

Numerical study on performance enhancement of a square enclosure with multiple hot circular obstacles

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Received: 21 July 2020 / Accepted: 8 March 2021 / Published online: 1 April 2021 © Akadémiai Kiadó, Budapest, Hungary 2021

Abstract

Heat dissipation from thermal enclosures is critical for their endurance and efficiency. Through geometry optimization under natural convection, we can cost-efectively obtain maximum heat dissipation. Hence, the present study numerically investigates the fow and heat transfer characteristics of two hot circular bodies in a 2D simulation. By considering and reviewing several works of literature, a modifcation has been done, i.e., two cylinders are considered inside an enclosure with one of them at a fxed position and the other one at three diferent positions (considered as three diferent cases). So, the fuid (air) movement around the body has been increased and helps in appreciable heat transfer from the body. Our primary concern is as to how the utilization of this geometry will impact the heat transfer rate which we have quantifed in terms of Nusselt and Rayleigh numbers. The most optimum confguration is when one of the obstacles is inclined at the top of the enclosure concerning the other which is vertically symmetric, and this orientation gives an increase of 28.09% in terms of the heat transfer, and when two obstacles are considered inline is when we can witness least favorable conditions. In this orientation, a decrease of 70.08% is observed when compared to the most optimum condition.

Keywords Heat transfer · Fluid dynamics · Natural convection · CFD · Fluid flow

List of symbols

-
- T_h Hot wall temperature (°C)
 T_c Cold wall temperature (°C T_c Cold wall temperature (°C)
L Characteristic Length (m)
- Characteristic Length (m)
- g Gravitational Pull (m/s^2)
- d Diameter of the obstacle (m)
- Nu Nusselt number
- Ra Rayleigh number
- C_p Pressure Coefficient
- p Static pressure at the point where pressure coefficient is being measured
- *p*∞ Free stream static pressure
- p_{O} Free stream stagnation pressure
- ρ_{∞} Free stream fluid density
- *V*_∞ Free stream fluid velocity
- C_f Skin Friction coefficient
- *τ*w Skin shear stress on the surface
- *ρ* Fluid density
- ν Free stream fluid velocity

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- h Convective heat transfer coefficient $(w/k.m^2)$
- k Thermal conductivity (w/k.m)
- v Kinematic Viscosity (m^2/s)
- *β* Thermal expansion coefficient (k^{-1})
- α Thermal diffusivity (m²/s)
- T Surface temperature (K)
- T_{∞} Bulk mean temperature (K)
U Average velocity magnitude
- Average velocity magnitude (m/s)
- U^* Non-Dimensional Velocity
- W Vorticity Magnitude (1/s)
- W^* Non-Dimensional Vorticity
- BC Bottom Center
- BL Bottom Left
- BR Bottom Right
- UL Upper Left
- UR Upper Right

Introduction

Natural convection plays a pivotal role in understanding the flow of heat in the absence of any mechanical device used to enhance the fow rate since the wear caused by this will be overlooked until the damage is irreparable. All electronic

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assemblies nowadays have huge heat dissipation due to their need for compaction and extensive miniaturization. Even though the usage of fans with higher capacities and cooling agents is being promoted but they would signifcantly increase manufacturing costs and production time. Not to mention these components would require special care, adding to the manpower costs. Hence, the most ideal way would be to optimize the geometry is to utilize the features of the components themselves to maximize heat dissipation. This methodology can be applied in choosing an optimum geometry in heat transfer applications and in the placement of cylindrical luminescent sources (tube light) to ensure the endurance of the product. In the placement of condenser coils too, the results obtained from our experiment can fnd use. Solar energy is the dominant renewable energy source in today's generation, and our study of cylinders in enclosures will help us, better understand the heat flow between the liquid carrying tubes in the fat plate collectors. Even the petroleum industry, which employs tanks for the storage of crude oil, requires extremely long tubes for the transportation of fuid. One can optimize the geometry of these tubes to maintain the required temperature. Nuclear systems require the reaction to be constantly cooling to maintain an optimal temperature. Hence, a literature review has been carried out by considering natural convection in an enclosure with and without obstacles.

