

Bending Fatigue of Carburized Steel at Very Long Lives

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The bending fatigue behavior of two carburized steels is investigated for lives between approximately $10⁵$ and $10⁸$ cycles. Cracks are observed to start at sub-surface inclusions and develop features on fracture surfaces resembling "fish eyes" in appearance. This type of sub-surface cracking tends to govern fatigue strength at long lives. Previous studies of ''fish eye'' fatigue in carburized steel have been relatively few and have mainly considered failures originating at depths beneath a carburized case, where compressive residual stresses are minimal and hardness values approach those in the core. This study provides fatigue data for cracks originating within cases at various depths where compressive residual stresses are substantial and hardness is much higher than in the core. Fatigue strength is predicted by a simple model, accounting for the influence of residual stresses and hardness values at the different depths at which cracks started. Predictions of fatigue strength are compared with data generated in this study.

Keywords carburizing, fish eye, heat treatment, high-cycle fatigue, life prediction, residual stress, steel

1. Introduction

Numerous mechanical components made of hardened steel are designed to have fatigue lives exceeding millions of cycles, a regime in which failures tend to originate at sub-surface inclusions (Ref [1](#page-5-0)). In that regime, a penny-shaped crack typically grows from an inclusion with a distinct appearance known as a ''fish eye.'' An example is shown in Fig. [1](#page-1-0). This type of fatigue failure has been the subject of increasing study during the past decade and is considered in a number of review articles (Ref [2-6](#page-5-0)). A zone of special interest surrounding an inclusion is often found within a fish eye. The zone is referred to as an optically dark area (ODA) (Ref [1\)](#page-5-0) (or in some cases, a granular bright facet (Ref [7](#page-5-0)), fine granular area (Ref [3\)](#page-5-0), or rough surface area (Ref [8\)](#page-5-0)). ODAs are believed to start early in fatigue life (first 10% or so of life) and gradually increase in size, occupying most of the fatigue life (Ref [9,](#page-5-0) [10\)](#page-5-0). Little life is typically spent in the growth of a crack from the ODA size to fracture (Ref [11\)](#page-5-0). An example of an ODA is shown in Fig. [2](#page-1-0).

To date, studies of ''fish eye'' fatigue have primarily used through-hardened specimens. Many mechanical components are case hardened, producing a profile in hardness and residual stresses as a function of depth below the surface. Carburizing is a widely used case-hardening process applied to gears, bearings, camshafts, etc. Not only does it alter sub-surface hardness and create residual stresses, but it also leads to hydrogen around internal non-metallic inclusions (Ref [12](#page-5-0)). In this study, the highcycle fatigue behavior of carburized gear steels is explored to assess the influences of residual stresses and hardness on ''fish eye'' fatigue cracks occurring within carburized cases.

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A 1968 study by Funatani and Noda (Ref [13](#page-5-0)) on the bending fatigue of carburized specimens noted that fish eye cracks initiated at depths approximately equal to case depth, where compressive residual stresses were relatively minimal (less than -100 MPa) compared to bending stresses. Naito et al. (Ref [14\)](#page-5-0) tested carburized specimens with either non-martensitic or electropolished surface layers. Surface failures originated in non-martensitic layers even at lives approaching $10⁸$ cycles (the highest number of cycles tested). In electropolished specimens, several fish eye failures at inclusions were noted for lives of about 5×10^6 cycles and greater, but sizes and depths of inclusions were not reported except in one photo where the depth was approximately $250 \mu m$, within the carburized case. Murakami et al. (Ref [15](#page-5-0)) tested carbonitrided specimens with a case depth of approximately 2 mm. Fish eye cracks started at inclusions deeper than the case. Residual stresses versus depth within the case were not reported, which is understandable since cracks did not originate there. Huang et al. (Ref [16\)](#page-5-0) tested carburized specimens with a relatively shallow case (approx. 0.3 mm). Depths of inclusions at which cracks formed and residual stresses versus depth were not mentioned.

