ORIGINAL PAPER

Nonlinear stability of triangular equilibrium points in non-resonance case with perturbations

Saleem Yousuf · **Ram Kishor**

Received: 1 July 2022 / Accepted: 18 October 2023 / Published online: 20 December 2023 © The Author(s), under exclusive licence to Springer Nature B.V. 2023

Abstract The present study deals with the normalisation of Hamiltonian for the nonlinear stability analysis in non-resonance case of the triangular equilibrium points in the perturbed restricted three-body problem with perturbation factors as radiation pressure due to first oblate-radiating primary, albedo from second oblate primary, oblateness and a disc. The problem is formulated with these perturbations and Hamiltonian of the problem is normalised up to fourth order by Lie transform technique consequently a Birkhoff's normal form of the Hamiltonian is obtained. The Arnold– Moser theorem is verified for the nonlinear stability test of the triangular equilibrium points in non-resonance case with the assumed perturbations. It is found that in the presence of radiation pressure, stability range expanded, significantly with respect to the classical range of stability; however, because of albedo, oblateness and the disc, it contracted gradually. Moreover, it is observed that alike to the classical problem, in the perturbed problem under the impact of the assumed perturbations, there always exist one or more values of the mass ratio μ within the stability range at which discriminant $D_4 = 0$, which means the triangular equilibrium points are unstable in nonlinear sense.

S. Yousuf \cdot R. Kishor (\boxtimes)

S. Yousuf e-mail: sf07bhu@gmail.com **Keywords** Radiation pressure · Albedo · Oblateness · Disc · Nonlinear stability · Triangular equilibrium points

1 Introduction

The restricted three-body problem (RTBP) has been playing a significant role in dealing the dynamics of restricted mass for many space operations. It has been providing appropriate locations for placing several artificial satellites and helping to execute many space exploration missions beyond the solar system too. The RTBP consists of motion of infinitesimal mass under the gravitational influence of two massive bodies, which are orbiting about their centre of masses in circular path [\[1](#page-15-0)]. Several researchers have described the RTBP and its generalised form with or without perturbations in the form of radiation pressure, oblateness, P-R drag, disc, etc., in the context of existence of equilibrium points and their linear stability, basins of attraction associated to attractors as equilibrium points, Lyapunov exponents, periodic orbits and their families in the vicinity of equilibrium points $[2-14]$ $[2-14]$. The equilibrium points in the RTBP are being used for many space exploration missions from different space agencies. The location of equilibrium points in the planetary system would be an ideal location in future for space colonies apart from the favourite places for other space exploration missions. Therefore, the study of stability property of the linearly stable equilibrium points, i.e.

Department of Mathematics, Central University of Rajasthan, Kishangarh, Ajmer, Rajasthan 305817, India e-mail: kishor.ram888@gmail.com

triangular equilibrium points, is a necessary step as it describes the dynamics of small space objects such as Trojan asteroids for long-time duration and many other real-world applications. The stability of the equilibrium points in the restricted three-body problem over a long period of evolution is an important and critical issue, and as a result, many researchers of different fields, including mathematical physics, dynamical astronomy, celestial mechanics, etc., are studying the Hamiltonian system of the problem in the vicinity of the triangular equilibrium points.

In 1969, Deprit [\[15\]](#page-15-3) has presented the concept of Lie series to the case, in which the generating function is explicitly dependent on the small parameter. For the small parameter, Lie transformations naturally define a class of canonical mappings in the form of power series. They have demonstrated, how the canonical transformations contemplated by Von Zeipel's method, which are naturally defined. The canonical mappings obtained by using Lie transform comprise the natural ingredient for the Hamiltonian dynamical systems. Liu [\[16\]](#page-15-4) has studied the application of Birkhoff's normal form and singular perturbation in the stability of *L*⁴ point in the RTBP. In 1986, Meyer and Schmidt [\[17\]](#page-15-5) have computed the normal form of Hamiltonian up to six order and have applied the KAM theory for analysing the stability of triangular equilibrium points. Coppola and Rand [\[18](#page-15-6)] have used Lie transform method and the method of perturbation to a Hamiltonian systems with the help of computer algebra and have introduced a formula for transforming the normalised Hamiltonian to Birkhoff's form for describing nonlinear stability criteria. They have implemented these criteria for performing the nonlinear stability analysis of *L*⁴ point. Rao and Sharma [\[19\]](#page-15-7) and Ishwar [\[20\]](#page-15-8) individually, have investigated the stability of triangular equilibrium points in nonlinear sense, when one of the primaries is oblate spheroid and they have found that *L*⁴ point is stable for all μ within the stability range except at the critical value of the mass ratio.

Jorba and Villanueva [\[21](#page-15-9),[22](#page-15-10)] have initiated to execute efficient computation of Hamiltonian normal forms, first integrals of these systems, invariant tori and centre manifolds for understanding the dynamics of infinitesimal mass near the equilibrium points. This study is based on algebraic manipulation of formal series that have numerical coefficients, which is equipped with a very efficient software implementation. Further, Kushvah et al. [\[23,](#page-15-11)[24\]](#page-15-12) have studied the effect of PR-drag on the nonlinear stability of triangular equilibrium points in the photo-gravitational RTBP. They have computed the higher-order normal forms of Hamiltonian system associated to the problem and found that triangular equilibrium points are stable within the stability range except those values of μ for which KAM theory does not holds. Alvarez-Ramírez et al. [\[25](#page-15-13)] have obtained fourth-order normal form of the Hamiltonian in the presence of radiation pressure and have applied Arnold–Moser theorem to examine the nonlinear stability of *L*⁴ point. In the literature, several researchers [\[26](#page-15-14)[–32\]](#page-16-0) have studied either RTBP or its generalised form with different kinds of perturbations for nonlinear stability of equilibrium points and have analysed the effect of perturbations in the context of the nature of the stability property. Further, Zepeda Ramírez et al. [\[33,](#page-16-1)[34\]](#page-16-2) have presented the nonlinear stability analysis of equilibrium points in the planar equilateral restricted mass-unequal four-body problem. The RTBP has received very little attention in case of combined effect of several perturbations in the form of radiation pressure due to first primary, albedo of second primary, oblateness of both the primaries and presence of a disc.

The present study is focused on the nonlinear stability test of the triangular equilibrium points *L*4,⁵ by the use of Arnold–Moser theorem in non-resonance case under the influence of radiation pressure, albedo, oblateness and the disc. The paper is divided into five sections, which are as follows: Sect. [2](#page-1-0) deals with the formulation of the problem and determination of the triangular equilibrium points. Section [3](#page-4-0) presents the computation of Hamiltonian and estimation of its normalised form. Section [4](#page-7-0) is devoted to the higher-order normalisation of the Hamiltonian with the help of Lie series method and application of Arnold–Moser theorem. Section [5](#page-14-0) concludes the paper. The semi-analytical and numerical computations throughout the paper are performed by the use of latest version of Mathematica software. An extra care has been given to lengthy intermediate calculations and manipulations. For numerical computation, a fixed accuracy goal up to 7 places of decimal is used.

2 Formulation of the problem

Consider the motion of an infinitesimal mass *m*∗ (such as small space objects, asteroid, spacecraft, satellite,

etc.) under the gravitational influence of two oblate masses m_1 and m_2 (known as primaries) such that $m^* \ll m_2 \ll m_1$ and a disc of dusts in space, which is circumscribed the primaries and is rotating about the common centre of mass of the system. Suppose, primaries m_1 and m_2 are orbiting about their centre of masses in a circular path and gravitational influence of the infinitesimal mass on the remaining system is negligible. It is also assumed that first oblate primary m_1 is a radiating body and second oblate primary m_2 is such that it reflects the incident radiations from the radiating oblate primary *m*¹ and hence, second primary m_2 produces albedo effect. Suppose, F_r and F_a be the radiation pressure force due to first radiating primary and albedo of second primary then mass reduction factor *q* (also called radiation parameter) and albedo *QA* are, respectively, defined $[8,35]$ $[8,35]$ $[8,35]$ as

$$
q = \left(1 - \frac{F_r}{F_{g_1}}\right) \quad \text{and} \quad Q_A = \left(1 - \frac{F_a}{F_{g_2}}\right), \tag{1}
$$

where F_{g_1} and F_{g_2} are gravitational forces of the respective primaries. Let *A*²¹ and *A*²² be the oblateness coefficients of first and second primaries, which are, respectively, defined [\[5](#page-15-16)[,36](#page-16-4)] as

$$
A_{21} = \frac{R_{e_1}^2 - R_{p_1}^2}{5R^2} \quad \text{and} \quad A_{22} = \frac{R_{e_2}^2 - R_{p_2}^2}{5R^2},\tag{2}
$$

where R_{e_i} and R_{p_i} for $i = 1, 2$ are equatorial and polar radii of the primaries and R is the distance between both the primaries. Since it is assumed that a circular disc of dust in space circumscribed the system, then the gravitational force exerted due to the disc with mass *Md* on the infinitesimal mass defines a potential, which is expressed [\[37](#page-16-5),[38\]](#page-16-6) as

$$
\phi(r, 0) = \frac{M_d}{\sqrt{r^2 + T^2}} \quad \text{where} \quad r = \sqrt{X^2 + Y^2} \quad \text{and}
$$

$$
T = (a + b), \tag{3}
$$

with *a* and *b* as flatness and core parameters, which determine the density profile of the disc and *r* is the radial distance of the disc.

