

# Families of periodic orbits in the restricted four-body problem

A.N. Baltagiannis · K.E. Papadakis

Received: 16 May 2011 / Accepted: 14 June 2011 / Published online: 30 June 2011  
© Springer Science+Business Media B.V. 2011

**Abstract** In this paper, families of simple symmetric and non-symmetric periodic orbits in the restricted four-body problem are presented. Three bodies of masses  $m_1$ ,  $m_2$  and  $m_3$  (primaries) lie always at the apices of an equilateral triangle, while each moves in circle about the center of mass of the system fixed at the origin of the coordinate system. A massless fourth body is moving under the Newtonian gravitational attraction of the primaries. The fourth body does not affect the motion of the three bodies. We investigate the evolution of these families and we study their linear stability in three cases, i.e. when the three primary bodies are equal, when two primaries are equal and finally when we have three unequal masses. Series, with respect to the mass  $m_3$ , of critical periodic orbits as well as horizontal and vertical-critical periodic orbits of each family and in any case of the mass parameters are also calculated.

**Keywords** Asymptotic orbits · Four-body problem · Periodic orbits · Stability

## 1 Introduction and equations of motion

The objective of our paper is the investigation, numerically, of the families of simple (crossing the synodical line twice in one period) periodic orbits of a body of negligible mass studying its motion under the Newtonian gravitational attraction

of three bodies with masses  $m_1$ ,  $m_2$  and  $m_3$  which move, with the same angular velocity, in circular orbits around their center of mass fixed at the origin of the coordinate system. In the general problem of three bodies there is a particular solution in which the bodies lie at the vertices of an equilateral triangle, each moving in a Keplerian orbit. This is well known, and was first given by Lagrange in 1772. He found three-body motions in which the mutual distances are constant, and Euler extended them and found solutions in which the ratios of mutual distances are constant. Although Lagrange thought his equilateral triangle solutions were of no great practical significance, it was later realized that the Sun, Jupiter and the Trojan asteroids formed such a configuration in our Solar system. Many scientists, among others, Andoyer (1906), Lindow (1922, 1923), Schaub (1929), MacMillan and Bartky (1932), Hüttenhain (1933), Pedersen (1944, 1952), Brumberg (1957), Palmore (1973), Simo (1978), Majorana (1981), Álvarez-Ramirez and Vidal (2009), Baltagiannis and Papadakis (2011), and references therein, studied the four-body problem.

The same problem, in various versions, has been used by many scientists for practical applications such as, among others, Van Hamme and Wilson (1986), Kloppenborg et al. (2010) ( $\epsilon$  Aurigae system), Melita et al. (2008), Robutel and Gabern (2006) (Sun, Jupiter and Saturn system), Schwarz et al. (2009a) (star, two massive planets and a massless Trojan), Schwarz et al. (2009b) (star, brown dwarf, gas giant and a massless Trojan), Ceccaroni and Biggs (2010) (Sun, Jupiter, Trojan Asteroid, Spacecraft), and references therein.

We consider that three particles of masses  $m_1$ ,  $m_2$  and  $m_3$  always lie at the vertices of an equilateral triangle and one of them, say  $m_1$ , is on the positive  $x$ -axis at the origin of time. The motion of the system is referred to axes rotating with uniform angular velocity. The three bodies move in the same plane and their mutual distances remain unchanged

---

A.N. Baltagiannis · K.E. Papadakis (✉)  
Department of Engineering Sciences, Division of Applied  
Mathematics and Mechanics, University of Patras, Patras, 26504,  
Greece  
e-mail: k.papadakis@des.upatras.gr

A.N. Baltagiannis  
e-mail: abalt@upatras.gr

with respect to time. The motion of the primaries consists of circular orbits around their center of gravity.

The equations of motion of the infinitesimal mass of the restricted four-body problem, in the usual dimensionless rectangular rotating coordinate system  $(x, y, z)$  are written as Moulton (1900),

$$\begin{aligned} \ddot{x} - 2\dot{y} &= \frac{\partial\Omega}{\partial x} = x - \sum_{i=1}^3 \frac{m_i(x - x_i)}{r_i^3}, \\ \ddot{y} + 2\dot{x} &= \frac{\partial\Omega}{\partial y} = y - \sum_{i=1}^3 \frac{m_i(y - y_i)}{r_i^3}, \\ \ddot{z} &= \frac{\partial\Omega}{\partial z} = - \sum_{i=1}^3 \frac{m_i(z - z_i)}{r_i^3}, \end{aligned} \tag{1}$$

where dots denote time derivatives while the gravitational potential  $\Omega$  in synodic coordinates is defined as

$$\Omega = \frac{1}{2}(x^2 + y^2) + \frac{m_1}{r_1} + \frac{m_2}{r_2} + \frac{m_3}{r_3}, \tag{2}$$

and the distance of the fourth particle from each of the three primaries is:

$$r_i^2 = (x - x_i)^2 + (y - y_i)^2 + (z - z_i)^2, \quad i = 1, 2, 3. \tag{3}$$

The equations of motion admit a Jacobian type of integral,

$$\dot{x}^2 + \dot{y}^2 + \dot{z}^2 = 2\Omega - C, \tag{4}$$

where  $C$  is the Jacobian constant.

If we suppose that the three primary bodies move on the same plane and the axes are so chosen at the origin of time that  $m_1$  is on the  $x$ -axis then the coordinates of the three primaries are:

$$\begin{aligned} x_1 &= \frac{|K|\sqrt{m_2^2 + m_2m_3 + m_3^2}}{K}, \\ y_1 &= 0, \\ x_2 &= -\frac{|K|[(m_2 - m_3)m_3 + m_1(2m_2 + m_3)]}{2K\sqrt{m_2^2 + m_2m_3 + m_3^2}}, \\ y_2 &= \frac{\sqrt{3}}{2} \frac{m_3}{m_2^{3/2}} \sqrt{\frac{m_2^3}{m_2^2 + m_2m_3 + m_3^2}}, \\ x_3 &= -\frac{|K|}{2\sqrt{m_2^2 + m_2m_3 + m_3^2}}, \\ y_3 &= -\frac{\sqrt{3}}{2} \frac{1}{m_2} \sqrt{\frac{m_2^3}{m_2^2 + m_2m_3 + m_3^2}} \end{aligned} \tag{5}$$

where we have abbreviated

$$K = m_2(m_3 - m_2) + m_1(m_2 + 2m_3). \tag{6}$$

## 2 Case of equal primary bodies

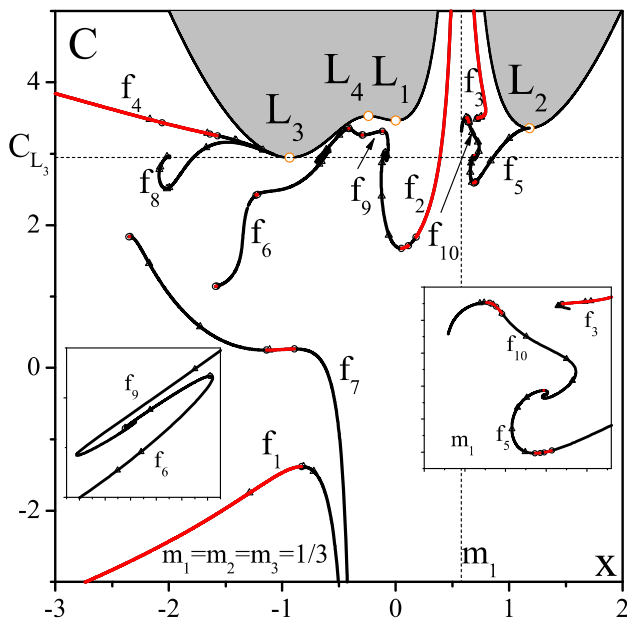
While in the classical restricted three-body problem there are five coplanar equilibrium points, here in the restricted four-body problem (Lagrangian equilateral triangle configuration) the existence as well as the number of the collinear and the non-collinear equilibrium points depends on the particular values of the mass parameters  $m_{1,2,3}$  of the primary bodies (Baltagiannis and Papadakis 2011). In our present case where all the primaries have the same mass  $m_1 = m_2 = m_3 = 1/3$ , the problem admits four collinear (on the  $x$ -axis) equilibrium points and six non-collinear (off the  $x$ -axis) ones. All these equilibrium points are unstable (for details see Álvarez-Ramirez and Vidal 2009 and Baltagiannis and Papadakis 2011).

