ORIGINAL PAPER



# A Full Dynamic Compound Inverse Method for output-only element-level system identification and input estimation from earthquake response signals

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Received: 23 December 2015 / Accepted: 6 April 2016 / Published online: 30 April 2016 © Springer-Verlag Berlin Heidelberg 2016

Abstract This paper proposes a new output-only elementlevel system identification and input estimation technique, towards the simultaneous identification of modal parameters, input excitation time history and structural features at the element-level by adopting earthquake-induced structural response signals. The method, named Full Dynamic Compound Inverse Method (FDCIM), releases strong assumptions of earlier element-level techniques, by working with a two-stage iterative algorithm. Jointly, a Statistical Average technique, a modification process and a parameter projection strategy are adopted at each stage to achieve stronger convergence for the identified estimates. The proposed method works in a deterministic way and is completely developed in State-Space form. Further, it does not require continuous- to discrete-time transformations and does not depend on initialization conditions. Synthetic earthquake-induced response signals from different shear-type buildings are generated to validate the implemented procedure, also with noisecorrupted cases. The achieved results provide a necessary condition to demonstrate the effectiveness of the proposed identification method.

**Keywords** Structural identification · Iterative optimization algorithm · Modal dynamic identification · Estimated unknown seismic input · Element-level system identification

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# **1** Introduction

The identification of structural dynamics properties appears fundamental in several engineering branches, and concerning specific technical-scientific contexts and practical applications within them. The field of earthquake engineering is considered here as a main target of interest, in particular as connected to potential Structural Health Monitoring approaches. These are devoted to the assessment of current structural properties and to the evaluation of potential damages that may occur in civil structures, due to the action of strong ground motion excitations.

In the past decades, the identification of structural dynamics properties at seismic input has been the subject of several researches, either within Experimental Modal Analysis (EMA) (e.g. [2,3,7,16,18,23]) where the excitation input is known and/or controlled, or within Operational Modal Analysis (OMA) (e.g. [17,20,31,35,44,45]), where the input action is unknown (output-only techniques). In these works, specifically-developed strategies have been developed to identify the modal parameters from earthquake-induced structural response signals.

In the branch of OMA methods, most common algorithms rely on the assumption of stationary white noise input. Then, the use of strong ground motion excitations has been attempted by a limited number of works, either in the Time Domain (e.g. [17,31,44,46,48]) or in the Frequency Domain (e.g. [25,34,35,49]).

Concerning output-only techniques in the Frequency Domain, a refined Frequency Domain Decomposition (rFDD) approach, Pioldi et al. [39,40], has been recently developed for the estimation of strong ground motion modal parameters, also at concomitant heavy damping, by addressing typical shortcomings of classical FDD [6] in this framework [49]. In both works, Pioldi et al. [39,40], synthetic response sig-

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nals have been adopted as a necessary validation condition. Then, in Pioldi et al. [41], systematic trials with real earthquake response signals have been effectively performed, by addressing also damage scenarios, retrofit stages and Soil Structure Interaction (SSI) effects. SSI effects have been also investigated in Pioldi et al. [43]. Finally, the rFDD technique has been improved by a coupled Chebyshev Type II filters procedure and a time-frequency Wavelets analysis, as outlined in Pioldi and Rizzi [42]. In such a research context, the present work makes a step forward, towards the development of an innovative parametric OMA algorithm in the Time Domain, devoted to assess not only the modal properties but also the earthquake input excitation and the unknown structural parameters of mass, damping and stiffness, as discussed below.

Indeed, even though modal parameters shall be considered as representative of the global dynamic properties of a given structure, referring to its mass, damping and stiffness characteristics, their usefulness may be insufficient in practical cases, in order to quantify the amount of damage that may have occurred within individual structural elements [19]. In fact, structural degradations will be reflected in modifications of the global structural behaviour. As a consequence, these damages are dependent on changes that may be detected in individual structural elements, i.e. at the so-called "elementlevel", specifically in terms of time variations of stiffness and damping structural characteristics (assuming time-invariant systems in terms of mass). Also, nearly all of the available techniques for modal dynamic identification and damage detection at seismic input pertain to EMA methods, i.e. they need the knowledge (or the acquisition) of the input ground motion (i.e. the knowledge of the external action) too.

At this stage, current OMA (output-only) methods may be viewed on different levels of sophistication. The first level appears to be the lower one: it leads to the estimation of structural modal properties only, i.e. of natural frequencies, mode shapes and modal damping ratios (and, in case of parametric algorithms, also of "realizations", in terms of a set of State-Space structural matrices referred to a given inputoutput behaviour [15]). The second level of detail pertains to the estimation of the "input excitation" acting on the system, specifically here of the shaking ground motion, which induces the structural response recorded from the available instrumentation. The third level of focus, finally, refers to the further refinement of mass, damping and stiffness matrix estimation. The aim is to target each of their characteristic parameters, i.e. to achieve the so-called "identification at the element-level". Thus, the simultaneous identification of modal parameters, excitation input and structural features at the element-level still appears to be rather challenging in the present dedicated OMA literature [12,30,51], and specifically in the earthquake engineering context, which constitutes a main target of the present work.

By using output-only acquisitions of the structural response, during the past years several identification techniques have been proposed to jointly estimate the structural parameters and the unknown seismic input at the base. On that, a literature overview on fundamental contributions is reported below, as an underlying framework for the innovative steps and approaches put forward later in the body of the present paper.

In 1989, Toki et al. [47], proposed that the "coda" of the response time history (i.e. the tail, namely the last seconds of the acquisition following the end of the earthquake excitation) may be representative of the free vibration response of the structural system. Their assumption was that the coda is not affected by the input ground motion, and then the ratio between mass and stiffness or damping coefficients shall be constant from this part of the record on. So, modal parameters may be identified by Extended Kalman Filtering (EKF), see Hoshiya and Sutoh [21]. Also, they improved the method to estimate the input ground motion by the Kalman Filter estimation error [36].

In 1994, Benedetti and Gentile [5], suggested a Frequency Domain algorithm, which worked in two stages, to identify the properties of structures subjected to input ground motion. With this method, the responses at two different locations of the structure were assumed to be available, while the requirements for the input excitation were avoided, by using the ratio between the Fourier amplitude transforms of the two acquired response signals. However, the input time history could not be determined by that identification procedure.

In the same year (1994) Wang and Haldar [51], proposed an iterative Finite Element-based procedure in the Time Domain to evaluate both stiffness and damping matrices at the element-level, jointly with the input force excitation (which could be a generic force or seismic loading). Only a small number of observation time points was required for the development of the algorithm, where the solicitations were assumed to be initially zero for the starting of the iterative Least-Squares (LS) procedure, and then were updated stepby-step, until convergence was reached. Finally, the complete input time history could be evaluated by using the estimated stiffness and damping parameters and the full-length measured structural responses.

In 1995, Hoshiya and Sutoh [22], suggested a method to identify both the input and the stiffness and damping matrix coefficients for shear-type frames. A combination of a smoothing algorithm on the input ground excitation and of an Extended Kalman Filtering was formulated, by adopting the coda of the system responses. They used a Weighted Global Iteration (WGI) procedure to obtain fast convergence to the optimum solution and to achieve stability, as previously outlined by Hoshiya and Sutoh [21]. A problem related to these methods concerned how to choose the coda of the response time history, i.e. the exact start of the free response part of the acquisition, without a prior knowledge of the input ground motion excitation.

Afterwards, in 1997, Haldar et al. [19], expanded their previous work to cases with limited response measurements (or limited observations), i.e. when the structure was monitored only at few dynamic degrees of freedom. They combined their former iterative LS algorithm with a Weighted Global Iteration procedure applied to an Extended Kalman Filter technique.

In 2000, Li and Chen [29], proposed an iterative Time Domain method for solving hybrid inverse problems. Structural parameters and inverse time history of input excitation were simultaneously estimated, by relying on structural response signals only. In that work, additional conditions were set on the input force, by relying on its mechanical characteristics. Then, a Statistical Average (SA) algorithm was developed in terms of a properly-defined transfer coefficient. The computational steps of the algorithm relied on their previous work, Li and Chen [28].

In the same year (2000), Cho and Paik [10], refined the algorithm by Wang and Haldar [51], by proposing an improved LS method based on Multiple models QR Decomposition (MQRD). This method enhanced the LS convergence and allowed for a more accurate identification of structural elements and potential damages.

In 2002, Chen et al. [8], took the element-level Time Domain method developed by Wang and Haldar [51] and implemented a refinement, which looked to identify both the earthquake excitation and the structural parameters (at the element-level) by using a combination of an iterative LS technique and of a specifically-developed Statistical Average method. They named the algorithm as *Dynamic Compound Inverse Method (DCIM)*, label which has been kept here as a base for the subsequent developments put forward in the present work.

Along the same line, in 2003, Li and Chen [30], published in the Computational Mechanics journal an improved version of their previous work Li and Chen [29], by improving the Statistical Average algorithm for the solution of the DCIM. The adopted solicitations spread from ambient to seismic ground motion, acting on different structures. In 2004, Chen and Li [9], again in Computational Mechanics, proposed a further implementation, which took into account the use of Rayleigh proportional damping  $\mathbf{C} = \alpha \mathbf{M} + \beta \mathbf{K}$ , instead of general viscous damping used until then. The unknown damping coefficients  $\alpha$  and  $\beta$  and the stiffness parameters of matrix **K** were coupled to each other, leading the classical MDOF equations of motion to constitute a set of non-linear identification equations in terms of the unknown structural parameters. Then, by relying on their previous algorithm, they developed a two-stage iterative method to avoid the nonlinear identification problem. An inner modification process was applied between each iterative step, to convert the spatial information of the external excitation into mathematical conditions for the iterative algorithm. In that work, the focus was on possible forced vibration surveys of the system, i.e. on cases where the structure was excited by actuators installed at several locations of the building. No ground motions or ambient vibrations were considered at that stage. The present paper relies much on the two contributions above, and attempts to provide a more general and robust analysis, as it is going to be outlined below, towards achieving a full identification approach (*Full Dynamic Compound Inverse Method-FDCIM*).

Still in 2004, Ling and Haldar [32], developed a further approach to take into account the possible use of Rayleigh proportionally-damped systems. They proposed a modified Iterative LS method with Unknown Input (ILS-UI) to identify such types of systems at the element-level. They used Taylor series approximations to transform the non-linear set of identification equations (arising from the equations of motion with Rayleigh damping) into a set of linear equations. Again, no ground motions or ambient vibrations were considered.

In 2006, Zhao et al. [54], demonstrated that the structural parameters and the earthquake ground motion could not be uniquely identified when absolute structural response signals were used, instead of relative ones. Then, they proposed a hybrid identification method to tackle the problem. First, that algorithm identified the structural parameters above the first-floor of a multi-storey shear-type building, by using LS. Later, the minimum modal information was introduced to find out the first floor parameters and to avoid the shortcoming of non-uniqueness. After that, all structural parameters were identified. The unknown earthquake ground motion was estimated by solving a first-order differential equation.

