



# PS–PVD Alumina Overlayer on Thermal Barrier Coatings Against CMAS Attack

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**Abstract** Glassy deposits mainly comprising of calcium–magnesium–alumina–silicate (CMAS) accelerate the spallation of thermal barrier coatings (TBCs). In this work, an Al<sub>2</sub>O<sub>3</sub> layer was produced on yttria-stabilized zirconia (YSZ) coating by plasma spray–physical vapor deposition (PS–PVD). The effects of processing parameters during PS–PVD process on the microstructures of the deposited Al<sub>2</sub>O<sub>3</sub> coatings were investigated. A homogeneous Al<sub>2</sub>O<sub>3</sub> coating with porosity less than 1% was deposited at the spray distance of 1400 mm, which is much denser than the coating produced by atmospheric plasma spray (APS) and other PS–PVD coatings sprayed at 1000 and 1900 mm. The CMAS corrosion test at 1250 °C showed that the APS Al<sub>2</sub>O<sub>3</sub> layer was dissolved in CMAS after 24 h thermal exposure, whereas a dense reaction layer mainly composed

of anorthite formed on the top region of the PS–PVD Al<sub>2</sub>O<sub>3</sub> layer effectively protected the underlying Al<sub>2</sub>O<sub>3</sub> layer from CMAS infiltration. Thermal cycling behavior of the PS–PVD Al<sub>2</sub>O<sub>3</sub>/YSZ TBC and the APS Al<sub>2</sub>O<sub>3</sub>/YSZ TBC at 1050 °C was compared. Bulk spallation of the APS TBC occurred within 300 h thermal cycling, while little spallation was observed on the PS–PVD TBC even after 500 h thermal cycling, indicating superior durability of the PS–PVD coating.

**Keywords** alumina · atmospheric plasma spray (APS) · calcium–magnesium–alumina–silicate (CMAS) · plasma spray–physical vapor deposition (PS–PVD) · thermal barrier coatings (TBCs) · thermal cycling

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## Introduction

Thermal barrier coatings (TBCs) applied on the hot components in gas turbines provide thermal and corrosion prevention for the underlying superalloys (Ref 1–3). The ceramic topcoat in the TBC system is made of 7–8wt % yttria-stabilized zirconia (YSZ), which is usually prepared by atmospheric plasma spray (APS) or electron beam–physical vapor deposition (EB–PVD) (Ref 4–7). With the increasing service temperature of advanced gas turbines and aero-engines, the failure of YSZ TBCs caused by environmental silicate deposits mainly composed of calcium–magnesium–aluminum–silicate (CMAS) has become an urgent issue to be solved. Molten CMAS deposits penetrate the TBCs through the interconnected microcracks and porosity in the APS coating or the inter-columnar gaps in the EB–PVD coating. On the one hand, thermal–chemical interaction between CMAS and YSZ coating results in the dissolution of yttrium from YSZ, which in turn causes

tetragonal–monoclinic phase transformation of  $ZrO_2$  accompanied with volume expansion. On the other hand, CMAS penetration accelerates sintering and stiffening of YSZ coating, leading to a reduction in porosity and loss of strain tolerance (Ref 8–10).

So far, many approaches have been attempted to mitigate CMAS penetration at high temperature for YSZ TBCs. One of the approaches is physical insulation by depositing a chemically inert film made of noble metals such as Pd or Pt which doesn't react with molten CMAS to arrest CMAS infiltration (Ref 11–14). However, its application was limited by the expensive cost, insufficient binding with the underlying TBC as well as restricted operating temperature. Another approach is chemical protection through preparing a sacrificial layer that could positively interact with CMAS glass to generate a dense and stable reaction layer for CMAS prevention (Ref 15). Or some Ti and Al were doped in the YSZ coating, which (Ref 16, 17) promotes the crystallization of CMAS glass and the formation of reaction layer, thus effectively against further CMAS attack before infiltration.

Alumina is recognized as a promising candidate for resisting CMAS attack because of its ability to shift the composition of molten CMAS to a field where glasses are easy to crystallize (Ref 18–20). Besides,  $Al_2O_3$  modified YSZ TBCs are demonstrated to have enhanced mechanical properties (Ref 21). Accordingly, the  $Al_2O_3$  coatings have been intensively used to prevent CMAS attack. Zhang et al. (Ref 19) have obtained an in-situ  $Al_2O_3$  layer by magnetron sputtering Al film on YSZ and vacuum heat treatment. Mohan et al. (Ref 20) have prepared an  $Al_2O_3$  film by electrophoretic deposition followed by sintering at 1200 °C. Nevertheless, sintering treatment is detrimental to the mechanical properties of the coating. Furthermore, the column gaps in the  $Al_2O_3$  coatings deposited by PVD provide an open pathway for the infiltration of glassy CMAS deposits (Ref 21–23). The  $Al_2O_3$  coatings prepared by APS possess internal defects, interconnected microcracks, and high porosity, which are also not effective in blocking CMAS penetration. Besides, premature spallation of the APS  $Al_2O_3$  overlayer on the YSZ TBC system occurs due to the poor thermal cycling durability (Ref 24, 25).

