#### **ORIGINAL PAPER**



# **Resistance of a Mo–Si–B‑Based Coating to Environmental Salt‑Based Hot Corrosion**

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## **Abstract**

During operation of gas turbine engines, diferent salts develop that lead to a hotcorrosion attack. While the hot corrosion associated with  $Na<sub>2</sub>SO<sub>4</sub>$  and  $Na<sub>3</sub>VO<sub>4</sub>$  has received considerable attention,  $MgSO<sub>4</sub>$  is also a main salt component, but it has received limited examination. For operation at temperatures beyond the capabilities of Ni-base superalloys, refractory-metal alloys with a Mo–Si–B-based coating have demonstrated a robust environmental resistance including hot corrosion attack by  $Na<sub>2</sub>SO<sub>4</sub>$  and  $Na<sub>3</sub>VO<sub>4</sub>$ . In order to evaluate the performance of Mo with a Mo–Si–Bbased amorphous coating, samples were exposed to  $MgSO<sub>4</sub>$  under thermal cycling at temperatures from 800 to 1300 °C. Mass change measurements revealed minimal attack from the  $MgSO<sub>4</sub>$  exposure except in local regions where cristobalite and MgMoO4 were detected. Even with the local region reaction, the mass change after a cumulative 100 h of exposure was about  $3.5 \text{ mg/cm}^2$ . Exposure to a composite salt mixture containing  $Na<sub>2</sub>SO<sub>4</sub>$ ,  $NaVO<sub>3</sub>$ ,  $MgSO<sub>4</sub>$ , CaCl, and KCl revealed signs of interactivity compared to the corrosion due to the separate efects of the individual salts, but again the mass change stabilized to a minimal level demonstrating that the Mo–Si–B-based coating is resilient to hot corrosion.

**Keywords** Hot corrosion · Magnesium sulfate · Mo–Si–B · Coating · Hightemperature materials

# **Introduction**

In the continuing effort to improve aircraft engine efficiency, it has become evident that new material systems with high-temperature capability are necessary. Turbine engine efficiency is improved with higher operating temperatures as modeled by the Brayton Cycle [[1\]](#page-11-0). This route requires new materials, such as refractory metals, that

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maintain their structural performance at high temperatures. Among these, Mo–Si–B alloys have been demonstrated to be a promising candidate [[2\]](#page-11-1). However, molybdenum and other refractory metals have poor oxidation resistance with Mo oxidizing to form low melting point and volatile oxides  $[3, 4]$  $[3, 4]$  $[3, 4]$  $[3, 4]$ . To prevent this, a Mo–Si–Bbased coating with an aluminoborosilica glass outer layer has been developed as a means to protect the substrate from oxidation [\[5](#page-11-4)]. The Mo–Si–B-based coatings are typically produced via a pack cementation method [\[6](#page-11-5)]. This process produces a multilayer system on the molybdenum or other refractory metal-based substrate [\[7](#page-11-6)].

In addition to oxygen attack, a variety of salt deposits from fuel impurities and the environment have been found in gas turbine engines [\[8](#page-11-7)]. The degradation of materials by molten salt deposits at the operational temperatures of turbines is defned as hot corrosion [[9\]](#page-11-8). The hot corrosion reaction occurs in two varieties: a lower temperature Type II and a higher temperature Type I [\[10](#page-11-9)]. Type II corrosion is typically active between 670 and 750 °C [[11\]](#page-11-10). Type I corrosion is a two-phase process that begins with a low corrosion incubation period that is related to the formation of an oxide scale. This is followed by accelerated attack as the scale breaks down and is likely caused by salt fuxing [[12\]](#page-11-11). For the most part, hot corrosion studies have focused on the attack by sodium sulfate  $(Na_2SO_4)/s$ odium metavanadate  $(NaVO_3)$  on chromia or alumina forming alloys. There has also been some study on the attack on silica forming materials such as SiC [\[13](#page-11-12)]. At the same time,  $MgSO<sub>4</sub>$  has been found to be as abundant as  $Na<sub>2</sub>SO<sub>4</sub>$  in the gas turbine environment, but there has been little attention to the attack by  $MgSO<sub>4</sub>$  or the combination of the magnesium and sodium sulfates.

