Pulsating fluid induced dynamic instability of visco-double-walled carbon nano-tubes based on sinusoidal strain gradient theory using DQM and Bolotin method

A. Ghorbanpour Arani · R. Kolahchi · M. Mosayyebi · M. Jamali

Abstract This research deals with the dynamic instability analysis of double-walled carbon nanotubes (DWCNTs) conveying pulsating fluid under 2D magnetic fields based on the sinusoidal shear deformation beam theory (SSDBT). In order to present a realistic model, the material properties of DWCNTs are assumed viscoelastic using Kelvin–Voigt model. Considering the strain gradient theory for small scale effects, a new formulation of the SSDBT is developed through the Gurtin–Murdoch elasticity theory in which the effects of surface stress are incorporated. The surrounding elastic medium is described by a visco-Pasternak foundation model, which accounts for normal, transverse shear and damping loads. The van der Waals interactions between the adjacent walls of the nanotubes are taken into account. The size dependent motion equations and corresponding boundary conditions are derived based on the Hamilton's principle. The differential quadrature method in conjunction with Bolotin method is applied for obtaining the dynamic instability region. The detailed parametric study is conducted, focusing on the combined effects of the nonlocal parameter, magnetic field, visco-Pasternak foundation, Knudsen number, surface stress and fluid velocity on the dynamic instability of DWCNTs. The results depict that the surface stress effects on the dynamic instability of visco-DWCNTs are very significant. Numerical results of the present study are compared with available exact solutions in the literature. The results presented in this paper would be helpful in design and manufacturing of nano/micro mechanical systems in advanced biomechanics applications with magnetic field as a parametric controller.

Keywords Dynamic instability · Pulsating fluid · SSDBT · Bolotin method · Strain gradient theory

1 Introduction

Nanotechnology is one of the most important and development frontiers in the modern science. It was originally used to define any work done on the molecular scale, or one billionth of a meter (nm or 10^{-9} m). Carbon nanotubes (CNTs) with nano scale dimension have been well-known over the past 15 years. It was first discovered by Iijima (1991), when he was studying the synthesis of fullerenes using electric arc discharge technique. Some unrivaled properties of CNTs are, chemical and thermal stability, extremely high tensile strength and elasticity, and

A. Ghorbanpour Arani (✉) · R. Kolahchi · M. Mosayyebi · M. Jamali
Faculty of Mechanical Engineering, University of Kashan, Kashan, Iran
e-mail: aghorban@kashanu.ac.ir;
a_ghorbanpour@yahoo.com

A. Ghorbanpour Arani
Institute of Nanoscience & Nanotechnology, University of Kashan, Kashan, Iran

high conductivity. And some amazingly characteristics of CNTs are, a SWCNT can be up to 100 times stronger than steel with the same weight, the Young's Modulus of SWCNT is up to 1TPa, which is 5 times greater than steel (230 GPa) while the density is only 1.3 g/cm^3 and the thermal conductivity (2,000 W/ m K) is five times greater than that of copper (400 W/ m K) (Terrones 2003). These over mentioned properties has made CNTs one of the best candidate materials for uses in various industrial applications, in stance, sensors, actuators, fluid storage, composite materials, coatings and films, microelectronics, biotechnology, battery, solar cell and water filter (Lim et al. 2010).

At nanometer length scale, the properties of a material such as its melting point, its electronic and optical properties will change. Sometimes, the classical theory cannot describe some phenomena of the material at atomic level. However, the change of these properties offers new opportunities for technological and commercial development and applications. There are two types of theories applied to calculate the stress in materials, one of them supposes that the stress at a defined point depends on the strain at the same point so it is named local elasticity theory but another one is assumes that the stress at a point is a function of strains at all points in the continuum therefore it is called nonlocal elasticity theory. Hitherto some different nonlocal elasticity theories are defined such as, Eringen, couple stress, modified couple stress and strain gradient. The Eringen's nonlocal elasticity theory which was initiated by Eringen (1983), has been widely utilized to study the mechanical manners in the micron and nano scale structures. Strain gradient theory was proposed by Mindlin (1965) and improved by Lam et al. (2003). This theory includes three additional material length scale parameters for linear elastic isotropic materials. The MSGT has been used by many researchers in order to analyze size-dependent structures. For instance, Akgöz and Civalek (2013a, b) studied free vibration analysis of axially functionally graded tapered Bernoulli–Euler microbeams based on the modified couple stress theory. Ghorbanpour Arani et al. (2012) investigated nonlocal wave propagation in an embedded double-walled boron nitride nanotubes (DWBNNT) conveying fluid via strain gradient theory. Akgöz and Civalek (2013a, b) presented a size-dependent shear deformation beam model based on the strain gradient elasticity theory. Bending and vibration of functionally graded sinusoidal

micro beams based on the strain gradient elasticity theory were carried out by Lei et al. (2013a). Analysis of micro-sized beams for various boundary conditions based on the strain gradient elasticity theory was investigated by Akgöz and Civalek (2012). They demonstrated that the bending values obtained by these higher-order elasticity theories have a significant difference with those calculated by the classical elasticity theory.

For the time being, the mechanical behaviors of beams are being studied by applying various beam theories. Euler–Bernoulli theory (EBT) which applied for slender beams is one of these theories but this is utilized in situation that the shear deformation effect is negligible. However, effects of shear deformation and rotary inertia become more prominent and cannot be ignored for moderately thick beams and vibration responses on higher modes. For achieving the higher accuracy in dynamic behaviors of beam structures, several higher-order shear deformation theories are such as parabolic (third-order) beam theory of Levinson (1981) and Reddy (1984), trigonometric beam theory of Touratier (1991), hyperbolic beam theory of Soldatos (1992), exponential beam theory of Karama et al. (2003), and a general exponential beam theory of Aydogdu (2009). Li et al. (2014) performed the comparison of various shear deformation theories for free vibration of laminated composite beams with general lay-ups. Thai and Vo (2012) carried out the a nonlocal SSDBT with application to bending, buckling, and vibration of nano beams.

Dynamic behavior of nano-tubes conveying fluid is one of the great important problem with the most applications in targeted drug delivery systems. Hence, there are many remarkable investigations on the vibration and dynamic behaviors of nano-tubes conveying fluid. Khodami Maraghi et al. (2013) presented the nonlocal vibration and instability of embedded DWBNNT conveying viscose fluid. They indicated that increasing flow velocity decreases system stability. Wang (2011) carried out a modified nonlocal beam model for vibration and stability of nanotubes conveying fluid. Kiani (2013) studied the effects of small-scale parameter, inclination angle, speed and density of the fluid flow on the maximum dynamic amplitude factors of longitudinal and transverse displacements. Recently, it is shown by Kaviani and Mirdamadi (2013) that considering the small-size effects of the flow field on the dynamic characteristics of CNTs

conveying fluid is essential. They investigated wave propagation analysis of CNTs conveying fluid including slip boundary condition and strain/inertial gradient theory. Mirramezani et al. (2013) showed that based on their result, they could have developed an innovative model for one dimensional coupled vibrations of CNTs conveying fluid using slip velocity of the fluid flow on the CNT walls as well as utilizing size-dependent continuum theories to consider the size effects of nano-flow and nano-structure. However, it should be noted that all of the research mentioned above, have considered constant flow velocity. Actually, in many NEMS/MEMS, the flow inside the pipes and tubes becomes pulsative type due to power systems and alternative pressurized devices. Hence, dynamic analysis of CNTs conveying pulsating fluid is very significant and essential. The stability analysis of a SWCNT conveying pulsating and viscous fluid with nonlocal effect was investigated by Liang and Su (2013). They showed that decrease of nonlocal parameter and increase of viscous parameter both increases the fundamental frequency and critical flow velocity. Ghorbanpour Arani et al. (2014b) carried out the nonlocal surface piezoelasticity theory for dynamic stability of DWBNNTs conveying viscose fluid based on different theories. In this study DIR of EBT, Timoshenko beam theory (TBT) and cylindrical shell theory are compared to each other. Nonlinear dynamic instability of DWCNT under periodic excitation is reported by Fu et al. (2009) based on EBB theory. Results show that DWCNT can be considered as single column when the vdW forces are sufficiently strong, also the area of DIR could be reduced by stiffness medium and increment of the aspect ratio of nanotubes. Ghorbanpour Arani et al. (2013) presented the nonlocal TBT for dynamic stability of DWBNNTs conveying nano-flow.

In macro structures, the surface free energy can be neglected in comparison with the bulk energy while in nano scale structures, the surface stress should be taken into account due to the high surface to volume ratio. Malekzadeh and Shojaee (2013) presented the surface and nonlocal effects on the nonlinear free vibration of non-uniform nano beams. They found that increase of the amplitude ratio causes reduction of the surface effects. Gheshlaghi and Hasheminejad (2011) investigated the surface effects on nonlinear free vibration of nano beams.

In some applications of nano-engineering, the investigation on dynamic characteristic of CNTs under

magnetic field as a parametric controller is useful. Ghorbanpour Arani et al. (2014a) presented that the magnetic field is fundamentally an effective factor on increasing resonance frequency leading to stability of system. Wang et al. (2010) carried out the influences of longitudinal magnetic field on wave propagation in CNTs embedded in elastic matrix. The results obtained show that wave propagation in CNTs embedded in elastic matrix under longitudinal magnetic field appears in critical frequencies at which the velocity of wave propagation drops dramatically. Kiani (2014) investigated the vibration and instability of a SWCNT in a three-dimensional magnetic field. The obtained results reveal that the critical transverse magnetic field increases with the longitudinally induced magnetic field. Further, its value decreases as the effect of the small-scale parameter increases. He studied the effects of the longitudinal and transvers magnetic fields on the longitudinal and flexural frequencies.

Viscoelasticity is the property of materials that exhibit both viscous and elastic characteristics when undergoing deformation. Viscous materials resist shear flow and strain linearly with time when a stress is applied. Elastic materials strain when stretched and quickly return to their original state once the stress is removed. Lei et al. (2013c) presented the Dynamic characteristics of damped viscoelastic nonlocal EBT. Ghorbanpour Arani and Amir (2013) studied the electro-thermal vibration of visco-elastically coupled BNNT systems conveying fluid embedded on elastic foundation via strain gradient theory. Vibration of nonlocal Kelvin–Voigt viscoelastic damped Timoshenko beams is carried out by Lei et al. (2013b).

However, to date, no report has been found in the literature on dynamic stability of viscoelastic DWCNTs conveying pulsating fluid based on surface sinusoidal strain gradient theory. Motivated by these considerations, in order to improve optimum design of nanostructures, we aim to present a realistic model for dynamic instability of DWCNTs resting on visco-Pasternak medium by considering the viscoelastic property of the nanotubes. DWCNTs are placed in 2D magnetic field and modeled by SSDBT as well as strain gradient theory. Visco- DWCNTs are conveying pulsating fluid in which the effect of slip boundary condition is considered using Knudsen number correct factor. The Gurtin–Murdoch elasticity theory is applied for incorporation the surface stress effects. DQM is used in order to calculate the DIR of visco-

DWCNTs induced by pulsating fluid. To confirm the validity of the present research, the results are compared with those reported in the literature. The effects of the nonlocal parameter, magnetic field, visco-Pasternak foundation, Knudsen number, surface stress and fluid velocity on the dynamic instability of visco-DWCNTs are elucidated.

2 Strain gradient elasticity theory

Unlike the modified couple stress theory of Yang et al. (2002) the strain gradient elasticity theory proposed by Lam et al. (2003) introduces additional dilatation gradient tensor and the deviatoric stretch gradient tensor in addition to the symmetric rotation gradient tensor. The strain energy U in a deformed isotropic linear elastic material occupying region Ω is given by Lei et al. (2013a), Ghorbanpour Arani et al. (2012):

$$U = \frac{1}{2} \int_{\Omega} \left(\sigma_{ij}\varepsilon_{ij} + p_i\gamma_i + \tau_{ijk}^{(1)}\eta_{ijk}^{(1)} + m_{ij}\chi_{ij} \right) dV, \quad (1)$$

where ε_{ij}, γ_i, $\eta_{ijk}^{(1)}$, χ_{ij} represent the strain, the dilatation gradient, the deviatoric stretch gradient and the symmetric rotation gradient tensors, respectively, which are defined by:

$$\varepsilon_{ij} = \frac{1}{2}\left(\frac{\partial u_j}{\partial x_i} + \frac{\partial u_i}{\partial x_j} \right), \quad (2)$$

$$\gamma_i = \frac{\partial \varepsilon_{mm}}{\partial x_i}, \quad (3)$$

$$\begin{aligned}
\eta_{ijk}^{(1)} = &\frac{1}{3}\left(\frac{\partial \varepsilon_{jk}}{\partial x_i} + \frac{\partial \varepsilon_{ki}}{\partial x_j} + \frac{\partial \varepsilon_{ij}}{\partial x_k} \right) \\
&- \frac{1}{15}\left[\delta_{ij}\left(\frac{\partial \varepsilon_{mm}}{\partial x_k} + 2\frac{\partial \varepsilon_{mk}}{\partial x_m} \right) \right. \\
&+ \delta_{jk}\left(\frac{\partial \varepsilon_{mm}}{\partial x_i} + 2\frac{\partial \varepsilon_{mi}}{\partial x_m} \right) \\
&\left. + \delta_{ki}\left(\frac{\partial \varepsilon_{mm}}{\partial x_j} + 2\frac{\partial \varepsilon_{mj}}{\partial x_m} \right) \right],
\end{aligned} \quad (4)$$

$$\chi_{ij} = \frac{1}{2}\left(e_{ipq}\frac{\partial \varepsilon_{qj}}{\partial x_p} + e_{jpq}\frac{\partial \varepsilon_{qi}}{\partial x_p} \right), \quad (5)$$

where u_i, δ_{ij} and e_{ijk} are the displacement vector, the knocker delta and the alternate tensor, respectively. The classical stress tensor, σ_{ij}, the higher-order

stresses, p_i, $\tau_{ijk}^{(1)}$ and m_{ij} are given by:

$$\sigma_{ij} = E\delta_{ij}\varepsilon_{mm} + 2G\left(\varepsilon_{ij} - \frac{1}{3}\varepsilon_{mm}\delta_{ij} \right), \quad (6)$$

$$p_i = 2l_0^2 G\gamma_i, \quad (7)$$

$$\tau_{ijk}^{(1)} = 2l_1^2 G\eta_{ijk}^{(1)}, \quad (8)$$

$$m_{ij} = 2l_2^2 G\chi_{ij}, \quad (9)$$

where E and G are the bulk modulus and the shear modulus, respectively, (l_0, l_1, l_2) are independent material length scale parameters.

