RESEARCH ARTICLE



# **Measuring three‑dimensional fow structures in the conductive airways using 3D‑PTV**

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Abstract Detailed information about flow patterns and mass transport in the conductive airways is of crucial interest to improve ventilation strategies as well as targeted drug delivery. Despite a vast number of flow studies in this feld, there is still a dearth in experimental data of threedimensional fow patterns, in particular for the validation of numerical results. Therefore, oscillating fow within a realistic model of the upper human conductive airways is studied here experimentally. The investigated range of Reynolds numbers is *Re* = 250–2000 and the Womersley number is varied between  $\alpha = 1.9 - 5.1$ , whereby physiological flow at rest conditions is included. In employing the three-dimensional particle tracking velocimetry measurement technique, we can directly visualize airway specifc fow structures as well as examine Lagrangian trajectory statistics, which has not been covered to date. The systematic variation of characteristic fow parameters in combination with the advanced visualization technique sheds new light on the mechanisms of evolving fow patterns. By determining Lagrangian properties such as pathline curvature and torsion, we fnd that both strongly depend on the Reynolds number. Moreover, the probability density function of the curvature reveals a unique shape for certain fow regions and resembles a turbulent like behavior at the small scales.

## **1 Introduction**

In the past, intense research has been carried out to gain a solid knowledge of the fuid dynamics inside the human airways with the aim of improving the existing ventilation strategies or optimizing drug delivery. One of the earliest experimental investigations on the actual flow behavior inside the upper bronchial tree was conducted by Schroter and Sudlow [\(1969](#page-10-0)). They used a simplifed representation of the human lung geometry based on data provided by Weibel [\(1963](#page-10-1)) and Horsfeld and Cumming [\(1967\)](#page-10-2). With this model, they were able to capture important flow characteristics, like flow separation at bifurcations causing asymmetric flow for inspiration and the development of four-vortex patterns for expiration, generated by flow unification from the previous generations. Since then, continuous effort has been made to build more realistic lung models and to introduce more advanced measurement techniques to this feld of research. Nevertheless, there is still a lack of experimental data, which motivated our work. In the following, we want to emphasize our motivation by recapturing some of the most recent research work, while laying our focus on the used models as well as measurement techniques and elaborate, where more work has to be done.

Due to the relatively simple geometry of the Weibel model, it has been investigated excessively. Measurements have been carried out using 1D techniques such as LDA (Kerekes et al. [2016\)](#page-10-3), 2D applications like particle image velocimetry (PIV) (Fresconi and Prasad [2007\)](#page-10-4), as well as magnetic resonance velocimetry (MRV) to reconstruct the whole three-dimensional velocity field (Jalal et al. [2016](#page-10-5)). Despite the usage of similar geometric models, the applied type of flow is rather different. The work of Kerekes et al.  $(2016)$  $(2016)$  just covered constant inspiratory flow at a flow rate of 30 l/min, representing breathing under rest conditions.

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Fresconi and Prasad ([2007](#page-10-4)) went a step further in investigating steady inspiratory/expiratory as well as oscillatory flow. They carried out a thorough analysis of the developing secondary flow and found a critical Dean number  $Dn =$ 10, at which noticeable secondary motion starts to occur. In incorporating the time-resolved PIV technique, Ramuzat and Riethmuller ([2002\)](#page-10-6) were able to resolve the fow feld during the whole breathing cycle (see also Theunissen and Riethmuller [2008\)](#page-10-7). Whereas they just covered the laminar regime (*Re <*1000), which limits the interpretation of the results to smaller, more distal branches of the conductive airways, the research of Jalal et al. [\(2016\)](#page-10-5) examined the laminar as well as the turbulent regime with a maximum Reynolds number of *Re* = 5000. One limitation of this research is that they just modeled the steady case of inspiration. However, they showed that the strength of secondary motion reaches a maximum at around *Re* = 2000 and weakens with further increased velocities due to the transition into the turbulent regime. They further found a change of the evolving vortex pattern in the daughter branches with increasing Reynolds numbers.

A next step towards more realistic geometries has been done by Adler and Brücker ([2007\)](#page-9-0). They modifed the commonly used planar and symmetric models to mimic a more realistic lung by rotating the daughter branches to introduce strong three-dimensional flow as well as incorporating an asymmetric branching of the main bifurcation. The investigated cases covered the fow at rest conditions as well as high-frequency oscillatory ventilation (HFOV). Velocity patterns in various cross sections of the three-dimensional lung model were measured using the particle image velocimetry (PIV) technique.