In this section we will present the network of ten planar symmetric (with respect to  $x$ -axis) simple-periodic orbits of the problem when the masses of the primaries are equal. We have chosen the ten families which correspond (qualitatively) to the most basic families of the classical Copenhagen three-body problem (see Szebehely 1967, p. 455). By symmetric simple-periodic orbits we mean the simplest solutions which have just two perpendicular intersections with the horizontal  $x$ -axis.

All calculations reported in this paper were performed using the variable step R-K 8th-order direct integration and settings the allowable energy variation  $\Delta C = |C_{start} - C_{end}| < 10^{-12}$  and  $|x_0 - x_T| < 10^{-8}$  (initial and final conditions at  $t = 0$  and  $t = T$ ).

In Fig. 1 we present the network of ten families of the simple symmetric periodic orbits for equal masses of the primary bodies. The presentation is made in the  $(x, C)$  plane, where the initial conditions generating closed solutions re-entering after one oscillation, are shown as characteristic periodic family curves. We note here that we have considered the symmetric periodic orbits as represented by their initial conditions  $x_0, y_0 = 0, \dot{x}_0 = 0$  and  $\dot{y}_0 > 0$  i.e. “positive” perpendicular intersections of the  $x$ -axis. The position of the primary body  $m_1$  is denoted by a vertical dashed line. Small red circles indicate the positions of the equilibrium points  $L_i, i = 1, \dots, 4$ . The shaded area is non-accessible to motion due to the Jacobian integral.

At the same time we study the linear stability of the periodic solutions of the problem. The horizontal and vertical isoenergetic stability of each periodic orbit of the ten families was computed using the stability parameters  $a_h$  and  $a_v$ , defined by Hénon (1965). According to this paper the condition for a symmetric periodic orbit to be horizontal or/and vertical stable is  $|a_h| < 1$  or/and  $|a_v| < 1$  respectively. For high accuracy (i.e. the accuracy of the numerical integration, which in the present work is 12 significant figures), we integrated the equations of motion simultaneously with equations of variation. In Fig. 1 the stability arcs of the families are presented by red lines.

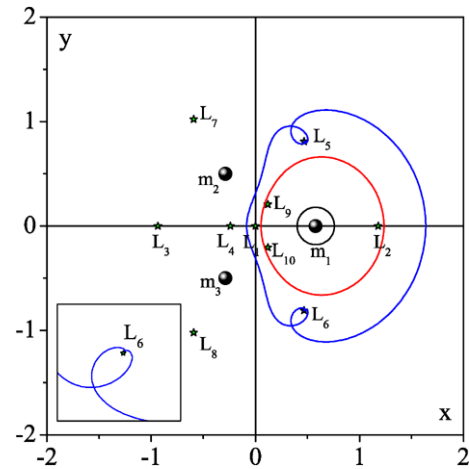


**Fig. 1** The characteristic curves of ten families of the simple symmetric periodic orbits for  $m_1 = m_2 = m_3 = 1/3$ . Red lines indicate the horizontal stability arcs of the families. The small black circles and triangles show the horizontal and the vertical critical periodic orbits correspondingly

Finally, we calculated the special generating planar orbits i.e. the horizontal and vertical critical periodic orbits of each family. A periodic orbit is a horizontal-critical or vertical-critical periodic orbit if the stability index  $|a_h| = 1$  or  $|a_v| = 1$  correspondingly (for details see Hénon 1973) and from them we have, generally, the opportunity to calculate other families of planar or three dimensional periodic orbits. The small black circles and triangles indicate the horizontal and the vertical critical periodic orbits of the families correspondingly.

Family  $f_1$  consists of symmetric simple periodic orbits around the three primary bodies. These are retrograde orbits in the rotating as well as in the fixed system like family  $m$  of the classical three-body problem. Family  $f_1$ , from one side, has orbits where tend to collision with the primaries while on the other side the periodic orbits grow away from the three primary bodies as the Jacobian constant decreases. A large part of the family is stable (red line in Fig. 1) while one horizontal critical periodic orbit and five vertical ones exist in the part of the family where we have calculated (the family exist and for  $C < -3$  which is out of the region  $-3 \leq C \leq 5$  in which we study the families in this paper).

The periodic orbits of family  $f_2$  are retrograde orbits around the primary body  $m_1$  like family  $h$  of the classical three-body problem. The characteristic curve of this family, from one side, evolves inside the funnels of the  $(x, C)$  diagram, that are formed between the zero velocity boundaries, and always increases as  $C$  increases. From the other

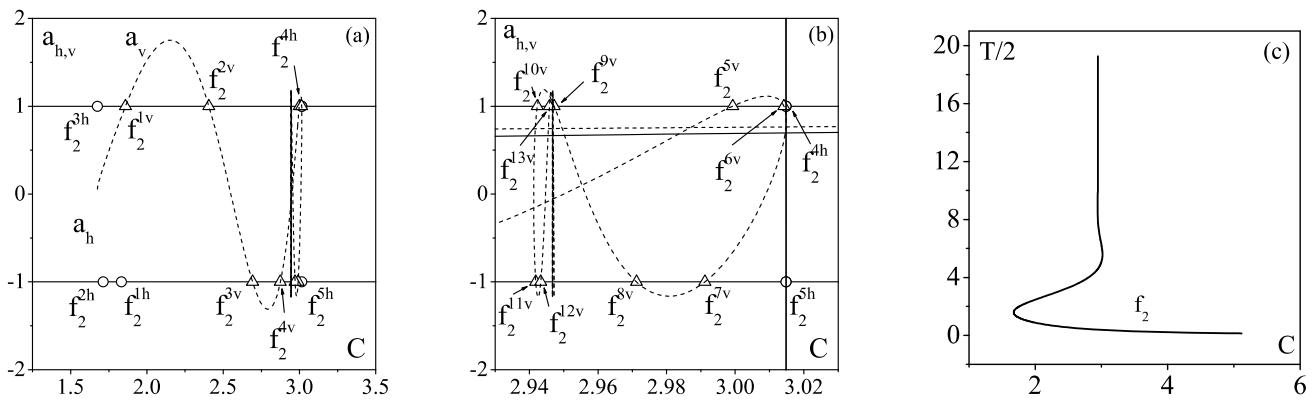


**Fig. 2** Three symmetric periodic orbits of family  $f_2$  for  $m_1 = m_2 = m_3 = 1/3$ . The “end” of the family is a heteroclinic asymptotic orbit on  $L_5$  and  $L_6$  (blue line). The small black points and the green stars denote the positions of the primary bodies and the equilibria of the problem correspondingly

side, spirals around point in the  $(x, C)$  plane which has ordinate  $C = C_{L(3,5,6)} = 2.94673$  (Fig. 1). So, the termination members of the family are asymptotic orbits which intersect the  $x$ -axis perpendicularly and tend asymptotically to  $L_5$  for  $t \rightarrow +\infty$  and to  $L_6$  for  $t \rightarrow -\infty$ , spiralling into (and out of) this point (Fig. 2). The termination of this family has a characteristic behaviour. As the period tends to infinity, the energy of the orbits oscillates in a small shrinking interval around the energy of the asymptotic orbit which is the energy of the equilibrium point  $L_{5(6)}$ . At each change in the sign of the derivative of the energy, the stability of the orbit changes. This type of termination has been called a “blue sky catastrophe” by Devaney (1977).

