### **RESEARCH ARTICLE**



# **Internal fows of ventilated partial cavitation**

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# **Abstract**

Our study provides the frst experimental investigation of the internal fows of ventilated partial cavitation (VPC) formed by air injection behind a backward-facing step. The experiments are conducted using fow visualization and planar particle image velocimetry with fog particles for two diferent cavity regimes of VPC, i.e., open cavity (OC) and two-branch cavity (TBC), under various ranges of free stream velocity (*U*) and ventilation rates (*Q*). Our experiments reveal similar fow patterns for both OC and TBC, including forward fow region near the air–water interface, reverse fow region, near-cavitator vortex, and internal fow circulation vortex. However, OC internal fow exhibits highly unsteady internal fow features, while TBC internal fow shows laminar-like fow patterns with a Kelvin–Helmholtz instability developed at the interface between forward and reverse fow regions within the cavity. Internal fow patterns and the unsteadiness of OC resemble those of turbulent fow separation past a backward-facing step (BFS fow), suggesting a strong coupling of internal fow and turbulent external recirculation region for OC. Likewise, internal fow patterns of TBC resemble those of laminar BFS fow, with the presence of unsteadiness due to the strong velocity gradient across the forward–reverse fow interface. The variation in the internal fow upon changing *U* or *Q* is further employed to explain the cavity regime transition and the corresponding change of cavity geometry. Our study suggests that the ventilation control can potentially stabilize the cavity in the TBC regime by delaying its internal fow regime transition from laminar-like to highly unsteady.

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#### **Graphic abstract**



Instantaneous tracer streak image



# **1 Introduction**

Ventilated partial cavitation (VPC) is an air pocket that is artifcially formed by gas injection behind a fow separation device or cavitator, in a liquid flow. Such phenomenon has been applied to air lubrication on ship operation for friction drag reduction (Ceccio [2010](#page-13-0)), also known as partial cavity drag reduction (PCDR). When properly designed, PCDR has been shown to achieve up to 95% drag reduction for the covered area and 20% energy savings (Mäkiharju et al. [2012\)](#page-14-0).

To utilize this promising technique on energy savings for ship operations, extensive work has been done on PCDR, mainly focusing on the cavitator design and establishing a relationship between relevant fow conditions and ventilation cost. For instance, Amromin et al. [\(2006](#page-13-1)) and Kopriva et al.  $(2008)$  $(2008)$  measured the lift and drag coefficients on a hydrofoil cavitator to validate the efectiveness of cavitation theory for drag reduction schemes. Makiharju et al. ([2010\)](#page-14-1) also used imaging techniques on cavity closure to determine the cavity shape, growth rate and collapse rate, closure oscillation, and closure type and observed the efect of each fow condition on the ventilation cost. To ensure the industrial applicability of PCDR, Mäkiharju et al. [\(2013](#page-14-2)) later conducted both large- and small-scale experiments to examine the scaling factors of relevant flow parameters to the required ventilation rate for a partial cavity to be sustained under various flow parameters.

In addition to studies on PCDR, studies on cavity shape and topology have also been conducted. Laberteaux and Ceccio ([2001\)](#page-14-3) used particle image velocimetry (PIV) and particle streak photography to investigate the efects of external flow on cavity geometry and air entrainment of a natural partial cavity formed behind a wedge. Barbaca et al. [\(2017](#page-13-3)) and Barbaca et al. ([2018\)](#page-13-4) used bottom view imaging of a VPC that formed behind a 3D and 2D fence type cavitator, respectively, to classify the cavity topology based on gas leakage mechanisms. They also performed pressure measurements to calculate the cavitation number in each regime. Most recently, using high-speed imaging on the side and bottom views, Qin et al. ([2019\)](#page-14-4) have investigated VPC under various flow conditions and ventilation rates and classifed cavity regimes based on gas leakage mechanisms and explained the transitions between regimes. They further suggested the importance of understanding the role of internal fow in a gas leakage mechanism. They pointed out that gas leakage occurs as the ventilated gas inside the cavity is dragged by the internal boundary layer and dispersed into bubbles by the external recirculating vortex that is formed behind the closure of the cavity. However, no prior studies have provided the internal fow characterization of VPC. Thus, the purpose of this study is to experimentally investigate the internal flow of VPC to further understand the ventilation physics of VPC.

In contrast to the dearth of internal fow studies in VPC, only two studies have experimentally examined the internal flow of ventilated supercavitation. Wang et al. ([2018\)](#page-14-5) conducted internal fow PIV on a 2D cylinder cavity by seeding fuorescent-painted glass microspheres and acquired timeaveraged velocity feld. However, the spatial resolution of the acquired flow field was not sufficient enough to reveal flow structures that occur in the instantaneous internal flow feld. Wu et al. [\(2019](#page-14-6)) used the ray tracing mapping method that enabled distortion correction of a curved surface to perform high-resolution internal flow PIV with fog particles seeded. The resolution of their experiments was sufficient enough to observe the coherent structures and to provide quantitative measurements for supercavity with diferent closure types. They not only provided information on the internal boundary layer that was not available in the literature but further provided a detailed explanation on cavity evolution, closure formation, gas leakage mechanism based on the coupling of internal and external fows.

