

Modifed lattice Boltzmann solution for non‑isothermal rarefed gas fow through microchannel utilizing BSR and second‑order implicit schemes

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Abstract

Thermal microscale gas fow was simulated into a coplanar microchannel was simulated at a broad range of Knudsen numbers. Attempts were made to improve the accuracy of slip velocity on walls using a modifed model with two relaxation times based upon the mesoscopic method. The temperature jump of fuid fow at the wall was captured by a model with a single relaxation time using a second-order implicit method. The Zou–He boundary conditions were employed at both inlet and outlet boundaries, and bounce-back/specular refection distribution functions were applied to the impermeable walls. The non-equilibrium distribution functions were also used as the inlet temperature boundary condition. A fully developed temperature profle was considered at the microchannel outlet. A pressure ratio of 2 was considered in the simulations, and various parameters such as dimensionless pressure, pressure deviation from the linear pressure, dimensionless velocity at various Knudsen numbers, centerline velocity and slip velocity of the fuid, centerline temperature and fuid temperature on the wall, Nusselt number with changing Knudsen and Prandtl numbers, parameter *k* along the microchannel length and C_f Re values were evaluated in the slip and transition flow regimes. The results of the direct simulation Monte Carlo were used to evaluate the correctness of the numerical model. The consistency of the two methods indicated the accuracy of the proposed method.

Keywords Microscale gas fow · Mesoscopic method · Second-order implicit method · Bounce-back/specular refection · Thermodynamics

List of symbols B Molecular slip coefficient BBC Bounce-back BSR Bounce-back/specular refection *c* Lattice speed $(m s^{-1})$ \boxtimes Ouang-Vu Bach bachquangvu@duytan.edu.vn

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Greek symbols

- δt Time step (s)
- δx Step by step (m)
- λ Molecular mean free path (m)
- μ Dynamic viscosity (N s m⁻²)
- *v* Kinematic viscosity $(m^2 s^{-1})$
- *𝛱* Pressure ratio
- ρ Density (kg m⁻³)
- σ TMAC coefficient
- τ_a Antisymmetric relaxation time (based on slip boundary)
- τ_f Symmetric relaxation time for fluid flow
- $\tau_{\rm g}$ Symmetric relaxation time for thermal fluid flow
- $\tau_{\rm s}$ Symmetric relaxation time (based on viscosity)
- *𝛺* Collision operator
- ω Mass factors

Subscripts/superscripts

- l Linear
- out Outlet
- s Symmetric

seq Symmetric equilibrium w Wall

Introduction

Gas flow modeling in micro-/nanoscale is dependent on various parameters, the most important of which include compressibility and rarefaction. These two parameters cannot be ignored in gas fow modeling when the gas molecules–wall interaction cannot be ignored at the molecular scale. For this reason, the impact of fuid molecules–solid wall collisions is more signifcant than intermolecular collisions in the fuid flow. The type of flow is of great importance for the advancement of microfuidic instruments, and numerous investigations have been recently conducted on rarefed gas fow simulation in microchannels of diferent geometries. The gas rarefaction in microfuidic devices considerably infuences the molecular mean free path, which in turn may change the thermodynamic equilibrium in such devices. Rarefaction and non-equilibrium of fuid are measured by the dimensionless Knudsen number $\left(\text{Kn} = \frac{\lambda}{H} \right)$), where λ and *H* are the molecular mean free path and the characteristic length, respectively. Gas flow in microfluidic devices, rarefaction and compressibility are interdependent, and their effects on each other cannot be neglected. This interdependency causes difficulties in the analysis of gas fow in such devices. Conventionally, the compressibility of gas fow is measured concerning its Mach number. When the Mach number of the gas flow approaches one, the compressibility turns into an efective factor in the fuid fow, such that it cannot be ignored for gas flows with a Mach number greater than 0.3. According to the gas kinetic theory, Knudsen, Reynolds and Mach numbers are related as $Kn = \sqrt{\frac{\gamma \pi}{2}}$ 2 $\frac{Ma}{Re}$, and the range of the Reynolds number can be related to fow compressibility or incompressibility [[1](#page-15-0)].

At high Knudsen numbers, the compressibility of gas flow can be ignored only at very low Reynolds numbers. At low Knudsen numbers, the flow can be considered incompressible in a wide range of Reynolds numbers. Gas fow can be considered as a rarefed fow on a small scale and at high altitudes. The gas fow is classifed into four types based on the Knudsen number: (1) *continuous regime* (Kn<0.01): fuid continuity and thermodynamic equilibrium hypotheses hold in the Navier–Stokes equations. It is noteworthy that the Navier–Stokes equations with the no-slip condition can be used in microchannels. (2) *Slip regime* (0.01<Kn<0.1): thermodynamic equilibrium does not hold on the walls and in the Knudsen layer. Moreover, the Navier–Stokes equations (based on the fow continuity assumption) can be used assuming a fuid slip condition on the walls. (3) *Transition regime* (0.1 < Kn < 10): The Navier–Stokes equations do not hold under any condition, and rarefaction plays a key role in this case. The impact of Knudsen number in this regime is significant, posing a challenge for rarefied gas flow simulation in microchannels. The wall function or the modifed Knudsen number based on the gas kinetic theory (because of the viscosity correction) for dealing with this problem should be employed. (4) *Free molecular regime* (Kn>10): The collisions between gas molecules can be disregarded in comparison with gas molecules–solid wall collisions as a key factor in the gas flow modeling $[2]$ $[2]$.

Lattice Boltzmann method (LBM) has recently become a good alternative to direct simulation Monte Carlo (DSMC) for studying rarefed gas fow in microchannels due to its lower computational time and the ability to simulate complex geometries. Numerous investigations have been carried out on gas fow in micro- and nanoscales. Dongari et al. [[3\]](#page-15-2) analytically studied gas fow in a long microchannel and found a direct relationship between the mass fow rate and pressure ratio. In accordance with their results, the normalized mass fow rate initially decreased up to a Knudsen number of 0.886, after which it started to increase. Dongari et al. [\[4\]](#page-15-3) examined the infuence of Knudsen layer variations in micro-/nanoscale on the gas fow. Providing an analytical function called power law, the impact of Knudsen number on transition gas fow was studied with reasonable accuracy.