With increasing performance of medical imaging methods patient specifc geometries could be introduced into the research community. Such geometries have been used by Große et al. [\(2007](#page-10-8)), Soodt et al. ([2013](#page-10-9)) as well as by Banko et al. [\(2015\)](#page-9-1). By analogy to the previously described research works, they all difer in the covered fow regimes and employed measurement techniques. Große et al. ([2007\)](#page-10-8) conducted their investigations using multi-plane PIV with Reynolds numbers between *Re* = 1050–2100 for oscillatory motion as well as steady-state conditions. Their results emphasize a strong diference between steady-state measurements and oscillatory fow measurements. A further improvement of the multi-plane PIV approach has been done by the use of a scanning stereo PIV system by Soodt et al. ([2013\)](#page-10-9). With this quasi-volumetric measurement technique, they were able to reconstruct three-dimensional velocity felds within the trachea and the frst bifurcation for laminar and turbulent flow. Yet, the highest achievable Reynolds number ( $Re = 2650$ ) was only investigated for steady inspirational and expirational fow. Only the lower laminar regime has been examined under oscillatory fuid motion.

One of the frst true three-dimensional, three-component measurement techniques using a patient specifc geometry has been applied by Banko et al. [\(2015\)](#page-9-1). They used a phasedlocked MRV technique to resolve the flow field at a single but recognizable high Reynolds number of *Re* = 4200 with a sinusoidal modeled fow. The investigated model geometry reaches from the mouth down to the fourth bifurcating generation. As a major conclusion, they emphasize that the strength of the secondary flow is strongly dependent on the used geometry and is much higher in the realistic model compared to more simplifed ones.

Looking at this brief overview of the most recent research activities, we found that true time-resolved threedimensional velocity data for oscillatory fow under resting conditions are still missing. To fll this gap, we carry out three-dimensional Particle Tracking Velocity measurements (3D-PTV) to investigate the oscillatory flow ( $Re =$ 250–2000,  $\alpha$  between 1.9 and 5.2) inside a three-dimensional model of the upper human bronchial tree, incorporating six bifurcating generations. With the use of the 3D-PTV technique, we are able to visualize fow features, which have not been reported yet, as well as to derive Lagrangian statistics for particle acceleration, torsion, and curvature. We fnd that these statistics are giving a new insight to the description of the occurring fow structures and that these can be used as a promising tool to characterize the overall fow inside the human lungs.

## **2 Methods**

#### <span id="page-1-0"></span>**2.1 Lung model and fow parameters**

For our investigations, we are using a three-dimensional artifcial representation of the human lung based on data proposed by Weibel [\(1963](#page-10-1)) and Horsfeld et al. [\(1971](#page-10-10)). The same model has been used in the previous studies (e.g., Adler and Brücker [2007](#page-9-0); Bauer et al. [2012;](#page-9-2) Bauer and Brücker [2015\)](#page-9-3) and has proven its capability of representing important fow structures. For a detailed explanation of the model geometry and its manufacturing, the reader is thus referred to Adler and Brücker ([2007](#page-9-0)) as well as to Ramuzat and Riethmuller [\(2002\)](#page-10-6). The fow through the lung is modeled as an idealized sinusoidal breathing cycle, parameterized by the breathing frequency *f* and the tidal volume  $V_T$ . For such a fow, the Reynolds number *Re* and the Womersley number  $\alpha$  can be defined as follows:

<span id="page-1-1"></span>
$$
Re = \frac{4 \cdot V_{\rm T} \cdot f}{d_0 \cdot v}, \qquad \alpha = \frac{d_0}{2} \cdot \sqrt{\frac{2 \cdot \pi \cdot f}{v}}.
$$
 (1)

Where  $d_0$  denotes the diameter of the trachea and  $\nu$  the kinematic viscosity.

We conducted experiments by varying the Reynolds number in a range from *Re* = 250–2000 and the Womersley number between 1.9 and 5.2. For all investigated cases, data were acquired at peak inspiration and expiration phases. For  $Re = 2000$  and  $\alpha = 3.0$ , we additionally measured the flow feld at accelerating/decelerating inspiration and expiration phases, respectively, resulting in a total number of six phase positions.

Besides the Reynolds and the Womersley behavior, the Dean number *Dn* represents an important parameter in denoting secondary flow intensity in curved airway branches, defned by the following:

$$
Dn_i = Re_i \cdot \sqrt{\frac{d_i}{2 \cdot r_i}}.
$$
 (2)

The Dean number varies for each branch *i* and scales with the local Reynolds number, the branch's diameter  $d_i$ , as well as the local curvature radius  $r_i$  of the model's geometry. All characteristic fow parameters are summarized in Table [1.](#page-2-0) The indices "L" and "R" at the Dean number refer to the right and left main branch with respect to the anatomical definition.