A previous study that investigated the resistance of the Mo–Si–B coating to sodium sulfate (Na<sub>2</sub>SO<sub>4</sub>)/sodium metavanadate (NaVO<sub>3</sub>) reported good corrosion protection and a self-healing ability for the Mo–Si–B in this environment [\[7](#page-11-6)]. In the present study, the protective ability of the coating was investigated in the environments of magnesium sulfate  $(MgSO<sub>4</sub>)$  and a salt mixture analogous to real-world conditions. This mixture that is based upon analysis from Bornstein and Allen also contains vanadium, as another component present in fuel impurities [\[8](#page-11-7), [14](#page-11-13)] (Table [1](#page-1-0)).

The coating in this composite system would be afected by both attack from the individual salts and the results of the interactivity of the salts. Previous experiments on silica forming materials showed that sodium and magnesium salt combinations

<span id="page-1-0"></span>

**Table** 1

increased magnesium mobility while decreased sodium mobility [\[15](#page-11-14)]. The presence of sodium, magnesium, vanadium, and sulfur enables the formation of a range of vanadates, oxides, and sulfates that can both promote and inhibit hot corrosion [[14\]](#page-11-13). As a result, a composite salt mixture can behave diferently than the contribution of each individual component.

## **Experimental Procedures**

The efects of the salt systems on Mo–Si–B coatings were explored by producing a series of coated molybdenum coins that were exposed to hot corrosion environments. Pure molybdenum (99.95%) rods were cut into coins approximately 1.25 cm in diameter and .3 cm thick. The edges of these coins were rounded and polished to a 5 micron fnish.

A coating from a source of Si and B in the ratio of 35:1 silicon to boron was deposited via pack cementation. The coins were surrounded by a powder mixture and heated in an argon-filled tube furnace at 1000  $\degree$ C for 50 h. The powder mixture contained 2.5 wt% NaF salt as an activator to promote the coating formation. In the mixture, 35 wt% was the 35:1 silicon to boron powder, with the remaining 62.5 wt% being inert 180 grit alumina fller. The void space between the alumina particles provided pathways for the active components to interact with the molybdenum substrate.

After the pack cementation, the coins were conditioned in laboratory air at 1300 °C for 10 h. This conditioning served two functions. Primarily, it oxidized the surface forming the vitreous outer layer [\[16](#page-12-0)]. Secondly, it verifed the integrity and completeness of the coating. If the coating was incomplete, the molybdenum substrate would rapidly oxidize to molybdenum trioxide  $(MoO<sub>3</sub>)$  and vaporize. This failure of the coating was visually apparent and caused a signifcant mass loss allowing for easy detection of poorly coated samples.

For hot corrosion exposure, an aqueous slurry of magnesium sulfate or the composite salts was applied to the coated coins using an airbrush in sequential thin coatings in order to achieve a uniform deposit. The amount of salt applied was approximately 25 mg/cm<sup>2</sup> for the magnesium sulfate and 10 mg/cm<sup>2</sup> for the composite salt. This amount is several orders of magnitude beyond reported salt amounts known to promote corrosive efects [[17\]](#page-12-1). The composite salt slurry was a mixture of salts found in existing jet engines, along with environmental salts, and elements related to fuel impurities [\[8](#page-11-7), [18](#page-12-2)]. The fnal constituents are presented in Table [2](#page-2-0). The coated

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Ion	Na		SO <sub>4</sub>	Mg	Ca		
Approximate molar percent	38	4.5	27		7.5	10	
Salt Species	Na <sub>2</sub> SO <sub>4</sub>		NaVO <sub>3</sub>	MgSO <sub>4</sub>	CaCl		KCl
Approximate molar percent	41		20		14		

<span id="page-2-0"></span>**Table 2** Molar percentages of each ion present in composite salt slurry

specimens were dried in open air until the salt slurry densifed. Vapor phase mass loss experiments using a salt reservoir in the furnace were not attempted as it has been shown to yield signifcantly lower mass losses [\[7](#page-11-6)].