2.1 Surface effect theory

In nano structure such as nanotubes and nano plates, the ratio of surface to volume is high, therefore the surface effect should be considered. The classical constitutive relation of the surface can be calculated as given by Gurtin and Murdoch (1975, 1978). Hence the classical stress tensor related to the surface, σ^s, the higher-order stresses of surface, p_i^s, $\tau_{ijk}^{s(1)}$ and m_{ij}^s can be expressed as Malekzadeh and Shojaee (2013):

$$\sigma^s = \tau_s + E_s\varepsilon_x^s, \quad (10)$$

$$p_i^s = 2l_0^2 G_s\gamma_i^s, \quad (11)$$

$$\tau_{ijk}^{s(1)} = 2l_1^2 G_s\eta_{ijk}^{s(1)}, \quad (12)$$

$$m_{ij}^s = 2l_2^2 G_s\chi_{ij}^s, \quad (13)$$

where τ_s, E_s and G_s, the residual surface tension in the axial direction, the surface elastic modulus and surface shear modulus.

2.2 Viscoelastic SSDBT

A schematic of visco-DWCNT conveying pulsating fluid embedded in visco Pasternak foundation under 2D magnetic field is shown in Fig. 1. The displacement fields of DWCNTs based on SSDBT can be described as Simsek and Reddy (2013):

$$u_x(x, z, t) = u(x, t) - z\frac{\partial w(x, t)}{\partial x} + \Phi(z)\varphi(x, t), \quad (14a)$$
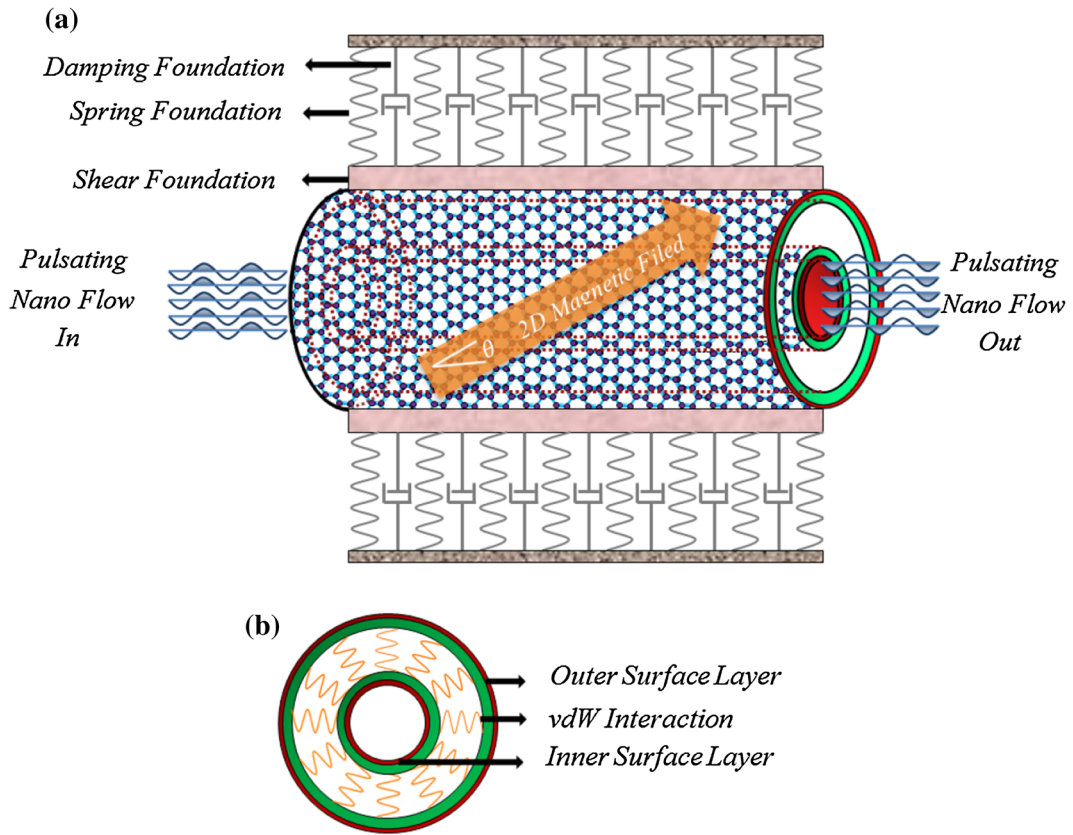
$$u_y(x, z, t) = 0, \quad (14b)$$

Fig. 1 **a** Schematic of DWCNT conveying pulsating fluid embedded in visco Pasternak foundation under 2D magnetic field. **b** Cross section of the nanotube

$$u_z(x, z, t) = w(x, t), \tag{14c}$$

$$\varphi(x, t) = \frac{\partial w(x, t)}{\partial x} - \phi(x, t), \tag{14d}$$

$$\Phi(z) = \frac{h}{\pi} \sin\left(\frac{\pi z}{h}\right), \tag{14e}$$

where u and w are the axial and the transverse displacement of any point on the neutral axis, t denotes time φ and ϕ are the transverse shear strain of any point on the neutral axis and the total bending rotation of the cross-sections at any point on the neutral axis. $\Phi(z)$ is a function of z, that characterizes the transverse shear and stress distribution along the thickness of the beam.

All materials exhibit some viscoelastic response. According to Kelvin–Voigt (Lei et al. 2013b) at real life, nano structure mechanical properties depend on the time variation. This model represents, as the stress is released, the material gradually relaxes to its undeformed state. By considering this model, Young's modulus, E, shear modulus, G, Young's modulus of surface, E_s and shear modulus of surface, G_s, are as follows:

$$E^* = E\left(1 + g\frac{\partial}{\partial t}\right), \tag{15a}$$

$$E_s^* = E_s\left(1 + g\frac{\partial}{\partial t}\right), \tag{15b}$$

$$G^* = G\left(1 + g\frac{\partial}{\partial t}\right), \tag{15c}$$

$$G_s^* = G_s\left(1 + g\frac{\partial}{\partial t}\right). \tag{15d}$$

3 Formulation

The energy method is applied to derive equations of motion, in this study. Total potential energy Π, is given by:

$$\Pi = U_s - (K + W), \tag{16}$$

U_s, K and W indicate total strain energy, total kinetic energy and the total external work in DWCNTs system.

Hamilton's principle is used to derive the motion equations of embedded DWCNTs conveying pulsating viscose fluid as follows:

$$\int_{t_0}^{t_1} \left[\delta U_s - \left(\delta K_{nanotube} + \delta W_{fluid} + \delta W_{lorentz} \right. \right. \tag{17}$$
$$\left. \left. + \delta W_{visco-pasternak} + \delta W_{vdw} \right) \right] = 0.$$

3.1 Strain gradient theory and surface effect

The total strain energy is come to result by combining the strain gradient theory and the surface effect theory as follows:

$$U_s = U_s^b + U_s^s = \frac{1}{2} \int_\Omega \left(\sigma_{ij}\varepsilon_{ij} + p_i\gamma_i + \tau_{ijk}^{(1)}\eta_{ijk}^{(1)} + m_{ij}\chi_{ij} \right) dAdx$$
$$+ \frac{1}{2} \int_\Omega \left(\sigma_{ij}^s\varepsilon_{ij}^s + p_i^s\gamma_i^s\tau_{ijk}^{s(1)}\eta_{ijk}^{s(1)} + m_{ij}^s\chi_{ij}^s \right) dSdx, \tag{18}$$

Here U_s^b is the strain energy of bulk and U_s^s is the strain energy of surface. Total strain energy is came into result step by step as follows.

By substituting Eq. (14) into Eq. (2), the strain is

$$\begin{pmatrix} \varepsilon_x \\ \varepsilon_x^s \end{pmatrix} = \begin{pmatrix} 1 & -1 & 1 & -1 \\ 1 & -1 & 1 & -1 \end{pmatrix} \begin{pmatrix} \dfrac{\partial u}{\partial x} \\ z\dfrac{\partial^2 w}{\partial x^2} \\ \dfrac{h}{\pi}\dfrac{\partial^2 w}{\partial x^2} f^{(\sin)} \\ \dfrac{h}{\pi}\dfrac{\partial \phi}{\partial x} f^{(\sin)} \end{pmatrix}, \tag{19a}$$

$$\begin{pmatrix} \varepsilon_{xz} \\ \varepsilon_{xz}^s \end{pmatrix} = \begin{pmatrix} \dfrac{1}{2} & -\dfrac{1}{2} \\ 0 & 0 \end{pmatrix} \begin{pmatrix} \dfrac{\partial w}{\partial x}f^{(\cos)} \\ \phi f^{(\cos)} \end{pmatrix}, \tag{19b}$$

where

$$f^{(\sin)} = \sin\left(\frac{\pi z}{h}\right), f^{(\cos)} = \cos\left(\frac{\pi z}{h}\right). \tag{19c}$$

And from Eqs. (19) and (3), it results that

$$\begin{pmatrix} \gamma_x \\ \gamma_x^s \end{pmatrix} = \begin{pmatrix} 1 & -1 & 1 & -1 \\ 1 & -1 & 1 & -1 \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 u}{\partial x^2} \\ z\dfrac{\partial^3 w}{\partial x^3} \\ \dfrac{h}{\pi}\dfrac{\partial^3 w}{\partial x^3} f^{(\sin)} \\ \dfrac{h}{\pi}\dfrac{\partial^2 \phi}{\partial x^2} f^{(\sin)} \end{pmatrix}, \tag{20a}$$

$$\begin{pmatrix} \gamma_z \\ \gamma_z^s \end{pmatrix} = \begin{pmatrix} -1 & 1 & -1 \\ -1 & 1 & -1 \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 w}{\partial x^2} \\ \dfrac{\partial^2 w}{\partial x^2} f^{(\cos)} \\ \dfrac{\partial \phi}{\partial x} f^{(\cos)} \end{pmatrix}, \tag{20b}$$

$$\gamma_y = \gamma_y^s = 0. \tag{20c}$$

By applying Eqs. (19) and (4) gives

$$\begin{pmatrix} \eta_{111}^{(1)} \\ \eta_{111}^{s(1)} \end{pmatrix} = \begin{pmatrix} \dfrac{2}{5} & -\dfrac{2}{5} & \dfrac{2}{5} & -\dfrac{2}{5} & \dfrac{1}{5} & -\dfrac{1}{5} \\ \dfrac{2}{5} & -\dfrac{2}{5} & \dfrac{2}{5} & -\dfrac{2}{5} & 0 & 0 \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 u}{\partial x^2} \\ z\dfrac{\partial^3 w}{\partial x^3} \\ \dfrac{h}{\pi}\dfrac{\partial^3 w}{\partial x^3} f^{(\sin)} \\ \dfrac{h}{\pi}\dfrac{\partial^2 \phi}{\partial x^2} f^{(\sin)} \\ \dfrac{\pi}{h}\dfrac{\partial w}{\partial x} f^{(\sin)} \\ \dfrac{\pi}{h}\phi f^{(\sin)} \end{pmatrix}, \tag{21a}$$

$$\begin{pmatrix} \eta_{113}^{(1)} \\ \eta_{113}^{s(1)} \end{pmatrix} = \begin{pmatrix} \eta_{131}^{(1)} \\ \eta_{131}^{s(1)} \end{pmatrix} = \begin{pmatrix} \eta_{311}^{(1)} \\ \eta_{311}^{s(1)} \end{pmatrix}$$
$$= \begin{pmatrix} -\dfrac{4}{15} & \dfrac{8}{15} & -\dfrac{8}{15} \\ -\dfrac{4}{15} & \dfrac{4}{15} & -\dfrac{4}{15} \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 w}{\partial x^2} \\ \dfrac{\partial^2 w}{\partial x^2} f^{(\cos)} \\ \dfrac{\partial \phi}{\partial x} f^{(\cos)} \end{pmatrix}, \tag{21b}$$

$$\begin{pmatrix} \eta_{122}^{(1)} \\ \eta_{122}^{s(1)} \end{pmatrix} = \begin{pmatrix} \eta_{221}^{(1)} \\ \eta_{221}^{s(1)} \end{pmatrix} = \begin{pmatrix} \eta_{212}^{(1)} \\ \eta_{212}^{s(1)} \end{pmatrix}$$

$$= \begin{pmatrix} -\dfrac{1}{5} & \dfrac{1}{5} & -\dfrac{1}{5} & \dfrac{1}{5} & \dfrac{1}{15} & -\dfrac{1}{15} \\ -\dfrac{1}{5} & \dfrac{1}{5} & -\dfrac{1}{5} & \dfrac{1}{5} & 0 & 0 \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 u}{\partial x^2} \\ z\dfrac{\partial^3 w}{\partial x^3} \\ \dfrac{h}{\pi}\dfrac{\partial^3 w}{\partial x^3} f^{(\sin)} \\ \dfrac{h}{\pi}\dfrac{\partial^2 \phi}{\partial x^2} f^{(\sin)} \\ \dfrac{\pi}{h}\dfrac{\partial w}{\partial x} f^{(\sin)} \\ \dfrac{\pi}{h}\phi f^{(\sin)} \end{pmatrix},$$

