Freckle Formation and Freckle Criterion in Superalloy Castings

P. AUBURTIN, T. WANG, S.L. COCKCROFT, and A. MITCHELL

The evaluation of a numerical criterion to provide quantitative insight on freckling conditions is critical to the successful manufacture of large superalloy castings. Of the criteria reported in the literature, those based on the Rayleigh number seem best suited to predict the onset of freckle formation. However, in their current form, these criteria cannot explain why freckles develop predominantly at the surface of single crystal (SX) castings and at midradius in VAR/ESR ingots. An experimental Bridgman-type furnace has been built to directionally solidify freckle-prone superalloys, CMSX-11B, RENÉ88, NIM80A, WASPALOY, MAR-M247, and a variation of IN718 with high silicon content, at various angles to the vertical. Under typical industrial solidification conditions (thermal gradient between 500 and 4000 K m⁻¹ (5 $<$ G $<$ 40 °C cm⁻¹) and solidification rate between 1.67 \times 10⁻⁵ and 1.0×10^{-4} m s⁻¹ ($1 \le R \le 6$ mm min⁻¹)), the results indicate a dependency of freckling on growth front angle likely related to the anisotropy in permeability. A modified Rayleigh criterion has been developed which accounts for directional permeability and orientation of the growth front relative to the gravity vector. Application to the experimental data shows good correlation with the onset of freckling for the range of solidification conditions examined in the study. The approximate threshold value for the modified Rayleigh number was estimated to be for $CMSX-11B$, 0.88, for RENÉ88, 0.90, for NIM80A, 0.85, for WASPALOY, 0.95, for MAR-M247, 0.86, and for IN718-Si, 0.65.

Freckles are a common defect found in nickel-based undesirable in critical applications because of their deleteri-
ous effect on mechanical performance. Moreover, castings straightforward "heat-flow" mathematical models. ous effect on mechanical performance. Moreover, castings. containing freckles must be scrapped, causing considerable economic loss because they cannot be removed by postcast, thermomechanical treatments. B. *Literature Review* Development in large aircraft engines and also in large

and-based gas turbines for power generation requires a con-
siderable scale-up in the diameter of turbine disks and in
the size of turbine blades. One of the main problems encoun-
tered in scaling-up has been the extensiv tion rates. Traditionally, for example, freckle formation in
in thermal gradient G , local solidification rate R , and local
ingots can be avoided by keeping ingot diameters and melt-
ing rates below critical values, wh

I. INTRODUCTION mathematical criteria;^[1–4] the other, based on solution of mass, momentum, energy, and species conservation equa-A. *Background*

finals, momentum, energy, and species conservation equa-

Freckles are a common defect found in nickel-based and may ultimately be applicable to a broader range of superalloy castings. They appear as a long trail of equiaxed conditions, the computational severity of this approach rains with a composition shift consistent with alloy segrega-
together with the absence of high-temperatu grains with a composition shift consistent with alloy segrega-
together with the absence of high-temperature data presently
ion. Some porosity and feeding shrinkage may also be pres-
precludes its practical application to tion. Some porosity and feeding shrinkage may also be pres-
ent in and adiacent to the freckle line. Freckles are highly Thus, given the present limitations, the best approach seems ent in and adjacent to the freckle line. Freckles are highly Thus, given the present limitations, the best approach seems
undesirable in critical applications because of their deleteri-
to be to adopt mathematical criteria

occurrence has been avoided by maintaining high thermal cally more correct criterion based on the Rayleigh number

gradients at the solidification front.

