RESEARCH ARTICLE

An accurate optical fow estimation of PIV using fuid velocity decomposition

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Received: 14 October 2020 / Revised: 23 February 2021 / Accepted: 1 March 2021 / Published online: 30 March 2021 © The Author(s), under exclusive licence to Springer-Verlag GmbH Germany, part of Springer Nature 2021

Abstract

In this study, we propose a novel optical fow formulation for estimating high-accuracy velocity felds from tracer particle image sequences. According to the Helmholtz velocity decomposition theorem, the proposed optical fow method decomposes the two-dimensional velocity feld into four components: translation motion, linear distortion motion, shear distortion motion and rotation motion. In this context, regularization terms for diferent motion components are designed, which have a reasonable physical interpretation for the fow characteristics of the fuid. Subsequently, we design specifc regularization parameters for the corresponding regularization terms according to the fow characteristics of the motion components. These regularization parameters can be adaptively adjusted with changes in the image space and velocity feld. In addition, the data term of the optical fow formulation is based on the projected-motion equation derived from the continuity equation, which maintains the compressibility of the fuid in the two-dimensional plane. Velocity felds are estimated from synthetic tracer particle images and hypersonic experimental image sequences, and the velocity results are compared to those of an advanced cross-correlation-based PIV method and previous advanced optical fow methods. The results and comparisons indicate that the proposed method shows good performance and high measurement accuracy when acquiring compressible flow structures from fluid measurements.

Graphic abstract

This work was supported by the National Natural Science Foundation of China (Grant No. 51875228), the National Key R&D Program of China (Grant No. 2020YFA0405700), and the National Defense Science and Technology Innovation Special Zone Project (Grant No. 193-A14-202-01-23).

1 Introduction

With the development of technology, fluid flow measurement techniques have gradually replaced flow visualization techniques in the study of fuid dynamics. Extracting

Extended author information available on the last page of the article

the velocity feld through measurement techniques allows researchers to analyse complex fow feld structures more deeply (Adrian and Westerweel [2011](#page-14-0)). As a non-contact measurement method, particle image velocimetry (PIV) has become one of the main technical methods for fuid flow measurement. Tracer particles are seeded into the fluid flow to follow its movement, and a sheet of pulsed laser light is employed to illuminate these tracer particles undergoing short displacements with a preset interval time. Meanwhile, a camera is utilized to synchronously capture two image frames of the tracer particles. Finally, the velocity feld is calculated by determining the displacements of the particle felds between the two image frames and using the preset interval time (Raffel et al. [2018](#page-15-0)). Because it allows the velocity feld to be determined over an entire image domain, PIV is widely used in many applications, such as aerospace, biological medicine, and nanofabrication. Although PIV has achieved very impressive performance, the measurement accuracy of PIV is signifcantly infuenced by optical imaging technology and PIV measurement methods. Therefore, improving the accuracy and dynamic range of the estimated velocity feld remains an important task in the PIV measurement method (Westerweel et al. [2013](#page-15-1)).

Over the past few decades, various PIV measurement methods have been proposed and developed to improve the measurement accuracy of the velocity feld. These methods can be classifed into two main groups: cross-correlationbased methods and optical fow-based methods (Heitz et al. [2008,](#page-14-1) [2010\)](#page-14-2). The traditional cross-correlation-based method divides the particle image into many fxed-size interrogation windows and assumes that all pixels in an interrogation window have the same velocity vector. When calculating the correlation coefficient of two interrogation windows across two successive image frames, one velocity vector that gives the correlation peak is produced for each interrogation window. The cross-correlation method is widely used in various felds because of its reliability, accuracy, and robustness, but the resolution of the estimated velocity feld is restricted by the size of the interrogation window (Kähler et al. [2012](#page-14-3)). The spatial resolution limitations of the cross-correlationbased method are a well-recognized issue, and there have been many efforts to improve the spatial resolution of the velocity feld, such as velocity feld interpolation and image deformation (Astarita [2008,](#page-14-4) [2009\)](#page-14-5), combined with particle tracking velocimetry (Kähler et al. [2012\)](#page-14-3), variational adaptive Gaussian interrogation window (Becker et al. [2012](#page-14-6)), and adjusting the interrogation window adaptively according to seeding density and velocity information (Theunissen et al. [2007](#page-15-2), [2010](#page-15-3); Yu and Xu [2016;](#page-15-4) Simonini et al. [2019](#page-15-5)). Because of the smoothing efect of the interrogation window on the velocity feld, the estimation error of the crosscorrelation-based method is relatively large in regions with large velocity gradients, such as shock waves, recirculation areas, shear flow and strong vortices (Seong et al. [2019](#page-15-6)).

An alternative to the cross-correlation-based method is the optical fow-based method, which was originally applied in the computer vision feld. Compared to the cross-correlation-based method, which provides a sparse velocity feld (one velocity vector for each interrogation window), the optical fow-based method enables the estimation of the dense velocity feld (one velocity vector for each pixel). First proposed by Horn and Schunck [\(1981\)](#page-14-7), the original optical fow formulation depends on an objective function composed of a data term and a regularization term. The data term is associated with the brightness constancy assumption, which assumes that a given point retains the same intensity in the image plane along its trajectory. To cope with the under-constrained vectorial estimation problem (i.e., aperture problem in the computer vision feld), a regularization term (based on the smoothness constancy assumption) is added in the optical fow formulation. Over the past 20 years, the original optical fow formulation proposed by Horn and Schunck ([1981](#page-14-7)) has been extended to various revised optical flow-based methods for fluid flow motion estimation. Quénot et al. [\(1998\)](#page-15-7) proposed an optical fow technique based on dynamic programming constraints, which was the frst application of the optical fow-based method in PIV. The multi-resolution optical fow approach proposed by Ruhnau et al. [\(2005](#page-15-8)) is a continuous variational formulation for globally estimating the dense velocity feld from particle images. Zhong et al. ([2017\)](#page-15-9) modifed the brightness constancy to the brightness gradient constancy to resist the infuence of the illumination variation in the experiment.

However, these approaches usually rely on computer vision theory and are not rigorously derived from the fundamental equations of fuid mechanics. Therefore, to more reasonably apply optical flow to fluid flow measurements, it is necessary to establish optical fow formulations based on the theory of fuid mechanics (Heitz et al. [2010\)](#page-14-2). Corpetti et al. ([2000,](#page-14-8) [2002,](#page-14-9) [2006\)](#page-14-10) presented a fuid-fow dedicated formulation based on the integrated continuity equation, which is derived from the Reynolds transport equation of fuid in the image plane. Liu and Shen [\(2008](#page-14-11)) built a quantitative connection between the optical fow and the fuid fow for various fow visualizations and proposed a physics-based optical fow method to provide a rational foundation for applying the optical fow method to PIV measurements (Liu et al. [2015](#page-14-12)). In this context, more data terms in the optical fow formulation are combined with the fundamental equations of fuid mechanics, such as the eddy-difusivity model (Cassisa et al. [2011\)](#page-14-13) and the transport equation of turbulent flow (Chen et al. [2015;](#page-14-14) Cai et al. [2018\)](#page-14-15), to improve the performance in estimating fuid fow motion. In addition to data terms based on the physics of fuid fow, various modifcations to the regularization term have also been frequently proposed.