Employing a similar approach to that of Wu et al. [\(2019](#page-14-6)), the current study presents an experimental investigation on the internal fow of VPC using fow visualization and PIV. This study aims to characterize the internal fow feld under distinct cavity regimes and investigate the connection of the internal fow with general cavity behaviors (i.e., cavity length growth and cavity regime). The remaining sections are listed as follows: Sect. [2](#page-2-0) provides detailed description of the experimental methods. The results from our study are presented in Sect. [3](#page-5-0). Specifcally, Sect. [3.1–](#page-5-1)[3.2](#page-8-0) reports the results of the internal flow visualization and the mean flow feld of each cavity regime. A comparison is made between the VPC internal fow and the single-phase fow past a backward-facing step in Sect. [3.3](#page-10-0) to further explain the cavity behavior upon changes in flow parameters and regime transitions. Section [4](#page-12-0) provides a summary of the current study.

# <span id="page-2-0"></span>**2 Experimental methodology**

# **2.1 Experimental facility and setup**

The experiments are conducted in the high-speed cavitation water tunnel at Saint Anthony Falls Laboratory (SAFL), University of Minnesota. The tunnel is a closed recirculating tunnel with a large volume dome-shaped settling chamber designated for fast bubble removal during ventilation experiments. For the past few years, the SAFL water tunnel has been used for a number of ventilated cavitation experiments (Karn et al. [2015,](#page-13-5) [2016](#page-13-6); Jiang et al. [2018](#page-13-7)). The free stream velocity is calculated through a Rosemount DP sensor that measures the diferential pressure between the settling chamber and test section. Correspondingly, the free stream velocity is regulated by a feedback control of the motor speed of the water tunnel so that the measured velocity meets the desired velocity. The dimension of the test section is 1.20 m  $\times$  0.19 m  $\times$  0.19 m (length, height, and width), and the bottom and two side windows of the test section are made up of Plexiglas for optical access. The center of the test section ceiling is also transparent for the PIV experiment. A detailed description of the experimental facility can be found in Karn et al. ([2016\)](#page-13-6). The 3D printed backward-facing step cavitator with a height of 16 mm is installed on the ceiling of test section as shown in Fig. [1a](#page-3-0), which leads to an expansion ratio (ER) of 1.09 and an aspect ratio of 11.9. As shown from the back view, the cavitator has an injection hole with a uniform thickness of 10 mm between each test section wall and the step edge. VPC is generated by ventilating gas behind a backward-facing step cavitator during the operation of the water tunnel.

Figure [1b](#page-3-0) depicts the experimental setup for the PIV experiment. The transparent center portion of the test section ceiling enables the projection of laser sheet from the top in order to ensure that the laser sheet is not distorted by the air–water interface. The laser sheet thickness is calibrated to be  $1 \pm 0.2$  mm. Particles generated from the Rosco 1700 fog machine act as a tracer that are frst flled in 42 gallons for a chamber where the ventilation line is connected so that the particles are laden in the air and injected into the test section. The mass flow rate of the ventilated air is controlled by an Omega Engineering FMA-2609A mass fow controller that has a proportional–integral–derivative algorithm for controlling at a constant desired fow rate up to 55 SLPM with the uncertainty within  $\pm$  0.2%FS. The fog particle size is estimated to be  $29 \pm 8$  µm. A more detailed explanation on the VPC setup can be found in Qin et al. [\(2019](#page-14-4)). The seeding procedures and the calculations of the particle size and laser sheet thickness are detailed in Wu et al. ([2019\)](#page-14-6). A Photonics DM30-527 Nd/YLF high-speed laser (maximum pulse energy of 30 mJ/pulse) is synchronized with the time resolution of Photron APX-RS high-speed camera (full sensor size of  $1024 \times 1024$  pixels) to perform time-resolved PIV measurements.

# **2.2 Selection of experimental conditions**

Among the four cavity regimes (i.e., foamy cavity, transition cavity, open cavity, and two-branch cavity) reported by Qin et al. ([2019](#page-14-4)), open cavity (OC) and two-branch cavity (TBC) are selected for the internal fow measurement. It is worth noting that the internal fow measurement is only viable for the OC and TBC regimes due to the presence of the large air pocket with a smooth air–water interface, whereas the other cavity regimes consist of dispersed bubbles.

Table [1](#page-3-1) shows the experimental conditions for the VPC internal fow measurement. Experimental conditions are selected using the cavity regime map of Qin et al. [\(2019\)](#page-14-4)



<span id="page-3-0"></span>**Fig. 1** Schematics of **a** the test section with backward-facing step (BFS) cavitator mounted on the ceiling, and **b** the experimental setup. The shaded region in the back view of the cavitator indicates the ventilation hole. During the experiments, water fows in *x* direction. For the imaging setup, the measurement feld of view covers from the rear

part of the cavitator to the closure of the internal fow feld. The time resolution varies between 2000 and 2500 Hz, and the spatial resolution varies between 32.79 μm/pixel and 56.82 μm/pixel, depending on the cavity regimes and the free stream velocities

<span id="page-3-1"></span>



*w* stands for the width of the test section

to cover a wide range of fow and ventilation settings for OC and TBC regimes. The cavity with various free stream velocities  $(U)$  and ventilation rates  $(Q)$  is examined to evaluate the effect of ventilation coefficients  $(C<sub>O</sub>)$  on the internal flow. All experimental conditions are referred to by their case numbers. As shown in Table [1](#page-3-1), for OC, we only investigated the experimental conditions with diferent *Q* that is precisely controlled (in precision of two decimal places) by the proportional–integral–derivative control algorithm embedded in the mass fow controller. In the current experiment, the investigation of OC at higher Froude number (Fr) is limited by the substantial cavity length fuctuation and interface distortion occurring with a further increase in Fr which prohibits accurate internal flow PIV measurements.