In 2008, Perry and Koh [38], proposed an element-level parameter and input estimation with a non-classical approach based on a Genetic Algorithm (GA). They made use of a modified GA-based method, to adopt the strategy of the Search Space Reduction Method, Perry et al. [37]. This method was revised by Perry and Koh [38] through a modification of the numerical integration scheme, and was allowed to be used also for output-only identification. Still in 2008, Wang and Cui [52], suggested a two-step method, which generalized the work of Chen et al. [8], apt to identify both element-level structural parameters and input time history.

Concerning the intelligent monitoring of multi-storey buildings by a wireless sensor network, in 2012, Lei et al. [27], suggested an element-level and input ground motion identification algorithm based on a two-stage Kalman Filter estimator. That method worked with absolute coordinates through the use of sub-structuring methods. Still in 2012, Lourens et al. [33], proposed an extension of an existing joint input-state estimation Kalman Filter, which was derived by using a linear minimum-variance unbiased estimation. The novelty consisted in the use of an optimal estimate in place of the true value of the input for the Kalman Filter steps.

Then, in 2014, Ding et al. [14], developed a method which could be used to identify structural parameters and input time history, either for linear or non-linear hysteretic frames with limited observations. With this method, they decomposed the structural excitation by orthogonal approximation and then applied an Extended Kalman Filter for the identification. Only forces acting on the floors of the frames were considered at that stage.

Recently, Bayesian methods also started to be applied for OMA input and element-level identification. Relevant examples may be found in Behmanesh et al. [4], where a probabilistic Finite Element (FE) model updating technique based on Hierarchical Bayesian modelling was proposed for the identification of civil structural systems under changing ambient/environmental conditions. Further, the works of Au and Zhang [1] and Zhang and Au [53], suggested the adoption of a modified two-stage Bayesian identification approach, towards the estimation of modal properties and structural parameters.

Still in 2015, Concha et al. [12], suggested an adaptive observer which simultaneously estimated the damping/mass and the stiffness/mass ratios, as well as the state of seismically-excited shear buildings. Their work started from the research of Jiménez and Icaza [24], where the parametrization of a shear-type frame equipped with a Magneto-Rheological Damper (MRD) had been considered; the assumptions were the knowledge of the structural parameters (at a first approximation, accounting for some uncertainties) and of the seismic ground acceleration, the accelerations of each floor and the MRD damper force (direct measurement). Similar assumptions were assumed in Concha et al. [12], where the identification scheme relied on the knowledge (i.e. on the measurement) of ground and floor accelerations and of the first floor mass  $m_1$ . This method achieved the estimation of a reduced-order model, if some floors only may be equipped with available instrumentation. Computationally, the algorithm relied on LS and on a Luenberger state estimator for the estimation process.

A crucial common denominator between all the aforementioned works is the fact that they take the full mass matrix to be given for granted, both in its matrix structure and in its mass parameters. Then, despite the output-only feature of the algorithms, the knowledge of structural responses only (either in terms of accelerations, velocities or displacements) is not sufficient for their functioning. A slight exception comes from the work of Concha et al. [12], where the first floor mass  $m_1$  and the seismic ground acceleration has been considered to be known, to return the element-level identifications for all the stiffness, damping and mass matrices.

In this framework, the present paper proposes a new, complete element-level system identification and input esti-

mation algorithm, hereafter named Full Dynamic Compound Inverse Method (FDCIM), following the terminology introduced in Computational Mechanics by Chen et al. [8]. This is specifically developed to operate with earthquake-induced response signals, collected from seismically-excited MDOF shear-type frames. The method releases strong assumptions implied by earlier techniques, especially the required knowledge of the full mass matrix (or at least of its specific elements), to provide an effective identification of the modal parameters of the system, i.e. natural frequencies, mode shapes and modal damping ratios, of the excitation input and of a realization of the state matrices. In fact, the present FDCIM algorithm requires the knowledge of structural response signals only, while mass, damping (handled here as general viscous damping; thus, an independent feature to be identified) and stiffness matrices, and obviously the exciting earthquake input, may be completely unknown. Accurate estimations of the input ground motion time history and of the states at the element-level are provided through a two-stage iterative algorithm. This method works jointly with a Statistical Average technique, a modification process and a parameter projection strategy, which are adopted at each iteration to provide correct and stronger constraints for the estimates, allowing for faster and much reliable convergence.

Moreover, the proposed FDCIM method (as opposed to other methods operating in the stochastic framework) appears to be completely deterministic and it is fully developed in *State-Space form.* It does not require transformations from continuous-time to discrete-time and it does not depend on the adopted initial conditions or on the state estimation, in order to identify the modal parameters and the input ground motion. Further, all structural features as element-level mass, damping and stiffness matrices, may be accurately identified merely by knowing any single component of one of these matrices, or even by relying on an estimate of a global structural parameter, like for instance the total mass of the building. Then, the adopted formulation is suitable for integration or support to other common output-only methods working with State-Space parametric Time Domain frameworks.

Presentation of the paper is structured as follows. Subsequent Sect. 2 presents the mathematical model, the basic governing equations and the derivation and theoretical framework of the present Full Dynamic Compound Inverse Method. The case of general viscous damping is detailed as well, in all its formulations. Section 3 presents a first validation of the developed algorithm with simulated data from a 3-storey frame, by describing results for the estimated modal parameters and input ground motion excitation and for the accurate element-level identification, as a function of the adopted number of record points. Section 4 outlines comprehensive results on the application of the FDCIM approach to noise-corrupted earthquake-induced synthetic response signals, coming from the same examples adopted in Sect. 3. Various levels of added noise have been considered, by aiming at getting closer to real conditions, i.e. resembling those that may be encountered in practice for real seismic response recordings. Then, further demonstrative analyses on a challenging 10-storey realistic structure from the literature are additionally produced and reported in Sect. 5, where results are considered again as a function of the adopted number of record points and by taking into account noise-corrupted signals. In the end, main conclusions are gathered in Sect. 6.

#### 2 Full Dynamic Compound Inverse Method

#### 2.1 Basic mathematical model

The response of a linear MDOF shear-type building subjected to earthquake base excitation, characterized by ground acceleration  $\ddot{u}_g(t)$ , is governed by the following classical set of second-order time-differential equations:

$$\mathbf{M}\ddot{\mathbf{u}}(t) + \mathbf{C}\dot{\mathbf{u}}(t) + \mathbf{K}\mathbf{u}(t) = \mathbf{f}(t) = -\mathbf{M}\ddot{\mathbf{u}}_g(t) = -\mathbf{M}\mathbf{r}\ddot{u}_g(t)$$
(1)

where **M**, **C** and **K**  $\in \mathbb{R}^{n \times n}$  are the mass, damping and stiffness matrices of the structural system, respectively, with the following classical definitions:

**C**<sub>1</sub>

ü

k,

h,

ü

 $\mathbf{c}_1 \mid \mathbf{k}_1$ 

ü,

$$\mathbf{M} = \begin{bmatrix} m_1 & 0 & 0 & \dots & 0 \\ 0 & m_2 & 0 & \dots & 0 \\ 0 & 0 & m_3 & \dots & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & \dots & m_n \end{bmatrix} [kg],$$





$$\mathbf{C} = \begin{bmatrix} c_1 + c_2 & -c_2 & 0 & \dots & 0 \\ -c_2 & c_2 + c_3 & -c_3 & \dots & 0 \\ 0 & -c_3 & c_3 + c_4 & \dots & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & \dots & c_n \end{bmatrix} [\text{Ns/m}] (2)$$
$$\mathbf{K} = \begin{bmatrix} k_1 + k_2 & -k_2 & 0 & \dots & 0 \\ -k_2 & k_2 + k_3 & -k_3 & \dots & 0 \\ 0 & -k_3 & k_3 + k_4 & \dots & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & \dots & k_n \end{bmatrix} [\text{N/m}]$$

where *n* is the number of floors,  $m_i$  are the floor masses,  $c_i$  and  $k_i$  (i = 1, ..., n) are the lateral column damping and stiffness coefficients, respectively, between the *i*th and the (i - 1)th floor. Matrix **M** is diagonal and matrices C, K share a tridiagonal (symmetric) structure. Vector  $\mathbf{u} = [u_1 \ u_2 \ u_3 \dots, u_n]^{\mathrm{T}} \in \mathcal{R}^{n \times 1}$  gathers the absolute (relative to the ground) displacements of each floor. Consistently, vectors  $\dot{\mathbf{u}}$  and  $\ddot{\mathbf{u}}$  represent absolute (relative to the ground) velocities and accelerations. Finally, the input force vector  $\mathbf{f}(t) \in \mathcal{R}^{n \times 1}$  is the vector containing the acting excitation input, here coming from the ground acceleration. It is defined as  $\mathbf{f}(t) = -\mathbf{M}\ddot{\mathbf{u}}_{g} = -\mathbf{M}\mathbf{r}\ddot{u}_{g}$ , where  $\mathbf{r} = [1 \ 1 \ 1 \ \dots \ 1]^{\mathrm{T}} \in$  $\mathcal{R}^{n \times 1}$  is the influence coefficient (rigid body motion) vector for the analysed case and  $\ddot{\mathbf{u}}_g = \mathbf{r}\ddot{u}_g$  is the ground acceleration vector. Figure 1 sketches the initial geometry, the lumpedmass scheme and the displacements of the adopted shear-type

tion  $\dot{\mathbf{x}}(t)$  and observer equation  $\mathbf{y}(t)$ , by sticking to classical literature definitions (see e.g. [15]):

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{f}(t) \\ \mathbf{y}(t) = \mathbf{C}_o \mathbf{x}(t) + \mathbf{D}\mathbf{f}(t) \end{cases}$$
(3)

where  $\mathbf{x}(t) = [\mathbf{u}(t) \ \dot{\mathbf{u}}(t)]^{\mathrm{T}} \in \mathcal{R}^{2n \times 1}$  is the state vector and  $\dot{\mathbf{x}}(t)$  is its derivative, while  $\mathbf{y}(t) \in \mathcal{R}^{n \times 1}$  is the output vector. Moreover,  $\mathbf{A} \in \mathcal{R}^{2n \times 2n}$  is the state matrix,  $\mathbf{B} \in \mathcal{R}^{2n \times n}$  the input matrix,  $\mathbf{C}_o \in \mathcal{R}^{n \times 2n}$  the output matrix and  $\mathbf{D} \in \mathcal{R}^{n \times n}$  the feed-through matrix, which can be defined as follows:

$$\mathbf{A} = \begin{bmatrix} \mathbf{0}_{n \times n} & \mathbf{I}_{n \times n} \\ -\mathbf{M}^{-1}\mathbf{K} & -\mathbf{M}^{-1}\mathbf{C} \end{bmatrix}, \quad \mathbf{B} = \begin{bmatrix} \mathbf{0}_{n \times n} \\ \mathbf{M}^{-1} \end{bmatrix}$$
(4)  
$$\mathbf{C}_o = \begin{bmatrix} -\mathbf{M}^{-1}\mathbf{K} & -\mathbf{M}^{-1}\mathbf{C} \end{bmatrix}, \qquad \mathbf{D} = \mathbf{M}^{-1}$$

Matrix terms  $\mathbf{0}_{n \times n}$  and  $\mathbf{I}_{n \times n} \in \mathcal{R}^{n \times n}$  indicate zero and identity matrices of the specified dimensions, respectively.