Among the recently developed optical fow methods, the regularization term is usually formulated in the forms of a high-order constraint (Corpetti et al. [2006;](#page-14-10) Liu and Ribeiro [2011;](#page-14-16) Lu et al. [2019](#page-14-17)), orthogonal decomposition (Yuan et al. [2007](#page-15-10)), wavelet expansion (Kadri-Harouna et al. [2013](#page-14-18); Dérian et al. [2011,](#page-14-19) [2013;](#page-14-20) Schmidt and Sutton [2019](#page-15-11), [2020\)](#page-15-12) or physical constraint (Kohlberger et al. [2003](#page-14-21); Ruhnau and Schnörr [2007;](#page-15-13) Heitz et al. [2008;](#page-14-1) Chen et al. [2015;](#page-14-14) Cai et al. [2018](#page-14-15)). These works indicate that the combination of the physics of fuid fow and optical fow schemes is a very promising direction. However, optical fow methods are not without their potential drawbacks. Compared to the crosscorrelation-based method, optical fow methods may exhibit difficulties with experimental noise or illumination variation that commonly occur in experiments. Moreover, the regularization parameters of most optical flow methods have no physical interpretation and are previously tuned empirically. However, the regularization parameters with physical interpretation can be optimally set and are more suitable for opti-cal flow methods dedicated to fluid flow (Cai et al. [2018\)](#page-14-15).

The objectives of this work are to present an optical flow method for processing particle image sequences that addresses the potential drawbacks listed above. The main ideas and contributions of this work consist of proposing a novel optical fow formulation by considering both particle image information and physics theory of fuid fow. According to the Helmholtz velocity decomposition theorem, the proposed optical fow method decomposes the twodimensional velocity feld into four components: translation motion, linear distortion motion, shear distortion motion and rotation motion. In this context, regularization terms for diferent motion components are designed, which have a reasonable physical interpretation for the fow characteristics of the fuid. Subsequently, we design specifc regularization parameters for the corresponding regularization terms according to the fow characteristics of the motion components. These regularization parameters can be adaptively adjusted with changes in the image space and velocity field. In addition, the data term of the optical flow formulation is based on the projected-motion equation derived from the continuity equation, which protects the compressibility of the fuid in the two-dimensional plane. A bilateral flter kernel function is used to resist image noise and illumination changes in the experiment. The proposed velocity decomposition-based variational optical fow method (VD-VOF) is based on the continuity equation and Helmholtz decomposition theorem. It is applicable not only to turbulent or incompressible flow but also to more flow conditions.

The rest of this paper is organized as follows. In Sect. [2,](#page-2-0) the optical fow formulation under fuid velocity decomposition is presented in detail. Then, the discretization scheme and multiscale technique are described in Sect. [3.](#page-6-0) Experimental evaluations on synthetic images and real data are demonstrated in Sects. [4](#page-8-0) and [5,](#page-11-0) respectively. Comparisons with other fuid motion estimation techniques are also discussed in these sections. Finally, a conclusion is discussed in Sect. [6.](#page-13-0)

2 Methodology

In this section, Helmholtz velocity decomposition is frst introduced in detail. Subsequently, regularization terms dedicated to Helmholtz decomposition and the velocity decompositionbased optical fow method are presented and discussed.

2.1 Helmholtz velocity decomposition

The Helmholtz velocity decomposition states that a velocity feld can be uniquely decomposed into a sum of a translational component, a divergence-free (solenoidal) component and a curl-free (irrotational) component. Such a decomposition simplifies the analysis of flow fields since important properties of fow representing physical phenomena such as incompressibility and vorticity can be studied on the decomposed components directly. The Helmholtz decomposition has been proved to be an important tool for fuid analysis and one of the fundamental theorems of fuid dynamics (Bhatia et al. [2013\)](#page-14-22). It has been used by various research communities for a wide variety of applications, such as weather modelling, oceanology, and geophysics (McWilliams [2006\)](#page-15-14). Under the zero boundary condition, the translational component can be ignored, and the vector feld can be expressed as a sum of the gradient of a stream potential and the curl of a velocity potential (Kohlberger et al. [2003](#page-14-21)). This implicit Helmholtz decomposition has been used for the optical fow method in the literature (Corpetti et al. [2002](#page-14-9), [2006](#page-14-10); Kohlberger et al. [2003](#page-14-21); Yuan et al. [2007\)](#page-15-10) and is different from the explicit Helmholtz decomposition into four components involved in the current work.

Let $\mathbf{x} = (x, y)^T$ denote the coordinate location of a point in the two-dimensional velocity field plane; $u(x) = (u(x), v(x))^T$ is the velocity vector at location *x*, and $x + \delta x$ is a neighbouring point near *x* with the velocity vector $u(x + \delta x)$. If the dis- $\tance \delta x = (\delta x, \delta y)^T$ between the point *x* and the point $x + \delta x$ is small enough, by applying the frst-order Taylor expansion, the following equations can be obtained:

$$
u(x + \delta x) = u(x) + \frac{\partial u}{\partial x} \delta x + \frac{\partial u}{\partial y} \delta y.
$$
 (1)

The detailed expression is as follows:

$$
\delta u(x) = \frac{\partial u}{\partial x} \delta x + \frac{\partial u}{\partial y} \delta y = \chi \cdot \delta x + \theta \cdot \delta x + \zeta \cdot \delta x, \tag{2}
$$

$$
\chi = \begin{bmatrix} \varepsilon_{xx} & 0 \\ 0 & \varepsilon_{yy} \end{bmatrix}, \ \theta = \begin{bmatrix} 0 & \varepsilon_{xy} \\ \varepsilon_{xy} & 0 \end{bmatrix}, \ \zeta = \begin{bmatrix} 0 & -\omega \\ \omega & 0 \end{bmatrix}, \tag{3}
$$

where χ , θ , ζ denote the linear distortion rate tensor, shear distortion rate tensor, and rotation tensor, respectively; $\epsilon_{xx} = \partial u / \partial x$ and $\epsilon_{yy} = \partial v / \partial y$ are the linear distortion rates in the *x* and *y* directions, respectively; $\varepsilon_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}$ /2 is the shear distortion rate; and $\omega = (\partial v / \partial x - \partial u / \partial y) / 2$ is the rotation angular rate. Therefore, Eq. [1](#page-2-1) can be derived as:

$$
u(x + \delta x) = u(x) + \chi \cdot \delta x + \theta \cdot \delta x + \zeta \cdot \delta x. \tag{4}
$$

On the right-hand side of Eq. [4,](#page-3-0) the frst term represents the translation motion of the fuid particle from point *x* to point $\mathbf{x} + \delta \mathbf{x}$ at velocity $\mathbf{u}(\mathbf{x})$, and the following three terms represent linear deformation motion, shear deformation motion and rotation motion. In this way, the velocity vector $u(x + \delta x)$ at point $x + \delta x$ can be decomposed into four components, which is the explicit function of Helmholtz velocity decomposition. There are two important diferences between Helmholtz velocity decomposition and general rigid object velocity decomposition: fuid motion has one more deformation velocity component than rigid object motion; rigid object velocity decomposition is valid for the entire rigid body, while Helmholtz velocity decomposition is valid only within a fluid particle and is a local theorem.

2.2 Regularization term based on Helmholtz velocity decomposition

To address the under-constrained vectorial estimation problem in the optical fow formulation, a classical regularization term was proposed in the seminal work of Horn and Schunck ([1981\)](#page-14-7). The regularization term proposed by Horn and Schunck [\(1981](#page-14-7)) is based on the smoothness constancy assumption that all neighbouring points have similar motions and is expressed as the magnitude of the frst-order velocity gradient:

$$
\int_{\Omega} \left(|\nabla u|^2 + |\nabla v|^2 \right) dxdy,\tag{5}
$$

where ∇ is the gradient operator in two-dimensional directions, and Ω denotes the two-dimensional image domain. Hereafter, the coordinate location of a point in the velocity feld plane is equivalent to the pixel coordinate in the image domain, defined as $\mathbf{x} = (x, y)^T$, and the velocity vector is equivalent to the optical fow vector, defned as $u(x) = (u(x), v(x))$ ^T. This classical first-order regularization term is actually the frst-order Tikhonov regularization for ill-posed problems (Liu and Shen [2008\)](#page-14-11) and has been used in the physics-based optical fow method proposed by Liu et al. [\(2015\)](#page-14-12). Furthermore, this regularization term can be interpreted as derived from a homogeneous divergence-free

uncertainty random feld and has been extended with a small-scale difusion tensor in Cai et al. ([2018\)](#page-14-15).