For TBC, various sets of  $C_Q$  are chosen to examine its efect on the internal fow under diferent *U* and *Q* with the remaining parameters fxed. However, it should be noted that Fr for the TBC regime is selected to be as low as possible since the cavity length of TBC is in the order of  $Fr<sup>2</sup>$  and that the cavity extends beyond the test section for Fr over 3.4 (Qin et al. [2019](#page-14-4)).

#### <span id="page-4-1"></span>**2.3 Measurement settings and data processing**

We conducted PIV measurements on the streamwise wallnormal plane (*xy*-plane) at the center spanwise (*z*) location of the channel. The PIV recording settings for each case are summarized in Table [2](#page-4-0). For each PIV case, the corresponding cavity length is measured using the calibrated cavity images. It should be noted that the cavity length defned for OC is the distance from the cavitator to the streamwise location where air entrainment occurs, which is uniform along the spanwise  $(z)$  direction when time averaged. The cavity length of TBC is defned as the distance from the cavitator to the bifurcation point, since TBC has a bifurcation point at the center *xy*-plane with two branches near the wall of the test section that are elongated downstream (Qin et al. [2019](#page-14-4)).

Figure [2](#page-5-2)a shows a sample image of the tracer feld illustrating the measurement feld of view (FOV) and image quality. The corresponding instantaneous velocity field is presented in Fig. [2b](#page-5-2), illustrating the internal fow feld

<span id="page-4-0"></span>**Table 2** Spatial and temporal resolution of the experiment

Case no.	Spatial resolution $(\mu \text{ m/pixel})$	fps $(Hz)$	Sampling time(s)
1	56.82	2500	1.64
$\overline{c}$	56.82	2500	1.64
3	32.79	2000	2.05
$\overline{4}$	32.79	2000	2.05
5	32.79	2500	1.64
6	32.79	2500	1.64

measured from the PIV calculation. For clarity, only one in two vectors is plotted in both *x* and *y* directions. The FOV for the OC internal flow PIV (1024  $\times$  512 pixels) is chosen to capture the entire fow feld at once. Tracer images are processed using an adaptive multi-pass cross-correlation algorithm in *LaVision DaVis* 8.2 to obtain instantaneous velocity vector felds. The interrogation window starts from  $96 \times 96$  pixels with a 50% overlap, decreasing to  $32 \times 32$ pixels with 75% overlap. The resulting parameters yield the spacing between vectors of 450.6 μm/vector. As shown in Fig. [2a](#page-5-2), b, highlighted with red dotted line, the region near the closure cannot yield reliable vectors due to the unsteady closure. Such spurious vectors (closure region of Fig. [2b](#page-5-2)) are generated due to the non-uniform refection of the light sheet that illuminates the unsteady closure. To cope with the cavity length fuctuations (Fig. [2](#page-5-2)d) and the spurious vectors beyond the closure, we have set a threshold based on the standard deviation of velocity vectors (Fig. [2](#page-5-2)c) which results in the reliable measurement feld of view of PIV covering 85% of the cavity length. To acquire the mean internal flow feld of OC from the instantaneous fow felds with fuctuating length, the horizontal axis of each instantaneous fow feld is normalized by the corresponding cavity length, and then, the normalized fow felds are averaged.

Due to the extended cavity length of TBC, the internal flow of TBC is recorded with separate FOVs of  $1024 \times 1024$ pixels, with the spatial interval of 3 cm to ensure the presence of the overlapping region between FOVs. Figure [3a](#page-6-0) shows the sample image of the tracer feld illustrating the measurement feld of view and image quality. The corresponding instantaneous velocity feld is presented in Fig. [3b](#page-6-0). For clarity, only one in two vectors is plotted in both *x* and *y* directions. For TBC, the interrogation window for PIV starts from  $96 \times 96$  pixels with a 50% overlap to  $32 \times 32$ pixels with 50% overlap. The resulting parameters yield the spacing between vectors of 524.6 μm/vector. To obtain the entire mean flow of TBC from separate velocity vector fields (Fig. [3](#page-6-0)b), each FOV is combined with a weighted average of distance for the overlapping regions. The heights of adjacent FOVs are matched to yield the minimum mean squared error between the two vector matrices in each overlapping region.

For all experimental conditions, temporal resolutions are designed to ensure that the smallest interrogation window  $(32 \times 32 \text{ pixels})$  captures forward-moving tracers driven by an air–water interface within 10 pixel displacement per frame. The sampling time for each experimental condition is chosen to be long enough to ensure statistical convergence. Here we assume that the recorded data are statistically converged when the sampling time is more than three times greater than the time required for the reverse fow to be fully advected from the closure to the cavitator. The reverse fow velocity for calculating the sampling time was empirically determined by preliminary experiments, being roughly a



<span id="page-5-2"></span>**Fig. 2** Samples of **a** the tracer feld image and **b** the corresponding instantaneous velocity vector feld superimposed with the streamwise velocity contour map showing the internal fow measurements for VPC in the OC regime (Fr = 3.0 and  $C_Q = 3.20 \times 10^{-3}$ ). The red dotted lines of **a** and **b** mark the location of the air–water interface of an instantaneous sample at the closure. For clarity, only one in two vectors is plotted in both directions. The internal fow region of OC

half of the forward fow velocity driven by the air–water interface.