In the present identification implementation, the known quantities are only the dynamic responses, i.e.  $\mathbf{x}(t)$  and  $\dot{\mathbf{x}}(t)$  state vectors, while state matrices (i.e.  $\mathbf{A}$ ,  $\mathbf{B}$ ,  $\mathbf{C}_o$  and  $\mathbf{D}$  matrices) and input vector  $\mathbf{f}(t)$  are the unknown variables to be identified. Accordingly, by switching back to physical space, the parameters which will be estimated are the modal characteristics, the seismic ground acceleration  $\ddot{u}_g$  and the mass, damping and stiffness matrices (i.e. their coefficients).

Thus, while typically Eqs. (1) or (3) are conceived to be formed and solved for the unknown structural responses, in the present identification process the role of given information and unknown variables is reversed and effective algebraic rewriting of the equations of motion, leading to appropriate identification equations, is sought, in view of expressing the new role of the unknown variables (the identification variables). This is pursued next.

It can be noticed that the estimation problem, by taking into account the desired unknowns, turns out to be undetermined, in the sense that is going to be revealed in the following. In order to handle this issue, from Eq. (1) it is possible to write the two following concatenated relations, namely the two basic equations on which the present recursive algorithm is built:

$$\mathbf{C}\dot{\mathbf{u}}(t) + \mathbf{K}\mathbf{u}(t) = -\mathbf{M}\left(\ddot{\mathbf{u}}(t) + \ddot{\mathbf{u}}_g(t)\right)$$
(5)

$$\mathbf{M}^{-1} \Big( \mathbf{C} \dot{\mathbf{u}}(t) + \mathbf{K} \mathbf{u}(t) \Big) = - \ddot{\mathbf{u}}(t) - \ddot{\mathbf{u}}_g(t)$$
(6)

where Eq. (6) may be derived directly from Eq. (5), by  $\mathbf{M}^{-1}$  pre-multiplication. Notice that mass matrix  $\mathbf{M}$  can always be inverted, since it is taken as a diagonal, non-singular matrix. The switching of Eqs. (5) and (6) to State-Space representation is immediate, and leads to the following formulations:

$$\mathbf{G}_{ck}\mathbf{x}(t) = \mathbf{N}_{ck}\dot{\mathbf{x}}(t) + \mathbf{L}_{ck}\mathbf{f}(t) = \mathbf{N}_{ck}\dot{\mathbf{x}}(t) - \mathbf{L}_{ck}\mathbf{M}\ddot{\mathbf{u}}_g(t)$$
(7)  
$$\mathbf{G}_m\mathbf{x}(t) = -\dot{\mathbf{x}}(t) - \mathbf{L}_m\ddot{\mathbf{u}}_g(t)$$
(8)

where  $\mathbf{G}_{ck} \in \mathcal{R}^{2n \times 2n}$ ,  $\mathbf{N}_{ck} \in \mathcal{R}^{2n \times 2n}$ ,  $\mathbf{L}_{ck} \in \mathcal{R}^{2n \times n}$ ,  $\mathbf{G}_m \in \mathcal{R}^{2n \times 2n}$  and  $\mathbf{L}_m \in \mathcal{R}^{2n \times n}$  are matrices to be specifically assembled, defined as follows:

$$\mathbf{G}_{ck} = \begin{bmatrix} \mathbf{0}_{n \times n} & \mathbf{I}_{n \times n} \\ \mathbf{K} & \mathbf{C} \end{bmatrix}, \quad \mathbf{N}_{ck} = \begin{bmatrix} \mathbf{I}_{n \times n} & \mathbf{0}_{n \times n} \\ \mathbf{0}_{n \times n} & -\mathbf{M} \end{bmatrix}$$
$$\mathbf{L}_{ck} = \begin{bmatrix} \mathbf{0}_{n \times n} \\ \mathbf{I}_{n \times n} \end{bmatrix}$$
(9a)

$$\mathbf{G}_{m} = \begin{bmatrix} \mathbf{0}_{n \times n} & -\mathbf{I}_{n \times n} \\ \mathbf{M}^{-1}\mathbf{K} & \mathbf{M}^{-1}\mathbf{C} \end{bmatrix}, \qquad \mathbf{L}_{m} = \begin{bmatrix} \mathbf{0}_{n \times n} \\ \mathbf{I}_{n \times n} \end{bmatrix}$$
(9b)

Then, Eq. (7) can be rewritten, with the purpose of a firststage identification (described in subsequent Sect. 2.2), in the following form (see [30]):

$$\mathbf{H}_{ck}(t)\boldsymbol{\theta}_{ck} = \mathbf{P}_{ck}(t)$$

$$\Rightarrow \underbrace{\begin{bmatrix} \mathbf{H}_{ck}^{1}(t) & \mathbf{0}_{n \times 2n} \\ \mathbf{0}_{n \times n} & \mathbf{H}_{ck}^{2}(t) \end{bmatrix}}_{2n \times 3n} \underbrace{\begin{bmatrix} \boldsymbol{\theta}_{ck}^{1} \\ \boldsymbol{\theta}_{ck}^{2} \\ 3n \times 1 \end{bmatrix}}_{3n \times 1} = \underbrace{\begin{bmatrix} \mathbf{P}_{ck}^{1}(t) \\ \mathbf{P}_{ck}^{2}(t) \end{bmatrix}}_{2n \times 1} \quad (10)$$

Last Eq. (10) is defined for every time instant  $t = t_i$ , i = 1, ..., L, being L the total number of points of the acquired input signals, i.e. the length of the signal. By collecting together all the sampling time instants, the subsequent formulation can be reached:

$$\underbrace{\mathbf{H}_{ck}}_{(L\times2n)\times3n} \quad \underbrace{\boldsymbol{\theta}_{ck}}_{3n\times1} = \underbrace{\mathbf{P}_{ck}}_{(L\times2n)\times1} \tag{11}$$

where:

$$\mathbf{H}_{ck} = [\mathbf{H}_{ck}(t_1) \ \mathbf{H}_{ck}(t_2) \ \dots \ \mathbf{H}_{ck}(t_L)]^{\mathrm{T}}$$
(12a)  
$$\boldsymbol{\theta}_{ck} = \{1 \ 1 \ 1 \ \dots \ 1, \ c_1 \ c_2 \ c_3 \ \dots \ c_i \ \dots \ c_n, \ k_1 \ k_2 \ k_3 \ \dots \ k_n \ \}^{\mathrm{T}}$$

$$\underbrace{\underbrace{1\times n}_{1\times n}}_{1\times n},\underbrace{\underbrace{1\times 2\times 3\times 1\times 1}_{1\times n},\underbrace{1\times 2\times 3\times 1\times 1}_{1\times n}}_{1\times n}$$
(12b)

$$\mathbf{P}_{ck} = \{\mathbf{P}_{ck}(t_1) \ \mathbf{P}_{ck}(t_2) \ \dots \ \mathbf{P}_{ck}(t_L)\}^{\mathrm{T}}$$
(12c)

Then,  $\mathbf{H}_{ck}(t)$  is a matrix containing the (known) velocity and displacement responses,  $\boldsymbol{\theta}_{ck}$  is a vector containing all the (unknown, to be identified) damping and stiffness parameters and  $\mathbf{P}_{ck}(t)$  is a vector related to the (unknown) mass terms, containing both (known) acceleration responses and (unknown) input ground motion excitation. Their elements can be computed, at the generic time instant  $t_i$ , as:

$$\mathbf{H}_{ck}^{1}(t_{i}) = \begin{bmatrix} \dot{u}_{1} & 0 & 0 & \cdots & 0 \\ 0 & \dot{u}_{2} & 0 & \cdots & 0 \\ 0 & 0 & \dot{u}_{3} & \cdots & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & \cdots & \dot{u}_{n} \end{bmatrix}, \quad \boldsymbol{\theta}_{ck}^{1} = \begin{bmatrix} 1 \\ 1 \\ 1 \\ \vdots \\ 1 \\ \vdots \\ 1 \\ \vdots \\ 1 \\ n \times 1 \end{bmatrix},$$

$$\mathbf{P}_{ck}^{1}(t_{i}) = \underbrace{ \begin{cases} \dot{u}_{1} \\ \dot{u}_{2} \\ \dot{u}_{3} \\ \vdots \\ \dot{u}_{n} \\ \vdots \\ \dot{u}_{n} \\ \vdots \\ \vdots \\ u_{n} \\ n \times 1 \end{bmatrix}}$$
(13)

By collecting all the sampling time instants together one has:

$$\underbrace{\mathbf{H}_{m}}_{(L\times2n)\times2n} \quad \underbrace{\boldsymbol{\theta}_{m}}_{2n\times1} = \underbrace{\mathbf{P}_{m}}_{(L\times2n)\times1}$$
(17)

where:

$$\mathbf{H}_m = [\mathbf{H}_m(t_1) \ \mathbf{H}_m(t_2) \ \dots \ \mathbf{H}_m(t_L)]^{\mathrm{T}}$$
(18a)

$$\boldsymbol{\theta}_{m} = \{\underbrace{1 \ 1 \ 1 \ \dots \ 1}_{1 \times n}, \underbrace{1/m_{1} \ 1/m_{2} \ 1/m_{3} \ \dots \ 1/m_{i} \ \dots \ 1/m_{n}}_{1 \times n}\}^{1}$$
(18b)

$$\mathbf{P}_m = \{\mathbf{P}_m(t_1) \ \mathbf{P}_m(t_2) \ \dots \ \mathbf{P}_m(t_L)\}^{\mathrm{T}}$$
(18c)