(21c)

$$\begin{pmatrix} \eta_{133}^{(1)} \\ \eta_{133}^{s(1)} \end{pmatrix} = \begin{pmatrix} \eta_{331}^{(1)} \\ \eta_{331}^{s(1)} \end{pmatrix} = \begin{pmatrix} \eta_{313}^{(1)} \\ \eta_{313}^{s(1)} \end{pmatrix}$$

$$= \begin{pmatrix} -\dfrac{1}{5} & \dfrac{1}{5} & -\dfrac{1}{5} & \dfrac{1}{5} & -\dfrac{4}{15} & \dfrac{4}{15} \\ -\dfrac{1}{5} & \dfrac{1}{5} & -\dfrac{1}{5} & \dfrac{1}{5} & 0 & 0 \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 u}{\partial x^2} \\ z\dfrac{\partial^3 w}{\partial x^3} \\ \dfrac{h}{\pi}\dfrac{\partial^3 w}{\partial x^3} f^{(\sin)} \\ \dfrac{h}{\pi}\dfrac{\partial^2 \phi}{\partial x^2} f^{(\sin)} \\ \dfrac{\pi}{h}\dfrac{\partial w}{\partial x} f^{(\sin)} \\ \dfrac{\pi}{h}\phi f^{(\sin)} \end{pmatrix},$$

(21d)

$$\begin{pmatrix} \eta_{223}^{(1)} \\ \eta_{223}^{s(1)} \end{pmatrix} = \begin{pmatrix} \eta_{232}^{(1)} \\ \eta_{232}^{s(1)} \end{pmatrix} = \begin{pmatrix} \eta_{322}^{(1)} \\ \eta_{322}^{s(1)} \end{pmatrix}$$

$$= \begin{pmatrix} \dfrac{1}{15} & -\dfrac{2}{15} & \dfrac{2}{15} \\ \dfrac{1}{15} & -\dfrac{1}{15} & \dfrac{1}{15} \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 w}{\partial x^2} \\ \dfrac{\partial^2 w}{\partial x^2} f^{(\cos)} \\ \dfrac{\partial \phi}{\partial x} f^{(\cos)} \end{pmatrix},$$

(21e)

$$\begin{pmatrix} \eta_{333}^{(1)} \\ \eta_{333}^{s(1)} \end{pmatrix} = \begin{pmatrix} \eta_{333}^{(1)} \\ \eta_{333}^{s(1)} \end{pmatrix} = \begin{pmatrix} \eta_{333}^{(1)} \\ \eta_{333}^{s(1)} \end{pmatrix}$$

$$= \begin{pmatrix} \dfrac{1}{5} & -\dfrac{2}{5} & \dfrac{2}{5} \\ \dfrac{1}{5} & -\dfrac{1}{5} & \dfrac{1}{5} \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 w}{\partial x^2} \\ \dfrac{\partial^2 w}{\partial x^2} f^{(\cos)} \\ \dfrac{\partial \phi}{\partial x} f^{(\cos)} \end{pmatrix}.$$

(21f)

And by substituting Eq. (19) into Eq. (5) gives the non-zero symmetric rotation gradient tensor as

$$\begin{pmatrix} \chi_{xy} \\ \chi_{xy}^s \end{pmatrix} = \begin{pmatrix} \chi_{yx} \\ \chi_{yx}^s \end{pmatrix}$$

$$= \begin{pmatrix} -\dfrac{1}{2} & \dfrac{1}{4} & -\dfrac{1}{4} \\ -\dfrac{1}{2} & \dfrac{1}{2} & -\dfrac{1}{2} \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 w}{\partial x^2} \\ \dfrac{\partial^2 w}{\partial x^2} f^{(\cos)} \\ \dfrac{\partial \phi}{\partial x} f^{(\cos)} \end{pmatrix},$$

(22a)

$$\begin{pmatrix} \chi_{yz} \\ \chi_{yz}^s \end{pmatrix} = \begin{pmatrix} \chi_{zy} \\ \chi_{zy}^s \end{pmatrix} = \begin{pmatrix} -\dfrac{1}{4} & +\dfrac{1}{4} \\ 0 & 0 \end{pmatrix} \begin{pmatrix} \dfrac{\pi}{h}\dfrac{\partial w}{\partial x} f^{(\sin)} \\ \dfrac{\pi}{h}\phi f^{(\sin)} \end{pmatrix}.$$

(22b)

By using Eqs. (19), (10), (6) and (15) the stresses are

$$\begin{pmatrix} \sigma_x \\ \sigma_x^s \end{pmatrix} = \begin{pmatrix} E^* & E_s^* \end{pmatrix} \begin{pmatrix} 1 & -1 & 1 & -1 \\ 1 & -1 & 1 & -1 \end{pmatrix}$$

$$\begin{pmatrix} \dfrac{\partial u}{\partial x} \\ z\dfrac{\partial^2 w}{\partial x^2} \\ \dfrac{h}{\pi}\dfrac{\partial^2 w}{\partial x^2} f^{(\sin)} \\ \dfrac{h}{\pi}\dfrac{\partial \phi}{\partial x} f^{(\sin)} \end{pmatrix} + \begin{pmatrix} 0 \\ \tau_s \end{pmatrix},$$

(23a)

$$\begin{pmatrix} \sigma_{xz} \\ \sigma_{xz}^s \end{pmatrix} = \begin{pmatrix} G^* & G_s^* \end{pmatrix} \begin{pmatrix} 1 & -1 \\ 0 & 0 \end{pmatrix} \begin{pmatrix} \dfrac{\partial w}{\partial x} f^{(\cos)} \\ \phi f^{(\cos)} \end{pmatrix}.$$

(23b)

By considering Eq. (15) and substituting Eq. (20) into Eq. (11) and (7)

$$
\begin{pmatrix} p_x \\ P_x^s \end{pmatrix} = 2l_0^2 (G^* \quad G_s^*) \begin{pmatrix} 1 & -1 & 1 & -1 \\ 1 & -1 & 1 & -1 \end{pmatrix}
$$

$$
\begin{pmatrix} \dfrac{\partial^2 u}{\partial x^2} \\[2mm] z\dfrac{\partial^3 w}{\partial x^3} \\[2mm] \dfrac{h}{\pi}\dfrac{\partial^3 w}{\partial x^3} f^{(\sin)} \\[2mm] \dfrac{h}{\pi}\dfrac{\partial^2 \phi}{\partial x^2} f^{(\sin)} \end{pmatrix}, \tag{24a}
$$

$$
\begin{pmatrix} p_z \\ P_z^s \end{pmatrix} = 2l_0^2 (G^* \quad G_s^*) \begin{pmatrix} -1 & 1 & -1 \\ -1 & 1 & -1 \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 w}{\partial x^2} \\[2mm] \dfrac{\partial^2 w}{\partial x^2} f^{(\cos)} \\[2mm] \dfrac{\partial \phi}{\partial x} f^{(\cos)} \end{pmatrix}. \tag{24b}
$$

From Eqs. (21), (12), (8) and (15), the non-zero higher-order $\tau_{ijk}^{(1)}$, $\tau_{ijk}^{s(1)}$ are

$$
\begin{pmatrix} \tau_{111}^{(1)} \\ \tau_{111}^{s(1)} \end{pmatrix} = 2l_1^2 (G^* \quad G_s^*) \begin{pmatrix} \dfrac{2}{5} & -\dfrac{2}{5} & \dfrac{2}{5} & -\dfrac{2}{5} & \dfrac{1}{5} & -\dfrac{1}{5} \\ \dfrac{2}{5} & -\dfrac{2}{5} & \dfrac{2}{5} & -\dfrac{2}{5} & 0 & 0 \end{pmatrix}
$$

$$
\begin{pmatrix} \dfrac{\partial^2 u}{\partial x^2} \\[2mm] z\dfrac{\partial^3 w}{\partial x^3} \\[2mm] \dfrac{h}{\pi}\dfrac{\partial^3 w}{\partial x^3} f^{(\sin)} \\[2mm] \dfrac{h}{\pi}\dfrac{\partial^2 \phi}{\partial x^2} f^{(\sin)} \\[2mm] \dfrac{\pi}{h}\dfrac{\partial w}{\partial x} f^{(\sin)} \\[2mm] \dfrac{\pi}{h}\phi f^{(\sin)} \end{pmatrix}, \tag{25a}
$$

$$
\begin{pmatrix} \tau_{113}^{(1)} \\ \tau_{113}^{s(1)} \end{pmatrix} = \begin{pmatrix} \tau_{131}^{(1)} \\ \tau_{131}^{s(1)} \end{pmatrix} = \begin{pmatrix} \tau_{311}^{(1)} \\ \tau_{311}^{s(1)} \end{pmatrix}
$$

$$
= 2l_1^2 (G^* \quad G_s^*) \begin{pmatrix} -\dfrac{4}{15} & \dfrac{8}{15} & -\dfrac{8}{15} \\ -\dfrac{4}{15} & \dfrac{4}{15} & -\dfrac{4}{15} \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 w}{\partial x^2} \\[2mm] \dfrac{\partial^2 w}{\partial x^2} f^{(\cos)} \\[2mm] \dfrac{\partial \phi}{\partial x} f^{(\cos)} \end{pmatrix}, \tag{25b}
$$

$$
\begin{pmatrix} \tau_{122}^{(1)} \\ \tau_{122}^{s(1)} \end{pmatrix} = \begin{pmatrix} \tau_{221}^{(1)} \\ \tau_{221}^{s(1)} \end{pmatrix} = \begin{pmatrix} \tau_{212}^{(1)} \\ \tau_{212}^{s(1)} \end{pmatrix}
$$

$$
= 2l_1^2 (G^* \quad G_s^*) \begin{pmatrix} -\dfrac{1}{5} & \dfrac{1}{5} & -\dfrac{1}{5} & \dfrac{1}{5} & \dfrac{1}{15} & -\dfrac{1}{15} \\ -\dfrac{1}{5} & \dfrac{1}{5} & -\dfrac{1}{5} & \dfrac{1}{5} & 0 & 0 \end{pmatrix}
$$

$$
\begin{pmatrix} \dfrac{\partial^2 u}{\partial x^2} \\[2mm] z\dfrac{\partial^3 w}{\partial x^3} \\[2mm] \dfrac{h}{\pi}\dfrac{\partial^3 w}{\partial x^3} f^{(\sin)} \\[2mm] \dfrac{h}{\pi}\dfrac{\partial^2 \phi}{\partial x^2} f^{(\sin)} \\[2mm] \dfrac{\pi}{h}\dfrac{\partial w}{\partial x} f^{(\sin)} \\[2mm] \dfrac{\pi}{h}\phi f^{(\sin)} \end{pmatrix}, \tag{25c}
$$

$$
\begin{pmatrix} \tau_{133}^{(1)} \\ \tau_{133}^{s(1)} \end{pmatrix} = \begin{pmatrix} \tau_{331}^{(1)} \\ \tau_{331}^{s(1)} \end{pmatrix} = \begin{pmatrix} \tau_{313}^{(1)} \\ \tau_{313}^{s(1)} \end{pmatrix}
$$

$$
= 2l_1^2 (G^* \quad G_s^*) \begin{pmatrix} -\dfrac{1}{5} & \dfrac{1}{5} & -\dfrac{1}{5} & \dfrac{1}{5} & -\dfrac{4}{15} & \dfrac{4}{15} \\ -\dfrac{1}{5} & \dfrac{1}{5} & -\dfrac{1}{5} & \dfrac{1}{5} & 0 & 0 \end{pmatrix}
$$

$$
\begin{pmatrix} \dfrac{\partial^2 u}{\partial x^2} \\[2mm] z\dfrac{\partial^3 w}{\partial x^3} \\[2mm] \dfrac{h}{\pi}\dfrac{\partial^3 w}{\partial x^3} f^{(\sin)} \\[2mm] \dfrac{h}{\pi}\dfrac{\partial^2 \phi}{\partial x^2} f^{(\sin)} \\[2mm] \dfrac{\pi}{h}\dfrac{\partial w}{\partial x} f^{(\sin)} \\[2mm] \dfrac{\pi}{h}\phi f^{(\sin)} \end{pmatrix}, \tag{25d}
$$

$$
\begin{pmatrix} \tau_{223}^{(1)} \\ \tau_{223}^{s(1)} \end{pmatrix} = \begin{pmatrix} \tau_{232}^{(1)} \\ \tau_{232}^{s(1)} \end{pmatrix} = \begin{pmatrix} \tau_{322}^{(1)} \\ \tau_{322}^{s(1)} \end{pmatrix} = 2l_1^2 (G^* \quad G_s^*)
$$

$$
\begin{pmatrix} \dfrac{1}{15} & -\dfrac{2}{15} & \dfrac{2}{15} \\ \dfrac{1}{15} & -\dfrac{1}{15} & \dfrac{1}{15} \end{pmatrix} \begin{pmatrix} \dfrac{\partial^2 w}{\partial x^2} \\[2mm] \dfrac{\partial^2 w}{\partial x^2} f^{(\cos)} \\[2mm] \dfrac{\partial \phi}{\partial x} f^{(\cos)} \end{pmatrix}, \tag{25e}
$$

$$\begin{pmatrix} \tau_{333}^{(1)} \\ \tau_{333}^{s(1)} \end{pmatrix} = \begin{pmatrix} \tau_{333}^{(1)} \\ \tau_{333}^{s(1)} \end{pmatrix} = \begin{pmatrix} \tau_{333}^{(1)} \\ \tau_{333}^{s(1)} \end{pmatrix}$$

$$= 2l_1^2 (G^* \quad G_s^*) \begin{pmatrix} \frac{1}{5} & -\frac{2}{5} & \frac{2}{5} \\ \frac{1}{5} & -\frac{1}{5} & \frac{1}{5} \end{pmatrix} \begin{pmatrix} \frac{\partial^2 w}{\partial x^2} \\ \frac{\partial^2 w}{\partial x^2} f^{(\cos)} \\ \frac{\partial \phi}{\partial x} f^{(\cos)} \end{pmatrix}. \tag{25f}$$