For non-dimensionalising the parameters, let the characteristic mass be defined as the sum of masses of the primaries, the characteristic length be taken as the sum of distance between the primaries and the characteristic time be assumed as the orbital period of the primaries in rotating frame. However, authors are interested to analyse the nature of motion in orbital plane, i.e. in *XY* -plane, only. Therefore, the motion along *Z*direction is uncoupled and behaves like a harmonic

oscillator with frequency 1, which is already in real normal from. Hence, formulation of the problem in hand and whole analysis are performed in *XY* -plane, in which the frame of reference of the model is described as follows: the barycentre is placed at origin of the frame, the line joining the primaries defines the *X*-axis; however, the *Y* -axis is perpendicular to the *X*-axis. Let $(-\mu, 0)$ and $(1 - \mu, 0)$ be the coordinates of the first and second primary, respectively, and (*X*, *Y*) be that of infinitesimal mass in the *XY*-plane, where $\mu = \frac{m_2}{m_1 + m_2}$ is the mass ratio. Then, the equations governing the motion of infinitesimal mass in the orbital plane under the synodic frame are written $[8,39]$ $[8,39]$ $[8,39]$ as

$$
\ddot{X} - 2n\dot{Y} = \frac{\partial \Omega}{\partial X},\tag{4}
$$

$$
\ddot{Y} + 2n\dot{X} = \frac{\partial \Omega}{\partial Y},\tag{5}
$$

where effective potential Ω is expressed as

$$
\Omega(X, Y) = \frac{n^2}{2}(X^2 + Y^2) + \frac{q(1 - \mu)}{r_1} \left(1 + \frac{A_{21}}{2r_1^2}\right) + \frac{\mu Q_A}{r_2} \left(1 + \frac{A_{22}}{2r_2^2}\right) + \frac{M_d}{\sqrt{r^2 + T^2}}.
$$
\n(6)

In Eq. [\(6\)](#page-2-0), $r_1 = \sqrt{(X + \mu)^2 + Y^2}$, $r_2 = \sqrt{(X + \mu - 1)^2 + Y^2}$ and $r = \sqrt{X^2 + Y^2}$ are the distances of the infinitesimal mass from first and second primaries and from the origin, respectively, and *n* is the perturbed mean motion, which is given $[8,40]$ $[8,40]$ $[8,40]$ as

$$
n = \sqrt{1 + \frac{3(A_{21} + A_{22})}{2} + \frac{2M_d r_c}{(r_c^2 + T^2)^{3/2}}},
$$
(7)

where r_c is the radial distance of the test body in the classical RTBP [\[41](#page-16-9)]. The Hamiltonian function of the problem in hand is expressed [\[4](#page-15-17)] as

$$
H(X, Y, P_X, P_Y) = \frac{P_X^2 + P_Y^2}{2} + n(YP_X - XP_Y) - \Omega(X, Y) + \frac{n^2}{2}(X^2 + Y^2),
$$
\n(8)

where $P_X = \dot{X} - nY$ and $P_Y = \dot{Y} + nX$ are momentum coordinates.

Fig. 1 Semi-analytical solutions relative to **a** linear order terms of perturbation parameters, **b** higher-order terms of perturbation parameters and **c** combination of (**a**) and (**b**)

2.1 Triangular equilibrium points

In a dynamical system, the point at which velocity of the infinitesimal mass vanishes, is called an equilibrium point. There are two types of equilibrium points, namely, collinear equilibrium points and triangular equilibrium points. In circular RTBP, collinear equilibrium points lie on the line joining both the primaries and are unstable in nature for all values of mass ratio $\mu \in (0, 0.5]$, hence unimportant for nonlinear stability. On the other hand, triangular equilibrium points lie in the orbital plane except the collinear axis and constitute equilateral triangle with the primaries. Triangular equilibrium points are denoted by L_4 and L_5 and these are stable in the range $0 < \mu < 0.0385209$. These are obtained by solving $\frac{\partial \Omega}{\partial X} = 0$ and $\frac{\partial \Omega}{\partial Y} = 0$, simultaneously for *X* and *Y*. In the present study, only linear order terms of the perturbing parameters are included, whereas second- and higher-order terms are ignored to reduce the complexity in the semi-analytical computations for finding the normal forms. For the numerical treatment, following fixed values as $T = 0.11$ and $r_c = 0.90$ are used so that mean motion *n* given by Eq. [\(7\)](#page-2-1) changes its form as

$$
n^2 = 1 + \frac{3(A_{21} + A_{22})}{2} + \frac{3M_d}{2}.
$$
 (9)

Now, to avoid the lengthy and complex computations with the available resources, only linear order terms of the perturbation parameters, throughout the study, are considered. However, all the perturbation parameters are very small quantities; hence, their contributions for higher-order terms are negligible. That is nature of motion in both the cases is very similar. In Fig. [1,](#page-3-0) semi-analytical solutions of the linearised equations of motion [\(22\)](#page-5-0) are displayed, in which red trajectory (Fig. [1a](#page-3-0)) corresponds to the linear order terms of perturbation parameters in the coefficients a_5 , a_6 , a_7 , which are expressed by Eqs. $(18-20)$ and blue-dashed trajectory (Fig. [1b](#page-3-0)) corresponds to that of higher-order terms of perturbation parameters in the coefficients a_5 , a_6 , a_7 , i.e. nonlinearised actual expression of a_5 , a_6 , a_7 , whereas Fig. [1c](#page-3-0) represents combination of Fig. [1a](#page-3-0), b. From Fig. [1c](#page-3-0), it can be seen that the solution with higher-order terms of perturbations

seems coincident with that of linear order terms. Therefore, the higher-order terms of the perturbations are neglected and for further analysis and discussion, only linear order terms of perturbation parameters are considered, so that complexity in the intermediary expressions can be minimised.

As, *q* and Q_A range over (0, 1], so taking $q = 1 - \epsilon_1$ and $Q_A = 1 - \epsilon_2$ with $0 < \epsilon_1, \epsilon_2 < 1$, the coordinates $(X_0, \pm Y_0)$ of the triangular equilibrium points *L*4,⁵ are expressed in linear order terms of perturbing parameters as

$$
X_0 = \frac{1}{2} - \mu - \frac{2}{3} (\epsilon_1 - \epsilon_2) + (A_{21} + A_{22}),
$$
(10)

$$
Y_0 = \frac{\sqrt{3}}{2} \left(1 - \frac{2}{9} (\epsilon_1 + \epsilon_2) - \frac{1}{3} (A_{21} + A_{22}) - \frac{2}{3} M_d \right).
$$
(11)

As, the points $L_{4,5}(X_0, \pm Y_0)$ are symmetric in nature with respect to the collinear axis, so it is enough to analyse only $L_4(X_0, Y_0)$ point and analysis for $L_5(X_0, -Y_0)$ point will be followed. Now, for the nonlinear stability test of *L*⁴ point, we determine the linear order normal form of the Hamiltonian of the system in the vicinity of *L*⁴ point and record the corresponding symplectic changes in the state variables. Further, these normal forms are utilised to obtain higher-order normal forms of the Hamiltonian to predict the nature of *L*⁴ point as summarised in [\[22\]](#page-15-10).

3 Linear order normal form of Hamiltonian

For computation of linear order normal form, we start from translation of the origin to the triangular equilibrium point $L_4(X_0, Y_0)$ by the use of symplectic change in the coordinates, which are given as

$$
x = X - X_0, \t y = Y - Y_0, \t(12)
$$

$$
p_x = P_X + Y_0, \t\t p_y = P_Y - X_0. \t\t(13)
$$

The changes in the coordinates $(12-13)$ $(12-13)$ transform the effective potential Ω given by Eq. [\(6\)](#page-2-0) to Ω , which is given as

$$
\tilde{\Omega} = \left[(x + X_0)^2 + (y + Y_0)^2 \right] + \frac{q(1 - \mu)}{\tilde{r}_1} \left(1 + \frac{A_{21}}{2 \tilde{r}_1^2} \right)
$$

$$
+ \frac{\mu Q_A}{\tilde{r}_2} \left(1 + \frac{A_{22}}{2 \tilde{r}_2^2} \right) + \frac{M_d}{\sqrt{\tilde{r}^2 + T^2}},
$$
(14)

where

$$
\tilde{r}_1 = \sqrt{(x + X_0 + \mu)^2 + (y + Y_0)^2},
$$

\n
$$
\tilde{r}_2 = \sqrt{(x + X_0 + \mu - 1)^2 + (y + Y_0)^2}
$$
 and
\n
$$
\tilde{r} = \sqrt{(x + X_0)^2 + (y + Y_0)^2}.
$$

and the Hamiltonian function [\(8\)](#page-2-2) takes the form

$$
\tilde{H}(x, y, p_x, p_y) = \frac{(p_x - Y_0)^2 + (p_y + X_0)^2}{2}
$$

+ $n[(y + Y_0)(p_x - Y_0)$
 $-(x + X_0)(p_y + X_0)]$
 $-\tilde{\Omega} + \frac{n^2}{2} [(x + X_0)^2 + (y + Y_0)^2].$ \n(15)