Saris et al. [\[1\]](#page-17-0) studied the phenomenon of free convection in an enclosure with a sinusoidally varying temperature profle that is on the upper wall for the Rayleigh numbers $10^2 - 10^8$ and found that the fluid recirculation pattern increases in intensity, shifts to the upper walls with the increase in Rayleigh number and the aspect ratio. This subsequently increased the rate of heat transfer. Lee et al. [[2\]](#page-17-1) numerically investigate a cylinder in an enclosure with an aspect ratio of one for the Rayleigh numbers 10^3 – 10^6 and notices peak values of Nusselt number on the cylinder surface and the boundaries when the position of the cylinder is varied horizontally and diagonally. Garrpeters [[3\]](#page-17-2) considered a horizontal cavity flled with inert fuid and took into account the periodic instabilities in the fuid. He analyzed the critical zone temperature, disturbances in steady fow for various spatial orientations and zone depths. Sadegh et al. [[4\]](#page-17-3) investigates isothermal cylinders arranged in vertical arrays for low Rayleigh numbers (Ra) and concluded that there exists an optimum separation distance for maximum possible heat transfer for each array, and the array Nusselt number depends upon diameter ratio, number of cylinders in the array and cylinder spacing. Bahrami and Safkhani [[5\]](#page-17-4) in their study of an "eccentric cylinder with an inner rotating wall using a porous media" were successful in determining a relationship between Rayleigh, Darcy, Richardson number and the heat transfer capability of the object. Marcel et al. [\[6](#page-17-5)] studied two horizontally heated cylinders,

separated vertically in an enclosure with the horizontal walls maintained at the heat sink temperature for Rayleigh numbers $10^3 - 10^6$. The Nusselt number depends on the thermal conductivity ratio of the wall, the fuid enclosed, and the Rayleigh number. The part of the top horizontal surface in the heat dissipation decreases as the Ra number keeps increasing while it's vice versa for the bottom surface. Asan et al. [[7](#page-17-6)] studied two isothermal concentric square ducts and the annular region enclosed by them and computed up to a Ra. number of 10^6 at three different dimension ratios. Kim et al. [[8\]](#page-17-7) considered the domain of a square enclosure consisting of a hot circular obstacle at multiple locations, vertically symmetric to the enclosure. The size and formation of the fuid cell are predominantly determined by the location of the cylinder within the enclosure and the Ra. number. Afzal et al. [[9\]](#page-17-8) analyzed the "effect of spacing on thermal performance characteristics of Li-ion battery cells". When the heat generation was assumed to be uniform, the central battery temperature increased on increase of spacing. Bhowmick et al. [\[10\]](#page-17-9) investigated square and round cylinders surrounded by an enclosure with sinusoidal temperature distribution on the walls. The streamlines start losing their regularity on increasing the wavelength, there's the formation of two cells within the enclosure and one dominates (one near the cold wall) over the other (one near the hot wall) on increase in Ra. number.

TetsuFujii et al. $[11]$ $[11]$, have justified the use of a thick boundary layer around a circular cylinder through the use of relevant equations employing the fnite-diference method. It was found that the effect of convective terms was evident even at $Gr = 10^{-4}$, as witnessed in the radial temperature distribution of the local Nusselt number. Bilgen [[12](#page-17-11)] in his study considered an enclosure with isothermal vertical boundaries and adiabatic horizontal boundaries and accounting for partitions at various lengths and Rayleigh numbers 10^4 – 10^{11} . The flow regime was found to be laminar for Ra up to 10^4 and thereafter turbulent. Dutta et al. [\[13](#page-17-12)] analyzed a porous rhombic enclosure at an inclination angle of 30° for heat transfer and entropy generation characteristics. The bottom and top walls of the enclosure are heated non-uniformly. Nusselt number increases but not before initially decreasing in its value when the phase angle increases. Girish et al. [\[14\]](#page-17-13) studied mixed convection in an upright annular region which is double open-ended and consists of a perfectly conducting baffle in between the inner hot and the outer cylinder. The efect of viscous dissipation was taken into consideration. Littlefeld et al. [\[15\]](#page-17-14) considered a uniform annular section of a comparatively higher aspect ratio. The outer wall was specifed to be isothermal, and constant fux was established. It was found that the changes in the Rayleigh number and heat flux ratio, significantly affect the rate of heat and mass transfer. Lee [\[16](#page-17-15)] examined a horizontal layer of fuid bounded by two parallel plates and an array, periodically arranged square

