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A performance evaluation of the ejector refrigeration system based on thermo-economic criteria through multi-objective approach

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Abstract

This work investigates an exhaust heat-driven ejector refrigeration system with the thermo-economic considerations. The system is thermally modelled and is optimized considering the performance coefficient and the total annual cost as two objectives using heat transfer search algorithm. Generator temperature, evaporator temperature and condenser temperature are considered as design variables. A 2-D shock circle model is used to simulate the ejector component with R245fa refrigerant. The results of multi-objective optimization are discussed using the Pareto frontier obtained between both conflicting objective functions. The effect of varying the nozzle throat diameter and the ecological function on the thermo-economic objectives is presented and discussed. The sensitivity analysis of the objective functions to the decision variables is investigated. Further, the exergo-economic results at the optimal point are also presented and discussed. The results reveal that the ejector and the generator are the leading contributors to the exergy destruction and hence to the total annual cost. The coefficient of performance and total annual cost at the best optimal point are 0.3 and 25,903 \$/year, respectively. The optimized product unit cost of the system is 53.8 \$/GJ with 10.5% exergy efficiency.

Graphic abstract



Keywords Ejector refrigeration · Optimization · Thermo-economic analysis · Heat transfer search algorithm

List of symbols

- A Area (m^2)
- $C_{\rm p}$ Specific heat (J/kg K)
- COP Coefficient of performance (-)
- CRF Capital recovery factor (year $^{-1}$)

- \dot{C} Exergy cost rate (\$/year)
- \dot{C}_Q Exergy cost rate associated with heat transfer (\$/ year)
- \dot{C}_W Exergy cost rate associated with work transfer (\$/ year)
- D Diameter (m)
- \dot{E} Exergy rate (W)
- *h* Specific enthalpy (J/kg K)
- *H* Pump energy head (m)

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М	Mach number (–)
MF	Maintenance factor (-)
ṁ	Mass flow rate (kg/s)
Ν	Velocity distribution exponent (-)
Р	Pressure (Pa)
Ż	Heat transfer rate (W)
R	Radius (m)
$R_{\rm gas}$	Gas constant (J/kg K)
S	Specific entropy (J/kg K)
Т	Temperature (K)
TAC	Total annual cost (\$/ywar)
и	Unit cost (\$/GJ)
v	Specific volume (m ³ /kg)
V	Velocity (m/s)
\overline{V}	Mean velocity (m/s)
Ŵ	Work transfer rate (W)
Ζ	Component cost value (\$)
à	

Z Component cost rate (\$/year

Greek letters

η	Efficiency
$\eta_{\rm s,p}$	Pump isentropic efficiency

- $\eta_{m,p}$ Pump mechanical efficiency
- *Γ* Adiabatic index

Subscripts

or condend

- Evap Evaporator
- g Generator
- p Primary flow
- s Secondary flow
- d Diffuser
- D Destruction
- F Fuel
- t Nozzle throat
- *e*1 Primary nozzle exit
- e2 Constant area section
- pa Primary flow at a–a
- sa Secondary flow at a-a
- m Mixed flow
- i, in Inlet
- e, ex Exit
- O Reference state
- K Component

Abbreviations

- ECF Ecological function
- HTS Heat transfer search
- IC Internal combustion engine

Introduction

In today's era, the increasing need for air conditioning has led to its larger share in electricity consumption worldwide. Due to this, low-grade energy-driven refrigeration systems are drawing more attention to the researchers. Ejector refrigeration system is one of such systems being more reliable as it contains no mobile parts and can have more environment-friendly potential compared to the conventional compression-based refrigeration systems. However, ejector refrigeration systems are unable to penetrate the market due to its low performance coefficient and difficulty in design. Many researchers have investigated the performance of the ejector refrigeration systems and the ejector component especially.

To predict the ejector performance, Huang et al. (1999) used a 1-D model. The authors have assumed the chocked flow condition with constant pressure mixing and uniform velocity distribution in the mixing chamber. The experimental and numerical performances were noted and compared with choked flow condition. With the experimental results, the 1-D model was able to predict the entrainment ratio and the area ratio values within 23% and 9% maximum deviation, respectively. This 1-D model algorithm was later used with modifications by Chen et al. (2013)and Tashtoush et al. (2015). Chen et al. (2013) suggested a 1-D model for ejector at the critical as well as sub-critical operational mode. The mixing of primary and secondary streams at constant pressure in the constant area zone was assumed, and the entrainment ratio values predicted by the model were compared with the experimental values of Huang et al. (1999) and found the maximum deviation of 14.2%. Recently, a simplified ejector model was proposed by Liu et al. (2017) based on the model of Huang et al. (1999). The model has expressed the ejector entrainment ratio and the backpressure by linear expressions. The authors reported the performance prediction comparison with experimental results within 4% deviation. These 1-D models were assuming the uniform velocity at a particular y-y section and were not considering the viscous effect near the ejector walls in the mixing chamber. To consider the viscosity effects at the inner wall of the mixing chamber of the ejector, Zhu et al. (2007) proposed a 2-D shock circle model considering the exponential velocity distribution at any y-y section. It was assumed that the mixed flow of primary and secondary streams gets chocked (Mach number = 1) at the thin intersection layer between these two streams. This very thin layer sheet is termed as the shock circle. The authors claimed that this 2-D model calculation was simpler to perform than the previously used 1-D models with better accuracy. The entrainment ratio values predicted by the model matched more accurately with the experimental results (of Huang et al. 1999) compared to 1-D model used by the later. The largest deviation observed in predicting the entrainment ratio was 11%. Using the shock circle model concept, Zhu and Li (2009) further used a novel model for 2-phase ejector performance evaluation for wet and dry working fluids at the chocked (critical) mode.