In the frames (a) and (b) of Fig. 3 we present the stability of the family  $f_2$  i.e. we plot the horizontal (solid line) and the vertical (dashed line) stability indices versus the Jacobian constant  $C$ . The second figure is an enlargement of the first one, around the value of  $C = C_{L_{3,5,6}}$  where the “blue sky catastrophe” termination of the family is occurred. Many horizontal ( $f_2^{ih}$ ) and vertical-critical ( $f_2^{iv}$ ) periodic orbits of the family are denoted by small circles and triangles correspondingly since the stability of the family changes all the time as the period tends to infinity and the characteristic curve of the family spirals around the point with ordinate  $C = C_{L_{3,5,6}}$  in the  $(x, C)$  plane. The evolution of the period  $T$  of the family  $f_2$  is presented in the third frame of Fig. 3.

Family  $f_3$  has as members direct simple symmetric periodic orbits around the primary body  $m_1$  like family  $g$  of the classical three-body problem. As  $C$  decreases the orbits increase in size and then change multiplicity. The parts of the families where correspond to higher-multiplicity orbits are not considered in this paper. The majority of orbits in family



**Fig. 3** (a) Horizontal and vertical stability diagram of family  $f_2$ . The small circles and triangles indicate the horizontal and the vertical critical periodic orbits correspondingly. (b) Zoomed area around the Jaco-

bian constant  $C = C_{L_{(3,5,6)}} = 2.94673$  where the stability-instability of the family changes all the time as the period tends to infinity (frame (c))

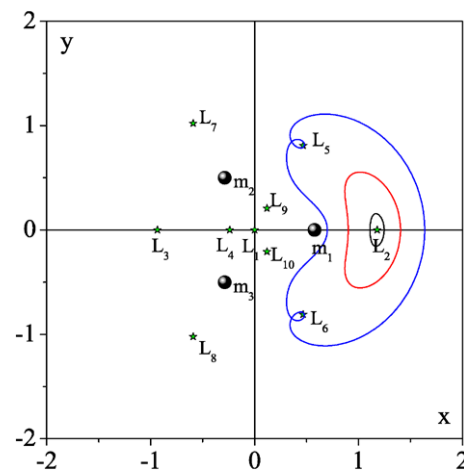
$f_3$  are stable (red line in Fig. 1) and one horizontal and four vertical-critical periodic orbits exist.

The periodic orbits of family  $f_4$  are retrograde in the rotating system and direct in the fixed system, around the primary bodies  $m_1, m_2$  and  $m_3$  like family 1 of the classical three-body problem. This family has periodic orbits, from one side, which grow away from the three primary bodies as the Jacobian constant increases while, from the other side, the family “ends” on the zero velocity curve (Fig. 1). Most of the periodic orbits of the family are stable and there are three horizontal and five vertical-critical periodic solutions.

Family  $f_5$  contains retrograde simple symmetric periodic orbits around the equilibrium point  $L_2$  like family a of the classical three-body problem. The infinitesimal elliptic retrograde orbits around  $L_2$  increase in size, as decreasing Jacobian constant, until a heteroclinic asymptotic orbit on equilibrium points  $L_5$  and  $L_6$  is reached (Fig. 4). The majority of the periodic orbits of this family are unstable but there are horizontal and vertical-critical orbits since the family “ends” under the “blue sky catastrophe” termination. The characteristic curve of the family has two small arcs of stable periodic orbits as the Jacobian constant tends to its lower value (red lines in Fig. 1).

Retrograde periodic orbits around the primary body  $m_1$  of the problem are the members of family  $f_6$ . From one side the family tends to collision with the primaries  $m_2$  and  $m_3$  while from the other side the family terminates with heteroclinic asymptotic orbits on the equilibrium points  $L_5$  and  $L_6$ . The periodic solutions of the family are unstable but two small arcs with stable ones exist (Fig. 1). As the period of the family tends to infinity and the characteristic of the family spirals around a point with  $C = C_{L_3}$ , horizontal and vertical-critical periodic orbits are created.

Family  $f_7$  has retrograde periodic orbits around the primary bodies  $m_2$  and  $m_3$ . The family begins with orbits very close to the two primaries and ends with a collision orbit



**Fig. 4** Three symmetric periodic orbits of family  $f_5$  around the equilibrium point  $L_2$  for  $m_1 = m_2 = m_3 = 1/3$ . The “end” of the family is a heteroclinic asymptotic orbit on  $L_5$  and  $L_6$  (blue line). The small black points and the green stars denote the positions of the primary bodies and the equilibria of the problem correspondingly

with the primary body  $m_1$ . The majority of the periodic solutions of the family are unstable but there are small arcs with stable ones in the middle and close to the end of its characteristic curve (red lines in Fig. 1).

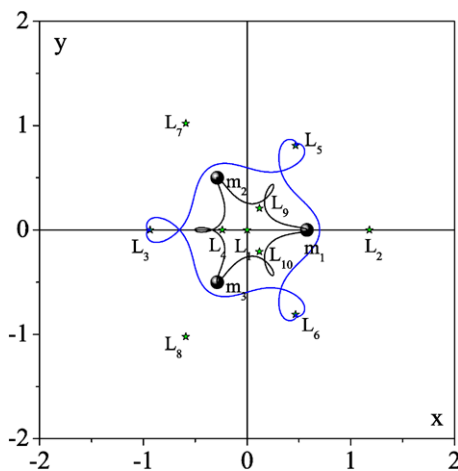
Family  $f_8$  from one side ends on the zero velocity curve while from the other side terminates with heteroclinic asymptotic orbit at  $L_5$  and  $L_6$  and all its members are retrograde periodic orbits around the three primary bodies. The periodic solutions of this family are unstable and we found horizontal and vertical-critical periodic orbits only in its “blue sky catastrophe” termination.

Family  $f_9$  is like family  $x$  of the classical three-body problem and consists of retrograde periodic orbits around the primary bodies  $m_2$  and  $m_3$ . The termination orbits of both sides of this family are heteroclinic asymptotic orbits

at  $L_5$  and  $L_6$ . There are small arcs of its characteristic curve with stable periodic solutions.

Finally, family  $f_{10}$  consists of direct symmetric periodic orbits around the three primary bodies of the problem and corresponds to family  $k$  of the classical three-body problem. The periodic orbits of this family, on both sides, tend to change multiplicity. We continued and calculated the parts of the family with higher-multiplicity and found the termination periodic orbits of this family. Family  $f_{10}$  ends, from one side, with a collision periodic orbit with the three primary bodies (black orbit in Fig. 5) and from the other side terminates with asymptotic orbit at the three equilibrium points i.e. the collinear  $L_3$  and the non-collinear  $L_5$  and  $L_6$  (blue orbit in Fig. 5). Family  $F_{10}$  has two small arcs of its characteristic curve with stable periodic solutions.