# <span id="page-5-0"></span>**3 Results**

# <span id="page-5-1"></span>**3.1 OC internal fow feld**

Flow visualization based on tracer streak images provides a qualitative view on the fow features of the OC internal fow.

is generated by removing the spurious vectors beyond the closure by setting the upper threshold based on **c** the normalized standard deviation of velocity vectors. **d** 85% of the mean cavity length is chosen as a threshold to ensure that the internal fow feld is not afected by unstable closure. The cavity length fuctuations of case 1 and 2 are colored in red and blue lines, respectively. Dotted lines correspond to 85% of the mean values

The tracer streak images are generated with a maximum intensity projection of stack of ten tracer images. As shown in Fig. [4](#page-6-1) (fow visualization of case 1) and supplemental video S1 (fow visualization of case 2), the OC internal flow field exhibits distinct regions: the forward flow region, reverse flow region, near-cavitator vortex, and internal flow circulation vortex near the closure. The forward fow region is driven by the moving air–water interface at the speed of a free stream velocity. The reverse fow region, driven by an adverse pressure gradient, covers a large portion of the



<span id="page-6-0"></span>**Fig. 3** Samples of **a** tracer feld images and **b** the corresponding instantaneous velocity vector felds superimposed with streamwise velocity contour maps showing the internal flow measurements for

VPC in the TBC regime (Fr = 1.6 and  $C_Q = 8.44 \times 10^{-3}$ ). For clarity, only one in two vectors is plotted in both directions



<span id="page-6-1"></span>**Fig. 4** Tracer streak image of case 1 (Fr = 3.0 and  $C_Q = 3.20 \times 10^{-3}$ ) showing the internal flow of OC. Clockwise near-cavitator vortex, counterclockwise internal fow circulation vortex, and intermittent transverse fow are commonly observed in tracer streak images

internal fow feld, and the yellow dashed line marks the boundary between the forward and reverse fow regions. A gradual increase in the internal boundary layer thickness is observed along the air–water interface. Near-cavitator vortex refers to the region of the fow circulation near the cavitator, which is a result of the interaction between forward-moving ventilation jet and the reverse fow. It is noteworthy that the shape and location of the near-cavitator vortex resemble the

*secondary vortex* in a single-phase flow past a backwardfacing step, but its property relies on the ventilation coefficient  $(C<sub>O</sub>)$ , which depends both on  $Q$  and  $U$ . The internal flow circulation vortex near the closure is formed by the interaction between the forward and reverse fows. While Qin et al. ([2019\)](#page-14-4) hypothesized that the gas leakage mechanism of OC (i.e., the dispersal of small bubbles from the air pocket) is primarily driven by the external recirculation vortex, our internal fow visualization reveals that the dynamics of internal fow circulation also contributes to the gas leakage of OC. Specifcally, the presence and the movement of the internal fow circulation vortex infuence the stability of the air–water interface, which ultimately leads to the intermittent shed off of the air pocket at the closure. Remarkably, as shown by the short tracer streak patterns in Fig. [4](#page-6-1) and supplemental video S1, our internal fow visualization also suggests the presence of intermittent transverse fow in the reverse fow region. These out-of-plane motions can be identifed by rather shorter tracer streaks with their direction diferent from that of the neighboring fows. Such transverse fow may be caused by the non-uniform pressure distribution along the spanwise direction associated with the unsteady cavity closure. It should be noted that these motions do not signifcantly afect the mean fow statistics presented in the current paper, since even near the closure such out-of-plane motions are observed intermittently while the majority of the recorded sequence exhibits in-plane motions.

To provide a quantitative view of the internal fow feld, the mean internal fow felds under constant *U* with diferent *Q* are investigated where such flow fields are generated by following the procedure mentioned in Sect. [2.3.](#page-4-1) Figure [5](#page-7-0) demonstrates the mean internal fow feld for case 1 with the streamwise velocity and vorticity contours superimposed with streamlines. The streamline pattern of case 1 clearly

diferentiates the distinct fow regions: the forward and reverse fow regions, near-cavitator vortex, and the internal flow circulation vortex. The distance from the cavitator to the center of the internal fow circulation vortex coincides with the mean cavity length  $(x = L_c)$  where the air film (air–water interface) is dispersed into bubbles, suggesting the role of internal fow circulation vortex in the gas leakage mechanism of OC. The increase in internal boundary layer thickness with streamwise location can be observed from the streamwise velocity contour (Fig. [5](#page-7-0)a), which is consistent until the streamlines become disconnected at the closure. The vorticity contour map (Fig. [5](#page-7-0)b) shows two strong shears in the mean fow feld, one corresponding to the forward fow that is dragged by the moving air–water interface and another one corresponding to the reverse fow with highvorticity shear near the ceiling. In addition, the vorticity map clearly shows the presence of the near-cavitator vortex and the internal fow circulation vortex, with the latter being signifcantly stronger and having an opposite sign of vorticity compared to the former. This trend is consistent with the general pattern observed from the instantaneous sample (Fig. [4](#page-6-1)). However, it is worth noting that the near-cavitator vortex on the mean vorticity contour map appears to be a little distorted and has a weaker strength compared to the observation from the instantaneous sample. Such discrepancy between the instantaneous and the averaged results suggests a fuctuation of the centroid location and the strength of the near-cavitator vortex potentially associated with the unsteadiness of the internal fow.