Furthermore, Wang et al. (2009) have experimentally assessed an ejector cooling system with thermal pumping by a multi-functional generator. The authors found the COP up to 0.45 when operated at higher evaporator temperatures using R365mfc. To encounter the ejector geometry effects, Yan et al. (2012) used a CFD-based model with the experimental investigation of the same. The ejector area ratio (mixing constant area/nozzle throat area) and the nozzle exit position were the dominating geometrical parameters to improve the ejector entrainment ratio. Śmierciew et al. (2014) has experimentally studied the generator and condenser temperature effects on a solar heat-operated ejector refrigeration system with refrigerant isobutene. The COP varied between 0.15 and 0.2. Eldakamawy et al. (2017) used retrograde (dry) refrigerants and investigated the ejector refrigeration system and found the use of dry refrigerant promising in the ejector-based systems. Further, Varga et al. (2017) have experimentally reported a small-scaled ejector refrigerator utilizing solar heat installed at Portugal. 1.5-kW ejector with R600a refrigerant was considered in the study. The thermal COP was varying between 0.15 and 0.40 with 4.6 average mechanical COP. Recently, Narimani et al. (2019) performed numerical and experimental study to investigate the generator pressure effect in the R245fabased ejector refrigeration system. The authors found that ejector operates at single chocking mode and double chocking mode operations before and after an optimum generator pressure, respectively. Furthermore, more studies have been carried out to study the COP of ejector refrigeration systems with various refrigerants and found that the environmentalfriendly refrigerants have lower COP compared to the obsoleting refrigerants (Gil and Kasperski 2015; Shestopalov et al. 2015; Zhang et al. 2015).

Several studies on optimization of ejector refrigeration systems are reported in the literature. Dai et al. (2009) optimized the ejector refrigeration cycle combined with the Rankine power cycle using genetic optimization algorithm. Chen et al. (2014) numerically presented the optimum COP with variation in the generator temperature keeping evaporator and condenser temperature to a fixed value. Also, the optimum ejector entrainment ratio and area ratio with variation in the condenser temperature are obtained using R123 and R141b as the refrigerants. Further to this, Sadeghi et al. (2015) have performed the exergo-economic study and the multi-objective optimization using genetic optimization algorithm of an ejector refrigeration system using R141b. The concept and equations of 2-D model were utilized to simulate the ejector component. The system temperatures were adjusted by the algorithm to optimize the exergy efficiency and the product unit cost. The authors have considered the exergy destruction cost to evaluate the total cost of the system as the ejector is found as the leading exergy destructing component of such systems. Sanaye et al. (2019) optimized a system combined of vapour–vapour ejector cycle and a liquid–vapour ejector cycle from energy, exergy, economic and environment viewpoints. With five design variables and two conflicting objective functions, the authors obtained about 18% performance coefficient improvement and 8% lower annual cost.

The useful concluding information from the above reviews is: (a) the concept of 2-D shock circle model is found more practical in assessing the ejector performance as it considers the viscosity effects and hence the non-uniform distribution of velocity in y-y plane of mixing chamber, (b) very fewer studies on the multi-objective optimization of conventional ejector refrigeration system. The multi-objective optimization using COP and cost as objectives is also not found in the literature, (c) exergy destruction is mostly highest in the ejector and can be reduced by considering the cost associated with this exergy destruction (exergo-economic study) which is considered by Sadeghi et al. (2015) only as per our foremost knowledge, (d) COP optimization with operating temperatures as design variables and use of newly developed optimization algorithms can explore more improved operating conditions to the ejector refrigeration systems.

The present work focuses on the multi-objective optimization of an ejector refrigeration system considering COP and TAC as two objective functions. For low-grade energy-driven systems (such as ejector refrigeration systems, absorption refrigeration system), the COP values are quite lower (main contributing reason for not penetrating into the market). So, this makes the COP, a very important performance parameter to be enhanced to reduce the input energy requirement for the desired output. Secondly, the TAC value includes components initial cost, operating cost and the cost associated with the exergy destruction in the components (exergo-economic analysis). By doing this, reduction in the TAC value will improve the second law-based performance of the system as well. Even, TAC is also a major factor for the system's market opportunities. Further, COP and TAC are found conflicting objectives (if one improves, other has to degrade). This seeks for the need of multi-objective optimization wherein the input parameters (decision variables) improved in such a way that both the objectives can take optimum value.

The main novelty of this work is to present the multiobjective optimization of an ejector refrigeration system using a novel multi-objective heat transfer search (HTS)

algorithm with the thermo-economic approach. The consideration of COP and the total annual cost (TAC) of the system as two objective functions (thermo-economic) adds novelty to this work further as the optimized outputs can contribute in trading off the ejector refrigeration system between thermal performance (COP) and the economy. The COP is selected as it is the main reason for the inability of such low-grade energy-driven systems to penetrate the market. Secondly, the TAC value considers the economic as well as the second law-based performance by taking the exergy destruction costs into the account. The outcome of the thermo-economic optimization is presented in the form of Pareto front. Additionally, the sensitivity analysis of the objectives and the exergo-economic analysis at the optimal condition are also presented. The concept and equations of 2-D shock circle model are used to simulate the ejector component in MATLAB tool. A dry refrigerant R245fa with positive slope of the saturated vapour line is used as a working fluid, and therefore, the working fluid will not condense in the ejector (Chunnanond and Aphornratana 2004). This hydrofluorocarbon (HFC) refrigerant is selected as it has good thermo-physical properties with zero potential to ozone depletion (Shestopalov et al. 2015).

System description and modelling

This section describes the ejector refrigeration system and the models for thermodynamic and the exergo-economic analyses.

System description

The low-grade energy-driven ejector refrigeration system is shown in Fig. 1. It consists of an ejector, generator, evaporator, condenser, expansion valve and pump. Exhaust gases from an IC engine are considered as the heat source to generator wherein the working fluid (R245fa) is vaporized to generator temperature. An ejector comprises of a primary nozzle, suction zone, mixing zone and a diffuser as shown in Fig. 2. The high pressure saturated vapour generated in the generator will enter the ejector as the primary flow stream. The converging-diverging nozzle will impart supersonic velocity to the primary stream (chocked at the nozzle throat), and hence, the pressure at the nozzle exit will be low. This low pressure will form the suction chamber for the secondary fluid stream (low pressure vapour refrigerant) from the evaporator. These primary and secondary vapour streams will mix at the effective area where the secondary stream will get chocked. The mixed flow will then undergo a shock wave while moving through the constant area section. This leads to increase in the pressure, and the velocity becomes



ejector refrigeration system



Fig. 2 Ejector component

subsonic (Huang et al. 1999). This stream will exit the ejector at the condenser pressure after further increasing its pressure in the diffuser section. The saturated liquid stream from the condenser will separate itself into two streams. One stream will be pumped to the generator where it is heated to the vapour state (primary stream). The second stream will be expanded to the evaporator pressure after passing through the expansion valve. In the evaporator, this secondary stream will get converted to the vapour state after receiving heat of the cooling load. The exhaust gas from an IC engine is considered as the heat source in the generator.