Arkilic et al. [\[5](#page-15-4)] investigated gas fow in a long microchannel using the perturbation method. In accordance with pressure variations along the streamlines, they concluded that the pressure decreases with raising the Knudsen number, and pressure changes nonlinearly in the slip regime. They also found that the mass fow rate is boosted by growing the pressure. However, they considered the frst-order derivative of velocity in the transverse direction to provide a relation for slip velocity. Armaly et al. [[6\]](#page-15-5) analyzed gas fow in a microchannel with a step both experimentally and theoretically and obtained transverse velocities in this geometry. They obtained the separation distance enhanced by increasing the Reynolds number up to 1200 in the laminar fow regime. A considerable rise in the Reynolds number formed oscillation as well as an increase in the velocity. Graur et al. [\[7](#page-15-6)] studied a relatively rarefied gas flow in a microchannel with a variable cross section both experimentally and theoretically. They found the mass fow rate increased as pressure increased in the nozzle and difuser. Hemadri et al. [[8\]](#page-15-7) studied rarefed gas fow in a microchannel with a non-uniform cross-section experimentally. They proposed a correlation between the average Knudsen number and average rarefaction and reported a Knudsen minimum of roughly 0.9. They showed that the mass fow rate improved by enhancing the angle, whereas it was decreased as the aspect ratio increased.

Sazhin [[9](#page-15-8)] scrutinized the rarefied gas flow in a rough microchannel utilizing the particle-based method called DSMC. In short/long channel, the mass flow rate at a broad range of Knudsen numbers from a free molecular regime to the near hydrodynamic was reported. The roughness has a signifcant impact on the fow rate in the free molecular and transitional regimes. Sabouri and Darbandi [[10](#page-15-9)] investigated the species separation for rarefed gas mixture fow in a micronozzle using the DSMC scheme. They conceived two parameters that have an important effect on the separation of mixture fow, including lateral species separation and streamwise separation. Biswas et al. [[11\]](#page-15-10) studied a step flow at low and moderate Reynolds numbers and found a larger separation length with increasing the Reynolds number. Darbandi and Roohi [\[12](#page-15-11)] evaluated subsonic fow in stepped nanochannels in micro-/nanoscale by employing the DSMC method. According to their outcomes, pressure linearly decreased along the microchannel, and the pressure deviated parabolically from the linear pressure. The dimensionless velocity along *y*-direction (to wall velocity) was higher at higher *x/l* values, and the velocity was higher along the step because of a reduction in the cross section. The velocity also decreased with increasing the Knudsen number. The Mach number for the lower wall along the longitudinal direction decreased from the inlet to the step length and then increased up to the microchannel outlet, but again slightly decreased at the outlet. Hadjiconstantinou and Simek [[13](#page-15-12)] simulated 2D thermal gas fow in a micro-/ nanochannel by DSMC method and calculated the Knudsen number (for constant wall temperature) and C_f ·Re values. They found that C_f : Re declines with the Knudsen number. Mozaffari and Roohi $[14]$ studied the thermally driven flow inside a divergent micro-/nanochannel with divergence angle at the range of 0–7 degrees with two aspect ratios $(L/H=6$ and 20) using DSMC method. They found that while the opening angle is small, the velocity and pressure distributions have a constant treatment approximately. Roohi et al. [[15\]](#page-15-14) extended the 2D solution of Navier–Stokes equations to the transition regime utilizing the information preservation method. They obtained the transverse velocities using the linear Boltzmann method and DSMC, IP (by models 1 and 2) and Beskok methods. The ratio of the mass fow rate obtained from the above methods to that obtained from the Navier–Stokes equations (the parameters) decreased with the Knudsen number at the outlet.

Chen and Tian $[16]$ $[16]$ simulated gas flow in a microchannel applying the LBM and found an increase in the slip velocity by growing the Knudsen number. They presented the variations in dimensionless velocity along the transverse direction of the microchannel and showed that the non-dimensional velocity in terms of inlet velocity rises with growing the Knudsen number. Liu and Feng [[17](#page-15-16)] examined the gas flow through a microchannel by using the cascade LBM. The Bosanquet-type effective viscosity is used to capture the rarefaction efect, and the BSR scheme plays a role as the wall boundary condition. The dimensionless parameters, such as

velocity and mass fow rate, are reported and discussed in detail.

Bakhshan and Omidvar $[18]$ $[18]$ studied fluid flow in a microchannel with a step for a broad range of Knudsen numbers using LBM with multiple relaxation times. They obtained velocities at various cross sections of the microchannel and found that the size of eddies increased with growing pressure ratio in the slip regime. They also found the transverse dimensionless velocities (in terms of average velocity) were inversely related to the Knudsen number. However, the changes in the cross section were represented by the relation $Kn(x) \cdot P(x) \cdot H(x) = \text{const.}$ Ginzburg [[19](#page-15-18)] proposed a model with two relaxation times (TRT) using symmetric distribution functions and symmetric equilibrium distribution function and also for asymmetric functions for the frst time. This model considerably reduced computational cost compared to a model with multiple relaxation times in the LBM. Esfahani and Norouzi [\[20](#page-15-19)] investigated rarefed gas fow in a microchannel with two relaxation times applying the LBM and calculated transverse velocities up to a Knudsen number of 0.338. With the increasing Knudsen number, the slip velocities obtained from the LBM were diferent from those calculated by DSMC method due to neglecting the efect of Knudsen layer in the transition regime and applying the slip velocity up to the frst derivative.

Guo et al. [[21\]](#page-15-20) simulated rarefied gas flow in microscale using the modifed LBM with multiple relaxation times and obtained transverse velocities and mass fow rates at different Knudsen numbers. As reported, the mass flow rate initially decreased and then increased. The model was, however, unable to provide the correct slip velocity and produced diferent results compared to those obtained from the DSMC. He and Luo [[22](#page-15-21)] provided the LBM theory and, additionally, provided an approximate velocity-based distribution function for low Much numbers by studying equilibrium distribution functions and their effective parameters. They also extended the LBM to two and three dimensions. Homayoon et al. [[23\]](#page-15-22) studied rarefied gas flow in a microchannel in a broad range of Knudsen numbers using the LBM. In this research, a single relaxation time was used, and the slip velocity was considered up to the frst order. They reported transverse velocities in the microchannel up to a Knudsen number of 8. However, their model failed to provide the slip velocity with acceptable accuracy, and values obtained from LBM difered from those calculated by the DSMC method. Kharmiani and Roohi [[24\]](#page-15-23) studied rarefed gas fow in a convergent–divergent nano-/micronozzle using two relaxation times in the LBM and presented the diagrams for pressure and slip Mach numbers. Assigning individual slip coefficients for each regime (by trial and error), they obtained a pressure close to obtained outcomes from the direct simulation Monte Carlo. However, the Mach number was obtained with a signifcant deviation on account of disregarding the impact of Knudsen layer in the analytical power-law function. Although ofering a good accuracy for two planar surfaces, this function lacks sufficient accuracy for geometries with a variable cross-section. Lim et al. [[25\]](#page-15-24) examined gas fow in microchannels using the LBM. They reported a relatively parabolic relation for pressure deviation from the linear pressure (δP) and an increase in the slip velocity along the channel. Bounce-back distribution functions were considered for the walls. The apparent diference in the δP was due to the inability of the relaxation time to provide the desired values.

