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# **OPTIMIZATION OF THE T6 HEAT TREATMENT TO IMPROVE THE MECHANICAL PROPERTIES OF ALLOY AlSi12CuNiMg**

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A study is made of the effect of the T6 heat treatment on the mechanical properties (strength, hardness, elongation, impact toughness) of aluminum alloy AlSi12CuNiMg modified with master alloy Al – 10% Sr. Differential thermal analysis is used to obtain curves that describe the heating of the alloy before it melts. The regimes that are optimum for solution treatment and aging of the alloy are determined.

*Key words:* aluminum alloys, thermal analysis, heat treatment, mechanical properties.

## **INTRODUCTION**

The automotive industry is currently giving preference to the use of aluminum over the heavier steel and copper in order to meet standards that have been introduced to reduce the weight of automobiles [1, 2]. For the manufacturer, this means a reduction in the weight of the engine without a loss of performance, which in turn allows a reduction in the weight of components of the transmission and the volume of the fuel tank [3]. Predictions indicate that over the period from 1996 to 2015 the total weight of the aluminum parts in an automobile will have increased from 110 to 250 or 340 kg with or without allowance for the panels and the structural elements of the body, respectively [4]. Cast aluminum alloys of the silumin type are being widely used to make various automotive parts: pistons, cylinder heads, tubes and transmissions, wheels and brakes, brake components, suspensions, the steering mechanism, etc.  $[5 - 9]$ .

The mechanical properties of silumins are improved by alloying  $[10 - 13]$ , by modification of the melt (by altering the form of the particles of silumin's brittle phase, which helps improve the mechanical and processing properties)  $[14 - 18]$ , and by administering heat treatments that induce precipitation  $[19 - 21]$ .

The introduction of additional alloying elements (mainly Mg and Cu) in combination with heat treatment further improves mechanical properties  $[22 - 24]$ . The parameters of the individual operations that are part of the T6 heat treatment have a direct effect on the mechanical properties of silumin  $[25 - 29]$ . However, those parameters also determine the cost of the finished product. The consequent need to reduce the cost of making castings is in turn making it necessary to develop a heat-technology which is of the optimum (short) duration.

The goal of this study is to determine the T6 heat-treatment parameters for foundry alloy AlSi12CuNiMg that are optimum from the standpoint of improving its mechanical characteristics (tensile strength, elongation, impact toughness, and hardness) while shortening the solution treatment and aging operation as much as possible.

## **METHODS OF STUDY**

The chemical composition of the aluminum alloy that was studied was determined with glow-discharge spectrometer GDS 850A. It was established that the alloy contains the following, wt.%: 11 Si; 1.15 Cu; 1.35 Ni; 1.1 Mg; 0.50 Fe; 0.35 Mn; 0.15 Zn; 0.09 Cr; 0.05 Ti; remainder — Al.

In the first stage of the investigation, we refined the melt by Rafal 1 method. This method entails the treatment of aluminum alloys (or magnesium alloys) with a degassing flux at 730°C for 5 min. The flux is based on hexachloroethane. The flux is in the form of 50-g pellets submerged on the bottom of the metal bath and is used in an amount corresponding to 0.4% of the weight of the melt. After refining, the oxides and the slag are pumped from the surface of the melt and the melt is modified with strontium by adding master alloy AlSr10A to it in an amount corresponding to 0.4% of the weight of the charge. After modification, the Crystaldimat analyzer

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**Fig. 1.** DTA curves of refined and modified alloy AlSi12CuNiMg:  $t<sub>s</sub>$ ) solution-treatment temperature;  $t<sub>a</sub>$ ) aging temperature.

[30 – 33] is used to record DTA (differential thermal analysis) curves during the solidification process.

Figure 1 shows DTA curves of the heating (melting) and crystallization of the refined and modified alloy. The solidsolution treatment temperature  $(t_a)$  and the aging temperature  $(t<sub>s</sub>)$  are indicated in the figure.

After being modified, the alloy was poured into permanent molds corresponding to standard specimens used to test castings in tension. The molds were preheated to 250°C. The cast specimens underwent dispersion-hardening while they were being held at a temperature near the solidus. The heat treatment included heating of the cast specimen to the solution-treatment temperature, holding at that temperature, cooling in water, and aging.