Thermal modelling

Thermal modelling and simulation of the system are carried out in MATLAB tool. The following assumptions are considered for the system simulation (Zhu et al. 2007; Huang et al. 1999; Sadeghi et al. 2015):

- Steady-state simulation with 25 °C and 100 kPa as ambient conditions.
- The condition at the outlet of the generator and evaporator is saturated vapour while the condition at the outlet of the condenser is saturated liquid.
- Frictional pressure drops in the pipes are neglected, typically calculated at the detailed design stage.
- Adiabatic inner walls of the ejector.
- Kinetic energies of the entering primary and secondary streams as well as the exit mixed stream from the diffuser are negligible.
- The secondary stream flow gets chocked and becomes critical at the effective a-a area where mixing starts.

- The pressure of the entering secondary flow stream is assumed equal to the evaporator pressure.
- Refrigerant feeding pump which pumps the refrigerant from the condenser to the generator remains cavitation-free.
- The mechanical efficiency of pump is considered as 75%.
- For the simplicity in simulation, the isentropic efficiencies are assumed for considering the frictional losses. Isentropic efficiencies of primary flow, secondary flow and the flow in the diffuser are considered as 0.95, 0.85 and 0.90, respectively.

Ejector model

In this work, the ejector component is modelled considering the concepts of shock circle model proposed by Zhu et al. (2007). This model predicts the ejector performance by considering the 2-D velocity distribution (exponential) in the constant area mixing chamber. This will include the viscosity effects near the wall in the mixing chamber. Table 1 shows the equations used for the ejector component modelling. Also, the simulation flowchart, which focuses on the ejector component modelling, is shown in Fig. 3. The thermo-physical properties $C_{\rm p}$, γ and $R_{\rm gas}$ are considered to be temperature dependent, and hence, they are taken from REFPROP fluid library in MATLAB. The output parameters from the ejector are temperature (condenser temperature) and pressure of mixed flow, mass flow rates of primary and secondary flow streams and the constant area section diameter.

 Table 1
 Ejector component modelling (Zhu et al. 2007; Sadeghi et al. 2015)
 Event

Primary stream flow rate at nozzle throat

Temperature, velocity and Mach number at the nozzle exit

Primary flow Mach number, temperature, velocity and effective diameter at effective area (a-a)

$$\begin{split} \dot{m}_{p} &= P_{p}A_{t} \sqrt{\left(\frac{r_{0}}{R_{pas}T_{p}}\right)\left(\frac{z}{1+\gamma}\right)^{(\gamma-1)}} \\ \frac{D_{c1}}{D_{t}} &= \sqrt{\left[\frac{2+(\gamma-1)M_{c1}^{2}}{2+(\gamma-1)}\right]^{\left(\frac{\gamma+1}{2(\gamma-1)}\right)}\left(\frac{1}{M_{c1}}\right)} \\ \frac{T_{p}}{T_{c1}} &= 1 + \frac{1}{2}\left(\gamma - 1\right)M_{e1}^{2} \\ V_{e1} &= M_{e1}\sqrt{\gamma R_{gas}T_{e1}} \\ \frac{P_{p}}{P_{s}} &= \left[1 + \frac{1}{2}\left(\gamma - 1\right)M_{pa}^{2}\right]^{\left(\frac{\gamma}{\gamma-1}\right)} \\ \frac{T_{p}}{T_{pa}} &= 1 + \frac{1}{2}\left(\gamma - 1\right)M_{pa}^{2} \\ \sqrt{pa} &= M_{pa}\sqrt{\gamma R_{gas}T_{pa}} \\ \left(\frac{D_{pa}\sqrt{\eta_{p}\eta_{s}}}{D_{c1}}\right)^{2} &= \left[\frac{2+(\gamma-1)M_{pa}^{2}}{2+(\gamma-1)M_{c1}^{2}}\right]^{\left(\frac{\gamma+1}{2(\gamma-1)}\right)}\left(\frac{M_{c1}}{M_{pa}}\right) \\ \overline{V}_{sa} &= \frac{2V_{pa}}{\left(R_{c2}^{2} - R_{pa}^{2}\right)} \left[\frac{nR_{c2}^{2}}{n+1}\left(1 - \frac{R_{pa}}{R_{c2}}\right)^{\left(\frac{n+1}{n}\right)} - \frac{nR_{c2}^{2}}{2n+1}\left(1 - \frac{R_{pa}}{R_{c2}}\right)^{\left(\frac{2n+1}{n}\right)}\right] \\ \dot{m}_{s} &= \frac{2\pi P_{s}V_{pa}}{\left(R_{c2}^{2} - R_{pa}^{2}\right)} \left[\frac{nR_{c2}^{2}}{n+1}\left(1 - \frac{R_{pa}}{R_{c2}}\right)^{\left(\frac{n+1}{n}\right)} - \frac{nR_{c2}^{2}}{2n+1}\left(1 - \frac{R_{pa}}{R_{c2}}\right)^{\left(\frac{2n+1}{n}\right)}\right] \\ n &= \frac{\ln\left(1 - \frac{D_{pa}}{D_{c2}}\right)}{\ln\left(\sqrt{\frac{\sqrt{T_{sa}/T_{pa}}}{M_{pa}}}\right)} \\ V_{m} &= \frac{\dot{m}_{p}V_{pa} + \dot{m}_{s}V_{sa}}{\dot{m}_{p} + \dot{m}_{s}} \\ T_{m} &= \frac{\dot{m}_{p}\left(C_{p}T_{pa} + \frac{V_{ca}^{2}}{2}\right) + \dot{m}_{s}\left(C_{p}T_{sa} + \frac{V_{ca}^{2}}{2}\right) - \left(\dot{m}_{p} + \dot{m}_{s}\right)\left(\frac{V_{ca}^{2}}{2}\right)}{\left(\dot{m}_{p} + \dot{m}_{s}\right)C_{p}} \\ v_{m} &= \frac{V_{m}A_{c2}}{\dot{m}_{p} + \dot{m}_{s}} \\ P_{c} &= 1 + \frac{2\gamma}{(\gamma+1)}\left(M_{m}^{2} - 1\right) \\ M_{e2} &= \sqrt{\frac{1+\left(\frac{\gamma-1}{2}\right)M_{m}^{2}}{\gamma A_{m}^{2} - \left(\frac{\gamma-1}{2}\right)}} \\ \eta_{d} &= \frac{T_{c2} - T_{d}}{T_{c2} - T_{m}} \end{array}$$