cylinders enclosed by the layer. The horizontal extent of the enclosure is varied by including more square cylinders and a large aspect ratio is employed to overcome approximation errors. Pishkar et al. [\[17](#page-17-16)] investigated "unsteady natural convection of Newtonian and non-Newtonian fuids in a square enclosure". They placed a "heat source with an oscillating heat fux on the bottom wall of the enclosure". It was found that that the Nusselt number is inversely proportional to the power-law index, and the converse is true for the maximum temperature. Ho et al. $[18]$ $[18]$ examined the fluid flow between two horizontal cylinders to understand the effect of insulated boundaries of the round enclosure wall on the heat transfer and fow feld characteristics in the region enclosed by the horizontal cylinders and the wall. Basak et al. [[19\]](#page-17-18) studied the fow inside a trapezoidal enclosure with two diferent conditions. Minea [\[20](#page-17-19)] analyzed convection and radiation in a rectangular and oval enclosure. The heat was transferred from two lateral panels through natural convection. Mohammed Rabhi et al. [[21\]](#page-17-20) analyzed the heat transfer due to natural convection in an inclined rectangular enclosure with multiple partitions. It was found that the partitions contributed to a decrease in heat transfer within the enclosure since it disrupted the fow feld and the radiative heat. Fusegi et al. [\[22\]](#page-17-21) found that in their study differentially heated sidewalls were seen in the cavity, and a uniform internal heat generation was witnessed. Seyyedi et al. [\[23](#page-17-22)] studied a square cylinder enclosing a circular pipe, meant to model a hot pipe buried in the soil. The reduction in cylinder radius and the Darcy number caused the Nusselt number to reduce. Kim et al. [[24](#page-17-23)] too conducted an immersed boundary study of natural convection in a square enclosure comprising of a circular cylinder for the Rayleigh numbers $10^3 - 10^6$. On the further increase of temperature, the Nusselt number drops due to the formation of secondary vortices at the apex of the cylinder and then reaches its maximum value (for the study) at $Ra = 10^6$. Corvaro et al. [[25](#page-17-24)] conducted an experimental interferometric analysis of natural convection in a square enclosure with partially active cold and hot walls at atmospheric pressure (Pr=0.71). Shadlaghani et al. $[26]$ $[26]$ studied "the serrated fns and their efect on annular spaces of varying cross-sections" and found that the inverted triangular section had the best rate of heat transfer. Paroncini et al. [\[27\]](#page-17-26) conducted an experimental and numerical analysis on natural convective heat transfer in an enclosure with partially active sidewalls and air as a medium. They found that an increase in the number of heat sources has a direct impact on the velocity feld and ultimately increased the heat transfer. Yuan et al. [[28](#page-17-27)] analyzed horizontal concentric annuli of varying inner shapes with the inner and outer shape kept at a constant temperature. They witnessed that increasing space at the top and the presence of corners increase the rate of heat transfer. The surface radiation had a huge role to play in this phenomenon.

Previous studies had made the authors curious regarding the fow feld interactions and natural convection heat dissipation if more than one obstacle existed in an enclosure. In today's components and systems, where there are multiple heat emanating substances inside enclosures, and we are tasked with enhancing the heat dissipation and also keeping power & materials constraints in mind. This triggered the authors to visualize two heated circular obstacles at fve diferent combinations and the fuid interactions between them inside a square enclosure to fnd the most optimum orientation of the heated bodies to ensure maximum heat dissipation.

Problem formulation

In this study, a two-dimensional enclosure with cold walls is studied, and enclosed within are two circular obstacles of constant dimensions, in varying positions. A total of fve cases are considered and of which three are visible in Fig. [1](#page-2-0). One of the obstacles is placed at a vertically symmetric position (Prime obstacle) with the other obstacle (case obstacle) being inline in one of the cases and inclined at 45° (mirrored) in the other two cases. The cases in which the 2nd obstacle is at an inclination are horizontally mirrored to obtain two more cases. The numerical analysis is carried out through ANSYS *Fluent 18* employing suitable assumptions. The flow is steady, 2D, laminar, and incompressible. The physical properties are specified at the bulk mean temperature i.e., $[(T_c+T_h)/2]$. The length of the square enclosure is *5d*, and the diameter of both the obstacles is *d*. The (adiabatic) walls of the enclosure are at the constant cold wall temperature of T_c , and the obstacles are

Fig.1 Physical domain of the study

at a hot wall temperature of T_h , which varies according to the Rayleigh Number as shown in Fig. [1](#page-2-0).

Governing equations and formulae

Appropriate boundary conditions are selected with keeping Fig. [1](#page-2-0) as a reference, and the problem is solved by employing the fnite volume method. The walls of the enclosure and circular obstacles are assigned a no-slip boundary condition and are kept at specifed operating temperatures. The diference in these temperatures establishes the fow feld. The Boussinesq approximation is used to account for the buoyancy efects. To reduce the approximation, 2nd order upwind scheme is employed. We make use of the SIMPLE algorithm to obtain a solution. The residuals set for the convergence criteria for the mass and momentum conservation are 10^{-6} and 10^{-12} for energy.