A dense protective layer is essential in order to prevent molten CMAS infiltration. In the present work,  $Al_2O_3$  coatings were produced by plasma spray–physical vapor deposition (PS–PVD). For this application, at an intermediate electric net power of approximately 40 kW, the vaporization degree of the feedstock can be limited, and mainly liquid phase deposition is obtained (Ref 26, 27). In this way, a thin and dense protective  $Al_2O_3$  overlayer for

the YSZ TBCs against CMAS attack can be produced. The influences of processing parameters on the microstructures of the sprayed  $Al_2O_3$  coatings were investigated. The resistance to CMAS corrosion and thermal cycling behavior of the PS–PVD  $Al_2O_3$  overlayer/YSZ TBC system and APS  $Al_2O_3$  overlayer/YSZ TBC system were studied comparatively and the associated mechanisms were discussed.

## Materials and Methods

### Preparation of TBCs

Nickel-based superalloy K403 was chosen as the substrate materials, which were cut into disk shape samples of 20 mm in diameter and 3 mm in thickness. The feedstock materials were commercial 8YSZ powder (METCO 204 NS) and Ni-21Co-17Cr-12Al-1Y (Ni 192-8) powder for the YSZ topcoat and metallic bond coat, respectively. The YSZ topcoat approx. 200  $\mu\text{m}$  thick and the NiCoCrAlY bond coat approx. 150  $\mu\text{m}$  thick were both prepared using an APS unit (Sulzer Metco) with a 9M gun.

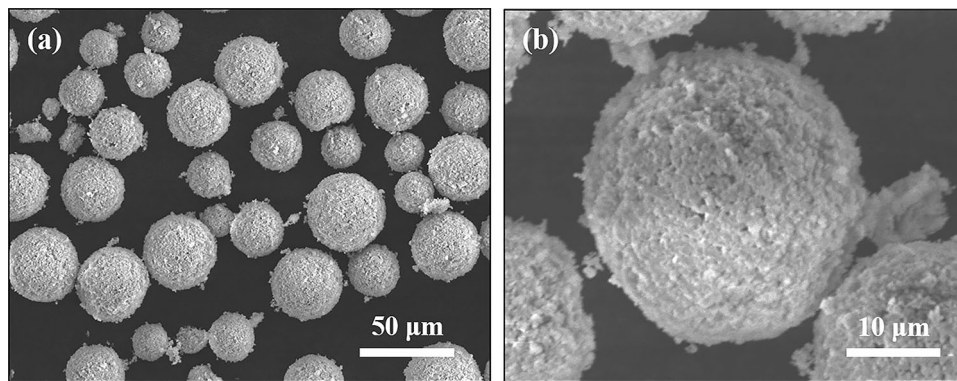
The powder feedstock for the  $Al_2O_3$  overlayer was prepared by spray drying, which was micrometer agglomerated, as shown in Fig. 1, with a size distribution:  $d_{10} = 20 \mu\text{m}$ ,  $d_{50} = 27 \mu\text{m}$  and  $d_{90} = 37 \mu\text{m}$ . The  $Al_2O_3$  overlayer was deposited by PS–PVD (Medicoat AG, Switzerland) with an MC-100 plasma torch, which allowed a maximum current of 2500 A and a peak power of 150 kW. The spray parameters for  $Al_2O_3$  coatings are shown in Table 1, in which three spray distances of 1000 mm, 1400 mm, and 1900 mm were used for obtaining dense coating. For comparison, an  $Al_2O_3$  overlayer with similar thickness was also prepared by APS with the processing parameters as shown in Table 2.

### CMAS Preparation and Corrosion Test

CMAS powders with a chemical composition of 31CaO–8MgO–12 $Al_2O_3$ –49 $SiO_2$  were prepared for the following corrosion test to simulate the environmental deposits in an aero-engine service environment. The preparation of the CMAS powders was described elsewhere (Ref 28).

Before the corrosion test, the ball-milled CMAS powder was dissolved in ethanol and deposited homogeneously onto the surfaces of the TBC specimens, then dried and weighed to maintain a deposit concentration of 15  $\text{mg}/\text{cm}^2$ . The TBC specimens with CMAS deposits were calcined in an air furnace at 1250 °C for 24 h to exam the corrosion resistance of the  $Al_2O_3$  overlayers.

**Fig. 1.** Surface morphologies of feedstock powder for spraying  $\text{Al}_2\text{O}_3$  coatings: (a) lower magnification and (b) higher magnification.



**Table 1** PS–PVD processing parameters for spraying  $\text{Al}_2\text{O}_3$  coatings

Plasma gas	Ar 30 slpm/He 60 slpm
Net power	40 kW
Powder feed rate	5 g/min
Carrier gas	Ar 3 slpm
Pressure	10 mbar
Spray distance	1000, 1400, 1900 mm
Substrate temperature	900–1000 °C

*Slpm* standard liters per minute

**Table 2** APS processing parameters for spraying  $\text{Al}_2\text{O}_3$  coating

Parameters	$\text{Al}_2\text{O}_3$ coating
Power	63 kW
Plasma gas	Ar 86 slpm/ $\text{H}_2$ 60 slpm
Spray distance	110 mm
Powder feeding	13–18 g/min

### Thermal Cycling Tests

The thermal cycling test was carried out in an air tube furnace with a mechanical automation system. During the testing, the TBC-coated specimens were first heated in the furnace at 1050 °C for 50 min, then followed by 10 min compressive air cooling from both sides of the specimen. The surface condition of the coated specimens was examined after every 10 cycles.