Li et al. $[26]$ $[26]$ simulated transitional flow in a microchannel by the LBM. Using multiple relaxation times and bounce-back/specular refection distribution functions, they reported transverse velocities and mass fow rates in the microchannel in terms of the Knudsen number. The slip velocities were signifcantly diferent from those obtained from the DSMC at high Knudsen numbers (around 4.5). They reported a Knudsen minimum of 0.886. Ahangar et al. [[27\]](#page-15-26) studied the isothermal diluted gas flow in a microchannel with a backward-facing step employing the TRT-LBM and reported pressure and velocity at different points. Different slip coefficients were provided in the LBM for various regimes to obtain values close to those computed by the DSMC approach. According to their results, the size of eddies decreased with increasing the rarefaction parameter, and the length of separation zone approached zero at high Knudsen numbers (in the transition regime). They also investigated the rectangular barrier efect on the rarefed fow properties in the long microchannel utilizing the power-law analytical function approach [[28](#page-15-27)]. Gokaltun and Dulikravich [[29](#page-15-28)] simulated rarefed gas fow in a channel with heat transfer by LBM with single relaxation time. The temperature of the upper and lower walls was ten times that of the inlet temperature. They found the transverse dimensionless temperature (in terms of wall temperature) declined with the rising Knudsen number. The transverse dimensionless velocity (in terms of inlet velocity) also increased with increasing *x/l*. Since a frst-order slip velocity was considered in the simulations, the proposed model showed an acceptable performance up to a Knudsen number of 0.14 and lacked a reasonable accuracy at higher Knudsen numbers, although no validation was performed to evaluate the accuracy of the calculated temperatures. Chen and Tian [\[30\]](#page-15-29) investigated a thermal microscale fow under the slip regime by LBM and observed a higher temperature at the lower wall at high Knudsen numbers at Eckert number $(Ec)=2$. The Nusselt number decreased with increasing the Knudsen number at two diferent *Ec* numbers so that the fow with a higher *Ec* showed a higher Knudsen number. They neglected the higher derivative orders in the calculation of velocity, leading to the low accuracy of the results.

Other works can be addressed in this way, concerning fuid fow and heat transfer at the microscale level via LBM or molecular dynamic (MD) approach [[31](#page-15-30)[–39\]](#page-16-0).

Previous researchers [[23,](#page-15-22) [26](#page-15-25), [29](#page-15-28)] considered single relaxation time and multiple relaxation times for solving the fuid fow. They also utilized three types of non-equilibrium distribution functions for particles in the D2Q9 lattice network to solve the thermal lattice Boltzmann equation against our proposed model. As novel research, the rarefed gaseous fow with heat transfer in a regularshape microchannel was simulated by incorporating the LBM into a modified model (TRT_SRT solver). Two relaxation times, namely symmetric relaxation time and antisymmetric relaxation time, were used for the diluted gas flow by applying the Zou–He and BSR boundary conditions at the inlet–outlet and wall surfaces, respectively. Single relaxation time and the second-order implicit method (used for geometry walls) were adopted to handle the thermal treatment of rarefed fow. Non-equilibrium distribution functions and a fully developed temperature profle were, respectively, considered in inlet and outlet boundary conditions. The acceptable consistency between the DSMC and LBM results shows the accuracy of the proposed model.

Modifed TRT_SRT model

Problem statement

Figure [1](#page-4-0) shows the gas flow in a microchannel with a height of *H*, length of *L* and an aspect ratio (*L/H*) of 20. A pressure ratio ($\overline{II} = \frac{P_{\text{in}}}{P_{\text{out}}}$) of 2 was considered in the simulations. The upper and lower microchannel walls were considered to be at the same temperature $(T_{bottom} = T_{top} = T_w = 0$ Tu), which were lower than the fluid inlet temperature $(T_{in} = 1 \text{ Tu})$. A fully developed temperature profle was considered at the microchannel outlet. The parameters $H = 40$, $L = 800$ and $Pr = 0.7$ were considered for simulation of the gas fow with heat transfer in the

Fig. 1 2D gas fow with heat transfer in a microchannel at a pressure ratio of 2

microchannel, and the dimensionless parameters *X**, *Y**, U^* and T^* were defined as *X/L*, *Y/H, U/U*_{mean} and T/T _{mean}, respectively. U_{mean} and T_{mean} , respectively, indicate the mean velocity and mean temperature.

Governing equations of LB

Lattice Boltzmann simulation includes two steps, namely collision and streaming. The Boltzmann equation is impossible to solve directly due to its complexity and the nonlinearity of the collision operator. To simplify the collision operator, Bhatnagar–Gross–Krook (BGK) proposed a frst-order linear model with a reasonable modeling accuracy [\[40](#page-16-1)]:

$$
\Omega = \omega(f^{\text{eq}} - f) = \frac{1}{\tau}(f^{\text{eq}} - f) \tag{1}
$$

where τ and ω , respectively, represent the relaxation time and collision frequency, f^{eq} is the equilibrium distribution function, also known as Maxwell–Boltzmann distribution function, showing the particle state precisely at the moment of collision. Relaxation time indicates conditions under which the distribution function *f* approaches the equilibrium distribution function (f^{eq}) . Relaxation time is an indispensable parameter in the LBM simulation and should be within the range of 0.5–3 for the model to convergence. The equilibrium distribution function is defned as follows [[40](#page-16-1)]:

$$
feq = \frac{\rho}{(2\pi RT)^{1.5}} \exp\left(-\frac{(\vec{\varepsilon} - \vec{u})^2}{2RT}\right)
$$
 (2)

where $\vec{\epsilon}$ shows the microscopic velocity, ρ , vector *u* and *T* represent macroscopic density, velocity and temperature, respectively, and *R* is the ideal gas constant.

In recent years, both single and multiple relaxation times have been used in the LBM for rarefied gas flow simulations. The model with a single relaxation time lacks sufficient precision in calculating the slip velocity. In contrast, a model with multiple relaxation times suffers from extensive computational times. Ginzburg [[19\]](#page-15-18) provided a model with two relaxation times (TRT). Ahangar et al. [\[41](#page-16-2)] reported the lattice Boltzmann equation with two relaxation times as follows:

$$
f_i(\mathbf{x} + \mathbf{c}_i \delta t, t + \delta t) - f_i(\mathbf{x}, t) = -\frac{1}{\tau_s} \left(f_i^s(\mathbf{x}, t) - f_i^{s, eq}(\mathbf{x}, t) \right)
$$

$$
- \frac{1}{\tau_a} \left(f_i^a(\mathbf{x}, t) - f_i^{a, eq}(\mathbf{x}, t) \right)
$$
(3)

where f_i represents the local distribution functions, f_i^s symmetric distribution functions, $f_{\text{A}}^{\text{s},\text{eq}}$ symmetric equilibrium distribution functions and $f_i^{\text{a},\text{e}\text{q}}$ asymmetric equilibrium distribution functions. Moreover, τ_s denotes symmetric relaxation time (based on corrected viscosity), τ_a asymmetric relaxation time (based on wall boundary conditions),

x particle position, *c*ⁱ discretized velocity vectors for each particle and δt time step. The simulation used the D_2Q_9 grid with time and spatial steps of unity.