So,  $\mathbf{H}_m(t)$  is again a matrix containing the (known) velocity and displacement responses,  $\boldsymbol{\theta}_m$  is a vector containing all the inverse of the (unknown, to be identified) mass parameters and  $\mathbf{P}_m(t)$  is a vector collecting the (known) acceleration

$$\mathbf{H}_{ck}^{2}(t_{i}) = \underbrace{\begin{bmatrix} \dot{u}_{1} \ \dot{u}_{1} - \dot{u}_{2} & 0 & \cdots & 0 & u_{1} \ u_{1} - u_{2} & 0 & \cdots & 0 \\ 0 \ \dot{u}_{2} - \dot{u}_{1} \ \dot{u}_{2} - \dot{u}_{3} \cdots & 0 & 0 \ u_{2} - u_{1} \ u_{2} - u_{3} \cdots & 0 \\ 0 & 0 & \dot{u}_{3} - \dot{u}_{2} \cdots & 0 & 0 & 0 \ u_{3} - u_{2} \cdots & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & \cdots & \dot{u}_{n} - \dot{u}_{n-1} & 0 & 0 & 0 & \cdots & u_{n} - u_{n-1} \end{bmatrix}}_{n \times 2n}$$

$$\boldsymbol{\theta}_{ck}^{2} = \{\underbrace{c_{1} \ c_{2} \ c_{3} \ \ldots \ c_{i} \ \ldots \ c_{n}, \ k_{1} \ k_{2} \ k_{3} \ \ldots \ k_{i} \ \ldots \ k_{n}}_{1 \times 2n}\}^{\mathrm{T}}$$

$$\mathbf{P}_{ck}^{2}(t_{i}) = \underbrace{\begin{bmatrix} -m_{1}(\ddot{u}_{1} + \ddot{u}_{g}) \\ -m_{2}(\ddot{u}_{2} + \ddot{u}_{g}) \\ \vdots \\ -m_{n}(\ddot{u}_{n} + \ddot{u}_{g}) \end{bmatrix}}_{n \times 1}$$

$$(14a)$$

Finally, the  $\theta_{ck}$  parameters may be estimated by applying a LS technique to previous Eq. (11):

$$\boldsymbol{\theta}_{ck} = \mathbf{H}_{ck}^{\dagger} \mathbf{P}_{ck} = \left( \left[ \mathbf{H}_{ck}^{\mathrm{T}} \mathbf{H}_{ck} \right]^{-1} \mathbf{H}_{ck}^{\mathrm{T}} \right) \mathbf{P}_{ck}$$
(15)

where superscript symbol † indicates the Moore-Penrose pseudo-inverse.

Similarly, for a second identification stage (see Sect. 2.2), Eq. (8) can be rewritten, for every time instant  $t = t_i$ , i = 1, ..., L, as:

$$\mathbf{H}_{m}(t)\boldsymbol{\theta}_{m} = \mathbf{P}_{m}(t)$$

$$\Rightarrow \underbrace{\begin{bmatrix} \mathbf{H}_{m}^{1}(t) & \mathbf{0}_{n \times n} \\ \mathbf{0}_{n \times n} & \mathbf{H}_{m}^{2}(t) \end{bmatrix}}_{2n \times 2n} \underbrace{\begin{bmatrix} \boldsymbol{\theta}_{m}^{1} \\ \boldsymbol{\theta}_{m}^{2} \\ 2n \times 1 \end{bmatrix}}_{2n \times 1} = \underbrace{\begin{bmatrix} \mathbf{P}_{m}^{1}(t) \\ \mathbf{P}_{m}^{2}(t) \end{bmatrix}}_{2n \times 1}$$
(16)

responses and the (unknown) input ground motion excitation. By considering a generic time instant  $t_i$ , each of their elements can be calculated as:

$$\mathbf{H}_{m}^{1}(t_{i}) = -\begin{bmatrix} \dot{u}_{1} & 0 & 0 & \cdots & 0\\ 0 & \dot{u}_{2} & 0 & \cdots & 0\\ 0 & 0 & \dot{u}_{3} & \cdots & 0\\ \vdots & \vdots & \vdots & \ddots & \vdots\\ 0 & 0 & 0 & \cdots & \dot{u}_{n} \end{bmatrix}, \quad \boldsymbol{\theta}_{m}^{1} = \begin{bmatrix} 1\\ 1\\ 1\\ \vdots\\ 1\\ \vdots\\ 1 \end{bmatrix},$$
$$\mathbf{P}_{ck}^{1}(t_{i}) = -\begin{bmatrix} \dot{u}_{1}\\ \dot{u}_{2}\\ \dot{u}_{3}\\ \vdots\\ \dot{u}_{n} \end{bmatrix},$$
$$(19)$$

$$\mathbf{H}_{m}^{2}(t_{i}) = \mathbf{K} \begin{bmatrix} u_{1} \ 0 \ 0 \cdots \ 0 \\ 0 \ u_{2} \ 0 \cdots \ 0 \\ 0 \ 0 \ u_{3} \cdots \ 0 \\ \vdots \ \vdots \ \vdots \ \ddots \ \vdots \\ 0 \ 0 \ 0 \cdots \ u_{n} \end{bmatrix} + \mathbf{C} \begin{bmatrix} \dot{u}_{1} \ 0 \ 0 \cdots \ 0 \\ 0 \ \dot{u}_{2} \ 0 \cdots \ 0 \\ 0 \ \dot{u}_{3} \cdots \ 0 \\ \vdots \ \vdots \ \ddots \ \vdots \\ 0 \ 0 \ 0 \cdots \ \dot{u}_{n} \end{bmatrix}$$

$$\boldsymbol{\theta}_{m}^{2} = \{\underbrace{1/m_{1} \ 1/m_{2} \ 1/m_{3} \ \dots \ 1/m_{i} \ \dots \ 1/m_{n}}_{1 \times n} \}^{\mathrm{T}}$$
(20a)
$$\left( \underbrace{-(\ddot{u}_{1} + \ddot{u}_{g})}_{(\ddot{u} + \ddot{u} \ y)} \right)$$

$$\mathbf{P}_{m}^{2}(t_{i}) = \underbrace{\begin{cases} -(\ddot{u}_{1} + \ddot{u}_{g}) \\ -(\ddot{u}_{2} + \ddot{u}_{g}) \\ -(\ddot{u}_{3} + \ddot{u}_{g}) \\ \vdots \\ -(\ddot{u}_{n} + \ddot{u}_{g}) \end{cases}}_{n \times 1}$$
(20c)

As seen in Eq. (15), the mass parameters collected in  $\theta_m$  can be estimated by applying a LS technique to Eq. (17):

$$\boldsymbol{\theta}_m = \mathbf{H}_m^{\dagger} \mathbf{P}_m = \left( \left[ \mathbf{H}_m^{\mathrm{T}} \mathbf{H}_m \right]^{-1} \mathbf{H}_m^{\mathrm{T}} \right) \mathbf{P}_m$$
(21)

Notice that, as commented earlier right after Eq. (4), the achieved algebraic representations in Eqs. (11) and (17) invert the role of unknown parameters and given quantities, as in Eqs. (1) and (3), so as to allow solving for the identification process in the unknown quantities to be identified, for a total number of 3n unknown structural parameters and unknown input ground motion signal.

By summarizing the introduced problem and its formulation so far, it can be noted that it does not look one of a simple solution, since unknown quantities are collected inside  $\mathbf{P}_{ck}$ and  $\mathbf{H}_m$ . Then, a two-stage iterative solution provided by the present Full Dynamic Compound Inverse Method (FDCIM) is innovatively developed below, to deal with the present identification inverse problem, as it is presented in the next section.

#### 2.2 Two-stage identification algorithm

The developed Full Dynamic Compound Inverse Method (FDCIM) is an algorithm that, by working with a State-Space representation, allows for the estimation, in series, of modal parameters, input ground motion and structural characteristics at the element-level. This is possible through a two-stage iterative algorithm, which consists of a LS optimization technique for parameter identification, Lawson and Hanson [26], and of a Statistical Average (SA) method, Chen et al. [8], which make the estimated input excitation to comply with the dynamic equilibrium of the considered frame, at each

time instant, and rely on a modification process, allowing for faster convergence.

The only known quantities are the time histories acquired from the building responses; especially, only acceleration responses may be known, since velocities and displacements may be integrated numerically from accelerations, see [12]. Then, the development of the FDCIM algorithm, divided in two subsequent stages, may be summarized in the computational steps outlined below.

# • First stage (realization of stiffness and damping parameters):

- 1.1 For the first iteration, since both  $\theta_{ck}$ ,  $\theta_m$  and  $\ddot{u}_g$  are unknown quantities, each parameter vector must be assigned to an arbitrary initial value, for example  ${}^{0}\theta_{ck} =$  $\{1 \ 1 \ 1 \ \dots \ 1\}^{T}, {}^{0}\theta_m = \{1 \ 1 \ 1 \ \dots \ 1\}^{T}$ , where left apex symbol 0 stays for initial starting value. For the subsequent recursions, a similar left apex *i* denotes the current iteration. Notice that the performance of the method is not sensitive to these initial values, which may be whatever between, say,  $10^{-6}$  and  $10^{6}$ . Then, initial matrices  ${}^{0}\mathbf{M}, {}^{0}\mathbf{C}, {}^{0}\mathbf{K}$  (through Eqs. (2)) and  ${}^{0}\mathbf{H}_{ck}$ (through Eq. (12a)) can be computed.
- 1.2 An estimate of vector  $\mathbf{P}_{ck}$  can be calculated by using the terms of Step 1, from Eq. (11), getting  ${}^{0}\tilde{\mathbf{P}}_{ck} = {}^{0}\mathbf{H}_{ck}{}^{0}\boldsymbol{\theta}_{ck}$ . Overmarking symbol ~ stays here for estimated value.
- 1.3 By using  ${}^{0}\mathbf{P}_{ck}$ , matrix  $\mathbf{G}_{ck}$  can be estimated from the following LS expression:

$${}^{0}\tilde{\mathbf{G}}_{ck} = {}^{0}\tilde{\mathbf{P}}_{ck}\mathbf{X}^{\dagger} = {}^{0}\tilde{\mathbf{P}}_{ck}\left(\left[\mathbf{X}^{\mathrm{T}}\mathbf{X}\right]^{-1}\mathbf{X}^{\mathrm{T}}\right), \qquad (22)$$

where  $\mathbf{X} = {\{\mathbf{x}(t_1) \ \mathbf{x}(t_2) \ \dots \ \mathbf{x}(t_L)\}^T \in \mathcal{R}^{(L \times 2n) \times 1}$  is the global state matrix, accounting for all *L* time instants.

