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A data‑driven approach for linear and nonlinear damage detection using variational mode decomposition and GARCH model

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Abstract

In this article, an original data-driven approach is proposed to detect both linear and nonlinear damage in structures using output-only responses. The method deploys variational mode decomposition (VMD) and generalized autoregressive conditional heteroscedasticity (GARCH) model for signal processing and feature extraction. To this end, VMD decomposes the response signals that are frst decomposed to intrinsic mode functions (IMFs), and then, GARCH model is utilized to represent the statistics of IMFs. The model coefficients' of IMFs construct the primary feature vector. Kernel-based principal component analysis (PCA) and linear discriminant analysis (LDA) are utilized to reduce the redundancy from the primary features by mapping them to the new feature space. The informative features are then fed separately into three supervised classifers: support vector machine (SVM), *k*-nearest neighbor (kNN), and fne tree. The performance of the proposed method is evaluated on two experimental scaled models in terms of linear and nonlinear damage assessment. Kurtosis and ARCH tests proved the compatibility of GARCH model. The results demonstrate that the proposed technique reaches the accuracy of 100% and 98.82% in classifying linear and nonlinear damage, respectively. Also, its accuracy is higher than 80% in the presence of noise with a signal-to-noise ratio (SNR) of more than 10 dB.

Keywords Data-driven SHM · Variational mode decomposition (VMD) · GARCH model · Linear and nonlinear damage

1 Introduction

Today's current structural engineering industry requires consideration to be directed towards structural health monitoring (SHM) and optimizing safety. With forecasts of increasing worlds' population, structural infrastructure shall be subject to increased loading and deformation. To decrease

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the effects and consequences of structural deterioration, SHM processes are required more frequently, with high levels of accuracy necessary to achieve asset preservation. Hence, there has been a surge in interest surrounding SHM and the development of automated defect evaluation systems in an attempt to maintain existing structural networks and allow for asset expansion.

Concerning structural behavior, damage leads to deviations in the structure's dynamic characteristics and is considered a reliable indication of anomaly diagnosis. Also, it might cause a system with typically linear behavior to demonstrate nonlinear responses, including cracking, impacts and rattling, delamination, stick or slip, rub, or deformation in connections [[1,](#page-16-0) [2](#page-16-1)]. Nonlinear behavior is supposed to be unpredictable and more sophisticated compared to the linear one. As a case in point, it has been proven through experimental investigation that natural frequencies could rise instead of decrease in breathing phenomena [[3](#page-16-2)]. This reaction originates from the fact that the crack conversely opens and closes in the experimental test. Subsequently, the detection of nonlinear anomalies is considered more challenging compared to linear damage [[4](#page-16-3)].

Over the decades, researchers have proposed several techniques in terms of anomaly identifcation. Generally speaking, such methods are divided into physics-based (or model-based) and data-driven approaches [\[5](#page-16-4)]. In the physics-based, anomalies are tracked utilizing monitoring variations within the simulated responses from the structural numerical model [[6\]](#page-16-5). This model is a detailed mathematical abstraction linking a studied system's input and output variables employing known or pre-sumed properties [\[7\]](#page-16-6). Post analysis is demanded for determining damage location and qualifcation. Finite-element methods (FEMs), boundary element methods (BEMs), and spectral fnite-element methods (SFEMs) are some of the techniques used in this regard. However, FEMs are considered the systematic method compared to the others due to their compliance in modeling complicated structures [\[8](#page-16-7)]. In the occurrence of damage, particular parameters of the simulated models are updated according to response measurements. Optimization algorithms are typically used to minimize variations between experimental and numerical responses by comparing mechanical characteristics of stifness, damping, or mass [\[6\]](#page-16-5).

Despite the broad potential of physics-based approaches in damage assessment, especially for the evolution of complex systems such as multi-stories buildings and multi-span bridges, they have some limitations. For example, exact modeling of a structure entails sufficient information regarding diferent components of a monitored system, such as loading states, boundary conditions, material properties, and precise coordinates of members. Moreover, optimization solutions commonly experience numerical instability as well as ill-conditions dilemma [[9\]](#page-16-8). The performance of such optimization techniques substantially degrades proportionally to the number of variables in the problem.

On the other side, data-driven SHM provides bottom–up solutions founded on tracking changes within the output signals appropriate for complex systems where the knowledge about geometries, properties, and initial conditions is limited [\[5](#page-16-4)]. Any sudden changes in the output signals are observed and analyzed through signal processing tools and pattern recognition procedures to determine probable damage. Independence for having an initial model and prior knowledge causes data-driven SHM to be a faster technique and an economical and practical solution for online SHM. Signal processing techniques synthesize, modify and analyze the recorded responses, and highlight diferent features in time, frequency, and frequency domains. Machine Learning algorithms are typically employed to identify and interpret features extracted from signals and recognize generated patterns in conjunction with such methods. Machine learning includes clustering, regression, neural networks, ensemble learning, deep learning, Bayesian methods, instance-based, decision trees, and dimensionality reduction [\[10](#page-16-9)].

Data-driven methods are helpful compared to physicsbased techniques when [[11\]](#page-16-10), frst, the structure's physical characteristics are unavailable or challenging to be modeled.

Second, there are an adequate number of sensors installed for capturing the structure's responses. Third, the computational operations are costly in the SHM project; in addition, multi-physics models consist of more physical processes in a system (e.g., thermal interactions, water precipitation, and magnetostatic and chemical reactions) may not seem efficient for utilizing a large amount of sensor data. The accuracy of physics-based depends on the response measurements; the best performance is achieved in an environment with the slightest amount of noise. In real-world structures and especially for in-servicing conditions, however, the amount of noise is considerable. As such, data-driven damage identifcations deploying actual responses have revealed preferable adaptability and thereby turned into an inspiring solution in the realm of SHM [\[10](#page-16-9)].

1.1 Need for research

Although nonlinear damage has been studied before and practical solutions are proposed in this realm, most focus on damage identifcation as the frst level based on Rytter's classifcation levels in SHM [[12\]](#page-16-11). Hence, limited research has been conducted to reach higher levels (e.g., damage localization and classifcation). This study attempts to address nonlinear damage detection in building structures through a robust data-driven approach. Adverse conditions such as environmental and operational efects in recording responses and analyzing signals are the other crucial points that should be considered. These issues become more though in the case of buildings where the story correlations can afect the structural responses. Therefore, proposing a robust model with appropriate precision in identifying diferent kinds of linear and nonlinear anomalies considering these issues leads to a practical approach in assessing real-world structures under adverse conditions.

Accordingly, the rest of the paper is organized as follows. In Sect. [2,](#page-1-0) related works are discussed, and gaps are highlighted once again. Case studies are presented in detail in Sect. [3](#page-1-1). Section [4](#page-2-0) provides the details of the proposed data-driven approach. Experimental results and discussion are given in Sect. [5.](#page-4-0) Finally, Sect. [6](#page-11-0) concludes the work and suggests future directions.