 $\left(\begin{array}{c} & \\ \end{array} \right) \left(\begin{array}{c} \end{array} \right) \left(\begin{array}{c} \\ \end{array} \right) \left(\begin{array}{c} \\ \end{array} \right) \left(\begin{array}{c} \end{array} \right) \left(\end{array} \right) \left(\begin{array}{c} \end{array} \right) \left(\end{array} \right) \left(\end{array} \right) \left(\begin{array}{c} \end{array} \right) \left(\end{array}$

Velocity and mass flow rate of secondary entrained stream at a-a

Velocity, temperature and specific volume of mixed flow before shock

Pressure and Mach number of mixed flow after shock

Diffuser efficiency

Energy analysis

Steady-state energy analysis is performed using the following energy balance for a system component:

$$\dot{Q} - \dot{W} = \sum_{i} \dot{m}_{i} h_{i} - \sum_{e} \dot{m}_{e} h_{e}$$
⁽¹⁾

where h, \dot{Q} and \dot{W} are the state enthalpy, heat transfer rate and work transfer rate, respectively. These energy balance equations for generator, evaporator, condenser and pump are listed in Table 2.

Exergo-economic analysis

The analysis of any thermal system is worthy when it is carried out from both thermodynamic and economic point of view. Further, in order to consider the second law of thermodynamics (exergy and irreversibility), the cost associated with the energy destruction is accountable. Keeping this point in mind, the exergo-economic analysis is performed for the system based on the procedure given by Lazzaretto and Tsatsaronis (2006). There are two kinds of components in the system: productive and dissipative



Fig. 3 Simulation flowchart

(Lazzaretto and Tsatsaronis 2006). In the present system, the cooling unit consists of evaporator as productive component while condenser and expansion valve as its dissipative components. The total annual cost (TAC) for the system from exergo-economic analysis can be equated as (Sadeghi et al. 2015):

$$TAC(\$/year) = \sum \dot{Z}_k + \sum \dot{C}_{D,k} + \dot{C}_f$$
(2)

where \dot{Z}_k , $\dot{C}_{D,k}$ and \dot{C}_f are the cost rates of component investment, component exergy destruction and system fuel. \dot{Z}_k can be determined by:

Table 2 Energy analysis-based governing equations

Heat transfer rate in generator	$\dot{Q}_{\rm g} = \dot{m}_{\rm p} (h_1 - h_8)$	(8)
Heat transfer rate in evaporator	$\dot{Q}_{\rm evap} = \dot{m}_{\rm s} (h_2 - h_7)$	(9)
Heat transfer rate in condenser	$\dot{Q}_{\rm cr} = \left(\dot{m}_{\rm s} + \dot{m}_{\rm p}\right) \left(h_3 - h_4\right)$	(10)
Pump work	$\dot{W}_{\text{pump}} = \frac{\dot{m}_{\text{p}}gH_{\text{pump}}}{\eta_{\text{pump}}} = \frac{\dot{m}_{\text{p}}}{\eta_{\text{pump}}} \left(\frac{P_{\text{g}} - P_{\text{cr}}}{\rho}\right)$	(11)
СОР	$\frac{\dot{Q}_{\text{evap}}}{\dot{Q}_{\text{g}} + \dot{W}_{\text{pump}}}$	(12)

 Table 3
 Exergy destruction in components

Component	Exergy destruction rate
Generator	$\dot{E}_{\rm D} = (\dot{E}_8 - \dot{E}_1) + (\dot{E}_{13} - \dot{E}_{14})$
Ejector	$\dot{E}_{\rm D} = \left(\dot{E}_1 + \dot{E}_2\right) - \dot{E}_3$
Evaporator	$\dot{E}_{\rm D} = (\dot{E}_7 - \dot{E}_2) + (\dot{E}_{11} - \dot{E}_{12})$
Condenser	$\dot{E}_{\rm D} = (\dot{E}_3 - \dot{E}_4) + (\dot{E}_9 - \dot{E}_{10})$
Pump	$\dot{E}_{\rm D} = \left(\dot{E}_5 - \dot{E}_8\right) + \dot{W}_{\rm pump}$

Table 4 Cost balance in components

Component	Equation of cost balance
Generator	$\dot{C}_{13} + \dot{C}_8 + \dot{Z}_g = \dot{C}_{14} + \dot{C}_1$
Ejector	$\dot{C}_1 + \dot{C}_2 = \dot{C}_3$
Evaporator	$\dot{C}_{11} + \dot{C}_7 + \dot{Z}_{evap} = \dot{C}_{12} + \dot{C}_2$
Condenser	$\dot{C}_9 + \dot{C}_3 + \dot{Z}_{cr} = \dot{C}_{10} + \dot{C}_4$
Pump	$\dot{C}_5 + \dot{Z}_{\text{pump,motor}} = \dot{C}_8 - \dot{C}_{\text{W,p}}$
Separation point	$\dot{C}_4 = \dot{C}_5 + \dot{C}_6$

$$\dot{Z}_{k} = Z_{k} \times CRF \times MF \tag{3}$$

where Z_k , CRF and MF are the component investment (capital) cost, capital recovery factor and the maintenance factor, respectively. The component capital cost equations and their finding procedure are adopted from the work of Misra et al. (2003) and Sadeghi et al. (2015).