The following distribution function, *g*, with single relaxation time, was used to obtain the temperature profles [[40\]](#page-16-1):

$$
g_{\rm i}\big(\mathbf{x}+\mathbf{c}_{\rm i}\delta t, t+\delta t\big)-g_{\rm i}(\mathbf{x}, t)=-\frac{1}{\tau_{\rm s}}\big(g_{\rm i}^{\rm s}(\mathbf{x}, t)-g_{\rm i}^{\rm seq}(\mathbf{x}, t)\big) \tag{4}
$$

To use Eq. ([2\)](#page-4-1) in programming the LBM, its discretized form is written using the Taylor series expansion [[40\]](#page-16-1) as follows:

$$
f_i^{eq} = \omega_i \rho \left[1 + \frac{c_i \cdot V}{c_s^2} + \frac{(c_i \cdot V)^2}{2c_s^4} - \frac{V \cdot V}{2c_s^2} \right]
$$
 (5)

where ρ is density, ω_i mass coefficients for each particle, c_i particle velocity equal to $\frac{\delta x}{\delta t}$, *V* velocity vector and *c*_s sound velocity, which is equal to $\frac{a}{\sqrt{3}}$.

The distribution function, *g*, is used in the simulations to obtain the temperature profles. The equilibrium distribution functions g_i^{eq} are defined as follows [\[40](#page-16-1)]:

$$
g_i^{\text{eq}} = \omega_i T \left[1 + \frac{c_i \cdot V}{c_s^2} \right] \tag{6}
$$

Ginzburg [\[19](#page-15-18)] defned local symmetric distribution functions (f_i^s) , asymmetric distribution functions (f_i^a) and symmetric $(f_i^{s,eq})$ and asymmetric $(f_i^{a, eq})$ equilibrium distribution functions for each particle as follows:

$$
f_i^s = \frac{1}{2} (f_i + f_{-i}), f_i^a = \frac{1}{2} (f_i - f_{-i})
$$

$$
f_i^{s, eq} = \frac{1}{2} (f_i^{eq} + f_{-i}^{eq}), f_i^{a, eq} = \frac{1}{2} (f_i^{eq} - f_{-i}^{eq})
$$
 (7)

The discretized velocities for the particles in the D_2Q_9 grid are defned as follows [\[42](#page-16-3)]:

$$
c_{i} = \begin{cases} (0,0) & i = 0\\ \left(\cos\left[\frac{(i-1)\pi}{2}\right], \sin\left[\frac{(i-1)\pi}{2}\right]\right)c & i = 1,2,3,4\\ \left(\cos\left[\frac{(2i-9)\pi}{4}\right], \sin\left[\frac{(2i-9)\pi}{4}\right]\right) \sqrt{2}c & i = 5,6,7,8 \end{cases}
$$
(8)

Any particle in the lattice grid has its mass coefficients indicating the efect of these particles on the local equilibrium distribution function. These coefficients for the D_2Q_9 grid are expressed as follows [\[27](#page-15-26)]:

$$
\omega_{i} = \begin{cases} \frac{4}{9} & i = 0\\ \frac{1}{9} & i = 1, 2, 3, 4\\ \frac{1}{36} & i = 5, 6, 7, 8 \end{cases} \tag{9}
$$

Fluid properties such as macroscopic density, velocity and pressure can be calculated by the numerical LBM [[40\]](#page-16-1):

$$
\rho = \sum_{i=0}^{8} f_i, \rho V = \sum_{i=0}^{8} c_i f_i, T = \sum_{i=0}^{8} g_i, P = \rho c_s^2
$$
 (10)

As the primary assumption in the LBM, the intermolecular forces are disregarded, and the gas is considered to be an ideal gas. Hence, equating $P = \rho RT$ with Eq. [10](#page-5-0) gives $RT = c_s^2 = \frac{1}{3}$.

The kinematic viscosity (ν) related to the symmetric relaxation time of the gas flow (τ_f) and thermal diffusion (α) related to the relaxation time of heat transfer in gas (τ_{α}) are defned as follows [[40\]](#page-16-1), [\[43\]](#page-16-4):

$$
v = (\tau_{\rm f} - 0.5)c_s^2 \cdot \delta t, \alpha = (\tau_{\rm g} - 0.5)c_s^2 \cdot \delta t \tag{11}
$$

Efective relaxation times

Gas viscosity is a crucial parameter for momentum transfer between gas molecules. By utilizing the gas kinetic theory, the viscosity of gas fow is defned by Guo et al. [[44](#page-16-5)] as follows:

$$
\mu = 1.016 \times \frac{5}{16} \frac{m}{d^2} \sqrt{\frac{RT}{\pi}} \approx 0.5 \rho \bar{c} \lambda, \ \lambda = \frac{m}{\sqrt{2\pi \rho d^2}} \tag{12}
$$

where *m* is molecular weight, *d* molecular diameter and $\bar{c} = \sqrt{\frac{8RT}{\pi}}$ mean molecular velocity. According to Eq. [12,](#page-5-1) the molecular mean free path equals $\lambda = \frac{\mu}{p}$ $\sqrt{\frac{\pi RT}{2}}$, in which the pressure (P) equals ρRT . This equation only holds for unconfned gas fows [\[44\]](#page-16-5). For the gas fows restricted by solid walls, the path of some molecules colliding the walls will be smaller than the mean free path as mentioned by Nie et al. [\[45](#page-16-6)], Guo et al. [[21\]](#page-15-20), Meng et al. [[46\]](#page-16-7) and Tang et al. [[47\]](#page-16-8). In general, the collision between gas molecules and solid walls can be ignored in comparison with intermolecular collisions when the Knudsen number is considerably lower than one. According to Eq. [12,](#page-5-1) viscosity (μ_0) equals $a_0 \rho \bar{c} \lambda$ for such conditions, where $a_0 = 0.5$ and the effective viscosity is equal to μ_0 . For transition gas flow, intermolecular collisions signifcantly decrease with increasing rarefaction effects. Bosanquet interpolated the diffusion coefficients and obtained the Bosanquet-type correlation for rarefed gas flows at low Knudsen numbers $[48]$ $[48]$ $[48]$:

$$
\frac{1}{\mu_e} = \frac{1}{\mu_0} + \frac{1}{\mu_{\infty}}
$$
\n(13)

where μ_e , μ_0 and μ_∞ , respectively, denote the effective viscosity of gas fow in transition regime, slip regime and free molecular regime. Only the collisions between the gas molecules and the solid wall should be regarded in the free molecular regime, in which case $\mu_{\infty} = a_{\infty} \rho \bar{c}H$. According to Beskok and Karniadakis [[49](#page-16-10)], the efective viscosity of