2 Background

Signal processing techniques play a fundamental role in data-driven SHM for analysis responses in time, frequency, or time–frequency domains. Fourier spectra, spectrum analysis, diference frequency analysis, and the high-frequency resonance technique are appropriate for damage identifcation, especially for gear faults and roller bearings $[13]$ $[13]$ $[13]$. Wavelets proved the efficiency for damage and deterioration detection in building structures based on a stochastic approach $[14]$ $[14]$ $[14]$. Fourier transform (FT) and fast Fourier transform (FFT) are considered the main concepts for anomaly detection. A time-series model is a promising tool for simulating and predicting structural signals in the time-domain. Since this method is based on a partial structural dynamics model, it can identify even a small number of vibrations [\[15](#page-16-14)]. In this area, autoregressive (AR) models are investigated for damage and deterioration detection in buildings and bridges [[16–](#page-16-15)[18\]](#page-16-16). Auto-regressive and moving average model (ARMA), as well as generalized autoregressive conditional heteroscedasticity model (GARCH), have proved to be benefcial for nonlinear damage identifcation in building specimens [[19](#page-16-17)]. Transient behaviors caused by damage or adverse environmental conditions can be recognized through a signal's time–frequency form [[20\]](#page-16-18).

In a broad perspective, the real-world signals are linear and stationary and are coupled with noise. Consequently, linear signal processing techniques, such as spectral analysis, are not appropriate in this realm of scope [\[21](#page-16-19)]. Hilbert–Huang transform (HHT), introduced by Huang et al. [\[22\]](#page-16-20), consists of two sequential steps. The first step, called empirical mode decomposition (EMD), separates the complicated initial signal into a determined and commonly limited number of intrinsic mode functions (IMFs) or modes. Each mode is an oscillatory function with time-varying frequencies that reveals the input signals' local features and corresponds to diferent frequencies and a residue [[23,](#page-16-21) [24](#page-16-22)]. The algorithm detects the maxima/minima recursively, assesses the envelopes using the extrema, and removes the average envelopes, which leads to isolating high-frequency bands.[[25](#page-16-23)]. In the next step, the Hilbert transform (HT) includes each IMF's orthogonal pair with 90° diference in the phase [\[26](#page-16-24)]. As a result, each IMF set and the corresponding pair can evaluate instant variations of signal magnitude and frequency concerning time. Compared to wavelet analysis and Fourier transform, EMD benefts from tracing out the IMFs by interpolating between the extremums instead of using any given wavelet basis. Despite the wide usage of EMD in a variety of time–frequency applications such as medical [[27\]](#page-16-25), economics [[28\]](#page-16-26), climate predictions [[29](#page-16-27)], SHM [\[30](#page-17-0), [31\]](#page-17-1), and many other felds, it may dace with some issues like sensitivity to noise and sampling frequency which cause the performance relies on the frequency ratio [[25,](#page-16-23) [31,](#page-17-1) [32](#page-17-2)].

Some modifed algorithms have been developed, including ensemble EMD (EEMD), complete ensemble EMD with adaptive noise (CEEMDAN), and variational mode decomposition (VMD) [\[32](#page-17-2)] to address these limitations. VMD is a relatively new algorithm that decomposes a signal into distinctive amplitude and frequency adjusted sub-signals where together they reproduce the primary input signal [[32\]](#page-17-2). This approach is entirely non-recursive, and the sub-signals are extracted simultaneously; it is proven that VMD outperforms the EMD algorithm in various areas such as signal analysis and damage detection.

Variational mode decomposition has been deployed in the real SHM by some researchers. For instance, Bagheri et al. [[31](#page-17-1)] calculated damping ratios for each extracted modal response obtained from VMD. The mode shape vector was obtained for each decomposed structure mode, which was then practiced for damage identifcation in three specimens, including numerical, experiment, and feld case studies. Xin et al. [\[33\]](#page-17-3) established two damage indices relying on modal parameters obtained from VMD. An experimental and numerical assessment demonstrated the efficiency of the method for nonlinear to fnd the location and severity of nonlinear damage scenarios in the models. Das and Saha [\[34](#page-17-4)] investigated the impact of a heavy noise environment on a new hybrid algorithm using VMD along with frequencydomain decomposition (FDD). It was deducted that the hybrid method could detect damage location accurately for noises above 20%. A novel methodology is illustrated and assessed in the following sections on two experimental specimens with linear and nonlinear damage scenarios.

3 Case studies

In this section, two case studies used in this work are thoroughly explained and discussed.

3.1 Case study 1: linear damage

The frst case study is a three-story metal frame with aluminum columns and foors, investigated in linear damage simulation [[35](#page-17-5)]. A roller at the base supports the specimen and can move horizontally using a hydraulic jack. Piezoelectric single-axis accelerometers instrument each foor. Nine linear damage scenarios are imitated employing stifness reduction of columns and replacement of a 1.2 kg mass. Hence, 50 signals are recorded for each status with a sample rate of 320 Hz. Therefore, 450 signals are acquired for all scenarios, as illustrated in Table [1.](#page-3-0) As depicted, there are nine statuses, including healthy condition (S1) showing the intact structures without any changes in components, two scenarios simulate the operational and environmental effects by changing mass of foors (S2 and S3), and six damage scenarios by changing the stifness of columns (S4–S9).

Additionally, Fig. [1](#page-3-1) presents sample recorded signals in different scenarios, where $y_1(t)$, $y_2(t)$, and $y_3(t)$ represent the recorded data by sensor 1, sensor 2, and sensor 3, respectively. It is evident that the recorded responses for all damage scenarios follow a random pattern, and usage of time-domain data cannot discriminate damage status

Table 1 Damage scenarios of case study 1 [\[17\]](#page-16-28)

Scenario	Records	Description
S1	$0 - 50$	Healthy state
S ₂	$51 - 100$	$Mass = 1.2$ kg at the base
S ₃	$101 - 150$	$Mass = 1.2$ kg on the 1st level
S4	$151 - 200$	87.5% stiffness decrease in column 1BD
S5	$201 - 250$	87.5% stiffness decrease in column 1AD and 1BD
S6	$251 - 300$	87.5% stiffness decrease in column 2BD
S7	$301 - 350$	87.5% stiffness decrease in column 2AD and 2BD
S8	$351 - 400$	87.5% stiffness decrease in column 3BD
\$9	401–450	87.5% stiffness decrease in column 3AD and 3BD

from healthy cases. Thus, there is a need to model output responses through signal processing techniques to fnd suitable features indicating variations in the signals.

3.2 Case study 2: nonlinear damage

This case is the adjusted model of the frst case study and is used for studying the impact of nonlinear damage. The

Fig. 1 Samples from some scenarios in linear damage (scenario 1)

sampling rate is the same as the linear model and is set to 322.58 Hz with 8192 data points for each record. Ten measurements are recorded for each state. Likewise, in the initial specimen, this frame also glides on rails that enable a transmission in one direction with the aid of an actuator. Four accelerometers with a sensitivity of 1000 mV/g are attached on the opposite side of the shaker at the center of the foors; thus, they do not help determine the specimen's torsion models.