## **2.2 Experimental set‑up**

The key component of the measurement set-up is the model of the human airways, described in Sect. [2.1.](#page-1-0) The oscillatory motion of the fow is realized by driving a piston diaphragm pump using a linear motor (G400 series, MOOG). The piston motion follows a sinusoidal function with adjustable frequency and amplitude. While the piston's motion frequency directly correlates to the breathing frequency *f*, the tidal volume  $V_T$  can be adjusted by changing the amplitude.

Three high-speed cameras (Phantom V12.1, Vision Research), each equipped with a 50 mm lens (Nikon Nikkor

<span id="page-2-0"></span>**Table 1** Overview of the chosen parameters frequency *f* and tidal volume  $V_T$  and their corresponding dimensionless numbers  $Re$ ,  $\alpha$ , and *Dn* in the right and left main bronchus

f[Hz]	$V_T$ [ml]	$Re_0[-]$	$\alpha$ [-]	$Dn_{1,R}[-]$	$Dn_{1,L}[-]$
0.06	156	248	1.9	24	19
0.06	313	496	1.9	48	39
0.06	469	744	1.9	71	58
0.15	125	496	3.0	48	39
0.15	250	992	3.0	95	78
0.15	375	1488	3.0	143	116
0.15	500	1984	3.0	190	155
0.30	187	1484	4.3	142	116
0.45	125	1488	5.2	143	116

50 mm/f1.8D), for image recording and three high-power LEDs (PT-121, Luminus Devices) for illumination are used as the imaging system. The camera exposure time was set to 500 μs during which a light-pulse of 450 μs length was fashed. A frame rate of 500 Hz for the cameras and the light sources has been used.

With using a water–glycerin mixture (43:57 mass ratio) as the working fuid, we matched the refractive indices between model and liquid. Polyamide particles (Vestosint) with a mean diameter of  $d_p = 100 \,\mu\text{m}$  were used to seed the flow. These particles are neutrally buoyant, as they have the same density as the liquid. The theoretical Stokes number of these particles can be determined by the following:

$$
St = \frac{t_{\rm p} \cdot u_0}{d_0} = \frac{\rho_{\rm p} \cdot d_{\rm p}^2}{18 \cdot \eta_{\rm fl}} \cdot \frac{u_0}{d_0} \,. \tag{3}
$$

This non-dimensional number denotes a ratio of the different time-scales of the solid particles and the fluid flow. For  $St \ll 0.01$ , the lag-introduced error for particle-based velocity measurements can be assumed to be under 0.7% (Dring [1982\)](#page-10-11). With the particle density  $\rho_p = 1150 \text{ kg/m}^3$ , the fluid's dynamic viscosity  $\eta_{\text{fl}} = 9.66 \times 10^{-3}$  Pa s, as well as a maximum expected velocity of  $u_0 = 2$  m/s in the trachea, the resulting Stokes number is  $St = 0.0074$ . Following the previously given assumption, the slip error of these relatively large particles can still be neglected.

# **2.3 Three‑dimensional particle tracking velocimetry (3D‑PTV)**

The raw images are evaluated using a three-dimensional particle tracking algorithm. The algorithm is a self-developed code using the MATLAB software package (MATLAB R2015a, Mathworks Inc.) as the developing framework. First tests introducing the code and showing the capability of capturing important fow structures have been presented recently by Janke and Bauer [\(2016\)](#page-10-12).

The three-dimensional reconstruction of the position of each tracer particle is realized by implementing photogrammetric methods, as presented by Maas et al. [\(1993\)](#page-10-13). To minimize the occurrence of ghost particles, an iterative particle reconstruction routine has been added. This technique is adapted from the proposed steps by Wieneke ([2013\)](#page-10-14).

For the particle tracking itself, a combination of a fourframe search algorithm (Malik and Papantoniou [1993\)](#page-10-15) to initialize particle tracks, and an algorithm incorporating the 'Shake-The-Box' idea (Schanz et al. [2013](#page-10-16)) for further tracking, is employed.

Due to the cyclic fow nature, we can use an ensemble averaging approach by performing successive measurements at phase-locked positions. With such a technique, we can

allow for a lower particle seeding density (ppp  $\approx 0.001$ ) but yet yield enough data for an adequate reconstruction.