the gas flow in the transition regime is expressed as $\mu_e = \frac{\mu_0}{(1+aKn)} = \frac{a_0 \rho \bar{c} \lambda}{(1+aKn)}$, where $a = \frac{a_0}{a_\infty}$ and a value of 2 was considered for *a* by Yuhong and Chan [\[50](#page-16-11)]. The next objective is to fnd the relationships between Knudsen number (Kn) and symmetric relaxation time (τ_s) to simulate the gas flow in slip and transition regimes. According to the above data, Li et al. [[51](#page-16-12)] proposed the following correlations for slip and transition gas flows:

$$
\tau_{\rm f} = \tau_{\rm s} = 0.5 + \sqrt{\frac{6}{\pi}} N \cdot \text{Kn (Slip flow regime)} \tag{14}
$$

$$
\tau_{\rm f} = \tau_{\rm s} = 0.5 + \sqrt{\frac{\pi}{6}} \frac{N \cdot \text{Kn}}{(1 + a \text{Kn})}
$$
 (Transition flow regime) (15)

where $N = \frac{H}{\delta x}$ represents the number of grids along the characteristic length, *H* is the characteristic length and δx shows the time step. The following single relaxation time was used to obtain the thermal solution [[41\]](#page-16-2). τ_g and τ_f represent the relaxation times related to fluid flow and thermal fluid flow, respectively:

$$
\tau_g = 0.5 + \frac{\alpha}{c_s^2 \cdot \delta t} \tag{16}
$$

Boundary conditions

Boundary conditions used for simulation of rarefed gas fow with heat transfer are of great importance. A common relation called second-order slip boundary condition [\[27](#page-15-26)] and the temperature jump relation for the gas flow $[29]$ were considered in the simulations:

$$
u^{\text{slip}} = B_1 \sigma_v \lambda \left. \frac{\partial u}{\partial y} \right|_{\text{W}} - B_2 \lambda^2 \left. \frac{\partial_u^2}{\partial y^2} \right|_{\text{W}} \tag{17}
$$

$$
T^{\text{jump}} = \phi \left(\frac{2\gamma}{\gamma + 1} \right) \left(\frac{\text{Kn}}{\text{Pr}} \right) \left(\frac{\partial T}{\partial y} \right) \tag{18}
$$

In Eq. [17,](#page-6-0) u^{slip} represents slip velocity and σ_{ν} is the tangential momentum accommodation coefficient (TMAC) defined as $(2-\sigma)$ where $\tau = 1$. The clip coefficients P , and P , regnes σ), where $\sigma = 1$. The slip coefficients B_1 and B_2 , respectively, assume values of 1 and 0.5, 0.8183 and 0.6531, and 1.11 and 0.61 according to Hsia and Domoto [\[52](#page-16-13)], Guo et al. [\[21\]](#page-15-20) and Cercignani [\[53\]](#page-16-14).

If it is equal to or even higher than 1, B_1 is unable to express slip velocity with reasonable accuracy as it directly impacts the calculation of slip velocity. Accordingly, through correcting the first-order Maxwell slip coefficient, Loyalka et al. $[54]$, and Loyalka $[55]$ $[55]$ $[55]$ expressed this coefficient as $B_1 = \left(\frac{2-\sigma}{\sigma}\right)$ σ $(1 - 0.1817\sigma)$. Reported by Li et al. [[51](#page-16-12)], the slip coefficients $B_1 = (1 - 0.1817\sigma)$ and $B_2 = 0.8$ were used in the simulations.

The temperature jump for the fluid on the wall is expressed using the frst-order derivative of temperature (Eq. [18\)](#page-6-1), where γ is specific heat ratio, Pr is Prandtl number and $\phi = \left(\frac{2-\sigma_{\eta}}{\sigma}\right)$ σ ⁷), in which σ_T represents the thermal accommodation coefficient and is usually considered equal to 1 [[56\]](#page-16-17).

Zou–He boundary conditions were considered at the inlet and outlet of the microchannel as follows [[57\]](#page-16-18):

The flow boundary condition at microchannel inlet is given as:

$$
u(0,j) = 1 - \left(\frac{f_0 + f_2 + f_4 + 2(f_3 + f_6 + f_7)}{\rho_{\text{in}}}\right) \tag{19}
$$

$$
f_1 = f_3 + \frac{2}{3} \rho_{\text{in}} u(0, j) \tag{20}
$$

$$
f_5 = f_7 - 0.5(f_2 - f_4) + \frac{1}{6}\rho_{\text{in}}u(0,j)
$$
 (21)

$$
f_8 = f_6 + 0.5(f_2 - f_4) + \frac{1}{6}\rho_{\text{in}}u(0, j)
$$
 (22)

where ρ_{in} is flow density at the inlet, which equals $\rho_{\text{in}} = 3P_{\text{in}}$ due to the pressure ratio ($\overline{II} = \frac{P_{\text{in}}}{P_{\text{out}}} = 2$) used in the problem under study.

The flow boundary condition at the microchannel outlet is given as:

$$
u(n,j) = -1 + \left(\frac{f_0 + f_2 + f_4 + 2(f_1 + f_5 + f_8)}{\rho_{\text{out}}}\right) \tag{23}
$$

$$
f_3 = f_1 - \frac{2}{3} \rho_{\text{out}} u(n,j) \tag{24}
$$

$$
f_7 = f_5 + 0.5(f_2 - f_4) - \frac{1}{6}\rho_{\text{out}}u(n,j)
$$
 (25)

$$
f_6 = f_8 - 0.5(f_2 - f_4) - \frac{1}{6}\rho_{\text{out}}u(n, j)
$$
 (26)

where ρ_{out} is flow density at the outlet, which equals $\rho_{\text{out}} = 3P_{\text{out}}$ for the gas flow.

Bounce-back/specular refection boundary conditions were used for the upper and lower walls [\[58](#page-16-19)].

For the upper wall,

$$
f_4 = f_2 \tag{27}
$$

$$
f_7 = rf_5 + (1 - r)f_6 \tag{28}
$$

$$
f_8 = rf_6 + (1 - r)f_5 \tag{29}
$$

For the lower wall,

$$
f_2 = f_4 \tag{30}
$$

$$
f_5 = rf_7 + (1 - r)f_8 \tag{31}
$$

$$
f_6 = rf_8 + (1 - r)f_7 \tag{32}
$$

Non-equilibrium distribution functions were used as a temperature boundary condition at the microchannel inlet. A fully developed temperature profle was also considered at the microchannel outlet [[40\]](#page-16-1).