By the use of Taylor's series expansion, Hamiltonian \hat{H} is expanded about the origin (i.e. about L_4 point) as

$$
H = H_0 + H_1 + H_2 + H_3 + H_4
$$

+H₅ + ··· + H_n + ··· , (16)

where H_n in the expansion [\(16\)](#page-4-4) is given as

$$
H_n = \sum_{j+k+l+m=n} H_{jklm} x^j y^k p_x^l p_y^m.
$$

Since the terms H_0 and H_1 in [\(16\)](#page-4-4) do not affect the dynamics of the infinitesimal mass so, we take H_2 as starting term for higher-order normalisation of Hamiltonian about *L*⁴ point, which is expressed as

$$
H_2 = \frac{p_x^2 + p_y^2}{2} + n(yp_x - xp_y) + a_5x^2 + a_6y^2 + a_7xy,
$$
 (17)

where the coefficients a_5 , a_6 and a_7 are

$$
a_5 = \frac{1}{8} \left(1 + \frac{11}{2} \epsilon_1 - \frac{15}{2} \epsilon_2 - \frac{51}{4} A_{21} + \frac{39}{4} A_{22} - \frac{10}{21} M_d \right),\tag{18}
$$

$$
a_6 = -\frac{5}{8} \left(1 + \frac{3}{2} \epsilon_1 - \frac{19}{10} \epsilon_2 + \frac{9}{20} A_{21} + \frac{15}{4} A_{22} + \frac{15}{8} M_d \right),
$$
\n
$$
a_7 = -\gamma \left(1 - \frac{1}{18} \epsilon_1 + \frac{5}{18} \epsilon_2 + \frac{35}{12} A_{21} + \frac{17}{12} A_{22} + \frac{14}{5} M_d \right)
$$
\n
$$
(19)
$$
\n(20)

with $\gamma = \frac{3\sqrt{3}}{4}(1 - 2\mu)$.

Now, Hamiltonian H_2 in Eq. [\(17\)](#page-4-5) can be expressed as a sum of several parts corresponding to different kind of perturbations such as

$$
H_2 = H_{2c} + H_{2\epsilon_1}\epsilon_1 + H_{2\epsilon_2}\epsilon_2 + H_{2A_{21}}A_{21} + H_{2A_{22}}A_{22} + H_{2M_d}M_d, \tag{21}
$$

where

$$
H_{2c} = \frac{p_x^2 + p_y^2}{2} + n(yp_x - xp_y) + \left(\frac{1}{8}x^2 - \frac{5}{8}y^2 - \gamma xy\right),
$$

\n
$$
H_{2\epsilon_1} = \left(\frac{11}{16}x^2 - \frac{15}{16}y^2 + \frac{\gamma}{18}xy\right),
$$

\n
$$
H_{2\epsilon_2} = \left(-\frac{15}{2}x^2 + \frac{19}{16}y^2 - \frac{5\gamma}{18}xy\right),
$$

\n
$$
H_{2A_{21}} = \left(-\frac{51}{32}x^2 - \frac{9}{32}y^2 - \frac{35\gamma}{12}xy\right),
$$

\n
$$
H_{2A_{22}} = \left(\frac{39}{32}x^2 - \frac{75}{32}y^2 - \frac{17\gamma}{12}xy\right),
$$

\n
$$
H_{2M_d} = \left(-\frac{10}{168}x^2 - \frac{75}{64}y^2 - \frac{14\gamma}{5}xy\right)
$$

are the parts of Hamiltonian H_2 corresponding to classical term, radiation parameter, albedo, oblateness of first primary, oblateness of second primary and the disc, respectively. Now, consider the Jordan matrix

$$
J = \begin{pmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ -1 & 0 & 0 & 0 \\ 0 & -1 & 0 & 0 \end{pmatrix}
$$
 and write it as

$$
J = \begin{pmatrix} 0 & I_2 \\ -I_2 & 0 \end{pmatrix}
$$
, where $I_2 = \begin{pmatrix} 1 & 0 \\ 0 & 1 \end{pmatrix}$.

Then, the linearised equations of motion for the transformed Hamiltonian function H_2 in Eq. [\(17\)](#page-4-5) are given as

$$
\begin{pmatrix}\n\dot{x} \\
\dot{y} \\
\dot{p}_x \\
\dot{p}_y\n\end{pmatrix} = J \cdot Hess(H_2) \begin{pmatrix}\nx \\
y \\
p_x \\
p_y\n\end{pmatrix}
$$
\n
$$
= \begin{pmatrix}\n0 & n & 1 & 0 \\
-n & 0 & 0 & 1 \\
-2a_5 & -a_7 & 0 & n \\
-a_7 & -2a_6 & -n & 0\n\end{pmatrix} \begin{pmatrix}\nx \\
y \\
p_x \\
p_y\n\end{pmatrix}.
$$
 (22)

Suppose, $\mathbf{M} = J \cdot Hess(H_2)$, then the characteristic equation corresponding to matrix **M** is written as

$$
\lambda^4 + P \lambda^2 + Q = 0,\tag{23}
$$

where

$$
P = 2a_5 + 2a_6 + 2n^2,\tag{24}
$$

$$
Q = n4 + 2a5n2 - 2a6n2 - a72 + 4a5a6.
$$
 (25)

The characteristic equation (23) concludes that the sys-tem [\(22\)](#page-5-0) is stable for $\mu \in (0, \mu_R]$ and unstable for $\mu \in (\mu_R, 0.5]$ [\[1](#page-15-0),[39\]](#page-16-7), where μ_R is called Routh's value of mass ratio of the problem in hand (see Table [1\)](#page-6-0). From Table [1,](#page-6-0) it is clear that Routh's value of mass ratio μ_R increasing rapidly with respect to radiation parameter ϵ_1 from classical value 0.0385209 to onwards within the upper limit of μ . However, it decreases almost in the same trends with respect to the values of albedo parameter ϵ_2 . Moreover, it decreases gradually from $\mu_R = 0.0385209$ towards lower limit of μ relative to oblateness of the primaries A_{2i} , $i = 1, 2$ and that of disc parameter M_d . Thus, it is concluded that in the presence of radiation pressure, the stability range is expanded, whereas it is contracted because of albedo of second primary, oblateness of the primaries and the disc. Since the nonlinear stability test is to be performed within the stable range $\mu \in (0, \mu_R]$ of the triangular equilibrium points so, it is assumed that all the solutions of the characteristic equation [\(23\)](#page-5-1) are purely imaginary, i.e. $\lambda_{1,2,3,4} = \pm i \omega_{1,2}, \omega_{1,2} \in \mathbb{R}$.

Now, we require a real symplectic change of variables which transforms the Hamiltonian function *H*² to a real normal form. For this, we first determine the eigenvector of matrix **M**. The matrix $E_{\lambda} = \mathbf{M} - \lambda I$ (where *I* is identity matrix of the order of **M**) associated to the eigenvalue λ can be written as

$$
E_{\lambda} = \begin{pmatrix} A & I_2 \\ B & A \end{pmatrix}, \text{ where } A = \begin{pmatrix} \lambda & n \\ -n & \lambda \end{pmatrix} \text{ and}
$$

$$
B = \begin{pmatrix} -2 a_5 & -a_7 \\ -a_7 & -2 a_6 \end{pmatrix}.
$$
 (26)

Thus, the eigenvectors are obtained by solving the system of equations

$$
\begin{pmatrix} A & I_2 \ B & A \end{pmatrix} \begin{pmatrix} w_1 \ w_2 \end{pmatrix} = \begin{pmatrix} 0 \ 0 \end{pmatrix} \text{ with}
$$

$$
w_1 = \begin{pmatrix} x \ y \end{pmatrix} \text{ and } w_2 = \begin{pmatrix} p_x \ p_y \end{pmatrix}.
$$
 (27)