It has been demonstrated that the classical frst-order regularization term $(Eq. 5)$ $(Eq. 5)$ $(Eq. 5)$ is equivalent to the first-order div-curl regularization term, which is the smoothing penalization of the magnitude of divergence and vorticity (Corpetti et al. [2006](#page-14-10)):

$$
\int_{\Omega} \left(|\operatorname{div} \boldsymbol{u}|^2 + |\operatorname{curl} \boldsymbol{u}|^2 \right) dxdy,\tag{6}
$$

where div \boldsymbol{u} and curl \boldsymbol{u} are the divergence and vorticity of the velocity feld, respectively. Generally, the regularization term in the optical fow cost functional is set from a regularity condition on the solution. However, such frst-order regularization terms are particularly suited for rigid movement and are difficult to relate to kinematical or dynamical properties of the fuid (Chen et al. [2015\)](#page-14-14). To protect the key information contained in the vorticity feld of fuid motion, Chen et al. ([2015](#page-14-14)) simplifed the frst-order divergence-curl regularization term to the penalization of the magnitude of divergence, which addresses 2D incompressible turbulent flows only.

Compared with general stable object motion, fuid motion has additional deformation velocity components, so that the fuid velocity feld normally exhibits areas with high values of vorticity and divergence. The underestimation of deformation movement may become a source of error for velocity feld estimation. In this case, higher-order regularization terms are considered, and the third derivatives of the velocity feld have been used in the wavelet domain by Schmidt and Sutton [\(2019](#page-15-11)). Another more commonly used higher-order regularization term is the second-order div-curl regularization term proposed by Corpetti et al. ([2006](#page-14-10)), which minimizes the gradients of the divergence and the vorticity:

$$
\int_{\Omega} \left(|\nabla \operatorname{div} \boldsymbol{u}|^2 + |\nabla \operatorname{curl} \boldsymbol{u}|^2 \right) dxdy. \tag{7}
$$

This regularization term is consistent with the physical properties of fuid fows. Schmidt and Sutton ([2020](#page-15-12)) implemented it in the wavelet domain and has achieved better results than the third-order velocity gradient regularization term (Schmidt and Sutton [2019\)](#page-15-11). Kohlberger et al. [\(2003\)](#page-14-21) used the third derivatives of the stream potential and velocity potential in the implicit Helmholtz decomposition as the regularization term, which ignored the translation component of the fuid fow. These works indicate that higher-order regularization terms with physical properties of fuid fow can obtain more accurate velocity estimations.

However, these optical flow methods completely ignore the image intensity information when establishing the regularization term, which is not advisable in the optical fow method in the computer vision feld (Sun et al. [2014](#page-15-15)). In the current work, we use Helmholtz decomposition to design a new regularization term, which considers not only the flow characteristics of the fluid but also the image information. This new regularization term enforces a smoothing constraint not on the velocity feld *u* itself but only on four motion components we are interested in. The explicit function of Helmholtz decomposition is used to combine the physical properties of the fuid fow with the image information within reasonable assumptions.

First, we consider only fluid flow characteristics and use the following notations:

$$
|\nabla u|^2 := |\nabla u|^2 + |\nabla v|^2,
$$
 (8)

$$
|\nabla \chi|^2 := |\nabla \varepsilon_{xx}|^2 + |\nabla \varepsilon_{yy}|^2, \tag{9}
$$

$$
|\nabla \theta|^2 := 2 |\nabla \varepsilon_{xy}|^2, \tag{10}
$$

$$
|\nabla \zeta|^2 := 2|\nabla \omega|^2. \tag{11}
$$

After the Helmholtz decomposition, the four motion components of fuid fow are smoothly constrained. From the previous velocity decomposition equation (Eq. [4\)](#page-3-0) in Sect. 2.1, a novel regularization term for optical fow estimation can be proposed:

$$
\int_{\Omega} \left(|\nabla u|^2 + |\nabla \chi|^2 + |\nabla \theta|^2 + |\nabla \zeta|^2 \right) dxdy \tag{12}
$$

where the translation motion *u*, linear distortion rate tensor χ , shear distortion rate tensor θ and rotation tensor ζ are smoothly constrained. Substituting Eqs. [8–](#page-4-0)[11](#page-4-1) into Eq. [12,](#page-4-2) we obtain the following formulation:

$$
\int_{\Omega} \left(|\nabla u|^2 + |\nabla v|^2 + |\nabla \epsilon_{xx}|^2 + |\nabla \epsilon_{yy}|^2 \right) dxdy + \int_{\Omega} \left(2|\nabla \epsilon_{xy}|^2 + 2|\nabla \omega|^2 \right) dxdy.
$$
\n(13)

However, such a regularization term does not accommodate discontinuities in the velocity feld. To capture locally non-smooth motion, such as shock waves, it is necessary to allow outliers in the regularization term (Sun et al. [2014](#page-15-15)). This can be achieved by a non-quadratic penalty function $\psi(s^2) = \sqrt{s^2 + \tau^2}$. The symbol τ is a prefixed small constant that is usually set as $\tau = 0.001$ to ensure that $\psi(s^2)$ is strictly convex. The use of the non-quadratic penalty function ensures a unique solution to the minimization problem and allows the construction of simple global convergence algorithm. In this way, a piecewise smooth fow feld is modelled, and we propose the minimisation of the following formulation:

$$
\int_{\Omega} \psi\left(|\nabla u|^2 + |\nabla v|^2\right) dxdy \n+ \int_{\Omega} \left(\psi\left(\left|\nabla \epsilon_{xx}\right|^2\right) + \psi\left(\left|\nabla \epsilon_{yy}\right|^2\right)\right) dxdy \n+ \int_{\Omega} \left(2\psi\left(\left|\nabla \epsilon_{xy}\right|^2\right) + 2\psi\left(\left|\nabla \omega\right|^2\right)\right) dxdy.
$$
\n(14)

On the basis of Eq. [14,](#page-4-3) we add the image intensity information to construct an image-driven self-adaptive regularization parameter to mitigate the regularization at fow motion boundaries. In supersonic or hypersonic fow, there are discontinuous fow structures such as shock wave, expansion wave, and shear layer (Lu et al. [2019\)](#page-15-16). The abrupt change in velocity between the fuid and the wall can also be regarded as discontinuous. Here, we regard these discontinuous fow structures as flow motion boundaries, similar to image edges. In the regions of the fow motion boundaries, the fuid density changes abruptly, causing a corresponding change in the average concentration of tracer particles. This is refected in the particle image in which the image intensity changes drastically in the regions of the fow motion boundaries and forms the image edges (Zhong et al. [2017](#page-15-9)). The self-adaptive regularization parameter $e^{-|\nabla I|^2}$ is based on a monotonically decreasing function of the magnitude of the image gradient, and it respects fow motion boundaries (Drulea and Nedevschi [2011](#page-14-23)). In the regions with uniform change in image intensity, this regularization parameter increases, while in the regions of the image edges, this regularization parameter automatically decreases, so that the fow motion boundaries can be protected from oversmoothing and blurring.