The temperature boundary condition at the inlet is:

$$
g_1 = T_{\text{in}}(\omega(1) + \omega(3)) - g_3 \tag{33}
$$

$$
g_5 = T_{\text{in}}(\omega(5) + \omega(7)) - g_7 \tag{34}
$$

$$
g_8 = T_{\text{in}}(\omega(8) + \omega(6)) - g_6 \tag{35}
$$

The temperature boundary condition at the outlet $\left(\frac{\partial T}{\partial x} = 0\right)$ is:

$$
g_i(n) = g_i(n-1), \text{ for } i = 1-8 \tag{36}
$$

The temperature jump of the fuid on the walls is defned as follows. First, the fuid temperature on the wall is calculated using this coefficient, and then the temperature boundary conditions are employed at the microchannel inlet and outlet. The following relationships were obtained using the second-order implicit scheme [\[29](#page-15-28)].

 T_1 and T_2 show the temperature at nodes near the wall surface $(y=0$ and $y=1)$. The temperature boundary condition for the lower wall and the unknown distribution function related to thermal rarefied gas flow at the bottom wall are calculated as follows:

$$
T_{y=0} = \frac{\left[C^*\left(4T_1 - T_2\right) + 2T_w\right]}{\left(2 + 3C^*\right)}\tag{37}
$$

$$
g_2 = T_{y=0}(\omega(2) + \omega(4)) - g_4 \tag{38}
$$

$$
g_5 = T_{y=0}(\omega(5) + \omega(7)) - g_7 \tag{39}
$$

$$
g_6 = T_{y=0}(\omega(6) + \omega(8)) - g_8 \tag{40}
$$

*T*_{m−1} and *T*_{m−2} demonstrate the temperature at *y* = *m* and *y*=*m*−1. The temperature boundary condition for the upper wall is:

$$
T_{\rm y=m} = \frac{\left[C^*\left(4T_{\rm m-1} - T_{\rm m-2}\right) + 2T_{\rm w}\right]}{(2 + 3C^*)} \tag{41}
$$

$$
g_4 = T_{y=m}(\omega(2) + \omega(4)) - g_2 \tag{42}
$$

$$
g_8 = T_{y=m}(\omega(6) + \omega(8)) - g_6 \tag{43}
$$

$$
g_7 = T_{y=m}(\omega(7) + \omega(5)) - g_5 \tag{44}
$$

The temperature jump coefficient (C^*) used in the rarefied gas fow simulation in the microchannel with heat transfer is defned as follows [[29\]](#page-15-28):

$$
C^* = \phi \operatorname{Kn}\left(\frac{2\gamma}{(\gamma+1)\operatorname{Pr}}\right) = (K)\operatorname{Kn}, \quad K = \phi \cdot \frac{2\gamma}{(\gamma+1)\cdot\operatorname{Pr}}, \phi = 1
$$
\n(45)

Guo and Shu [\[40](#page-16-1)] expressed the slip velocity of the gas flow between two planes based on the bounce-back/specular refection boundary conditions for the wall, where fow rarefaction level plays a crucial role in Eq. [46](#page-7-0). They also provided an analytical equation (Eq. [47\)](#page-7-1) for the second-order slip velocity in terms of slip coefficients. Equating the relations from lattice Boltzmann and analytical solutions, the asymmetric relaxation time (based on wall boundary conditions) and the parameter (dependent on the slip coefficient B_1) are obtained from Eqs. ([48](#page-7-2)) and [\(49](#page-7-3)). The asymmetric relaxation time along with the symmetric relaxation time increases the accuracy of slip velocities for the gas flow:

$$
u_{s} = \frac{4(1-r)}{r} \sqrt{\frac{6}{\pi}} Kn + \frac{2\varpi}{\pi (r_{s} - 0.5)^{2}} Kn^{2}
$$
 (46)

$$
us = 4B1Kn + 8B2Kn2
$$
\n(47)

$$
r = \left(1 + \sqrt{\frac{\pi}{6}}B_1\right)^{-1} \tag{48}
$$

$$
\tau_{\rm a} = 0.5 + \frac{\pi B_2 (2\tau_{\rm s} - 1)^2 + 3}{8(2\tau_{\rm s} - 1)}\tag{49}
$$

where $\varpi = 16(\tau_s - 0.5)(\tau_a - 0.5)$.

Results and discussion

Grid independency test

The optimal grid size plays a crucial role in reducing computational costs. The gas flow in the microchannel for obtaining the proper mesh size was simulated under conditions of

aspect ratio (AR) = $\frac{L}{H_0}$ = 20, pressure ratio (PR = \overline{H} = 2) and $\text{Kn}_{\text{in}} = 0.055$ with four mesh sizes of $200 \times 10, 400 \times 20$, 800×40 , and 1600×80 . The velocity was obtained in section $Y^* = 0.5$. According to the results in Table [1,](#page-8-0) the lower difference between the mesh sizes of 800×40 and 1600×80 is acceptable, and the mesh size of 800×40 is selected as the optimal choice reducing the computational cost.

DSMC–linear Boltzmann validations

The test case for validation of fuid fow and thermal fuid fow is diferent. The direct simulation Monte Carlo method was used to evaluate model accuracy. The helium flow in a microchannel with $AR = 4$, $Kn_{in} = 0.055$ and different pressure ratios was considered to evaluate the temperature accuracy according to Table [2](#page-8-1). To examine the accuracy of transverse velocity in the microchannel, a pressure ratio of 2 and an aspect ratio of 20 were used.

Figure [2](#page-9-0) shows the accuracy of the proposed model. Figure [2](#page-9-0)a shows the dimensionless velocity at *y*-direction (relative to the average velocity) for the slip regime (Kn=0.096). The values obtained from LBM are very close to the analytical solution [\[59](#page-16-20)]. In this regime, the infuences of the Knudsen layer are considerably lower than the slip regime, and the symmetric relaxation time in Eq. [14](#page-6-2) was used in the simulations. Figure [2](#page-9-0)b shows the dimensionless velocity in the transverse direction for the transition rarefied gas flow $(Kn=0.667)$. Due to the signifcant impact of the Knudsen layer in this regime and the correctness of the suggested model, the DSMC method [\[15](#page-15-14)] and linear Boltzmann method [[60](#page-16-21)] were used to improve model accuracy. The transverse velocity is parabolic in both slip and transition regimes. The lowest slip velocities were observed on the walls, and the highest velocity was observed at the microchannel centerline. Figure [2c](#page-9-0), d displays the dimensionless temperatures (related to T_{in}) along the microchannel and temperature jumps with diferent heat conditions of wall

Table 1 Velocities at *y*-direction at $Kn_{in} = 0.055$

Mesh sizes			
200×10	400×20	800×40	1600×80
Velocity at $Y^* = 0.5$			
1.238	1.310	1.349	1.367
$Error\%$			
5.816	2.977	1.334	

surfaces (300 *K* and 350 *K*) at an inlet Knudsen number of 0.055. As seen, the dimensionless temperature decreases along the microchannel, and a maximum error of 4% occurs about X^* =0.8. The DSMC method $[61]$ $[61]$ was used to increase the accuracy of dimensionless temperature. The maximum error of temperature jump occurred at a microchannel inlet (Fig. [2d](#page-9-0)), and the value is roughly 7%. The diagrams show the excellent accuracy of the proposed model for gas fow in a microchannel with heat transfer.