To simulate nonlinear damage, a mechanical bumper and a center column are installed onto the frame. This mechanism imitates the breathing crack and will cause nonlinear behaviors in the condition that the installed column hits the bumper, which is placed on the second foor. The adjustable gap between the bumper and the installed column is used for defning diferent degrees of nonlinearity. Hence, the larger the gap is, the smaller the nonlinear behavior becomes. The specimen's outline and the damage scenarios are provided in Fig. [2](#page-4-1) and Table [2,](#page-4-2) respectively. Some recorded nonlinear signals are given in Fig. [3,](#page-5-0) where $y_1(t)$, $y_2(t)$, $y_3(t)$, and $y_4(t)$, respectively, represent the recorded data by sensor 1, sensor 2, sensor 3, and sensor 4. Similar to the previous case, the time-domain

Fig. 2 Three-story bookshelf (adapted from [[14](#page-16-13)])

Table 2 Damage scenarios of case study 2 [\[17,](#page-16-28) [35\]](#page-17-5)

Scenario	Records	Description
S1	$1 - 10$	$Mass = 1.2$ kg on the 1st floor
S ₂	$11 - 20$	$Mass = 1.2$ kg at the base
S ₃	$21 - 30$	$Gap = 0.13$ mm
S4	$31 - 40$	$Gap = 0.10$ mm
S5	41–50	$Gap = 0.05$ mm
S6	$51 - 60$	$Gap = 0.15$ mm
S7	$61 - 70$	$Gap = 0.20$ mm
S8	71–79	Healthy State
S9	80–89	Gap = 0.20 mm and mass = 1.2 kg at the 1st floor
S ₁₀	$90 - 99$	Gap = 0.10 mm and mass = 1.2 kg at the 1st floor
S ₁₁	100–109	$Gap = 0.20$ mm and mass = 1.2 kg at the base
S ₁₂	110–119	50.0% stiffness reduction in column 1BD
S ₁₃	120–129	50.0% stiffness reduction in column $1AD + 1BD$
S ₁₄	130–139	50.0% stiffness reduction in column 3BD
S15	140–149	50.0% stiffness reduction in column 3AD and 3B _D
S ₁₆	150–159	50.0% stiffness reduction in column 2AD and 2BD
S17	160–169	50.0% stiffness reduction in column 2BD

presentation of responses cannot indicate variations due to damage properly.

As noted, two three-story models were presented for linear and nonlinear damage scenarios. Linear damage were simulated by reducing the cross-section area of columns, while nonlinear behavior was considered as hitting a bumper with a mid-column in the second case study. The environmental and operational conditions were also considered by adding a mass to diferent damage scenarios. Story accelerations were recorded for damage identifcation and classifcation, with a novel methodology discussed in the following section.

4 Proposed method

In this work, anomaly detection is performed in three steps. First, VMD decomposes the signal into several sub-signals with separated bandwidths. Second, primary features are extracted using the time-series modeling, and then, the number of features is reduced by KPCA and KDA. Finally, three supervised classifers are separately deployed to discriminate diferent damage states within three specimens. A schematic workflow of the proposed method is depicted in Fig. [4](#page-6-0). In the following, these stages are illustrated thoroughly.

4.1 Signal processing

Herein, the input acceleration signals are decomposed using VMD, so that an input signal *S*(*t*) is broken down into *d* limited-bandwidth IMFs depicted as [[36\]](#page-17-6)

$$
u_k(t) = A_k(t)\cos(\omega_k(t)),
$$
\n(1)

where $A_k(t)$ and $\omega_k(t)$ present the instantaneous amplitude and frequency of $u_k(t)$, respectively. The constructed variational problem is obtained using Hilbert transform as follows:

$$
\min_{u_k, \omega_k} \left\{ \sum_k \left\| \partial_t \left(\sigma(t) + \frac{j u_t(t)}{t} \right) e^{-j \omega_k t} \right\|_2^2 \right\},\tag{2}
$$

such that

$$
S_t = \sum_k u_k(t),\tag{3}
$$

where $\partial(t)$ denotes the partial derivative of *t*, ${u_k(t)} = {u_1(t), ..., u_n(t)}$ and ${\omega_k} = {\omega_1, ..., \omega_n}$ shows the IMFs of signal S_t and their center frequencies of each signal sub-band, respectively. Equation [\(2](#page-4-3)) is presented in a Lagrange function using λ and α as a multiplier operator and penalty factor, respectively, to solve the optimization problem

0 5 10 15 20 25 Time (s)

0 5 10 15 20 25 Time (s)

والمستقطعة

0 5 10 15 20 25 Time (s)

-2 0 2 $y_2(\mu)$

-2 0 2 $y_1 (\mu)$

-2 0 2 y3 (V)

-2 0 2 $y_4(\mu)$

-2 0 2 $y_1 (\mu)$

-2 0 2 $y_2(\mu)$

-2 0 2 y3 (V)

-2 -1 0 1 y4 (V)

 $L(\lbrace u_k \rbrace, \lbrace \omega_k \rbrace, \lambda) = \alpha \sum$

 $u_k(t)$ ‖

 $S(t) - \sum$ *k*

 $+$ ‖

Afterward, Eq. [\(4](#page-5-1)) is transformed into the time–frequency space, and the equivalent extremum solution is solved to obtain the frequency-domain form of the modal element $u_k(t)$ as well as the center frequency ω_k

 $\left(\sigma(t) + \frac{j}{t}u_t(t)\right)$

k

 $u_k(t)$ ⟩ .

 $\lambda(t), S(t) - \sum$

 $\left\| e^{-j\omega_k t} \right\|$

‖

2

2

k ‖ ‖ ‖ ‖ ‖ ∂_t

Fig. 3 Samples from nonlinear scenarios in case study 2

2

2 + ⟨

$$
u_k^{n+1}(\omega) = \frac{f(\omega) - \sum_{i=1, i \neq k}^k u_i(\omega) + 0.5\lambda(\omega)}{1+2}\alpha(\omega - \omega_k)^2, \tag{5}
$$

$$
\omega_k^{n+1} = \frac{\int_0^\infty \omega |u_k(\omega)|^2 d\omega}{\int_0^\infty |u_k(\omega)|^2 d\omega}.
$$
\n(6)

Finally, the alternative direction of multipliers (ADMM) is deployed to optimize the constrained variational model. Subsequently, the initial signal *S*(*t*) is broken down by *d* IMFs as described in the following:

- Initialize the parameters $\{u_k\}$, $\{\omega_k\}$, $\{\lambda^1\}$ and $n \to 0$
- The value of u_k^{n+1} and ω_k^{n+1} is updated according to ([5\)](#page-5-2) and ([6\)](#page-5-3).
- • The λ^{n+1} is updated as stated in

(4)

-2 0 2 y1 (V)

Fig. 4 Workfow of the proposed method

$$
\lambda^{n}(\omega) + \tau \left(f(\omega) - \sum_{k=1}^{n+1} u_{k}(\omega) \right). \tag{7}
$$

• Equation [\(7\)](#page-6-1) is continued till the following criteria are satisfed:

$$
\frac{\sum_{k} \|u_{k}^{n+1} - u_{k}^{n}\|_{2}^{2}}{\|u_{k}^{n}\|_{2}^{2}} < \varepsilon.
$$
 (8)

Proved that the above condition is met, the iteration procedure stops.