### Characterization

The surface and cross-sectional morphologies of as-sprayed TBCs and CMAS-interacted TBCs were characterized by a Field Emission-Scanning Electron Microscope (FE-SEM, Apollo 300) equipped with the energy dispersive X-ray spectrometry (EDS) and backscattered electron

(BSE) detector. The phase constituents of the coatings before and after interaction with CMAS were determined by X-ray diffraction (XRD, Rigaku D/max2200PC) using  $\text{Cu-K}\alpha$  radiation with a step size of 6°/min and scanning angle from 20° to 80°.

## Results and Discussion

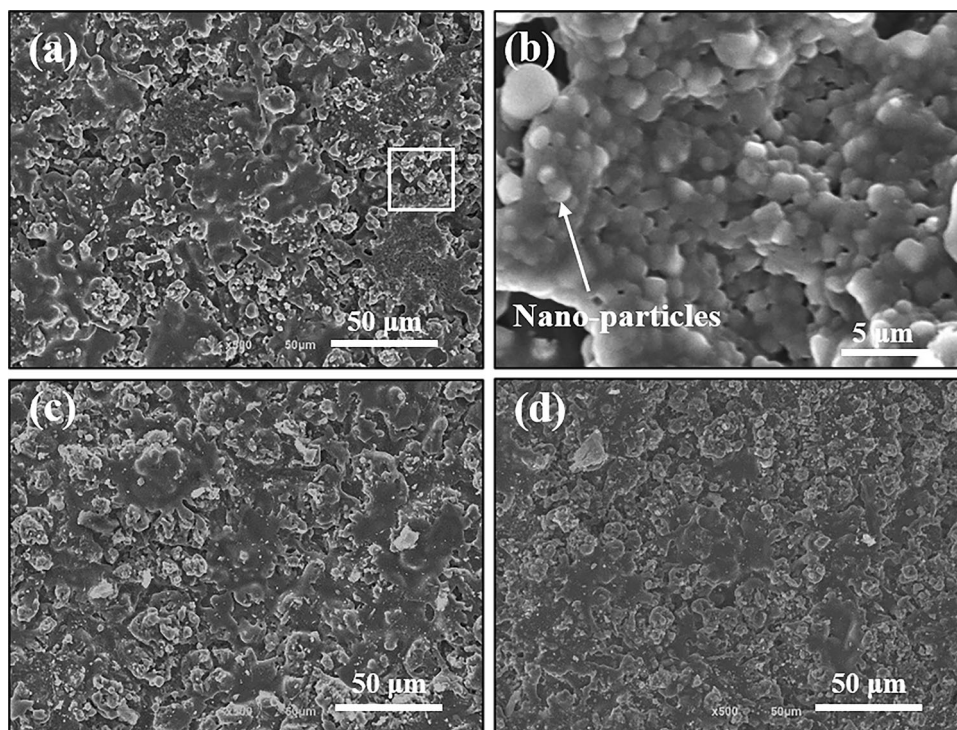
### Microstructures of the As-Sprayed $\text{Al}_2\text{O}_3$ Coatings

Figure 2(a), (c), and (d) shows the surface morphologies of the as-sprayed coatings deposited at 1000, 1500, and 1900 mm, respectively. All the sprayed coatings mainly consist of splats, resulting from impingement, spreading, and subsequent solidification of liquid droplets. Besides, there are some sintered nano-sized particles as seen in the high magnification SEM micrograph (Fig. 2b), which are probably from un-melted or partially melted primary particles in the feedstock powder. Note that the coating sprayed at 1000 mm exhibits more nanoparticles as compared to the coatings sprayed at 1400 and 1900 mm.

The porosity measurement was taken on polished cross sections of the  $\text{Al}_2\text{O}_3$  coatings based on image analyses, and the corresponding results are given in Table 3. The maximum porosity is only 2.6% which belongs to the coating sprayed at 1000 mm. The lowest porosity of ~ 0.8% was achieved in the coating sprayed at 1400 mm. From these results, the  $\text{Al}_2\text{O}_3$  overlayers by PS–PVD are quite dense, which means that the coatings were deposited mainly from well-melted liquid droplets.

Figure 3(a), (b), and (c) shows the fracture morphologies of PS–PVD  $\text{Al}_2\text{O}_3$  coatings deposited at 1000, 1400, and 1900 mm, respectively. The coatings reveal a different thickness although the coatings were sprayed within the same duration. In particular, the coating sprayed at 1000 mm is nearly twice as thick as the other two coatings sprayed at longer distances and falls off the substrate after coating process as shown in Fig. 3(a).

**Fig. 2.** Surface morphologies of PS–PVD Al<sub>2</sub>O<sub>3</sub> coatings deposited at different distances: (a) 1000 mm, (b) higher magnification of the area selected in (a, c) 1400 mm, and (d) 1900 mm



**Table 3** Porosities of PS–PVD Al<sub>2</sub>O<sub>3</sub> coatings

Spray distance (mm)	1000	1400	1900
Porosity (%)	2.61 ± 0.05	0.82 ± 0.02	1.53 ± 0.03

Compared to typical PS–PVD conditions for columnar coatings deposition, the plasma jet under the conditions in this paper is more concentrated due to higher chamber pressure 1000 Pa. As spraying distance increases, the heating effects of the plasma jet on powder feedstock become weaker, and the velocity of particles decreases. Based on this analysis, one possible reason for thinner coatings at long spraying distances could be some large particles were out of the deposition area due to reduced velocity and gravity. Accordingly, the porosity of the coating sprayed at 1400 mm is smaller than that of the coating sprayed at 1000 mm due to less un-melted particle deposition. However, with further increasing the spray distance to 1900 mm, solidification of liquid droplets in the plasma jet was likely occurred because of insufficient heat transfer from the plasma jet. This could result in a slightly increased porosity due to the deposition of solidified particles. Furthermore, a low substrate temperature at long spray distance due to less heat transfer from the plasma jet is not conducive to droplet deposition, which may be also responsible for the low deposition rates at long distances.