Results

In this section, the parameters $AR = 20$, $\Pi = 2$, different inlet Knudsen numbers, $T_{\text{bottom}} = T_{\text{top}} = T_{\text{wall}} = 0$ (Tu), $T_{\text{inlet}} = 1$ (Tu) and Pr=0.7 are considered for the present simulation.

Figure [3](#page-10-0) represents the dimensionless pressure (in terms of outlet pressure) and pressure deviation from the linear pressure as a function of the inlet Knudsen number. Figure [3](#page-10-0)a displays the dimensionless pressure along the microchannel. The pressure ratio of the microchannel is 2. As shown, the dimensionless pressure reduces with the growing inlet Knudsen number. The nonlinearity of the pressure profle decreases as the fow regime changes from the slip to the transition regime so that a linear pressure profle can be considered at an inlet Knudsen number of 0.5. Given the importance of deviation from linear pressure for rarefied gas fow in microchannels, Liu and Guo [[62\]](#page-16-23) expressed the linear pressure as $P_1 = P_{\text{in}} + (x/l)(P_{\text{out}} - P_{\text{in}})$. They also expressed the dimensionless δP in terms of outlet pressure as $\delta P = [P(x) - P_l] / (P_{out})$. Figure [3](#page-10-0)b represents the δP characteristic in terms of inlet Knudsen number with a parabolic trend. The δP parameter declines with the rising inlet Knudsen number. As seen, these two parameters are at a minimum at both the inlet and outlet and reach a maximum at a length of about 0.55.

Figure [4](#page-11-0) displays the transverse velocity profles in terms of inlet Knudsen number, centerline velocity and slip velocities along the *x*-axis. Figure [4a](#page-11-0) shows the dimensionless transverse velocity (in terms of mean velocity) in slip and transition regimes concerning the inlet Knudsen number. With the increasing rarefaction parameter, the velocity decreases along the centerline while slip velocity on the wall increases at a specific length of the microchannel $(x = 200 \text{ lu})$. As seen in Fig. [2](#page-9-0)b, the dimensionless centerline velocity decreases as the fow regime changes from the slip to the transition regime. However, the centerline velocity increases at the microchannel outlet. The gas fow with an inlet Knudsen number of

Fig. 2 Dimensionless velocity profles for **a** slip regime and **b** transition regime. LBM solution compared with Sreekanth analytical solution [[59](#page-16-20)], direct simulation Monte Carlo [[15](#page-15-14)], Ohwada linear

0.075 and a pressure ratio of 2 changes from a slip regime to a transition regime, leading to a decrease in the nonlinearity (relative to the slip regime). Figure [4c](#page-11-0) exhibits the efect of the rarefaction parameter on the dimensionless slip velocity along the microchannel. The slip velocity grows with the rising rarefaction parameter and slightly decreases at the end of the microchannel.

Figure [5](#page-11-1) shows C_f ·Re profiles at different Knudsen numbers for rarefed gas fow in a microchannel. Zhang et al. [[63\]](#page-16-24) introduced these values for a slip gas fow between two parallel plates using an analytical solution. They used the analytical relation $[96/(1+6A_1Kn+12A_2Kn^2)]$ to report these values up to a Knudsen number of 0.1. A_1 and A_2 show the slip coefficient for fluid flow. Due to the lack of generalization for the transition regime, Bakhshan and Omidvar [[18\]](#page-15-17) p r o p o s e d the r e l a t i o n

Boltzmann [[60](#page-16-21)], **c** and **d** dimensionless centerline temperatures, temperature jumps and DSMC solution (Roohi et al. [\[61\]](#page-16-22))

 $C_f \cdot \text{Re} = 3.113 + \frac{2.915}{1+2\text{Kn}} + 0.641 \exp\left(\frac{3.203}{1+2\text{Kn}}\right)$ up to a Knudsen number of 10. This relation has been used in both transition and slip regimes for simulation of rarefed gas fows in microchannels.

Figure [5](#page-11-1)a displays the efect of the rarefaction parameter on C_f Re along the microchannel in the slip regime. As seen, C_f : Re diminishes with improving the inlet Knudsen number. The nonlinearity of the C_f ·Re profiles also diminishes with the growing Knudsen number of the gas flow in the slip regime. Figure [5](#page-11-1)b shows C_f : Re for the rarefied gas flow in the transition regime. As clearly seen, C_f Re declines with the increasing rarefaction parameter. The range of variations for C_f ·Re is larger at lower inlet Knudsen numbers (0.5) compared to large Knudsen numbers $(Kn=1.5)$, and a completely linear trend is observed with the increasing Knudsen

Fig. 3 a Dimensionless pressure profles along the microchannel and **b** pressure deviation from the linear pressure along the longitudinal direction of the geometry at diferent inlet Knudsen numbers

number. The Knudsen number boosts along the microchannel (from the inlet to the outlet) according to the relation $Kn_{out} = Kn_{in} \cdot H$ due to a decrease in the gas pressure $(\Pi = \frac{P_{\text{in}}}{P_{\text{out}}} = 2)$. Consequently, C_f Re decreases in both slip and transition regimes according to the correlation presented for this parameter.

Figure [6](#page-12-0) shows the dimensionless temperature profles (relative to the mean temperature) of the gas fow at diferent microchannel abscisses as a function of Knudsen and Prandtl numbers. In Fig. [6a](#page-12-0), the Knudsen number is constant $(Kn_{in}=0.035)$, and transverse temperature profiles are shown along the microchannel. The dimensionless temperature decreases as the gas fow passes through the microchannel from the inlet to the outlet. Figure [6](#page-12-0)b shows the effect of rarefaction parameter at Kn=1.667 in the transverse direction of the microchannel on the dimensionless temperatures. With the increasing inlet Knudsen number, temperature at *X**=0.02375 increases and reaches 0.0027, 0.00913 and 0.0118 on the upper wall at Knudsen numbers of 0.01, 0.035 and 0.045, respectively. Figure [6c](#page-12-0) shows the impact of Prandtl number on the transverse temperature profles in the microchannel at the inlet Knudsen number of 0.05. As the Prandtl number increases, the temperature at the center in the transverse direction decreases, whereas the temperature on the walls increases. The dimensionless temperature at *X**=0.00875 and *Y**=0 equals 0.00896, 0.00933, 0.0108, 0.0121 and 0.0129 at Prandtl numbers of 0.7, 2, 5, 7.88 and 10, respectively.