Herein, the iteration is stopped; otherwise, it returns to step 2, and *d* IMFs can be extracted [\[31](#page-17-1), [36\]](#page-17-6). In Figs. [5,](#page-7-0) [6,](#page-8-0) [7,](#page-9-0) and [8,](#page-10-0) the IMFs of linear and nonlinear signals are shown. Due to space limitations, we only present the two IMFs.

4.2 Feature extraction

4.2.1 GARCH modeling of IMFs

Generally speaking, a signal can be modeled via ARMA timeseries to evaluate the conditional mean. As an illustration, the ARMA(*p*, *q*) prediction for the conditional mean is formulated as [[37\]](#page-17-7)

$$
S_{t} = \sum_{i=1}^{p} \varphi_{i} S_{t-i} + \sum_{j=1}^{q} \theta_{j} \varepsilon_{t-j} + \varepsilon_{t} + c,
$$
\n(9)

where p denotes the autoregressive model order, φ ^{*i*} presents the autoregressive variable, *q* stands for the moving average model order, θ_j shows the moving average variable,

 ε_t denotes the residual, and *c* is a constant. However, the residual is usually considered to have a mean of zero with constant variance. In some time-series, it is not homoscedastic and has no constant variance [[37\]](#page-17-7). In this case, the time-varying variance is called conditional variance that is described as

$$
\sigma_t^2 = \text{var}_{t-1}(\varepsilon_t) = E_{t-1}\left(\varepsilon_t^2\right). \tag{10}
$$

The GARCH model, established by Bollersl [[38](#page-17-8)], is a dynamic model that addresses the conditional heteroscedasticity or volatility clustering for an innovation process using a weighted combination of past heteroscedasticity functions coupled with the squared residuals of the past. It causes a reduction in the parameters and complexity of the model. A GARCH(*r*, *m*) model for the conditional variance of residual ε_t is formed as

$$
\sigma_t^2 = \beta + \sum_{i=1}^r b_i \sigma_{t-j}^2 + \sum_{j=1}^m a_j \varepsilon_{t-j}^2.
$$
 (11)

In which β , b_i , and a_j are the parameters of the GARCH model. Herein, the following constraints are defned to ensure that the conditional variance is positive:

$$
\beta > 0, b_i \ge 0, a_j \ge 0. \tag{12}
$$

Moreover, the following formula is defned to make the covariance stationary:

$$
\sum_{i=1}^{r} b_i + \sum_{j=1}^{m} a_j < 1. \tag{13}
$$

This paper utilizes the GRACH model to create the conditional variance model for IMFs obtained from VMD. The

0 5 10 15 20 25 Time (s) 0 5 10 15 20 25 Time (s) 0 5 10 15 20 25 Time (s)

Fig. 5 IMF of signals from scenario S1 (healthy) in case study 1

GARCH model showed reliable performance in nonlinear problems, as discussed in [[19\]](#page-16-17). The coefficients of GARCH(r , m), i.e., $\{b_i\}$ and $\{a_j\}$, are considered as features. Hence, *k*th IMF is described by $\left\{ b_1^{(k)}, \ldots, b_r^{(k)}, a_1^{(k)}, \ldots, a_m^{(k)} \right\}$ } . Considering *d* IMFs, the feature vector of signal with $d(r + m)$ features, $\mathbf{f}_{(d(r+m))\times 1}$, is constructed as

$$
\mathbf{r} = \left[b_1^{(1)}, \dots, b_r^{(1)}, a_1^{(1)}, \dots, a_m^{(1)}, b_1^{(d)}, \dots, b_r^{(d)}, a_1^{(d)}, \dots, a_m^{(d)} \right]^\mathrm{T} .
$$
\n(14)

Finally, since each signal is recorded by several sensors, each record is described with $n_f = \sum_{i=1}^{n} d_i (r + m)$ features, where *n* shows the number of sensors and d_i stands for the number of IMFs is used to decompose the signal of the *i*th sensor. Hence, the feature vector of a signal with *n* sensors is given as $\mathbf{f} = [\mathbf{r}_1^T, ..., \mathbf{r}_n^T]^T$. All obtained features are not suitable for classifcation, and feature vectors may sufer from redundant features. Hence, we should utilize feature reduction techniques to remove such features from the feature vector.

4.2.2 Feature reduction

The general concept of kernel-based feature reduction is based on deploying a particular sort of nonlinear mapping function to protrude the initial vector **f** into a high-dimensional feature space as *F*. Regarding the new feature space, the principal components are obtained through the regular principal component analysis (PCA). In other words, the principal nonlinear components in the initial space correspond to the principal components in feature space *F*. Afterward, the kernel functions, including polynomial, radial basis function, and sigmoid, are used to perform the nonlinear mapping in KPCA [[39](#page-17-9)].

Assume nonlinear mapping ϕ ; the initial data space \mathbb{R}^{n_f} is mapped into a new feature space like H as [[40\]](#page-17-10)

$$
\phi: \mathbb{R}^{n_f} \to H
$$

$$
\mathbf{F} \mapsto \phi(\mathbf{F}).
$$
 (15)

For a training sample set $f_1, f_2, ..., f_M$ in \mathbb{R}^{n_f} , where *M* denotes training sample numbers. Subsequently, the covariance matrix is formulated as [\[40\]](#page-17-10)

0.1 Amplitude **Amplitude 0 -0.1 0 5 10 15 20 25 Time (s) 0.05** Amplitude **Amplitude 0 -0.05 0 5 10 15 20 25 Time (s) 0.1** Amplitude **Amplitude 0 -0.1 0 5 10 15 20 25 Time (s)**

Fig. 6 IMF of signals from scenario S9 in case study 1

$$
\mathbf{S}^{\phi} = \frac{1}{M} \sum_{j=1}^{M} \left(\phi(\mathbf{f}_{j}) - \mathbf{m}_{0}^{\phi} \right) \left(\phi(\mathbf{f}_{j}) - \mathbf{m}_{0}^{\phi} \right)^{\mathrm{T}}, \tag{16}
$$

such that

$$
\mathbf{m}_0^{\phi} = \frac{1}{M} \sum_{j=1}^{M} \phi(\mathbf{f}_j). \tag{17}
$$