- 1.4 Knowing  ${}^{0}\theta_{m}$ ,  ${}^{0}\tilde{N}_{ck}$  can be reconstructed from Eq. (9a).
- 1.5 Starting from Eq. (7), an estimate of the input ground motion vector  ${}^{0}\ddot{\mathbf{u}}_{g,k}(t)$  can be obtained:

$${}^{0}\tilde{\mathbf{\tilde{u}}}_{g,k}(t) = -\left(\mathbf{L}_{ck}{}^{0}\mathbf{M}\right)^{\dagger} \left({}^{0}\tilde{\mathbf{G}}_{ck}\mathbf{x}(t) - {}^{0}\tilde{\mathbf{N}}_{ck}\dot{\mathbf{x}}(t)\right),$$
(23)

where  ${}^{0}\tilde{\mathbf{u}}_{g,k}(t) = \{{}^{0}\tilde{\vec{u}}_{g,k}^{(1)}(t) {}^{0}\tilde{\vec{u}}_{g,k}^{(2)}(t) {}^{0}\tilde{\vec{u}}_{g,k}^{(3)}(t) \dots {}^{0}\tilde{\vec{u}}_{g,k}^{(i)}(t) \dots {}^{0}\tilde{\vec{u}}_{g,k}^{(n)}(t)\}^{\mathrm{T}}$  contains the estimate of the input ground motion relative to each storey  $i = 1, \dots, n$  and for every time instant  $t = t_j, j = 1, \dots, L$ .

1.6 Then, a Statistical Average (SA) process is introduced in the algorithm, since the *n*th ground accelerations  ${}^{0}\tilde{u}_{g,k}^{(i)}(t)$  obtained from previous Eq. (23) should be the same for every storey, at each time instant  $t = t_j$ . Actually, this is not true because of the difference between the assumed initial values  ${}^{0}\theta_{ck}$ ,  ${}^{0}\theta_{m}$  and their true values. So, the SA method can be applied as follows, by computing the average (mean value)  ${}^{0}\bar{\tilde{u}}_{g,k}(t_{j})$  between the *n* different  ${}^{0}\tilde{\tilde{u}}_{g,k}^{(i)}(t_{j})$ , at each time instant  $t = t_{j}$ :

$${}^{0}\bar{\ddot{u}}_{g,k}(t_j) = \frac{1}{n} \sum_{i=1}^{n} {}^{0}\tilde{\ddot{u}}_{g,k}^{(i)}(t_j), \quad j = 1, \dots, L \quad (24)$$

1.7 After the computation of the average ground motion  ${}^{0}\bar{u}_{g,k}(t)$ , a modified version of vector  $\mathbf{P}_{ck}(t)$  may be reconstructed, by using the following modification procedure:

$${}^{0}\hat{\mathbf{P}}_{ck}(t) = {}^{0}\tilde{\mathbf{N}}_{ck}\dot{\mathbf{x}}(t) - \mathbf{L}_{ck} {}^{0}\mathbf{Mr} {}^{0}\ddot{\ddot{u}}_{g,k}(t)$$
(25)

where superscript  $\land$  means modified value.

1.8 Finally, an improved estimation of the  $\boldsymbol{\theta}_{ck}$  parameters can be computed by the use of the modified  ${}^{0}\hat{\mathbf{P}}_{ck}$  vector, leading to estimates of damping and stiffness parameters at iteration 1 as follows:

$${}^{1}\boldsymbol{\theta}_{ck} = {}^{0}\mathbf{H}_{ck}^{\dagger} {}^{0}\tilde{\mathbf{P}}_{ck} = \left( \left[ {}^{0}\mathbf{H}_{ck}^{\mathrm{T}} {}^{0}\mathbf{H}_{ck} \right]^{-1} {}^{0}\mathbf{H}_{ck}^{\mathrm{T}} \right) {}^{0}\tilde{\mathbf{P}}_{ck}$$
(26)

1.9 On the latter LS formulation, a parameter projection technique is adopted, in order to arrive at strictly-positive estimates of  $({}^{1}\theta_{ck})_{ij}$  parameters, i.e. of  $c_i$  and  $k_i$  terms, when i = j (i.e. for the diagonal terms of the damping and stiffness matrices), without the need of solving constrained LS problems:

$$({}^{1}\boldsymbol{\theta}_{ck})_{ij} = \operatorname{sgn}\left[({}^{1}\boldsymbol{\theta}_{ck})_{ij}\right]({}^{1}\boldsymbol{\theta}_{ck})_{ij}, \quad \forall i, j = 1, \dots, n$$
(27)

where sgn[...] represents the sign function, that is the odd mathematical function which extracts the sign of the real number in its argument.

Then, from the estimate of  ${}^{1}\theta_{ck}$ , updated estimates  ${}^{1}\mathbf{C}$  and  ${}^{1}\mathbf{K}$  of the damping and stiffness matrices can be computed.

# • Second stage (realization of mass parameters):

Similarly to what seen for the first stage, the second stage will be outlined by the following computational steps.

- 2.1 The first iteration of the second identification stage starts from the achieved estimates of  ${}^{1}\theta_{ck}$ ,  ${}^{1}C$  and  ${}^{1}K$ , aiming at the computation of the  ${}^{0}H_{m}$  matrix, through Eq. (18a).
- 2.2 Vector  $\mathbf{P}_m$  can be estimated by using the terms of Step 2.1, from Eq. (17) to  ${}^0\tilde{\mathbf{P}}_m = {}^0\mathbf{H}_m{}^0\boldsymbol{\theta}_m$ , where  ${}^0\boldsymbol{\theta}_m$  was defined in previous Step 1.1.

2.3 Matrix  $\mathbf{G}_m$  can be estimated, through the estimated value of  ${}^0\tilde{\mathbf{P}}_m$ , from the LS problem:

$${}^{0}\tilde{\mathbf{G}}_{m} = {}^{0}\tilde{\mathbf{P}}_{m}\mathbf{x}^{\dagger} = {}^{0}\tilde{\mathbf{P}}_{m}\left(\left[\mathbf{x}^{\mathrm{T}}\mathbf{x}\right]^{-1}\mathbf{x}^{\mathrm{T}}\right)$$
(28)

Notice that from matrix  $\mathbf{G}_m$ , an estimate of state matrix **A** can be directly computed, leading to  ${}^{0}\tilde{\mathbf{A}}$  (see Eq. (9b) for more details).

2.4 Input ground motion vector  ${}^{0}\ddot{\mathbf{u}}_{g,c}(t)$  can be again estimated, by taking into account previous Eq. (8), as:

$${}^{0}\tilde{\ddot{\mathbf{u}}}_{g,c}(t) = -\mathbf{L}_{m}^{\dagger} \Big( {}^{0}\tilde{\mathbf{G}}_{m}\mathbf{x}(t) - \dot{\mathbf{x}}(t) \Big)$$
(29)

where, again, vector  ${}^{0}\tilde{\mathbf{u}}_{g,c}(t)$  holds the input ground motion estimates for each storey i = 1, ..., n and for every time instant  $t = t_j, j = 1, ..., L$ .

2.5 As in previous Step 1.6, the Statistical Average method is applied, at each time instant  $t = t_j$ , to the *n* different  ${}^{0}\tilde{u}_{g,c}^{(i)}(t_j)$  ground motion accelerations:

$${}^{0}\bar{\ddot{u}}_{g,c}(t_{j}) = \frac{1}{n} \sum_{i=1}^{n} {}^{0}\tilde{\ddot{u}}_{g,c}^{(i)}(t_{j}), \quad j = 1, \dots, L \quad (30)$$

2.6 Through the following modification procedure, the modified version of vector  $\mathbf{P}_m(t)$  can be reconstructed as:

$${}^{0}\hat{\mathbf{P}}_{m}(t) = -\dot{\mathbf{x}}(t) - \mathbf{L}_{ck}\mathbf{r} \,{}^{0}\ddot{\vec{u}}_{g,c}(t) \tag{31}$$

2.7 Then, through the use of modified vector  ${}^{0}\hat{\mathbf{P}}_{m}$ , the improved estimation of  $\boldsymbol{\theta}_{m}$  can be computed, by giving rise to iteration 1 for the second identification stage:

$${}^{1}\boldsymbol{\theta}_{m} = {}^{0}\mathbf{H}_{m}^{\dagger} {}^{0}\tilde{\mathbf{P}}_{m} = \left( \left[ {}^{0}\mathbf{H}_{m}^{\mathrm{T}} {}^{0}\mathbf{H}_{m} \right]^{-1} {}^{0}\mathbf{H}_{m}^{\mathrm{T}} \right) \mathbf{P}_{m} \quad (32)$$

2.8 As before, on previous Eq. (32) the parameter projection technique is implemented, towards getting strictly-positive estimations of the  $({}^{1}\theta_{m})_{ij}$  parameters  $\forall i, j = 1, ..., n$  (since mass matrix elements must be all strictly-positive), without the use of constrained LS problems:

$$({}^{1}\boldsymbol{\theta}_{m})_{ij} = \operatorname{sgn}\left[\left({}^{1}\boldsymbol{\theta}_{m}\right)_{ij}\right]\left({}^{1}\boldsymbol{\theta}_{m}\right)_{ij} \quad \forall i, j = 1, \dots, n$$
(33)

Then, from the  ${}^{1}\theta_{m}$  estimate, the updated estimate of mass matrix  ${}^{1}\mathbf{M}$  can be computed.

2.9 In the end,  ${}^{0}\mathbf{M}$ ,  ${}^{0}\mathbf{C}$ ,  ${}^{0}\mathbf{K}$ ,  ${}^{0}\boldsymbol{\theta}_{ck}$  and  ${}^{0}\boldsymbol{\theta}_{m}$  can be replaced by updated parameters  ${}^{1}\mathbf{M}$ ,  ${}^{1}\mathbf{C}$ ,  ${}^{1}\mathbf{K}$ ,  ${}^{1}\boldsymbol{\theta}_{ck}$  and  ${}^{1}\boldsymbol{\theta}_{m}$  in Step 1.1. The two-stage recursive algorithm iterates then from Step 1.2 to Step 2.8, until the following convergence criteria are met:

$$\max \left| \frac{{}^{i} \boldsymbol{\theta}_{ck}(p) - {}^{i-1} \boldsymbol{\theta}_{ck}(p)}{{}^{i} \boldsymbol{\theta}_{ck}(p)} \right| < \varepsilon_{\theta},$$
  
$$\max \left| \frac{{}^{i} \boldsymbol{\theta}_{m}(q) - {}^{i-1} \boldsymbol{\theta}_{m}(q)}{{}^{i} \boldsymbol{\theta}_{m}(q)} \right| < \varepsilon_{\theta}$$
(34a)  
$$\| {}^{i} \tilde{\boldsymbol{\mu}}_{m}^{(r)}(t; ) - {}^{i} \tilde{\boldsymbol{\mu}}_{ck}(t; ) \|$$

$$\max \left\| \frac{u_{g,k}(t_j) - u_{g,k}(t_j)}{i\bar{u}_{g,k}(t_j)} \right\| < \varepsilon_g,$$
  
$$\max \left\| \frac{i\tilde{u}_{g,c}^{(r)}(t_j) - i\bar{u}_{g,c}(t_j)}{i\bar{u}_{g,c}(t_j)} \right\| < \varepsilon_g$$
(34b)

where left superscript *i* indicates the current iteration step (and *i* – 1 the previous one), and *p* and *q* are the *p*th and *q*th elements of vectors  $\theta_{ck}$  and  $\theta_m$ , respectively. Superscript *r* is the *r*th DOF of the estimated ground motion, while *j* indicates the *j*th time instant of interest. Finally,  $\varepsilon_{\theta}$  and  $\varepsilon_g$  represent the selected tolerances, which generally can be chosen to be between  $10^{-4}$  and  $10^{-6}$ . Of course,  $\varepsilon_{\theta}$  refers to a convergence criterion on the structural parameters, while  $\varepsilon_g$  refers to a convergence criterion on the estimated input ground motion.