On simplification, we obtain two sets for *x* and *y* as

$$
x = a_7 - 2n\lambda
$$
, $y = -(2a_5 - n^2 + \lambda^2)$ and (28)

$$
x = 2a_6 + n^2 + \lambda^2, \ \ y = -(a_7 + 2\lambda). \tag{29}
$$

For further analysis, one can take any one of these two sets of x and y . Using (29) , in Eq. (27) , we get corresponding momentum coordinates as

$$
p_x = na_7 + 2a_6\lambda + n^2\lambda + \lambda^3,
$$

Table 1 Routh's value of mass ratio μ_R with respect to perturbation parameters

ϵ_1	ϵ_2	A_{21}	A_{22}	M_d	μ_R
$\mathbf{0}$	Ω	Ω	Ω	Ω	0.0385209
0.04	Ω	$\overline{0}$	Ω	Ω	0.0478447
0.08	Ω	Ω	Ω	Ω	0.0573648
0.12	Ω	Ω	Ω	Ω	0.0670942
0.12	0.02	$\overline{0}$	θ	Ω	0.0596718
0.12	0.06	Ω	Ω	Ω	0.0451903
0.12	0.08	$\overline{0}$	θ	Ω	0.0381198
$\mathbf{0}$	$\mathbf{0}$	0.002	θ	Ω	0.0346414
$\boldsymbol{0}$	Ω	0.003	Ω	Ω	0.0327137
$\boldsymbol{0}$	Ω	0.004	Ω	Ω	0.0307939
$\boldsymbol{0}$	Ω	$\overline{0}$	0.0001	Ω	0.0384916
$\boldsymbol{0}$	Ω	Ω	0.0002	Ω	0.0384624
$\boldsymbol{0}$	Ω	Ω	0.0003	Ω	0.0384331
$\boldsymbol{0}$	Ω	Ω	$\mathbf{0}$	0.0005	0.0377790
$\boldsymbol{0}$	Ω	Ω	Ω	0.0010	0.0370382
θ	Ω	Ω	$\mathbf{0}$	0.0020	0.0355602
0.02	0.01	0.003	0.0003	0.0020	0.0307876

$$
p_y = 2na_6 - n^3 - a_7\lambda - n\lambda^2. \tag{30}
$$

Thus, the resulting eigenvector is written as

$$
\begin{pmatrix} x \ y \ p_x \ p_y \end{pmatrix} = \begin{pmatrix} 2a_6 - n^2 + \lambda^2 \\ -a_7 - 2n\lambda \\ na_7 + 2a_6\lambda + n^2\lambda + \lambda^3 \\ 2na_6 - n^3 - a_7\lambda - n\lambda^2 \end{pmatrix}.
$$
 (31)

However, eigenvalue $\lambda = i\omega$ is the solution of charac-teristic equation [\(23\)](#page-5-1); hence, the frequencies $\omega_{1,2}$ can be obtained from the equation

$$
\omega^4 - P\omega^2 + Q = 0,\tag{32}
$$

If ω_1^2 and ω_2^2 are roots of the quadratic equation [\(32\)](#page-6-1) in ω^2 , then we have the relations

$$
\omega_1^2 + \omega_2^2 = P = -2a_5 - 2a_6 - n^2,
$$
\n
$$
\omega_1^2 \omega_2^2 = Q = n^4 + 2a_5n^2 - 2a_6n^2 - a_7^2 + 4a_5a_6.
$$
\n(34)

From Eq. (32) , the frequencies are expressed as

$$
\omega_1 = \pm \sqrt{\frac{P + \sqrt{P^2 - 4Q}}{2}}, \ \omega_2 = \pm \sqrt{\frac{P - \sqrt{P^2 - 4Q}}{2}}.
$$
\n(35)

where the coefficients *P* and *Q* are written in linear order terms of the perturbation parameters as

$$
P = 1 - \frac{1}{2}\epsilon_1 + \frac{1}{2}\epsilon_2 - \frac{13}{12}A_{21} + \frac{5}{12}A_{22} + \frac{26}{33}M_d,
$$

(36)

$$
Q = \frac{27}{16} - \gamma^2 - \frac{27}{16}\epsilon_1 + \frac{39}{16}\epsilon_2 + \frac{371}{32}A_{21} + \frac{65}{32}A_{22} + \frac{342}{31}M_d.
$$
 (37)

On substituting $\lambda = i \omega$ in eigenvector [\(31\)](#page-6-2) and separating the real and imaginary parts as

$$
U(\omega) = \left(-2a_6 + n^2 + \omega^2, \ a_7, -na_7, -na_6 + n^3 - n\omega^2\right)^T,
$$
\n(38)

$$
V(\omega) = \left(0, 2n\omega, -2a_6\omega - n^2\omega + \omega^3, a_7\omega\right)^T, \tag{39}
$$

where*U* and *V* represent the real and imaginary parts of eigenvector associated to each frequency ω_i , $i = 1, 2$ and *T* denotes the transpose of matrix. Now, construct a matrix **C** as $C = (V_1, V_2, U_1, U_2)$ which represents real symplectic change in state variables corresponding to each frequency ω_1 and ω_2 . For symplectic change, matrix **C** must satisfy the condition $C^T \cdot J \cdot C = J$. By the use of Eqs. (33) and (34) , symplectic condition is verified and found that

$$
\mathbf{C}^T \cdot J \cdot \mathbf{C} = \begin{pmatrix} 0 & 0 & d(\omega_1) & 0 \\ 0 & 0 & 0 & d(\omega_2) \\ -d(\omega_1) & 0 & 0 & 0 \\ 0 & -d(\omega_2) & 0 & 0 \end{pmatrix}, \tag{40}
$$

where $d(\omega) = \omega \left[4a_6^2 + a_7^2 + 4a_6n^2 - 3n^4 \right]$ $+(2n^2 - 4a_6) \omega^2 + \omega^4$. In order to satisfy the symplectic condition, a scaling is performed in the columns of matrix **C**, so that $C^T \cdot J \cdot C = J$. To do this, define $c_{1,2} = +\sqrt{d(\omega_{1,2})}$ and rewrite the matrix **C** as $C = (V_1/c_1, V_2/c_2, U_1/c_1, U_2/c_2)$. Again, matrix **C** to be real, we must take $d(\omega_{1,2}) > 0$. In other words, frequencies must satisfy $\omega_1 > 0$ and $\omega_2 > 0$. With this transformation, matrix **C** become symplectic, and hence, the Hamiltonian function [\(17\)](#page-4-5) reduces to a real normal from as

$$
H_2 = \frac{\omega_1}{2} (x^2 + p_x^2) + \frac{\omega_2}{2} (y^2 + p_y^2).
$$
 (41)

The elements of the final symplectic matrix $\mathbf{C} = [c_{ij}]$ are given as

$$
c_{11} = 0, c_{12} = 0, c_{13} = \frac{-2 a_6 + n^2 + \omega_1^2}{\sqrt{d(\omega_1)}},
$$

\n
$$
c_{14} = \frac{-2 a_6 + n^2 + \omega_2^2}{\sqrt{d(\omega_2)}},
$$

\n
$$
c_{21} = \frac{2n\omega_1}{\sqrt{d(\omega_1)}}, c_{22} = \frac{2n\omega_2}{\sqrt{d(\omega_2)}}, c_{23} = \frac{a_7}{\sqrt{d(\omega_1)}},
$$

\n
$$
c_{24} = \frac{a_7}{\sqrt{d(\omega_2)}},
$$

\n
$$
c_{31} = \frac{-(2a_6 + n^2)\omega_1 + \omega_1^3}{\sqrt{d(\omega_1)}}, c_{32} = \frac{-(2a_6 + n^2)\omega_2 + \omega_2^3}{\sqrt{d(\omega_2)}},
$$

\n
$$
c_{33} = \frac{na_7}{\sqrt{d(\omega_1)}}, c_{34} = \frac{n a_7}{\sqrt{d(\omega_2)}}, c_{41} = \frac{a_7\omega_1}{\sqrt{d(\omega_1)}},
$$

\n
$$
c_{42} = \frac{a_7\omega_2}{\sqrt{d(\omega_2)}},
$$

\n
$$
c_{43} = \frac{-2n a_6 + n^3 - n\omega_1^2}{\sqrt{d(\omega_1)}}, c_{44} = \frac{-2n a_6 + n^3 - n\omega_2^2}{\sqrt{d(\omega_2)}}.
$$

In the absence of perturbation parameters ϵ_1 , ϵ_2 , A_{21} , A_{22} and M_d , the elements of the symplectic matrix **C** agree with the results in [\[17,](#page-15-5)[22\]](#page-15-10). Next, we obtain generating function for the solution of intermediary homological equations. To obtain generating function in simpler way, real normalised Hamiltonian [\(41\)](#page-7-1) is transformed in complex normal form by the use of complex variables defined [\[22](#page-15-10)] as

$$
x = \frac{X_1 + iX_3}{\sqrt{2}}, \ y = \frac{X_2 + iX_4}{\sqrt{2}}, \tag{42}
$$

$$
p_x = \frac{iX_1 + X_3}{\sqrt{2}}, \ \ p_y = \frac{iX_2 + X_4}{\sqrt{2}}, \tag{43}
$$

and complex normal form of Hamiltonian is obtained as

$$
H_2 = i \omega_1 X_1 X_3 - i \omega_2 X_2 X_4, \tag{44}
$$

where real variables X_1 , X_2 correspond to space coordinates and X_3 , X_4 to that of corresponding momentum coordinates. Equation [\(44\)](#page-7-2) represents the linear order normal form of the Hamiltonian of the problem in question, which is to be used to obtained higher-order normal form of the Hamiltonian.