By utilizing Eq. (15) and inserting Eq. (22) into (13) and (9) the higher-order stresses m_{ij}, m_{ij}^s are

$$\begin{pmatrix} m_{xy} \\ m_{xy}^s \end{pmatrix} = \begin{pmatrix} m_{yx} \\ m_{yx}^s \end{pmatrix}$$

$$= 2l_2^2 (G^* \quad G_s^*) \begin{pmatrix} -\frac{1}{2} & \frac{1}{4} & -\frac{1}{4} \\ -\frac{1}{2} & \frac{1}{2} & -\frac{1}{2} \end{pmatrix} \begin{pmatrix} \frac{\partial^2 w}{\partial x^2} \\ \frac{\partial^2 w}{\partial x^2} f^{(\cos)} \\ \frac{\partial \phi}{\partial x} f^{(\cos)} \end{pmatrix}, \tag{26a}$$

$$\begin{pmatrix} m_{yz} \\ m_{yz}^s \end{pmatrix} = \begin{pmatrix} m_{zy} \\ m_{zy}^s \end{pmatrix}$$

$$= 2l_2^2 (G^* \quad G_s^*) \begin{pmatrix} -\frac{1}{4} & +\frac{1}{4} \\ 0 & 0 \end{pmatrix} \begin{pmatrix} \frac{\pi}{h} \frac{\partial w}{\partial x} f^{(\sin)} \\ \frac{\pi}{h} \phi f^{(\sin)} \end{pmatrix}. \tag{26b}$$

Substituting Eqs. (19)–(26) into Eq. (18) for each layer, leads to

$$\begin{aligned} U_s = \frac{1}{2} \int_0^L \Bigg\{ & B1 \frac{\partial^2 w_i}{\partial x^2} + B2 \left(\frac{\partial w_i}{\partial x} \right)^2 + B3_i \left(\frac{\partial^2 w_i}{\partial x^2} \right)^2 \\ & + B4 \left(\frac{\partial^3 w_i}{\partial x^3} \right)^2 + B5 \frac{\partial^3 w_i}{\partial x^3} \frac{\partial w_i}{\partial x} + B6 \frac{\partial^2 u_i}{\partial x^2} \frac{\partial w_i}{\partial x} \\ & + B7 \frac{\partial u_i}{\partial x} \frac{\partial^2 w_i}{\partial x^2} + B8 \frac{\partial^3 w_i}{\partial x^3} \frac{\partial^2 u_i}{\partial x^2} + B9 \phi_i \frac{\partial w_i}{\partial x} \\ & + B10 \frac{\partial^2 w_i}{\partial x^2} \frac{\partial \phi_i}{\partial x} + B11 \phi_i \frac{\partial^3 w_i}{\partial x^3} + B12 \frac{\partial^3 w_i}{\partial x^3} \frac{\partial^2 \phi_i}{\partial x^2} \\ & + B13_i \frac{\partial u_i}{\partial x} + B14_i \left(\frac{\partial u_i}{\partial x} \right)^2 + B15_i \left(\frac{\partial^2 u_i}{\partial x^2} \right)^2 \end{aligned}$$

$$\begin{aligned} & + B16 \frac{\partial \phi_i}{\partial x} \frac{\partial u_i}{\partial x} + B17 \phi_i \frac{\partial^2 u_i}{\partial x^2} + B18 \frac{\partial^2 u_i}{\partial x^2} \frac{\partial^2 \phi_i}{\partial x^2} \\ & + B19(\phi_i)^2 + b20 \frac{\partial \phi_i}{\partial x} + B21 \left(\frac{\partial \phi_i}{\partial x} \right)^2 \\ & + B22 \left(\frac{\partial^2 \phi_i}{\partial x^2} \right)^2 + B23 \phi_i \frac{\partial^2 \phi_i}{\partial x^2} + B24 \frac{\partial w_i}{\partial x} \frac{\partial^2 \phi_i}{\partial x^2} \Bigg\} dx, \end{aligned} \tag{27}$$

Subscript i denotes the number of nanotube where $i = 1, 2$ demonstrate inner and outer nanotubes, and B_{ij} are defined in Appendix 1.

The total kinetic energy of nanotubes can be expressed as:

$$k_{tube} = \frac{1}{2} \rho_t \int_0^L \left\{ \int_{A_i} \left[\left(\frac{\partial u_{xi}}{\partial t} \right)^2 + \left(\frac{\partial u_{zi}}{\partial t} \right)^2 \right] dA_i \right\} dx. \tag{28}$$

3.2 Virtual work of pulsating nano-flow

The well-known Navier–Stokes equations are stated as follows Mirramezani et al. (2013):

$$\rho \frac{D\overrightarrow{V}}{Dt} = -\nabla \overrightarrow{P} + \mu \nabla^2 \overrightarrow{V} + \overrightarrow{F}_{body}, \tag{29}$$

where $\frac{D}{Dt}$ is the material or total derivative and $\overrightarrow{V}$ is the flow velocity, $\overrightarrow{P}$ and μ are, respectively, the pressure and the viscosity of the flowing fluid, ρ is the mass density of the internal fluid, and $\overrightarrow{F}_{body}$ represents body forces.

According to the reference (Mirramezani et al. 2013), the viscosity parameter could not appear in the fluid–structure interaction (FSI) equation. So the force exerted due to the fluid flow on the nanotube can be obtained as follows:

$$\begin{aligned} F_f &= A_f \frac{\partial p_r}{\partial r} - \mu_0 A_f \left[\frac{\partial^3 W}{\partial x^2 \partial t} + V_f \frac{\partial^3 W}{\partial x^3} \right] \\ &= -\rho A_f \left[\frac{\partial^2 W}{\partial t^2} + 2V_f \frac{\partial^2 W}{\partial x \partial t} + V_f^2 \frac{\partial^2 W}{\partial x^2} \right], \end{aligned} \tag{30}$$

where F_f is the exerted force by fluid to the nanotube. A_f, ρ, V_f and μ denote the cross sectional area of the internal fluid, the fluid density, the velocity of the fluid flow in the longitudinal direction on the CNT wall and the viscosity of the flowing fluid, respectively.

For CNTs conveying fluid, the Kn may be larger than 10-2; consequently, the assumption of no-slip boundary condition is not true and the fluid slip velocity should be modified. The slip velocity is presented as follows Mirramezani et al. (2013):

$$V_{avg,slip} = \frac{\gamma}{1+\gamma} V_{avg,(no-slip)}, \tag{31a}$$

where

$$\gamma = 4\left(\frac{2-\sigma_v}{\sigma_v}\right)\left(\frac{Kn}{1+Kn}\right), \tag{31b}$$

Here σ_v is tangential moment accommodation coefficient and is considered to be 0.7 for most practical purpose.

The case of pulsating internal flow is assumed harmonically fluctuating, as follows Liang and Su (2013):

$$V_f = V_{avg,(no-slip)} = V_0(1 + \alpha\cos(\omega t)), \tag{32}$$

where V_0 is the mean flow velocity, α is the amplitude of the harmonic fluctuation (assumed small) and ω its frequency.

$$\overrightarrow{J} = \nabla \times \overrightarrow{h}, \tag{34a}$$

$$\nabla \times e = -\eta\left(\frac{\partial h}{\partial t}\right), \tag{34b}$$

$$\nabla \cdot \overrightarrow{h} = 0, \tag{34c}$$

$$e = -\eta\left(\frac{\partial D}{\partial t} \times \overrightarrow{H}\right), \tag{34d}$$

$$\overrightarrow{h} = \nabla \times (\overrightarrow{D} \times \overrightarrow{H}), \tag{34e}$$

where, $\overrightarrow{J}$, e, η and h represent the current density, strength vectors of electric field, the magnetic field permeability and disturbing vectors of magnetic field respectively. $\overrightarrow{D} = (u_x, u_y, u_z)$ is the displacement vector and magnetic field vector is:

$$\overrightarrow{H} = \begin{pmatrix} H_x \\ 0 \\ H_z \end{pmatrix} = \begin{pmatrix} H\cos(\theta) \\ 0 \\ H\sin(\theta) \end{pmatrix}. \tag{35}$$

By using Eq. (34) $\overrightarrow{h}$ and $\overrightarrow{J}$ are describe:

$$\overrightarrow{h} = \nabla \times \left(D \times \overrightarrow{H}\right)$$
$$= \begin{pmatrix} -H_z\dfrac{\partial w_i}{\partial x} + H_z\cos\left(\dfrac{\pi z}{h}\right)\dfrac{\partial w_i}{\partial x} - H_z\cos\left(\dfrac{\pi z}{h}\right)\phi_i \\ 0 \\ H_x\dfrac{\partial w_i}{\partial x} - H_z\dfrac{\partial u_i}{\partial x} + H_z z\dfrac{\partial^2 w_i}{\partial x^2} - H_z\dfrac{h}{\pi}\sin\left(\dfrac{\pi z}{h}\right)\dfrac{\partial^2 w_i}{\partial x^2} + H_z\dfrac{h}{\pi}\sin\left(\dfrac{\pi z}{h}\right)\dfrac{\partial\phi_i}{\partial x} \end{pmatrix}, \tag{36}$$

The total virtual work of pulsating nano-flow is

$$W_f = \int_0^l F_f w_1 dx$$
$$= \int_0^l \left\{ -\rho A_f\left[\frac{\partial^2 w_1}{\partial t^2} + 2U_f\frac{\partial^2 w_1}{\partial x\partial t} + U_f^2\frac{\partial^2 w_1}{\partial x^2}\right]\right\} w_1 dx. \tag{33}$$

It is noticed that $U_f = \frac{\gamma}{1+\gamma} \times V_f$ in the governing equations.

3.3 Maxwell's equation

Maxwell's equation are given by Wang et al. (2010):

$$\overrightarrow{J} = \nabla \times \overrightarrow{h} = \left(-H_z\frac{\pi}{h}\sin\left(\frac{\pi z}{h}\right)\frac{\partial w_i}{\partial x} + H_z\frac{\pi}{h}\sin\left(\frac{\pi z}{h}\right)\phi_i\right.$$
$$\left. -H_x\frac{\partial^2 w_i}{\partial x^2} + H_z\frac{\partial^2 u_i}{\partial x^2} - H_z z\frac{\partial^3 w_i}{\partial x^3} + H_z\frac{h}{\pi}\sin\left(\frac{\pi z}{h}\right)\frac{\partial^3 w_i}{\partial x^3}\right)j.$$
$$-H_z\frac{h}{\pi}\sin\left(\frac{\pi z}{h}\right)\frac{\partial^2\phi_i}{\partial x^2} \tag{37}$$

The Lorentz force in three directions is:

$$f = (f_x, f_y, f_z) = \eta\left(\overrightarrow{J} \times \overrightarrow{H}\right). \tag{38}$$

Introducing Eq. (37) to Eq. (38) and utilizing Eq. (35), the Lorentzian forces are obtained as:

$$f_x = -\eta H_z^2 \frac{\pi}{h} \sin\left(\frac{\pi z}{h}\right) \frac{\partial w_i}{\partial x} + \eta H_z^2 \frac{\pi}{h} \sin\left(\frac{\pi z}{h}\right) \phi_i$$
$$- \eta H_z H_x \frac{\partial^2 w_i}{\partial x^2} + \eta H_z^2 \frac{\partial^2 u_i}{\partial x^2} - \eta H_z^2 z \frac{\partial^3 w_i}{\partial x^3}$$
$$+ \eta H_z^2 \frac{h}{\pi} \sin\left(\frac{\pi z}{h}\right) \frac{\partial^3 w_i}{\partial x^3} - \eta H_z^2 \frac{h}{\pi} \sin\left(\frac{\pi z}{h}\right) \frac{\partial^2 \phi_i}{\partial x^2},$$
$$\text{(39a)}$$

$$f_y = 0, \tag{39b}$$

$$f_z = \eta H_x H_z \frac{\pi}{h} \sin\left(\frac{\pi z}{h}\right) \frac{\partial w_i}{\partial x} - \eta H_x H_z \frac{\pi}{h} \sin\left(\frac{\pi z}{h}\right) \phi_i$$
$$+ \eta H_x^2 \frac{\partial^2 w_i}{\partial x^2} - \eta H_x H_z \frac{\partial^2 u_i}{\partial x^2} + \eta H_x H_z z \frac{\partial^3 w_i}{\partial x^3}$$
$$- \eta H_x H_z \frac{h}{\pi} \sin\left(\frac{\pi z}{h}\right) \frac{\partial^3 w_i}{\partial x^3} + \eta H_x H_z \frac{h}{\pi} \sin\left(\frac{\pi z}{h}\right) \frac{\partial^2 \phi_i}{\partial x^2}.$$
$$\text{(39c)}$$

The resultant Lorentz's forces, $R = (R_x, R_y, R_z)$, and the corresponding bending moments, $M = (M_x, M_y, M_z)$, as follows Kiani ([2014]):

$$R_x = \int_{A_i} f_x \, dA_i = -\eta H_z^2 \frac{\pi}{h} p_0 \frac{\partial w_i}{\partial x} + \eta H_z^2 \frac{\pi}{h} p_0 \phi_i$$
$$- \eta H_z H_x A_i \frac{\partial^2 w_i}{\partial x^2} + \eta H_z^2 A_i \frac{\partial^2 u_i}{\partial x^2}$$
$$+ \eta H_z^2 \frac{h}{\pi} p_0 \frac{\partial^3 w_i}{\partial x^3} - \eta H_z^2 \frac{h}{\pi} p_0 \frac{\partial^2 \phi_i}{\partial x^2}, \tag{40a}$$

$$R_y = \int_{A_i} f_y \, dA_i = 0, \tag{40b}$$

$$R_z = \int_{A_i} f_z \, dA_i = \eta H_x H_z \frac{\pi}{h} p_0 \frac{\partial w_i}{\partial x}$$
$$- \eta H_x H_z \frac{\pi}{h} p_0 \phi_i + \eta H_x^2 A_i \frac{\partial^2 w_i}{\partial x^2}$$
$$- \eta H_x H_z A_i \frac{\partial^2 u_i}{\partial x^2} - \eta H_x H_z \frac{h}{\pi} p_0 \frac{\partial^3 w_i}{\partial x^3}$$
$$+ \eta H_x H_z \frac{h}{\pi} p_0 \frac{\partial^2 \phi_i}{\partial x^2}, \tag{40c}$$

$$M_x = M_z = 0, \tag{40d}$$

$$M_y = \int_{A_i} z f_x \, dA_i = -\eta H_z^2 p_1 \frac{\pi}{h} \frac{\partial}{\partial x} w_i + \eta H_z^2 p_1 \frac{\pi}{h} \phi_i$$
$$- I \eta H_z^2 \frac{\partial^3 w_i}{\partial x^3} + \eta H_z^2 \frac{h}{\pi} p_1 \frac{\partial^3 w_i}{\partial x^3} - \eta H_z^2 \frac{h}{\pi} p_1 \frac{\partial^2 \phi_i}{\partial x^2}$$
$$\text{(40e)}$$

Eventually, Lorentz work is written as Kiani ([2014]):

$$W_{lorentz} = \int_0^L (R_x u_i + R_z w_i + M_{y,x} w_i) dx. \tag{41}$$

3.4 Visco-Pasternak foundation and vdW interaction

The external work due to visco-Pasternak foundation and vdW forces are written as:

$$W_{visco-pasternak,vdw} = \int_0^L C_v \frac{R_1}{R_2} (w_2 - w_1) w_1 dx$$
$$+ \int_0^L C_v (w_1 - w_2) w_2 dx$$
$$+ \int_0^L (-K_w w_2 + G_P \nabla^2 w_2) w_2 dx$$
$$+ \int_0^L C_d \frac{\partial w_2}{\partial t} dx, \tag{42}$$

where K_w, C_d, C_v and G_P Winkler's spring modulus, damper, vdW interaction coefficient and Pasternak's shear modulus of elastic medium, respectively.