As the porosity is an important criterion for evaluating the capability of a coating resistant to CMAS corrosion and infiltration, the coating sprayed at 1400 mm was considered as the optimized microstructure for CMAS protection in this paper.

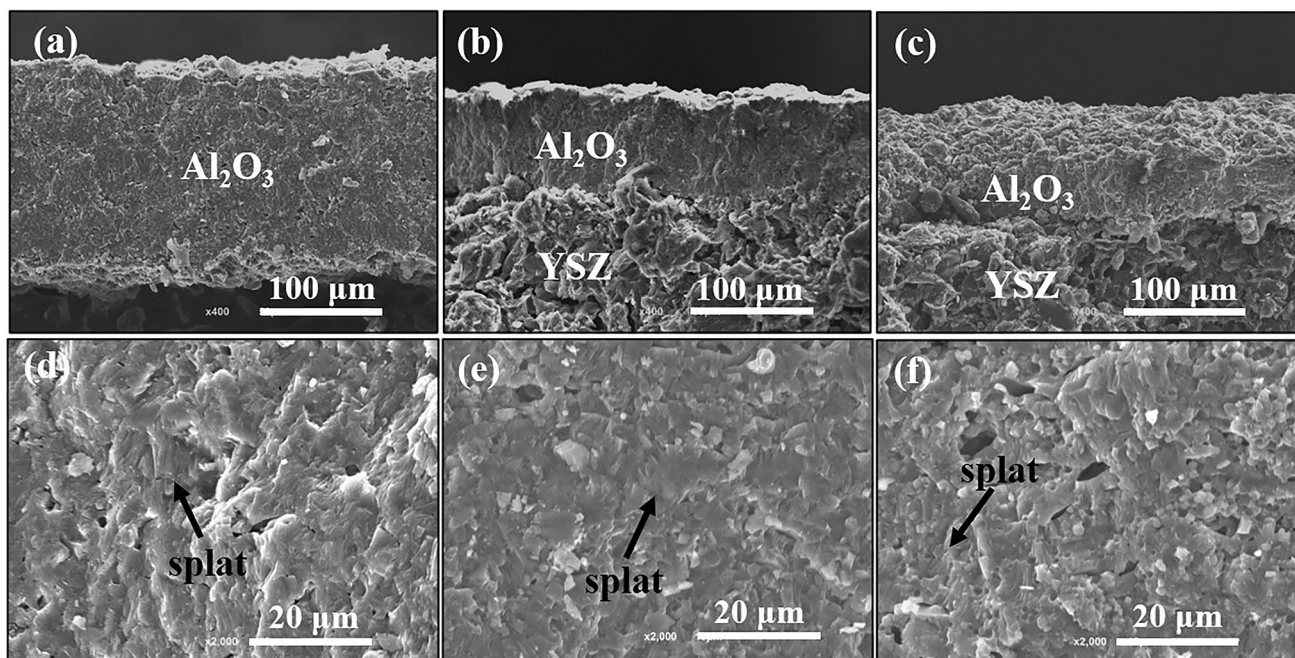
Figure 4(a) shows the cross-sectional micrographs of the PS–PVD Al<sub>2</sub>O<sub>3</sub> overlayer deposited at the spray distance of 1400 mm and a reference APS Al<sub>2</sub>O<sub>3</sub> overlayer. By optimized processing parameters for APS (listed in Table 2), the porosity (6.5%) of APS Al<sub>2</sub>O<sub>3</sub> coating is still higher than that of the PS–PVD Al<sub>2</sub>O<sub>3</sub> (only 0.82%).

### CMAS Corrosion Tests

CMAS infiltration tests were performed on both Al<sub>2</sub>O<sub>3</sub> coating and 8YSZ coating in order to compare the corrosion resistance of the two TBCs. After 24 h testing, many cracks and spallation were visible on the surface of YSZ coatings, as seen in Fig. 5(a), which led to surface wrinkle and coating spallation. The cross-sectional micrograph of YSZ TBC is shown in Fig. 5(b). The structure of the upper zone of YSZ coating is quite loose with obvious small globular particles which indicates dissolution and corrosion of YSZ by the molten CMAS. The EDS line scanning (Fig. 5c and d) showing the concentration of Si and Ca along the arrow direction in Fig. 5(b) demonstrates that the minimum infiltration depth of molten CMAS for YSZ coating was at least 50 μm.