In Table 1 we list the initial conditions of the symmetric periodic orbits which are plotted in Figs. 2, 4 and 5. For each orbit listed in this table,  $x_0$  is the initial position of the particle on the  $x$ -axis,  $\dot{y}_0$  is the vertical initial velocity ( $\dot{x}_0 = 0$ ),  $x_{T/2}$ ,  $\dot{y}_{T/2}$  are denoted the same quantities at the half



**Fig. 5** The termination symmetric periodic orbits of family  $f_{10}$  for  $m_1 = m_2 = m_3 = 1/3$ . The *small black points* and the *green stars* denote the positions of the primary bodies and the equilibria of the problem correspondingly

period,  $T/2$  is the half period and  $C$  is the Jacobian constant. In the last column we present the horizontal stability of the periodic orbit.

### 3 Two primary bodies with equal masses

In this section we will study the simple symmetric periodic solutions of the restricted four-body problem when the two primary bodies  $m_2$  and  $m_3$  have equal masses. The existence of this kind of orbits (symmetric with respect to horizontal  $x$ -axis), in the present case, is easily be proved if we insert the transformation  $(x, y, \dot{x}, \dot{y}, t) \rightarrow (x, -y, -\dot{x}, \dot{y}, -t)$  in the equation of motion (1). We verify that these equations remain unchanged under the previous transformation.

It is known, that in the Lagrange central configuration the necessary condition for the stability of the configuration is the inequality,

$$\frac{m_1 m_2 + m_2 m_3 + m_3 m_1}{(m_1 + m_2 + m_3)^2} < \frac{1}{27}, \tag{7}$$

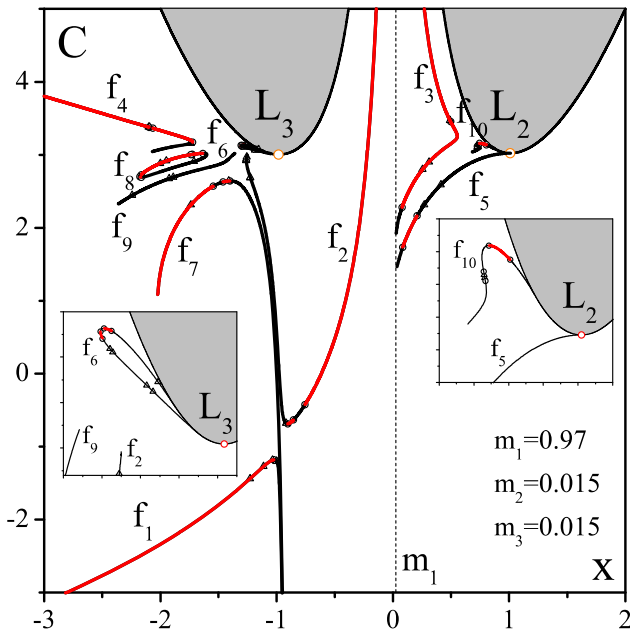
where  $m_1, m_2$  and  $m_3$  the three primary bodies (Gascheau 1843). We choose  $m_1 = 0.97$  and  $m_2 = m_3 = 0.015$  and so the inequality (7) is fulfilled.

And in this case of mass distribution, as in the first one, we calculated the network of the families of the simple symmetric periodic orbits of the problem and we present ten of them. Some of these families, namely  $f_{1,2,3,4,5,7}$ , are the evolution of the old ones (when  $m_1 = m_2 = m_3$ ) and some of them are new ( $f_{6,8,9,10}$ ) since the old ones do not exist any more. In Fig. 6 we plot the characteristic curves of these ten families.

Comparing the Figs. 1 and 6 we see the differences in the forbidden areas (or the trapped areas), in the absence of the collinear equilibrium points  $L_1$  and  $L_4$  as well as in the characteristic curves of the families. According to the paper Baltagiannis and Papadakis (2011) the restricted four-body problem, when  $m_2 = m_3$  admits two or four collinear and four or six non-collinear equilibrium points (depends

**Table 1** Symmetric periodic orbits which are plotted in Figs. 2, 4 and 5

Family	$x_0$	$\dot{y}_0$	$x_{T/2}$	$\dot{y}_{T/2}$	$T/2$	$C$	Stability
$f_2$	0.40050000	1.56088315	0.75770121	-1.53644341	0.36611460	3.05967251	S
	0.05350000	1.34205190	1.23459900	-1.30198016	1.56756877	1.67506327	S
	-0.08435450	0.73239406	1.63558530	-1.01441055	19.27221938	2.94672728	U
$f_5$	1.10950000	0.21881797	1.24602285	-0.20446502	2.32565571	3.33382420	U
	0.90050000	0.93362977	1.40248144	-0.70365211	2.40189387	3.03584294	U
	0.69799469	2.06682108	1.63556565	-1.01438409	14.68467055	2.94673480	U
$f_{10}$	0.57952380	17.46633001	-0.50246977	0.08137413	4.00510710	3.31516544	U
	0.60309000	4.90830820	-0.36515435	-0.00036434	3.74842644	3.47668129	U



**Fig. 6** The network of ten families of the simple symmetric periodic orbits for  $m_1 = 0.97$  and  $m_2 = m_3 = 0.015$ . Red lines indicate the horizontal stability arcs of the families. The small black circles and triangles show the horizontal and the vertical critical periodic orbits correspondingly

on  $m_{2,3}$ ). For  $m_2 = m_3 = m_{crit} = 0.2882762$  the equilibrium point  $L_1$  coincides with  $L_4$  and for values less than  $m_{crit}$  the equilibria  $L_1$  and  $L_4$  do not exist. So, in the present case the problem has 2 collinear and 6 non-collinear equilibria. All the equilibrium points are unstable except the non-collinear equilibria  $L_5$  and  $L_6$  which are stable.

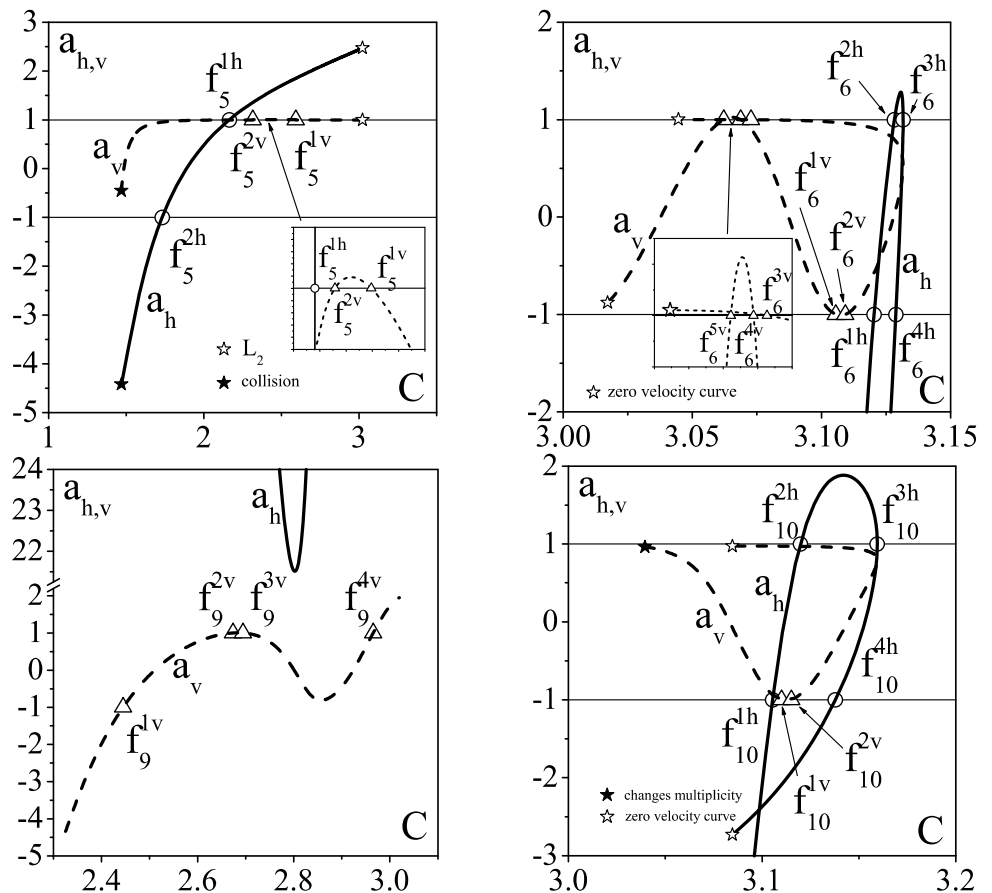
For  $m_1 = 0.97$  and  $m_2 = m_3 = 0.015$  we found that families  $f_1$  and  $f_7$  present, qualitatively, the same behaviour as in the case with three equal primary bodies (both families tend to collision with the primaries  $m_2$  and  $m_3$ ).