Since S^{ϕ} it is a bounded, compact, positive, and symmetric matrix, its nonzero values are also positive. For the sake of fnding these nonzero values, Schölkopf et al. [\[41\]](#page-17-11) suggested linearly express every eigenvector of S^{ϕ} by [[40](#page-17-10)]

$$
\beta = \sum_{i=1}^{M} \alpha_i \phi(\mathbf{f}_i). \tag{18}
$$

To compute expansion coefficients, the Gram matrix is formed as $\tilde{R} = \mathbf{Q}^T \mathbf{Q}$, where $\mathbf{Q} = [\phi(\mathbf{F}_1), ..., \phi(\mathbf{F}_M)]$. Consequently, each component **Q** is computed using kernel tricks as [[40\]](#page-17-10)

$$
\tilde{R}_{ij} = \phi(\mathbf{f}_i)^{\mathrm{T}} \phi(\mathbf{f}_j) = (\phi(\mathbf{f}_i) \cdot \phi(\mathbf{f}_j)) = K(\mathbf{f}_i, \mathbf{f}_j).
$$
\n(19)

Accordingly, $\tilde{\mathbf{R}}$ is centralized by $[40]$ $[40]$ $[40]$

$$
\mathbf{R} = \tilde{\mathbf{R}} - \mathbf{1}_M \tilde{\mathbf{R}} - \tilde{\mathbf{R}} \mathbf{1}_M + \mathbf{1}_M \tilde{\mathbf{R}} \mathbf{1}_M, \tag{20}
$$

where

$$
\mathbf{1}_M = \left(\frac{1}{M}\right)_{M \times M}.\tag{21}
$$

Afterward, the orthonormal eigenvectors $\gamma_1, \dots, \gamma_{n_p}$ of **R** are calculated related to n_p the most significant positive eigenvalues, such that $\lambda_1 \geq \lambda_2 \geq \cdots \geq \lambda_{n_p}$. Consequently, the orthonormal eigenvectors β_1 , β_2 , ..., β_{n_p} of corresponding S^{ϕ} are obtained via [\[40\]](#page-17-10)

$$
\beta_j = \frac{1}{\sqrt{\lambda_j}} Q \gamma_j; \quad j = 1, ..., n_p.
$$
\n(22)

After that, the KPCA transformed feature $\mathbf{y} = (y_1, ..., y_{n_p})^{\text{T}}$ vector is obtained by the projection of the mapped sample $\phi(\mathbf{f})$ onto the eigenvector $\beta_1, \beta_2, ..., \beta_{n_p}$ as formulated below [[40](#page-17-10)]

$$
\mathbf{y} = (\beta_1, \beta_2, ..., \beta_{n_p})^{\mathrm{T}} \phi(\mathbf{f}).
$$
 (23)

The training matrix $\mathbf{F} = \begin{bmatrix} \mathbf{f}_1^T; \ \mathbf{f}_2^T; \dots; \mathbf{f}_M^T \end{bmatrix}^T$ with the size of $n_f \times M$ is mapped to the matrix $\mathbf{Y} = \left[\mathbf{y}_1^T \mathbf{y}_2^T \mathbf{y}_3^T \dots \mathbf{y}_M^T \right]^T$ with the size of $n_p \times M$.

Fig. 7 IMF of signals from scenario S7 in case study 2

The aim of linear LDA is as follows [[42\]](#page-17-12):

$$
\mathbf{a}_{\text{opt}} = \arg \max \frac{\mathbf{a}^{\text{T}} \mathbf{S}_{\text{b}} \mathbf{a}}{\mathbf{a}^{\text{T}} \mathbf{S}_{\text{w}} \mathbf{a}},\tag{24}
$$

where S_b and S_w reveal the between-class and within-class scatter matrices, which are obtained as

$$
\mathbf{S}_{\mathrm{b}} = \sum_{k=1}^{n_{\mathrm{c}}} m_k (\boldsymbol{\mu}^{(k)} - \boldsymbol{\mu}) (\boldsymbol{\mu}^{(k)} - \boldsymbol{\mu})^{\mathrm{T}}, \qquad (25)
$$

$$
\mathbf{S}_{w} = \sum_{k=1}^{n_{c}} \left(\sum_{i=1}^{m_{k}} \left(\mathbf{y}_{i}^{(k)} - \boldsymbol{\mu}^{(k)} \right) \left(\mathbf{y}_{i}^{(k)} - \boldsymbol{\mu}^{(k)} \right)^{\mathrm{T}} \right), \qquad (26)
$$

where μ is the global mean, m_k stands for the number of samples in the *k*th class, and $\mu^{(k)}$ denotes the mean of the *k*th class. Afterward, the total scatter matrix is defned as $S_t = S_b + S_w$. The optimum values of **a** correspond to the nonzero eigenvalue of eigenproblem

$$
\mathbf{S}_{\mathbf{b}}\mathbf{a} = \lambda \mathbf{S}_{\mathbf{a}}\mathbf{a}.\tag{27}
$$

A maximum number of $n_c - 1$ eigenvectors are obtained corresponding to nonzero eigenvalues, because the rank of \mathbf{S}_b is limited to $n_c - 1$. Similar mapping [\(15](#page-7-1)) is considered to extend the LDA to the nonlinear case. Hence, S_5^{φ} , S_{w}^{φ} , and S_t^{φ} , respectively, stand for the between-class, within-class, and total scatter matrices in feature space, which are obtained by the following formulation:

$$
\mathbf{S}_{\mathrm{b}}^{\varphi} = \sum_{k=1}^{n_{\mathrm{c}}} m_k \Big(\boldsymbol{\mu}_{\varphi}^{(k)} - \boldsymbol{\mu}_{\varphi} \Big) \Big(\boldsymbol{\mu}_{\varphi}^{(k)} - \boldsymbol{\mu}_{\varphi} \Big)^{\mathrm{T}}, \tag{28}
$$

$$
\mathbf{S}_{w}^{\varphi} = \sum_{k=1}^{n_{c}} \left(\sum_{i=1}^{m_{k}} \left(\varphi \left(\mathbf{y}_{i}^{(k)} \right) - \mu_{\varphi}^{(k)} \right) \left(\varphi \left(\mathbf{y}_{i}^{(k)} \right) - \mu_{\varphi}^{(k)} \right)^{\mathrm{T}} \right) \tag{29}
$$

$$
\mathbf{S}_{t}^{\varphi} = \sum_{i=1}^{M} \left(\varphi(\mathbf{y}_{i}) - \boldsymbol{\mu}_{\varphi} \right) \left(\varphi(\mathbf{y}_{i}) - \boldsymbol{\mu}_{\varphi} \right)^{\mathrm{T}}. \tag{30}
$$

Assume that ν shows the projective function in feature space, and the associated objective function in feature space is defned as

$$
\mathbf{v}_{\rm opt} = \arg \max \frac{\mathbf{v}^{\rm T} \mathbf{S}_{\rm b}^{\varphi} \mathbf{v}}{\mathbf{v}^{\rm T} \mathbf{S}_{\rm f}^{\varphi} \mathbf{v}}.
$$
 (31)