This study differs from those just reviewed by reporting the results of fatigue tests in which fish eye cracks started within the cases of carburized specimens, at depths where magnitudes of compressive residual stresses were significant relative to applied stresses and where hardness values were much higher than in the core. To date, approaches for predicting ''fish eye'' fatigue strength have been developed largely using test data from specimens with minimal residual stresses and hardness nearly uniform with depth. This study provides an opportunity to assess the usefulness of a prominent approach for predicting fatigue strength for situations where failures may originate within the case, at depths where residual stresses are substantial and hardness values vary with depth.

2. Experimental Procedure

Two Cr-Ni carburized gear steels, 20Cr2Ni4 and 17Cr2Ni2, designated A and B, respectively, were tested. Table [1](#page-1-0) shows their nominal chemical compositions and tensile properties. The geometry and nominal dimensions of a first set of fatigue specimens made of alloys A and B are shown in Fig. 3. Fatigue tests with the first set involved loading levels that unexpectedly caused ball bearings in a rotating bending fatigue test machine to fail prematurely. A second set of smaller specimens with a nominal diameter of 5.6 mm allowed additional testing at lower load levels, avoiding further bearing failures. The second set is designated NA, NB. Both sets of specimens were carburized at 910 °C for 20 h with a carbon potential of 1.2% and then aircooled and held at 650 \degree C for 2 h. After that, the specimens were austenitized at 880 °C for 2 h, oil quenched, and tempered at 180 \degree C for 2 h. After carburizing and hardening, specimens were gently ground and polished in the longitudinal direction with abrasive paper (to grade #2000) to achieve a mirror-like finish.

Carburized cases were etched with picric acid to reveal prior austenite grain boundaries. Applying ASTM E112-10 (Ref [17\)](#page-5-0),

Fig. 1 Typical ''fish eye'' originating at a sub-surface inclusion in this study. (Specimen surface is shown to the left of the fish eye)

the average grain size was determined to be G8.5 and G9 for steels A and B, respectively, or 19 and $16 \mu m$. Micro-hardness profiles of carburized cases determined according to ASTM E384-10 (Ref [18\)](#page-5-0) are reported in Fig. [4.](#page-2-0) Specimens made of alloy B developed a somewhat harder case. Longitudinal residual stresses versus depth were measured by x-ray diffraction using conventional Cr-Ka radiation (2.291A wavelength), [211] planes, and the $\sin^2\psi$ method combined with layer removal by electropolishing. Stresses were corrected for the effect of layer removal (Ref [19](#page-5-0)). Residual stresses measured in specimens that had been cycled without failure are shown in Fig. [5.](#page-2-0) Compressive residual stresses in representative specimens A and NA were larger than those in the corresponding B and NB specimens. In the second set of specimens (NA,NB) with a somewhat smaller diameter, significant residual stresses were present over shallower depths than in the first set. Magnitudes of residual stresses in Fig. [5](#page-2-0) for depths greater than about 0.2 mm are comparable to typical values for carburizing in the range of -250 to -350 MPa (Ref [20](#page-5-0)), while values near the surface are in line with those approaching -800 MPa reported for carburized and ground gear teeth (Ref [21\)](#page-5-0). The magnitudes of residual stresses measured suggest that little or no residual stress relaxation occurred from fatigue cycling. The yield strength near the surface of specimens was estimated to be approximately 2000 to 2100 MPa based on correlations between hardness and yield strength for carburized steel (Ref [22](#page-6-0)). With the exception of one specimen, the bending stress amplitude at the surface was 1150 MPa or less in the testing to be described shortly, so that the sum of applied and residual stresses would not be expected to cause significant relaxation of residual stresses by plastic deformation (Ref [23](#page-6-0)).