Two general approaches appear in the literature for pre-

dicting t per unit area) and, as such, may be employed to characterize the onset of fluid flow in unstable systems.^[6,13]

P. AUBURTIN, formerly Postdoctoral Candidate, Department of While the basic mechanism of channeling as a precursor etals and Materials Engineering, University of British Columbia, is Pro-
to freckling appears clear, there cess Modeling Research Engineer with PSA-Peugeot Citroen, Bievres, to where the channel actually nucleates in the mushy region.
France 91570. T. WANG, Postdoctoral Candidate, S.L. COCKCROFT, For example Magirl and Incroper France 91570. T. WANG, Postdoctoral Candidate, S.L. COCKCROFT, For example, Magirl and Incropera^[14] and Sarazin and Hella-
Associate Professor, and A. MITCHELL, Professor, are with the Department wall[15] proposes that Associate Professor, and A. MITCHELL, Professor, are with the Department
of Metals and Materials Engineering, University of British Columbia, Can-
ada V6T 1Z4.
Manuscript submitted August 23, 1999.
Manuscript submitted Aug Fowler,^[5] Worster,^[6] and Tait and Jaupart^[16] conclude that

Metals and Materials Engineering, University of British Columbia, is Process Modeling Research Engineer with PSA-Peugeot Citroen, Bievres,

the onset of convection and channels in the mushy zone is independent of boundary layer instabilities at the solid/liquid interface. And in yet another study, Auburtin *et al.*^[17] have

and Pb-Sb alloys, Sarazin and Hellawell^[15] employed the regime (*i.e.*, at what fraction solidified), what the appropriate Rayleigh number of the form given in Eq. [1] to describe characteristic length should be, and w Rayleigh number of the form given in Eq. $[1]$ to describe the onset of flow. The conset of flow. The conset of Rayleigh number should be adopted as the criterion in

$$
Ra = \frac{g \frac{d\rho}{dz}}{\frac{\eta D_T}{h^4}} \tag{1}
$$

list. By assuming a critical Rayleigh number of unity, they together with a modified, or alternative, Rayleigh number
were able to back calculate the magnitude of the characteris- for superalloy casting processes. Employin were able to back calculate the magnitude of the characteris-
tic length scale, h, which was found to be on the order of leigh number, a characteristic length is set and critical values, tic length scale, *h*, which was found to be on the order of the primary dendrite spacing λ_1 . Other authors^[6,16,18,19] have Ra*, are estimated quantitatively for the experimental adopted a Rayleigh number of the form castings.

$$
Ra = \frac{g\Delta\rho K_0 h}{\eta D_T} \tag{2}
$$

where K_0 is the representative flow permeability of the mushy zone. The characteristic length *h* has been assigned various values. For example, Tait and Jaupart^[16] and Worster and Kerr^[20] considered h as the thickness of the mushy zone. Whereas, Worster and Wettlaufer^[21] and Lu and Chen^[19] assumed *h* to be the thermal diffusion length $h = D_T / R$. Owing to variations in the characteristic length, the numerical value of the critical Rayleigh number in Eq. [2] is reported to vary in magnitude from 20 to several hundred.^[22,23]

In addition to the uncertainties discussed, there are other troubling limitations with the basic Rayleigh number/criterion from a process application standpoint. For example, in vacuum arc remelting and electroslag remelting (VAR/ESR) ingots, freckles are found predominately at the midradius in ingots and not in the center where the Rayleigh number is at a maximum.[2] One possible explanation may lie in results of an early study conducted by Copley *et al.*[3] on the transparent analog system ammonium-chloride/water (NH4Cl/ $H₂O$). In their investigation, they found that in straightforward vertical solidification experiments, where the growth front was horizontal and flat, the freckles seemed to be randomly distributed across the casting. However, when the growth front was convex (higher at the center than the edges), the freckles appeared preferentially at the center. Alternatively, when the growth front was concave (higher at the edges than the center), freckles formed preferentially at the outside surface of the casting.

The importance of orientation on flow initiation in an unstable system is confirmed in another study conducted by Hart^[13] aimed at investigating the evolution of convection patterns in water. In this investigation, Hart found that the Fig. 1—Schematic illustration depicting freckle formation and associated
fluid flow pattern.
onset of unstable convection was best described by the value
of the Rayleigh number of the system and that the critical threshold value was dependent on the tilt angle of the experimental apparatus *via* the following relationship:

Unstable flow for
$$
Ra > Ra^* = 1708/\sin \alpha
$$
 [3]

interface. And in yet another study, Auburtin *et al.*^[17] have
found that the channels appear to originate at a solid fraction
of 0.5 in directionally cast superalloy samples.
The exact form of the Rayleigh number and t superalloys. Moreover, it would appear that the basic criterion could benefit from incorporation of a dependency on growth front angle relative to the gravity vector.