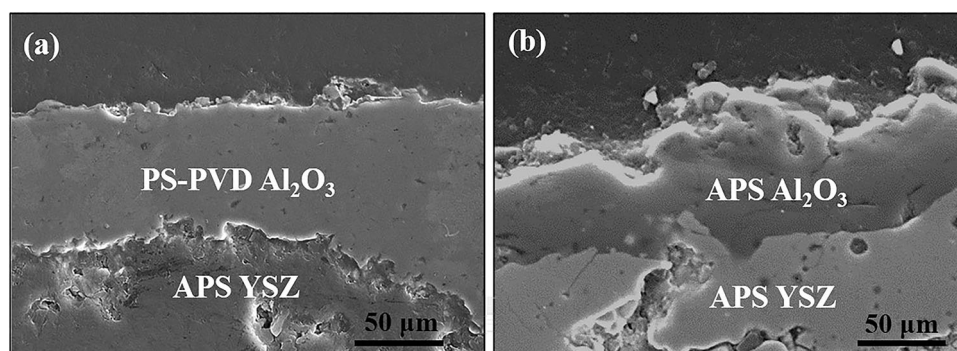
Similarly, the bilayer  $\text{Al}_2\text{O}_3/\text{YSZ}$  TBCs were calcined with CMAS glass under same isothermal treatment conditions to compare the infiltration resistance of APS  $\text{Al}_2\text{O}_3$  and PS–PVD  $\text{Al}_2\text{O}_3$  coatings. The cross-sectional morphologies of the bilayer  $\text{Al}_2\text{O}_3/\text{YSZ}$  coatings after interaction with molten CMAS are shown in Fig. 6. As for APS  $\text{Al}_2\text{O}_3/\text{YSZ}$  coatings (Fig. 6a), the CMAS infiltration depth evidenced by the Si detection was about 50  $\mu\text{m}$  (Fig. 6e). What's more, those loose and small particles at the  $\text{Al}_2\text{O}_3/\text{YSZ}$  interface indicate the underlying YSZ also slightly reacted with CMAS. From the cross-sectional morphology in Fig. 6(b), CMAS has infiltrated through the pores into the interior coating and reacted with YSZ at the interface. This means that APS  $\text{Al}_2\text{O}_3$  coating offers weak resistance to CMAS corrosion.

Figure 6(c) shows the cross-sectional SEM image of PS–PVD  $\text{Al}_2\text{O}_3/\text{YSZ}$  coatings after interaction with CMAS at 1250  $^\circ\text{C}$  for 24 h, where the respective elemental distributions of Ca and Si are given in Fig. 6(f) and (g). Note that Ca and Si content decreased dramatically at the detection depth of about 10  $\mu\text{m}$ , indicating that the infiltration of CMAS glass was successfully resisted by PS–PVD  $\text{Al}_2\text{O}_3$  overlay. Figure 6(d) shows the dense reaction layer on top of PS–PVD  $\text{Al}_2\text{O}_3/\text{YSZ}$  coatings and corresponding Ca, Mg, Al, and Si elemental content at points 1 to 6 is summarized in Table 4. The presence of Ca, Mg, Al, and Si elements is detected at point 1 to point 4, particularly, the content of Ca and Si reduces gradually through the coating thickness, which indicates that a dense 4  $\mu\text{m}$  thick reaction layer generated during the period of CMAS corrosion. At point 6, a depth of around 10  $\mu\text{m}$ , the  $\text{Al}_2\text{O}_3$  layer is intact