3. Experimental Results

Fatigue tests were conducted at room temperature in air. The loading frequency was 75 Hz. The results of fatigue testing are

Fig. 2 Optically dark area surrounding an inclusion Fig. 3 Geometry of fatigue specimen (dimensions in mm)

Table 1 Chemical composition (wt.%) and tensile properties of the steels prior to carburizing

Alloy		Si	Mn	Cr	Ni	Mo	$S_{\rm w}$ MPa	$S_{\rm m}$, MPa	RA, %
A	$0.17 - 0.23$	$0.17 - 0.37$	$0.3 - 0.6$	1.25-1.65	3.25-3.65	$0.1 - 0.15$	1292	1483	
B	$0.14 - 0.19$	$0.17 - 0.37$	$0.3 - 0.6$	$1.5 - 1.8$	1.4-1.7	$0.25 - 0.35$	1150	1400	35
	S_v yield strength, S_u ultimate tensile strength, RA reduction of area								

Fig. 4 Micro-hardness profiles

Fig. 5 Residual stress profiles, with arrows showing depths at which cracks started

shown in Fig. 6, where bending stress amplitude refers to values at the surface. The fracture surfaces of the failed specimens were examined by optical microscopy as well as with a scanning electron microscope (SEM) equipped with an x-ray energy-dispersive spectrometer (EDS). Based on the result of EDS, all internal fractures originated at non-metallic inclusions that were either Al_2O_3 or TiN. Sizes of inclusions in terms of a $\sqrt{\text{area}}$ parameter ranged between 16 and 49 μ m, where *area* is the projected area of a defect or crack on a plane normal to the direction of maximum tensile stress (Ref [1](#page-5-0)). The $\sqrt{\text{area}}$ parameter will be used later in a model for fatigue strength. In the case of a circular area, the parameter $\sqrt{\text{area}}$ = 0.89 \times diameter. Inclusions causing fish eye cracks were observed at depths ranging from 10 to 745 μ m. ODAs could be identified at most crack initiation sites, with $\sqrt{\text{area}}$ values (including the area of the inclusion within an ODA) between 30 and 99 μ m. Figure [7](#page-3-0)(a) shows an example of a fish eye fracture that initiated from an unbroken Al_2O_3 inclusion, while Fig. [7\(](#page-3-0)b) shows a pore left by an inclusion debonded from the matrix, suggesting relatively weak bonding between the inclusion and matrix. Several radial ridges emanated from the crack origin. The length of these ridges ranged from 16 to 100 μ m, and many of the short ridges stopped at the outer border of ODAs. Nishijima and Kanazawa (Ref [24\)](#page-6-0) have suggested that such ridges indicate shear cracking similar to what might be found for cracks that initiate at a surface. The size of each ODA, normalized by the size of the corresponding

Fig. 6 Fully reversed bending fatigue data, with symbol S indicating cracking originating at the surface and C denoting a combination of surface cracking and sub-surface fish eye cracking

inclusion from which a fish eye crack started, is shown in Fig. [8.](#page-3-0) Although there is significant scatter, the ratio of ODAto-inclusion size tends to increase at longer lives. Finally, it is of interest to note that one NB specimen tested at 850 MPa had an unexpectedly short life, as seen in Fig. 6. Examination of the fracture surface suggested that failure originated at a large surface inclusion, with no evidence of fish eye cracking.

Figure 5 shows depths of inclusions at which fish eye cracks formed relative to compressive residual stress versus depth. Most of the cracks started at locations with significant residual stress. There was no evidence of oxidation or decarburization at any of the locations where cracks started.

The fatigue strength, σ_w (MPa), of each specimen with a fish Fire rangue such gut, σ_w (wir a), or each specified with a fishing eye fracture was estimated by Murakami's $\sqrt{\text{area}}$ parameter model (Ref [1](#page-5-0)) given in Eq 1, which was derived from a fracture mechanics analysis along with data from specimens of uniform hardness and assumed minimal residual stress. The model has become prominent in evaluations of fatigue strength governed by cracking from inclusions:

$$
\sigma_{w} = \frac{1.56 \left(\text{Hv} + 120 \right) \left[(1 - R)/2 \right]^{\beta}}{\left(\sqrt{\text{area}_{\text{inc}}} \right)^{1/6}}, \tag{Eq 1}
$$

where Hv is Vickers hardness, area_{inc} is the projected area (μm) of an inclusion normal to the direction of stress, R is the ratio of minimum to maximum stress in a cycle, and β is given by

$$
\beta = 0.226 + 10^{-4} \,\text{Hv} \tag{Eq 2}
$$

$$
R = (\sigma_{\rm m} - \sigma_{\rm w})/(\sigma_{\rm m} + \sigma_{\rm w}), \tag{Eq 3}
$$

where $\sigma_{\rm m}$ is the mean value of applied stress.