Figure [6](#page-8-1) examines how the change of  $C_Q$  due to an increased  $Q$  affects the internal flow of OC. It is important to note that during the OC experiments a small portion of the ceiling is occasionally covered by water droplets generated due to the unavoidable breakup of the air–water interfaces.



<span id="page-7-0"></span>**Fig. 5 a** Streamwise velocity and **b** vorticity contour maps superimposed with streamlines for case 1 (Fr = 3.0 and  $C_Q = 3.20 \times 10^{-3}$ )



<span id="page-8-1"></span>**Fig. 6** Vorticity contour maps superimposed with streamlines for case 2 (Fr = 3.0 and  $C_Q = 4.11 \times 10^{-3}$ ). The red dotted lines indicate the region that cannot be resolved due to the condensed droplets that block the optical access as mentioned in Sect. [2.3](#page-4-1)

Although the presence of these droplets blocks the tracer signal in some small regions near the ceiling, we have ensured that their presence does not distort the laser sheet within the recorded time sequence. Such regions are excluded during the PIV calculation to eliminate the spurious vectors associated with the presence of the droplets. Such regions are marked with red dotted lines in Fig. [6](#page-8-1). As the figure shows, besides an increase in the mean cavity length as reported by Qin et al.  $(2019)$  $(2019)$  $(2019)$ , the general features of internal flow of OC, i.e., the presence of forward and reverse fows as well as the near-cavitator and internal fow circulation vortices, remain unchanged with increasing *Q*. Particularly, the center of the internal flow circulation vortex stays around  $x = L_c$  regardless of ventilation rates. However, the near-cavitator vortex shows an increase in strength and size with increasing *Q*. Additionally, the region of internal fow circulation appears to be more confned to the region near the closure.

# <span id="page-8-0"></span>**3.2 TBC internal fow feld**

Flow visualization with tracer streak images of TBC provides qualitative view on the internal fow structure at the center plane. Tracer streak images of TBC are generated following the same method as for the OC cases. Figure [7a](#page-8-2) illustrates the snapshot of tracer streaks of each FOV that is characterized by case 3. Similar to OC internal fow, tracer streak images of TBC enable a clear observation of the forward and reverse fow regions, the near-cavitator vortex, and the internal fow circulation vortex. However, as shown from supplemental videos S2–S6, the internal flow structure of TBC is more laminar-like and steady over time, compared to the instantaneous fow visualization of OC. Particularly, the intermittent transverse fow occurring in the reverse fow region is no longer observed in TBC due to the absence of unsteady closure (see supplemental video S6).

However, as shown in Fig. [7b](#page-8-2), the reverse fow driven by the adverse pressure gradient is still unsteady, in spite of the steady cavity length with the air–water smoothly attached to the ceiling at the center  $xy$ -plane. The figure shows the stagnant fow regions marked by the short tracer streaks slowly advecting upstream and inducing wavy pattern of the reverse



<span id="page-8-2"></span>**Fig. 7 a** Compilation of the tracer streak images at diferent streamwise locations case 3 (Fr = 1.6 and  $C_Q = 8.44 \times 10^{-3}$ ) illustrating the presence of distinct fow regions inside TBC. Yellow dotted lines mark the boundary between forward and reverse fows. **b** Sequence of tracer streak images showing the presence of KH instability in the

internal fow region. Yellow dotted circles mark the stagnant region of internal fows that is advecting upstream, providing an evidence of KH instability between forward and reverse fow regions. Such stagnant regions moving upstream cause the reverse fow to have an oscillatory motion

flow. Such unsteadiness suggests the presence of an instability in the internal flow region that affects the instantaneous flow structure of TBC. In particular, we believe that such instability is related to the Kelvin–Helmholtz (KH) instability, triggered by the strong velocity gradient across the interface of forward and reverse fows (see supplemental video S7 for clear visualization of such behavior).

Mean fow felds of TBC are investigated to provide a quantitative view of the internal fows. The internal fow feld of TBC is acquired by combining the mean fow felds of separate FOVs into a single fow feld. Similar to the qualitative observation of TBC internal flow, the velocity contour map of case 3 (Fig. [8a](#page-9-0)) shows the four distinct fow regions of TBC: the near-cavitator vortex, forward and reverse fow regions, and the internal flow circulation vortex. Moreover, besides the substantial increase in cavity length compared to OC, a small forward fow region near the cavitator can be clearly observed from the velocity contour map, due to the relatively higher  $C_Q$  for TBC cases. The streamline pattern of TBC is diferent from that of the OC near the closure as the internal fow circulation vortex is fully connected, implying no interaction between internal and external fows at the center *xy*-plane that infuences the stability of the air–water interface near the closure. The internal boundary layer thickness of TBC grows with streamwise direction to a certain extent, but then decreases due to the thinning of the cavity thickness near the closure. In addition, near the closure, reverse flow velocity appears to have the highest magnitude, and the slight attenuation of the forward flow is also observed. Such behaviors suggest the (spanwise) bifurcation of the internal fows due to the reduction in the cross-sectional area. The vorticity contour map from Fig. [8](#page-9-0)b shows the two strong shear regions with high vorticity near the air–water interface and near the ceiling, respectively. In addition, unlike for OC, the near-cavitator vortex is more clear from the vorticity contour map. An observation of the stronger vorticity at the near-cavitator vortex also indicates that the internal fow of TBC is more laminar-like compared to OC cases with appreciable ventilation jet near the cavitator. The strength of the near-cavitator vortex is still weaker compared to that of the internal fow circulation vortex with opposite sign of vorticity.