For the analyses presented here, convergence tolerances have been generally set to  $\varepsilon_{\theta} = 10^{-6}$  and  $\varepsilon_g = 10^{-4}$ . For the ground motion, a lower convergence level is selected, since the estimation error is "diluted" over a considerable greater amount of values, with respect to that for the vector parameters. A stopping criteria is set if after 2000 iterations the convergence criteria have not reached at least  $\varepsilon_{\theta} = 10^{-4}$ and  $\varepsilon_g = 10^{-2}$ . Then, the algorithm definitively stops if after 5000 iterations convergence has not been reached, or if any of the  $\theta_i$  identified parameters go to zero or to infinity.

Once the iterations made on the two stages of the algorithm go to convergence, the final  $\theta_{ck}$  and  $\theta_m$  render realizations of the final mass, damping and stiffness matrices, jointly with realizations of state matrix **A** and output matrix **C**<sub>0</sub>, which can be directly derived from **G**<sub>m</sub>. At a glance, a realization of the State-Space model can be reconstructed from these matrices. Then, the averaged value  $\bar{u}_{g,c}(t)$  obtained from Step 2.5 represents the estimated time history of the ground motion.

#### 2.3 Element-level identification procedure

As previously explained, from the FDCIM, estimates of the modal parameters, of the input ground motion excitation and of the structural (mass, damping and stiffness) matrices at the element-level can be achieved. These matrices, as well as the identified State-Space matrices (**A**, **B**, **C**<sub>0</sub> and **D**), represent a realization of the system under investigation, i.e. a possible combination of matrices which is able to reconstruct the acquired structural responses  $\mathbf{u}(t)$ ,  $\dot{\mathbf{u}}(t)$  and  $\ddot{\mathbf{u}}(t)$ , known the input ground motion excitation  $\ddot{u}_g(t)$ . The latter can be estimated from Step 2.5, as exposed earlier.

Then, from the estimated realization of state and output matrices (**A** and **C**<sub>0</sub>) or, equivalently, of mass, damping and stiffness matrices, the modal properties (natural frequencies, mode shapes and modal damping ratios) can be obtained in a straight-forward manner [11, 13].

The last issue concerns the accurate identification of the *element-level* values, since by this iterative procedure matrices **M**, **C** and **K** are identified correctly, unless for a real positive multiplying parameter  $\delta$ , hereafter called *Fixing Factor*. This means that the proper orders of magnitude between each element are preserved, and only a multiplying parameter  $\delta$  is required to restore the real amount of each mass, damping and stiffness element. In other words, different realizations of matrix **M**, **C** and **K** differ only for an unknown proportionality factor.

The Fixing Factor  $\delta$  can be computed through the knowledge (or at least, an evaluation) of a single structural parameter, allowing then for rescaling the realizations. For instance, the total mass of the building under exam may be known or estimated, or any other single parameter of one of three matrices **M**, **C** and **K** (namely any single parameter among  $m_i$ ,  $c_i$  and  $k_i$ ). Basically, one of these comparison values *P* comes from its knowledge (or estimation) from the real building ( $P_{real}$ ), while the other is taken from the estimated model ( $P_{est}$ ). So, the Fixing Factor can be simply computed as the ratio between them:

$$\delta = \frac{P_{real}}{P_{est}}, \quad \text{being } P = m_{tot}, \ m_i, \ c_i \text{ or } k_i. \tag{35}$$

Then, by multiplying the estimated mass, damping and stiffness matrices by the Fixing Factor, the real value of the element-level identified parameters is achieved. In Sects. 3 and 4, the element-level estimates achieved from the analysed cases will be presented, by adopting always  $P = m_{tot}$  as rescaling parameter (a parameter that may be judged in practical cases, based on several types of information that may be known for the building).

### **3 Identification outcomes**

The effectiveness of the further developed Full Dynamic Compound Inverse Method is now presented through numerical examples, generated from synthetic earthquake-induced response signals. The adopted structure is a three-storey

 Table 1
 Structural properties of the adopted three-storey shear-type frame [8]

Floor	1	2	3
Stiffness $k_i$ (×10 <sup>6</sup> [kN/m])	4728.40	315.23	157.61
Damping $c_i$ (×10 <sup>3</sup> [kN s/m])	4369.00	291.30	145.60
Mass $m_i$ (×10 <sup>3</sup> [kg])	350.25	262.29	175.13

shear-type frame, taken from the work of Chen et al. [8], whose characteristics are reported in Table 1.

The modal characteristics of the building can be computed through classical modal analysis, as reported in Table 2.

The adopted input ground motion excitation is the El Centro 1940 earthquake, as also used by Toki et al. [47], Li and Chen [30], Perry and Koh [38], Lei et al. [27], whose main features are recalled in Table 3.

The time duration is 40 s, sampled at 100 Hz, resulting in a total number of 4000 signal points. The response of the building, in terms of absolute displacements, velocities and accelerations is computed via numerical integration of the equations of motions by Newmark's method, see e.g. [11]. These responses are taken as the only known quantities in the identification process. Of course, if only accelerations results were known (as usual, from the acquisitions out of the monitored structure), specifically-developed integration schemes may be implemented, to obtain also velocities and displacements (e.g. like the ones presented in Concha et al. [12]).

Then, the so-obtained structural responses are taken as input, i.e. as known quantities, for the application of the present FDCIM technique. The performance of the method has been assessed with several different initial values of vectors  ${}^{0}\theta_{ck}$  and  ${}^{0}\theta_m$ , between  $10^{-6}$  and  $10^6$ . No measurable differences have been found among the attempted values, both in terms of achievable estimates ( $\Delta < 0.00001\%$ ) and number of iterations ( $\Delta < 5\%$ ), which turn out to be exactly

 Table 2
 Modal characteristics of the adopted three-storey shear-type frame [8]

Mode	1	2	3
Natural frequencies (Hz)	3.448	7.376	19.16
Modal damping ratios (%)	1.00 %	2.14 %	5.56%
Mode shapes	0.0279	0.0595	0.9955
	0.4315	0.8096	-0.0946
	0.9017	-0.5839	0.0063

the same for all the examined cases. So, the developed identification method can be considered to be insensitive on the adopted initial values, which have been finally chosen as  ${}^{0}\boldsymbol{\theta}_{ck} = \{1 \ 1 \ 1 \ \dots \ 1\}^{\mathrm{T}}$  and  ${}^{0}\boldsymbol{\theta}_{m} = \{1 \ 1 \ 1 \ \dots \ 1\}^{\mathrm{T}}$ .

The same with regards to the estimated Fixing Factor  $\delta$ . Attempts with  $m_{tot}$  and any single parameter  $m_i$ ,  $c_i$  and  $k_i$  have been performed, and confirmed that the achievable estimates of the rescaling factor do not result sensitive with the assumed known structural parameter (the differences among deviations calculated from the achieved estimates, as a function of the adopted  $P_i$ , are less than 5%). Here, only results from the use of  $P = m_{tot}$  are presented. Moreover, different number of points of the signals are used for the present analysis, starting from the initial time instant, namely from a minimum of 50 points (equal to a duration of 0.5 s) to a maximum of 4000 points for the entire signal (corresponding to the total duration of 40 s). Tolerance levels are set as  $\varepsilon_{\theta} = 10^{-6}$  and  $\varepsilon_g = 10^{-4}$ .

Sample, characteristic results are presented in Tables 4 and 5, for the length of 100, 200, 500 and 1000 sampling points, while the deviations of the estimates for all the examined cases are summarized in Figs. 2 and 3. For the natural frequencies, the maximum percentage deviation, calculated as  $(f_{i,id} - f_{i,targ})/f_{i,targ} \cdot 100$ , is 0.0049% (100 points case), while for the modal damping ratios is 0.0115% (50 points case). The mode shapes are always identified in an extremely effective way, since they always lead to unitary Modal Assurance Criterion (MAC) indexes, for every considered case. For the identified element-level parameters, the maximum deviations are: 0.0105% (50 points case) for  $k_i$ ; 0.0179% (100 points case) for  $c_i$ ; 0.0181% (100 points case) for  $m_i$ . It can be seen that, despite the very poor number of adopted points, all estimates result very effective.

As it is visible in Fig. 2 and likely to be expected, deviations go to lower magnitudes (better estimates) for the natural frequencies with respect to the modal damping ratios. In Fig. 3 this applies to the stiffness estimates, which go to lower magnitudes for the second mode. Otherwise, results for  $m_i$ ,  $c_i$  and  $k_i$  appear quite similar, and not much sensitive on the adopted number of points.

Then, considering the acquisitions (4000 points) and concerning the identified input ground motion excitation, the deviations between real and estimated peak ground acceleration (PGA) and between real and estimated Root Mean Square (RMS) value are only 0.0041 and 0.0026 %, respectively (see results later shown in Sect. 4). This confirms the goodness of the achieved estimates of the input ground

 Table 3
 Main characteristics of the adopted earthquake (El Centro, 1940)

Earthquake	Date	Station	Duration (s)	$f_s$ (Hz)	М	Comp.	PGA $(m/s^2)$	RMS (m/s <sup>2</sup> )
El Centro (EC)	18/05/1940	0117	40	100	7.1	NS	3.069	0.5158

Case	Target	Estimated (100 pts.)	$\Delta\%$	Estimated (200 pts.)	$\Delta\%$	Estimated (500 pts.)	$\Delta\%$	Estimated (1000 pts.)	$\Delta\%$
Natural	frequencies (Hz)								
$f_1$	3.448	3.448	0.0049	3.448	0.0018	3.448	0.0018	3.448	0.0019
$f_2$	7.376	7.376	0.0001	7.376	0.0000	7.376	0.0001	7.376	0.0001
$f_3$	19.16	19.16	0.0016	19.16	0.0007	19.16	0.0004	19.16	0.0005
Modal d	lamping ratios (%	)							
$\zeta_1$	1.00%	1.00%	0.0102	1.00%	0.0059	1.00%	0.0037	1.00%	0.0040
ζ2	2.14%	2.14%	0.0048	2.14%	0.0035	2.14%	0.0018	2.14%	0.0020
ζ3	5.56%	5.56%	0.0080	5.56%	0.0049	5.56%	0.0025	5.56%	0.0029
Modal a	ssurance criteria								
MAC <sub>1</sub>	_	1.000	_	1.000	_	1.000	_	1.000	_
MAC <sub>2</sub>	_	1.000	_	1.000	_	1.000	_	1.000	_
MAC <sub>3</sub>	_	1.000	_	1.000	_	1.000	_	1.000	-