Here, the influence of residual stresses σ_{res} is taken into account as if it was a mean stress $\sigma_{\rm m}$ in Eq 3. The bending stress amplitude σ_a at the sub-surface location of an inclusion was obtained from

$$
\sigma_{\rm a} = \sigma_{\rm nominal} \left(1 - \frac{d_{\rm inc}}{r_s} \right),\tag{Eq 4}
$$

where $\sigma_{nominal}$ is the bending stress amplitude at the surface, d_{inc} is the depth of an inclusion from the surface, and r_s is the measured radius of the minimum section of specimens. The values of hardness and residual stress measured at the depth of a given inclusion were used in Eq 1-3. Since σ_w is a function of R in Eq [1,](#page-2-0) and R is a function of σ_w in Eq [3,](#page-2-0) determination of both can start by taking $\sigma_w = \sigma_a$ to obtain an initial value of R, and then to find an updated value of σ_w , which is used to find an updated value of R , and so forth until convergence is obtained.

Figure $9(a)$ $9(a)$ shows the relation between the ratio of the applied stress amplitude to the estimated fatigue strength, $(\sigma_{\alpha}/\sigma_{\alpha})$ σ_w), and the number of cycles to failure, N_f. The ratio (σ_a/σ_w) is conservative, except where it drops slightly below unity approaching 10⁸ cycles. The values of $(\sigma_{\alpha}/\sigma_{\rm w})$ provide evidence that Murakami's model in Eq [1](#page-2-0) can also be effective for fatigue strength estimates of carburized and ground steel when fish eye cracks start at depths in carburized cases where compressive residual stresses are significant and fatigue lives are in the range of 10⁵ to 10⁸ cycles. The relation between ($\sigma_{\alpha}/\sigma_{\rm w}$) and the depth of inclusions is shown in Fig. [9](#page-4-0)(b). There is no apparent dependence of (σ_{a}/σ_{w}) on the depth. If desired, the use of $\sqrt{\text{area}_{ODA}}$ rather than $\sqrt{\text{area}_{inc}}$ in Eq [1](#page-2-0) would provide lower, more conservative estimates of predicted fatigue strength σ_w (approximately 7% more conservative for the data here).

It has been proposed that there may be two kinds of crack growth thresholds for fish eye failures. A larger one, characterized by the stress intensity factor range ΔK at the outer border of ODAs, represents a threshold for a transition from small crack growth to "Paris law" behavior (Ref [25\)](#page-6-0), and a smaller one, characterized by ΔK at the border of inclusions, represents a threshold for crack growth to occur from an inclusion (Ref [26](#page-6-0)). To account for the effect of compressive residual stress, the stress intensity factor range at the border of an inclusion, ΔK_{inc} , and at the outer border of an ODA, ΔK_{ODA} , an incrusion, ΔR_{inc} , and at the outer border of an ODA, ΔR_{ODA} , was computed using an $\sqrt{\text{area}}$ model (Ref [1](#page-5-0)) modified here to account for the effect of residual stresses:

$$
\Delta K_{\text{inc}} = 0.5 \left(\sigma_a + \sigma_{\text{res}} \right) \sqrt{\pi \sqrt{\text{area}_{\text{inc}}}} \tag{Eq 5}
$$

$$
\Delta K_{\rm ODA} = 0.5 \left(\sigma_a + \sigma_{res} \right) \sqrt{\pi \sqrt{\text{area}_{\rm ODA}}}, \tag{Eq 6}
$$

where $\sqrt{\text{area}_{\text{inc}}}$ and $\sqrt{\text{area}_{\text{ODA}}}$ are the projected areas of an inclusion and an ODA, respectively. The values of applied stress amplitude, σ_a , and residual stress, σ_{res} , were taken at the sub-surface location of a given inclusion. (Compressive σ_{res} has a negative sign in Eq 5 and 6).