Each acquired phase ensemble itself represents a small time integration window ( $\Delta \varphi = 6/40 \pi$ ) during the breathing cycle (see Fig. [1](#page-3-0)). To take diferent breathing frequencies into account, while keeping the integration windows comparable, their start and ending point is defined by  $\varphi \pm 3/40 \pi$ . Where  $\varphi$  denotes the investigated phase position. Based on the sinusoidal piston motion, inspiration starts at  $\varphi = 2/4 \pi$ with its peak flow at  $\varphi = 4/4 \pi$ . Flow reversal occurs at  $\varphi$  $= 6/4 \pi$  and maximum expirational flow is present at  $\varphi =$  $0/4 \pi$ . Accelerating inspiration/expiration is defined to be at  $\varphi = 3/4 \pi$  and  $\varphi = 7/4 \pi$ . The decelerating phases are at  $\varphi =$ 5/4  $\pi$  and  $\varphi = 1/4 \pi$ , respectively. The size of the integration windows has been chosen arbitrarily to be 7.5% of the whole cycle. Yet, we consider this duration to be short enough to assume a steady state during the time integration.

# **3 Results and discussion**

#### **3.1 Quality of reconstruction**

The calibration of the multi-camera system is carried out by traversing a fat checkerboard through the measurement volume. The plate is positioned at six diferent locations within the measurement volume with a distance of 10 mm



<span id="page-3-0"></span>**Fig. 1** Piston stroke *s* over the phase angle  $\varphi$  follows a sinus function. The central positions of the diferent measurement points for different phase angles  $\varphi$  are marked with a *dashed line* 

between each calibration plane. The resulting three-dimensional distribution of calibration points is used to calculate the projection matrix of each camera using a pinhole camera model. Due to practical reasons, the measurements had to be split into four measurement series. Before each measurement series, the calibration was repeated.

The error of reconstruction can be expressed as the diference between the real-image coordinates of each point and the back-projected coordinates of the reconstructed threedimensional position of this point (Wieneke [2008](#page-10-17)). This diference is shortly called disparity. The volume averaged disparities after calibration  $\delta_{\text{cal}}$  for each camera and each measurement series are summarized in Table [2](#page-3-1).

In addition, the calibration procedure could not be conducted with the lung model being in place. Therefore, the model was removed during the calibration procedure. Due to this fact, reconstruction errors, caused by introducing the model to the light path, cannot be included in the disparity  $\delta_{\text{cal}}$ . To receive a quantification of this influence, the disparity values  $\delta_{\rm run}$  of around 5000 particles for a representative measurement run within each series have been calculated and averaged. The values are noted in Table [2](#page-3-1), as well.

The calibration itself leads to a sub-pixel accurate reconstruction with disparities under 0.3 px. After re-installing the model and calculating the disparities of the reconstructed particles, the error rises up to 2.1 px. Two main reasons are assumed to be responsible for this. On one hand, an inhomogeneous refractive index of the model, caused by the casting process and the aging of the silicone, could be observed. On the other hand, there could be slight shifts in the refractive index of the water–glycerin mixture. Although the refractive index of the liquid was measured regularly, some minor deviations across the whole volume of around 60 l cannot be fully excluded.

An overview over some overall trajectory statistics is given in Table [3.](#page-4-0) For most of the runs, the reconstructed database consists of around 11,000 trajectories. To minimize the infuence of ghost particles/ghost trajectories, only paths with a minimum length greater than five time steps are considered for evaluation. This reduces the number of available trajectories by around 30%. For most cases, the maximum tracked length is over a hundred time steps, while the mean value is at around 20 time steps.

<span id="page-3-1"></span>



<span id="page-4-0"></span>

# <span id="page-4-2"></span>**3.2 Characteristic fow structures**

One motivation in using the 3D-PTV approach is to resolve the three-dimensional velocity feld while visualizing important fow structures in a very intuitive way by showing actual particle paths.

The typical flow fields for peak inspiration and peak expiration at  $Re = 2000$  and  $\alpha = 3.0$  are illustrated in Fig. [2](#page-4-1). For both cases, the fgure provides two diferent views on the reconstructed volume. For later references, the trachea will also be referred to as  $G_0$ , the (anatomical) left main branch as  $G_{1,L}$ , and the (anatomical) right main branch as  $G_{1,R}$ , respectively. The results show that we were able to track particles down to the fourth bifurcating generation for a specifc region of the model. The particle paths in this fgure are color-coded with respect to their local stream-wise velocity. To obtain the velocity information for each line segment, the time derivative of the trajectory's coordinates, using a central diferencing scheme, is calculated.