In addition to translation motion, linear deformation and shear deformation motion, rotation motion exists in the shock wave, shear layer and vortex structure. However, the image edge of the vortex structure in turbulent fow is not obvious. To further improve the optical fow performance in the region of the vortex structure, a vorticity bilateral flter is added as a regularization parameter of the rotation tensor constraint. The bilateral flter consists of a Gaussian (space) flter and a range (vorticity) flter (Lin et al. [2014](#page-14-24)):

$$
BF_{\omega} = G_s \cdot G_{\omega}
$$

= $\exp\left(-\frac{|x - x_0|^2}{2\sigma_s}\right) \cdot \exp\left(-\frac{|\omega_x - \omega_{x_0}|^2}{2\sigma_{\omega}}\right),$ (15)

where G_s denotes the spatial Gaussian filter of size ρ and with standard deviation σ_s , and G_ω denotes the vorticity Gaussian filter of size ρ and with standard deviation σ_{ω} . The weight of a pixel x decreases as the distance from the centre x_0 of the filter kernel increases. In addition, the weight decreases if the vorticity ω_x of the pixel x is different from the filter kernel centre's vorticity ω_{x_0} . The vorticity bilateral

flter considers the spatial distance and vorticity diference between the pixel and its neighbouring pixels to better protect the vortex structure.

The improved regularization term, which includes the non-quadratic penalty function, a bilateral flter kernel function and a self-adaptive parameter based on the image gradient, is proposed as follows:

$$
E_{reg} = \int_{\Omega} e^{-|\nabla I|^2} \psi \left(|\nabla u|^2 + |\nabla v|^2 \right) dxdy
$$

+
$$
\int_{\Omega} e^{-|\nabla I|^2} \left(\psi \left(|\nabla \varepsilon_{xx}|^2 \right) + \psi \left(|\nabla \varepsilon_{yy}|^2 \right) \right) dxdy \qquad (16)
$$

+
$$
\int_{\Omega} \left(2e^{-|\nabla I|^2} \psi \left(|\nabla \varepsilon_{xy}|^2 \right) + 2BF_{\omega} * \psi (|\nabla \omega|^2) \right) dxdy,
$$

where ∗ denotes the convolution operation. Compared to the classical regularization term, the proposed regularization term considers both particle image information and physics theory of fuid fow. According to the Helmholtz velocity decomposition, the proposed optical fow method decomposes the two-dimensional velocity feld into four components: translation motion, linear distortion motion, shear distortion motion and rotation motion. In this context, regularization terms for diferent motion components are designed, which have a reasonable physical interpretation for the fuid fow. Subsequently, a non-quadratic penalty function is used to capture the locally non-smooth motion. In addition, we design the specifc regularization parameters for the corresponding regularization terms according to the flow characteristics of the motion components. The monotonically decreasing function $e^{-|\nabla I|^2}$ automatically changes according to the image intensity to protect the fow motion boundaries from being over-smoothed. The vorticity bilateral filter BF_{ω} automatically changes according to the vorticity feld to protect the vortex structure. These regularization parameters can be adaptively adjusted with changes in the image space and velocity feld.

2.3 Velocity‑decomposition‑based optical fow

Optical flow-based PIV methods estimate the motion between two sequential particle images taken at times *t* and $t + dt$ at every pixel position. When $I(x, t)$ is the intensity of the particle image obtained at time t , and the position (x, t) is moved by d*x* and d*t* between the two images, the intensity of these two pixel points can generally be assumed to be invariable (Horn and Schunck [1981](#page-14-7)):

$$
I(x, t) \approx I(x + dx, t + dt). \tag{17}
$$

The right-hand side of Eq. [17](#page-5-0) can be expanded using the Taylor series, and the following basic optical fow equation can be obtained:

$$
\frac{\partial I}{\partial t} + \nabla I \cdot \mathbf{u} = 0,\tag{18}
$$

Corpetti et al. ([2006](#page-14-10)) proposed the integrated continuity equation in the image plane as an alternative based on the Reynolds transport equation of fuid under the assumption that the image intensity is proportional to an integral of the fuid density across a measurement domain. The integrated continuity equation is expressed as follows:

$$
\frac{\partial I}{\partial t} + \nabla I \cdot \boldsymbol{u} + I \operatorname{div} \boldsymbol{u} = 0. \tag{19}
$$

The projected-motion equations for typical flow visualizations were derived in the pivotal work of Liu and Shen ([2008\)](#page-14-11), and those authors showed that the optical flow \boldsymbol{u} is proportional to the path-averaged velocity of particles across the laser sheet. The physics-based optical fow equation proposed by Liu and Shen ([2008](#page-14-11)) in the image plane is given by

$$
\frac{\partial I}{\partial t} + \nabla \cdot (Iu) = f(x, I),\tag{20}
$$

where $f(\mathbf{x}, I) = D\nabla^2 I + Dc\mathbf{B} + c\mathbf{n} \cdot (\Psi \mathbf{u})\Big|_{\Gamma_1}^{\Gamma_2}$ $\frac{1}{\Gamma_1}$, *D* is a diffusion coefficient, c is a coefficient for particle scattering/absorption, and *B* is a boundary term related to the considered transported quantity Ψ and its derivatives coupled with the derivatives of the control surfaces confning the laser illumination volume. Because the control surfaces are planar, there is no particle difusion through the molecular process, and the rate of particle accumulation in the laser illumination volume is ignored, the right-hand side term $f(x, I)$ in Eq. [20](#page-5-1) could be approximately equal to 0, and Eq. [20](#page-5-1) is equivalent to Eq. [19](#page-5-2) (Heitz et al. [2010](#page-14-2)). Then, the integrated continuity equation proposed by Corpetti et al. [\(2006\)](#page-14-10) is consistent with the projected-motion equation presented by Liu and Shen ([2008\)](#page-14-11) in the case of a laser-sheet-illuminated particle image.

To acquire the compressible structures in the two-dimensional flow field, consistent with the objective of the proposed regularization term, the data term of the proposed optical fow method is based on the projected-motion equation derived from the continuity equation, which protects the compressibility of the fuid in the two-dimensional plane. A bilateral flter kernel function is used to resist image noise and illumination changes in the experiment (Lin et al. [2014](#page-14-24)). The data term of the VD-VOF estimation cost function is defned as follows:

$$
E_{data} = \int_{\Omega} BF_I * \left| \frac{\partial I}{\partial t} + \nabla I \cdot \mathbf{u} + I \text{div } \mathbf{u} \right|^2 \text{d}x \text{d}y,\tag{21}
$$

where $BF_I = G_s \cdot G_I$, G_s and G_I denote the spatial Gaussian filter and the intensity Gaussian filter, respectively. The weight of the bilateral flter is automatically adjusted according to the spatial distance and image intensity diference. As mentioned earlier, in the region of the fow motion boundary, the particle concentration changes drastically, and the image intensity changes accordingly. The data term with the bilateral flter is more sensitive to particle concentration, and the automatically adjusted weight can reduce the propagation or difusion of the fow motion boundary, for example, to protect the shock wave from being oversmoothed or the boundary of the shear layer from being blurred. Furthermore, the spatial Gaussian flter in the bilateral flter can make the data term more robust, thereby resisting image noise.