Figure [10](#page-4-0) shows ΔK_{inc} and ΔK_{ODA} values versus cycles to failure N_f . For the A specimens, ΔK_{inc} varies between 2.6 and 3.4 MPa \sqrt{m} with an average of 3.0 MPa \sqrt{m} , while ΔK_{ODA} varies between 3.6 and 4.9 MPa \sqrt{m} with an average of 4.3 MPa \sqrt{m} . For the NA specimens, ΔK_{inc} varies between 1.1 and 3.1 MPa \sqrt{m} with an average of 2.3 MPa \sqrt{m} , while ΔK_{ODA} varies between 1.7 and 5.5 MPa \sqrt{m} with an average of 4.5 MPa \sqrt{m} . For the B specimens, ΔK_{inc} varies between 2.3 and 3.9 MPa \sqrt{m} with an average of 2.9 MPa \sqrt{m} , while ΔK_{ODA} varies between 3.8 and 5.4 MPa \sqrt{m} with an average of 4.7 MPa \sqrt{m} . For the NB specimens, ΔK_{inc} varies between 2.9 and 3.8 MPa \sqrt{m} with an average of 3.3 MPa \sqrt{m} , while ΔK_{ODA} varies between 4.2 and 4.9 MPa \sqrt{m} with an average of 4.5 MPa \sqrt{m} . The average value of ΔK_{inc} for all of the specimens in which $\alpha \sqrt{m}$, the average value of ΔK_{inc} for an or the specificity in this study was 2.9 MPa \sqrt{m} , while the average value of ΔK_{ODA} was 4.5 MPa \sqrt{m} . Those values, which were computed to account for the presence of significant residual stresses, compare well with the values in the range of ΔK_{inc} from 2 to EUREATE WITH THE VALUES IN THE TANGE OF ΔK_{inc} from 2 to 4 MPa \sqrt{m} for steels with similar martensitic microstructures and hardness but without residual stresses from case hardening (Ref [26-35](#page-6-0)). The values of ΔK_{ODA} computed here fall in the range of 3-6 MPa \sqrt{m} typically reported for such steels without residual stresses (Ref [26](#page-6-0), [27](#page-6-0), [30-37\)](#page-6-0).

If residual and applied stresses versus depth within a carburized case are known, then ΔK_{inc} and ΔK_{ODA} can be computed as a function of depth for a range of inclusion sizes expected to assess the potential for fish eye failures. This could

Fig. 8 Ratio of the size of an ODA to that of the inclusion from which fish eye cracking developed

Fig. 7 (a) Inclusion at the center of a fish eye for a specimen of steel A, $\sigma_a = 1041$ MPa, $N_f = 2{,}651{,}000$ cycles, $\sqrt{\text{area}} = 28$ µm; (b) pore at the center of a fish eye for a specimen of steel B, $\sigma_a = 1067$ MPa, $N_f = 1,149,800$ cycles, $\sqrt{\text{area}} = 35$ µm

Fig. 9 Fish eye failures evaluated using Murakami's \sqrt{area} model (Eq [1](#page-2-0)): ratio of applied stress amplitude to estimated fatigue strength vs. (a) cycles to failure and (b) depth of inclusions

Fig. 10 Stress intensity factor range at inclusions and at ODAs vs. cycles to failure for (a) alloy A and (b) alloy B

Fig. 11 Signature of hydrogen at an inclusion, observed by tritium microautoradiography. Image modified from (Ref 38) and \odot 2005 by the American Nuclear Society, LaGrange, Illinois

provide information useful for alloy selection, design analysis, and heat treatment. For instance, trade-offs between fish eye fatigue strength and inclusion sizes, applied stress levels, etc. could be explored.