According to Hamilton's principle (i.e. Eq. ([17])), the motion equations are obtained as follows:

$$\delta u_1 : B_6 \frac{\partial^3 w_1}{\partial x^3} - B_7 \frac{\partial^3 w_1}{\partial x^3} + B_{17} \frac{\partial^2 \phi_1}{\partial x^2} + B_8 \frac{\partial^5 w_1}{\partial x^5}$$
$$- B_{16} \frac{\partial^2 \phi_1}{\partial x^2} + B_{18} \frac{\partial^4 \phi_1}{\partial x^4} - 2 B_{14_1} \frac{\partial^2 u_1}{\partial x^2}$$
$$+ 2 B_{15_1} \frac{\partial^4 u_1}{\partial x^4} + \rho_t A_1 \frac{\partial^2 u_1}{\partial t^2}$$
$$+ \frac{\rho_t h P_0}{\pi} \frac{\partial^3 w_1}{\partial x \partial t^2} - \frac{\rho_t h P_0}{\pi} \frac{\partial^2 \phi_1}{\partial t^2} - R_x, \tag{43}$$

$$
\delta w_1: -B_9 \frac{\partial \phi_1}{\partial x} - 2B_4 \frac{\partial^6 w_1}{\partial x^6} - B_{11} \frac{\partial^3 \phi_1}{\partial x^3} - B_8 \frac{\partial^5 u_1}{\partial x^5}
$$
$$
+ B_{10} \frac{\partial^3 \phi_1}{\partial x^3} - 2B_2 \frac{\partial^2 w_1}{\partial x^2} - B_{12} \frac{\partial^5 \phi_1}{\partial x^5}
$$
$$
- B_{24} \frac{\partial^3 \phi_1}{\partial x^3} - B_6 \frac{\partial^3 u_1}{\partial x^3} + B_7 \frac{\partial^3 u_1}{\partial x^3} - 2B_5 \frac{\partial^4 w_1}{\partial x^4}
$$
$$
+ 2B_{3_1} \frac{\partial^4 w_1}{\partial x^4} - \frac{\rho_t h P_1}{\pi} \frac{\partial^3 \phi_1}{\partial x \partial t^2} - \frac{\rho_t h^2 L}{\pi^2} \frac{\partial^4 w_1}{\partial x^2 \partial t^2}
$$
$$
+ \frac{\rho_t h^2 L}{\pi^2} \frac{\partial^3 \phi_1}{\partial x \partial t^2} - I \rho_t \frac{\partial^4 w_1}{\partial x^2 \partial t^2} + \rho_t A_1 \frac{\partial^2 w_1}{\partial t^2}
$$
$$
- \frac{\rho_t h P_0}{\pi} \frac{\partial^3 u_1}{\partial x \partial t^2} + 2 \frac{\rho_t h P_1}{\pi} \frac{\partial^4 w_1}{\partial x^2 \partial t^2}
$$
$$
- 2\tau_s(h+b) \frac{\partial^2 w_1}{\partial x^2} - R_z - M_{y,x}
$$
$$
- \frac{c_v R_1}{R_2}(w_2 - w_1) + \rho A_f \frac{\partial^2 w_1}{\partial t^2}
$$
$$
+ 2\rho A_f U_f \frac{\partial^2 w_1}{\partial x \partial t} + \rho A_f U_f^2 \frac{\partial^2 w_1}{\partial x^2}, \tag{44}
$$

$$
\delta \phi_1: -B_{10} \frac{\partial^3 w_1}{\partial x^3} + B_{12} \frac{\partial^5 w_1}{\partial x^5} + B_{18} \frac{\partial^4 u_1}{\partial x^4}
$$
$$
+ 2B_{22} \frac{\partial^4 \phi_1}{\partial x^4} + B_9 \frac{\partial w_1}{\partial x} + B_{24} \frac{\partial^3 w_1}{\partial x^3} + B_{17} \frac{\partial^2 u_1}{\partial x^2}
$$
$$
+ B_{11} \frac{\partial^3 w_1}{\partial x^3} + 2B_{19} \phi_1 + 2B_{23} \frac{\partial^2 \phi_1}{\partial x^2}
$$
$$
- 2B_{21} \frac{\partial^2 \phi_1}{\partial x^2} - B_{16} \frac{\partial^2 u_1}{\partial x^2} - \frac{\rho_t h P_0}{\pi} \frac{\partial^2 u_1}{\partial t^2}
$$
$$
+ \frac{\rho_t h P_1}{\pi} \frac{\partial^3 w_1}{\partial x \partial t^2} - \frac{\rho_t h^2 L}{\pi^2} \frac{\partial^3 w_1}{\partial x \partial t^2} + \frac{\rho_t h^2 L}{\pi^2} \frac{\partial^2 \phi_1}{\partial t^2}, \tag{45}
$$

$$
\delta u_2: B_6 \frac{\partial^3 w_2}{\partial x^3} - B_7 \frac{\partial^3 w_2}{\partial x^3} + B_{17} \frac{\partial^2 \phi_2}{\partial x^2} + B_8 \frac{\partial^5 w_2}{\partial x^5}
$$
$$
- B_{16} \frac{\partial^2 \phi_2}{\partial x^2} + B_{18} \frac{\partial^4 \phi_2}{\partial x^4} - 2B_{14_2} \frac{\partial^2 u_2}{\partial x^2}
$$
$$
+ 2B_{15_2} \frac{\partial^4 u_2}{\partial x^4} + \rho_t A_2 \frac{\partial^2 u_2}{\partial t^2} + \frac{\rho_t h P_0}{\pi} \frac{\partial^3 w_2}{\partial x \partial t^2}
$$
$$
- \frac{\rho_t h P_0}{\pi} \frac{\partial^2 \phi_2}{\partial t^2} - R_x, \tag{46}
$$

$$
\delta w_2: -B_9 \frac{\partial \phi_2}{\partial x} - 2B_4 \frac{\partial^6 w_2}{\partial x^6} - B_{11} \frac{\partial^3 \phi_2}{\partial x^3} - B_8 \frac{\partial^5 u_2}{\partial x^5}
$$
$$
+ B_{10} \frac{\partial^3 \phi_2}{\partial x^3} - 2B_2 \frac{\partial^2 w_2}{\partial x^2} - B_{12} \frac{\partial^5 \phi_2}{\partial x^5} - B_{24} \frac{\partial^3 \phi_2}{\partial x^3}
$$
$$
- B_6 \frac{\partial^3 u_2}{\partial x^3} + B_7 \frac{\partial^3 u_2}{\partial x^3} - 2B_5 \frac{\partial^4 w_2}{\partial x^4} + 2B_{3_2} \frac{\partial^4 w_2}{\partial x^4}
$$

$$
- \frac{\rho_t h P_1}{\pi} \frac{\partial^3 \phi_2}{\partial x \partial t^2} - \frac{\rho_t h^2 L}{\pi^2} \frac{\partial^4 w_2}{\partial x^2 \partial t^2} + \frac{\rho_t h^2 L}{\pi^2} \frac{\partial^3 \phi_2}{\partial x \partial t^2}
$$
$$
- I \rho_t \frac{\partial^4 w_2}{\partial x^2 \partial t^2} + \rho_t A_2 \frac{\partial^2 w_2}{\partial t^2} - \frac{\rho_t h P_0}{\pi} \frac{\partial^3 u_2}{\partial x \partial t^2}
$$
$$
+ 2\frac{\rho_t h P_1}{\pi} \frac{\partial^4 w_2}{\partial x^2 \partial t^2} - 2\tau_s(h+b) \frac{\partial^2 w_2}{\partial x^2} - R_z - M_{y,x}
$$
$$
- c_v(w_1 - w_2) - G_p \nabla^2 w_2 + K_w w_2 + c_d \frac{\partial w_2}{\partial t}, \tag{47}
$$

$$
\delta \phi_2: -B_{10} \frac{\partial^3 w_2}{\partial x^3} + B_{12} \frac{\partial^5 w_2}{\partial x^5} + B_{18} \frac{\partial^4 u_2}{\partial x^4}
$$
$$
+ 2B_{22} \frac{\partial^4 \phi_2}{\partial x^4} + B_9 \frac{\partial w_2}{\partial x} + B_{24} \frac{\partial^3 w_2}{\partial x^3} + B_{17} \frac{\partial^2 u_2}{\partial x^2}
$$
$$
+ B_{11} \frac{\partial^3 w_2}{\partial x^3} + 2B_{19} \phi_2 + 2B_{23} \frac{\partial^2 \phi_2}{\partial x^2}
$$
$$
- 2B_{21} \frac{\partial^2 \phi_2}{\partial x^2} - B_{16} \frac{\partial^2 u_2}{\partial x^2} - \frac{\rho_t h P_0}{\pi} \frac{\partial^2 u_2}{\partial t^2}
$$
$$
+ \frac{\rho_t h P_1}{\pi} \frac{\partial^3 w_2}{\partial x \partial t^2} - \frac{\rho_t h^2 L}{\pi^2} \frac{\partial^3 w_2}{\partial x \partial t^2} + \frac{\rho_t h^2 L}{\pi^2} \frac{\partial^2 \phi_2}{\partial t^2}. \tag{48}
$$

Boundary conditions at $x = 0$ and $x = L$ read (Lei et al. 2013a):

$$
\frac{\partial^2 \delta w_i}{\partial x^2} = 0, \ \frac{\partial \delta w_i}{\partial x} = 0, \ \delta w_i = 0,
$$
$$
\frac{\partial \delta \phi_i}{\partial x} = 0, \ \delta \phi_i = 0, \tag{49}
$$
$$
\frac{\partial \delta u_i}{\partial x} = 0, \ \delta u_i = 0.
$$

3.5 Solution procedure

3.5.1 DQM

DQM is employed in this section which in essence approximates the partial derivative of a function, with respect to a spatial variable at a given discrete point, as a weighted linear sum of the function values at all discrete points chosen in the solution domain of the spatial variable (Khodami Maraghi et al. 2013). Let F be a function representing u_1, u_2, w_1, w_2 and φ_1, φ_2 with respect to variable x in the domain of $(0 < x < L)$ having N_x grid points along these variable. The nth-order partial derivative of $F(x)$ with respect to x may be expressed discretely as

$$\frac{d^n F(x_i)}{dx^n} = \sum_{k=1}^{N_x} A_{ik}^{(n)} F(x_k) \quad n = 1, \ldots, N_x - 1, \quad (50)$$

where $A_{ik}^{(n)}$ is the weighting coefficient, whose recursive formula are described in Khodami Maraghi et al. (2013). The Chebyshev–Gauss–Lobatto polynomials (Civalek 2004, 2006) was used to determine the unequally spaced position of the grid points as follows

$$x_i = \frac{L}{2}\left[1 - \cos\left(\frac{2i-1}{N_x - 1}\right)\pi\right], \quad (51)$$

Combining all the motion equations along with the corresponding boundary conditions using DQM and rewritten them in matrix form yields

Hence, the components of $\{d\}$ can be written in the Fourier series with period $2T$ as

$$\{d\} = \sum_{k=1,3,\ldots}^{\infty}\left[\{a\}_k \sin\frac{k\omega t}{2} + \{b\}_k \cos\frac{k\omega t}{2}\right], \quad (53)$$

According to this method, the first instability region is usually the most important in studies of structures. It is due to the fact that the first DIR is wider that other DIRs and structural damping in higher regions becomes neutralize (Lanhe et al. 2007). Substituting Eq. (53) into Eq. (52) and setting the coefficients of each sine and cosine as well as the sum of the constant terms to zero, yields

$$\begin{bmatrix} [0]_{14\times14} & [0]_{14\times(6N_x-14)} \\ [M_{db}]_{(6N_x-14)\times14} & [M_{dd}]_{(6N_x-14)\times(6N_x-14)} \end{bmatrix}_{6N_x\times6N_x} \left\{ \begin{array}{c} \{\ddot{Y}_b\} \\ \{\ddot{Y}_d\} \end{array} \right\}$$

$$+ \left(\begin{bmatrix} [0]_{14\times14} & [0]_{14\times(6N_x-14)} \\ [C_{db}]_{(6N_x-14)\times14} & [C_{dd}]_{(6N_x-14)\times(6N_x-14)} \end{bmatrix} \right.$$
$$\left. + (V_0 + V_0\alpha\cos(\omega t)) \begin{bmatrix} [0]_{14\times14} & [0]_{14\times(6N_x-14)} \\ [C_{db}]^f_{(6N_x-14)\times14} & [C_{dd}]^f_{(6N_x-14)\times(6N_x-14)} \end{bmatrix} \right)_{6N_x\times6N_x} \left\{ \begin{array}{c} \{\dot{Y}_b\} \\ \{\dot{Y}_d\} \end{array} \right\}$$

$$+ \left(\begin{bmatrix} [K_{bb}]_{14\times14} & [K_{bd}]_{14\times(6N_x-14)} \\ [K_{db}]_{(6N_x-14)\times14} & [K_{dd}]_{(6N_x-14)\times(6N_x-14)} \end{bmatrix} \right.$$
$$\left. + (V_0 + V_0\alpha\cos(\omega t))^2 \begin{bmatrix} [K_{bb}]^f_{14\times14} & [K_{bd}]^f_{14\times(6N_x-14)} \\ [K_{db}]^f_{(6N_x-14)\times14} & [K_{dd}]^f_{(6N_x-14)\times(6N_x-14)} \end{bmatrix} \right)_{6N_x\times6N_x} \left\{ \begin{array}{c} \{Y_b\}_{14\times1} \\ \{Y_d\}_{(6N_x-14)\times1} \end{array} \right\} = 0,$$

$$(52)$$

where $[M]$, $[C]$ and $[K]$ are the mass, damping and stiffness matrixes, respectively; $[C^f]$ and $[K^f]$ are the respectively, damping and stiffness matrixes related to pulsating fluid; $\{Y\}$ is the displacement vector (i.e. $\{Y\} = \{u_i, v_i, w_i\}$ $i = 1, 2$); subscript b and d represent boundary and domain points.