Family  $f_2$  now tends to collision with the primaries  $m_1$  and  $m_2$  and it has not simple asymptotic orbits any more.

The evolution of the families  $f_{3,4,5}$ , as  $C$  varies, is different now since  $f_3$  and  $f_5$  tend to collision with  $m_1$  while  $f_4$ , as  $C$  decreases, changes multiplicity. In the first frame of Fig. 7 the stability diagram of family  $f_5$  is illustrated. The two horizontal and the two vertical critical orbits of this family are marked with small circles and triangles correspondingly.

Family  $f_6$  consists of retrograde symmetric periodic orbits around the three bodies  $m_{1,2,3}$  and from both sides terminates on the zero velocity curve (inside zoomed image in Fig. 6). It has two small arcs in its characteristic curve with stable periodic orbits. In this family four horizontal and

**Fig. 7** Horizontal (solid lines) and vertical (dashed lines) stability diagram of families  $f_{5,6,9,10}$  for  $m_1 = 0.97$ ,  $m_2 = m_3 = 0.015$ . The small circles and triangles indicate the horizontal and the vertical critical periodic orbits correspondingly



five vertical-critical periodic orbits exist (second frame in Fig. 7).

Family  $f_8$  has retrograde periodic orbits around  $m_{1,2,3}$ . The characteristic curve of this family is closed. The half periodic orbits of this family are stable (red line in Fig. 6). Six horizontal and four vertical-critical symmetric periodic orbits exist.

The only family where all its periodic orbits are unstable is the family  $f_9$ . Consists of retrograde periodic orbits around the three bodies  $m_{1,2,3}$ . From one side, this family, tends to collision with  $m_2$  and  $m_3$  while from the other side changes multiplicity. Four vertical-critical periodic orbits exist (third frame in Fig. 7).

Finally, the members of the family  $f_{10}$  are direct symmetric periodic orbits around the primary  $m_1$ . The family has stable periodic orbits, four horizontal and two vertical-critical periodic orbits (last frame in Fig. 7). This family terminates, from one side, on the zero velocity curve while from the other side changes multiplicity (inside zoomed image in Fig. 6).

From the horizontal and vertical stability curves of the families  $f_5$ ,  $f_6$  and  $f_{10}$  it is seen that these families consist mostly of orbits which are vertically stable but horizontally unstable.

#### 4 Primary bodies with unequal masses

Collinear equilibrium points in the restricted four-body problem (Lagrangian configuration) with three unequal masses, do not exist. The problem, in this case, admits eight or ten (depends on the mass parameters) non-collinear equilibrium points (for details see Baltagiannis and Papadakis 2011). In this section we study the families of simple periodic orbits when the mass distribution is  $m_1 = 0.97$ ,  $m_2 = 0.02$  and  $m_3 = 0.01$ . The stability condition (7), for these  $m_i$ , is fulfilled. For these values of masses the problem has eight equilibrium points. All equilibria are unstable except  $L_5$  and  $L_6$  which are stable. In this case of the problem, the horizontal  $x$ -axis is not a symmetric axis any more (as in the two previous cases in the present paper). So, in this section we will study the families of simple non-symmetric periodic orbits for the specific values of masses of the three primary bodies.

In the two previous cases we presented the characteristic curves of the families of simple symmetric periodic orbits using the initial conditions of the periodic orbits namely  $\mathbf{x}_0 = \{x_0, y_0 = 0, \dot{x}_0 = 0, \dot{y}_0(C)\}$  while now the characteristic curves of the families of non-symmetric periodic orbits in the  $(x_0, C)$  plane do not define the complete set of the initial conditions of the orbits since they do not provide any information about the values of the horizontal ( $\dot{x}_0 \neq 0$  now) or the vertical component of the velocity of the fourth infinitesimal

body. For comparison reasons we will use the same presentation of the families on the  $(x_0, C)$  plane and we will give additional information about the velocity of the fourth body in appropriate tables of data.

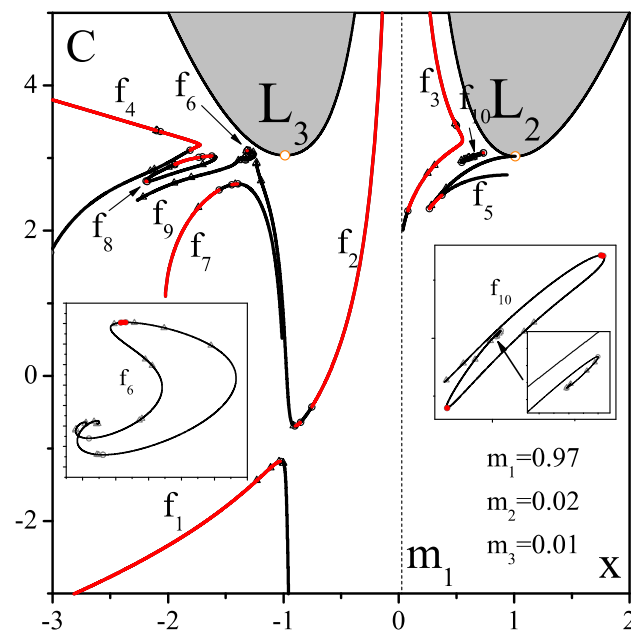
While the determination of the initial conditions of an orbit close to a symmetric periodic orbit is possible using the grid method (Markellos et al. 1974), this is not possible for non-symmetric periodic orbits since the velocity of the fourth body is not vertical with respect to horizontal  $x$ -axis for  $T = 0$ . So, we need to know how a symmetric periodic orbit changes as the mass parameters vary. From the previous section we know ten families of simple symmetric periodic orbits with  $m_1 = 0.97$  and  $m_2 = m_3 = 0.015$ . We can obtain a one-parameter set of horizontal or vertical critical periodic orbits by varying the mass parameter (say,  $m_3$ ). If the parameter is fixed the horizontal (or vertical) critical orbits are isolated. In the present case, we keep constant the mass of the dominant body  $m_1 = 0.97$  and we computed monoparametric sets of horizontal or vertical critical orbits for values of the mass parameter of the third primary  $m_3$  varying in the range  $[0.015, 0.01]$  (the mass of the second primary body is defined by the equation  $m_2 = 1 - m_1 - m_3$ ). So, we determined critical non-symmetric periodic orbits for  $m_1 = 0.97$ ,  $m_3 = 0.01$  and  $m_2 = 0.02$ . We shall call the set of initial conditions and other quantities describing a critical orbit for a range of values of the mass parameters, a bifurcation series or a series of critical orbits. Once a non-symmetric orbit has been found, we can extend the search to the whole one-parameter family to which that orbit belongs. All the families of symmetric periodic orbits  $f_i$ ,  $i = 1, \dots, 10$  of the previous case (two equal primaries) have at least one horizontal critical periodic orbit except family  $f_9$ . In this case we determined the series of one (it has four) vertical critical periodic orbit of family  $f_9$ .