Continuity equation

$$
\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0\tag{1}
$$

x—Momentum equation

$$
\rho \left(u \frac{\partial u}{\partial t} + v \frac{\partial u}{\partial y} \right) = -\frac{\partial p}{\partial x} + \mu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right) \tag{2}
$$

*y—*Momentum equation

$$
\rho \left(u \frac{\partial v}{\partial t} + v \frac{\partial v}{\partial y} \right) = -\frac{\partial p}{\partial y} + \mu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) + g \beta \left(T - T_{\infty} \right) \tag{3}
$$

Energy equation

$$
u\left(\frac{\partial T}{\partial x}\right) + v\left(\frac{\partial T}{\partial y}\right) = \alpha \left(\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2}\right) \tag{4}
$$

Formulae

$$
T_{\rm h} = \frac{Ra \cdot \alpha \cdot \vartheta}{g \cdot \beta \cdot L^3} + T_{\rm c}
$$
 (5)

$$
Nu = \frac{hL}{k} \tag{6}
$$

$$
C_{\rm p} = \frac{p - p_{\infty}}{\frac{1}{2}\rho_{\infty}V_{\infty}^2} = \frac{p - p_{\infty}}{p_{\circ} - p_{\infty}}
$$
\n(7)

$$
C_{\rm f} = \frac{\tau_{\rm w}}{\frac{1}{2}\rho v^2} \tag{8}
$$

$$
U^* = \frac{U \cdot L}{\alpha} \tag{9}
$$

Fig. 2 Unstructured mesh distribution of computational domain

Table 1 Grid independence study

No. of Elements	Nusselt number (Nu)	Variation in value/%
14,696	14.139	
17,434	14.135	0.028
19,022	14.131	0.028

$$
W^* = \frac{W \cdot L}{U} \tag{10}
$$

Grid generation and grid independence

In this study, due to the curvilinear design of the physical domain and the square enclosure around it, an unstructured mesh has been chosen for better computational results. In order to understand the fow physics and heat transfer on the boundary, we have made use of the edge sizing feature on the wall boundaries and the obstacles (Fig. [2](#page-3-0)). Grid independence study has been conducted in the feld of interest (Table [1\)](#page-3-1), where diferent number of grid elements are used. Convergence was achieved on nearly 14,696 elements, and it was found that successive iterations have the least impact on Nusselt number, which is a signifcant parameter in this study.

Results and discussion

The square enclosure with two hot circular obstacles has been studied in different orientations. Their effects on flow and heat transfer characteristics are depicted and analyzed in detail in this section.

Verifcation of computational study

Kim et al. [[7](#page-17-6)] examined the phenomenon of natural convection caused by a temperature diference between a cold enclosure and a hot inner circular obstacle. He further studied the efect of the location of the obstacle on the heat transfer and fuid fow when the temperature of the bottom wall is varied. The position of the obstacle is varied along the vertical line of symmetry and investigated for verifcation, which is presented in Table [2](#page-4-0). The appropriate velocity and temperature contours are presented in Fig. [3.](#page-5-0)

Ramesh and Venkatesan [\[29](#page-17-28)] experimentally studied natural convection in a square enclosure with air as a medium. The domain comprised of diferentially heated vertical walls and adiabatic horizontal walls. Ramesh and Venkatesan make use of a diferential interferometer placed at the top and bottom walls and the trapezoidal rule to calculate the average heat transfer coefficient. Ramesh and Venkatesan found that the contribution of radiation was negligible to the flow owing to the low emissivity of the hot wall. The boundary conditions played a pivotal role in the fow regime and heat transfer characteristics. Ramesh and Venkatesan also recommended using extra insulation behind the walls to achieve adiabatic boundary conditions. Ramesh and Venkatesan had validated Ostrach's [\[30](#page-17-29)] study of enclosures by comparing their experimentally found Local Nusselt number with the latter's empirical calculation. The present study took this a step further by comparing the numerical values found with previous experimental and empirical results (Fig. [4\)](#page-6-0).

Heat Transfer enhancement of an enclosure with heated circular obstacle [\[8](#page-17-7)]

As one is aware that a hotter fuid is less dense as compared to a colder one and hence is quick to rise to the top in an

enclosure whereas the colder fuid remains at the bottom owing to its mass. This phenomenon is also noticeable in B.S Kim's [\[8](#page-17-7)] study when one analyses the velocity contour of an enclosure consisting of a single heated circular obstacle with no-slip boundary conditions applied to the walls of the enclosure. The rate of heat transfer was quantized in terms of Nusselt & Rayleigh numbers. Mathematically speaking, the Nusselt number is the ratio of convective and conductive heat transfer and a measure of comparison of the amount of heat transfer by the fuid and solid in the domain. Rayleigh number considered essential in characterizing the fow regime and is a product of Prandtl number (the ratio of momentum and thermal difusivity) and Grashof number (the ratio of buoyant and viscous forces). With the increase in the Rayleigh number, fuid transitions from a laminar to turbulent flow. Present study was carried out for three Rayleigh Numbers 10^4 , 10^5 & 10^6 . The pressure and skin friction coefficient maps for all the above cases were also obtained. The former is needed to describe the relative pressures throughout the fow domain, and the latter mathematically is a ratio of the skin shear stress on the surface of the body due to the viscous pressure and the dynamic pressure of the free stream. Once the fuid (air) starts gaining temperature after coming in contact with the obstacle, it starts rising toward the top of the enclosure and since the walls of the enclosure are at a lower temperature, the fuid loses its heat through convection. After which, the fuid moves to the bottom of the enclosure due to an increase in its density, and the entire cycle is repeated (Fig. [3a](#page-5-0)). One can witness a consistency in the fow owing to less variation in the density which in turn can be attributed to the small temperature diference between the walls & the obstacle at a low Ra. number $[10^4]$ (Fig. [3](#page-5-0)b).