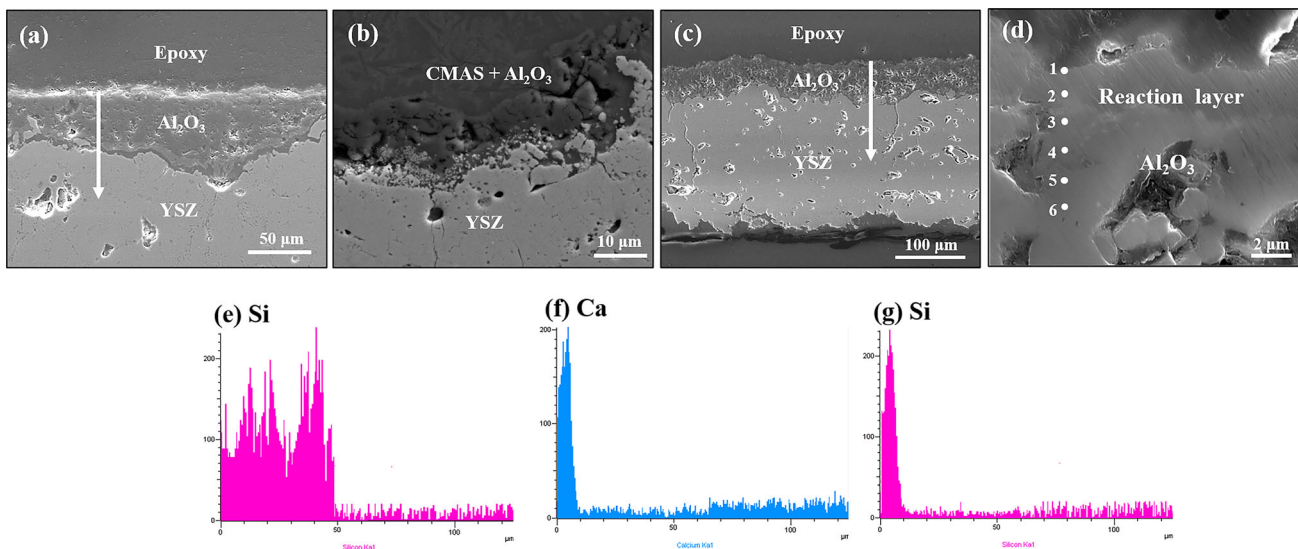
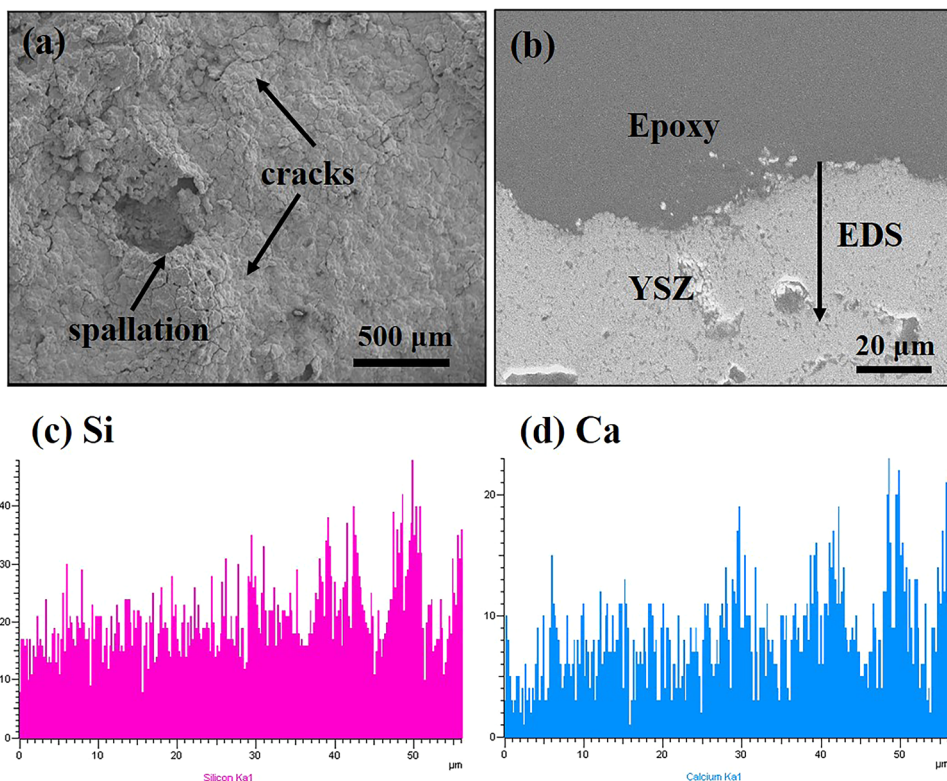


**Fig. 3.** Fracture morphologies of cross sections of PS–PVD  $\text{Al}_2\text{O}_3$  coatings deposited at different distances: (a) 1000 mm, (b) 1400 mm, (c) 1900 mm, and (d, e, f are magnifications of (a, b, and c), respectively

**Fig. 4.** SEM micrographs of cross sections of  $\text{Al}_2\text{O}_3$  coatings prepared by PS–PVD at 1400 mm (a) and by APS (b)



**Fig. 5.** SEM morphologies of the surface (a) and cross section (b) of the YSZ coating after CMAS corrosion at 1250 °C for 24 h, and element distribution of Ca (c) and Si (d) across the coating thickness toward the arrow direction in (b)



**Fig. 6.** Cross-sectional micrographs of coatings after CMAS corrosion at 1250 °C for 24 h. (a) APS Al<sub>2</sub>O<sub>3</sub>/YSZ coating, (b) magnification at the interface of APS Al<sub>2</sub>O<sub>3</sub>/YSZ, (c) PS-PVD Al<sub>2</sub>O<sub>3</sub>/YSZ

coating, (d) magnification of the top area in (c), (e) Si distribution along the arrow direction in (a), and (f) Ca and (g) Si distribution along the arrow direction in (c)

and CMAS components can hardly be found. In addition, this result also indicates that in order to obtain a better protection the thickness of Al<sub>2</sub>O<sub>3</sub> coating must be at least 10 μm.

Figure 7(a) and (b) shows the surface morphologies of the PS-PVD alumina overlayer with bare holes and cracks. The chemical composition of the region marked by the red

rectangle in Fig. 7(b) is identified by EDS and listed in Table 5. The composition with atom ratio of Ca (12.35 at.%), Al (22.38 at.%), and Si (24.20 at.%) close to 1:2:2 is deviated from the original CMAS composition. To identify the reaction products, XRD analyses are performed on as-sprayed Al<sub>2</sub>O<sub>3</sub> coating and the one after CMAS corrosion (Fig. 8). According to the phase structural

analyses, the original PS–PVD  $\text{Al}_2\text{O}_3/\text{YSZ}$  coating is composed of  $\alpha\text{-Al}_2\text{O}_3$  and  $\text{ZrO}_2$ . After 24 h interaction with CMAS, the peaks of anorthite and spinel could be detected. Combined with the EDS results in Table 5, the composition of anorthite could be  $\text{CaAl}_2\text{Si}_2\text{O}_8$ . Anorthite and spinel were demonstrated to start to melt at temperature above  $1550\text{ }^\circ\text{C}$ , which is much higher than CMAS melting point and operating temperature of TBCs ( $1250\text{ }^\circ\text{C}$  in this paper) (Ref 16, 29, 30). Thus, these high melting point reaction products can inhibit the infiltration of molten CMAS on one hand. On the other hand, compared with APS  $\text{Al}_2\text{O}_3$  coating, the PS–PVD  $\text{Al}_2\text{O}_3$  coating possesses higher density which did not provide infiltration paths to molten CMAS before formation of stable high melting point production. These should be reasons for the better CMAS resistance of the PS–PVD  $\text{Al}_2\text{O}_3$  overlayer at  $1250\text{ }^\circ\text{C}$ .