4 Nonlinear stability

In this section, the nonlinear stability of the equilibrium points in non-resonance case with the assumed perturbations is studied. In resonance case, one can perform the nonlinear stability test by the use of theorems as in [\[42](#page-16-10)], also stated as in [\[43\]](#page-16-11). For non-resonance case, Arnold–Moser theorem is to be applied, which predicts the nonlinear stability of equilibrium points in the restricted three-body problem. Before applying Arnold–Moser theorem, we have to determine the Birkhoff's normal form of the Hamiltonian up to fourth order about the equilibrium point as a function of action-angle variable $(I_1, I_2, \phi_1, \phi_2)$ as follows [\[17](#page-15-5)[,30](#page-16-12)[,44](#page-16-13),[45\]](#page-16-14):

4.1 Birkhoff's normal form

A Hamiltonian function, which is expressed in actionangle variables $(I_1, I_2, \phi_1, \phi_2)$ as

$$
K = K_2 + K_4 + \dots + K_{2n} + K_{2n+1},\tag{45}
$$

is of Birkhoff's normal form [\[45](#page-16-14)[,46](#page-16-15)] up to the terms *n* if *K*2*ⁿ* denotes homogeneous polynomial of degree *n* in action variables I_1 , I_2 and K_{2n+1} is that of higher degree polynomial than *n*; $K_2 = \omega_1 I_1 - \omega_2 I_2$ with constants $\omega_{1,2} > 0$; K_4 is given as $K_4 = -(A I_1^2 + B I_1 I_2 + A_2 I_2)$ CI_2^2) with *A*, *B*, *C* as constants to be determined. This form can be found by imposing non-resonance

Fig. 2 Frequency ω_1 (bold line) and ω_2 (dashed line) with respect to mass ratio μ at (**i**) $\epsilon_1 = 0$ (red) (**ii**) $\epsilon_1 = 0.04$ (blue) (**iii**) $\epsilon_1 = 0.08$ (green) (**iv**) $\epsilon_1 = 0.12$ (black) (**v**) $\epsilon_1 = 0.20$ (cyan)

condition on the frequencies ω_1 , ω_2 described as in $[30,31,47]$ $[30,31,47]$ $[30,31,47]$ $[30,31,47]$ and which is stated as if frequencies of infinitesimal mass in linear dynamics are ω_1 , ω_2 and $s \in \mathbb{Z}$ is such that $s > 2$, then

$$
s_1\omega_1 + s_2\omega_2 \neq 0 \tag{46}
$$

for all $s_1, s_2 \in \mathbb{Z}$ with $|s_1| + |s_2| \leq 2s$. This condition of irrationality insures the existence of a symplectic normalising transformation which transforms the Hamiltonian (16) in the form of Hamiltonian (45) and the coefficients of the normalised Hamiltonian are independent to the integer *s* as well as to the transformation determined. In particular, the determinant

$$
\det \begin{vmatrix} \frac{\partial^2 K}{\partial I_1^2} & \frac{\partial^2 K}{\partial I_1 \partial I_2} & \frac{\partial K}{\partial I_1} \\ \frac{\partial^2 K}{\partial I_2 \partial I_1} & \frac{\partial^2 K}{\partial I_2^2} & \frac{\partial K}{\partial I_2} \\ \frac{\partial K}{\partial I_1} & \frac{\partial K}{\partial I_2} & 0 \end{vmatrix}_{I_1, I_2=0}
$$
(47)

is an invariant of the Hamiltonian [\(45\)](#page-7-3) with respect to this symplectic transformation. Now, the Arnold– Moser theorem is stated [\[17,](#page-15-5)[48\]](#page-16-18) as

Arnold–Moser Theorem: *The origin is stable for the system whose Hamiltonian is in Birkhoff 's normal form provided for some integer* $i \in [2, n]$ *,* $D_{2i} =$ $K_{2i}(\omega_2, \omega_1) \neq 0$.

In this paper, we are interested to compute the Birkhoff's normal form of the Hamiltonian [\(16\)](#page-4-4) in action variables and then compute the determinant *D*⁴ corresponding to K 's (45) in terms of assumed perturbations.

Fig. 3 Frequency ω_1 (bold line) and ω_2 (dashed line) with respect to mass ratio μ at (**i**) $\epsilon_2 = 0$ (red) (**ii**) $\epsilon_2 = 0.02$ (blue) (**iii**) $\epsilon_2 = 0.04$ (green) (**iv**) $\epsilon_2 = 0.06$ (black) (**v**) $\epsilon_2 = 0.08$ (cyan)

Fig. 4 Frequency ω_1 (bold line) and ω_2 (dashed line) with respect to mass ratio μ at (**i**) $A_{21} = 0$ (red) (**ii**) $A_{21} = 0.002$ (blue) (**iii**) $A_{21} = 0.006$ (green) (**iv**) $A_{21} = 0.008$ (black) (**v**) $A_{21} = 0.020$ (cyan)

Fig. 5 Frequency ω_1 (bold line) and ω_2 (dashed line) with respect to mass ratio μ at (**i**) $A_{22} = 0$ (red) (**ii**) $A_{22} = 0.002$ (blue) (**iii**) $A_{22} = 0.006$ (green) (**iv**) $A_{22} = 0.008$ (black) (**v**) $A_{22} = 0.020$ (cyan)

Fig. 6 Frequency ω_1 (bold line) and ω_2 (dashed line) with respect to mass ratio μ at (**i**) $M_d = 0$ (red) (**ii**) $M_d = 0.002$ (blue) (**iii**) $M_d = 0.006$ (green) (**iv**) $M_d = 0.008$ (black) (**v**) $M_d = 0.020$ (cyan)

Fig. 7 Comparison between the frequencies up to linear order (red line) and higher-order (black dashed line) terms of perturbation parameters with respect to mass ratio μ

Fig. 8 D_4 Vs μ in the absence of perturbations

4.2 Fourth-order Normalised Hamiltonian

However, at least fourth-order Birkhoff's normal form of the normalised Hamiltonian is required to verify the Arnold–Moser theorem, which can be obtained from second order normalised Hamiltonian [\(44\)](#page-7-2) by using the Lie transform method described in [\[18](#page-15-6),[22,](#page-15-10)[30](#page-16-12)[,31](#page-16-16)[,49](#page-16-19)]. Suppose, higher-order normalised Hamiltonian [\[49](#page-16-19),[50\]](#page-16-20) is given as

$$
K = K_2 + K_3 + K_4 + K_5 + \dots + K_n + \dots, \quad (48)
$$

where K_n is defined as

$$
K_n = \sum_{j+k+l+m=n} K_{jklm} X_1^j X_2^k X_3^l X_4^m.
$$

The first term K_2 of the normalised Hamiltonian K , equals to H_2 [\(44\)](#page-7-2) and the n^{th} term of *K* [\(48\)](#page-9-0) is expressed as [\[18](#page-15-6)]

$$
K_n = \frac{1}{n} \{H_2, W_n\} + \text{terms previously known},
$$

where $\{.,.\}$ denotes Lie bracket and W_n is generating function to be determined. Now, generating function W_n is chosen in such a way that K_n is reduced in its better form. The Lie bracket ${H_2, W_n}$ is expanded as

$$
\{H_2, W_n\} = i \omega_1 \left[X_3 \frac{\partial W_n}{\partial X_3} - X_1 \frac{\partial W_n}{\partial X_1} \right] - i \omega_2 \left[X_4 \frac{\partial W_n}{\partial X_4} - X_2 \frac{\partial W_n}{\partial X_2} \right].
$$
 (49)

Now, we have to determine W_n such that the partial derivatives of W_n result in cancellation of terms as much as possible in K_n . It is noticed that these terms be of the form $M^*X_1^jX_2^kX_3^lX_4^m$ where M^* is constant and W_n is chosen to be the sum of terms that cancels each terms, given as $W_n = N^* X_1^j X_2^k X_3^l X_4^m$ where N^* is to be determined. The expression

$$
\frac{1}{n} \{H_2, W_n\} = \frac{1}{n} [i\omega_1 (l-j) - i\omega_2 (m-k)]
$$

$$
N^* X_1^j X_2^k X_3^l X_4^m
$$
 (50)

leads to

$$
N^* = \frac{i n M^*}{\omega_1 (l - j) - \omega_2 (m - k)}.
$$
 (51)

If the denominator in Eq. (51) vanishes, then the above scheme fails. But, we are discussing the case of non-resonance (in which frequencies are incommensurable), so the denominator equals to zero only if $l = j$ and $m = k$, simultaneously. This implies that the terms in normalised Hamiltonian

Fig. 9 *D*₄ Vs μ in the presence of radiation pressure **a** $\epsilon_1 = 0.04$, **b** $\epsilon_1 = 0.08$, **c** $\epsilon_1 = 0.12$ and **d** $\epsilon_1 = 0.20$

K will be a homogeneous polynomial of the form $(X_1 X_3)^j (X_2 X_4)^k$. Thus, in non-resonance case, the terms of the normalised Hamiltonian K [\(48\)](#page-9-0) reduce to the form as

$$
K_2 = H_2 = i \omega_1 X_1 X_3 - i \omega_2 X_2 X_4,
$$

\n
$$
K_3 = 0,
$$

\n
$$
K_4 = K_{2020} (X_1 X_3)^2 + K_{1111} (X_1 X_3) (X_2 X_4)
$$

\n
$$
+ K_{0202} (X_2 X_4)^2,
$$

\n
$$
K_5 = 0,
$$

\n
$$
K_6 = K_{3030} (X_1 X_3)^3 + K_{2211} (X_1 X_3)^2 (X_2 X_4)
$$

\n
$$
+ K_{1122} (X_1 X_3) (X_2 X_4)^2 + K_{0303} (X_2 X_4)^3,
$$

\n
$$
K_7 = 0,
$$

\nand so on.