In Table 2 horizontal critical periodic orbits with  $S_h = 1$  for the families  $f_i$ ,  $i = 1, \dots, 8, 10$  and vertical critical ones with  $S_v = 1$  for the family  $f_9$  are given. For each family the first critical periodic orbit is the symmetric critical solution for  $m_1 = 0.97$  and  $m_2 = m_3 = 0.015$  while the second one is the non-symmetric critical solution for  $m_1 = 0.97$ ,  $m_2 = 0.02$  and  $m_3 = 0.01$  correspondingly. In the last column of Table 2 we present the stability index  $S_h = (a_h + d_h)/2$  or  $S_v = (a_v + d_v)/2$ , where  $a_h$ ,  $d_h$ ,  $a_v$  and  $d_v$  are the isoenergetic stability horizontal (vertical) parameters, of each periodic orbit (for details see Hénon 1965, 1973).

Starting with the initial conditions of Table 2, for  $m_1 = 0.97$ ,  $m_2 = 0.02$  and  $m_3 = 0.01$ , we calculated the ten families of simple non-symmetric periodic orbits. In Fig. 8 we present these ten families with their corresponding stable arcs (red lines). We note again that the characteristic curves presented here do not give any information about the velocity of the non-symmetric periodic orbits of the families  $f_i$ ,  $i = 1, \dots, 10$ .

**Table 2** Series, with respect to the mass parameter  $m_3$  ( $m_1 = 0.97$  is fixed and  $m_2 = 1 - m_1 - m_3$ ), of critical symmetric and non-symmetric periodic orbits

Family	$m_3$	$x_0$	$\dot{x}_0$	$\dot{y}_0$	$C$	$T$	Stability index
$f_1$	0.015	-1.02160447	0.00000000	2.04679983	-1.18104196	2.82991261	$S_h = 1.00000000$
	0.01	-1.02506173	-0.03888367	2.05336268	-1.17807072	2.84078467	$S_h$
$f_2$	0.015	-0.75612119	0.00000000	1.89645568	-0.42598534	2.65805735	$S_h$
	0.01	-0.75507284	0.01748971	1.90269870	-0.42825831	2.66038694	$S_h$
$f_3$	0.015	0.49660519	0.00000000	0.97708718	3.45614347	3.19119978	$S_h$
	0.01	0.50001684	0.00161738	0.97122228	3.44521168	3.25098464	$S_h$
$f_4$	0.015	-2.09768018	0.00000000	1.40690658	3.37872619	9.39826718	$S_h$
	0.01	-2.09833939	-0.00119843	1.40823351	3.37878217	9.39762923	$S_h$
$f_5$	0.015	0.20901246	0.00000000	2.92071648	2.16398820	6.24319928	$S_h$
	0.01	0.26503026	0.26031156	2.42511743	2.30234921	6.23835093	$S_h$
$f_6$	0.015	-1.29832006	0.00000000	0.34444543	3.12037989	14.34282809	$S_h$
	0.01	-1.31371897	0.08828457	0.40643256	3.10118167	15.15805445	$S_h$
$f_7$	0.015	-1.54435005	0.00000000	1.05782952	2.57088764	5.64506825	$S_h$
	0.01	-1.55780271	0.01646369	1.08000666	2.55924268	5.70518828	$S_h$
$f_8$	0.015	-1.72477906	0.00000000	1.06562373	3.00644003	18.17937805	$S_h$
	0.01	-1.70207170	0.00374697	1.03334713	3.01572943	18.04783658	$S_h$
$f_9$	0.015	-2.24258103	0.00000000	1.86532152	2.44520826	17.63176140	$S_v = 1.00000000$
	0.01	-2.21748105	0.03965238	1.83278958	2.46324526	17.58806770	$S_v$
$f_{10}$	0.015	0.73304548	0.00000000	0.46045805	3.10542112	9.37317852	$S_h$
	0.01	0.73452874	-0.14568935	0.47286556	3.07075149	10.71055379	$S_h$

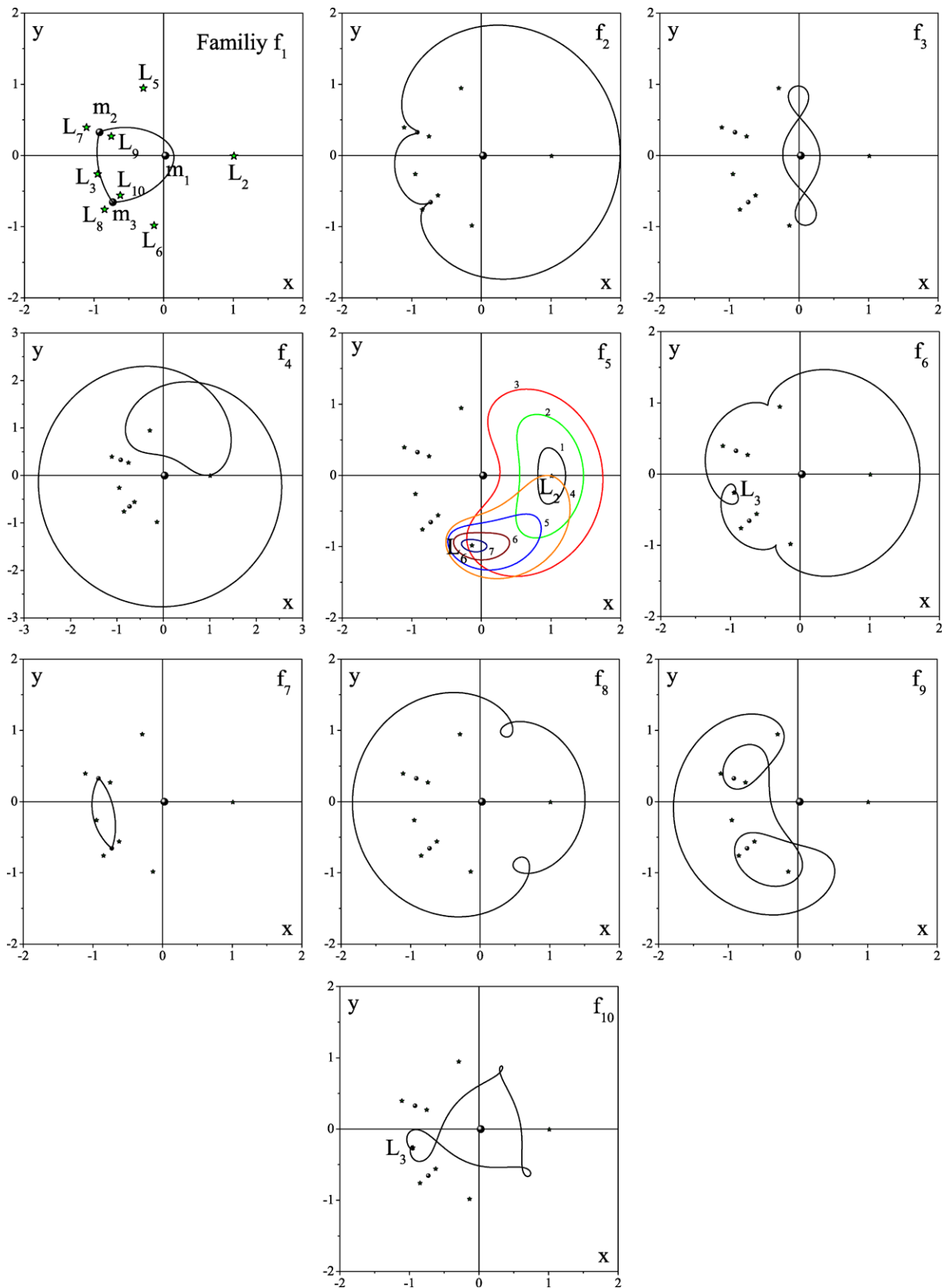
**Fig. 8** The characteristic curves of ten families of the simple non-symmetric periodic orbits for  $m_1 = 0.97$ ,  $m_2 = 0.02$  and  $m_3 = 0.01$ . Red lines indicate the horizontal stability arcs of the families. The small black circles and triangles show the horizontal and the vertical critical periodic orbits correspondingly