<span id="page-4-1"></span>**Fig. 2** Reconstructed particle trajectories for *Re* = 2000 and  $\alpha = 3.0$  at  $\varphi = 4/4\pi$  and  $\varphi = \frac{0}{4\pi} \frac{color\text{-} \text{coded}}{inter\sqrt{1-\rho^2}}$ their local stream-wise velocity. **a**, **b** perspective view, **c**, **d** top view

For the case of inspiration (Fig. [2a](#page-4-1), c), the flow starts to split into the daughter branches already a few millimeters upstream the carina (①). At this point, particle trajectories near the wall start to curl inwards, forming the frequently reported counter-rotating helices (Adler and Brücker [2007](#page-9-0); Freitas and Schröder [2008\)](#page-10-18). This structure repeats in every branch, where the symmetry line of the helical pairs changes accordingly to the direction of the model's curvature. Mass being already at the center line seems to be unafected by the helical motion. Following  $G_{1,R}$ , it becomes apparent that this strong helical motion can also lead to a cross-fow of nearwall particles towards opposite branches relative to their origin position in the trachea (②). Thus, especially near-wall particles are prone to change their pathway direction, which is crucial for targeted drug delivery.

During expiration (Fig. [2](#page-4-1)b, d), the typical double vortex pair develops within the tracheal area (③). The evolving helical structure reveals that the infuence of distal branches on upstream fow patterns is well preserved. Symmetry line positions of the helix pairs strongly depend on the branching geometry of the daughter branches.

To investigate the fow felds in more detail, Fig. [3](#page-5-0) shows three diferent cuts through the measurement volume at four different phase positions  $\varphi$  (see Fig. [1\)](#page-3-0) during the breathing cycle for  $Re = 2000$  and  $\alpha = 3.0$ . The three cross sections correspond to the trachea, and the right and the left main bronchus, respectively. The color coding is based on a qualitative representation of the local torsion. The torsion  $\tau$  as well as the curvature  $\kappa$  are two characteristic scalar values to defne a three-dimensional curve. Both can be calculated by evaluating the Frenet–Serret formulas. For our case, the curvature and torsion are determined by the following relations:

$$
\kappa = \frac{|\dot{\underline{x}} \times \ddot{\underline{x}}|}{|\dot{\underline{x}}|^3},\tag{4}
$$

$$
\tau = \frac{\dot{x} \cdot (\ddot{x} \times \ddot{x})}{|\dot{x} \times \ddot{x}|^2}.
$$
\n(5)

In these formulas,  $x$  is the position vector and time derivatives are marked with a dot. Both scalars have the unit  $m^{-1}$ . Mathematically, the curvature can be explained as the deviation of a curve from a straight line and the torsion can be described as a measure of how strong the curve bends away from the plane of curvature. While positive values (red) of torsion indicate a clockwise sense of rotation and negative values (blue) a counter clockwise, respectively.

Overall, we can show that the torsion is suitable in featuring characteristic fow structures, which has been reported in various previous studies (e.g., Ramuzat and Riethmuller [2002](#page-10-6); Adler and Brücker [2007;](#page-9-0) Fresconi and Prasad [2007\)](#page-10-4).

We start with a more detailed analysis by first considering inspiration. Interestingly, while the Dean-type vortices in  $G_{1L}$  (Fig. [3](#page-5-0)L.a, b) are rather symmetric and capture the whole cross section, as also reported for simplifed symmetric model geometries (Fresconi and Prasad [2007;](#page-10-4) Jalal et al. [2016](#page-10-5)), the helix pair in  $G_{1R}$  is very small and shifted towards the outer wall (Fig. [3R](#page-5-0).a, b), obviously indicating the infuence of the next daughter generation.

<span id="page-5-0"></span>**Fig. 3** Detailed top-to-down view showing reconstructed particle trajectories in diferent cross sections for *Re* = 2000 and  $\alpha = 3.0$ *, color*-coded with their local sign of torsion  $\tau$  at **a**  $\varphi = 3/4 \pi, \mathbf{b} \varphi = 5/4 \pi, \mathbf{c}$  $\varphi = \frac{7}{4} \pi$  and **d**  $\varphi = \frac{1}{4} \pi$ . Separation areas are marked with *bold lines*. Dean vortices are highlighted by *dashed circles*



During expiration, we are able to observe a delayed development of the four-vortex pattern inside  $G_0$ , which has, to the best of our knowledge, not been reported yet. At the accelerating expiration phase (Fig. [3T](#page-5-0).a), only a single helix pair, originating from the right primary bronchus, is detectable. When reaching the state of maximum expiration, an additional pair of helices has developed, forming the typical four-helix pattern (Fig. [2d](#page-4-1)), which further persists during the deceleration phase (Fig. [3T](#page-5-0).d). This diference between acceleration and deceleration suggests that despite the same Reynolds number, the development of certain structures obviously occurs as a certain value of the Reynolds number, and thus, Dean number has been exceeded near the peak velocity phase. Inertia contributes to the enduring existence of the structures during the deceleration phase, although the flow parameters are already below the critical values.