The parameters of the heat treatment are shown in Table 1. Completion of the three-level factorial experiment with four variables made it possible to the temperatures and duration of the solution treatment and aging operation based on the properties of the alloy: tensile strength  $\sigma_r$ , elongation  $\delta$ , Brinell hardness *HB*, and impact toughness measured on a circular specimen. Figure 2 shows a diagram of the experiments.

After the heat treatment, a ZD-20 testing machine conforming to the standard EN ISO 6892-1:2010 was used to perform static tests in tension. Brinell hardness was measured on a PRL 82 testing machine in accordance with the standard EN ISO 6506-1:2008. The diameter of the ball was 10 mm, the load was 9800 N, and the holding time was 30 sec.

Figure 3 presents a sketch of the specimen used to determine impact toughness. The simplified method used for the impact-toughness test involved the use of a less expensive and simpler specimen of cylindrical form [34]. In the case being discussed, impact toughness is determined as the ratio of the work done in fracturing the specimen to the surface area of a cross section of the specimen located near the notch.



**Fig. 2.** Diagram of experiments for determining the optimum heat-treatment regimes for alloy AlSi12CuNiMg:  $t_s$  and  $\tau_s$ ) temperature and time of solution treatment, respectively;  $t_a$  and  $\tau_a$ ) aging temperature and time respectively;  $c_1$  ) material;  $c_2$  ) temperature of the coolant water;  $c_3$ ) refining and modifying additions;  $n_1$ ) fluctuations of furnace temperature;  $n_2$ ) casting defects in the specimen;  $o_1$ ) tensile strength;  $o_2$ ) elongation;  $o_3$ ) impact toughness;  $o_4$ ) Brinell hardness *HB*.



**Fig. 3.** Specimen for impact-toughness tests.

#### **RESULTS AND DISCUSSION**

The tensile strength of the specimen increased to  $\sigma_r$  = 218 MPa after modification, while it rose from  $\sigma_r$  = 184 MPa to  $\sigma_r$  = 426 MPa after the T6 heat treatment. Figure 4*a* and *b* present three-dimensional diagrams that show how  $\sigma_r$  is affected by the temperatures and duration of the solution treatment and aging operation.

The value of  $\sigma_u$  of the heat-treated alloy was as much as 95% higher than the value for the original (modified) alloy. The highest value of strength was reached after solution treatment in the range  $520 - 530^{\circ}$ C and aging at  $180^{\circ}$ C for

**TABLE 1.** Heat-Treatment Parameters for Alloy AlSi12CuNiMg

$t_{s}$ , <sup>o</sup> C	$\tau_{s}$ , h	$t_a$ , <sup>o</sup> C	$\tau_a$ , h
$t_{s1} = 510$	0.5	$t_{a1} = 180$	
$t_{s2} = 520$	1.5	$t_{a2} = 230$	
$t_{s3} = 530$	3.0	$t_{a3} = 310$	

**Notes:**  $t_s$  and  $\tau_s$  are the holding temperature and time for the specimen in the solution treatment;  $t_a$  and  $\tau_a$  are the temperature and duration of the aging operation.

400 360 320 28 450 350 250 150 3.2 1 2.4  $\frac{1}{2}$   $\frac{1}{2$ 1.6 5 0.8 7 9  $\sigma_{\rm r}$ , MPa  $\sigma_{\rm r}$ , MPa  $\frac{2}{3}$ , h *<sup>s</sup>* , <sup>h</sup> *<sup>a</sup>* , <sup>h</sup> *<sup>a</sup>* , <sup>h</sup>  $\mathcal{S}_{\mathbf{z}_{O}}$  $\mathcal{S}$  $\partial^{\hspace{-3pt}\cdot\cdot\hspace{3pt}\cdot\hspace{3pt}}_\mathscr{O}$  $\delta_{\cal O}$  $\frac{1}{60}$  $\mathfrak{G}_O$ 2<br>P  $\mathcal{Z}_{\mathcal{O}}$  $\mathcal{S}_{\mathcal{L}_{O}}$  $\mathcal{E}_{\mathcal{E}_{\mathcal{O}}}$  $\delta_{\!\! \rho_{\! \! \rho}}$  $\partial \alpha$  $\partial_{\mathcal{O}}$ **RO**  $\mathcal{S}_{\mathcal{E}}$  $\frac{6}{1}$ <sub>s</sub>, *<sup>t</sup><sup>s</sup>* , °C  $\frac{2}{c}$ ,  $\frac{160}{c}$ *ta* $\int_{0}^{\infty}$ *à*  $c \rightarrow$  defined by  $\sim$  defined by  $d$ *b*  $\delta$ , % δ, 4.0 3.5 3.0 2.5 2.0 1.5 5 43  $\overline{2}$ 1  $540 \approx 320$ 530 280 <sup>520</sup> <sup>240</sup> <sup>510</sup>  $200$   $200$ 490°<del>3</del>  $\geq$ . જ ಾ 1 2.4  $\frac{1}{6}$  $o_{\phi}$ 