Fig. 8 IMF of signals from scenario S9 in case study 2

This function can be solved by eigenproblem as

$$
S_b^{\varphi} \nu = \lambda S_t^{\varphi} \nu. \tag{32}
$$

And we have

$$
\mathbf{v} = \sum_{i=1}^{M} \alpha_i \varphi(\mathbf{y}_i). \tag{33}
$$

Then, we can define an equivalent problem as:

$$
\alpha_{\text{opt}} = \arg \max \frac{\alpha^{\text{T}} \mathbf{K} \mathbf{W} \mathbf{K} \alpha}{\alpha^{\text{T}} \mathbf{K} \mathbf{K} \alpha},
$$
\n(34)

where $\boldsymbol{\alpha} = [\alpha_1, ..., \alpha_M]^T$. The corresponding eigenproblem is as $\mathbf{K} \mathbf{W} \mathbf{K} \alpha = \lambda \mathbf{K} \mathbf{K} \alpha$, where **K** shows the kernel matrix, i.e., $K_{ij} = \kappa(\mathbf{y}_i, \mathbf{y}_j)$ and W is defined as

$$
W_{ij} = \begin{cases} 1/m_k, & \text{if } \mathbf{y}_i \text{ and } \mathbf{y}_j \text{ both belongs to the } k\text{th class} \\ 0, & \text{otherwise} \end{cases}
$$
 (35)

Each eigenvector α provides a projective function ν in the feature space. Let **y** a data, and then, we have

$$
\langle v, \varphi(y) \rangle = \sum_{i=1}^{m} \alpha_i \langle \varphi(y_i), \varphi(y) \rangle = \sum_{i=1}^{m} \alpha_i \kappa(y_i, y) = \alpha^{\mathrm{T}} \kappa(:,y), \tag{36}
$$

where κ (:, *y*) \doteq $[\kappa$ (*y*₁, *y*), ..., κ (*y_m*, *y*)]^T. Let { α ₁, ..., α _{*n*_c-1}} be the $n_c - 1$ eigenvectors of the eigenproblem concerning nonzero eigenvalues. The transformation matrix $\Theta = [\boldsymbol{\alpha}_1, \dots, \boldsymbol{\alpha}_{n_c-1}]$ is $M \times (n_c - 1)$ a matrix that embeds the data sample **y** into $n_c - 1$ dimensional subspace by

$$
\mathbf{y} \to \mathbf{z} = \Theta^{\mathrm{T}} \kappa(:,\mathbf{y}). \tag{37}
$$

4.3 Classifcation

In the next section, three classifiers are applied to the selected features previously taken and are called predictors. These classifers are prevailing in the realm of Machine Learning, including support vector machine (SVM), fne tree, and *k*-nearest neighbor (*k*NN). SVM is a supervised training algorithm founded on the fact that measurements can be considered two-dimensional space. Each sample

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denotes a data point in the space and can be separated by a line in the case of a two-dimensional problem and a plane in the case of the dimensional system [[43\]](#page-17-13). Regarding *k*NN, despite its simplicity, it is common in terms of suing in large training datasets. It allocates an estimated value to a new sample on the ground of plurality or weighted of the *k*-nearest neighbors in the training set [[44\]](#page-17-14). Classifcation using a decision tree (fne tree) algorithm is very fast and suitable for high-dimensional classifcation problems. A fne tree is a predictive algorithm mapping from samples about an item to conclusions about its target value. In this model, leaves represent the labels, nodes are the features, and branches denote the junction of features, resulting in label classifcation [[45\]](#page-17-15). Subsequently, the prediction using these classifers is compared with each in the following sections.

5 Results and discussion

This section provides the experimental results and relevant discussions. We considered the fvefold cross-validation to assess the performance of the proposed method. To this end, data were randomly partitioned into fve equal-sized groups, and then, the training and testing procedures were repeated for five trials. One group was considered for testing data in each trial, and other groups were used to train the classifer. Finally, results were averaged.

5.1 The efect of the number of IMFs on residual

The number of IMFs has a considerable effect on the number of extracted features and the complexity of the proposed method. Here, we determine the efficient number of IMFs based on the mean absolute of residuals, shown in Fig. [9](#page-11-1) for diferent numbers of IMFs of nonlinear signals. It is observed that residual generally reduces as the number of IMFs increases. However, the slope of reduction varies for diferent sensors. The residuals of sensors 2, 3, and 4 reduce faster than that of sensor 1. As observed, the residual of sensor one does not have a signifcant variation when the number of IMFs are greater than ten. On the other side, the reduction in residuals of sensors 2, 3, and 4 is not notable for the number of IMFs greater than seven. Hence, we consider the ten IMFs for sensor one and seven IMFs for the remaining sensors. Considering 31 IMFs and two features extracted from each IMF, each recording is described with 62 features.

Following the linear case, as observed in Fig. [10,](#page-12-0) the residuals of all sensors dwindle gradually at nearly the same pace. For any fgures over eight IMFs, the residual does not show signifcant deviations. Thus, for the linear signals, the eight values of IMFs are assigned for all sensors of stories. Considering two features for each IMF, each record is denoted through 48 features.

5.2 Classifcation accuracy

To assess the stability of the proposed method and evaluate the effect of features on results, the authors considered four cases as follows:

- SA: no feature reduction method is employed
- SB: only KPCA is used for feature reduction
- SC: only KDA is employed for feature reduction
- SD: at first, KPCA and then KDA is considered for feature reduction.

The number of features in conditions SB, SC, and SD is obtained based on the normalized cumulative summation of eigenvalues (NCSE). When the NCSE reaches higher than 0.95 for the first time, the efficient number of features is

Fig. 9 The residual of VMD of nonlinear data for diferent numbers of IMFs and data length. **a** Length of 512, **b** length of 2048, and **c** length of 8192

Fig. 10 The residual of VMD of linear data for diferent numbers of IMFs and data length. **a** Length of 512, **b** length of 2048, and **c** length of 8192

obtained. Considering $[\lambda_1, \dots, \lambda_{n_f}]$ as sorted eigenvalues in descending order, the NSCE is calculated as follows:

$$
\Lambda_i = \frac{\sum_{k=1}^i \lambda_k}{\sum_{k=1}^{n_f} \lambda_k}; \quad i = 1, \cdots, n_f.
$$

Classifcation accuracy of the proposed method for nonlinear and linear data considering *k*NN, SVM, and fne tree classifers and diferent lengths of signals obtained from sensors are given in Tables [3](#page-12-1) and [4](#page-13-0), respectively.

Concerning the nonlinear case, the minimum and maximum performance are observed in scenario SA and SD with 76.92% and 98.82%, respectively. In all scenarios, fine tree classifiers seem to be more efficient compared to the other classifers. Moreover, *k*NN is the second accurate classifer, and SVM indicates the lowest performance in this case. It is noteworthy that the signal length has the highest impact on the SB and the lowest on SD with the relative variation (Δ_{max}) of 9.09% and 3.69%, respectively.

Regarding the nonlinear case study, the highest and lowest performance, likewise the nonlinear case, were observed in SA and SD with the accuracy of 100.0% and 89.56%, respectively. Similar to the previous case, the fne tree is the suitable classifer in all proposed scenarios. Except for the SB, *k*NN indicates higher performance in comparison with SVM. Scenario SB reveals less sensitivity to the signal length, whereas scenario SA shows the highest sensitivity to the signal variations based on Δ_{max} .