Based on Eqs. [16](#page-5-3) and [21](#page-5-4), the model of the proposed VD-VOF method is as follows:

$$
E = \lambda E_{data} + E_{reg}
$$

\n
$$
= \lambda \int_{\Omega} BF_I * \left| \frac{\partial I}{\partial t} + \nabla I \cdot \boldsymbol{u} + I \text{div } \boldsymbol{u} \right|^2 \text{d}x \text{d}y
$$

\n
$$
+ \int_{\Omega} e^{-|\nabla I|^2} \psi \left(|\nabla u|^2 + |\nabla v|^2 \right) \text{d}x \text{d}y
$$

\n
$$
+ \int_{\Omega} e^{-|\nabla I|^2} \left(\psi \left(|\nabla \varepsilon_{xx}|^2 \right) + \psi \left(|\nabla \varepsilon_{yy}|^2 \right) \right) \text{d}x \text{d}y
$$

\n
$$
+ \int_{\Omega} \left(2e^{-|\nabla I|^2} \psi \left(|\nabla \varepsilon_{xy}|^2 \right) + 2BF_{\omega} * \psi (|\nabla \omega|^2) \right) \text{d}x \text{d}y,
$$
\n(22)

where $\lambda > 0$ is a smoothness parameter controlling the balance between the data term and the regularization term. The frst two terms on the right side of the equation can be regarded as the modifed model of the physics-based optical fow method proposed by Liu et al. [\(2015\)](#page-14-12). The last two terms on the right side of the equation are constrained based on Helmholtz velocity decomposition and can also be seen as higher-order constraints of the velocity feld. These two terms are often referred to as postprocessing constraints in the computer vision feld.

3 Discretization and multiscale

In this section, the optimization strategies based on the discretization scheme and multiscale technique are frst presented and discussed. Then, the complexity analysis and parameter settings are introduced in detail.

3.1 Discretization scheme

For the functional problem in Eq. [22](#page-6-1), auxiliary variables are usually introduced to minimize the convex approximation of the original functional (Corpetti et al. [2006;](#page-14-10) Drulea and Nedevschi [2011](#page-14-23); Lu et al. [2019\)](#page-14-17). We introduce four auxiliary scalar fields, $\hat{\epsilon}_{xx}$, $\hat{\epsilon}_{yy}$, $\hat{\epsilon}_{xy}$ and $\hat{\omega}$, which constitute direct estimates of the distortion rate and rotation rate. Then, the VD-VOF model is decomposed into fve parts:

$$
E_1(u, v, \varepsilon_{xx}, \varepsilon_{yy}, \varepsilon_{xy}, \omega)
$$

= $\lambda \int_{\Omega} BF_I * \left| \frac{\partial I}{\partial t} + \nabla I \cdot u + I \text{div } u \right|^2 \text{d}x \text{d}y$
+ $\int_{\Omega} e^{-|\nabla I|^2} \psi \left(|\nabla u|^2 + |\nabla v|^2 \right) \text{d}x \text{d}y$
+ $\int_{\Omega} \left(\left| \varepsilon_{xx} - \hat{\varepsilon}_{xx} \right|^2 + \left| \varepsilon_{yy} - \hat{\varepsilon}_{yy} \right|^2 \right) \text{d}x \text{d}y$
+ $\int_{\Omega} \left(2 \left| \varepsilon_{xy} - \hat{\varepsilon}_{xy} \right|^2 + 2 |\omega - \hat{\omega}|^2 \right) \text{d}x \text{d}y$, (23)

$$
E_2(\hat{\epsilon}_{xx}) = \int_{\Omega} e^{-|\nabla I|^2} \psi\left(|\nabla \hat{\epsilon}_{xx}|^2\right) dxdy
$$

+
$$
\int_{\Omega} |\epsilon_{xx} - \hat{\epsilon}_{xx}|^2 dxdy,
$$
 (24)

$$
E_3(\hat{\epsilon}_{yy}) = \int_{\Omega} e^{-|\nabla I|^2} \psi\left(\left|\nabla \hat{\epsilon}_{yy}\right|^2\right) dxdy + \int_{\Omega} \left|\epsilon_{yy} - \hat{\epsilon}_{yy}\right|^2 dxdy,
$$
 (25)

$$
E_4(\hat{\epsilon}_{xy}) = \int_{\Omega} 2e^{-|\nabla I|^2} \psi\left(\left|\nabla \hat{\epsilon}_{xy}\right|^2\right) dxdy + \int_{\Omega} 2\left|\hat{\epsilon}_{xy} - \hat{\epsilon}_{xy}\right|^2 dxdy,
$$
\n(26)

$$
E_5(\hat{\omega}) = \int_{\Omega} 2BF_{\omega} * \psi \left(|\nabla \hat{\omega}|^2 \right) dxdy
$$

+
$$
\int_{\Omega} 2|\omega - \hat{\omega}|^2 dxdy.
$$
 (27)

Because the distortion rate (ϵ_{xx} , ϵ_{yy} , ϵ_{xy}) and the rotation rate ω are the first-order differential forms of the velocity field (u, v) , the minimization problem of $E_1(u, v, \varepsilon_{xx}, \varepsilon_{yy}, \varepsilon_{xy}, \omega)$ can be regarded as equivalent to that of $E_1(u, v, u_x, u_y, v_x, v_y)$. This convex minimization problem can be optimized by alternating steps updating either (u, v) or $(\hat{\epsilon}_{xx}, \hat{\epsilon}_{yy}, \hat{\epsilon}_{xy}, \hat{\omega})$ in every iteration. For E_1 , $\hat{\epsilon}_{xx}$, $\hat{\epsilon}_{yy}$, $\hat{\epsilon}_{xy}$ and $\hat{\omega}$ are considered fixed, and (u, v) is unknown. The minimum of the energy model E_1 can be found by solving the associated Euler-Lagrange equations, given by

$$
0 = \lambda BF_I * I\delta_x \left(\frac{\partial I}{\partial t} + \nabla I \cdot \boldsymbol{u} + I \text{div} \, \boldsymbol{u} \right)
$$

+ $e^{-|\nabla I|^2} \text{div} (\boldsymbol{\psi}' \cdot \nabla \boldsymbol{u}) + \nabla^2 \boldsymbol{u} - \delta_x \hat{\epsilon}_{xx} - \delta_y \hat{\epsilon}_{xy} + \delta_y \hat{\omega},$ (28)

$$
0 = \lambda BF_I * I\delta_y \left(\frac{\partial I}{\partial t} + \nabla I \cdot \boldsymbol{u} + I \text{div} \,\boldsymbol{u}\right) + e^{-|\nabla I|^2} \text{div}(\boldsymbol{\psi}' \cdot \nabla \nu) + \nabla^2 \nu - \delta_y \hat{\varepsilon}_{yy} - \delta_x \hat{\varepsilon}_{xy} - \delta_x \hat{\omega},
$$
\n(29)

with $\psi'(s^2) = 1/(2\sqrt{s^2 + \tau^2})$. To simplify the equations, we use the following notation:

$$
\delta_x := \frac{\partial}{\partial x} \n\delta_y := \frac{\partial}{\partial y} \n\psi' := \psi' (|\nabla u|^2 + |\nabla v|^2).
$$
\n(30)

The Euler-Lagrange equations as expressed in Eqs. [28](#page-6-2) and [29](#page-7-0) can be solved by Gauss-Seidel iteration or successive over relation (SOR) iteration. The details of the numerical algorithm for solving the above equations, Eqs. [28](#page-6-2) and [29,](#page-7-0) have been derived in detail in Sánchez et al. [\(2013](#page-15-17)), and we do not discuss this process further here.

For the minimum of the energy model Eqs. [24–](#page-6-3)[27](#page-6-4), the numerical schemes are expected to be similar. Taking E_2 as an example, $\hat{\epsilon}_{xx}$ is unknown, and (u, v) and ϵ_{xx} are considered fxed. To make this problem easier to solve, we modify the penalty function $\psi(s^2)$ to the L1 norm as follows:

$$
E_2(\hat{\epsilon}_{xx}) = \int_{\Omega} e^{-|\nabla I|^2} |\nabla \hat{\epsilon}_{xx}| dxdy + \int_{\Omega} |\epsilon_{xx} - \hat{\epsilon}_{xx}|^2 dxdy.
$$
 (31)

The Euler-Lagrange equation for Eq. [31](#page-7-1) is

$$
- \operatorname{div} \left(e^{-|\nabla I|^2} \frac{\nabla \hat{\varepsilon}_{xx}}{|\nabla \hat{\varepsilon}_{xx}|} \right) + 2 \hat{\varepsilon}_{xx} - 2 \varepsilon_{xx} = 0, \tag{32}
$$

which can be solved using a fxed-point iteration scheme. The details of the numerical algorithm for solving Eq. [32](#page-7-2) have been derived in detail in Drulea and Nedevschi [\(2011](#page-14-23)); as before, we do not discuss this process further here.