3.6 Bolotin method

In order to determinate the DIR of visco-DWCNTs, the method suggested by Bolotin (1964) is applied.

$$\left| \left([K] + \left(1 \pm \alpha + \frac{\alpha^2}{2}\right)[K]^f\right) \right.$$
$$+ \left(\pm[C]\frac{\omega}{2} + \left(\frac{\alpha\omega}{4} \pm \frac{\omega}{2}\right)[C]^f\right) \quad (54)$$
$$\left. - [M]\frac{\omega^2}{4} \right| = 0,$$

Solving the above equation based on eigenvalue problem, the variation of ω with respect to α can be plotted as DIR.

4 Results and discussion

Based on DQM and Bolotin method, the effects of nonlocal parameter, magnetic field, visco-Pasternak foundation, Knudsen number, surface stress and fluid velocity on the DIR of visco-DWCNTs are investigated. The material properties of the DWCNTs related to bulk are: Young's modulus of E = 1 Tpa, Poisson's ratio of $\upsilon = 0.27$, density of $\rho = 2,300$ kg/m³ and thickness of $h = 0.34$ mm (Lei et al. 2012). Generally, the surface material properties can be calculated by atomic simulations. However, the material properties of the DWCNTs related to surface are: surface Young's modulus of, $E_s = 35.3$ N/m and residual surface stress of, $\tau_s = 0.31$ N/m (Lei et al. 2012).

4.1 Convergence of DQM

The convergence and accuracy of the DQM in evaluating the DIR of the visco-DWCNTs are shown in Table 1 for two cases of with and without considering surface effects. The results are prepared for different values of the DQM grid points. Fast rate of convergence of the DQM is quite evident. As can be seen, eleven grid points can yield accurate results.

4.2 Validation

To the best of the authors' knowledge no published literature is available for visco-DWCNTs pulsating fluid embedded in a visco-Pasternak foundation based on surface sinusoidal strain gradient theory.

Table 1 Convergence behavior and accuracy of the DQM for DIR of DWCNTs

Grid points	Without surface effects		With surface effects	
	Longitudinal magnetic field	Lateral magnetic field	Longitudinal magnetic field	Lateral magnetic field
5 × 5	7.9043	4.1204	8.8819	5.3167
7 × 7	8.1952	5.5809	9.3902	6.5979
9 × 9	8.2288	5.6501	9.4324	6.6025
11 × 11	8.2592	5.6519	9.4338	6.6034
14 × 14	8.2592	5.6519	9.4338	6.6034
16 × 16	8.2592	5.6519	9.4338	6.6034

Table 2 Validation of present work with Ref. [16] for non-dimensional fundamental frequency of firs mode

h/l	Theory	L = 10h, (Present work)			L = 30h Gurtin and Ian Murdoch (1975)		
		Local theory	Modified couple stress theory	Strain gradient theory	Local theory	Modified couple stress theory	Strain gradient theory
1	EBT	13.4981	24.6135	40.9691	13.4975	24.6031	40.9674
	TBT	13.4981	24.5171	39.9337	13.4595	24.5061	39.9336
	SSDBT	13.4981	24.5957	40.8981	13.4595	24.5801	40.8962
5	EBT	13.4981	14.1114	15.5579	13.4975	14.1105	15.5572
	TBT	13.4981	14.0715	15.4996	13.4595	14.0707	15.4989
	SSDBT	13.4981	14.0799	15.5190	13.4595	14.0794	15.5184
10	EBT	13.4981	13.6535	14.0409	13.4595	13.6533	14.0408
	TBT	13.4981	13.6150	13.9980	13.4595	13.6149	13.9979
	SSDBT	13.4981	13.6161	14.0028	13.4595	13.6160	14.0027

Since, no reference to such a work is found to-date in the literature, its validation is not possible. However, in an attempt to validate this work as far as possible, a simplified analysis of this paper is carried out without considering the surface effect, pulsating fluid, visco-Pasternak foundation and viscoelastic property of DWCNTs. Present results are compared with the work of Akgöz and Civalek (2013a, b) who studied nonlocal vibration analysis

of microbeams based on SSDBT and strain gradient theory. Considering the material properties the same as Ref. (Akgöz and Civalek 2013a, b) and dimensionless frequency as $\bar{\omega} = \omega L^2 \sqrt{m_0/EI}$, the results are shown in Table 2. It is evident that the present results are in a good agreement with those obtained results by Ref. (Akgöz and Civalek 2013a, b). It should be noted that setting $\Phi(z) = 0$ and $\Phi(z) = z$ yields the EBT and TBT, respectively. It is also concluded that

Fig. 2 Dimensionless pulsation amplitude versus dimensionless pulsation frequency for different nonlocal theories
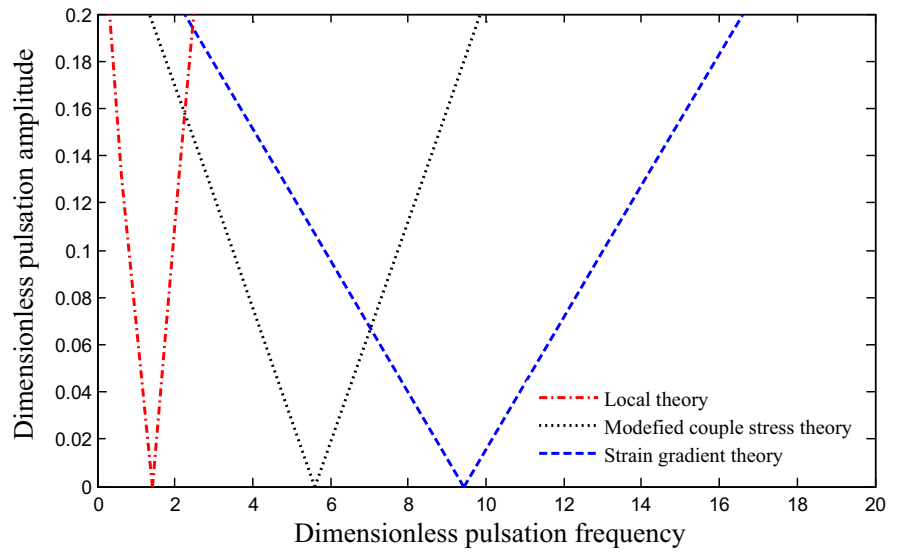
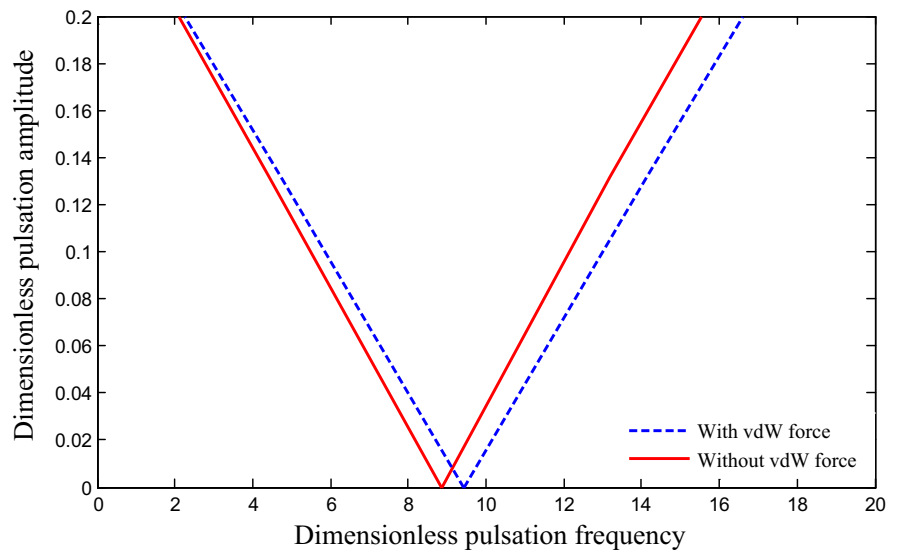**Fig. 3** Dimensionless pulsation amplitude versus dimensionless pulsation frequency for different values of Knudsen number
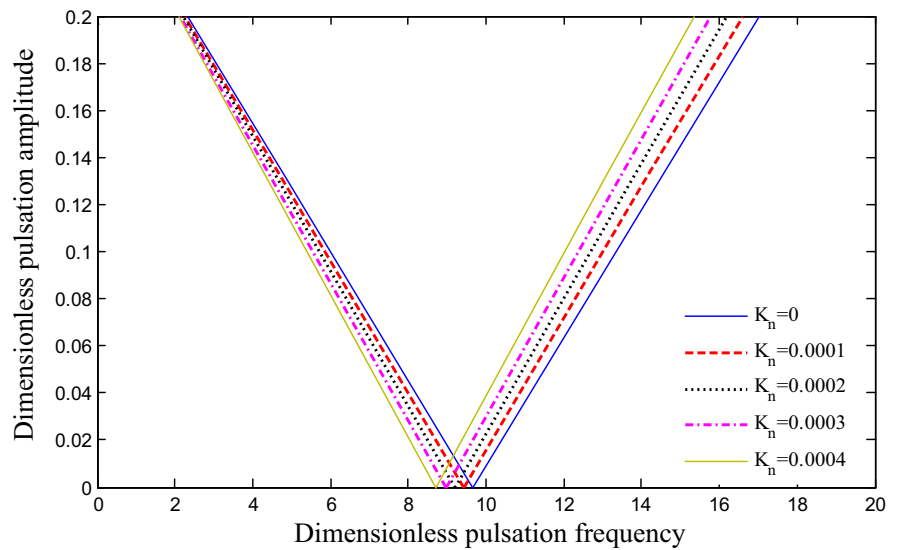
the frequency of SSDBT is higher than EBT and TBT. In addition, one can see that excellent agreement exist between the results of the DQM in the present paper with exact solution in Ref. (Akgöz and Civalek 2013a, b).

4.3 The effect of different parameters on DIR

Figure 2 represents the effect of different nonlocal theories including classical, modified couple stress

and strain gradient on the dimensionless pulsation frequency (i.e. $\Omega = \omega L \sqrt{\rho_t / E}$) versus the dimensionless pulsation amplitude. Comparing classical theory with two non-classical theories, it can be concluded that the DIR predicted by the strain gradient theory is higher than the classical and modified couple stress theories. This is because the strain gradient theory expresses the three additional dilatation gradient tensor, the deviatoric stretch gradient tensor and the rotation gradient tensor.

Fig. 4 Dimensionless pulsation amplitude versus dimensionless pulsation frequency for different elastic medium
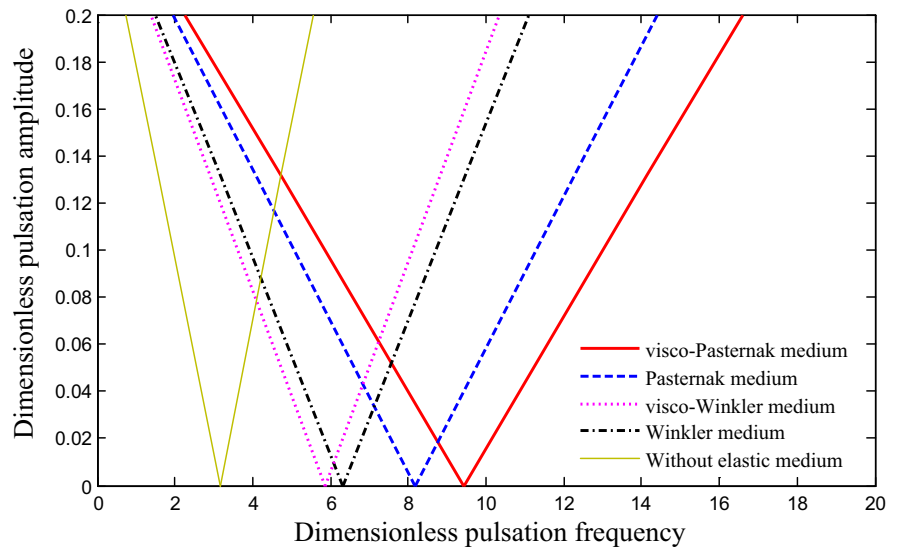
Fig. 5 Dimensionless pulsation amplitude versus dimensionless pulsation frequency for vdW effect

Figure 3 illustrates the effect of Knudsen number on the dimensionless pulsation frequency with respect to the dimensionless pulsation amplitude. It is came to know by enhancing Knudsen number, DIR and the dimensionless pulsation frequency shift to left and decrease. It is because that, when the Knudsen number increases, the mean free path of liquid molecules increases and results in lower stiffness.