The coated coins were loaded into a tube furnace. In the furnace, the coins were exposed to repeated thermal cycling of 5 h periods at either 900 °C or 1000 °C. During the cycles, the air was either left static in the furnace or laboratory air was fowed over the coins at a rate of approximately 10 l/min. The large quantities of salt and relatively slow air speeds make the conditions tested far more favorable for hot corrosion attack than turbine operation. After each cycle, the mass of the coin was measured, and X-ray difraction was performed periodically. The salt coating was not reapplied during the cycles as Gray et al. [[19\]](#page-12-3) have shown that redeposition has little efect on corrosion for the time scales used here. Select coins were sectioned and polished for scanning electron microscopy. Before sectioning, the coins were coated with a sputtered gold and an electroplated nickel layer to preserve the glassy silica coating during the polishing according to a procedure outlined by Richter [[20\]](#page-12-4).

## **Results**

## **Coating Structure**

The pack cementation method produced a coating with several diferent phase layers covering the molybdenum substrate. The coating structure after exposure to  $MgSO<sub>4</sub>$ for 20 thermal cycles at 1000  $\degree$ C in flowing air is presented in the cross-section images in Fig. [1](#page-3-0). The initial coating structure after conditioning was not altered by the exposure to  $MgSO<sub>4</sub>$  at 1000 °C in flowing air. Moving away from the molybdenum substrate, a thin  $T_2$  layer of Mo<sub>5</sub>SiB<sub>2</sub> develops, on top of that a  $T_1$  layer of boron saturated  $Mo<sub>5</sub>Si<sub>3</sub>$ , then a layer of MoSi<sub>2</sub>, and finally there is an outer layer of aluminoborosilica that develops during conditioning from the codeposition of Si and B with the incorporation of some alumina fller [\[7](#page-11-6)]. During cooling, a few cracks



<span id="page-3-0"></span>**Fig. 1 a** Backscatter electron image and **b** secondary electron image of Mo–Si–B coating after 20 thermal cycles at 1000 °C in fowing air

develop in the MoSi<sub>2</sub> containing layer, but they are arrested by the  $T_1$  layer and do not reach the substrate. The vitreous outer layer is capable of fowing at elevated temperatures to fll the cracks and can be replenished by the lower layers in case of damage  $[21]$  $[21]$ . The  $T<sub>2</sub>$  phase serves as a diffusion barrier and a source of silicon and boron [\[7](#page-11-6)]. This enables a self-healing mechanism that further mitigates corrosive attacks.

#### **Magnesium Sulfate**

Magnesium sulfate comprises about 22% of the salt deposits found in turbines [[8\]](#page-11-7). Both MgSO<sub>4</sub> and Na<sub>2</sub>SO<sub>4</sub> have similar decomposition routes producing SO<sub>3</sub> and an oxide  $[22, 23]$  $[22, 23]$  $[22, 23]$ . It has been proposed that sodium oxide  $(Na<sub>2</sub>O)$  diffuses into the outer vitreous layer of the Mo–Si–B-based coating promoting the nucleation of cristobalite [\[7](#page-11-6)]. The resulting crystallization front advances into the silica and causes some of the coating to break off via spalling during cooling. In contrast, magnesium ions have virtually no mobility in silica [\[15](#page-11-14)]. Consequently, magnesium is restricted to forming cristobalite only on the surface and maintains a larger amount of the vitreous surface which enhances the preservation of the Mo–Si–B coating.

The mass change of the sample in the  $MgSO<sub>4</sub>$  environment after each cycle is presented in Fig. [2](#page-4-0). There is an initial gain in mass from the deposited salt coating. After this gain, the sample stabilizes and the total mass change plateaus to near the initial mass before the salt coating application. The coated coins that were in a fowing air environment showed a stable mass with little mass change. The 1000 °C cycles in the static environment showed some mass change, but the amount appears to level off at about 3.5 mg/cm<sup>2</sup> which is still minimal. The 900  $\degree$ C static salt environment did not have a net mass change as some initial amount of salt remained on



<span id="page-4-0"></span>**Fig. 2** Mass change of the coated Mo coins with a Mo–Si–B-based coating in different MgSO<sub>4</sub> environments

the surface throughout the experiment. The presence of some amount of cristobalite on these coins, as determined by XRD, indicates that some corrosion has occurred but the mass of the salt remaining overcompensates for this mass loss (Fig. [3\)](#page-5-0).