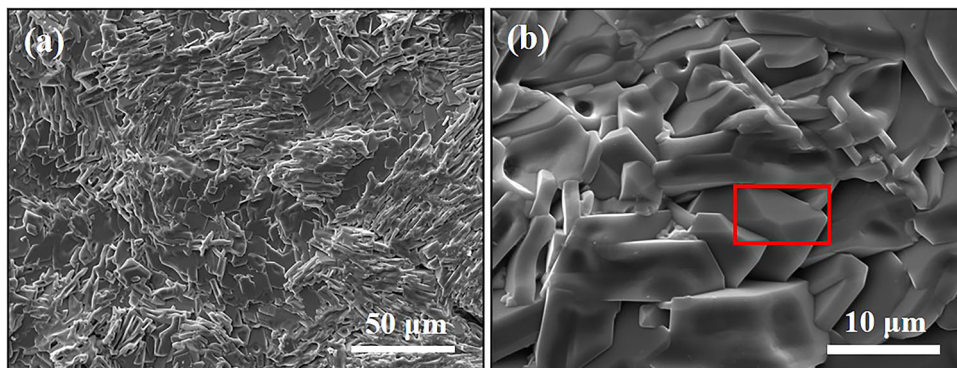
### Thermal Cycling Behavior of PS–PVD $\text{Al}_2\text{O}_3/\text{YSZ}$ Coating System

Considering the potential spallation of an in-service environment induced by mismatch coefficients of thermal expansion (CTE) between  $\text{Al}_2\text{O}_3$  and YSZ, thermal cycling of bilayer PS–PVD  $\text{Al}_2\text{O}_3/\text{YSZ}$  TBCs was studied. Figure 9 shows the surface photographs of the as-sprayed  $\text{Al}_2\text{O}_3/\text{YSZ}$  coating cycled 100 and 500 h at  $1050\text{ }^\circ\text{C}$ ,

**Table 4.** Chemical compositions of dots 1–6 in Fig. 6d

Element (at. %)	Ca	Al	Si	Mg	O
1	12.35	22.28	24.20	0.36	Bal.
2	8.73	27.72	19.63	0.69	Bal.
3	2.54	46.77	6.18	0.47	Bal.
4	1.24	64.56	2.10	0.67	Bal.
5	...	53.03	1.84	...	Bal.
6	...	42.83	...	...	Bal.

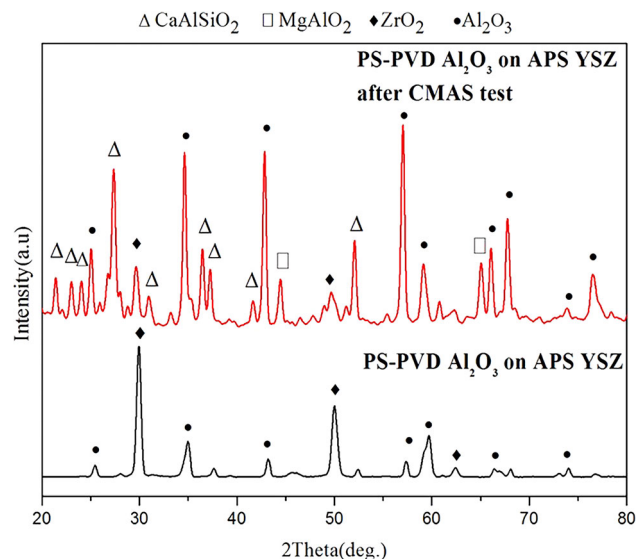
**Fig. 7.** Surface morphologies of PS–PVD  $\text{Al}_2\text{O}_3/\text{YSZ}$  coatings after CMAS corrosion at  $1250\text{ }^\circ\text{C}$  for 24 h: (a) at lower magnification and (b) at higher magnification



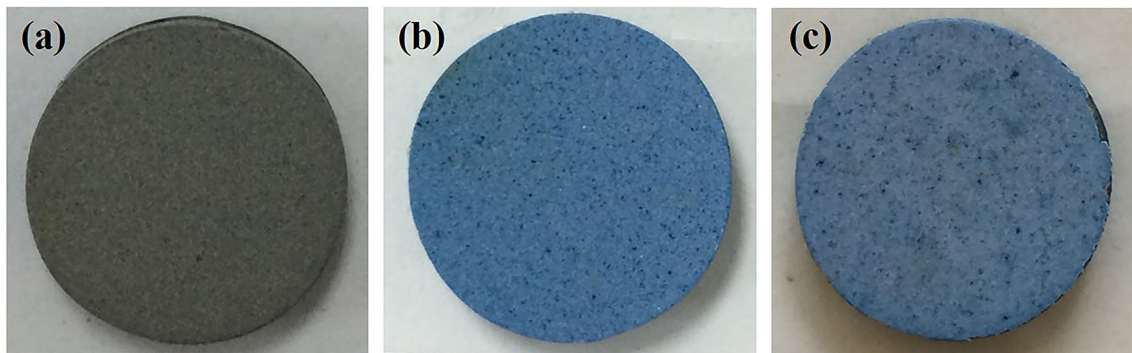
respectively. It can be seen that the coating surface kept almost intact except for a little spallation at the sample edge even though the surface changed to blue color due to oxidation, which implies an excellent lifetime of more than 500 h for PS–PVD  $\text{Al}_2\text{O}_3/\text{YSZ}$  TBCs. The surface morphologies of the as-sprayed  $\text{Al}_2\text{O}_3/\text{YSZ}$  and after 100, 500 h thermal cycling tests are shown in Fig. 10. The white small spots in Fig. 10(b) and (c) implied that the YSZ coating was already exposed attributed to the spallation of the  $\text{Al}_2\text{O}_3$  overlayer. But it is good that the spallation area