Figure [7](#page-13-0) shows the impact of inlet Knudsen number on the dimensionless temperature profles (relative to mean temperature) at the centerline and on the microchannel walls. Figure [7](#page-13-0)a depicts temperature variations at the centerline

along the microchannel as a function of the rarefaction parameter. As seen, the temperature at the centerline rises with the increasing Knudsen number. Figure [7](#page-13-0)b displays the temperature profles of the gas fow on the microchannel walls. When the flow regime type shifts from slip $(Kn_{in}=0.035)$ to transition (according to $\Pi = 2$), the temperature jump profle declines along the microchannel at a given inlet Knudsen number despite the increasing trend of temperature profles with the Knudsen number.

Hadjiconstantinou and Simek [\[13\]](#page-15-12) defned the Nusselt number for gas fow in a microchannel with a fully developed temperature profile as Nu = $\frac{2H\left(\frac{\partial T}{\partial y}\right)}{(T-T)^2}$ $\frac{\sqrt{dy}/w}{(T_w - T_B)}$ using the DSMC method, where T_B (bulk temperature) and *H* represent the bulk temperature and channel width, respectively. In the simulations conducted using the proposed model, a Prandtl number of 0.7 was considered. Figure [8](#page-13-1) indicates the impact of rarefaction, Prandtl number and *k* on the Nusselt number of a rarefed gas fow along the microchannel.

Figure [8](#page-13-1)a shows the Nusselt number at diferent inlet Knudsen numbers along the microchannel obtained from the numerical LBM. As the Knudsen number increases, the Nusselt number decreases along the microchannel. However, the Nusselt number remains unchanged due to a fully developed temperature profle from a certain length up to the end of the microchannel. Figure [8b](#page-13-1) illustrates the impact of Prandtl number on the Nusselt number along the microchannel at an inlet Knudsen number of 0.08. The Nusselt number of rarefed gas fow increases with the increasing Prandtl number. As previously pointed out by Shokouhmand and Esfahani [[64\]](#page-16-25), when the Prandtl number increases signifcantly, the Nusselt number of the gas fow experiences a sharp change at lengths close to the microchannel inlet as shown in Fig. [2](#page-9-0)b

Fig. 4 a Dimensionless transverse velocity profles, **b** centerline velocity and **c** slip velocity in terms of inlet Knudsen numbers

Fig. 5 C_f Re profiles in a slip regime along the microchannel and **b** transition regime at different inlet Knudsen numbers

Fig. 6 a Dimensionless temperature profles in *y*-direction at diferent microchannel lengths, **b** dimensionless temperature profle at diferent inlet Knudsen numbers and **c** the effect of Prandtl number on dimensionless temperature profile in *y*-direction at $Kn_{in} = 0.05$

at the Prandtl numbers 7.88 and 10. Figure [8](#page-13-1)c shows the Nusselt number of the gas fow as a function of *k* along the microchannel. As seen, the Nusselt number decreases with increasing *k*. According to Eq. [45](#page-7-4), *k* is dependent on the specifc heat ratio (*γ*) and Prandtl number.

Mass fow rate and normalized mass fow rates have been reported in terms of Knudsen number [[1\]](#page-15-0), [\[20\]](#page-15-19), [[65](#page-16-26)], and Roohi and Darbandi [[15\]](#page-15-14) proposed the dimensionless mass flow rate as $\text{Mn} = \frac{m}{\left(\rho_{\text{av}} \cdot \sqrt{2RT} \cdot H/2\right)}$, where ρ_{av} is the average density equal to $(\rho_{\text{in}} + \rho_{\text{out}})/2$, *R* gas constant, *T* temperature and *H* the characteristic length of the microchannel. The mass flow rate (m) is defined as $\int_0^H \rho(y) \cdot u(y) \cdot dy$.

Figure [9](#page-14-0) depicts the impact of the pressure ratio on the mass flow rate in terms of Knudsen number. Figure [9a](#page-14-0) shows the normalized mass fow rate up to an inlet Knudsen number of 10 for the gas fow obtained from the LBM and DSMC methods. The mass fow rate decreases up to an inlet Knudsen number of about 0.9 and then increases. This infection point is defned as the Knudsen minimum. Gavasane et al. [\[66\]](#page-16-27) reported an inflection point of $\delta_{\rm m} = 1$ based on the average rarefaction parameter ($\delta_{\rm m}$) from the relation $\delta_{\rm m} = \sqrt{\pi}/(2Kn_{\rm m})$. Figure [9](#page-14-0)b illustrates the three pressure ratios effects on the mass flow rate. As seen in Table [3](#page-14-1), the normalized mass flow rate of the gas in the microchannel rises with the rising pressure ratio. An aspect ratio of 20 was considered for the microchannel in Fig. [9](#page-14-0)a, b.

Fig. 7 a Dimensionless temperature profles at microchannel centerline at diferent rarefaction parameters and **b** temperature jump in the gas flow at different inlet Knudsen numbers along the microchannel

Fig. 8 a Nusselt profles at diferent inlet Knudsen numbers along the microchannel, **b** the efect of Prandtl number on the Nusselt number at $Kn_{in} = 0.08$ and **c** effect of *k* on the Nusselt number in the rarefied gas flow

Fig. 9 a Normalized mass fow rate in terms of Knudsen number obtained from LBM and DSMC methods, **b** pressure ratio efect on the dimensionless mass fow rate at diferent inlet Knudsen numbers

Conclusions

Non-isothermal microscale gas flow $(T_w < T_{in})$ was studied by a modifed TRT_SRT model based on the LBM. A symmetric relaxation time was separately used in each regime. In the transitional regime, for considering the efect of the Knudsen layer, a single relaxation time based on viscosity correction (Bosanquet-type) was used. Bounce-back and specular-boundary conditions were considered at walls, and the accuracy of slip velocities was improved using the asymmetric relaxation time obtained based on this boundary condition. The fuid temperature jump was obtained by using a second-order implicit scheme and the temperature jump coefficient C^* . The main results are summarized below:

- With the increasing Knudsen number, the nonlinear pressure profles along the microchannel turned into linear profiles, leading to a decrease in the δP .
- As the Knudsen number increased, the transverse dimensionless velocity decreased at the centerline, while the slip velocities on the wall increased.
- The Cf·Re profles in the slip and transition regimes were, respectively, nonlinear and linear and decreased with the rising rarefaction parameter.
- With the increasing Prandtl number, the transverse dimensionless temperature decreased relative to the inlet temperature but increased on the walls.
- The dimensionless temperature at the centerline (relative to the average temperature) increased with the increasing Knudsen number. The fuid temperature on the wall decreased from the slip regime to the transition regime with the increasing Knudsen number. Additionally, this temperature showed an increasing trend in the transition regime.
- The Nusselt number declined with the rising rarefaction parameter, assuming a fully developed temperature profle for the gas fow. The Nusselt profles increased with the increasing Prandtl number.
- For different Knudsen numbers at the inlet, the normalized mass fow rate profle experienced a decreasing trend up to a point with the Knudsen minimum, after which it starts an increasing trend reaching $\delta_{\rm m} = 1$.

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