5.3 Confusion matrix

In this part, the classifcation performance for both case studies is provided through confusion matrices. Considering the confusion matrix, we provide the recall or sensitivity (Sens.),

Table 4 Classifcation accuracy of the proposed method for diferent classifers and lengths of linear data

precision (Prec.), total accuracy (Acc.), and F-score, which are defned as

$$
Sens. = \frac{TP}{TP + FN},
$$
\n(38)

$$
\text{Prec.} = \frac{\text{TP}}{\text{TP} + \text{FP}},\tag{39}
$$

$$
Acc. = 100 \frac{TP + TN}{TP + TN + FP + FN},
$$
\n(40)

$$
F\text{-score} = 2\frac{\text{Prec.} \times \text{Sens.}}{\text{Prec.} + \text{Sens.}} = \frac{\text{TP}}{\text{TP} + 0.5(\text{FP} + \text{FN})},\tag{41}
$$

where TP, TN, FP, and FN denote the true positive, true negative, false positive, and false negative, respectively.

The results are given in Table [5](#page-13-1) for the linear damage and the performance metrics are computed for the nine scenarios described earlier. As indicated, the proposed method determines all damage scenarios with no errors. Consequently, this approach expresses the highest performance for discriminating linear damage based on reference to this study.

Regarding the nonlinear case study, 17 separate states of the specimen are predicted through the presented technique, and the results are presented by the confusion matrix as depicted in Table [6.](#page-14-0) As noted, in the majority of the damage states, the prediction accuracy is 100%. Regarding the remaining cases, which are two out of seventeen scenarios, the classifcation performance is 90.0%. Subsequently, the established strategy revealed considerable performance for recognizing nonlinear and linear damage with signifcant precision.

Linear case			Predicted class		Sens	Prec						
		S_1	S ₂	S ₃	S_4	S_5	S_6	S_7	S_8	S_9		
Actual class	S_1	50	$\overline{0}$	Ω	$\overline{0}$	$\overline{0}$	$\mathbf{0}$	$\mathbf{0}$	θ	$\overline{0}$		
	S_2	θ	50	Ω	$\overline{0}$	θ	$\overline{0}$	θ	$\mathbf{0}$	Ω		
	S_3	$\boldsymbol{0}$	Ω	50	$\overline{0}$	$\overline{0}$	$\boldsymbol{0}$	$\overline{0}$	$\mathbf{0}$	Ω		
	S_4	$\boldsymbol{0}$	$\overline{0}$	$\overline{0}$	50	$\overline{0}$	$\overline{0}$	θ	$\mathbf{0}$	$\overline{0}$		
	S_5	$\boldsymbol{0}$	$\overline{0}$	$\overline{0}$	$\overline{0}$	50	$\overline{0}$	θ	$\mathbf{0}$	$\overline{0}$		
	S_6	$\boldsymbol{0}$	$\overline{0}$	$\overline{0}$	$\overline{0}$	$\overline{0}$	50	$\overline{0}$	$\mathbf{0}$	$\overline{0}$		
	S_7	θ	θ	$\overline{0}$	$\overline{0}$	$\overline{0}$	$\overline{0}$	50	θ	$\overline{0}$		
	S_8	θ	Ω	$\overline{0}$	$\overline{0}$	$\overline{0}$	$\mathbf{0}$	$\mathbf{0}$	50	$\overline{0}$		
	S_9	$\boldsymbol{0}$	$\overline{0}$	$\overline{0}$	$\overline{0}$	θ	$\mathbf{0}$	$\overline{0}$	$\mathbf{0}$	50		

Total accuracy: 100%; F-score: 1

Table 5 Confusion matrix for

linear case study

Table 6 Confusion matrix for nonlinear case study

Nonlinear case		Predicted label														Sens	Prec			
		S_1	S_2	S_3	S_4	S_5	S_6	S_7	S_8	S_9	S_{10}	\mathbf{S}_{11}	S_{12}	S_{13}	S_{14}	S_{15}	S_{16}	S_{17}		
Actual label	S_1	10	0	0	$\mathbf{0}$	$\boldsymbol{0}$	$\overline{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\overline{0}$	$\mathbf{0}$	$\mathbf{0}$	$\boldsymbol{0}$	$\mathbf{0}$	$\overline{0}$	θ	$\mathbf{0}$		
	S_2	$\overline{0}$	10	0	$\mathbf{0}$	$\overline{0}$	$\mathbf{0}$	$\overline{0}$	0	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	1	
	S_3	$\overline{0}$	$\mathbf{0}$	10	θ	$\boldsymbol{0}$	$\mathbf{0}$	θ	$\overline{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\boldsymbol{0}$	θ	$\mathbf{0}$	$\mathbf{0}$	$\boldsymbol{0}$	1	
	S_4	0	0	$\boldsymbol{0}$	10	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	0	$\mathbf{0}$	$\bf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\boldsymbol{0}$	$\boldsymbol{0}$	$\mathbf{0}$	$\mathbf{0}$	$\boldsymbol{0}$		
	S_5	$\mathbf{0}$	0	0	θ	10	$\mathbf{0}$	$\boldsymbol{0}$	0	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\boldsymbol{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\boldsymbol{0}$		
	S_6	$\mathbf{0}$	0	$\overline{0}$	θ	$\overline{0}$	10	$\mathbf{0}$	$\overline{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	0	$\mathbf{0}$	$\mathbf{0}$	θ	$\mathbf{0}$	$\mathbf{0}$	1	
	S_7	$\mathbf{0}$	$\mathbf{0}$	$\overline{0}$	$\mathbf{0}$	$\boldsymbol{0}$	$\mathbf{0}$	10	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\boldsymbol{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	1	
	S_8	$\mathbf{0}$	$\mathbf{0}$	0	$\mathbf{0}$	$\boldsymbol{0}$	$\mathbf{0}$	θ	9	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\boldsymbol{0}$	$\boldsymbol{0}$	$\mathbf{0}$	θ	$\boldsymbol{0}$	1	
	S_9	$\overline{0}$	0	$\overline{0}$	θ	$\boldsymbol{0}$	$\mathbf{0}$	$\mathbf{0}$	$\overline{0}$	9	$\mathbf{0}$	1	$\mathbf{0}$	$\mathbf{0}$	θ	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	0.9	
	\mathbf{S}_{10}	$\mathbf{0}$	0	$\overline{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	θ	$\overline{0}$	$\mathbf{0}$	10	$\mathbf{0}$	$\mathbf{0}$	$\boldsymbol{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	1	0.909
	S_{11}	$\mathbf{0}$	0	0	$\mathbf{0}$	$\boldsymbol{0}$	$\mathbf{0}$	θ	0	$\mathbf{0}$	$\mathbf{0}$	10	$\mathbf{0}$	$\boldsymbol{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{1}$	0.909
	S_{12}	$\mathbf{0}$	0	$\overline{0}$	θ	$\boldsymbol{0}$	$\mathbf{0}$	$\mathbf{0}$	0	$\mathbf{0}$		$\mathbf{0}$	9	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	0.9	
	\mathbf{S}_{13}	$\mathbf{0}$	0	$\overline{0}$	θ	$\boldsymbol{0}$	$\mathbf{0}$	θ	$\overline{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	Ω	10	θ	Ω	$\mathbf{0}$	$\mathbf{0}$	1	
	\mathbf{S}_{14}	θ	0	0	$\mathbf{0}$	$\boldsymbol{0}$	$\mathbf{0}$	$\mathbf{0}$	0	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	0	$\mathbf{0}$	10	Ω	θ	$\mathbf{0}$	1	
	S_{15}	$\overline{0}$	0	$\overline{0}$	θ	$\overline{0}$	$\mathbf{0}$	θ	$\overline{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	θ	10	$\mathbf{0}$	$\mathbf{0}$	1	
	S_{16}	$\mathbf{0}$	0	$\overline{0}$	$\mathbf{0}$	θ	$\mathbf{0}$	θ	$\overline{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	$\mathbf{0}$	θ	θ	10	$\mathbf{0}$	1	
	S_{17}	0	0	0	$\mathbf{0}$	$\boldsymbol{0}$	$\boldsymbol{0}$	$\boldsymbol{0}$	0	$\mathbf{0}$	$\boldsymbol{0}$	$\mathbf{0}$	$\mathbf{0}$	$\boldsymbol{0}$	0	$\mathbf{0}$	$\mathbf{0}$	10	1	