Further in exergo-economic assessment, the cost rate associated with exergy destruction in the component $(\dot{C}_{D,k})$ can be given as:

$$C_{\mathrm{D,k}} = u_{\mathrm{f,k}} \times E_{\mathrm{D,k}} \tag{4}$$

Here $u_{f,k}$ is the unit cost of the component fuel source and $\dot{E}_{D,k}$ is the exergy destruction rate within the component. The exergy rate of a flowing stream is given by:

$$\dot{E}_{k} = \dot{m}_{k} \left(\left(h_{k} - h_{0} \right) - T_{0} \left(s_{k} - s_{0} \right) \right)$$
(5)

Here subscript k and 0 represent particular component and reference state conditions which are considered as 25 °C and 100 kPa. Table 3 lists the exergy destruction rate for each Table 5 Auxiliary equations for components

Component	Auxiliary equation
Generator	$\frac{\dot{c}_{13}}{\dot{E}_{13}} = \frac{\dot{c}_{14}}{\dot{E}_{14}}$ or $u_{13} = u_{14}$
Evaporator	$\frac{\dot{c}_2}{\dot{E}_2} = \frac{\dot{c}_7}{\dot{E}_7} \text{ or } u_2 = u_7$
Condenser	$\frac{\dot{c}_3}{\dot{E}_3} = \frac{\dot{c}_4}{\dot{E}_4} \text{ or } u_3 = u_4$
Expansion valve	$\frac{\dot{c}_6}{\dot{E}_6} = \frac{\dot{c}_7}{\dot{E}_7} \text{ or } u_6 = u_7$
Separation point	$\frac{\dot{C}_6}{\dot{C}_5} = \frac{\dot{m}_s}{\dot{m}_p}$

component of this system which is obtained by applying exergy balance. Further to evaluate $u_{f,k}$, we have:

$$u_{\rm f,k} = \dot{C}_{\rm f,k} / \dot{E}_{\rm f,k} \tag{6}$$

where suffix f represents the fuel source of particular component. To evaluate these exergy-based cost rates, the cost balance equation formulation is needed for each component. The general cost balance equation is:

$$\sum \dot{C}_{in,k} + \dot{Z}_{k} + \dot{C}_{Q,k} = \sum \dot{C}_{ex,k} + \dot{C}_{W,k}$$
(7)

The above equation makes us to understand that the sum of rates of the exergy cost of the leaving streams is equal to the summation of rates of the exergy cost of incoming streams and the net cost (capital and operational) of a particular component. This cost balancing is applied to each component and presented in Table 4. The total number of exergy streams in a system will always be higher (in general) compared to the total number of components and hence the numbers of cost balance equations. These require auxiliary equations to equalize their number. Table 5 lists the auxiliary equations for this system (Sadeghi et al. 2015). These auxiliary equations are obtained by applying F-principle and P-principle to the components. These principles are described in detail by Lazzaretto and Tsatsaronis (2006).

Model validation

The ejector refrigeration system model is validated with the results obtained for similar system from the work of Sadeghi et al (2015) operating with R141b refrigerant. The results of the present model (Table 9a) match significantly with the results in the reference within 3% error and hence validate the system model. As the thermal modelling focuses mainly on the ejector component, its validation with the experimental results available in the work of Huang et al. (1999) is shown in Table 9b. In this validation, for different nozzle geometries and various operating temperatures (with R141b as refrigerant),

two ejector output parameters are compared. These are entrainment ratio $(\dot{m}_{\rm s}/\dot{m}_{\rm p})$ and the area ratio $(D_{e2}/D_{\rm t})^2$. The values obtained from the present model agree well with the experimental results of the same within 10% deviation. This % error limit is acceptable in an engineering analysis.

Design variables, objective functions and constraints

The multi-objective-based optimization is performed for the system using HTS algorithm. The selected design (decision) variables, objective functions and the associated constraints are discussed in this section.

Design variables and associated constraints

In this ejector-based thermal system, operating temperature in the components significantly influences the system performance. Three decision variables, namely evaporator temperature, condenser temperature and generator temperature, are selected for the optimization. The selected ranges of these design variables and the constraints are listed in Table 6. Ecological function (ECF), the difference between net heat supplied and the entropy generation rate, is also considered as an ecological parameter. The physical importance of ECF is to favour our physical environment by controlling the entropy generation from the system operation (Ahmadi et al. 2014). ECF is considered as a constraint with a fixed value of 11 kW.

Objective functions

As per thermo-economic criteria, a thermal objective, namely the coefficient of performance (COP) of the system,

and an economic objective, namely the total annual cost (TAC) of the system, are selected as the objective functions. Coefficient of performance is to be maximized while TAC is set to be minimized.

Heat transfer search (HTS) algorithm and its multi-objective form

Heat transfer search is an evolutionary algorithm created by Patel and Savsani (2015). This HTS algorithm mimics the law of thermodynamics and heat transfer which state that every system in the universe always tries to attain thermal equilibrium with its surrounding through various heat transfer processes. Like other evolutionary algorithms, HTS is also a population-based algorithm. The search process of HTS algorithm is made up of 'conduction phase', 'convection phase' and 'radiation phase'. In the optimization algorithm, heat transfer terms, namely molecules, are treated as population, temperature of the molecules is treated as design variable, and energy level of the molecules is treated as objective function value of the population.

The HTS algorithm initiates with the random solution generation for the considered population. This population then updates through the execution of any one of the phases mentioned above until the predefined termination criteria are reached. During each generation, only one of the abovementioned phases will execute to update the population. The execution of any of the phase to update the population depends on the execution probability associated with that phase. In the HTS algorithm, equal probability is assigned to each of the three phases involved during the course of optimization. This equal probability is decided by a random number which varies between 0 and 1. Further, the algorithm also incorporates the greedy selection components which

	Range		Reason	
	Lower limit	Upper limit		
Design variables				
Evaporator temperature, $T_{\text{evap}}(\mathbf{K})$	271	285	Availability in the market	
Generator temperature, $T_{g}(\mathbf{K})$	348	368	Temperature range avail- able from heat source	
Condenser temperature, $T_{\rm cr}$ (K)	306	313	Practical range for water based cooling in con- denser	
Constraints used				
$T_{14} > T_{\rm g}$		To ensure the positive temperature gradient through- out the generator		
$\dot{Q}_{\rm g}$ < 33 kW		The constant heat input from source is 33 kW		
ECF = 11		To fix the ecological function to a value		

Table 6 Design variable ranges and constraints

ensure that only improved solutions are accepted during the course of optimization. The elitism concept is also implemented in the algorithm to achieve the better output. The detailed description of each of the phase of the HTS algorithm along with its mathematical formulation is available in the pioneer work of Patel and Savsani (2015).

The success of any optimization algorithm depends on the proper balance between the exploration and exploitation during the search process. The search process of the HTS algorithm consists of three phases, and each phase builds with two different search mechanisms which balance the exploration and exploitation during search process. The search processes of HTS algorithm design in such a way so that during early generations each phase explores the search space, while in the second half each phase exploits the search space. In this way, the HTS algorithm balances the exploration and exploitation of an entire search process. However, the computational time of the proposed algorithm is more due to its search mechanism.