**Table 5.** Chemical compositions of the region enclosed by the red rectangle in Fig. 7(b)

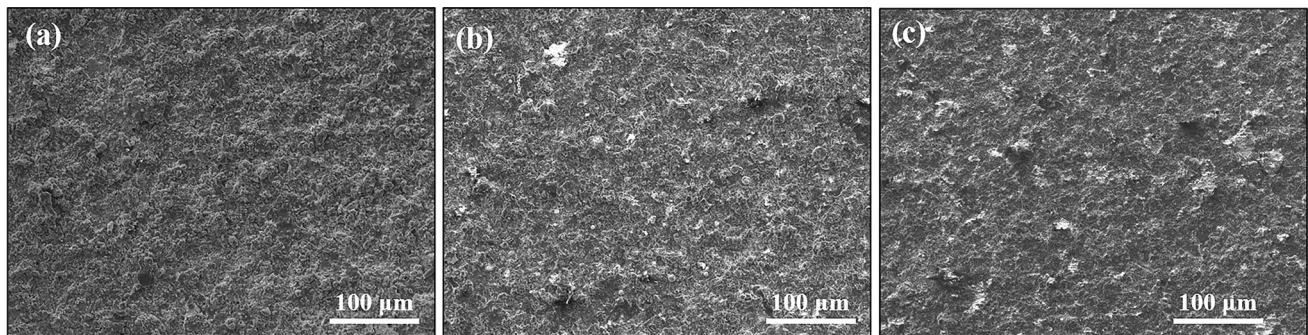
Element	Ca	Al	Si	Mg	O
at. %	12.35	22.28	24.20	0.36	Bal



**Fig. 8.** XRD patterns of (a) as-sprayed  $\text{Al}_2\text{O}_3/\text{YSZ}$  coating and (b) the coating with CMAS deposit after 24 h exposure at  $1250\text{ }^\circ\text{C}$



**Fig. 9.** Photographs of surfaces of as-sprayed PS–PVD  $\text{Al}_2\text{O}_3/\text{YSZ}$  coating sample (a) and the coating sample thermal cycled at 1050 °C for 100 h (b) and 500 h (c)



**Fig. 10.** Surface morphologies of as-sprayed PS–PVD  $\text{Al}_2\text{O}_3/\text{YSZ}$  coating sample (a) and the coating thermal cycled at 1050 °C for 100 h (b) and 500 h (c)

is no more than 2% of the whole surface after 500 h thermal cycling.

As a control, the thermal cycling test was also conducted for APS  $\text{Al}_2\text{O}_3/\text{YSZ}$  TBCs, whose failure occurred as soon as the thermal cycling time reached 300 h. This comparison shows that PS–PVD  $\text{Al}_2\text{O}_3/\text{YSZ}$  TBCs owns better thermal cycling durability than APS  $\text{Al}_2\text{O}_3/\text{YSZ}$  TBCs. A possible reason could be the better bonding strength between PS–PVD  $\text{Al}_2\text{O}_3$  and YSZ coating due to the high substrate temperature and well-melted  $\text{Al}_2\text{O}_3$  droplets.

## Conclusions

A thin and dense  $\text{Al}_2\text{O}_3$  overlayer was deposited by PS–PVD on YSZ TBC. Its resistance to CMAS interaction and thermal cycling behavior were investigated. Some conclusions can be drawn as follows:

1. The PS–PVD  $\text{Al}_2\text{O}_3$  overlayers PS–PVD are mainly deposited from well-melted droplets. They possess much lower porosity (optimal 0.82% deposited at the spray distance of 1400 mm) than that of APS  $\text{Al}_2\text{O}_3$  overlayer.
2. The thin and dense PS–PVD  $\text{Al}_2\text{O}_3/\text{YSZ}$  TBCs exhibit excellent CMAS resistance because of low porosity and formation of dense reaction layer composed of anorthite and spinel preventing the penetration of molten CMAS.
3. The PS–PVD- $\text{Al}_2\text{O}_3$ -coated specimens show a good thermal cycling lifetime more than 500 h, about 65% longer than that of APS- $\text{Al}_2\text{O}_3$ -coated specimens (300 h).

In summary, the  $\text{Al}_2\text{O}_3$  overlayer deposited on APS YSZ coating by PS–PVD exhibits excellent CMAS resistance and thermal cycling durability. The PS–PVD  $\text{Al}_2\text{O}_3$  overlayer can be regarded as one of the promising protective coatings for YSZ TBCs.

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