#### **3.3 Infuence of Reynolds and Womersley number**

Cross sections of the trachea during peak expiration are presented in Fig. [4](#page-6-0) to show the infuence of diferent Reynolds *Re* and the Womersley numbers  $\alpha$ . All trajectories are colorcoded with the local stream-wise velocity, normalized with respect to the calculated mean velocity  $u_0$  within  $G_0$  used in Eq. ([1\)](#page-1-1).

The effect of the Womersley number seems to be negligible, at least in the parameter range investigated here. However, for the diferent Reynolds numbers, three

diferent regimes can be identifed. The frst one is a fourvortex structure with a very symmetric arrangement at  $Re = 250$ . The second pattern is characterized by a single pair of counter-rotating vortices for  $250 < Re \le 1000$  with a high velocity jet in between them. The third structure reveals again the development of two pairs of counterrotating helices for  $Re \ge 1500$  but with a stronger asymmetry compared to the case of *Re* = 250. To our knowledge, such an efect has not been reported yet.

We want to investigate this in further detail by considering the corresponding helicity felds for *Re* = 250, 750, 1000, 1500. The helicity has been calculated by interpolating the Lagrangian velocity vectors onto an equidistant grid using a Gaussian-weighted interpolation (see Fig. [5](#page-7-0)b). The size of each cell was set to  $1.2 \times 1.2 \times 1.2$  mm<sup>3</sup>. Using this Eulerian representation of the velocity feld *u*, the helicity *H* can be calculated for each cell volume *V* as follows:

$$
H = \int_{V} \underline{u} \cdot (\nabla \times \underline{u}) \, \mathrm{d}V. \tag{6}
$$

The isosurfaces of the resulting helicity felds, shown in Fig. [5](#page-7-0)c, envelope values, which are at least  $\pm 10\%$  of the maximum calculated helicity. The vortex structures do not change signifcantly inside the right main bronchus throughout the whole range of Reynolds numbers. This leads to the conclusion that the variation in the emerging vortex pattern

<span id="page-6-0"></span>**Fig. 4** Detailed top-to-down view showing reconstructed particle trajectories for diferent  $Re$  and  $\alpha$ , *color*-coded with their normalized local stream-wise velocity at  $\varphi = 0/4 \pi$ . For chosen parameter values, vortices are highlighted for identifying diferent vortex regimes



<span id="page-7-0"></span>**Fig. 5 a** Measurement volume, in which the helicity *H* is calculated (*dark gray*). **b** Interpolation of unstructured Lagrangian velocity vectors using a Gaussian-weighted averaging scheme. **c** Isosurfaces of helicity *H* for *Re* = 250, 750, 1000, 1500 at  $\varphi = 0/4 \pi$  reveal different vortex patterns for diferent *Re*. The important vortices in  $G<sub>1</sub>$  involved in this development are marked with ①–④



mainly originates from the left branch. Within this section, a large area with negative helicity (①) can be found, starting from the second bifurcation of the model and reaching up to the trachea. This vortex has two partnering counter-rotating vortices (② and ③). The frst one (②) is located right after the second bifurcation and rapidly disappears in downstream direction. The second structure (③) develops at around the middle of the bronchus (②). This second structure then builds the fourth observable vortex in the tracheal area. At  $Re = 750$ , the counter-clockwise rotating vortex  $\circ$ ) is displaced because of a strengthened clockwise rotating vortex (②) originating from the second bifurcation. This displacement leads to a suppression of the former anterior clockwise rotating vortex (③). It is further noticeable that the vortices, sharing the same sense of rotation and originating each from  $G<sub>LL</sub>$  and  $G<sub>LR</sub>$ , have unified at the main carina. This forms the first observed two-vortex pattern (see Fig.  $4$ ,  $Re = 750$ ). At the end of the left main bronchus, a small additional counterrotating vortex (④) has evolved at the posterior side. This small structure grows in size with further increasing *Re* and starts to disturb its neighboring vortex  $(2)$ . At first ( $Re =$ 1000), with no signifcant impact, but at *Re* = 1500, it has grown large enough to prevent the unifcation of the vortices of both branches and starts to reach into the trachea. The very prominent counter-clockwise rotating vortex (①) at the lower Reynolds numbers is now prevented to propagate further and vanishes right at the main carina. At its position, a clockwise rotating vortex (③) can evolve without disturbance and is able to reach up to  $G_0$ .