In the present work, the influence of growth front angle on freckle formation in industrial superalloys is demonstrated in a series of experiments. Following this, a mechanism is The parameters in Eq. [1] are defined in the Nomenclature proposed to explain the influence of growth front angle
list. By assuming a critical Rayleigh number of unity, they together with a modified, or alternative, Raylei

*CMSX-11B is a trademark of Cannon-Muskegon Corp., Muskegon, MI. Figure 2.

 \ddagger Waspaloy is a trademark of Pratt & Whitney, East Hartford, CT. Mar-M247 is a trademark of Martin-Marrietta Corp., Baltimore, MD. IN718 is then be withdrawn from the hot zone toward the cold zone a trademark of Inco Alloys International, Huntington, WV. a prescribed rate. Further details o

and IN718. All these alloys are nickel-based superalloys
prone to freckle formation. The nominal compositions and
melting range of these alloys can be found in Table I. The
melting ranges of alloys CMSX-11B, NIM80A, and I and MAR-M247 were taken from Reference 25. In an adjusted by the operator between 1.07 × 10 and 1.0 ×
attempt to induce upward channel flow in the IN718, pure
Si was added to the commercially available IN718 alloy to
make

So was a commonlered to the matter in Ty18 have been reported to

make IN718-Si (freckles in IN718 have been reported to

make IN718-Si (freckles in IN718 have been reported to the temperature of the suceptor relative to and chill, and a variable speed withdrawal mechanism. The trackie train was completed using an energy dispersion spec-
main feature of this furnace that differentiates it from its trometer (EDX) microprobe (KEVEX* detector industrial counterparts is its ability to be tilted from 0 deg *KEVEX is a trademark of Kevex Corporation, Foster City, CA. (conventional vertical operation) to 35 deg to the vertical.

solidification, type-D tungsten-rhenium (W-3 pct Re/W-25 tron microscope (SEM). The procedure involved locating

II. EXPERIMENTAL METHODOLOGY pct Re) thermocouples were used. All thermocouples were Six industrial alloys were selected for this work: CMSX-

11B,* RENÉ88,** NIM80A† WASPALOY, \ddagger MAR-M247,

11B,* RENÉ88,** NIM80A† WASPALOY, \ddagger MAR-M247,

119,* RENÉ88,** NIM80A† WASPALOY, \ddagger MAR-M247,

FRENESS IS a trademark of General Electric Aircraft Engines, Fair-

Field, CT.

Thims0A is a trademark of Inco Alloys International, Huntington, WV.

Thims0A is a trademark of Prot & Whitney East Hartford CT Mar.

allow th at a prescribed rate. Further details on the furnace can be

 10^{-4} m s⁻¹ (1 and 6 mm min⁻¹). The thermal gradient at the

To facilitate control and the recording of data relating to XOne analyzer) attached to a Hitachi S-2300 scanning elec-

Alloy	$CMSX-11B$	RENÉ 88	NIM80A	IN718*	WASPALOY	MAR-M247
T_S - T_L (K)	1548 to 1609	1523 to 1628	1586 to 1652	1526 to 1620	1603 to 1628	1553 to 1633
Ni	bal.	bal.	76.0	53.46	bal.	bal.
Co	7.0	13.0		0.2	12.3	10.0
Cr	12.5	16.0	19.5	18.12	19.0	8.5
Al	3.6	2.1	1.4	0.46	1.2	5.5
Ti	4.2	3.7	2.4	1.0	3.0	1.0
Hf	0.04					\cdot 4
Mo	0.5	4.0		2.96	3.8	0.6
Ta	5.0					3 \cdot
Nb	0.1	0.7		5.27		
W	5.0	4.0				10.0
C		0.03	0.06	0.031		
B		0.015	0.03			
Mn			0.3			
Si.			0.3	0.08		
Zr			0.06			
Fe				18.24		

Table I. Nominal Composition (in pct) and Melting Range of Cast Superalloys

*The tested alloy IN718-Si was made by adding 0.5 pct Si into IN718 alloy, with the resulting composition, as shown in Table II under "Matrix."