As, fourth-order Birkhoff's normal form is sufficient to apply Arnold–Moser theorem so we are restricted to perform analysis with K_2 , K_3 and K_4 , only. Now, in the action variables $I_1 = iX_1X_3$ and $I_2 = iX_2X_3$, *K*2, *K*³ and *K*⁴ are expressed as

$$
K_2 = H_2 = \omega_1 I_1 - \omega_2 I_2, \tag{52}
$$

$$
K_3 = 0,\t(53)
$$

$$
K_4 = -(K_{2020}I_1^2 + K_{1111}I_1I_2 + K_{0202}I_2^2),\tag{54}
$$

where K_{2020} , K_{1111} and K_{0202} are the coefficients of homogeneous quadratic terms [\(54\)](#page-10-0) in action variables. Equations [\(52–54\)](#page-10-0) together constitute the Birkhoff's normal form of the Hamiltonian of the problem in question up to order four, which is similar to that in [\[18](#page-15-6)[,22](#page-15-10)]. Now, these coefficients are further expressed in terms of perturbation parameters ϵ_1 , ϵ_2 , A_{21} , A_{22} and M_d as follows

$$
K_{2020} = K_{2020}^c + K_{2020}^{\epsilon_1} \epsilon_1 + K_{2020}^{\epsilon_2} \epsilon_2 + K_{2020}^{A_{21}} A_{21} + K_{2020}^{A_{22}} A_{22} + K_{2020}^{M_d} M_d, \tag{55}
$$

$$
K_{1111} = K_{1111}^c + K_{1111}^{\epsilon_1} + K_{1111}^{\epsilon_2} + K_{1111}^{A_{21}} A_{21}
$$

$$
+ K_{1111}^{A_{22}} A_{22} + K_{1111}^{M_d} M_d,
$$
(56)

$$
K_{0202} = K_{0202}^c + K_{1111}^{\epsilon_1} \epsilon_1 + K_{0202}^{\epsilon_2} \epsilon_2 + K_{0202}^{A_{21}} A_{21} + K_{0202}^{A_{22}} A_{22} + K_{0202}^{M_d} M_d, \tag{57}
$$

where the coefficients K_{jklm}^c , $K_{jklm}^{\epsilon_1}$, $K_{jklm}^{\epsilon_2}$, $K_{jklm}^{A_{21}}$, $K_{jklm}^{A_{22}}$ and $K_{jklm}^{M_d}$ with *j*, *k*, *l*, *m* = 1, 2, 3, 4 such that $j + k + l + m = 4$, $j = l$ and $k = m$, in each terms of Eqs. [\(55–57\)](#page-10-1) correspond to the classical term, radiation parameter ϵ_1 , albedo parameter ϵ_2 , oblateness of first primary *A*21, oblateness of second primary *A*²² and

Fig. 10 *D*₄ Vs μ in the presence of albedo **a** $\epsilon_2 = 0.04$, **b** $\epsilon_2 = 0.06$, **c** $\epsilon_2 = 0.08$ and **d** $\epsilon_2 = 0.08393$

the disc parameter M_d , respectively. In the absence of all perturbing parameters, the expressions of the coefficients K_{2020} , K_{1111} and K_{0202} coincide with coefficients as in case of the classical RTBP [\[18,](#page-15-6)[22\]](#page-15-10). Now, the determinant $D_4(\omega_2, \omega_1)$, required to use the Arnold– Moser theorem [\[17](#page-15-5)], is computed to examine the nonlinear stability of triangular equilibrium points *L*4,⁵ in the non-resonance case, which is given as

$$
D_4(\omega_2, \omega_1) = D_4^c + D_4^{\epsilon_1} \epsilon_1 + D_4^{\epsilon_2} \epsilon_2 + D_4^{A_{21}} A_{21} + D_4^{A_{22}} A_{22} + D_4^{M_d} M_d,
$$
 (58)

where D_4^c , $D_4^{e_1}$, $D_4^{e_2}$, $D_4^{A_{21}}$, $D_4^{A_{22}}$ and $D_4^{M_d}$ are the parts of *D*⁴ correspond to classical case, radiation parameter, albedo parameter, oblateness of first primary, oblateness of second primary and disc parameter, respectively. In the absence of perturbing parameters, expression [\(58\)](#page-11-0) of *D*⁴ is agreed to that of classical result as in [\[18](#page-15-6)[,30](#page-16-12)[,47](#page-16-17)], which is

$$
D_4 = D_4^c = \frac{-36 + 541 \omega_1^2 \omega_2^2 - 644 \omega_1^4 \omega_2^4}{8(1 - 2 \omega_2^2)(-4 + 25 \omega_1^4 \omega_2^4)}.
$$
 (59)

In the present problem, corresponding parts of the *D*⁴ associated to the perturbing parameters ϵ_1 , ϵ_2 , A_{21} , A_{22} and M_d are, respectively, given as

$$
D_4^{\epsilon_1} = \frac{G_1}{G_2}, D_4^{\epsilon_2} = \frac{G_3}{G_2}, D_4^{\lambda_{21}} = \frac{G_4}{G_5}, D_4^{\lambda_{22}} = \frac{G_6}{G_5} \text{ and}
$$

$$
D_4^{M_d} = \frac{G_7}{G_8},
$$
 (60)

where

$$
G_1 = 93676500 - 558820053\omega_1^2\omega_2^2 - 362539454\omega_1^4\omega_2^4
$$

+470308824 $\omega_1^6\omega_2^6$
+160171776 $\omega_1^8\omega_2^8$ + 11577344 $\omega_1^{10}\omega_2^{10}$,

$$
G_2 = -111837348 + 786057993\omega_1^2\omega_2^2
$$

+2379308438 $\omega_1^4\omega_2^4$ - 2738662648 $\omega_1^6\omega_2^6$
-265223424 $\omega_1^8\omega_2^8$ - 62580736 $\omega_1^{10}\omega_2^{10}$,

$$
G_3 = -197618292 + 5093401725\omega_1^2\omega_2^2
$$

-24531749522 $\omega_1^4\omega_2^4$ + 13637287400 $\omega_1^6\omega_2^6$
-4361320192 $\omega_1^8\omega_2^8$ - 1622632448 $\omega_1^{10}\omega_2^{10}$,

$$
G_4 = -294388668 + 2672825103\omega_1^2\omega_2^2
$$

-5556998150 $\omega_1^4\omega_2^4$ + 4738270136 $\omega_1^6\omega_2^6$
+257103104 $\omega_1^8\omega_2^8$ - 384555008 $\omega_1^{10}\omega_2^{10}$,

$$
G_5 = -1148948460 + 7875474171\omega_1^2\omega_2^2
$$

Fig. 11 *D*₄ Vs μ in the presence of oblateness of first primary **a**, **b** $A_{21} = 0.002$, **c** $A_{21} = 0.004$ and **d** $A_{21} = 0.0044$

$$
-3538195892\omega_1^4\omega_2^4 + 3475970190\omega_1^6\omega_2^6
$$

+1372056620 $\omega_1^8\omega_2^8 - 676372950\omega_1^{10}\omega_2^{10}$,
 $G_6 = 4313088\omega_1^2\omega_2^2 - 60871680\omega_1^4\omega_2^4$
+1455906816 $\omega_1^6\omega_2^6 - 1572028416\omega_1^8\omega_2^8$
-58982400 $\omega_1^{10}\omega_2^{10}$,
 $G_7 = 8626176\omega_1^2\omega_2^2 - 121743360\omega_1^4\omega_2^4$
+552517632 $\omega_1^6\omega_2^6 - 784760832\omega_1^8\omega_2^8$
-117964800 $\omega_1^{10}\omega_2^{10}$,
 $G_8 = 119558799360\omega_1^2\omega_2^2$
-1687362969600 $\omega_1^4\omega_2^4 + 7657894379520\omega_1^6\omega_2^6$
-10876785131520 $\omega_1^8\omega_2^8 - 1634992128000\omega_1^{10}\omega_2^{10}$.