As one goes on to increase the same $(10⁵)$, one finds that there's a considerable increase in the magnitude of the vorticity in the upper regions of the enclosure and as the temperature diference increases so does the velocity. Also, the vortex formation is unevenly distributed with the majority of them occurring in the top half, and the fow pattern is still found to be symmetric to the vertical axis. (Fig. [3c](#page-5-0)) The contours are consistent at $10⁶$ too but a considerable increase in the velocity, and vorticity magnitude is noted. The bifurcation of flow moves even further away from the obstacle. The average Nusselt number obtained is very similar to those published in Table [2](#page-4-0), with negligible deviations.

Fig. 3 Velocity (**a**, **b**, **c**) and Temperature (**a′**, **b′**, **c′**) contours [[7](#page-17-6)]

Velocity and Temperature contours of Present Study at Diferent Orientations

Bottom center confguration (inline arrangement at center)

Firstly, when one considers the case of the inline geometry (BC), one can easily point out that a major portion of fuid interaction is restricted to the space between both obstacles. Since both of them are heated and each prevent the heat transfer of the other from occurring smoothly. Hence, they mutually disrupt each other's fow (Fig. [5](#page-7-0)a and a′). At Ra number $10⁴$, the bifurcation of the fluid layers occurs on the top of the upper obstacle ('case') and with marginal space between the two obstacles. As the Ra number (10^5) is increased, a predictable increase in the velocity and vortex formation rate is noticed due to the increased temperature gradient as mentioned earlier (Fig. [5b](#page-7-0) and b′). This is in agreement with H. Asan's study on an annular region between two isothermal square ducts [[7\]](#page-17-6). One is even able to observe pockets of hot regions on either side of the obstacles,

which might indicate that a fraction of fuid particles move downwards after losing their energy to the upper wall. These can absorb the heat from the region formed by the interaction of fuids between the two obstacles. An evident larger bifurcation too is witnessed atop the upper obstacle. At Ra number $10⁶$ there is no such bifurcation owing to the considerable increase in the velocity and vorticity magnitude (Fig. [5](#page-7-0)c and c′). The same could even be stated as the reason for the pockets of spaces formed, where the fuid interaction seems to be absent since the increased velocity does not give time to the downward traveling fuid to come in contact with the hotter air around the obstacles. This phenomenon is even witnessed in J.M Lee's analysis of a square enclosure with cold wall obstacles [\[2\]](#page-17-1)

Bottom inclined confguration ('BR and BL')

When the bottom right configuration (BR) is studied, it is noted that the fow of the fuid is quite evenly distributed throughout the enclosure for the Ra number 10^4 with the

most high-speed regions of space present between both the heated circular obstacles. Upon close observation, it is found that the heated fuid rises from the lower 'case' obstacle. It comes in contact in quick succession with the 'prime' obstacle and then the adjacent walls. This makes the fuid lose energy quite drastically as compared to the previous cases. The air again comes in contact with the lower obstacle which makes it gain temperature (Fig. [6a](#page-8-0) and a'). This rapid transfer of energy occurs through convection, causing the air to difuse almost immediately when it meets the lower obstacle and hence does not make it to the bottom corner, where the lower obstacle is present. At a higher Rayleigh number (10^5) , this process speeds up even more due to the increase in the velocity and vorticity magnitude (Figs. [6](#page-8-0)b and b′). Another inference made is that as the velocity of the fuid inside the enclosure keeps increasing, it doesn't fnd enough time to interact with the regions of enclosure beyond the 'prime' obstacle and hence leading to formation of low-speed regions of air in those areas. This phenomenon is most evident at Ra number $10⁶$, where the region beyond the 'prime' obstacle witnessed almost negligible fuid fow and the temperature contours too are uniform (Figs. [6c](#page-8-0) and 6c′). Similarly, when the orientation is changed from right to left, we could not observe any signifcant variations in the flow physics and heat transfer.

Top inclined confgurations ('UL and UR')

When we consider the upper left configuration('UL'), in surprising contrast to the bottom inclined configurations, we notice that a major proportion of fow circulation occurs between the heated obstacles and the far vertical wall (Figs. [7](#page-9-0)a and a′), which wasn't the case previously at Ra. number $10⁴$. This is remarkable as the fluid can disperse even more uniformly and has a lot of available space to spread out which in turn makes heat transfer more efficient.