**Table 4** Identified natural frequencies  $f_i$ , modal damping ratios  $\zeta_i$  and calculated Modal Assurance Criterion indexes  $MAC_i$  for the three-storey frame, as a function of adopted number of points;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake

**Table 5** Identified stiffness  $k_i$ , damping  $c_i$  and mass  $m_i$  parameters for the three-storey frame as a function of adopted number of points;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake

Case	Target	Estimated (100 pts.)	$\Delta\%$	Estimated (200 pts.)	$\Delta\%$	Estimated (500 pts.)	$\Delta\%$	Estimated (1000 pts.)	$\Delta\%$
Mass p	arameters (×10 <sup>-</sup>	<sup>3</sup> [kg])							
$m_1$	350.25	350.21	0.0107	350.24	0.0041	350.24	0.0037	350.24	0.0040
<i>m</i> <sub>2</sub>	262.29	262.30	0.0021	262.29	0.0010	262.29	0.0008	262.29	0.0009
<i>m</i> <sub>3</sub>	175.13	175.16	0.0181	175.14	0.0066	175.14	0.0061	175.14	0.0066
Dampi	ng parameters (>	×10 <sup>3</sup> [kN s/m])							
<i>c</i> <sub>1</sub>	4369.00	4368.22	0.0179	4368.63	0.0085	4368.74	0.0061	4368.71	0.0067
<i>c</i> <sub>2</sub>	291.30	291.29	0.0050	291.29	0.0046	291.29	0.0022	291.29	0.0023
<i>c</i> <sub>3</sub>	145.60	145.61	0.0063	145.60	0.0013	145.60	0.0020	145.60	0.0022
Stiffnes	ss parameters (×	(10 <sup>6</sup> [kN/m])							
$k_1$	4728.40	4728.02	0.0080	4728.27	0.0028	4728.25	0.0031	4728.24	0.0033
$k_2$	315.23	315.23	0.0006	315.23	0.0004	315.23	0.0000	315.23	0.0000
<i>k</i> <sub>3</sub>	157.61	157.63	0.0105	157.62	0.0040	157.62	0.0035	157.62	0.0038
Numbe	er of iterations (e	$\varepsilon_{ heta} = 10^{-6}, \ \varepsilon_g =$	$= 10^{-4}$ )						
Ν	_	1690	_	768	_	648	_	738	_

motion, which is effectively reconstructed by the present FDCIM technique and proves to be nearly the same as the original one. Some plots concerning the achieved estimates of the input ground motion excitations are also going to be presented in Sect. 4, for further illustration.

# 4 Addition of noise

With the purpose of assessing the goodness of the present FDCIM approach by getting closer to real operating conditions, analyses with noise-corrupted earthquake-induced

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synthetic response signals are also considered. Noisecorrupted cases are generated by adding a zero-mean Gaussian white noise time history to displacement, velocity and acceleration structural response acquisitions. The noise process is added to each single acquisition channel (i.e. at every storey) in terms of percentage of the Root Mean Square (RMS) ratio between the noise time history and the starting response measurement (to be corrupted by noise). Different levels of noise have been considered, starting from 0.5 %, then going to 1, 3, 5% and finally to heavy 10 and 20% noise cases.



Fig. 2 Percentage deviations of identified natural frequencies  $f_i$  and modal damping ratios  $\zeta_i$  with respect to target parameters as a function of adopted number of points; three-storey frame;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake



Fig. 3 Percentage deviations of identified stiffness  $k_i$ , damping  $c_i$  and mass  $m_i$  parameters with respect to target parameters as a function of adopted number of points; three-storey frame;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake

**Table 6** Identified natural frequencies  $f_i$ , modal damping ratios  $\zeta_i$  and calculated Modal Assurance Criterion indexes  $MAC_i$  for the three-storey frame, no-noise and noise-corrupted cases; 4000 points;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake

Case	Target	Estimated (No noise)	$\Delta\%$	Estimated (1% noise)	$\Delta\%$	Estimated (5% noise)	$\Delta\%$	Estimated (20% noise	$\Delta\%$
Natural	frequencies (	(Hz)							
$f_1$	3.448	3.448	0.0025	3.446	0.0684	3.447	0.0180	3.429	0.5544
$f_2$	7.376	7.376	0.0001	7.376	0.0059	7.376	0.0038	7.390	0.1938
$f_3$	19.16	19.16	0.0007	19.16	0.0389	19.13	0.1383	19.12	0.1677
Modal d	lamping ratio	s (%)							
ζ1	1.00%	1.00%	0.0054	1.00%	0.5113	1.03 %	2.8104	0.97%	2.9541
ζ2	2.14%	2.14%	0.0028	2.13%	0.3863	2.20%	2.6490	2.07%	3.5408
ζ3	5.56%	5.56%	0.0043	5.50%	1.0468	5.82%	4.6298	5.89%	5.8787
Modal a	ssurance crit	eria (1)							
MAC <sub>1</sub>	_	1.000	_	1.000	_	1.000	_	1.000	_
MAC <sub>2</sub>	-	1.000	_	1.000	-	1.000	_	1.000	-
MAC <sub>3</sub>	_	1.000	_	1.000	_	1.000	_	1.000	_

Case	Target	Estimated (No noise)	$\Delta\%$	Estimated (1 % noise)	$\Delta\%$	Estimated (5% noise)	$\Delta\%$	Estimated (20% noise)	$\Delta\%$
Mass pa	arameters ( $\times 10^3$	[kg])							
$m_1$	350.25	350.23	0.0054	349.66	0.1672	351.14	0.2547	354.76	1.2874
$m_2$	262.29	262.29	0.0013	262.44	0.0556	261.67	0.2372	256.48	2.2152
<i>m</i> <sub>3</sub>	175.13	175.15	0.0087	175.57	0.2512	174.86	0.1542	176.43	0.7428
Dampin	g parameters (×	$(10^3  [kN  s/m])$							
$c_1$	4369.00	4368.59	0.0094	4315.41	1.2266	4583.19	4.9025	4706.83	7.7325
<i>c</i> <sub>2</sub>	291.30	291.29	0.0030	289.85	0.4990	299.53	2.8245	286.69	1.5828
<i>c</i> <sub>3</sub>	145.60	145.60	0.0029	145.43	0.1143	148.60	2.0599	134.90	7.3515
Stiffnes	s parameters (×	10 <sup>6</sup> [kN/m])							
$k_1$	4728.40	4728.20	0.0041	4723.82	0.0968	4727.86	0.0114	4780.55	1.1029
$k_2$	315.23	315.23	0.0002	315.30	0.0199	314.59	0.2034	311.13	1.2999
<i>k</i> <sub>3</sub>	157.61	157.62	0.0051	157.84	0.1453	157.23	0.2423	155.93	1.0639
Estimat	ed input ground	motion-peak gr	ound accel	eration/root mean squ	are ([m/s	2])			
PGA	3.069	3.069	0.0041	3.062	1.8276	3.138	4.3708	3.642	21.123
RMS	0.5158	0.5158	0.0026	0.5162	0.0752	0.5282	2.4119	0.7055	36.780

**Table 7** Identified stiffness  $k_i$ , damping  $c_i$  and mass  $m_i$  parameters for the three-storey frame, no-noise and noise-corrupted cases; 4000 points;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake



Fig. 4 Percentage deviations of identified natural frequencies  $f_i$  and modal damping ratios  $\zeta_i$  with respect to target parameters, no-noise and noise-corrupted cases; three-storey frame, 4000 points;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake

As before, sample, characteristic results are shown in Tables 6 and 7, for no-noise, 1, 5 and 20% noise-corrupted cases, while all deviations for the examined cases are reported in Figs. 4 and 5. The maximum percentage deviations are 1.1939% (10% noise case) and 5.8787% (20% noise case) for the natural frequencies and modal damping ratios, respectively. Again, mode shape estimates turn out to be very effective, by showing always unitary MAC values, for all the modes and examined cases. As concerning the estimated element-level parameters, the maximum deviations are 2.0262% (10% noise case), 7.7325% (20% noise case) and 3.2652% (10% noise case) for  $k_i$ ,  $c_i$  and  $m_i$ , respectively. So, although the heavy noise-corrupted adopted cases

effective. By observing Figs. 4 and 5, it is visible that natural

(up to 20% of noise), system identification keeps very much

frequencies are more accurate than modal damping ratios, even if characterized by greater dispersions. Modal damping ratios, as well as damping parameters  $c_i$ , seem to be less sensitive to the amount of added noise. Element-level  $k_i$  and  $m_i$  are slightly more accurate with respect to the damping parameters, even though they are characterized by higher dispersions.

Finally, the input ground motion excitation is still identified effectively, despite for the added noise. Fig. 6 represents the trend of the deviations between real and estimated PGA



Fig. 5 Percentage deviations of identified stiffness  $k_i$ , damping  $c_i$  and mass  $m_i$  parameters with respect to target parameters, no-noise and noisecorrupted cases; three-storey frame, 4000 points;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake



Fig. 6 Percentage deviations of the Peak Ground Acceleration (PGA) and Root Mean Square (RMS) values of the identified input ground motion excitation with respect to target excitation, no-noise and noise-

and RMS, respectively. The estimates are affected by slight errors up to a 10 % noise level; only for the 20 % noise case the deviations grow up to 21.123 and 36.780 %, for PGA and RMS, respectively. Also, Fig. 7 represents the noisecorrupted acceleration responses adopted for the analysis, at different levels of added noise, with the comparison between estimated and target input ground motion time histories  $\ddot{u}_g$ . It is clear that  $\ddot{u}_g$  can be effectively estimated until the level of 10 % noise; only with a higher noise level of 20% its reconstruction starts to get into troubles.

Nevertheless, a further careful observation of the estimates of the input ground motion suggests that the acquisitions are affected by a distributed noise, which can be largely removed by post-filtering such obtained data. In fact, as it can be seen in Fig. 8, data filtering can greatly improve the estimates. The application of a Chebyshev Type II lowpass filter to the identification data brings to: PGA =  $3.0472 \text{ m/s}^2$  ( $\Delta = 0.7103\%$ ) and RMS =  $0.5077 \text{ m/s}^2$  ( $\Delta = 1.5704\%$ ) for the 10% noise case; PGA =  $3.0438 \text{ m/s}^2$  ( $\Delta = 0.82113\%$ ) and RMS =  $0.5347 \text{ m/s}^2$  ( $\Delta = 3.6642\%$ ) for the 20% noise case. That again confirms the goodness of the achieved results, in getting closer to real scenarios.



corrupted cases; three-storey frame, 4000 points;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake

#### 5 Identification of a realistic ten-storey frame

In the present section, the effectiveness of the developed Full Dynamic Compound Inverse Method is further demonstrated through a rather realistic structure from the literature. The building is a ten-storey RC frame taken from the work of Villaverde and Koyama [50], whose characteristics are reported in Table 8.