Table II. Chemical Analysis of the Matrix and Freckles Found in Experimental Castings

		WASPALOY		MAR-M247		CMSX-11B		$IN718-Si$	
Alloy	Matrix (Wt Pct)	Freckle (Wt Pct)							
Ni	56.89	56.96	57.42	56.58	59.02	61.75	53.41	51.88	
Co	13.90	13.30	10.35	9.79	5.51	4.49			
Cr	19.95	19.04	8.30	8.25	11.34	5.59	18.58	16.08	
Al	1.64	1.59	5.79	5.91	2.93	3.32	0.38	0.33	
Ti	3.03	4.12	0.78	1.01	3.93	7.08	0.90	1.39	
Hf			1.73	2.71	0.16	0.22			
Mo	4.59	4.99	0.62	0.76	0.48	0.26	2.86	3.79	
Ta			4.08	5.16	6.23	9.40			
Nb					0.13	0.11	4.05	8.86	
W			10.93	9.83	9.90	7.77			
Si							0.62	1.35	
Fe							19.45	16.33	

Fig. 2—Schematic diagram of the Bridgman-type furnace used in this study and the locations of thermocouples TC1, TC2, TC3, TC4, and TC5 used during the calibration of the PROCAST model. Fig. 3—Axisymetric mesh geometry used in PROCAST to model the exper-

freckle sites first at the etched cross section of samples by the darker appearance of freckles than the matrix. The freckle sites were then marked on the polished surface for SEM furnace enclosure. The furnace enclosure was modeled to analysis.

tice.^[10] and was calibrated against thermocouple measurements at various locations within the sample taken under a this manner, the initial temperature distribution in the casting susceptor temperature). The FE mesh used in the analysis is shown in Figure 3. The model included the casting, alu- Overall, the model was able to reproduce the observed

imental furnace.

alysis.
Because the thermocouple data provides an incomplete is withdrawn from the hot zone. Figure 4 schematically Because the thermocouple data provides an incomplete is withdrawn from the hot zone. Figure 4 schematically picture of the solidification conditions prevailing at the illustrates the various boundary conditions applied in picture of the solidification conditions prevailing at the illustrates the various boundary conditions applied in the growth front, the furnace was also modeled using the com-
model. The values adopted for the various inte growth front, the furnace was also modeled using the com-
model. The values adopted for the various interfacial heat
mercial finite element (FE) package ProCAST*. The model
transfer coefficients, material emissivities, and transfer coefficients, material emissivities, and temperature *ProCAST is a trademark of UES Inc., Dayton, OH. constraints are presented in Table III together with the initial conditions for the various materials. To simulate the casting used the physical properties of IN718, as is common prac-
tice,^[10] and was calibrated against thermocouple measure-
casting stationary until thermal equilibrium is reached. In variety of operating conditions (*i.e.*, withdrawal rate and and furnace components is determined prior to commencing susceptor temperature). The FE mesh used in the analysis withdrawal of the ingot.

mina tube (mold), steel spacer, water-cooling tubes, and temperature variation satisfactorily with deviations of less

$(I = \text{Interface heat transfer}, R = \text{Radiation})$

mental tiltable furnace. The gradient parallel to the gravity vector at the freckle

parable to the estimated precision of the thermocouple measurements. A typical comparison between the thermocouple measurements and the predicted temperatures is shown in Figure 5. In Figure 5, the thermocouple data appear as the
dark lines, and the model predictions appear as the lighter
lines. Additional details on the model and its calibration can (where R is the solidification rate, an be found in Reference 29. The remaining terms, *g*, *d_p*/ gradient at the liquidus). Finally, the remaining terms, *g*, *d_p*/