Multi-objective heat transfer search (MOHTS) algorithm is a multi-objective variant of the HTS algorithm aimed to solve multi-/many-objective problems (Patel et al. 2017; Patel 2018; Raja et al. 2017; Patel and Raja 2019). MOHTS algorithm adopts a non-dominated sorting process to generate the Pareto solutions. The algorithm identifies the non-dominated solutions and keeps it in the algorithm's external archive. During each generation, the external archive of the MOHTS is updated with the help of the ε -dominance method which inspects the domination of the solution. The dominated solutions are eliminated from the external archives, and remaining non-dominated solutions within the external archives are used to generate the Pareto front. The detailed description and working of the MOHTS algorithm are available in the literature (Patel et al. 2017; Raja et al. 2017; Patel 2018; Patel and Raja 2019).

Application example

The exhaust gas from an IC engine is considered as the generator heat source in the system. The constant heat input of 33 kW at 225 °C exhaust gas temperature and 0.25 kg/s mass flow rate is considered. To simplify the simulation process, the cooling water stream is assumed to enter and leave the condenser at $(T_{cr} - 5)$ °C and $(T_{cr} - 2)$ °C. Similarly, the cooling load (chilled water) stream is assumed to enter and leave the evaporator at $(T_{evap} + 8)$ °C and $(T_{evap} + 3)$ °C (Sadeghi et al. 2015). The isentropic efficiencies of primary and secondary stream flows are taken as 95% and 85% while the diffuser isentropic efficiency is considered to be 90% (Zhu et al. 2007). The primary nozzle throat diameter and nozzle exit diameter are taken as 9.5 mm and 19 mm, respectively. For economic analysis, interest rate of 15%, maintenance factor of 1.06 and system's expected life of 20 year are considered. Further, the unit cost of electricity for the pump and the heat source to the generator are taken as 10 \$/GJ and 23.59 \$/GJ, respectively (Sadeghi et al. 2015). Finally, the exergy cost rates of water streams entering the evaporator and condenser are taken as zero.



Results and discussion

The results obtained from the multi-objective optimization of the ejector-based refrigeration system using HTS algorithm are presented in this section.

Multi-objective optimization

Considering COP and TAC as two objectives simultaneously, the optimized result in form of Pareto front is presented in Fig. 4. All the points in the Pareto front are the optimal solutions. Both COP and TAC are showing the conflicting behaviour as the maximized COP solution point has the maximum value of TAC (which is to be minimized) and vice versa. The optimal point A has the minimum optimum TAC (25,657 \$/year) with minimum optimum COP (0.26) and the optimal point E has the maximum optimum COP (0.35) but with maximum optimum TAC (26,594 \$/year). Based on this behaviour, one can select either point E or point A as the optimum design solution if the only objective to be focused is COP or TAC, respectively. 4% and 31% are the increment observed in the optimal values of TAC and COP upon moving from A to E, respectively. Apart from the end points A and E, three more optimal points are chosen for the study (B, C and D). The design variables and the objective function values at all five optimal points are listed in Table 7. The evaporator and condenser temperature varies significantly from point A to E while generator temperature has insignificant scattering from A to E. This observation states that the former two temperatures dominantly produce the conflicting effect on the objective functions compared to the generator temperature. Further, the ideal (best) optimal solution that can be obtained by considering equal weightage of both the objective functions is point O while the least desirable solution (non-ideal) is point N. Therefore, an optimal point nearest to the ideal point should be selected in order to consider COP and TAC simultaneously. Among available solution points, point B is the nearest to the ideal solution obtained from this multiobjective optimization. Finally, one can conclude from a Pareto front that the decision-maker can select any of the optimal design points according to the priority of the objective functions.

Sensitivity analysis

Sensitivity analysis highlights the behaviour of objective functions upon variation in design variable values from their optimum points. The observations from sensitivity analysis can help the decision-maker in adjusting any of the design variables value from its value at the optimal solution while implementing the optimal design (Jaluria 2007). Here the design variables are varied individually for all five selected optimal solutions (A-E) and the results are plotted in Fig. 5. Figure 5a shows the effect of varying the generator temperature from its optimal value within its range. It demonstrates that with the increase in the generator temperature from its optimal point, the COP decreases while TAC initially increases and then decreases. The initial rise in TAC is less significant than the later decrement. On moving from optimal point A-E, TAC is almost unchanged initially and decreases afterwards as generator temperature increases. These observations can help the decision-maker in knowing that how sensitively the generator temperature variation changes the objective functions. Physically, the decrease in COP is because of rise in heat supplied for constant refrigeration effect as evaporator and condenser temperatures are unaltered (from their optimal point values). Further, it is found that the generator is the second leading component to the exergy destruction (Fig. 8), and as the temperature differential in the generator component (between the constant heat source temperature and the generator temperature) decreases with rise in the generator temperature, the exergy destruction in the generator decreases. Due to this, there is also noticeable drop in the exergy destruction cost (compared to the other capital and operating costs) of the generator and hence the drop in TAC.

Figure 5b shows the effect of variation in the evaporator temperature from its optimal values at all five different optimal points (A–E). With rise in the evaporator temperature, both objectives increase. The varying range of COP is almost same for all optimal points (A–E) while the varying range of TAC becomes costlier from A to E. These results can suggest the decision-maker to adjust the evaporator temperature according to the prior objective among the two.

Table 7 Optimal parameters atselected design points (A–E)

	А	В	С	D	Е
Evaporator temperature, T _{evap} (K)	272.4	274.6	275.3	275.9	276.4
Generator temperature, $T_{g}(\mathbf{K})$	351.9	351.6	351.2	350.8	350.4
Condenser temperature, $T_{\rm cr}$ (K)	309.2	308.5	307.6	306.8	306.0
Coefficient of performance, COP	0.26	0.30	0.32	0.33	0.35
Total annual cost, TAC (\$/year)	25,657	25,903	26,134	26,376	26,594



Fig. 5 Sensitivity of objective function to a generator temperature, b evaporator temperature, c condenser temperature

Physically, these trends for both the objective functions are due to increase in the refrigeration capacity at constant condenser and generator side parameters. This uplift in the refrigeration capacity will increase the COP and TAC both. The TAC increase can be understood by the fact that the heat exchanger (evaporator) size needs to be larger to accommodate the rise in the refrigeration capacity. This will increase the capital cost of the evaporator and hence the TAC.