Further investigation of the TBC internal flow is conducted with different *U* and *Q* to examine the effect of  $C<sub>O</sub>$ on the internal fow structures of TBC. Figure [9](#page-10-1)a presents the vorticity contour map superimposed with streamlines of case 4, which represents the increased *Q* (from 1 SLPM to 5 SLPM). As shown in the fgure, the cavity length of TBC is invariant under the change of the ventilation rate, which is consistent with the fndings reported by Qin et al. [\(2019](#page-14-4)). However, the internal flow structure near the cavitator differs with case 3 and the amplifcation of the near-cavitator



<span id="page-9-0"></span>**Fig. 8 a** Streamwise velocity and **b** vorticity contour map superimposed with streamlines of TBC case 3 (Fr = 1.6 and  $C_Q = 8.44 \times 10^{-3}$ )



<span id="page-10-1"></span>**Fig.** 9 Vorticity contours of **a** case 4 (Fr = 1.6 and  $C_Q = 4.22 \times 10^{-2}$ ) that represents an increase in  $Q$  from case 3, **b** case 5 (Fr = 2.0 and  $C_Q = 6.85 \times 10^{-3}$ ) that represents an increase in Fr from case 3 and

vortex with an increase in *Q* is clearly observed. The internal fow circulation vortex also exhibits strengthened vorticity with an increase in  $Q$ . Furthermore, the effect of  $C<sub>Q</sub>$  on the TBC internal flow with higher *U* (case 5) is visualized by the shrinkage in the size of the near-cavitator vortex (Fig. [9](#page-10-1)b). Figure [9](#page-10-1)c shows the consistency of the effect of  $C<sub>Q</sub>$  on the internal fow of TBC under diferent *U* and *Q*. The size of the near-cavitator vortex of case 6 (Fig. [9c](#page-10-1)) is enlarged compared to case 5 (higher *Q* with constant *U*) similar to the aforementioned comparison between diferent *Q* (i.e., Figs. [8](#page-9-0)b, [9a](#page-10-1)) and decreased compared to case 4 (higher *U* with constant *Q*) similar to the aforementioned comparison between diferent *U* (i.e., Figs. [8](#page-9-0)b and [9b](#page-10-1)). The strength of internal fow circulation vortex is also observed to be strengthened with increasing *Q*. Disconnected isocontours in the vorticity feld with increased *U* suggest an increase in the unsteadiness of the internal fow structure that is discussed

**c** case 6 (Fr = 2.0 and  $C_Q = 3.43 \times 10^{-2}$ ) that represents an increase in both Fr and *Q* from case 3. Each vorticity contour is superimposed with streamlines of the corresponding case

in Fig. [7b](#page-8-2). The location of the internal fow circulation vortex, however, appears to be located in similar regions despite the change in *U* or *Q*.

# <span id="page-10-0"></span>**3.3 Comparison with fow over a backward‑facing step**

To further understand the internal flow field of VPC, we provide a comparison of VPC internal flows with a singlephase flow over a backward-facing step (BFS), which is referred to as BFS flow hereafter. Since Nadge and Govardhan  $(2014)$  $(2014)$  noted that the expansion ratio (ER) greatly influences the reattachment length  $(X<sub>r</sub>)$  behavior among other variables (e.g., aspect ratio, free stream turbulent intensity, boundary layer state, and its thickness, etc.), only the studies with *ER <* 1.2 are selected and are tabulated in Table [3](#page-11-0) for comparing with our VPC internal fow cases which uses



<span id="page-11-1"></span>**Fig. 10** Reattachment lengths  $(X_r/h)$  from the literature of low expansion ratio and varying Reynolds number based on step height. Dotted line indicates the trend of the *X*r∕*h* of the single-phase fow separation with increase in Re<sub>h</sub>. Symbols: ■ Beaudoin et al. [\(2004](#page-13-8)); ▲ Goldstein et al. [\(1970](#page-13-9));  $\blacktriangledown$  Boiko et al. ([2011\)](#page-13-10);  $\blacktriangleright$  Etheridge and Kemp ([1978\)](#page-13-11); ◂ Kostas et al. [\(2002](#page-14-10)); ⧫ Scarano et al. [\(1999](#page-14-11)); ∙ Nadge and Govardhan  $(2014)$  $(2014)$ ; + Ma and Schröder  $(2017)$  $(2017)$ ;  $\times$  Jovic and Driver ([1995\)](#page-13-12);  $\Box X_r/h$  of external recirculation region of OC from Qin et al. ([2019\)](#page-14-4);  $\circ$  TBC length from the present study. Note that the *X<sub>r</sub>*/*h* of single-phase fows is marked with flled symbols, and the data points corresponding to VPC studies of ours and Qin et al. ([2019\)](#page-14-4) are marked with hollow symbols

a cavitator corresponding to a ER of 1.09. Figure [10](#page-11-1) illustrates the trend of  $X_r/h$  for BFS flow upon the change in Re<sub>h</sub>. As shown in the figure, for BFS flow, its  $X_r/h$  increases approximately linearly when the fow is laminar, from 3 to 20. Then, it drops abruptly to about 5 as the fow transitions from laminar to turbulent at a critical  $\text{Re}_{h}$  of around 2000. With further increase in  $\text{Re}_{h}$ ,  $X_r/h$  grows at a slower rate in comparison with that in the laminar regime, with a value ranging from 5 to 7 under  $Re_h$  of 3000–25,500.