3.2 Multiscale strategy

Generally, one typical weakness is that optical fow methods require small displacements to achieve sufficient accuracy because of the condition for establishing the Taylor series linearization (Heitz et al. [2010\)](#page-14-2). In the case of a large displacement between two sequential particle images, the discordance between the temporal derivative and the spatial gradient in the image plane may lead to poor velocity estimations. To address this, optical fow methods are typically embedded into a sequential coarse-to-fne multiscale strategy, which has been proved efective for improving the estimation range of optical flow estimations in the literature (Heitz et al. [2008,](#page-14-1) [2010;](#page-14-2) Chen et al. [2015](#page-14-14); Cai et al. [2018](#page-14-15); Lu et al. [2019](#page-14-17)).

Algorithm 1 VD-VOF method with multiscale strategy

- **Input:** The pair of PIV images, $I(t)$, $I(t+\Delta t)$; The smoothness parameter, λ ; The layer of the pyramid, K ; The scale factor for pyramid downsampling, η ; The number of warping iterations, L ; The number of alternating iterations, R ; The number of SOR iterations, P ; The number of fixed-point iterations, Q ;
- **Output:** Two-dimensional velocity field, $u = (u, v)$;
- 1: Build up pyramid $I^t pyr\{k\}$, $I^{t+\Delta t} pyr\{k\}$ of $I(t)$, $I(t+$ Δt) using η
- 2: for $k = K 1$ to 0 do

$$
B: \qquad I^t = I^t pyr\{k\}, \ I^{t+\Delta t} = I^{t+\Delta t} pyr\{k\};
$$

- $4:$ if $k == K - 1$ then
- $5:$ Initialize u, v to zero;
	- Initialize $\hat{\varepsilon}_{xx}, \hat{\varepsilon}_{yy}, \hat{\varepsilon}_{xy}$, and $\hat{\omega}$ to zero;
- $7:$ else 8:

 $6:$

- Upsample u, v from the upper pyramid level; 9: Upsample $\hat{\varepsilon}_{xx}$, $\hat{\varepsilon}_{yy}$, $\hat{\varepsilon}_{xy}$, and $\hat{\omega}$ from the upper pyramid level;
- $10:$ end if $11:$ for $l = L - 1$ to 0 do Compute $I_w^{t+\Delta t} = warping(I^{t+\Delta t}, u)$ 12:
- using bicubic interpolation; Compute I_x , I_y , and I_t , using I^t and $I_w^{t+\Delta t}$; $13[·]$ 14: for $r = R - 1$ to 0 do for $p = P - 1$ to 0 do $15:$ 16: Update δu and δv using Eqs. 28 and 29 with SOR iteration;

```
17:end for
18:
                    for q = Q - 1 to 0 do
19:
                        Update \hat{\varepsilon}_{xx}, \hat{\varepsilon}_{yy}, \hat{\varepsilon}_{xy}, and \hat{\omega} using Eq. 32
                           with fixed-point iteration;
20:end for
21:end for
22:Update u = u + \delta u and v = v + \delta v;
23:
          end for
24: end for
```
The main idea of this strategy is the image pyramid structure and image warping operation and is divided into the following steps: (a) an image pyramid structure of a particle image pair is constructed by a series of low-pass flters and a downsampling process on the image such that the resolutions of the images in the pyramid are successively reduced; (b) the principal component displacements are estimated from the coarse-resolution level and then projected onto the next fner resolution level of the image pyramid, and the incremental displacements are then estimated while going down the image pyramid; and (c) an image warping operation is performed iteratively at each pyramid level so that only small displacement increments are estimated in each image warping operation.

In the image warping iteration process, let u_0 denote the displacement field transferred from the previous warping iteration. Then, a warped intermediate image $I_w(t + \Delta t) = I(x + u_0 \Delta t, t + \Delta t)$ is obtained by warping the second image $I(x + u\Delta t, t + \Delta t)$ to the first image $I(x, t)$ with u_0 . In the current warping iteration, the small displacement increment δu is estimated between the first image and the warped image rather than the second image. Finally, the updated $u_0 = u_0 + \delta u$ is transferred to the next warping iteration. As the warping iteration progresses, δu gradually approaches 0, u_0 gradually approaches u , and the intermediate image $I_w(t + \Delta t)$ gradually approaches the first image $I(x, t)$. The framework of the minimization process of the proposed VD-VOF method is shown in Algorithm 1 with the image pyramid structure and image warping operation.

3.3 Complexity analysis and parameter settings

Complexity analysis is a common technique to compare the computational efficiency of different methods and has been used in the optical fow method literature (Kadri-Harouna et al. [2013](#page-14-18); Lu et al. [2019;](#page-14-17) Schmidt and Sutton [2020](#page-15-12)). Assuming that the resolution of the particle image is $N \times N$, for the space complexity, the proposed VD-VOF method saves the image pyramid $I^tpyr{k}$, the corresponding velocity feld (*u*, *v*) and auxiliary scalar fields $(\hat{\epsilon}_{xx}, \hat{\epsilon}_{yy}, \hat{\epsilon}_{xy}, \hat{\omega})$ at runtime, and its space complexity is $K \times N^2 = O(N^2)$. In the image warping operation, the space complexity is $K \times L \times N^2 = O(N^2)$ (the image size *N* is much larger than the layer of image pyramid *K* and the number of warping iterations *L*). Therefore, the space complexity of the proposed VD-VOF method is $O(N^2)$. For the time complexity, the calculation of (I_x, I_y, I_t) is related to the image resolution $N \times N$, and the time complexity is $O(N^2)$. The calculation of $(\delta u, \delta v)$ consists of iteratively solving partial diferential equations Eqs. [28](#page-6-2) and [29,](#page-7-0) the number of SOR iterations *P* is generally close to *N*, and the corresponding time complexity is $O(N^3)$. In the calculation of the auxiliary scalar fields $(\hat{\epsilon}_{xx}, \hat{\epsilon}_{yy}, \hat{\epsilon}_{xy}, \hat{\omega})$ using Eq. [32](#page-7-2) with fxed-point iteration, the number of fxed-point iterations is also close to *N*, and the corresponding time complexity is $O(N^3)$. Therefore, the time complexity of the proposed VD-VOF method is $O(N^3)$.

The range of the smoothness weight parameter is approximately $2000 \le \lambda \le 4000$. The size of the bilateral kernel is 5 pixels, the standard deviation for the space Gaussian is 0.85, and the standard deviation for the image intensity Gaussian and vorticity Gaussian is 0.1 (the image and vorticity intensities are normalized within [0, 1]). The number of image warping iterations *L* is fxed to 5, the image warping interpolation method is the bicubic interpolation method, and the number of alternating iterations *is fixed to 10. The* number of SOR iterations *P* is fixed to 200, and the number of fxed-point iterations *Q* is fxed to 200. The number of pyramid layers is approximately $2 \leq K \leq 5$, chosen based on the image size and fluid velocity, and the scale factor η for downsampling is set to 0.5.