Now, the variation in the frequencies ω_1 and ω_2 , which defines the determinant $D_4(\omega_2, \omega_1)$ at different values of perturbation parameters, is shown in Figs. [2,](#page-8-0) [3,](#page-8-1) [4,](#page-8-2) [5](#page-8-3) and [6.](#page-9-2) It is noticed that

because of radiation pressure $\epsilon_1 = 1 - q$, the value of ω_1 increases and that of ω_2 decreases, whereas due to the presence of albedo $\epsilon_2 = 1 - Q_A$, oblateness of the primaries A_{2i} , $i = 1, 2$, and the disc M_d , the value of frequency ω_1 decreases and that of the value of ω_2 increases. This significant change in the values of frequencies ω_1 and ω_2 leads to a considerable change in the value of zeros of the D_4 on μ -axis. Figure [7](#page-9-3) is drawn for the comparison between the frequencies which is computed by truncating the expansion for perturbation parameters up to linear order (red line) and those obtained by taking higher-order terms of the perturbing factors (black dashed line). Alike, in case of Fig. [1,](#page-3-0) it is observed that frequencies in both the cases, i.e. red curve and black dashed curve with respect to the mass ratio μ , seem coincident (Fig. [7\)](#page-9-3). As, in the absence of perturbations, it is seen that semi-analytical results agreed to that of classical one $[22,30,47]$ $[22,30,47]$ $[22,30,47]$ $[22,30,47]$ and in that case there exist only one value of μ such that $D_4 = 0$ (Fig. [8\)](#page-9-4). In order to analyse the effect of radiation pressure $\epsilon_1 = 1 - q$, albedo $\epsilon_2 = 1 - Q_A$, oblateness of the primaries A_{2i} , $i = 1, 2$ and the presence of disc M_d in the context of the nonlinear stability of triangular equilibrium points *L*4,5, the variations in the values of the determinant D_4 with respect to mass ratio μ within the stability range $0 < \mu < \mu_R$ are observed (Fig. [9\)](#page-10-2). Figure [9](#page-10-2) illustrates the changes in *D*⁴ and the existence of values of μ within the stability range $0 < \mu < \mu_R$

Fig. 12 D_4 Vs μ in the presence of oblateness of second primary **a**, **b** $A_{22} = 0.0001$, **c** $A_{22} = 0.00036$ and **d** $A_{21} = 0.00037$

at which *D*⁴ vanishes at different values of perturbing parameters, which are responsible for the changing the values of zeros of D_4 . It is seen that for the nonlinear stability, the region within the stability range is constant, where $D_4 \neq 0$ at different values of perturbing parameters. We found that there exist some values of μ at which Arnold–Moser theorem fails (Fig. [9\)](#page-10-2). In case of radiation pressure $\epsilon_1 = 1 - q$, there exist two values of μ for which $D_4 = 0$ and the difference in these two values increases with the increase in the value of ϵ_1 (Fig. [9\)](#page-10-2). But for albedo parameter ϵ_2 , there are two zeros of D_4 for $\epsilon_2 \in (0, 0.08393)$ and after that no such μ exists at which $D_4 = 0$ (Fig. [10\)](#page-11-1). Further, due to oblateness of the first primary *A*21, number of zeros of *D*⁴ is varying. At $A_{21} = 0.002$, $D_4 = 0$ for three values of μ and at $A_{21} = 0.004$, D_4 vanishes at two places on the μ -axis, whereas nearly after $A_{21} = 0.0043579$, D_4 is non-zero within the range of stability $0 < \mu < \mu_R$ $(Fig. 11)$ $(Fig. 11)$. A similar trend is obtained for the oblateness of the second primary A_{22} and noticed that D_4 crosses

 μ -axis four times when $A_{22} \in [0, 0.0001)$, twice when $A_{22} \in [0.0001, 0.000356]$ and once when $A_{22} \in$ $(0.000356, 0.013905)$ and for nearly $A_{22} > 0.013905$, no such values of μ are found, where Arnold–Moser theorem fails (Fig. [12\)](#page-13-0). In case of variation in the values of disc parameter M_d , there exists two values of $μ$ at which $D_4 = 0$ for $0 < M_d < 0.00241$ and for $0.00241 < M_d < 0.5$, $D_4 \neq 0$; hence, Arnold– Moser theorem is applicable for all values of μ within the stability range $0 < \mu < \mu_R$ (Fig. [13\)](#page-14-1). Now, it is summarised that because of perturbations, Arnold– Moser theorem fails at several places within the range of stability. Also, due to these perturbations, stability range increases and the triangular equilibrium points *L*4,⁵ are stable in nonlinear sense for those values of μ , where $D_4 \neq 0$. Thus, the effect of perturbations in the form of radiation pressure, albedo, oblateness of the primaries and the disc influenced the stability range as well as nature of triangular equilibrium points in nonlinear sense within the stability range.

ļμ.

IJ,

40

20

 $\mathbf 0$

 -20

 -40

 \mathbf{c}_4

Fig. 13 *D*₄ Vs μ in the presence of disc **a** $M_d = 0.0005$, **b** $M_d = 0.0024$ and **c** $M_d = 0.0024148$

5 Conclusion

The perturbed restricted three-body problem is studied in the context of nonlinear stability analysis of the triangular equilibrium points $L_{4,5}$ in non-resonance case by applying Arnold–Moser theorem under the influence of perturbations in the form of radiation due to first primary, albedo of second primary, oblateness of both the primaries and presence of the disc. First, equations of motion are established and then Hamiltonian of the proposed problem is obtained. Next, the linear order normalization of the Hamiltonian is performed by placing the origin at *L*⁴ point. Again, by the use of Lie series method, fourth-order normalised Hamiltonian is determined, which is similar to that of in [\[22](#page-15-10)[,47](#page-16-17)[,51](#page-16-21)], in the absence of perturbations. Further, the nonlinear stability test is performed with the help of Arnold–Moser Theorem. It is noticed that in the absence of perturbations, determinant D_4 vanishes at $\mu = 0.0109137$ within the classical stability range $0 < \mu < 0.0385209$ as in [\[22](#page-15-10)[,47](#page-16-17),[51\]](#page-16-21). However, in the presence of radiation pressure, the range of stability expanded, significantly

with respect to the classical range of stability but in the presence of other perturbations, i.e. albedo, oblateness and the disc, it is contracted gradually. Moreover, the number of zeros of the determinant $D_4(\omega_2, \omega_1)$ at μ axis is increased as compared to one as in classical case. The trend of existence of zeros of *D*⁴ within the corresponding stability ranges are different for different kind of perturbations. It is also found that frequencies $\omega_{1,2}$ are influenced due to the perturbations. Finally, it is concluded that in the presence of perturbations, Arnold–Moser theorem fails at many point within the respective stability ranges; hence, the triangular equilibrium points *L*4,⁵ are unstable in nonlinear sense for those value of mass ratio μ on the μ -axis at which $D_4 = 0$. In opposite of this, it is also mentioned that the points $L_{4,5}$ are stable within the range of stability except for those values of μ at which $D_4 \neq 0$. The current study and observations are applicable to other generalised problem of a restricted few body system and can be extended to higher-order normal form with various types of disturbances such as P-R drag, solar wind drag, etc.

0.0220

Acknowledgements Authors are thankful to UGC, Govt. of India, for providing partial support through the UGC startup research Grant No.-F.30-356/2017(BSR) and UGC-JRF (Ref. No.- 21/06/2015(i)EU-V). We are also thankful to Inter-University Center for Astronomy and Astrophysics (IUCAA), Pune (India), for sharing some of the references used in this article. Last but not least, we are thankful to all esteemed reviewer(s) for shaping the manuscript in its present form.

Author contributions SY performed conceptualisation, formal analysis, methodology, software, validation, visualisation, writing— original draft. RK helped in conceptualisation, formal analysis, methodology, resources, software, supervision, validation, visualisation, writing— reviewing and editing.

Funding This work was supported by UGC, Govt. of India (Grant numbers [UGC start-up research Grant No.-F.30-356/ 2017(BSR)] and [UGC-JRF (Ref. No.- 21/06/2015(i)EU-V).]).

Data availability Data sharing is not applicable to this manuscript as no datasets were generated or analysed during the current study.

Declarations

Conflict of interest The authors declare that they have no conflict of interest that are relevant to the content of this article.