4. Discussion

As noted in the introduction, most of the fatigue life associated with the fish eye phenomenon is believed to be spent in the formation of ODAs at inclusions. Different mechanisms have been suggested for the development of ODAs. Murakami et al. (Ref [15](#page-5-0)) suggested that hydrogen plays an important role in fish eye crack development, perhaps by embrittlement. Figure 11 shows an example of a "signature" of hydrogen trapped at an inclusion as observed by a specialized technique known as tritium automicroradiography (Ref [38](#page-6-0)). Shiozawa et al. (Ref [9,](#page-5-0) [39](#page-6-0)) proposed that ODAs are formed by carbide decohesion, growth of microcracks along boundaries between carbides and the matrix, plus microcrack coalescence. Hydrogen is known to have the potential to reduce cohesive strength at carbide-to-matrix interfaces (Ref [40](#page-6-0)), and it seems plausible that hydrogen may promote the mechanism suggested by Shiozawa et al. Sakai (Ref [3](#page-5-0)) suggested a mechanism for development of ODAs in which very fine sub-grains with different crystallographic orientations are formed by microscale polygonization and debonding, along with coalescence of the micro-debondings. Grad et al. (Ref [41](#page-6-0)) found that cyclic loading combined with the stress concentration created by inclusions caused significant grain refinement within ODAs. As Li (Ref [4](#page-5-0)) has noted, hydrogen-assisted local plasticity may

promote dislocation activity and multi-slip around inclusions. That influence may, in turn, encourage mechanisms such as polygonization and grain refinement.

When it comes to the influence of hydrogen on the phenomenon of fish eye fatigue originating within carburized cases, relatively little appears to be known. Gas carburizing and direct hardening have been found to produce a substantial initial hydrogen content of about 2 ppm (Ref [42\)](#page-6-0), while tempering at $200 \degree C$ in air for one hour in air resulted in approximately 50% hydrogen effusion. Tempering in vacuum caused 75% effusion in 30 min. Effusion during tempering in air was likely inhibited by an increasingly thick oxide layer (Ref [42](#page-6-0)). Empirical corrections to Eq [1](#page-2-0) have been proposed (Ref [43](#page-6-0), [44](#page-6-0)) to account for effects of hydrogen concentration on fatigue strength based on tests of small, through-hardened alloys. Similar relations for fish eye cracks originating within carburized cases remain to be explored. It may be of interest in future studies of fish eye fatigue to investigate how hydrogen content varies with depth within carburized cases using a measurement approach like that applied by Murakami and Matsunaga (Ref [45\)](#page-6-0) to hydrogen-charged specimens of stainless steels. Carburized specimens with and without oxidized layers might be tested to see how hydrogen content and subsurface concentration versus depth differ.

5. Conclusions

- 1. Previous studies on fish eye fatigue in carburized specimens in which depths of crack origins were reported showed that origins were typically at depths where compressive residual was minimal and hardness values were close to those in the core. By contrast, most fish eye fatigue cracks started at sub-surface inclusions within the case of carburized specimens in this study, at depths where compressive residual stresses were significant and where hardness values were much higher than in the core.
- core.
2. Murakami's $\sqrt{\text{area}}$ parameter model was found to be effective in predicting fatigue strength by taking into account hardness and compressive residual stresses at the depths of inclusions from which fish eye cracks originated.
- 3. An average value of stress intensity factor range ΔK_{inc} at the border of inclusions, computed to account for residual stresses and applied stresses, was 2.9 MPa \sqrt{m} , comparable to values in the literature for specimens that had minimal residual stresses at locations where fish eye cracks originated. ΔK_{inc} can be considered a threshold for crack growth to occur from an inclusion. An average value of $\Delta K_{\rm ODA}$ at the outer border of ODAs, also computed to ΔK_{ODA} at the other border of ODAs, also computed to account for residual stresses, was 4.5 MPa \sqrt{m} , again comparable to the values from specimens with minimal residual stresses. $\Delta K_{\rm ODA}$ can be considered a threshold for a transition from small crack growth to ''Paris law'' behavior. The ability to account for the influence of residual stresses in determining ΔK_{inc} and ΔK_{ODA} by superposition of applied and residual stresses is supported by the data generated in this study.

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