4 Evaluations of synthetic data

In this section, diferent synthetic image sequences are used to investigate the performance of the proposed VD-VOF method. The main advantage of testing on synthetic image sequences is that the true velocity feld is known and can be used for comparison with the velocity feld estimated by different methods. Although these synthetic image sequences are usually generated under diferent ideal conditions, they are quite suitable for evaluating the performance of diferent PIV measurement methods. A real experimental tracer particle image sequence is tested and discussed in the next section. The estimated velocity feld from the VD-VOF method is compared against the true velocity feld, as well as velocity estimations using a multiple-pass cross-correlation estimation (Insight 4G, TSI Inc.), the classical H $\&$ S R+S estimation by Ruhnau et al. ([2005\)](#page-15-8), the physics-based estimation by Liu [\(2017](#page-14-25)), the feld-segmentation-based estimation by Lu et al. [\(2019](#page-14-17)), and the stochastic model estimation by Cai et al. ([2018\)](#page-14-15). The parameter settings of all compared approaches are described in detail in our previous work (Lu et al. [2019\)](#page-14-17), and we do not discuss this process further here.

Hereafter, we follow a standard way to quantitatively evaluate the experimental results by computing the root mean square error (RMSE) and average angle error (AAE) over *N* pixels of the image:

$$
RMSE = \sqrt{\frac{1}{N} \sum_{i=1}^{N} \left| u_i^t - u_i^e \right|^2},
$$
\n(33)

$$
AAE = \frac{1}{N} \sum_{i=1}^{N} \arccos\left(\frac{\boldsymbol{u}_i^t \cdot \boldsymbol{u}_i^e}{\left|\boldsymbol{u}_i^t\right|\left|\boldsymbol{u}_i^e\right|}\right),\tag{34}
$$

where $u^t = (u^t, v^t)$ and $u^e = (u^e, v^e)$ denote the ground truth and the estimated velocity feld, respectively. *N* represents the total number of pixels of the estimated particle image, and the index *i* represents the pixel position where the velocity is computed. RMSE and AAE are commonly used as the evaluation criteria in the velocity estimation of fuid fow (Chen et al. [2015;](#page-14-14) Cai et al. [2018](#page-14-15); Lu et al. [2019\)](#page-14-17). To avoid the infuence of errors caused by the matrix extrapolation, the velocity values of the widths of the three pixels around the image are ignored when calculating RMSE and AAE. Moreover, the spectrum analysis of the velocity feld is introduced to evaluate the estimated fow structure, which is also a common evaluation in fluid flow motion estimation (Heitz et al. [2010](#page-14-2); Schmidt and Sutton [2019,](#page-15-11) [2020](#page-15-12)). This evaluation criterion can be used not only for simulation data but also for experimental data. Therefore, we compare the proposed method with other approaches based on these evaluation criteria.

4.1 Analytic fows

These synthetic particle image data contain six diferent flows, which were generated by the Cemagref team (Carlier) [2005](#page-14-26)). The analytic flows can be divided into two categories: viscous fows and potential fows. Viscous fows include Poiseuille flow and Lamb-Oseen vortex flow, which are two well-known examples with velocity gradients. Poiseuille flow is viscous flow between two parallel plates with a constant streamwise pressure gradient. The Lamb-Oseen vortex is a two-dimensional viscous fow with circular streamlines and a decreasing vorticity along the radial distance. The provided potential fows in the analytic fows data set are a uniform flow, a sink, a vortex and a potential flow around a cylinder with circulation. For each of the analytic fows,

41 successive images are given with the same parameters (particle size, concentration, intensity); the corresponding ground-truth velocity felds of the six diferent fows are presented in Fig. [1](#page-9-0).

As shown in Fig. [2](#page-9-1), quantitative comparisons of various PIV methods can be performed using the evaluation criteria RMSE and AAE. On these six synthetic particle images, the proposed VD-VOF method provides better results than the previous advanced optical fow methods. Compared to the original model (H $&$ S R+S estimation), different data terms (physics-based estimation) and smoothing terms (FS-VOF estimation and stochastic model estimation) can be used to reduce the RMSE and AAE. Moreover, the addition of the regularization term based on Helmholtz velocity decomposition in the VD-VOF estimation is more efective

Fig. 1 The ground-truth velocity felds for the analytic fows: **a** Poiseuille fow; **b** Lamb-Oseen vortex; **c** uniform fow; **d** sink fow; **e** vortex flow; **f** potential flow around a cylinder with circulation

Fig. 2 RMSE (**a**) and AAE (**b**) errors of velocity for analytic flow particle image sequences estimated by diferent methods: #01 Poiseuille fow; #02 Lamb-Oseen vortex; #03 uniform fow; #04 sink flow; #05 vortex flow; #06 potential fow around a cylinder with circulation

in reducing the error. However, in the third data set #03 uniform fow, the calculation accuracy of the VD-VOF method is slightly lower than that of the multiple-pass cross-correlation method. This is because there is no velocity gradient in the uniform fow, and the multiple-pass cross-correlation method based on the interrogation window will achieve smoother results. In later experiments, we demonstrate that our algorithm has comprehensive advantages in terms of resolution and accuracy over the multiple-pass cross-correlation method.

4.2 2D DNS turbulent fow

This synthetic particle image data set is a 2D, homogeneous, isotropic and incompressible turbulent flow, which is generated by direct numerical simulation of the Navier-Stokes equations at Reynolds number *Re* = 3000 and Schmidt number $Sc = 0.7$. The details of the simulation and synthetic particle generation are described in Carlier and Wieneke (2005) (2005) (2005) . This data set contains typical difficulties for PIV measurement methods such as high velocity gradients and a large dynamic range. It has been used by many researchers in the fuid measurement community, including in our previous work (Lu et al. [2019\)](#page-14-17). The data set consists of 100 successive images at the resolution of 256×256 pixels, and the maximum displacement between two successive images is approximately 3.5 pixels. Fig. [3](#page-10-0) demonstrates a particle image and the corresponding vorticity map with the groundtruth velocity feld.

The vorticity maps for a given instantaneous velocity field (at time $t = 50$) obtained by different methods are illustrated in Fig. [4](#page-11-1). The corresponding velocity vectors are superimposed on each vorticity map, and the vorticity maps share a common colour legend. The bicubic interpolation method is used to interpolate the estimated velocity feld of the multiple-pass cross-correlation method to obtain the same fne velocity feld as the optical fow method. The

vortex structures of diferent scales, which are comparatively blurred with the H $\&$ S R+S estimation and are slightly detected by the multiple-pass cross-correlation estimation and physics-based estimation, are substantially more accurately estimated with the FS-VOF estimation, stochastic model estimation and the proposed VD-VOF method such that the vortices have a similar behaviour to that of DNS. Upon carefully comparing the ground-truth vorticity map in Fig. [3](#page-10-0)b, at $x = 5$ pixel, $y = 250$ pixel, the FS-VOF and stochastic model estimations have an incorrect local vortex structure, but it does not exist in the vorticity map estimated by the VD-VOF method. We can observe from the vorticity maps that the proposed VD-VOF method performs better than the other methods, particularly in the area with high vorticity.

We present in Fig. [5](#page-11-2) the RMSE and AAE curves of the velocity for the 2D DNS particle image sequences estimated from diferent methods. This comparison result suggests that the proposed VD-VOF method outperforms the multiplepass cross-correlation and other advanced optical fow estimations. The turbulent energy spectra of these estimations are plotted in Fig. [6.](#page-12-0) All energy spectra generated by diferent methods are very close to the reference DNS spectrum at large scales. As clearly seen in Fig. [6,](#page-12-0) the proposed VD-VOF method performs much better than the other methods at small scales since the proposed method has introduced the velocity decomposition model and image-driven self-adaptive regularization parameter that can protect small-scale motions in the turbulent flow.