These results show a clear diference with the previous work on attack by  $Na<sub>2</sub>SO<sub>4</sub>/NaVO<sub>3</sub>$  salts where the mass change is shown in Fig. [4](#page-6-0) [[7\]](#page-11-6). In most cases, there is significantly less mass change for the exposed  $MgSO<sub>4</sub>$ -coated samples than those coated with  $Na<sub>2</sub>SO<sub>4</sub>/NaVO<sub>3</sub>$ . The mass change for both  $MgSO<sub>4</sub>$ - and  $Na<sub>2</sub>SO<sub>4</sub>$ -/ NaVO<sub>3</sub>-coated samples was similar only for the 900  $^{\circ}$ C flowing air environment, which was relatively inactive for both. The consistent reduction in mass change can be attributed to the lack of magnesium mobility in the amorphous surface layer. In static environments, the mass change during  $MgSO<sub>4</sub>$  exposure is consistently 0.1 times that of the  $Na<sub>2</sub>SO<sub>4</sub>$ -based mass change. The mass changes in flowing air environments were more temperature-dependent. At 900  $^{\circ}$ C, the MgSO<sub>4</sub> corrosion was 0.2 times that due to  $\text{Na}_2\text{SO}_4/\text{NaVO}_3$ , whereas at 1000 °C the corrosion was 0.05 times that due to  $\text{Na}_2\text{SO}_4/\text{NaVO}_3$ , as a result of increased reactivity in the  $\text{Na}_2\text{SO}_4/\text{NaVO}_3$  $NaVO<sub>3</sub>$  system at higher temperatures. Since the magnesium does not move through the amorphous surface layer advancing a crystallization front, large cristobalite regions do not form and spall off, as described in the  $\text{Na}_2\text{SO}_4/\text{NaVO}_3$  environments [\[7](#page-11-6), [15](#page-11-14)]. Additionally, a sample was tested at 1300  $^{\circ}$ C in a static environment. The mass change of this sample stabilized at about  $-3$  mg/cm<sup>2</sup>, comparable to the mass change in the 1000 °C static environment. This is in contrast to coated samples without salt deposits at 1000 °C which show no mass change during thermal cycling. This mass change could be a result of the time the sample was at lower temperatures, while the furnace was heating up and/or more aggressive corrosion above 1000 °C.

X-ray difraction was used to identify the composition and structure of the Mo–Si–B coatings after the conditioning and following the thermal cycling



<span id="page-5-0"></span>Fig. 3 Mass change data for coated Mo coins with a Mo–Si–B-based coating in both MgSO<sub>4</sub> and Na2SO4/NaVO<sub>3</sub> environments. The Na<sub>2</sub>SO<sub>4</sub>/NaVO<sub>3</sub> data are reproduced from [\[7](#page-11-6)]



<span id="page-6-0"></span>**Fig. 4** X-ray difraction data for coated samples after 20 5 h cycles at 1000 °C under a static condition **a** a typical region of Mo–Si–B coin surface after magnesium sulfate exposure and **b** regions containing  $MgMoO<sub>4</sub>$ 

exposure. The majority of the coating is the expected vitreous layer that appears as a broad amorphous maximum in the X-ray difraction scan. In addition, multiple series of peaks associated with the molybdenum silicide structures of  $T_1$  and MoSi<sub>2</sub> are detected clearly. Some areas of the coins showed peaks consistent with magnesium molybdate (MgMoO<sub>4</sub>) as shown in Fig. [4.](#page-6-0) The formation of MgMoO<sub>4</sub> from reaction of  $MgSO_4$  with  $MoO_3$  was evaluated by free energy calculations [[24,](#page-12-8) [25](#page-12-9)]. The result of this calculation is presented in Fig. [5](#page-6-1) and shows that the reaction

<span id="page-6-1"></span>

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between  $MgSO<sub>4</sub>$  and  $MoO<sub>3</sub>$  to yield  $MgMoO<sub>4</sub>$  and  $SO<sub>2</sub>$  and  $O<sub>2</sub>$  is favorable thermodynamically over the entire temperature range investigated.

The relative scarcity of  $MgMoO<sub>4</sub>$  and the lack of significant mass change implies that the presence of  $MoO<sub>3</sub>$  as a reactant with  $MgSO<sub>4</sub>$  occurs in localized regions of the coating. The SEM image in Fig. [6](#page-7-0) shows localized cracks in the silicide layers that likely formed upon cooling and were subsequently flled by fow of the amorphous surface layer during heating. These gaps are a probable source of the molybdenum that forms  $MoO<sub>3</sub>$  during heating before the glass has sufficient fluidity to cover the gap. This  $MoO<sub>3</sub>$  can then act as a reactant to form  $MgMoO<sub>4</sub>$  locally.