For these mechanisms, the location of the daughter generations in combination with the local Dean number is playing an important role and hence the possibility for the Dean mechanism to develop undisturbed or not. To conclude this section, we found a signifcant change of the secondary flow structure within this model and could show that the

evolution of the four-vortex structure is quite diferent for the cases of  $Re = 250$  and  $Re \ge 1500$  and thus an increasing infuence of distal bifurcations on downstream fow patterns at higher Reynolds and Dean numbers, respectively.

## **3.4 Lagrangian statistics**

The 3D-PTV method also offers a new insight to the human lung flow by deriving typical Lagrangian trajectory properties, which are not easily obtainable with Eulerian measurement approaches. One main property is the particle acceleration *a*, which can be calculated as the second time derivative of the three-dimensional particle path. For the results shown in this subsection, the infuence of varying Reynolds number is shown only, since no signifcant changes could be found by varying the Womersley number. The other two properties which we want to cover are the local torsion and curvature of the particle paths, which we introduced in Sect. [3.2](#page-4-2) already.

In Fig. [6](#page-8-0), the probability density functions (PDF) of the normalized particle acceleration *a*∕ √ *a*<sup>2</sup> (left) as well as of the normalized local torsion  $\tau \cdot d_0$  (right) are given for the area around the main carina (highlighted in dark gray) at diferent Reynolds numbers during peak inspiration ( $\varphi = 4/4 \pi$ ). Since no signifcant diferences between the peak inspiration and expiration could be found, results are only presented for peak inspiration.

The shape of the distributions of the local acceleration reveals to be nearly identical for all covered Reynolds numbers. We measure accelerations up to five times the root mean square with a symmetric distribution. In other words, the traced particles experience similar normalized accelerations while passing the airways, independently of the Reynolds number. The average fatness *F* of all distributions is  $F \approx 6.0$ . This can be interpreted as a small but noticeable

<span id="page-8-0"></span>

deviation from an idealized Gaussian distribution  $(F = 3)$ . We further calculated the averaged fatness factors for the distribution within  $G_{1,L}$  and  $G_{1,R}$  separately and received  $F_{1,L}$  $= 6.5$  and  $F_{1,R} = 4.6$ . We can see that the asymmetrical branching leads to diferences between those two areas. The acceleration within the right main bronchus tends to follow a more Gaussian distribution. However, note that the Dean number in  $G_{1,R}$  is larger than in  $G_{1,L}$  with the inverse relation of  $Dn_{1,L}/Dn_{1,R} \approx F_{1,R}/F_{1,L} \approx 0.75$ . From these results, we can conclude two major fndings. On one hand, we fnd a Reynolds independence of the normalized acceleration PDFs, but we can show that the local geometry of the model has a noticeable impact on the shape of the distribution. On the other hand, we measured acceleration distributions, which deviate from an idealized Gaussian distribution.

In contrast to the particle acceleration, we fnd a strong Reynolds dependency for the normalized local torsion  $\tau \cdot d_0$ . With increasing *Re*, the distributions become narrower. Whereas for lower *Re*, the probability for larger torsion values increases, resulting in a fat-tail shape. Those larger values indicate a stronger out-of-plane motion from the plane of curvature. As shown in Figs. [2](#page-4-1) and [3,](#page-5-0) most of the particles are following a helical path throughout the airways. With increasing torsion, these helical paths reveal a more elongated shape, probably due to weaker secondary motion at the low Reynolds numbers.

The PDFs for the curvature, normalized with the diameter of the trachea  $d_0$ , are presented in Fig. [7](#page-9-4) for peak inspiration ( $\varphi = 4/4 \pi$ ) and for peak expiration ( $\varphi = 0/4 \pi$ ). We further divide the fow feld around the main carina in three sub-areas, covering  $G_0$ ,  $G_{1,R}$ , and  $G_{1,L}$  separately. Thereby, we fnd a strong infuence of the local geometry on the resulting shape of the PDFs. This becomes especially apparent when comparing the PDFs inside  $G_{LR}$  and  $G_{1,L}$  for peak inspiration ( $\varphi = 4/4 \pi$ ). During this phase, a pair of Dean vortices develops in each of the branches. However, since the length of the left primary bronchus is longer than of the right branch, the developing Dean vortex pair is more prominent in  $G_{1,L}$  (see Fig. [3](#page-5-0)L.b, R.b), although the Dean number is lower in the left branch. A large cross-sectional area is under the infuence of these very symmetric vortices (see Fig. [3](#page-5-0)L.b). This forces the particles on paths with similar curvature values, resulting in a very sharp peak of the PDF in  $G<sub>11</sub>$  for the cases of higher Reynolds numbers. At low *Re*, however, the PDFs are similar within the primary bronchi, which indicates a less pronounced infuence of the airway geometry for the low Reynolds numbers.