But here also one can witness the same successive transfer of energy between hot and cold boundaries which leads to quicker difusion on the lower obstacle ('prime'), though not as quick as in bottom inclined cases. The heated fuid rising from the prime obstacle to the case gets difused toward the open space toward the upper wall and again to the adjacent one. This further pushes it downwards toward the lower obstacle but due to the availability of space the loss and gain of energy are gradual. (Fig. [7b](#page-9-0) and b′) On increasing the Rayleigh number, the rate of heat transfer increased drastically due to the increase in velocity and vorticity magnitude. This further enhances the heat transfer rate. There is hardly any region inside the enclosure where the fow circulation is negligible, though at Ra number 10^6 , (Fig. [7c](#page-9-0) and c') the flow circulation starts to follow a dedicated path and thus creating a pocket by encircling around it. The heat transfer is uniform otherwise, and there are possibilities of more pockets being created at higher Rayleigh numbers. Here also

Fig. 5 Velocity (**a**, **b**, **c**) and Temperature (**a′**, **b′**, **c′**) contours of BC

the top inclined right resembles the same fluid flow and heat transfer characteristics because its axis-symmetric in nature. When the heat transfer rate of UL & UR is compared with the validation study [[8](#page-17-7)], there was an increase of 28.09%. This indicates that we should prefer to arrange heated obstacles at inclined confgurations to obtain an enhanced heat dissipation rate. Furthermore, when the same confguration was compared to the bottom center arrangement, an increase of 70.08% was witnessed. We can gauge from this that the BC configuration would be the least suitable candidate when it comes to selecting the optimum arrangement to maximize heat dissipation.

Orientation efect on local and average pressure coefficients

When it comes to the BC configuration, the local pressure coefficients for all the three Rayleigh numbers are negative and in the range of−0.005 to – 0.045 for the 'case' obstacle (Fig. [8](#page-10-0)). They show a steep decline around 200° of inclination. The values become more negative with the increase of the Rayleigh number. For the 'prime' obstacle one witnessed a contrasting phenomenon. All the values are positive and only increase in magnitude with the increase in the Rayleigh number (Fig. [9](#page-11-0)). They are in the range of $+0.004-+0.0018$ and also witness a steep increase at 200° of inclination.

When it comes to the average pressure coefficient is seen that all the values are negative and positive for the 'case' and 'prime' obstacle, respectively. The magnitude (negative/ positive) only increased with an increase in the Rayleigh number (Fig. [10](#page-12-0)). Hence the static pressure on the surface is found to be less than the free stream pressure for the 'case' obstacle and more for the 'prime' obstacle. This implied that the fuid velocity is more than the free stream velocity on the surface of the 'case' obstacle and doesn't exceed

free stream velocity in the case of the 'prime' obstacle. The values found are similar to other bottom inclined confgurations. When it comes to the local skin friction coefficient, in the 'case' obstacle, the curve is periodic and witnesses the lowest values at around 200° for Ra. numbers 10^4 and 10⁵ but also peaks in the same region when considering Ra. number 10^6 (Fig. [11](#page-12-1)). The values are in the range of 0.0001 to 0.0082 and witness a major increase with the increase in Ra. number. In the 'prime' obstacle, the decline is more

Fig. 7 Velocity (**a**, **b**, **c**) and Temperature (**a′**, **b′**, **c′**) contours of upper left orientation (UL)

prolonged, and the lowest peak obtained is constant for almost 100° (Fig. [12](#page-13-0)). When the average skin friction coefficient is considered, it is seen that the slope of the curve is steep between Ra. numbers 10^4 and 10^5 but there's a drastic increase between 10^5 and 10^6 (Fig. [13\)](#page-13-1). So as the Rayleigh number keeps on increasing and fuid starts transitioning away from a laminar flow, the shear stress on the surface of heated obstacles increases.

In bottom inclined confgurations, since the velocity and vorticity undergo a considerable increase, the bottom corner witnesses' negligible fuid fow and the temperature contours are uniform. The local pressure coefficients for all the three Rayleigh numbers are negative and in the

range of -0.005 to -0.04 for the 'case' obstacle (Fig. [8](#page-10-0)). Although 'BL' shows a steep decline around 200° of inclination, the curve shoots up to around the same value for 'BR' at Ra numbers 10^4 and 10^5 . The values become more negative with the increase of the Rayleigh number. For the 'prime' obstacle, we witness a contrasting phenomenon. All the values are positive and only increase in magnitude with the increase in the Rayleigh number (Fig. [9\)](#page-11-0).

They are in the range of $+0.004-+0.082$ and also witness a steep increase at 200° of inclination for Ra number $10⁴$ and $10⁵$. Although for Ra number $10⁶$ one notices that the pressure coefficient dips at around 100° and then rises

steeply at around 200° in the case of 'BL' but for 'BR', the curve rises and falls. It fnally peaks at around 300°.