In order to assess the self-healing capability of the coating system, it is necessary to consider the fow velocity of the amorphous surface layer. The works of Papko and Levitskii and Saiz et al. [\[26](#page-12-10), [27\]](#page-12-11) provide the foundation to approximate the self-healing rate. Papko and Levitskii found the viscosity, *η* of a 70% silica glass at 900 °C, with other oxides including  $B_2O_3$  accounting for no more than 10% each, to be  $10^{5.2}$  Pas s [[26\]](#page-12-10). Saiz et al. found the spreading velocity, v of a silica glass on molybdenum at 1200 °C to be approximately 90 µm s−1 and related to the capillary number,  $C_a = \eta v / \gamma_{lv}$ . Using their capillary number of  $9 \times 10^{-2}$  and an interfacial tension,  $\gamma_{lv}$  of ~1 J m<sup>-1</sup> an approximate viscosity of 10<sup>5</sup> Pas s can be calculated [[27\]](#page-12-11). Given the proximity of these two viscosities, an overestimated spreading velocity of 90  $\mu$ m s<sup>-1</sup> can be used to generalize the flow speed of the glass. Using this, cracks such as A and B in Fig.  $6$  could be filled in about 10 s. At the same time, upon reheating the formation of  $MoO<sub>3</sub>$  becomes appreciable above 700 °C where the glass



<span id="page-7-0"></span>**Fig. 6** SE-SEM image of cracks in the surface coating that have been flled in by glass during thermal cycling at 1000 °C under a static condition

flow rate is slower. As a result, the interval between 700 and 1000  $\degree$ C which corresponds to about 20 min during heating is available for  $MoO<sub>3</sub>$  formation. However, due to the highly localized crack regions the impact on the mass change is small.

XRD comparisons between the coated samples exposed in the  $Na_2SO_4/NaVO_3$ environment and  $MgSO<sub>4</sub>$  environment are presented in Fig. [7](#page-8-0) [\[7](#page-11-6)]. Both coatings showed the expected peaks corresponding to molybdenum silicide phases such as  $MoSi<sub>2</sub>$  and  $MoSi<sub>3</sub>$ . However, the cristobalite peaks are more pronounced in the coating with  $Na<sub>3</sub>SO<sub>4</sub>/NaVO<sub>3</sub>$  deposit. This coupled with a reduced relative amorphous signal indicates some crystallization of the amorphous coating. These peak diferences are in agreement with the expected scarcity of cristobalite nucleation due to the immobility of magnesium in the amorphous surface layer.

In Fig. [8](#page-9-0) X-ray diffraction results are presented for samples exposed in a  $MgSO<sub>4</sub>$ environment at 1000 °C under a static condition for 20 5 h cycles. For a typical surface region, Fig. [8a](#page-9-0) reveals small peaks for cristobalite and the expected molybdenum silicides. Figure [8](#page-9-0)b from isolated regions shows peaks corresponding to cristobalite, molybdenum silicides, and magnesium molybdate. These regions are accompanied by a negligible amorphous signal associated with glass surface. This



<span id="page-8-0"></span>**Fig. 7** X-ray difraction information of typical surfaces in **a** magnesium sulfate environment and **b**  $Na<sub>2</sub>SO<sub>4</sub>/NaVO<sub>3</sub>$  environment [\[7](#page-11-6)]



<span id="page-9-0"></span>**Fig. 8** X-ray diffraction patterns of coating surface after exposure to MgSO<sub>4</sub> at 1000 °C for 20 5 h cycles under a static condition, **a** a typical surface region and **b** a surface region containing magnesium molybdate and cristobalite

is likely due to the immobile molten salt causing crystallization on the surface. The structural change and spallation from this transformation are likely the cause of the mass loss in this situation via the same mechanism described earlier [[7\]](#page-11-6). However, a more detailed examination into spallation and cristobalite formation mechanisms is beyond the scope of this paper.