The network of the ten families of periodic orbits we found is quite close to this network of families in the case where the two primaries are equal ( $m_2 = m_3 = 0.015$ ) but with a significant difference i.e. in the present case all the periodic solutions are non-symmetric. Namely, families  $f_{1,2,3,4,7,8,9}$  have the same, qualitatively, behaviour as in the previous case while families  $f_{5,6,10}$  are evolved different.

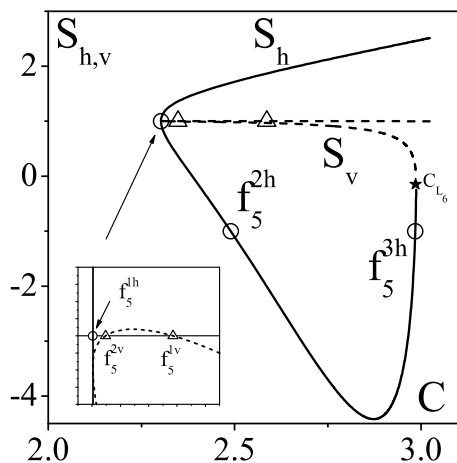
Family  $f_5$  consists of simple retrograde non-symmetric periodic orbits which some of them are Trojan type orbits (generally not crossing the synodical line). From one side has solutions around the equilibrium point  $L_2$  and from the other side has as members periodic orbits around the stable equilibrium point  $L_6$  ( $f_5$  in Fig. 9). The evolution of this family is like the short-period asymmetric family emanating from the Lagrangian equilibrium point  $L_5$  and terminates on a symmetrical periodic orbit belonging to the family  $b$ , which family  $b$  emanates from the collinear equilibrium point  $L_3$  in the classical restricted three-body problem (Deprit et al. 1967). Family  $f_5$  has stable periodic orbits and three horizontal and two vertical critical non-symmetric periodic solutions.

Family  $f_6$  has retrograde non-symmetric periodic orbits around the three primaries and now, in this mass distribution, from both sides terminates with solutions where are asymptotic orbits at the equilibrium point  $L_3$  (Fig. 8, inside

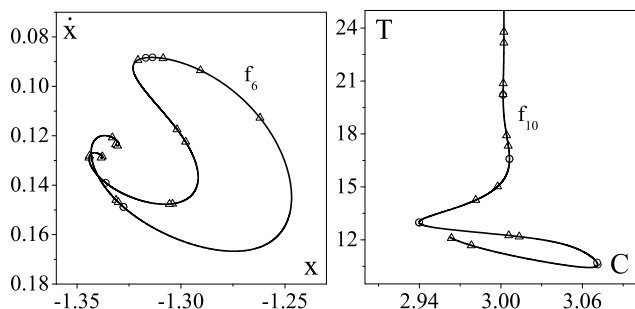




**Fig. 9** Non-symmetric periodic orbits from the ten families  $f_i$ ,  $i = 1, \dots, 10$  for  $m_1 = 0.97$ ,  $m_2 = 0.02$  and  $m_3 = 0.01$ . The *small black points* and the *green stars* denote the positions of the primary bodies and the equilibria of the problem correspondingly



**Fig. 10** Horizontal (solid line) and vertical (dashed line) stability diagram of the family  $f_5$  of the simple non-symmetric periodic orbits for  $m_1 = 0.97$ ,  $m_2 = 0.02$  and  $m_3 = 0.01$ . The small circles and triangles indicate the horizontal and the vertical critical periodic orbits of this family correspondingly



**Fig. 11** Left:  $x$  versus  $\dot{x}$  from the family  $f_6$  of the simple non-symmetric periodic orbits for  $m_1 = 0.97$ ,  $m_2 = 0.02$  and  $m_3 = 0.01$ . Right:  $C$  versus  $T$  from the family  $f_{10}$ . The small circles and triangles indicate the horizontal and the vertical critical periodic orbits of these families correspondingly

left window and frame  $f_6$  in Fig. 9). It has a small arc of stable periodic orbits (red line in Fig. 8) and we have calculated many horizontal and vertical periodic solutions close to its termination (small circles and triangles in left frame of Fig. 11).

Finally, family  $f_{10}$  consists of direct simple non-symmetric periodic orbits around the dominant primary body  $m_1$ . This family from one side has periodic orbits where change multiplicity and from the other side has asymptotic orbits at the equilibrium point  $L_3$  ( $f_{10}$  in Fig. 9). Many horizontal and vertical periodic orbits are calculated (small circles and triangles in right frame of Fig. 11).

In Fig. 9 we plot samples of simple non-symmetric periodic orbits from the ten determined families. In frames  $f_6$  and  $f_{10}$  asymptotic orbits at  $L_3$  from families  $f_6$  and  $f_{10}$  are presented correspondingly. In frame  $f_5$  we illustrate seven non-symmetric periodic orbits where they present the evolution of family  $f_5$  which we have described previously.

In Fig. 10 the stability diagrams from the family  $f_5$  is illustrated. The positions of the horizontal and vertical critical periodic orbits from this family are also presented. In Fig. 11 we plot two characteristic curves of families  $f_6$  and  $f_{10}$  correspondingly. In the second figure we denote the positions of the critical periodic orbits as the period  $T$  tends to infinity.

### 5 Conclusions and remarks

Our goal of this paper was the study of the families of the simple periodic orbits of the restricted four-body problem (Lagrangian equilateral triangle configuration), when the primary bodies of the problem have equal or not masses. We found ten families of periodic orbits of each case i.e. when the primary bodies are equal, when the two primaries  $m_2$  and  $m_3$  are equal and finally when  $m_1 \neq m_2 \neq m_3$ . We chosen, in the last two cases, one dominant primary body ( $m_1$ ) and two small ( $m_{2,3}$ ) in order the stability inequality (7) of the configuration to be valid.

We found a large number of various types of orbits, such as symmetric and non-symmetric, with respect to the horizontal  $x$ -axis, orbits around one, two or three primaries, orbits around an equilibrium position, collision orbits, Trojan orbits, as well as asymptotic orbits at one, two or three equilibria. The orbits are manly retrograde but there are families where consist of direct periodic orbits.

We have also studied the stability of each periodic orbit and we found the special generating planar horizontal and vertical critical solutions of each family.

In any case of the mass distribution, the families  $f_1$ ,  $f_2$ ,  $f_3$  and  $f_4$  have the majority of their periodic orbits to be stable (like the basic families  $m$ ,  $h$ ,  $g$  and  $l$  of the classical three-body problem). The majority of the periodic orbits of the rest families are unstable.