In order to further demonstrate the infuence and contribution of each module in the proposed VD-VOF method, the modifed physics-based estimation and the basic VD-VOF method were added as compared approaches. The modifed physics-based estimation comprises the frst two items in Eq. [22](#page-6-1), and the basic VD-VOF method is the form of Eq. [22](#page-6-1) without the regularization parameters $e^{-|\nabla I|^2}$ and BF_ω . As shown in Fig. [7](#page-12-1), both the regularization terms based on

Fig. 3 Illustration of a sample frame $(t = 50)$ from the 2D DNS synthetic image sequence: **a** particle image and **b** vorticity map with the ground-truth velocity feld

Fig. 4 Instantaneous vorticity feld for the 2D DNS turbulent fow data set at $t = 50$ estimated by different methods: **a** result of multiplepass cross-correlation; **b** result of H & S R+S estimation; **c** result of

physics-based estimation; **d** result of FS-VOF estimation; **e** result of stochastic model estimation; **f** result of the proposed VD-VOF method

Fig. 5 RMSE (**a**) and AAE (**b**) errors of velocity for the 2D DNS particle image sequence estimated by the diferent methods

Helmholtz decomposition and the adaptive regularization parameters have efects on reducing RMSE and AAE errors, among which the former contribute more signifcantly. From these experiments, we can conclude that the proposed VD-VOF method is very well adapted to the estimation of fuid flows with high vorticity and small-scale motion. In the next section, we present some experiments on real data of a hypersonic compressible flow.

5 Evaluations of experimental data

A Mach 5 hypersonic wind tunnel experiment was carried out to acquire the particle images of a compressible flow. The experiment was performed in the FD-30B hypersonic wind tunnel of the China Aerodynamics Research and Development Center (CARDC). The wind

Fig. 6 Spectrum analysis of the turbulent velocity for the 2D DNS particle image sequence, $E(k)$ denotes the energy at a given frequency *k*

tunnel was operated at a total pressure p_0 of 0.5 MPa and a total temperature T_0 of 350 K. A Quantel dual-cavity Nd:YAG-pulsed laser with a wavelength of 532 nm, a maximum energy of 430 mJ, a pulse rate of 10 Hz and a pulse duration of 60 ns was used. The particle images were recorded by a PowerView Plus 4MP-HS camera with a charge-coupled device (CCD) chip of 2048×2048 pixels and a Nikon AF Micro-Nikkor 200 mm f/4D prime lens, resulting in a visualized field of 100 mm \times 100 mm and a spatial resolution of 20 pixels/mm. The laser pulses were separated by $0.5 \mu s$, and a maximum of 200 image pairs were acquired in all experiments. A synchronizer with an accuracy of 0.25 ns ensured that the laser and the camera operated simultaneously. Titanium dioxide (TiO₂) particles with particle diameter 200 nm and relaxation time 0.41 μs were chosen for seeding.

A 40° compression ramp model was used to form complex flow field structures, such as the flow separation and shock wave/boundary layer interaction, as shown in Fig. [8](#page-12-2)a. Figure [8](#page-12-2)b is the average velocity feld calculated from 100 instantaneous velocity felds measured by commercial PIV software Insight 4G. The separation shock and the reattachment shock are represented by two white dotted lines, and the grey area indicates the compression ramp model. The hypersonic laminar flow has different stages such as boundary layer transition, fow separation and reattachment throughout the fow process. The shear layer forms and grows along the fow direction gradually, the shear layer surrounds the recirculation zone, and the free stream surrounds the shear layer. More details about the experimental design can be found in our previously published paper (Lu et al. [2019](#page-15-16)), and we omit that information here.

The instantaneous velocity felds and streamlines meas-ured by different methods are shown in Fig. [9.](#page-13-1) The flow structures of the hypersonic laminar flow over the 40° compression ramp are successfully captured through the diferent fluid flow estimations. The velocity field structures estimated

Fig. 8 The particle image (**a**) and average velocity feld (**b**) of hypersonic laminar flow over a 40◦ compression ramp

Fig. 9 Instantaneous velocity fields and streamlines of 40° compression ramp flow at Mach 5 estimated by different methods: a result of multiple-pass cross-correlation; **b** result of H & S R+S estimation;

by diferent methods are basically the same. The trend of the streamlines clearly indicates the separation and reattachment of the fow. The shear layer gradually forms and grows along the fow direction, and the shear structure and shearing degree increase. The shear layer surrounds the recirculation zone, and the free stream velocity outside the shear layer is uniform. Due to the infuence of the fow separation and shearing effect, the velocity direction is deflected and shows a tendency to coincide with the direction of compression of the flow channel.

The overall trend of the streamline refects both the recirculation motion in the separation zone and the reattachment of the fow on the compression ramp. In the result of multiple-pass cross-correlation, as shown in Fig. [9](#page-13-1)a, the thickness of the shear layer is too large because of the interrogation window. The recirculation motion is incomplete in the velocity feld measured by the H&S R+S and physics-based estimations, as shown in Fig. [9b](#page-13-1) and c, respectively. The separation shock and reattachment shock are not smooth in the result of stochastic model estimation, and the recirculation zone structure is compressed in the result of FS-VOF estimation. In comparison, the instantaneous velocity feld measured by the proposed VD-VOF method is closer to the average velocity feld in Fig. [8b](#page-12-2). Compared with other instantaneous velocity felds, the structures of the recirculation zone and shear layer in Fig. [9f](#page-13-1) is more complete, and the boundaries of the separation shock and reattachment shock are smoother. The non-quadratic penalty function and bilateral flter kernel function in the proposed method ensure that the shock position obtained by this method is consistent with that obtained by FS-VOF estimation. Similar to the synthetic data discussed in Sect. [4.2](#page-10-1), spectrum analysis is

c result of physics-based estimation; **d** result of FS-VOF estimation; **e** result of stochastic model estimation; **f** result of the proposed VD-VOF method

Fig. 10 Spectrum analysis of the velocity fields for 40° compression ramp flow

performed for the six processing methods and are shown in Fig. [10](#page-13-2). The spectrum obtained by the proposed VD-VOF method retains the inertial scales before reaching the energy dissipating range and is closer to the −5∕3 spectrum slope. From this experiment, we can conclude that the proposed VD-VOF method is very well adapted to the estimation of fluid flows with a compressible flow structure.

6 Conclusion

In this study, we propose a novel optical fow formulation by considering both particle image information and physics theory of fuid fow. According to the Helmholtz velocity decomposition theorem, the proposed optical fow method decomposes the two-dimensional velocity feld into four components: translation motion, linear distortion motion, shear distortion motion and rotation motion. In this context, regularization terms for diferent motion components are designed, which have a reasonable physical interpretation for the fow characteristics of the fuid. Subsequently, we design specifc regularization parameters for the corresponding regularization terms according to the fow characteristics of the motion components. These regularization parameters can be adaptively adjusted with changes in the image space and velocity feld. In addition, the data term of the optical flow formulation is based on the projected-motion equation derived from the continuity equation, which protects the compressibility of the fuid in the two-dimensional plane. A bilateral flter kernel function is used to suppress image noise and illumination changes in the experiment.

The proposed VD-VOF method is based on the continuity equation and Helmholtz decomposition theorem, considering both particle image information and physics theory of fluid flow. It is suitable not only for turbulent or incompressible flow but also for other flow conditions. To fully analyse the performance of the VD-VOF method, we analyse the comprehensive comparison results on synthetic and experimental image sequences using advanced cross-correlation and various optical fow approaches. For particle image data of analytic fows, the proposed VD-VOF method provides better results than previous advanced optical flow methods. For the 2D DNS turbulent fow particle image sequence, we present the efectiveness of our method for turbulent fow felds, and it is shown that the proposed velocity decomposition formulation outperforms some of the high-accuracy methods in the literature. Finally, we capture a compression ramp flow at a Mach 5 hypersonic flow field, and we demonstrate the good performance of the proposed VD-VOF method for a hypersonic compressible fow.

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