A. *Analysis of Experimental Results*

Longitudinal sections of typical castings solidified at angles of 0 and 35 deg to the vertical are shown in Figure 6. The initial liquidus isotherm (melt limit of charge) is identified by the dashed lines appearing on the photomicrographs. For the 0 deg sample, the liquidus isotherm is initially flat and perpendicular to the long axis of the casting. The
grains have grown parallel to the long axis of the casting In the present study, the freckle initiation temperature is grains have grown parallel to the long axis of the casting

convection in the melt that develops during the holding delay prior to commencing the withdrawal. It is also apparent that there was some vertical grain growth initially. However, after a short distance, the grains become aligned with the direction of heat flow. Thus, in the steady-state growth regime, it is possible to conclude that the growth front is approximately perpendicular to the long axis of the casting independent of furnace angle. Furthermore, if that holds, then the tilt angle of the furnace is representative of the angle of the growth front relative to the vertical or gravity vector.

The typical appearance of freckles in a casting is shown in Figure 7. It can be seen that freckles appear on the surface of the sample, similar to what has been observed by Giamei and Kear.^[30] The results of a typical microprobe analysis of the freckles are presented in Table II and indicate segregation of alloy constituents. The amount of the shift in composition is similar to that found in freckles in industrial castings, $[29]$ confirming that the segregates in the experimental samples are indeed freckles.

B. *Application of the Conventional Rayleigh Criterion*

Because of the uncertainty over choosing an appropriate characteristic length, it seems reasonable to adopt the approach suggested by Sarazin and Hellawell^[15] of setting $h = \lambda_1$, which should yield Rayleigh numbers around 1. For the sake of convenience, the original Rayleigh number from Eq. [1] has been rearranged to the form given in Eq. [4].

$$
Ra = 4.16 \times 10^{-6} G_F^V (G_L R)^{-1.32}
$$
 [4]

This manipulation has been accomplished as follows: first, the expression for the gradient in density in Eq. [1] has been replaced by

$$
\frac{d\rho}{dz} = \frac{d\rho}{dT} \times \frac{dT}{dz} = \frac{d\rho}{dT} G_F^V
$$
 [5]

Fig. 4—Schematic heat flow diagram in the PROCAST model of the experi-
Where G_F^V represents the vertical temperature gradient (*i.e.*, initiation temperature).

Second, the characteristic length, *h*, set equal to the prithan 15 K throughout the solidification range, which is com-
to the following expression:^[31,32]
the following expression:^[31,32]

$$
\lambda_1 = \frac{150 \times 10^{-6}}{(G_L \, R)^{0.33}} \tag{6}
$$

dT, η , and D_T have been evaluated as given subsequently **and have been lumped into a single proportionality constant independent of alloy:**^[2] **III. RESULTS AND DISCUSSION** independent of alloy:^[2]

$$
\frac{d\rho}{dT} = 30 \text{ (kg m}^{-3} \text{ K}^{-1})
$$
 [7]

$$
g = 9.81 \, \text{(ms}^{-2)} \tag{8}
$$

$$
\eta = 0.004 \text{ (kg m}^{-1} \text{ s}^{-1}) \tag{9}
$$

$$
D_T = 9 \times 10^{-6} \, (\text{m}^2 \, \text{s}^{-1}) \tag{10}
$$

in alignment with the direction of heat flow. In the sample assumed to be consistent with a fraction solidified $f_s = 0.5$, solidified at 35 deg, there is some curvature likely due to which is based on the results of a pre which is based on the results of a previous study.^[17,29] The

Fig. 5—Comparison between measured (bold lines) and predicted (thin lines) thermal history at five locations inside the casting (130, 122, 117, 110, and 100 mm above the top of the chill; see TC1, TC2, TC3, TC4, and TC5 in Fig. 3). (Control temperature: 1708 K; withdrawal rate: $3.3 \times$ 10^{-5} ms⁻¹).

term $d\rho/dT$ is a function of both thermal and solute effects and has been estimated using the code METALS provided by the National Physical Laboratories UK (NPL). The procedure involves input of the measured freckle composition and is outlined in Reference 29. The temperature gradient at the Fig. 6—Typical DS samples cast at 0 and 35 deg to the vertical. liquidus, *GL* , solidification rate, *R*, at the liquidus as well as the temperature gradient at the fraction solidified for freckle initiation of 0.5, $G_{0.5}$, have been evaluated using experimental error of the determination of those particular ProCAST according to the methodology presented in the properties at high temperature. Also, it permits a comparison User's Manual. The vertical temperature gradient G_F^V is simply equal to $G_{0.5}$ cos α , where α is the growth front angle to the horizontal.