When it comes to the average pressure coefficient it is seen that all the values are negative and positive for the 'case' and 'prime' obstacle, respectively, and the magnitude (negative/positive) only increases with an increase in the Rayleigh number (Fig. [10\)](#page-12-0). Hence the static pressure on the surface is found to be less than the free stream pressure for the 'case' obstacle and more for the 'prime' obstacle. This implied that the fuid velocity is more than the free stream velocity on the surface of the 'case' obstacle and doesn't exceed free stream velocity in the case of the 'prime' obstacle. The cause for this is attributed to the efective performance of the confguration. The values found are similar to the above inline confguration.

Orientation efect on local and average skin friction coefcients

For the local skin friction coefficient plot, in the case of the 'case' obstacle, the curve is periodic and witnessed the lowest values at around 200° for Ra. numbers 104 and 105 but also peaks in the same region when considering Ra. number $10⁶$ (Fig. [11\)](#page-12-1). The values are in the range of 0.0001–0.0082 and witnessed a major increase with the increase in Ra. number. In the 'prime' obstacle, the decline is more prolonged, and the lowest peak obtained is constant for almost 100° (Fig. [12\)](#page-13-0).

When the average skin friction coefficient is considered, it is observed that the slope of the curve is steep between Ra. numbers 10^4 and 10^5 but there's a drastic increase between $10⁵$ and $10⁶$ (Fig. [13](#page-13-1)). So as the Rayleigh number keeps on increasing and fuid starts transitioning away from a laminar flow, the shear stress on the surface of heated obstacles increases. The 'case' (bottom) obstacle has the maximum

Fig. 8 a Local pressure coefficients of 'case' obstacle for Rayleigh number $10⁴$ **b** Local pressure coefficients of 'case' obstacle for Rayleigh number 10^5 c Local pressure coefficients of 'case' obstacle for Rayleigh number 10^6

values for every Ra number as the heat fow is restricted between both obstacles and thus increased the shear stress on the bodies. But in the case of the 'prime' obstacle, the skin friction coefficient is intermediate between upper and lower inclined confgurations because of the enhanced heat transfer in the upper inclined confgurations. This phenomenon is witnessed due to the efective performance of the confguration.

Orientation efect on local and average nusselt number

The local Nusselt number $(10^4 \text{ and } 10^5)$ plot of the 'case' (lower) obstacle further strengthens the point of view of rapid transfer of energy, when a serious dip at an angle of inclination around 200° (Fig. [14](#page-14-0)) is witnessed in the curve. The plot conforms to the previous study when 10^6 is considered, which is justifed by the increased velocity and vorticity. The Local Nusselt number is conformal for all three

Rayleigh numbers in the case of 'prime' obstacles with the validation (Fig. [15\)](#page-15-0) results. The computed average Nusselt number values were close to those obtained during the validation, and hence, it can be deduced that this confguration does not give an edge over the others in terms of heat transfer enhancement (Fig. [16\)](#page-16-0). The bottom right confguration 'BR' turned out to be a mirror image of the former case with respect to all the parameters considered for this study. This belief is further strengthened by the fact that the average Nusselt number plot of the same overlaps that of bottom right confguration 'BR' (Fig. [16\)](#page-16-0).

Though one can notice a considerable dip in the local Nusselt number values of the 'prime' obstacle across all Rayleigh numbers as it has considerably way more available space to dissipate its heat through all directions. Since there's no immediate obstruction in the path of its fow (Fig. [15](#page-15-0)). The uniform fow circulation helps in the dissipation of the heat from the 'prime' (lower) obstacle. On

Fig. 9 a Local pressure coefficients of 'Prime' obstacle for Rayleigh number 10^4 **b** Local pressure coefficients of 'Prime' obstacle for Rayleigh number 10^5 c Local pressure coefficients of 'Prime' obstacle for Rayleigh number 10^6

Fig. 10 a Average pressure coefficients of 'case' obstacles for different configurations at various Rayleigh numbers **b** Average pressure coefficients of 'Prime' obstacles for diferent confgurations at various Rayleigh numbers

Fig. 11 a Local Skin friction coefficients of 'case' obstacle for Rayleigh number 10^4 b Local Skin friction coefficients of 'case' obstacle for Rayleigh number 10⁵ **c** Local Skin friction coefficients of 'case' obstacle for Rayleigh number 10⁶

Fig. 12 a Local Skin friction coefficients of 'Prime' obstacle for Rayleigh number 10^4 **b** Local Skin friction coefficients of 'Prime' obstacle for Rayleigh number 10⁵ **c** Local Skin friction coefficients of 'Prime' obstacle for Rayleigh number 10⁶

Fig. 13 a Average skin friction coefficients of 'case' obstacles for different configurations at various Rayleigh numbers **b** Average skin friction coefficients of 'Prime' obstacles for different configurations at various Rayleigh numbers