The effect of variation in the condenser temperature from its optimal point value is shown in Fig. 5c. With the increase in the condenser temperature, both objective functions show decrement wherein the COP reduction is almost insignificant compared to the reduction in TAC. This suggests that compared to COP, the TAC is more sensitive to the condenser temperature. Further, in contrast to the above evaporator temperature variation effects, here the varying range of TAC (for all 5 optimal points A–E) is almost same and it is not so for the varying range of COP. Physically, the reduction in the working fluid mass flow rate demands for the lower sized condenser component and will decrease the exergy destruction as well. This leads to significant TAC reduction. Finally, one can conclude from this sensitivity analysis that (a) for an optimal point (any among A–E), the TAC is affected significantly by all three design variables while the COP gets affected by the evaporator and generator temperature; (b) on moving from A to E, COP is sensitive to condenser temperature while TAC is sensitive to generator and evaporator temperatures.

The consideration of exergy destruction cost plays significant role in deciding the total cost value of the system. To justify this with numbers, further detailed analysis is carried out at optimal point B. In case of varying the evaporator temperature, for 6 K rise in the evaporator temperature (from 278 to 285 K) and keeping condenser and generator temperature at their optimal values, the total capital cost indeed gets decreased by 226 \$/year, but the total exergy destruction cost gets increased by 1007 \$/year. Among this 1007 \$/year rise in exergy destruction cost, the ejector component exergy destruction cost shows 240 \$/year increase. Hence, this trend leads to overall rise in the TAC. Further, in case of varying the condenser temperature, for 7 K rise in the evaporator temperature (from 306 to 213 K) and keeping evaporator





and generator temperature at their optimal values, the total capital cost gets increased by 323 \$/year, but the total exergy destruction cost gets decreased by 1955 \$/year. Among this 1955 \$/year drop in exergy destruction cost, the ejector component exergy destruction cost shows 682 \$/year decrement. Hence, this trend leads to overall drop in the TAC.

From the above analysis, one can note that the exergy destruction cost considerations (exergo-economic concept) in such systems where the highly irreversible components such as ejector and generator are present play a significant role in the objective function value (TAC) change when the design variables are varied from their optimum values. This is why the sensitivity analysis on the optimization results is necessary.

Effect of nozzle throat diameter and ecological function

Three different nozzle throat diameters are considered by keeping the nozzle exit diameter unchanged. The Pareto fronts in all three cases are plotted in Fig. 6. It shows that with the increase in nozzle throat, the Pareto shifts upward,



Table 8State properties andcosts at optimal point B

K. Mansuriya et al.

State	Fluid	$T(\mathbf{K})$	Mass flow rate	Exergy rate (kW)	Costs		
			(kg/s)		$\dot{C}(/\text{year})$	u (\$/GJ)	
1	R245fa	351.6	0.113	4.022	4210.1	33.2	
2	R245fa	274.6	0.043	-0.823	-1708.3	65.8	
3	R245fa	308.5	0.156	1.206	2501.8	65.8	
4	R245fa	308.5	0.156	0.034	71.4	65.8	
5	R245fa	308.5	0.113	0.025	51.6	65.8	
6	R245fa	308.5	0.043	0.010	19.7	65.8	
7	R245fa	274.6	0.043	-0.098	-202.5	65.8	
8	R245fa	308.7	0.113	0.052	164.3	99.8	
9	Water	303.5	2.741	0.336	0	0	
10	Water	306.5	2.741	1.128	2712.7	76.2	
11	Water	282.6	0.408	0.704	0	0	
12	Water	277.6	0.408	1.251	2123.3	53.8	

i.e. for same TAC, the COP improves with the throat diameter. Moreover, for same COP, the TAC value increases with the decrease in the throat diameter. The increase in the primary stream flow rate with nozzle throat diameter is the reason for this observed trend. Also, this behaviour tells that the exergy destruction within the ejector can be reduced by increasing the nozzle throat diameter. Further, the ecological function (ECF) is varied to see its effect on the optimization results. The Pareto fronts obtained for three different values of ECF are shown in Fig. 7. High ECF value favours the COP objective significantly. For same net heat input to the system, low ECF value indicates high exergy destruction in the system and hence the high TAC. Therefore, from ecology point of view, this ECF consideration can lead to better optimization solutions.

Exergo-economic analysis results

The system performance at the optimal point B (nearest to the ideal solution) is investigated from exergoeconomic point of view. Table 8 summarizes the state properties and the cost values at each stage of the system. The main product output from the current system is the water stream being chilled in the evaporator, and therefore, the product unit cost (u_{12}) at state point 12 is found to be 53.8 \$/GJ. The improvement in the exergy efficiency can decrease this product unit cost. The exergy destruction rate and the corresponding destruction cost rate of each component are presented in Fig. 8 and Fig. 9, respectively. From total exergy input of 5.24 kW, 74.4% (3.9 kW) is the total exergy destruction in the system, 15.1% (0.79 kW) is the exergy loss in the system, and the rest 10.5% (0.55 kW) is the system exergy output. Only 10.5% utilization of the input exergy is due to the highly irreversible components such as ejector and the generator. Also, 15.1% (0.79 kW) exergy loss takes place in the



Fig. 8 Exergy destruction rate in system components at optimal point ${\rm B}$

condenser due to heat rejection at a temperature higher than the surrounding temperature. In addition to this, the highest exergy destruction of ~2 kW (38%) is in the ejector component followed by the generator with 1.21 kW (23.1%) exergy destruction rate. As ejector undergoes irreversible processes such as mixing, friction and the shock wave, it shows the highest exergy destruction while the high temperature difference is the reason for significant exergy destruction in the generator. In conclusion, the exergy efficiency at the optimal point B is found to be 10.5%. From Fig. 9, the total exergy destruction cost at the optimal point B is 3846.2 \$/year with ejector as the highest contributor followed by the cooling unit. Table 7 gives the TAC at B as 25,903 \$/year. The cooling unit (evaporator, condenser and expansion valve) cost rate is



Fig. 9 Exergy cost rate in system components at optimal point B

1379 \$/year. Upon including the initial cost of components to this destruction cost, the cooling set is having the highest TAC.