The rationale for evaluating a single proportionality con-

between the various alloys on the basis of casting condi-
tions only.

Having evaluated the Rayleigh number for all of the cast-
ing conditions examined, the results can be plotted as a stant for all of the alloys examined in the study is that their function of furnace growth front angle for the six alloys. probable variation over the alloy range chosen is within the The resulting plot is presented in Figure 8. Given the range

of variation in thermal gradients and solidification rates $h = \lambda_1 \left(\frac{K}{K_y}\right)$ [11] observed in the quasi-steady-state sections of each sample, the Rayleigh numbers can be calculated with an estimated precision of approximately ± 15 pct.^[29] As can be seen,
except for alloys RENÉ88, NIM80A, and IN718-Si, which
have only five experimental datum points for each alloy, it
is not possible to draw a line horizontally t ling occurs and below which freckling occurs. Therefore, either the basic Rayleigh number as a criterion is incorrect or there is a dependency on growth front angle.

Fig. 9—Schematic illustration of typical curved growth front found in ingot.

C. *Modified Rayleigh Number*

In the original work of Sarazin and Hellawell, $^{[15]}$ the characteristic length for channeling in Pb-Sn and Pb-Sb alloys was determined by setting the Rayleigh number equal to unity, consistent with a balance between the driving force and resistance to flow. The fact that the characteristic length, *h*, determined in this manner was roughly consistent with the primary dendrite arm spacing may have been somewhat Fig. 7—Typical appearance of freckles on the etched surface of the DS
samples.
Samples.
Samples.
Samples.
Samples.
Samples.
Samples.
Samples. been rewritten to include the permeability of the medium in which liquid convection occurs. Both approaches have their merits since physical reasoning demands that the magnitude of the Rayleigh number should be around unity if a "balance" does indeed exist. Moreover, the characteristic length should include the permeability as it appears in the denominator in the Rayleigh number, contributing to the resistance to flow. The potential role of permeability can be illustrated by considering a schematic illustration of a typical mushy zone arising in an ingot cast *via* an ESR or VAR process (refer to Figure 9). It is obvious that at the center of the ingot (left-hand side of Figure 9, vertical channeling would be expected to see a "characteristic length" consistent with flow in a direction parallel to the primary dendrites. Whereas vertical channeling occurring toward the ingot surface, on the right-hand side of Figure 9, would be expected to see a "characteristic length" consistent with flow more perpendicular to the primary dendrite trunks.

Fig. 8—Original Rayleigh number *vs* growth front angle for the experimen- Based on the previously mentioned arguments, a modified characteristic length of the form presented in Eq. $[11]$ has been proposed as

$$
h = \lambda_1 \left(\frac{K}{K_{\rm y}}\right) \tag{11}
$$

$$
K_y = 3.75 \times 10^{-4} f_L^2 \lambda_1^2 \qquad \text{for } 0.17 < f_L < 0.61 \qquad [12]
$$

$$
K_x = 3.62 \times 10^3 f_L^{3.34} \lambda_1^{0.699} \lambda_2^{2.73} \text{ for } 0.19 < f_L < 0.66 \quad [13]
$$

Fig. 10—Modified Rayleigh number *Ra vs* growth front angle for various Fig. 11—Modified Rayleigh number *Ra vs* growth front angle for various thermal gradients G_L and $G_F^V = G_L + 1000 \text{ K m}^{-1}$ (constant growth rate $R = 3.3 \times 10^{-5} \text{ ms}^{-1}$ (2 mm min⁻¹)). thermal gradients G_L and $G_F^V = G_L + 1000 \text{ K m}^{-1}$ (constant growth rate growth rates *R* (constant thermal gradients $G_L = 1500 \text{ K m}^{-1}$ and $G_F^V = 3.3 \times 10^{-5} \text{ ms}^{-1}$ (2 mm min²¹)). (2500 K m^{-1}) .

$$
K = \frac{1}{\frac{\sin^2 \alpha}{K_x} + \frac{\cos^2 \alpha}{K_y}}
$$
 [14]

$$
\lambda_2 = \frac{40 \times 10^{-6}}{(G_L R)^{0.42}} \tag{15}
$$

where λ_2 is in meters $G_L R$ is in K s⁻¹.