Fig. 14 a Local Nusselt number of 'Case' obstacles for Rayleigh number 10^4 b Local Nusselt number of 'Case' obstacles for Rayleigh number 10⁵ **c** Local Nusselt number of 'Case' obstacles for Rayleigh number 106

Fig. 15 a Local Nusselt number of 'Prime' obstacles for Rayleigh number 10^4 b Local Nusselt number of 'Prime' obstacles for Rayleigh number 105 **c** Local Nusselt number of 'Prime' obstacles for Rayleigh number 106

Fig. 16 Average Nusselt Number at diferent Rayleigh numbers for different configurations

studying the average Nusselt number, it was found that the values are unimaginably high, and this was quite expected due to the above-stated reasons. The heat transfer rate even further kept increasing drastically when the Rayleigh numbers are increased (Fig. 16). The upper right configuration yields a result that is a mirror image of the upper left confguration for all the parameters considered. Although one noticeable fact is that the (Figs. [13](#page-13-1) and [14](#page-14-0)) local Nusselt number curves give the inverse of those found in UL confguration. This is logical as the fow circulation direction too is mirrored.

Conclusions

By conducting this study, the authors gained considerable insights into what geometry confguration might be suitable or unsuitable to employ in compact electronic components as natural convection is only the ideal solution when it comes to optimizing the heat transfer rate of those compact components. It was realized that the placement of heated circular obstacles directly in the line of each other (BC) should be avoided since they mutually disrupt each other's flow circulation, which acts quite detrimental to the objective of heat dissipation and the average Nusselt number plot approves of this notion. When the confgurations in which the obstacles at the bottom are inclined ('BL' and 'BR') are considered, it is found that the proximity of the enclosing walls to the heated obstacle plays a key factor. It leads to a moderate rate of heat transfer, whose value is quite close to that of the validation study. Quick successive energy transfers are witnessed between the fuid and the obstacle frst, followed by the fuid and the boundary walls and lastly with the obstacle in a repetitive manner. When the confgurations in which the obstacles at the top are inclined ('UL' and 'UR') are considered, a quite signifcant improvement in the heat transfer rate was noted, owing to the gradual successive energy transfer and increased available space for fow circulation. This made way to achieve uniform heat dissipation and the hypothesis is confrmed by the surface Nusselt number plot of UL and UR (which are overlapping), found to be comparatively high with respect to the other confgurations in this study. On analyzing the local pressure coefficients, it was found that in top inclined configurations the values of pressure coefficient are positive for 'case' obstacles and is negative for 'prime' obstacles. A contrasting phenomenon is witnessed in bottom inclined confgurations, where the values are negative for 'case' obstacles and positive for 'prime' obstacles. It was gauged from this that the static pressure is less than the free stream pressure for 'case' obstacles in bottom inclined confgurations and 'prime' obstacles in top inclined confgurations. The local velocity is more than the free stream velocity near these surfaces. Similarly for the 'case' obstacles in top inclined confgurations and 'prime' obstacles in bottom inclined confgurations, the static pressure is more than the free stream pressure leading us to the conclusion that the local velocity is less than the free stream velocity near these surfaces. When the average skin friction coefficient in 'case' obstacles of BC configuration was studied, it was found that the slope of the curve is steep between Ra. numbers 10^4 and 10^5 but there's a drastic increase between $10⁵$ and $10⁶$. So as the Ra. the number kept increasing and the fuid started transitioning away from a laminar fow, the shear stress on the surface of heated obstacles increased. In the 'prime' obstacle, the decline is more prolonged, and the lowest peak obtained is constant for almost 100°.In bottom inclined confgurations, The 'case' (bottom) obstacle had the intermediate values for every Ra. number but the 'prime' obstacle had the least skin friction coefficient since the bottom obstacle prevented the heat from making it to the surface of the upper obstacle. The body, in turn, ended up having a comparatively lower value of skin friction coefficient. In top inclined configurations, The 'case' (bottom) obstacle had the lowest values for every Ra. number but the 'prime' obstacle had the highest skin friction coefficient. Although the heat could impinge on the lower obstacle due to gravity, the fow dissipated way faster to the surrounding space rather than the upper obstacle due to availability of space. The increase in Ra. number lead to an increase in the rate of heat transfer as indicated by the Local Nu. number plots, the magnitude of local pressure coefficients, and the order of local skin pressure friction coefficients.

To quantify heat transfer rate between the least and the most optimum geometry, the study found an increase of 28.09% when the UL confguration is compared with the

validation results and found a staggering 70.08% increase when UL is compared to the least optimum inline configuration (BC).

Acknowledgements The authors would like to thank the Modeling and Computing facilities made available to us by the resource Coordinators at Product Development lab at the Mechanical Engineering Department, SRMIST.

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