**Fig. 4.** Dependence of the tensile strength  $(a, b)$  and elongation  $(c, d)$  of alloy AlSi12CuNiMg on treatment time and temperature: *a*, *c*) solution treatment (with subsequent aging at  $t_a = 180^\circ \text{C}$  and  $\tau_a$  = 5 h); *b*, *d*) aging (before aging) solution treatment with  $t_s$  = 520°C and  $\tau_s$  = 1 h).

several hours (Fig. 4*a* and *b* ). Aging decreased the value of  $\sigma_r$  when higher temperatures were used for the aging operation.

Elongation increased from 1.6 to 2% after modification, while  $\delta = 0.6 - 4\%$  after the T6 heat treatment. Figure 4c and *d* present three-dimensional diagrams of the change in elongation in relation to the heat treatment parameters.

The 100% increase in specimen elongation that resulted after the heat treatment corresponds to solution treatment at temperatures above 510°C for a longer period of time than in the case of the increase in  $\sigma_r$  and to aging above 280°C (Fig. 4*c* and *d* ). Elongation decreased after a brief solution treatment combined with aging above  $240^{\circ}$ C for  $5-8$  h (elongation decreased 70% in this case compared to the value of  $\delta$  for the modified alloy).

The hardness of the alloy increased to 77 *HB* after modification and was within the range 59 – 143 *HB* after T6 heat treatment. Figure 5*a* and *b*, show three-dimensional diagrams of the change in hardness in relation to the heat treatment regimes. Hardness changes in a manner similar to tensile strength. The increase in hardness after the T6 heat treatment was 85%. With the use of aging temperatures above  $310^{\circ}$ C and an aging time of  $5 - 8$  h, hardness decreased regardless of the conditions in the solution treatment.

Impact toughness increased to  $2.2 \text{ kJ/m}^2$  after modification and was within the range  $2.3 - 6.7 \text{ kJ/m}^2$  after the T6 heat treatment. This range corresponds to a 200% increase in impact toughness compared to the specimens that were not heat-treated. Three-dimensional diagrams of the change in impact toughness with heat treatment are shown in Fig. 5*c* and *d*. It is apparent that the highest values for impact toughness were reached after solution treatment above 520 $^{\circ}$ C for 1.5 – 3 h and aging at 310 $^{\circ}$ C for 5 – 8 h. Aging for less time and at lower temperatures (180 – 200°C) decreases the impact toughness of the alloy.



**Fig. 5.** Dependence of the Brinell hardness  $(a, b)$  and impact toughness  $(c, d)$  of alloy AlSi12CuNiMg on treatment temperature and time: *a*, *c*) solution treatment (with subsequent aging at  $t_a = 180^{\circ}\text{C}$  and  $\tau_a = 5$  h); *b*, *d*) aging (after solution treatment with  $t_s = 520^{\circ}\text{C}$  and  $\tau_s = 1$  h).

### **CONCLUSIONS**

1. Heat treatment of alloy AlSi12CuNiMg in the T6 regime — which includes holding the alloy close to the solidus temperature, quenching it in water, and subjecting it to artificial aging — helps alter its mechanical properties. The largest increases in tensile strength (up to 426 MPa) and hardness are obtained after solution treatment at 520 – 530°C for  $0.5 - 1.5$  h and aging at  $180^{\circ}$ C for  $5 - 8$  h.

2. The largest increases in elongation and impact toughness are obtained with solution treatment at 520 – 530°C for  $1.5 - 3$  h and aging at  $310^{\circ}$ C for  $5 - 8$  h.

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