References

- 1. Szebehely, V.: Theory of Orbits. The Restricted Problem of Three Bodies, pp. 1–40. Academic Press, New York pp (1967)
- 2. Chernikov, Y.A.: The Photogravitational Restricted Three-Body Problem. Sov. Astron. **14**(1), 176 (1970)
- 3. Kishor, R., Kushvah, B.S.: Periodic orbits in the generalized photogravitational chermnykh-like problem with power-law profile. Astrophys. Space Sci. **344**(2), 333–346 (2013)
- 4. Kishor, R., Kushvah, B.S.: Lyapunov characteristic exponents in the generalized photo-gravitational chermnykh-like problem with power-law profile. Planet. Space Sci. **84**, 93– 101 (2013)
- 5. Abouelmagd, E.I., Sharaf, M.: The motion around the libration points in the restricted three-body problem with the effect of radiation and oblateness. Astrophys. Space Sci. **344**(2), 321–332 (2013)
- 6. Luo, S.H., Sayanjali, M.: Fourth body gravitation effect on the resonance orbit characteristics of the restricted threebody problem. Nonlinear Dyn. **76**, 955–972 (2014)
- 7. Luo, T., Xu, M.: Dynamics of the spatial restricted three-body problem stabilized by Hamiltonian structurepreserving control. Nonlinear Dyn. **94**, 1889–1905 (2018)
- 8. Yousuf, S., Kishor, R.: Effects of albedo and disc on the zero velocity curves and linear stability of equilibrium points in the generalized restricted three body problem. Mon. Not. R. Astron. Soc. **488**(2), 1894–1907 (2019)
- 9. Luo, T., Pucacco, G., Xu, M.: Lissajous and halo orbits in the restricted three-body problem by normalization method. Nonlinear Dyn. **101**, 2629–2644 (2020)
- 10. Aslanov, V.S.: A splitting of collinear libration points in circular restricted three-body problem by an artificial electrostatic field. Nonlinear Dyn. **103**, 2451–2460 (2021)
- 11. Pousse, A., Alessi, E.M.: Revisiting the averaged problem in the case of mean-motion resonances in the restricted threebody problem. Nonlinear Dyn. **108**, 959–985 (2022)
- 12. Alrebdi, H.I., Dubeibe, F.L., Papadakis, K.E., Zotos, E.E.: Equilibrium dynamics of a circular restricted three-body problem with Kerr-like primaries. Nonlinear Dyn. **107**, 433– 456 (2022)
- 13. Yousuf, S., Kishor, R.: Impact of a disc and drag forces on the existence linear stability of equilibrium points and Newton– Raphson basins of attraction. Kinemat. Phys. Celest. Bodies **38**, 166–180 (2022)
- 14. Yousuf, S., Kishor, R., Kumar, M.: Motion about equilibrium points in the Jupiter-Europa system with oblateness. Appl. Math. Nonlinear Sci. (2022). [https://doi.org/10.2478/amns.](https://doi.org/10.2478/amns.2021.2.00124) [2021.2.00124](https://doi.org/10.2478/amns.2021.2.00124)
- 15. Deprit, A.: Canonical transformations depending on a small parameter. Celest. Mech. **1**(1), 12–30 (1969)
- 16. Liu, J.C.: The uniqueness of normal forms via lie transforms and its applications to hamiltonian systems. Celest. Mech. **36**(1), 89–104 (1985)
- 17. Meyer, K., Schmidt, D.: The stability of the Lagrange triangular point and a theorem of Arnold. J. Differ. Equ. **62**(2), 222 (1986)
- 18. Coppola, V.T., Rand, R.: Computer algebra implementation of lie transforms for hamiltonian systems: application to the nonlinear stability of l4. ZAMM-J. Appl. Math. Mech./Zeitschrift für Angewandte Mathematik und Mechanik **69**(9), 275–284 (1989)
- 19. Subba Rao, P., Krishan Sharma, R.: Effect of oblateness on the non-linear stability of l 4 in the restricted three-body problem. Celest. Mech. Dyn. Astron. **65**(3), 291–312 (1996)
- 20. Ishwar, B.: Non-linear stability in the generalized restricted three-body problem. Celest. Mech. Dyn. Astron. **65**(3), 253– 289 (1996)
- 21. Jorba, A., Villanueva, J.: Numerical computation of normal forms around some periodic orbits of the restricted threebody problem. Physica D **114**(3–4), 197–229 (1998)
- 22. Jorba, A.: A methodology for the numerical computation of normal forms, centre manifolds and first integrals of hamiltonian systems. Exp. Math. **8**(2), 155–195 (1999)
- 23. Kushvah, B., Sharma, J., Ishwar, B.: Nonlinear stability in the generalised photogravitational restricted three body problem with poynting-robertson drag. Astrophys. Space Sci. **312**(3), 279–293 (2007)
- 24. Kushvah, B., Sharma, J., Ishwar, B.: Normalization of hamiltonian in the generalized photogravitational restricted three body problem with poynting-robertson drag. Earth Moon Planet. **101**(1), 55–64 (2007)
- 25. Alvarez-Ramírez, M., Formiga, J., de Moraes, R., Skea, J., Stuchi, T.: The stability of the triangular libration points for the plane circular restricted three-body problem with light pressure. Astrophys. Space Sci. **351**(1), 101–112 (2014)
- 26. Markellos, V., Papadakis, K., Perdios, E.: Non-linear stability zones around triangular equilibria in the plane circular restricted three-body problem with oblateness. Astrophys. Space Sci. **245**(1), 157–164 (1996)
- 27. Chandra, N., Kumar, R.: Effect of oblateness on the nonlinear stability of the triangular liberation points of the

restricted three-body problem in the presence of resonances. Astrophys. Space Sci. **291**(1), 1–19 (2004)

- 28. Singh, J.: Effect of perturbations on the non linear stability of triangular points in the restricted three-body problem with variable mass. Astrophys. Space Sci. **321**(2), 127–135 (2009)
- 29. Singh, J.: Combined effects of perturbations, radiation, and oblateness on the nonlinear stability of triangular points in the restricted three-body problem. Astrophys. Space Sci. **332**(2), 331–339 (2011)
- 30. Kishor, R., Kushvah, B.S.: Normalization of hamiltonian and nonlinear stability of the triangular equilibrium points in non-resonance case with perturbations. Astrophys. Space Sci. **362**(9), 1–18 (2017)
- 31. Kishor, R., Raj, M.X.J., Ishwar, B.: Normalization of Hamiltonian and nonlinear stability of triangular equilibrium points in the photogravitational restricted three body problem with P-R drag in non-resonance case. Qual. Theory Dyn. Syst. **18**, 1055–1075 (2019)
- 32. Cárcamo-Díaz, D., Palacián, J.F., Vidal, C., Yanguas, P.: On the nonlinear stability of the triangular points in the circular spatial restricted three-body problem. Regular Chaotic Dyn. **25**(2), 131–148 (2020)
- 33. Zepeda Ramírez, J.A., Alvarez-Ramírez, M., García, A.: Nonlinear stability of equilibrium points in the planar equilateral restricted mass-unequal four-body problem. Int. J. Bifurc. Chaos **31**(11), 2130031–15 (2021). [https://doi.org/](https://doi.org/10.1142/S0218127421300317) [10.1142/S0218127421300317](https://doi.org/10.1142/S0218127421300317)
- 34. Zepeda Ramírez, J.A., Alvarez-Ramírez, M., García, A.: A note on the nonlinear stability of equilibrium points in the planar equilateral restricted mass-unequal four-body problem. Int. J. Bifurc. Chaos **32**(2), 2250029–15 (2022). [https://](https://doi.org/10.1142/S0218127422500298) doi.org/10.1142/S0218127422500298
- 35. Ragos, O., Zagouras, C.: On the existence of the "out of plane" equilibrium points in the photogravitational restricted three-body problem. Astrophys. Space Sci. **209**(2), 267 (1993)
- 36. McCuskey, S.W.: Introduction to Celestial Mechanics. Addison-Wesley Pub. Co., Reading (1963)
- 37. Miyamoto, M., Nagai, R.: Three-dimensional models for the distribution of mass in galaxies. Publ. Astron. Soc. Jpn. **27**, 533–543 (1975)
- 38. Kushvah, B.S.: Linear stability of equilibrium points in the generalized photogravitational chermnykh's problem. Astrophys. Space Sci. **318**(1), 41–50 (2008)
- 39. Murray, C.D., Dermott, S.F.: Solar System Dynamics, pp. 63–128. Cambridge University Press, Cambridge (2000)
- 40. Kushvah, B.S.: Linear stability of equilibrium points in the generalized photogravitational Chermnykh's problem. Astrophys. Space Sci. **318**, 41 (2008)
- 41. Singh, J., Amuda, T.O.: Stability analysis of triangular equilibrium points in restricted three-body problem under effects of circumbinary disc, radiation and drag forces. J. Astrophys. Astron. **40**(1), 5 (2019)
- 42. Markeev, A.P., Sokolskii, A.G.: On the stability of periodic motions which are close to Lagrangian solutions. Sov. Astron. **21**, 507–512 (1977)
- 43. Goździewski, K.: Nonlinear Stability of the Lagrangian Libration Points in the Chermnykh Problem. Celest. Mech. Dyn. Astron. **70**, 41–58 (1998)
- 44. Meyer, K., Hall, G.: Book-review-introduction to hamiltonian dynamical systems and the n-body problem. Science **255**, 1756 (1992)
- 45. Jorba, A., Masdemont, J.: Dynamics in the center manifold of the collinear points of the restricted three body problem. Physica D **132**(1–2), 189–213 (1999)
- 46. Birkhoff, G.D.: Dynamical System. American Mathematical Society Colloquium Publications, New York (1927)
- 47. Deprit, A., Deprit-Bartholome, A.: Stability of the triangular Lagrangian points. Astron. J. **72**(2), 173 (1967)
- 48. Meyer, K.R., Hall, G.R., Offin, D.C.: Introduction to Hamiltonian Dynamical Systems and the N-Body Problem. Springer, New York (1992)
- 49. Celletti, A.: Stability and Chaos in Celestial Mechanics. Springer, Berlin (2010)
- 50. Coppola, V.T., Rand, R.H.: Computer algebra, Lie transforms and the nonlinear stability of L4. Celest. Mech. **45**, 103–103 (1989)
- 51. Shevchenko, I.I.: Symbolic computation of the Birkhoff normal form in the problem of stability of the triangular libration points. Comput. Phys. Commun. **178**(9), 665–672 (2008)

Publisher's Note Springer Nature remains neutral with regard to jurisdictional claims in published maps and institutional affiliations.

Springer Nature or its licensor (e.g. a society or other partner) holds exclusive rights to this article under a publishing agreement with the author(s) or other rightsholder(s); author self-archiving of the accepted manuscript version of this article is solely governed by the terms of such publishing agreement and applicable law.