This building is characterized by close modes, with all natural frequencies laying in about a 5 Hz interval. The modal characteristics of the building are reported in Table 9. Mode shapes are omitted here for brevity.

As before, the El Centro 1940 earthquake is taken as benchmark ground motion base-excitation. Again, no variations were found in terms of achievable estimates ( $\Delta$ < 0.00001%) and number of iterations ( $\Delta$  < 5%), as a function of the adopted initial values of  ${}^{0}\theta_{ck}$  and  ${}^{0}\theta_{m}$  vectors, ranging from 10<sup>-6</sup> to 10<sup>6</sup>. Tolerances are set to  $\varepsilon_{\theta} = 10^{-6}$ and  $\varepsilon_{g} = 10^{-4}$ , with a Fixing Factor  $\delta$  set on  $P = m_{tot}$ .

Figures 9 and 10 display the deviations of the achievable estimates, as a function of the adopted number of points, starting from a minimum of 100 points (duration



Fig. 7 Acceleration structural responses  $\ddot{u}_i(t)$ ; estimated and target input ground motion excitation  $\ddot{u}_g(t)$ ; no-noise and noise-corrupted cases; three-storey frame, 4000 points;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake



Fig. 8 Estimated (with applied filtering) and target input ground motion excitation  $\ddot{u}_g(t)$ ; 10 and 20% noise-corrupted cases; three-storey frame, 4000 points;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake

Table 8 Structural properties of the adopted ten-storey shear-type frame [50]

Floor	1	2	3	4	5	6	7	8	9	10
Stiffness $k_i$ (×10 <sup>6</sup> [kN/m])	62.47	59.26	56.14	53.02	49.91	46.79	43.7	40.55	37.43	34.31
Damping $c_i$ (×10 <sup>4</sup> [kN s/m])	31.90	28.94	27.42	25.90	24.38	22.85	21.33	19.81	18.28	16.76
Mass $m_i$ (×10 <sup>3</sup> [kg])	179	170	161	152	143	134	125	116	107	98

 Table 9
 Modal characteristics of the adopted ten-storey shear-type frame [50]

		-		-	-					
Mode	1	2	3	4	5	6	7	8	9	10
Natural frequencies (Hz)	0.500	1.326	2.151	2.934	3.653	4.292	4.836	5.272	5.590	5.787
Modal damping ratios (%)	0.77~%	2.05~%	3.33 %	4.53 %	5.64 %	6.61 %	7.44 %	8.10 %	8.59%	8.88%

of 1 s) to a maximum of 4000 points (entire response signal).

The recorded maximum deviation is 0.000615 % (4000 points case) for the natural frequencies, while is 0.0049 % (100 points case) for the modal damping ratios. The mode shapes always lead to unitary MAC indexes, for every considered case. With regards to the identified element-level parameters, the maximum deviations show to be: 0.0039%(4000 points case) for  $k_i$ , 0.0295 % (100 points case) for  $c_i$  and 0.0051 % (4000 points case) for  $m_i$ . The number of required iterations is 3331, 329, 138, 281, 415 and 472 for the 100, 200, 500, 1000, 2000 and 4000 points cases, respectively. Despite the increase of the parameters to be estimated, jointly with the increase of complexity of the adopted structure, the estimates show to be very effective. Even with very short structural recordings (i.e. with a very poor number of adopted points), the algorithm reveals to be able to effectively converge, by rendering even better results than for the simpler 3-storey frame. In a sense, it looks like that, with the present 10-dofs system, namely with 10 available response signals as input source to be fed to the FDCIM algorithm, much information is available for the identification process and, even though the identification problem becomes more challenging, the inverse problem arrives at even better results.

As it can be seen in Figs. 9 and 10, the most effective estimates can be traced for the natural frequencies and for the stiffness parameters. In particular, the best estimates are located between the 2nd and the 4th modes of vibration for the natural frequencies and for the modal damping ratios (for the latter, very good estimates can be found also on the last modes of vibration). In parallel, the best estimates for  $m_i$ ,  $c_i$  and  $k_i$  can be found between the 4th and the 6th modes of vibration.

Further, some attempts with noise-corrupted signals are considered again for validation purposes. As seen in Sect. 4, the noise process is added to the response of each storey in terms of percentage of the Root Mean Square (RMS) ratio between the noise time history and the starting response measurement (to be corrupted by noise). The considered noise levels are 0.5, 1, 3, 5, 10, and 20%.

All the achieved deviations for the examined cases are reported in Figs. 11 and 12. The maximum deviations are 0.6657 and 5.7659% (both for the 20% noise case) for the natural frequencies and for the modal damping ratios, respectively. Again, mode shapes return always unitary MAC values, for all the modes and examined cases. As concerning the estimated element-level parameters, the maximum deviations are 3.9316, 6.5641 and 5.1500% (both for the 20%)



Fig. 9 Percentage deviations of identified natural frequencies  $f_i$  and modal damping ratios  $\zeta_i$  with respect to target parameters as a function of adopted number of points; ten-storey frame;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake



Fig. 10 Percentage deviations of identified stiffness  $k_i$ , damping  $c_i$  and mass  $m_i$  parameters with respect to target parameters as a function of adopted number of points; ten-storey frame;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake



Fig. 11 Percentage deviations of identified natural frequencies  $f_i$  and modal damping ratios  $\zeta_i$  with respect to target parameters, no-noise and noise-corrupted cases; ten-storey frame, 4000 points;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake

noise case) for  $k_i$ ,  $c_i$  and  $m_i$ , respectively. So, despite the use of heavy noise-corrupted cases, the identification keeps to be very effective.

As before, from Figs. 11 and 12 that it can be noted the natural frequencies are more accurate with respect to the modal damping ratios. The best estimates are located between the 2nd and the 4th modes of vibration, both for the natural frequencies and for the modal damping ratios. Then, in general, the element-level estimates  $k_i$  and  $m_i$  are slightly more accurate with respect to the  $c_i$  evaluations. More in detail, the best



Fig. 12 Percentage deviations of identified stiffness  $k_i$ , damping  $c_i$  and mass  $m_i$  parameters with respect to target parameters, no-noise and noise-corrupted cases; ten-storey frame, 4000 points;  $m_{tot}$  Fixing Factor parameter; general viscous damping; El Centro earthquake

estimates for  $m_i$ ,  $c_i$  and  $k_i$  can be found again between the 4th and the 6th modes of vibration.

Finally, the input ground motion excitation is still identified effectively, both with or without the added noise. For the full acquisition (4000 points, without noise), the deviations between real and estimated Peak Ground Acceleration (PGA) and between real and estimated Root Mean Square (RMS) value are only  $3.61 \times 10^{-4}$  and  $7.29 \times 10^{-5}$ %, respectively. Again, the deviations of those parameters are very limited until a 10% noise level. Only for the 20% noise case the deviations increase until 10.663 and 24.674%, for PGA and RMS values, respectively.

Then, by post-filtering the obtained input ground motions, it is possible to improve the estimates, leading to PGA =  $3.0637 \text{ m/s}^2$  ( $\Delta = 0.18547 \%$ ) and RMS =  $0.5187 \text{ m/s}^2$ ( $\Delta = 0.57269 \%$ ) for the 20% noise case. Those results demonstrate again the effectiveness of the developed algorithm, also by working with the present much challenging ten-storey case.

# **6** Conclusions

The present work has demonstrated the effectiveness of an innovative Time-Domain output-only modal dynamic identification, input estimation and element-level system identification technique, towards the simultaneous identification of modal characteristics, input excitation time history and structural parameters at the element-level by adopting earthquake-induced structural response data only. Specifically, no information on the structural system is required, for instance for the mass matrix, as typical from previous attempts in the literature, if not for a single structural parameter (for instance the total mass of the building) if a Fixing Factor shall be set to convert realizations into element-level estimates. This technique has been named *Full*  Dynamic Compound Inverse Method (FDCIM), in order of continuity with the earlier definition and formulation in [8].

In the end, main outcomes and results on the development and application of the present FDCIM approach, with specific reference to the seismic engineering context, may be summarized as follows:

- The theoretical framework of the FDCIM technique has been presented, starting from the underlying mathematical model and the basic governing equations, by aiming at a complete and exhaustive description of the method.
- The developed FDCIM approach releases strong assumptions of earlier known element-level techniques, towards the simultaneous identification of modal parameters, input ground motion excitation and mass, damping and stiffness matrices at the element-level through an innovative two-stage iterative algorithm.
- A Statistical Average technique, a modification process and a parameter projection strategy have been developed within the iterative algorithm and adopted to provide correct constraints and to achieve stronger and faster convergence on the identification estimates.
- The proposed FDCIM method works in a completelydeterministic way, it is fully developed in State-Space form and it does not require continuous- to discrete-time transformations. Further, it does not depend on computational initialization conditions or on the state estimation, as it has been proven with the performed validation analyses.
- Accurate estimates of all structural parameters have been provided, such as element-level mass, damping and stiffness matrices, by knowing only a single component of one of these matrices, or even by relying on an estimate of a global parameter of the structure, like for instance the total mass of the building.

- The algorithm has been validated first with simulated data from a three-storey shear-type frame, by showing the achieved estimates for modal parameters, input ground motion excitation and element-level structural parameters, as a function of the adopted number of signal points, starting from the initial time instant. Effectiveness has been proven by relying on a minimum of 50 points to a maximum of 4000 points, i.e. on the entire signal.
- The FDCIM approach is then applied to noise-corrupted earthquake-induced synthetic response signals, still on the three-storey frame, by considering various levels of added noise. The effective estimates coming out from the analyses further validated and confirmed the goodness of the present FDCIM identification approach, by aiming at getting closer to real conditions.
- Finally, additional analyses with a more complex structural case, i.e. a realistic ten-storey frame from the literature have been performed, as a function of the adopted number of points and by using again noisecorrupted signals. These further and fully confirmed the results achieved from the present inverse analysis identification method.

Further on-going research will concern attempts with even more complex structures and several seismic ground motion excitations (synthetic signals), jointly with the later adoption of real earthquake-induced response data. Then, integration or support by the FDCIM to other common output-only methods working with State-Space parametric Time Domain frameworks will be addressed. Additional theoretical investigations on the FDCIM approach and on its possible further improvement will be the target of future work.

Acknowledgments The Authors would like to acknowledge public research support from "Fondi di Ricerca d'Ateneo ex 60%" and a ministerial doctoral grant and funds at the ISA Doctoral School, University of Bergamo, Department of Engineering and Applied Sciences (Dalmine).

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