For the structure on the left-hand side of Figure 9, Eq. expression for the modified Rayleigh number, $\frac{1}{1}$ yields λ_1 because K will be equal to K_n, resulting in a ship may be simplified to the following form: [11] yields λ_1 because *K* will be equal to K_y , resulting in a Rayleigh number close to unity in agreement with Sarazin and Hellawell.^[15] For the structure on the right-hand side, *K* is increased reflecting the influence of the perpendicular

$$
\text{Ra} = \frac{gd\rho/dz}{\eta D_T} \left[\lambda_1 \left(\frac{K}{K_y} \right) \right]^4 \tag{16}
$$

growth rate, appear in Figure 10, and the results for different from the present experimental results are

where λ_1 and λ_2 are in meters, and K_x and K_y are in growth rates, at a constant gradient, appear in Figure 11. As square meters.
According to Scheideger,^[35] the permeability in the verti-
 10^{-4} ms⁻¹ (6 mm min⁻¹)), the Rayleigh number increases According to Scheideger,^[35] the permeability in the verti- 10^{-4} ms⁻¹ (6 mm min⁻¹)), the Rayleigh number increases cal direction is given by the expression with the solidification front angle.

D. Application of Modified Rayleigh Criterion to the *Experimental Results*

In the general case of ingot casting, the mushy zone and In Eqs. [12] and [13], λ_1 is the primary dendrite arm spacing,
 λ_2 is the secondary dendrite arm spacing, and f_L is the liquid

fraction; the rest of the parameters are defined in the nomen-

clature list at the study.

> To assess the applicability of the proposed modifications, the experimental results were reexamined, and new Rayleigh numbers have been calculated for each casting. Once the necessary numerical data has been incorporated into the expression for the modified Rayleigh number, the relation-

$$
\text{Ra} = 4.14 \times 10^{-6} \, G_F^V \, (G_L \, R)^{-1.32} \left(\frac{K}{K_y}\right)^4 \left(\pm 15 \, \text{pct}\right) \tag{17}
$$

component of permeability, which is between 2 and 4 times
greater than the parallel component at $f_L = 0.5$.
Employing the revised expression for characteristic
length, a modified Rayleigh number can, therefore, be writ-
t and *R* for each casting have been obtained from ProCAST as before. For each experimental casting, the Rayleigh number Ra has been plotted against growth front angle. The results are presented in Figures $12(a)$ through (d) together with a line delineating the freckled and freckle-free regions. To illustrate the effect of variation in growth front angle on As can be seen, in all cases, it is now possible to evaluate *Ra*, the modified Rayleigh number has been plotted against for each alloy a critical threshold value Ra* independent of growth front angle α for a variety of different solidification growth front angle, below which freckling does not occur conditions. The results for different gradients, at a constant and above which freckling occurs. Accordingly, estimates

Fig. 12—Modified Rayleigh number *vs* growth front angle for alloy:(*a*) CMSX-11B, (*b*) RENÉ88, (*c*) NIM80A, (*d*) IN718-Si, (*e*) WASPALOY, and (*f*) MAR-M247.