Figure 4 depicts variations of the dimensionless pulsation frequency as a function of the dimensionless pulsation amplitude by varying the type of DWCNT surrounding foundation. Four different elastic medium are considered namely as visco-Pasternak, Pasternak, visco-Winkler and Winkler mediums. As can be seen considering elastic foundation increases the magnitude of dimensionless pulsation frequency and

Fig. 6 Dimensionless pulsation amplitude versus dimensionless pulsation frequency for different values of magnetic field
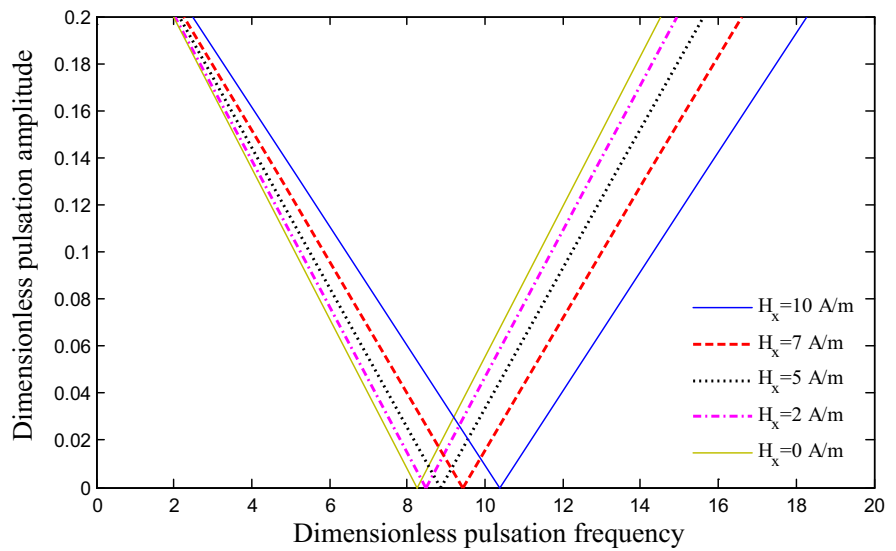
Fig. 7 Dimensionless pulsation amplitude versus dimensionless pulsation frequency for different values of magnetic field angle
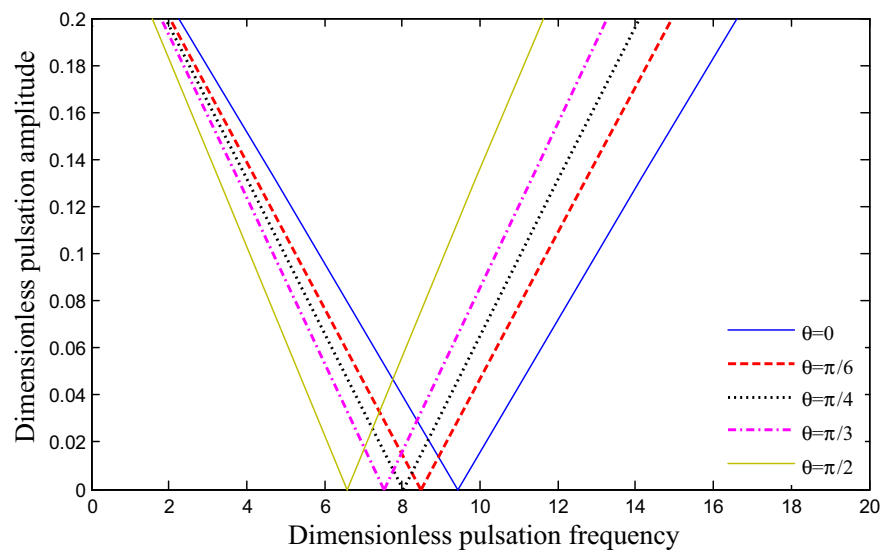
subsequently, DIR shifts to right. It is due to the fact that putting DWCNT in an elastic medium makes the system more stable and stiffer. It is also concluded that the DIR of Pasternak or visco-Pasternak model is higher than Winkler or visco-Winkler one. It is because Pasternak model considers not only the normal stresses but also the transverse shear deformation and continuity among the spring elements. Furthermore, the DIR predicted by visco-Pasternak

and visco-Winkler mediums is lower than Pasternak and Winkler models, respectively.

The effects of the vdW force on the dimensionless pulsation frequency with respect to the dimensionless pulsation amplitude are shown in Fig. 5. Stiffness of tube increases by considering vdW force so the dimensionless pulsation frequency increases.

Figure 6 demonstrates variations of the dimensionless pulsation frequency versus the dimensionless

Fig. 8 Dimensionless pulsation amplitude versus dimensionless pulsation frequency for different values of surface effect parameters
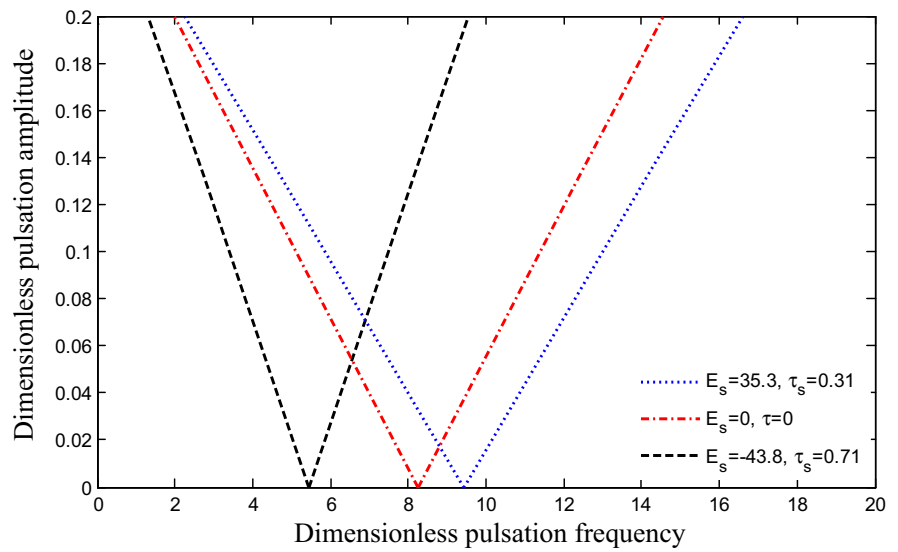
Fig. 9 Dimensionless pulsation amplitude versus dimensionless pulsation frequency for different values of fluid velocity
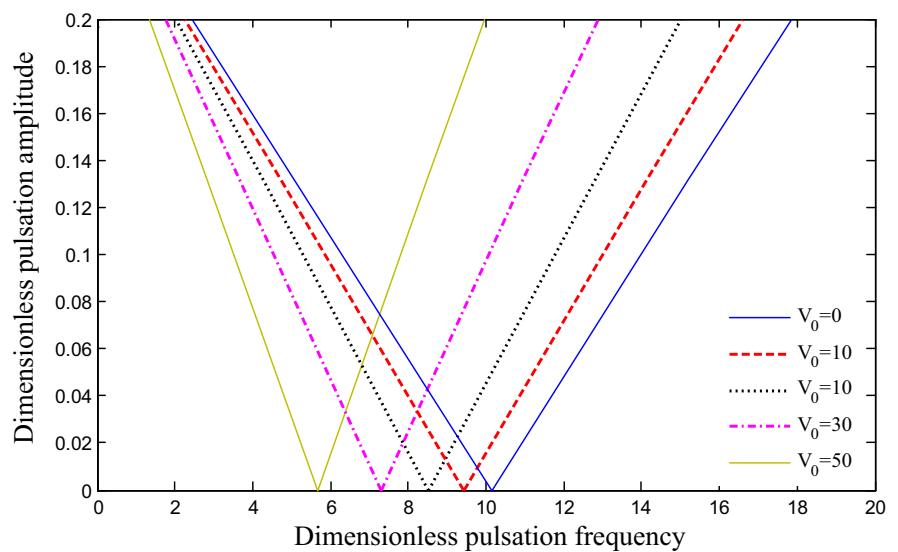
pulsation amplitude for different magnitude of longitudinal magnetic field intensity. It can be observed that increment of magnetic field intensity makes the system more stable, hence the DIR and dimensionless pulsation frequency shift to right and increases. In addition, the longitudinal magnetic field has little influence on DIR when its magnitude is very small.

In order to show the magnetic field angle effects, Fig. 7 is plotted where the dimensionless pulsation frequency is a function of the dimensionless pulsation amplitude. It is shown that when the magnetic field angle is zero (i.e. longitudinal magnetic field) the dimensionless pulsation frequency and DIR are maximum. With increasing magnetic field angle, the dimensionless pulsation frequency and DIR decrease and in magnetic field angle of $\pi/2$ (i.e. lateral magnetic field), those become minimum.

Figure 8 illustrates the surface stress effect on the dimensionless pulsation frequency with respect to the dimensionless pulsation amplitude. This figure shows that the surface stress effect is remarkable so that with ignoring this effect the DIR and the dimensionless pulsation frequency are not inaccurate. This is due to the fact that considering surface stress increases the stability of the DWCNT.

Figure 9 shows the dimensionless pulsation frequency with respect to the dimensionless pulsation amplitude for different values of fluid velocities. Increasing the fluid velocity through the inner nanotube frequency decreases the dimensionless pulsation frequency. Furthermore, shifting the DIRs is obvious in higher flow velocities. Flowing fluid through the DWCNT exerts compressive axial load and for higher velocities the magnitude of this load increases, so increasing the flow velocity results in frequency and DIR decreasing.

5 Conclusion

Dynamic response of DWCNTs have applications in designing many NEMS/MEMS devices such as sensors, actuators, fluid storage, solar cell and biomechanic. Pulsating fluid induced dynamic instability of DWCNTs considering structural damping, surface stress effect and viscoelastic foundation was the main contributions of the present paper. The DWCNTs were subjected to 2D magnetic field and simulated by sinusoidal strain gradient theory. Bolotin method in conjunction with DQM were used for calculating the DIR of the visco-DWCNT The Chebyshev–Gauss–Lobatto polynomials was used to determine the unequally spaced position of the grid points. The effects of nonlocal parameter, magnetic field, visco-Pasternak foundation, Knudsen number, surface stress and fluid velocity on the DIR of the visco-DWCNT were discussed. Results depict that the imposed magnetic field was an effective controlling parameter for dynamic instability of visco-DWCNTs. The DIR predicted by the strain gradient theory was higher than the classical and modified couple stress theories. It was shown that when the magnetic field angle is zero (i.e. longitudinal magnetic field) the dimensionless pulsation frequency and DIR are maximum. Furthermore, considering surface stress increases the stability of the DWCNT. The results of this study were in good agreement with those reported by Ref. (Akgöz and Civalek 2013a, b). The results presented in this work can be useful for the study and design of the next generation of nano/micro structures that make use of the nonlocal dynamic instability of visco-CNTs embedded in visco-Pasternak medium.

Appendix 1

$$B_1 = \frac{\tau_s h P_{0_s}}{\pi}, \quad B_2 = \frac{8}{15}\frac{l_1^2 G^* \pi^2 L}{h^2} + \frac{1}{4}\frac{l_2^2 G^* L \pi^2}{h^2} + G^* O,$$

$$B_{3_i} = -l_1^2\left(\frac{32}{15}G^* T_0 + \frac{16}{15}G_s^* T_{0_s}\right) + \frac{8}{15}l_1^2\left(G^* A_i + G_s^* S_i\right) + l_1^2\left(\frac{32}{15}G^* O + \frac{8}{15}G_s^* O_s\right) - l_2^2\left(G^* T_0 + 2G_s^* T_{0_s}\right)$$

$$+ l_2^2\left(\frac{1}{4}G^* O + G_s^* O_s\right) + l_2^2\left(G^* A_i + G_s^* S_i\right) - 4l_0^2\left(G^* T_0 + G_s^* T_{0_s}\right) + 2l_0^2\left(G^* O + G_s^* O_s\right)$$

$$+ 2l_0^2\left(G^* A_i + G_s^* S_i\right) + EI + E_s I_s - 2\frac{h}{\pi}\left(E^* P_1 + E_s^* P_{1_s}\right) + \frac{h^2}{\pi^2}\left(E^* L + E_s^* L_s\right),$$

$$B_4 = \frac{4}{5}l_1^2\left(G^* I + G_s^* I_s\right) + \frac{4}{5}\frac{l_1^2 h^2}{\pi^2}\left(G^* L + G_s^* L_s\right) - \frac{8}{5}\frac{l_1^2 h}{\pi}\left(G^* P_1 + G_s^* P_{1_s}\right) + 2l_0^2\left(G^* I + G_s^* I_s\right)$$

$$- 4\frac{l_0^2 h}{\pi}\left(G^* P_1 + G_s^* P_{1_s}\right) + 2\frac{l_0^2 h^2}{\pi^2}\left(G^* L + G_s^* L_s\right),$$

$$B_5 = \frac{4}{5}l_1^2 G^* L - \frac{4}{5}\frac{l_1^2 G^* \pi P_1}{h}, \quad B_6 = \frac{4}{5}\frac{l_1^2 G^* \pi P_0}{h}, \quad B_7 = 2\frac{h}{\pi}\left(E^* P_0 + E_s^* P_{0_s}\right),$$

$$B_8 = \frac{8}{5}\frac{l_1^2 h}{\pi}\left(G^* P_0 + G_s^* P_{0_s}\right) + 4\frac{l_0^2 h}{\pi}\left(G^* P_0 + G_s^* P_{0_s}\right), \quad B_9 = -\frac{16}{15}\frac{l_1^2 G^* \pi^2 L}{h^2} - \frac{1}{2}\frac{l_2^2 G^* L \pi^2}{h^2} - 2G^* O,$$

$$B_{10} = -l_1^2\left(\frac{64}{15}G^* O + \frac{16}{15}G_s^* O_s\right) + l_1^2\left(\frac{32}{15}G^* T_0 + \frac{16}{15}G_s^* T_{0_s}\right) - l_2^2\left(\frac{1}{2}G^* O + 2G_s^* O_s\right)$$

$$+ l_2^2\left(G^* T_0 + 2G_s^* T_{0_s}\right) + 4l_0^2\left(G^* T_0 + G_s^* T_{0_s}\right) - 4l_0^2\left(G^* O + G_s^* O_s\right) - 2\frac{h^2}{\pi^2}\left(E^* L + E_s^* L_s\right)$$