Economic comparison without and with sub-cooling arrangement

The cavitation in the pump can be avoided by using the sub-cooling arrangement that sub-cools the refrigerant leaving the condenser below saturation temperature. This sub-cooling will not allow the bubbles to get formed when the refrigerant will enter the pump and the assumption of cavitation-free pump can be justified. To consider the economic effect in the existing system with sub-cooling arrangement, a heat exchanger as sub-cooler is added before pump. It is assumed that sub-cooling of 5 °C is required to avoid the cavitation (Sokolov and Hershgal 1991). The cost results with this consideration are compared with the studied system's cost values at the best optimal point B. It is found that the rise in the TAC is 973 \$/ year (3.75% rise). Among this total increase of 973 \$/year, capital cost rises by 743 \$/year, operating cost rises by 156 \$/year, and the exergy destruction cost is increased by 74 \$/year (Table 9).

	Table 9	Thermal	model	validations
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Parameters					Values with R141b as refrigerant					
					Sadeghi	et al. (2015)		Present model		Error (%)
(a) Eject	or system m	iodel va	lidation							
Ejector s $D_{e1} = 1$	ystem mode 4.93 mm	el valida	ation with	the resul	ts of Sadeghi et al. (20	(15). $T_{\rm g} = 367.54$	K, $T_{\rm cr} = 30$	06.44 K, $T_{evap} = 273.03$	K, $D_t = 9.35 \text{ mm}$	1,
Coeffic	ient of perfe	ormance	e		0.14			0.1441		-2.93
Second	law efficien	ncy (%)			3.09			3.055		1.13
Primary	y flow rate,	ḿ _p (kg/s	s)		0.1573			0.1569		0.25
Second	ary flow rat	te, m _s (k	g/s)		0.0302			0.0299		0.99
Consta	nt area dian	neter, D	_{e2} (mm)		28.27			28.29		-0.06
Refrige	ration effec	ation effect, \dot{Q}_{evap} (kW) 6.03 5.9566			1.22					
$\overline{D_{t}(mm)}$	D _{e1} (mm)	$T_{\rm g}({\rm K})$	$T_{\text{evap}}(\mathbf{K})$	$T_{\rm cr}({\rm K})$	Entrainment ratio, (<i>i</i>	$n_{\rm s}/\dot{m}_{\rm p}$)		Area ratio, $(D_{e2}/D_t)^2$	2	
					Huang et al. (1999) Experimental values	Present model	Error %	Huang et al. (1999) Experimental values	Present model	Error %
(b) Eject	or compone	ent mode	el validatio	on						
2.64	4.5	368	281	315.1	0.1859	0.173	6.94	6.44	6.54	- 1.56
		363	281	309.7	0.304	0.2743	9.77	7.73	7.54	2.46
		368	281	306.6	0.3457	0.3276	5.24	9.41	9.32	0.94
		363	281	310.5	0.2718	0.2449	9.90	6.99	6.69	4.29
		357	285	309	0.3398	0.3634	-6.95	6.44	6.71	-4.23
		351	285	298.7	0.7412	0.6668	10.04	9.41	9.71	-3.16
2.82	5.1	368	281	304.3	0.4377	0.3965	9.41	10.64	10.74	-0.97
		368	281	306	0.3937	0.359	8.81	9.83	9.70	1.36
		368	285	307.2	0.4048	0.3654	9.73	9.17	9.43	-2.85
		368	285	312.3	0.304	0.2881	5.23	7.26	7.49	-3.21

Table 10	Effect of pump isentropic efficiency	

At optimal point B (COP= 0.3 ,	Isentropic efficiency		
TAC=25,903 \$/year)	0.6	0.5	0.4
% rise in Pump work	67	100	150
% drop in COP	0.8	1.4	2
% rise in TAC	0.3	0.6	1.1

Effect of pump isentropic efficiency on objectives

So far in the work, mechanical efficiency of 0.75 is considered in order to cover hydraulic and frictional losses. For low viscous refrigerant like R245fa, the additional pump work required due to entropy change (non-isentropic) condition can be considered using isentropic efficiency of the pump. Keeping this in mind, the overall pump efficiency can be considered as $\eta_{\text{pump}} = \eta_{\text{m,p}} \times \eta_{\text{s,p}}$ (where $\eta_{\text{m,p}}$ and $\eta_{\text{s,p}}$ are mechanical and isentropic efficiencies of pump, respectively). The isentropic efficiencies considered for pumping R245fa are 0.4, 0.5 and 0.6 (Surindra et al. 2019). At optimal point B, Table 10 shows that pump work rises significantly for chosen isentropic efficiencies compared to 100% isentropic efficiency of pump. But it is notable that there is negligible change in the COP objective even when pump isentropic efficiency is 40%. This is due to the low-grade heat-driven system where pump work is negligible compared to generator input heat. Further, in case of TAC, for 40% isentropic efficiency, the change in TAC is 1.1% which is somehow significant as running cost of less efficient pump is higher.

Conclusion

Thermodynamic modelling and multi-objective considered optimization of an exhaust heat-driven ejector refrigeration system are performed in this work. A newly developed and an efficient heat transfer search (HTS) optimization algorithm is used. The coefficient of performance and the total annual cost of the system are considered as the objective functions (thermo-economic consideration) while the system temperatures are selected as the design variables. In order to consider the cost of exergy destruction as a part of the total cost, the exergo-economic approach is adopted during economic evaluation. This exergo-economic approach helps in knowing the trade-off between the investment costs and the exergy destruction cost of the system components and hence helps in the real cost evaluation. The Pareto frontier obtained as a result of the multi-objective optimization gives the optimum design points (solutions) to the decision-maker who can implement any of the optimum design points based on the requirement and the priority of the objective functions. Also, compared to generator temperature, evaporator and condenser temperatures produce more conflicting effect between both the objective functions i.e. COP and TAC. The evaporator, condenser and generator temperatures take respective values of 274.6 K, 308.5 K and 351.6 K at the optimal point B which is found closest to the ideal point in the Pareto front. The objectives COP and TAC take respective values of 0.3 and 25,903 \$/year at this best optimal point. Further, the sensitivity analysis at 5 selected optimal points reveals that the TAC is affected significantly by all three design variables while evaporator and generator temperature affects significantly to COP as compared to condenser temperature. The exergo-economic results at optimal points B reveal that the product unit cost is 53.8 \$/ GJ and can be further decreased by increasing the exergetic efficiency of the system (10.5%). This can be achieved by reducing the exergy destruction in the ejector and the generator as they are found as the leading contributors for the same at the optimum condition.

Compliance with ethical standards

Conflict of interest The authors declare that they have no conflict of interest.

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