\n $\text{Ra}^*(\text{CMSX-11B}) \approx 0.88$ \n	\n D_T \n	\n thermal \n
\n $\text{Ra}^*(\text{RENÉ88}) \approx 0.90$ \n	\n $\Delta \rho$ \n	\n density \n
\n $\text{Ra}^*(\text{NIM80A}) \approx 0.85$ \n	\n m^{-4} \n	
\n $\text{Ra}^*(\text{IN718-Si}) \approx 0.65$ \n	\n $\text{d} \rho/dT$ \n	\n density \n
\n $\text{Ra}^*(\text{WASPALOY}) \approx 0.95$ \n	\n g \n	\n gravitation \n
\n Gos \n	\n temperature \n	

 Ra^* (MAR-M247) ≈ 0.86

It can be seen that the critical threshold values Ra^* are very similar for all alloys and are close to unity. It is to be recalled
that the analysis uses proportionality coefficients, and,
hence, the actual threshold values may be slightly different.
However, if the results of the pr However, if the results of the present study are assumed to η dynamic viscosity (kg m
be indicative of the relative propensity of each allow to form h characteristic length (m) be indicative of the relative propensity of each alloy to form h freckles then WASPAI OY appears to be less freckle-prope freckles, then WASPALOY appears to be less freckle-prone K permeability in vertical direction (m^2)
because it has the higher critical threshold value Ra^* K_0 representative permeability of mushy zone (m^2) because it has the higher critical threshold value Ra^* , K_0 representative permeability of must be whereas IN718-Si is the most prone to freckle formation K_r perpendicular permeability (m²) whereas IN718-Si is the most prone to freckle formation. K_x
 K_y

IV. SUMMARY AND CONCLUSIONS λ_1 primary dendrite arm spacing (m) secondary dendrite arm spacing (m) cles are presently one of the maior defects encoun-

Freckles are presently one of the major defects encoun- R solidification rate $(m s^{-1})$ tered in large superalloy castings. The evaluation of a numer-

ical criterion able to provide quantitative insight on the Ra^{*} critical threshold value of Rayleigh number ical criterion able to provide quantitative insight on the conditions of freckle formation is now recognized as a major key toward the successful manufacture of large diameter VAR/ESR ingots and large DS/SX castings. An experimental furnace was built to directionally solidify cylindrical castings **REFERENCES** at various angles to the vertical, thus simulating a tilted mushy zone. Analysis of the results using the conventional 1. T.M. Pollock and W.H. Murphy: *Metall. Mater. Trans. A*, 1996, vol. Rayleigh number reveals no single criterion above which 27A, pp. 1081-94.

freckling occurs and below which no freckling occurs for 2. P. Auburtin and A. Mitchell: Liquid Metal Processing and Casting Freckling occurs and below which no freckling occurs for
the alloys examined in the study, CMSX-11B, RENÉ88,
NIM80A, WASPALOY, MAR-M247, and a variation of
INT18 with high silicon content.
N718 with high silicon content.

number has been proposed. The modified number uses a
characteristic length that includes dendrite permeability and
also accounts for its dependence on growth front angle and
also accounts for its dependence on growth front on the local solidification conditions. The modified Rayleigh 7. J.C. Heinrich, S. Felicelli, and D.R. Poirier: *Num. Heat Transfer, Part* number has been used to analyze the results of the experi-
B, 1993, vol. 23, pp. 46 number has been used to analyze the results of the experi-
mental study. Using the modified number, a single threshold
value can be determined in all cases. For CMSX-11B, the
p. 277-96.
H. Combeau and G. Lesoult: in *Model* threshold value is approximately 0.88; for RENÉ88, it is *Advanced Solidification Processes VI*, T.S. Piwonka, V. Voller, and L. approximately 0.90; for NIM80A, it is approximately 0.85; Katgerman, eds., TMS, Warrendale, P approximately 0.90; for NIM80A, it is approximately 0.85; Katgerman, eds., TMS, Warrendale, PA, 1993, pp. 201-08.

for WASPAI OV it is approximately 0.95: for MAR-M247 it 10. M.C. Schneider, J.P. Gu, C. Beckermann, W.J. Bo for WASPALOY, it is approximately 0.95; for MAR-M247, it

is approximately 0.86; and for IN718-Si, it is approximately

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Overall, the results of the study clearly demonstrate the vol. 1, pp. 1209-20.

Suppose the consumption of the study clearly demonstrate the vol. 1, pp. 1209-20. influence of growth front angle on channel formation as it
was found that, under similar solidification conditions, tilted
castings exhibited freckles, whereas vertical castings were
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- K_y parallel permeability (m²)
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