To compare with BFS flow, a proper definition of  $Re<sub>h</sub>$ for VPC internal fow needs to consider the properties of both air and water phases since it is the interplay between the two phases that dictates the overall behavior of the cavity. Remarkably, as the schematic shown in Fig. [10](#page-11-1), in OC regime, the *X*r∕*h* of the external recirculation region outside the cavity and its variation upon  $Re<sub>h</sub>$  are comparable to that of BFS flow in turbulent regime under the same  $Re_h$  when it is determined using the water viscosity and the cavity thickness at the closure of OC  $(h_c)$ . This result suggests that the external recirculation region of OC is turbulent, consistent with the assessment from Qin et al. [\(2019\)](#page-14-4). Moreover, such turbulent external recirculation results in a fuctuation of the cavity length of OC (evidenced in Fig. [2](#page-5-2)d) up to one step height (*h*). This phenomenon resembles the fuctuation of the  $X_r$  in the turbulent BFS flow within a range of 3*h* (Schäfer et al. [2009](#page-14-8); Ma and Schröder [2017\)](#page-14-9), suggesting a strong coupling of the internal air flow with the water flow in the external recirculation region. For turbulent BFS fow, the fluctuation of  $X_r/h$  is attributed to the onset of KH instability that triggers the fapping motion in the separated shear layer (Ma and Schröder [2017](#page-14-9)). Accordingly, the presence of KH instability in the separated shear layer extended from the cavity interface may be responsible for the cavity length fuctuation of OC. In contrast, in TBC regime, the cavity length and its growth trend upon  $\text{Re}_{h}$  is found to match those of  $X_r/h$  of laminar BFS flow under the same Re<sub>h</sub> based on the viscosity of air. Such matching implies that the cavity behavior of TBC is dictated by the internal air fow and tends to be decoupled from the influence of external flow disturbance as the air–water interface is smoothly attached to the ceiling at the closure with an absence of the external recirculation region.

In accordance with the cavity geometry, the internal flow felds of VPC are also compared with the BFS fows. Specifcally, for OC, its internal mean fow resembles the turbulent BFS flow in terms of flow structures. For instance, the behavior of the internal flow circulation vortex and near-cavitator vortex of OC is similar to the recirculation vortex and

<span id="page-11-0"></span>**Table 3** List of experimental research on single-phase fow separation over a backwardfacing step with relevant expansion ratio (ER) and Reynolds number based on step height  $(Re<sub>h</sub>)$ 



secondary vortex of BFS flow, respectively. In addition, both flow fields show a maximum reverse flow velocity occurring at the center of the circulation vortex, i.e., the internal flow circulation vortex of OC (Fig.  $5a$ ) and the recirculation vortex of turbulent BFS flow (Ma and Schröder [2017](#page-14-9)). To provide a better comparison between the fow structures, we introduce the velocity ratio of the forward fow and the reverse flow along the same velocity profile at each streamwise location. It should be noted that the maximum of this ratio ( $r_{FR, max}$ ) yields 0.15 for OC internal flows and 0.14 for turbulent BFS fow (Ma and Schröder [2017\)](#page-14-9). Moreover, the unsteady behaviors of OC internal flow resemble those of turbulent BFS flow. As it can be observed from supplemental video S1, the internal flow circulation vortex of OC exhibits intermittent breakdown and its advection toward upstream. Such behavior is referred to as the convective instability, since the amplifed disturbance diminishes in its origin and convects to other locations (Huerre and Monkewitz [1990](#page-13-13)). Remarkably, such convective instability also occurs in the turbulent BFS fow but is manifested as KH instability in the separated shear layer shown as intermittent vortical structures advecting downstream. As a result, it is conceivable that the unsteady behavior of the internal fow circulation vortex may be infuenced by the presence of the unsteadiness within the separated shear layer in the external recirculation region of OC, which provides further support to the strong coupling of the internal and external fows that determines the cavity length fuctuation of OC.

Correspondingly, the internal fow of TBC shows some similarities to the BFS flow in the laminar regime. As observed in supplemental videos S2–S6, the internal fow structure of TBC is laminar-like, despite that some unsteadiness is observed within the reverse fow region (see Fig. [7](#page-8-2)b). Such unsteadiness may be attributed to the KH instability triggered by the strong velocity gradient between forward and reverse fows, as mentioned in Sect. [3.2](#page-8-0), which has not been reported in the laminar BFS flow studies (Goldstein et al. [1970](#page-13-9); Boiko et al. [2011\)](#page-13-10). Such velocity gradient can be characterized by  $r_{FR, max}$  defined earlier, which turns out to be roughly 0.40 for TBC cases, considerably higher than  $r_{FR, max}$  of the laminar BFS flow that has been reported as around 0.05 (Boiko et al. [2011](#page-13-10)). Such high  $r_{\text{FR,max}}$  (see Fig. [8a](#page-9-0)) near the closure, mainly due to the reduced crosssectional area and the bifurcation of the internal gas flow, results in stronger velocity gradient for TBC that triggers the KH instability. It is noteworthy that the location of the maximum reverse fow velocity is not directly comparable to the flow structure of the laminar BFS flow. However, such phenomenon clearly points out the diference from that of OC, where its maximum reverse fow velocity and its location, and their maximum velocity ratios are comparable to those of the turbulent BFS flow.