#### **Composite Salt**

The furnace cycles of the composite salt mixture were only performed in fowing laboratory air. The mass change stabilized after 6–8 iterations of 5 h cycles. The composite salt samples showed little variation in mass change at 900 °C and 1000  $\degree$ C as indicated in Fig. [9](#page-10-0). The samples cycled at 800  $\degree$ C exhibited lower mass change (less than 1  $mg/cm<sup>2</sup>$ ) compared to the higher temperature exposures. The extent of the corrosion at 900 °C is greater than the separate  $MgSO_4$  and  $Na<sub>2</sub>SO<sub>4</sub>/$  $NaVO<sub>3</sub>$  cycles indicating some increased activity from the interaction of the salts and the coating. At 900 °C, the salt composite exhibited twice the mass change as in the Na<sub>2</sub>SO<sub>4</sub>/NaVO<sub>3</sub> environment. At 1000 °C, the composite only had 0.25 the mass change as the  $\text{Na}_2\text{SO}_4/\text{NaVO}_3$  mixture. The cause of this increased corrosion at a lower temperature will be addressed in future study. A series of extended duration cycles of 20 h each were performed on the 900  $^{\circ}$ C sample as it showed the most corrosion. After a total of 150 h, no further mass change occurred beyond the 2 mg/  $\text{cm}^2$  that resulted from 10 five-hour cycles. The lack of significant change in activity



<span id="page-10-0"></span>**Fig. 9** Mass loss of composite salt mixture compared to  $Na_2SO_4/NaVO_3$  and  $MgSO_4$  salt environments

at 1000 °C is in agreement with previous research that the presence of magnesium in corrosive systems containing sodium and vanadium suppresses the corrosive activ-ity [\[14](#page-11-13)]. This is contrasted by the results of the  $\text{Na}_2\text{SO}_4/\text{NaVO}_3$  system that had a clear increase in corrosion at higher temperatures.

X-ray difraction of the composite salt surface showed the same regions of molybdenum silicides and a distinct cristobalite peak as illustrated in Fig. [10.](#page-10-1) This indicates that the salt mixture, likely due to the large  $Na<sub>2</sub>SO<sub>4</sub>$  portion, causes crystallization on the surface. However, the lower amount of mass change than that with the 1000 °C Na<sub>2</sub>SO<sub>4</sub> salt indicates that less spalling is occurring in this environment than in the environment with only  $Na<sub>2</sub>SO<sub>4</sub>$ . Some of the peaks potentially correspond to vanadium compounds such as vanadium sulfde (VS) and vanadium silicides ( $V_3Si/V_5Si_3$ ), which suggests that there may be an intermixing between



<span id="page-10-1"></span>**Fig. 10** Typical XRD pattern of the coating surface after exposure to the composite salt system after repeated thermal cycling at 900 °C

molybdenum and vanadium. Additionally, some peaks were observed corresponding to  $MgMoO<sub>4</sub>$ . Clear identification of components through X-ray diffraction was complicated by the number of phases present in the system, each with a variety of possible products.

## **Conclusions**

Magnesium sulfate in isolation has little detrimental efect on Mo–Si–B coatings for refractory metals in normal operating conditions. Unlike sodium sulfate, there is little crystallization of the outer vitreous silica layer. Examinations of the mass changes during repeated thermal cycles and changes in the X-ray difraction of the surface support this. It is noted that magnesium molybdate does form on the surface of some coatings in localized regions as a result of the reaction of the magnesium sulfate and molybdenum oxide.

Repeated thermal cycling of samples with the composite salt mixture of  $Na<sub>2</sub>SO<sub>4</sub>$ ,  $NaVO<sub>3</sub>, MgSO<sub>4</sub>, CaCl, KCl$  demonstrated a consistent level of impact on the coating. Following an initial period of corrosion and mass change during exposure to  $MgSO<sub>4</sub>$ , the amorphous coating stabilizes. This mass change remained at about or less than  $3.5 \text{ mg/cm}^2$  for all tested environments. After investigating the composite salt mixture, isolated sodium salts remain the most deleterious to the coating, but did not induce failures exposing the molybdenum core to oxygen. Against the range of salts and impurities that occur in turbine engines under normal operations, the amorphous aluminoborosilica coating derived from the Mo–Si–B system demonstrates signifcant and reliable resilience.

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