$$+ 2\frac{h}{\pi}\left(E^* P_1 + E_s^* P_{1_s}\right),$$

$$B_{11} = -\frac{4}{5}l_1^2 G^* L + \frac{4}{5}\frac{l_1^2 G^* \pi P_1}{h},$$

$$B_{12} = -\frac{8}{5}\frac{l_1^2 h^2}{\pi^2}\left(G^* L + G_s^* L_s\right) + \frac{8}{5}\frac{l_1^2 h}{\pi}\left(G^* P_1 + G_s^* P_{1_s}\right) - 4\frac{l_0^2 h^2}{\pi^2}\left(G^* L + G_s^* L_s\right)$$

$$+ 4\frac{l_0^2 h}{\pi}\left(G^* P_1 + G_s^* P_{1_s}\right),$$

$$B_{13_i} = \tau_s S_i, B_{14_i} = E^* A_i + E_s^* S_i, B_{15_i} = \frac{4}{5}l_1^2\left(G^* A_i + G_s^* S_i\right) + 2l_0^2\left(G^* A_i + G_s^* S_i\right),$$

$$B_{16} = -2\frac{h}{\pi}\left(E^* P_0 + E_s^* P_{0_s}\right), \quad B_{17} = -\frac{4}{5}\frac{l_1^2 G^* \pi P_0}{h},$$

$$B_{18} = -\frac{8}{5}\frac{l_1^2 h}{\pi}\left(G^* P_0 + G_s^* P_{0_s}\right) - 4\frac{l_0^2 h}{\pi}\left(G^* P_0 + G_s^* P_{0_s}\right), \quad B_{19} = \frac{8}{15}\frac{l_1^2 G^* \pi^2 L}{h^2} + \frac{1}{4}\frac{l_2^2 G^* L \pi^2}{h^2} + G^* O,$$

$$B_{20} = -\frac{\tau_s h P_{0_s}}{\pi},$$

$$B_{21} = l_1^2\left(\frac{32}{15}G^* O + \frac{8}{15}G_s^* O_s\right) + l_2^2\left(\frac{1}{4}G^* O + G_s^* O_s\right) + 2l_0^2\left(G^* O + G_s^* O_s\right) + \frac{h^2}{\pi^2}\left(E^* L + E_s^* L_s\right),$$

$$B_{22} = \frac{4}{5}\frac{l_1^2 h^2}{\pi^2}\left(G^* L + G_s^* L_s\right) + 2\frac{l_0^2 h^2}{\pi^2}\left(G^* L + G_s^* L_s\right), B_{23} = \frac{4}{5}l_1^2 G^* L, B_{24} = -\frac{4}{5}l_1^2 G^* L,$$

$$(55)$$

where the following integrals are defined

$$
\begin{pmatrix}
(A_i, I, P_0, P_1, T_0, L, O) \\
(S_i, I_s, P_{0_s}, P_{1_s}, T_{0_s}, L_s, O_s)
\end{pmatrix}
= \int_{A,S} \begin{pmatrix} \beta & 0 \\ 0 & \beta \end{pmatrix} \begin{pmatrix} dA_i \\ dS_i \end{pmatrix},
\tag{56}
$$

where

$$
\beta = \left(1, z^2, f^{(\sin)}, f^{(\cos)}, z f^{(\cos)}, \left(f^{(\sin)}\right)^2, \left(f^{(\cos)}\right)^2 \right).
\tag{57}
$$

References

Akgöz, B., Civalek, Ö.: A size-dependent shear deformation beam model based on the strain gradient elasticity theory. Int. J. Eng. Sci. **70**, 1–14 (2013a). doi:10.1016/j.ijengsci.2013.04.004

Akgöz, B., Civalek, Ö.: Free vibration analysis of axially functionally graded tapered Bernoulli–Euler microbeams based on the modified couple stress theory. Compos. Struct. **98**, 314–322 (2013b). doi:10.1016/j.compstruct.2012.11.020

Akgöz, B., Civalek, Ö.: Analysis of micro-sized beams for various boundary conditions based on the strain gradient elasticity theory. Arch. Appl. Mech. **82**(3), 423–443 (2012). doi:10.1007/s00419-011-0565-5

Aydogdu, M.: A new shear deformation theory for laminated composite plates. Compos. Struct. **89**(1), 94–101 (2009). doi:10.1016/j.compstruct.2008.07.008

Bolotin, V.V.: The dynamic stability of elastic systems. In: The Dynamic Stability of Elastic Systems. Holden-Day, San Francisco (1964)

Civalek, Ö.: Harmonic differential quadrature-finite differences coupled approaches for geometrically nonlinear static and dynamic analysis of rectangular plates on elastic foundation. J. Sound Vib. **294**, 966–980 (2006). doi:10.1016/j.jsv.2005.12.041

Civalek, Ö.: Application ofdifferential quadrature (DQ) and harmonic differential quadrature (HDQ) for buckling analysis of thin isotropic plates and elastic columns. Eng. Struct. **26**, 171–186 (2004). doi:10.1016/j.engstruct.2003.09.005

Eringen, A.C.: On differential equations of nonlocal elasticity and solutions of screw dislocation and surface waves. J. Appl. Phys. **54**(9), 4703–4710 (1983). doi:10.1063/1.332803

Fu, Y., Bi, R., Zhang, P.: Nonlinear dynamic instability of double-walled carbon nanotubes under periodic excitation. Acta Mech. Solida Sin. **22**(3), 206–212 (2009). doi:10.1016/S0894-9166(09)60267-6

Gheshlaghi, B., Hasheminejad, S.M.: Surface effects on nonlinear free vibration of nanobeams. Compos. B Eng. **42**(4), 934–937 (2011). doi:10.1016/j.compositesb.2010.12.026

Ghorbanpour Arani, A., Amir, S.: Electro-thermal vibration of visco-elastically coupled BNNT systems conveying fluid embedded on elastic foundation via strain gradient theory. Phys. B **419**, 1–6 (2013). doi:10.1016/j.physb.2013.03.010

Ghorbanpour Arani, A., Kolahchi, R., Vossough, H.: Nonlocal wave propagation in an embedded DWBNNT conveying fluid via strain gradient theory. Phys. B **407**(21), 4281–4286 (2012). doi:10.1016/j.physb.2012.07.018

Ghorbanpour Arani, A., Hashemian, M., Kolahchi, R.: Nonlocal Timoshenko beam model for dynamic stability of double-walled boron nitride nanotubes conveying nanoflow. Proc. Inst. Mech. Eng. (2013). doi:10.1177/1740349913513449

Ghorbanpour Arani, A., Amir, S., Dashti, P., Yousefi, M.: Flow-induced vibration of double bonded visco-CNTs under magnetic fields considering surface effect. Comput. Mater. Sci. **86**, 144–154 (2014a). doi:10.1016/j.commatsci.2014.01.047

Ghorbanpour Arani, A., Kolahchi, R., Hashemian, M.: Nonlocal surface piezoelasticity theory for dynamic stability of double-walled boron nitride nanotube conveying viscose fluid based on different theories. Proc. Inst. Mech. Eng. **28**(17), 3258–3280 (2014b). doi:10.1177/0954406214527270

Gurtin, M., Ian Murdoch, A.: A continuum theory of elastic material surfaces. Arch. Rational. Mech. Anal. **57**(4), 291–323 (1975). doi:10.1007/BF00261375

Gurtin, M.E., Ian Murdoch, A.: Surface stress in solids. Int. J. Solids Struct. **14**(6), 431–440 (1978). doi:10.1016/0020-7683(78)90008-2

Iijima, S.: Helical microtubules of graphitic carbon. Nature **354**(6348), 56–58 (1991)

Karama, M., Afaq, K.S., Mistou, S.: Mechanical behaviour of laminated composite beam by the new multi-layered laminated composite structures model with transverse shear stress continuity. Int. J. Solids Struct. **40**(6), 1525–1546 (2003). doi:10.1016/S0020-7683(02)00647-9

Kaviani, F., Mirdamadi, H.R.: Wave propagation analysis of carbon nano-tube conveying fluid including slip boundary condition and strain/inertial gradient theory. Comput. Struct. **116**, 75–87 (2013). doi:10.1016/j.compstruc.2012.10.025

Khodami Maraghi, Z., Ghorbanpour Arani, A., Kolahchi, R., Amir, S., Bagheri, M.R.: Nonlocal vibration and instability of embedded DWBNNT conveying viscose fluid. Compos. B **45**(1), 423–432 (2013). doi:10.1016/j.compositesb.2012.04.066

Kiani, K.: Vibration behavior of simply supported inclined single-walled carbon nanotubes conveying viscous fluids flow using nonlocal Rayleigh beam model. Appl. Math. Model. **37**(4), 1836–1850 (2013). doi:10.1016/j.apm.2012.04.027

Kiani, K.: Vibration and instability of a single-walled carbon nanotube in a three-dimensional magnetic field. J. Phys. Chem. Solids **75**(1), 15–22 (2014). doi:10.1016/j.jpcs.2013.07.022

Lam, D.C.C., Yang, F., Chong, A.C.M., Wang, J., Tong, P.: Experiments and theory in strain gradient elasticity. J. Mech. Phys. Solids **51**(8), 1477–1508 (2003). doi:10.1016/S0022-5096(03)00053-X

Lanhe, W., Hongjun, W., Daobin, W.: Dynamic stability analysis of FGM plates by the moving least squares differential quadrature method. Compos. Struct. **77**(3), 383–394 (2007). doi:10.1016/j.compstruct.2005.07.011

Lei, X.-W., Natsuki, T., Shi, J.-X., Ni, Q.-Q.: Surface effects on the vibrational frequency of double-walled carbon nanotubes using the nonlocal Timoshenko beam model. Compos. B **43**(1), 64–69 (2012). doi:10.1016/j.compositesb.2011.04.032

Lei, J., He, Y., Zhang, B., Gan, Z., Zeng, P.: Bending and vibration of functionally graded sinusoidal microbeams based on the strain gradient elasticity theory. Int. J. Eng. Sci. **72**, 36–52 (2013a). doi:10.1016/j.ijengsci.2013.06.012

Lei, Y., Adhikari, S., Friswell, M.I.: Vibration of nonlocal Kelvin–Voigt viscoelastic damped Timoshenko beams. Int. J. Eng. Sci. **66–67**, 1–13 (2013b). doi:10.1016/j.ijengsci.2013.02.004

Lei, Y., Murmu, T., Adhikari, S., Friswell, M.I.: Dynamic characteristics of damped viscoelastic nonlocal Euler-Bernoulli beams. Eur. J. Mech. A **42**, 125–136 (2013c). doi:10.1016/j.euromechsol.2013.04.006

Levinson, M.: A new rectangular beam theory. J. Sound Vib. **74**(1), 81–87 (1981). doi:10.1016/0022-460X(81)90493-4

Li, J., Wu, Z., Kong, X., Li, X., Wu, W.: Comparison of various shear deformation theories for free vibration of laminated composite beams with general lay-ups. Compos. Struct. **108**, 767–778 (2014). doi:10.1016/j.compstruct.2013.10.011

Liang, F., Su, Y.: Stability analysis of a single-walled carbon nanotube conveying pulsating and viscous fluid with nonlocal effect. Appl. Math. Model. **37**(10–11), 6821–6828 (2013). doi:10.1016/j.apm.2013.01.053

Lim, C., Li, C., Yu, J.-L.: Dynamic behaviour of axially moving nanobeams based on nonlocal elasticity approach. Acta. Mech. Sin. **26**(5), 755–765 (2010)

Malekzadeh, P., Shojaee, M.: Surface and nonlocal effects on the nonlinear free vibration of non-uniform nanobeams. Compos. B **52**, 84–92 (2013). doi:10.1016/j.compositesb.2013.03.046

Mindlin, R.D.: Second gradient of strain and surface-tension in linear elasticity. Int. J. Solids Struct. **1**(4), 417–438 (1965). doi:10.1016/0020-7683(65)90006-5

Mirramezani, M., Mirdamadi, H.R., Ghayour, M.: Innovative coupled fluid–structure interaction model for carbon nanotubes conveying fluid by considering the size effects of nano-flow and nano-structure. Comput. Mater. Sci. **77**, 161–171 (2013). doi:10.1016/j.commatsci.2013.04.047

Reddy, J.N.: A simple higher-order theory for laminated composite plates. J. Appl. Mech. **51**(4), 745–752 (1984). doi:10.1115/1.3167719

Şimşek, M., Reddy, J.N.: Bending and vibration of functionally graded microbeams using a new higher order beam theory and the modified couple stress theory. Int. J. Eng. Sci. **64**, 37–53 (2013). doi:10.1016/j.ijengsci.2012.12.002

Soldatos, K.P.: A transverse shear deformation theory for homogeneous monoclinic plates. Acta Mech. **94**(3–4), 195–220 (1992). doi:10.1007/BF01176650

Terrones, M.: Science and technology of the twenty-first century: synthesis, properties, and applications of carbon nanotubes. Annu. Rev. Mater. Res. **33**(1), 419–501 (2003). doi:10.1146/annurev.matsci.33.012802.100255

Thai, H.-T., Vo, T.P.: A nonlocal sinusoidal shear deformation beam theory with application to bending, buckling, and vibration of nanobeams. Int. J. Eng. Sci. **54**, 58–66 (2012). doi:10.1016/j.ijengsci.2012.01.009

Touratier, M.: An efficient standard plate theory. Int. J. Eng. Sci. **29**(8), 901–916 (1991). doi:10.1016/0020-7225(91)90165-Y

Wang, L.: A modified nonlocal beam model for vibration and stability of nanotubes conveying fluid. Phys. E **44**(1), 25–28 (2011). doi:10.1016/j.physe.2011.06.031

Wang, H., Dong, K., Men, F., Yan, Y.J., Wang, X.: Influences of longitudinal magnetic field on wave propagation in carbon nanotubes embedded in elastic matrix. Appl. Math. Model. **34**(4), 878–889 (2010). doi:10.1016/j.apm.2009.07.005

Yang, F., Chong, A.C.M., Lam, D.C.C., Tong, P.: Couple stress based strain gradient theory for elasticity. Int. J. Solids Struct. **39**(10), 2731–2743 (2002). doi:10.1016/S0020-7683(02)00152-X