Total accuracy: 98.82%; F-score: 0.988

Fig. 11 Efect of noise on classifcation performance

5.4 The efect of noise

The various intensity of noises is applied to the responses on the grounds of signal–noise ratio (SNR) to assess the stability of the proposed method against noise, as depicted in Fig. $11.$ As observed, the proposed method is efficient even in environments contaminated with severe noise $(SNR=1)$. Furthermore, the established approach can maintain its performance against noise where it shows insignifcant variations in the case of SNR of 20 and 15.

6 GARCH efect assessment

In this section, two tests are applied to demonstrate the compatibility of the GARCH model [[46](#page-17-16)]. Thus, Kurtosis and ARCH tests are provided in the following sections.

6.1 Kurtosis test

GARCH model is appropriate for those signals that have the shape of heavy tails. Therefore, the Kurtosis test is utilized to fnd out that signals have heavy tails or not. The Kurtosis for a distribution (s) is formulated as follows $[46]$:

$$
K(s) = \frac{E(s - \mu)^4}{\sigma^4},\tag{42}
$$

where μ and σ denote the mean and standard deviation of distribution *s*, respectively, and *E*(*s*) stands for the expected value of *s*. For Gaussian distribution, the Kurtosis value of three and higher values shows that the distribution of coeffcients has a heavier tail than the Gaussian distribution. This paper applies this test to the IMFs for each sensor, and the average results for minimum and maximum values of subbands are presented in Table [7.](#page-15-0) Regarding the results, it can be seen that the maximum values are higher than 3, which proves that the IMFs do not have Gaussian distribution.

Table 7 Kurtosis test for IMFs

Sensor	Statistic		IMF number											
			\mathcal{L}	3	$\overline{4}$	5	6	7	8	9	10			
	Min	2.81	2.75	2.75	2.80	2.87	2.87	2.99	2.78	2.72	2.88			
	Max	3.36	3.28	3.41	3.89	3.54	3.87	4.27	4.07	3.47	4.74			
2	Min	2.88	2.78	2.92	2.57	2.67	2.85	2.54	2.73	2.89	2.75			
	Max	4.61	6.93	8.18	4.92	6.50	4.10	3.63	3.91	3.99	6.70			
3	Min	2.77	2.84	2.82	2.88	2.92	2.94	2.89	2.86	2.77	2.94			
	Max	24.57	23.77	8.97	4.76	4.62	4.38	3.91	6.02	8.85	4.83			
$\overline{4}$	Min	2.88	2.98	2.91	2.92	2.81	2.93	2.93	2.93	2.87	2.97			
	Max	41.64	8.66	4.78	4.68	4.23	3.93	3.61	4.08	4.277	4.24			

Table 8 ARCH test for IMFs

6.2 ARCH test

Based on the hypothesis provided in [[47\]](#page-17-17), the ARCH test is deployed to see the existence of ARCH/GARCH impact in the IMFs of each sensor. In this reference, the Lagrange multiplier test is presented based on regression. Subsequently, the test statistic is asymptotically Chi-square distributed has q degrees of freedom [\[46](#page-17-16)].

Thus, in this part, the ARCH test is applied to the IMFs for diferent sub-bands, and the average results for signals are shown in Table [8](#page-15-1). In this table, *h* stands for the Boolean decision variable, where 1 shows the rejection of the null hypothesis, which depicts that no GARCH effect exists. p Value is the signifcance level at which the test rejects the null hypothesis. GARCHstat and CriticalValue are the ARCH test static and critical values of the Chi-square distribution, respectively. Based on this test, if GARCHstat is less than the critical value, no GARCH efect exists. In this study, the signifcance level is set to 0.05, frequently deployed in [\[48\]](#page-17-18). Notably, these results are the average of all signals; for example, the average value of h for the fourth IMF of the frst sensor is 0.74, which demonstrates that 74% of the signals have the GARCH efect. Thus, in general, the results of the table prove the existence of the GARCH efect in most cases.

7 Conclusion

In this paper, a novel methodology was proposed with the potential to identify and classify linear and nonlinear damage in building structures. Here, the VMD was applied to address the variational conditioning in the input signals and the GARCH model used for modeling the decomposed signals. Afterward, the IMFs were deployed as the features of input signals. It was revealed

that using all IMFs led to an increase in residuals. Thus, KPCA and KDA are applied to the extracted features, respectively, to find the optimum and appropriate features. It was observed that using kernel-based dimensional reduction could enhance classification performance using SVM, KNN, and fine tree algorithms. It was demonstrated through the use of two empirical models that the proposed method could discriminate linear damage states correctly and without any error and classify nonlinear damage with significant accuracy. Moreover, the proposed method proves its efficiency even in a highly noisy environment with an SNR of 20 and 15. Finally, to see the existence of the GARCH effect, Kurtosis and ARCH tests were deployed, and the results showed that IMFs followed the GARCH effect; thereby, they were appropriate candidates for the proposed method.

The authors suggest the application of VMD and the GARCH model for unsupervised approaches and reinforcement learning. Moreover, optimization algorithms such as particle swarm optimization (PSO) and grey wolf optimizer (GWO) could be deployed to fnd the optimum number of features. The current limitation of the proposed method is sensitivity to the noisy signals, which can be solved by SNR estimation and reducing the noise by signal processing approaches. Also, we can consider the semisupervised schemes to reduce the efect of noisy features on the performance of feature reduction schemes.

Declarations

Conflict of interest The authors declare that they have no confict of interest.

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