The network of the families of the periodic orbits changes substantially as the mass parameters vary and some families disappear while some news appear. More specifically, our main results can be summarized, in each case, as follows:

#### 5.1 Three equal masses

- There is not family without stable periodic orbits. The majority of the periodic orbits of the families  $f_1$ ,  $f_2$ ,  $f_3$  and  $f_4$  are stable. The rest families have small arcs of their characteristic curves with stable periodic orbits.
- Six of the ten families, under consideration, have as terminating members, asymptotic orbits. Namely, families  $f_2$ ,  $f_5$ ,  $f_6$ ,  $f_8$  and  $f_9$  terminate with heteroclinic asymptotic orbits at the equilibrium points  $L_5$  and  $L_6$ . One family,  $f_{10}$ , has as terminating member an asymptotic orbit at the three equilibrium points  $L_3$ ,  $L_5$  and  $L_6$ .

- Family  $f_5$  emanates from the collinear equilibrium point  $L_2$  and terminates with asymptotic orbits at the non-collinear equilibria  $L_5$  and  $L_6$ .
- All the families have horizontal and vertical-critical periodic orbits as members.
- All the symmetric periodic orbits we found are asymmetric with respect to the vertical  $y$ -axis.
- All the families consist of retrograde periodic orbits except families  $f_3$  and  $f_{10}$  where have direct ones.

### 5.2 Two equal masses

- Only one family, namely  $f_9$ , is entirely unstable. All the others families have stable periodic solutions.
- Simple asymptotic orbits, in the ten families under consideration, do not exist.
- All the symmetric periodic orbits we found are asymmetric with respect to the vertical  $y$ -axis.
- The characteristic curve of family  $f_8$  is closed.
- All the families consist of retrograde periodic orbits except families  $f_3$  and  $f_{10}$  where have direct ones.

### 5.3 Unequal masses

- All the families determined here consist of simple non-symmetric periodic orbits (both in horizontal and in vertical axis).
- All the families have stable periodic orbits.
- Two (of ten) families, namely,  $f_3$  and  $f_{10}$  have terminating orbits asymptotic solutions at the equilibrium point  $L_3$ .
- Family  $f_5$  emanates (or terminates) from the non-collinear equilibrium point  $L_2$  and terminates (or emanates) at the non-collinear equilibria  $L_6$ .
- The characteristic curve of family  $f_8$  is closed.
- All the families consist of retrograde periodic orbits except families  $f_3$  and  $f_{10}$  where have direct ones.

Closing this article we would like to express our intention to apply these results in the Sun-Jupiter-Asteroid system. Families of periodic orbits have already been calculated and we will present them in a future article.

**Acknowledgements** During this work the first Author was in receipt of a “K. Karatheodory” research grant.

## References

- Álvarez-Ramirez, M., Vidal, C.: Dynamical aspects of an equilateral restricted four-body problem. *Math. Probl. Eng.* (2009). doi:10.1155/2009/181360
- Andoyer, M.H.: Sur les solutions périodiques voisines des positions d'équilibre relatif, dans le problème des  $n$  corps. *Bull. Astron.* **23**, 129–146 (1906)

- Baltagiannis, A.N., Papadakis, K.E.: Equilibrium points and their stability in the restricted four-body problem. *Int. J. Bifurc. Chaos* (2011). doi:10.1002/ijbc.201100040
- Brumberg, V.A.: Permanent configurations in the problem of four bodies and their stability. *Sov. Astron.* **34**, 57–79 (1957)
- Ceccaroni, M., Biggs, J.: Extension of low-thrust propulsion to the autonomous coplanar circular restricted four body problem with application to future Trojan Asteroid missions. In: 61st International Astronautical Congress, IAC 2010, Prague, Czech Republic (2010)
- Deprit, A., Henrard, J., Palmore, J., Price, E.: The Trojan manifold in the system Earth-Moon. *Mon. Not. R. Astron. Soc.* **137**, 311–335 (1967)
- Devaney, R.: Blue sky catastrophes in reversible and Hamiltonian systems. *Indiana Univ. Math. J.* **26**, 247–263 (1977)
- Gascheau, M.: Examen d'une classe d'équations différentielles et application a un cas particulier du problème des trois corps. *Compt. Rend.* **16**, 393–394 (1843)
- Hénon, M.: Exploration numérique du problème restreint. II Masses égales, stabilité des orbites périodiques. *Ann. Astrophys.* **28**, 992–1007 (1965)
- Hénon, M.: Vertical stability of periodic orbits in the restricted problem. I. Equal masses. *Astron. Astrophys.* **28**, 415–426 (1973)
- Hüttenhain, E.: Räumliche infinitesimale Bahnen um die Librationspunkte im Geradenfall der  $(3 + 1)$ -Körper. *Astron. Nachr.* **250**, 298–316 (1933)
- Kloppenborg, B., et al.: Infrared images of the transiting disk in the  $\epsilon$  Aurigae system. *Nat. Lett.* **464**, 870–872 (2010)
- Lindow, M.: Ein Spezialfall des Vierkörperproblems. *Astron. Nachr.* **216**, 21–22 (1922)
- Lindow, M.: Eine Transformation für das Problem der  $n + 1$  Körper. *Astron. Nachr.* **219**, 142–154 (1923)
- MacMillan, W.D., Bartky, W.: Permanent configurations in the problem of four bodies. *Trans. Am. Math. Soc.* **34**, 838–875 (1932)
- Majorana, A.: On a four-body problem. *Celest. Mech.* **25**, 267–270 (1981)
- Markellos, V., Black, W., Moran, P.: A grid search for families of periodic orbits in the restricted problem of three bodies. *Celest. Mech.* **9**, 507–512 (1974)
- Melita, M.D., Licandro, J., Jones, D.C., Williams, I.P.: Physical properties and orbital stability of the Trojan asteroids. *Icarus* **195**, 686–697 (2008)
- Moulton, F.R.: On a class of particular solutions of the problem of four bodies. *Trans. Am. Math. Soc.* **1**, 17–29 (1900)
- Palmore, J.: Relative equilibria of the  $n$ -body problem. Thesis, University of California, Berkeley, California (1973)
- Pedersen, P.: Librationspunkte im restringierten vierkörperproblem. *Dan. Mat. Fys. Medd.* **21**, 1–80 (1944)
- Pedersen, P.: Stabilitätsuntersuchungen im restringierten vierkörperproblem. *Dan. Mat. Fys. Medd.* **26**, 1–37 (1952)
- Robutel, P., Gabern, F.: The resonant structure of Jupiter's Trojan asteroids—I. Long-term stability and diffusion. *Mon. Not. R. Astron. Soc.* **372**, 1463–1482 (2006)
- Schaub, W.: Über einen speziellen Fall des Fünfkörperproblems. *Astron. Nachr.* **236**, 34–54 (1929)
- Schwarz, R., Süli, À., Dvorac, R.: Dynamics of possible Trojan planets in binary systems. *Mon. Not. R. Astron. Soc.* **398**, 2085–2090 (2009b)
- Schwarz, R., Süli, À., Dvorac, R., Pilat-Lohinger, E.: Stability of Trojan planets in multi-planetary systems. *Celest. Mech. Dyn. Astron.* **104**, 69–84 (2009a)
- Simo, C.: Relative equilibrium solutions in the four body problem. *Celest. Mech.* **18**, 165–184 (1978)
- Szebehely, V.: Theory of orbits. Academic Press, New York (1967)
- Van Hamme, W., Wilson, R.E.: The restricted four-body problem and epsilon Aurigae. *Astrophys. J.* **306**, L33–L36 (1986)