During maximum expiration, such a dominant peak cannot be found. Due to the unifcation of the Dean vortices, originating from the daughter generations, the range of occurring curvatures is broader, which widens the PDFs around the maxima. We can also detect a change of the shape and the location of the maxima in dependency of the Reynolds number. An increasing *Re* results in a shift of the maximum's location towards smaller  $\kappa \cdot d_0$ . This means that, at higher *Re* particle, paths tend to have larger curvature radii. It also becomes apparent that the PDFs converge towards a fnal shape at *Re* = 2000.

The infuence of the local geometry primarily changes the curvature values between  $\kappa \cdot d_0 = 0.01$ –1. The slopes of the distributions for  $\kappa \cdot d_0 > 1$  are neither significantly infuenced by the geometry nor by the fow phases but only depend on the Reynolds number. This reveals a more universal characteristic of the large curvature values. These slopes can be approximated by a power function with exponents between −3/2 for *Re* = 250 and −5/2 *Re* = 2000. Such values have been already reported throughout the literature (Xu et al. [2007;](#page-10-19) Braun et al. [2006;](#page-9-5) Liberzon et al. [2012](#page-10-20)), but in the feld of turbulence research. The here reported behavior of the curvature PDFs can then be interpreted as an indicator of the slow transition towards the turbulent regime with increasing *Re*. Or at least, it shows a mimicking behavior of the fow features, resulting in a turbulence like fow mixing at the small scales. In how far this trend develops further for higher *Re* will be a subject of future investigations. At the larger scales, the cited research works reported a linear slope at the log–log scale, which we are not able to see. This underlines again the infuence of the lung geometry on the development of important fow structures, which the cited research works do not represent as they focused on modeling isotropic turbulence.

<span id="page-9-4"></span>**Fig. 7** PDF of the normalized local trajectory curvature  $\kappa \cdot d_0$  in the branches  $G_0$ ,  $G_{1,R}$ and  $G<sub>1,L</sub>$  at peak inspiration  $\varphi = 4/4 \pi$  and peak expiration  $\varphi = 0/4 \pi$  for different Reynolds numbers. *Re* = 250  $\Box$ ,  $Re = 500 \odot$ ,  $Re = 750$  $(\nabla)$ ,  $Re = 1000$  ( $\times$ ),  $Re = 1500$  $(\triangle)$ , and *Re* = 2000 ( $\star$ )



# **4 Conclusion**

The visualization technique of 3D-PTV in the conductive airways allowed us to gain new insights into the threedimensional nature of helical vortices, which develop in the upper branches of the human bronchial tree. We found that inertia plays a crucial role in maintaining secondary fow structures beyond the peak phases. The systematic variation of characteristic fow parameters revealed an increasing geometric infuence of upstream generations on the following branches with increasing *Re* and *Dn*, respectively. This infuence results in a changing vortex regime between low and high *Re* during peak expiration. The variation of the frequency, i.e., the Womersley number, does not infuence the evolving flow structures.

The 3D-PTV technique further allows to derive characteristic scalars, which can be used to identify fow features, as well as to discuss them in a more general approach by evaluating their corresponding statistics. The analysis of the spatial normalized Lagrangian particle acceleration distribution uncovers a non-Gaussian shape with slightly more pronounced tails, while being independent of the Reynolds number. In contrast to this, we found a Reynolds dependency of the probability density functions for the local torsion and curvature. The curvature results unveil a strong infuence of the developing fow feld with respect to the local model's geometry at the larger scales. At the small scales, a more universal behavior, following a power-law function with exponents between −3/2 and −5/2, could be found. This behavior is comparable to the previous studies covering turbulent fow and underlines the good mixing properties of the human lung, even though the fow is not fully turbulent, yet.

In the future, the evaluation of Lagrangian statistics seems to be well suited to compare diferent lung geometries and their efect on the resulting fow felds without the need of discussing single structures. Moreover, these statistics might be able to reveal an overall systematic of how well certain lung geometries perform with regard to a homogeneous mixing or an optimal drug transport.

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