provide a cost-effective solution to deep-sea missions and put human operators out of harm's way. An underwater vehicle equipped with a two-link and three-joint manipulator is studied here for performing routine tasks under uncertain and harsh environmental conditions. The FSMADRC method is used in the closed-loop system to address the tracking problem of the AUVMS, and the results are compared with those of the commonly used PID and CFL control schemes. The simulation results show that FSMA-DRC achieves better control performance than the PID and CFLC schemes. The proposed FSMADRC is independent of the AUVMS model and only needs limited knowledge of the system's model. Finally, the proposed scheme achieves better control performance than the other two reference control methods with less power consumption. In addition, the FSMADRC can determine (selftune) the appropriate parameters adaptively, which makes it very suitable for real-life applications and can be easily extended to other vehicle systems.

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