Finally, the response of the VPC to changing free stream velocity and ventilation rates can also be better understood through a comparison with the behavior of BFS fow upon changing in Reh. With increasing *U* under fxed ventilation, the cavity length of TBC frst increases and then drops abruptly as the cavity transitions from TBC to OC. This phenomenon is similar to that occurs in the change of  $X_r/h$ of BFS fow as the fow feld transitions from laminar to turbulent upon increasing  $Re_h$ . This comparison implies the change of cavity length in VPC is strongly associated with the transition of internal gas flow from laminar-like to highly unsteady, and the delay of such transition may help sustain the presence of TBC upon changing external flows. Our results suggest that such delay, or the control of state of the internal fow, can be achieved through ventilation. Particularly, it has been found that the internal pressure of VPC measured at the cavitator grows with ventilation (Qin et al. [2019\)](#page-14-4), indicating a growth of favorable pressure gradient with increasing *Q*. Such behavior explains the transition from OC to TBC with increasing *Q* under fxed *U* due to a laminarization process of turbulent fow with favorable pressure gradient (Patel and Head [1968\)](#page-14-12). Further increase in *Q* beyond the transition can help suppress the growth of instability within the shear region that affects the internal fow structure, allowing the cavity to be sustained in its TBC regime even at the higher  $Re<sub>h</sub>$  where the laminar BFS flow may transition into turbulent.

# <span id="page-12-0"></span>**4 Conclusions**

In this study, we conducted fow visualization and particle image velocimetry experiments using fog particles as tracers to provide the internal fow characterization of a ventilated partial cavitation. Various experiments are conducted at two diferent cavity regimes, i.e., open cavity (OC) and two-branch cavity (TBC) under various free stream velocity  $(U)$  and ventilation rates  $(Q)$ . Instantaneous flow visualization and mean fow feld show the presence of general internal flow structures for both OC and TBC, i.e., forward flow region near the moving air-water interface, reverse flow region occupying the large portion of the internal flow, and vortical structures (near-cavitator vortex and internal flow circulation vortex) swirling in opposite directions. The comparison between instantaneous and mean fow felds shows that the OC internal flow is highly unsteady, and its internal flow circulation vortex breaks down intermittently and advects upstream. In contrast, the TBC internal fow is laminar-like despite some unsteadiness within the reverse flow region associated with the Kelvin–Helmholtz (KH) instability triggered by the strong velocity gradient at the interface between the forward and reverse fows. The comparison between VPC internal fows and the single-phase

flow past a backward-facing step (BFS flows) further reveals the cause of the aforementioned unsteadiness in the internal flow regions. For OC, it is suggested that the internal flow in air phase is highly coupled with the external recirculation region which is in turbulent water phase, and the unsteadiness of the internal fow structure may be infuenced by the convective instability from the external recirculation region. For TBC, the internal fow structure resembles a laminar BFS flow, despite the presence of the unsteadiness within reverse flow due to the KH instability. However, such unsteadiness in TBC internal fow does not contaminate the overall flow structure. Based on such comparison, explanations are made on the cavity behavior and the regime transitions under the change of *U* and *Q*. As *U* increases from TBC, internal flow structure gradually becomes unstable and exhibits an increased unsteadiness within the reverse fow region, similar to the onset of convective instability within the separated shear layer of BFS fow that triggers the transition of its regime from laminar to turbulent. Further increase in  $U$  results in highly unstable internal flow with the cavity regime transitioning from TBC to OC, and its internal fow exhibits strong coupling with the external flow which is in turbulent regime. As *Q* increases from OC, in combination with favorable pressure gradient, highly unstable internal flow becomes more laminar which ultimately changes its regime to TBC. Correspondingly, the cavity length of OC grows due to an increased role of air phase until the internal flow of OC becomes decoupled with the external flows. Further increase in *Q* beyond the transition suppresses the growth of perturbation from the KH instability in the strong shear region, and the cavity sustains in TBC regime even at higher  $Re<sub>h</sub>$  where the laminar BFS flow may transition into turbulent.

Our measurements reveal the distinct fow structures of VPC under diferent cavity regimes and demonstrate the role of internal fow on the cavity geometry and regime transitions. In addition, our fndings suggest that the ventilation control can potentially stabilize the cavity in the TBC regime by delaying its internal fow transition from laminar-like to highly unsteady. Therefore, to further elucidate the effect of ventilation on the dynamics of VPC, it is required to scope into the gas leakage mechanism associated with cavity transition and internal flow changes. However, such investigation requires 3D flow characterization for unsteady closure of OC and internal fow bifurcation of TBC, which cannot be performed with the current planar PIV measured only at the single spanwise location. For instance, our measurements characterize the internal fows using planar PIV at the center plane. But to gain a better understanding of the unsteady closure of OC and the internal fow bifurcation of TBC, we recommend future works to measure and examine the three-dimensional fow characteristics. Particularly, for TBC, we believe that conducting PIV experiments at

multiple spanwise locations, possibly near the wall of the test section, may shed some light on characterizing the 3D internal fow feld and further estimating the discharge rate at the closure. However, such experiments are currently limited by the optical accessibility of the facility, since only an inch-wide center portion of the ceiling is transparent. Consequently, the current setup can only illuminate the internal flows in the vicinity of the center plane ( $z = 0$ ). In addition, our current study is limited in the relatively low range of Fr due to the restricted length of the test section of the water tunnel. As a result, our study does not include VPC in higher  $Re<sub>b</sub>$  range for TBC where one can examine how the ventilation infuences the unsteadiness of TBC internal fow. This limitation can be overcome by using a smaller step height for the backward-facing step cavitator in the future experiments.

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