ORIGINAL

# **Experimental and numerical study of ventilated supercavitation around a cone cavitator**

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**Abstract** In this paper, a numerical and experimental analysis of the ventilated super-cavitating flow around a cone cavitator is presented. For the frst time, the experiments are conducted in an open loop water tunnel. The fuid fow velocity in the test section is between 24 and 37 m/s at a constant rate of injection. The simulations of ventilated supercavitation are provided at different velocities. The corresponding governing equations are solved numerically using the fnite element method and the mixture model. Finally, the effect of signifcant parameters such as cavitation number, inlet velocity on drag coefficient and the shape of cavity are investigated. A comparison of the numerical and experimental results shows that the numerical method can accurately simulate the physics of the ventilated cavitation phenomenon such as the cavity shape. According to the results, the maximum cavity diameter and length declines as the cavitation number increases. With an increase in the fow velocity, the cavity length and diameter increased to 200 and 18 % respectively. At constant rate of the ventilated air, with an increase of cavitation number from 0.15 to 0.25, the drag force drops by 62  $\%$ .

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#### **1 Introduction**

When the pressure in a liquid flow falls below the particular saturation vapor pressure, the liquid evaporates. This phenomenon, known as cavitation, has several applications in the industry, and is categorized by a dimensionless cavitation number σ. Supercavitation is the extreme form of cavitation, in which a single bubble of gas is formed around a body moving rapidly through water, such as a projectile. It can signifcantly reduce the drag of an underwater body, thus enabling a dramatic increase in its maximum speed. When supercavitation occurs on an underwater body, the skin friction drag is reduced drastically, so the supercavitating body can reach higher speeds under water with the same amount of energy spent compared to the absence of supercavitation.

Supercavitation can be divided into natural supercavitation and ventilated supercavitation. The natural supercavitation occurs when the free stream velocity is increased above a certain limit  $(U > 45$  m/s at sea level, and increases with submersion depth (or  $p\infty$ ). This phenomenon can also be achieved by decreasing the ambient pressure, p $\infty$ , which is only feasible in cavitation tunnels. Increasing the cavity pressure,  $p_c$ , through ventilation of the cavity is called artifcial or ventilated supercavitation. The static hydrodynamic forces and cavity shape associated with a given cavitator have been modeled by a number of researchers. Further, drag reduction can be observed on bodies surrounded fully or partially by a natural or gas-ventilated cavity [\[1](#page-10-0)].

In the past decade, several researchers have experimentally investigated supercavitating flows. Most of these studies focused on cavitation shape, velocity and pressure distributions of fow feld and control and stability of the supercavitating vehicles. Zhang et al. [[2\]](#page-10-1) performed a series



of projectile and closed-loop water-tunnel experiments to study the shape properties of natural and ventilated supercavitation. They obtain empirical formulas for ventilated cavity shape and cavitation number under zero incidence conditions. Wosnik et al. [\[3](#page-10-2)] measured the amount of ventilation gas required for sustaining an artifcial cavity at different velocities and velocity distribution in cavitation wake based on the DSPI and PIV techniques. Savchenko [\[4](#page-10-3)] investigated the effect of gravity and leakage rate on the shape of ventilated supercavitation through water tunnel experiments. The behavior of supercavitating and cavitating flow around a conical body of revolution with and without ventilation at several angles of attack was studied experimentally by Feng et al. [[5\]](#page-10-4). They demonstrated that the presence of supercavity under the ventilation condition reduced the drag. Natural and ventilated cavitations generated on a smooth-nosed axisymmetric body were studied experimentally by Feng et al. [[6\]](#page-10-5).

In recent years, most studies on supercavitating fows have been carried out using numerical methods. Ji and Luo [\[7](#page-10-6)] investigated both natural and ventilated cavitation numerically using the solver of a commercial CFD code CFX. They just simulated the cavitating flow around an under-water vehicle under different cavitation conditions. Nesteruk [\[8](#page-10-7)] simulated the unsteady evolutions of the slender axisymmetric ventilated supercavity using one-dimensional inviscid fow of the incompressible gas in the channel between the cavity surface and the body of revolution. It was shown that the cavity shape depended strongly on the values of the ventilation rate and the cavitation number. Ventilated supercavities were studied both numerically and experimentally by Rashidi et al. [[9\]](#page-10-8). The simulations were conducted for two different algorithms in free-surface treatment, both using the VOF method, but only one uses Young's algorithm in the advection of the free surface. They showed that the numerical method using Young's algorithm can accurately simulate the physics of ventilated cavitation phenomena such as the cavity shape, gas leakage and re-entrant jet. The shape of ventilated cavity and underwater body drag have been numerically simulated using the commercial code Fluent by Li-Ping et al. [[10\]](#page-10-9). They showed that total drag coefficient can be increased to a minimum value while the ratio of cavity thickness to body's diameter is about 0.02.

Javadpour et al. [[11\]](#page-10-10) experimentally investigated important natural supercavitation parameters in an open water tunnel complemented by a numerical analysis with the CFX software. They found a good agreement between the numerical and experimental results. The unsteady simulation of cavitation around NACA 0015 was investigated by Asnaghi et al. [\[12\]](#page-10-11). For multiphase simulation, single-fuid Navier–Stokes equations, along with a volume fraction transport equation were employed. For discretization of equations, they used the fnite-volume approach written on a body ftted curvilinear coordinates on collocated grid. Baradaran fard and Nikseresht [\[13](#page-10-12)] were investigated unsteady turbulent cavitating flows around a circular disk and a cone cavitator. They have used in a fnite volume approach to promote robust solution for unsteady turbulent cavitat-ing flows. Ji and Luo [\[14](#page-10-13)] studied natural and ventilated cavitation around an underwater vehicle. They used  $k - \omega$ SST turbulence model and for convenience, the solver of a commercial CFD code CFX coupled with the proposed cavitation model. They showed that for the natural and ventilated cavitation simulation, the predicted cavitation characteristics including the cavity length, cavity diameter and cavity shape agrees satisfactorily with the analytic and experimental results. Jian-hong et al. [\[15\]](#page-10-14) investigated the multiphase cavitating flow around an underwater projectile. They applied a mixture model to simulate the multiphase cavitating fow including ventilated cavitation caused by air injection as well as natural cavitation They showed that the drag coeffcient is also hardly infuenced by the natural cavitation numbers as the gas fow rate is greater than 0.555. The unsteady cavitating flows around the Clark-Y hydrofoil at velocity of 10 m/s are studied by Haung and Wang [[16](#page-10-15)]. They showed that the unsteady cavity shapes and velocity distributions are better consistent with experimental visualizations as compared with the RANS model. Zou et al. [\[17\]](#page-10-16) investigated the gas-leakage rate of an unsteady ventilated supercavitating body. They calculated the rate of volume change of supercavity based on Logvinovich's principle and the empirical formula and compared the simulation results with the results from experiment.

Most experimental studies on supercavitation have been performed in a closed-loop water tunnel [\[2](#page-10-1), [5,](#page-10-4) [9](#page-10-8), [18–](#page-10-17)[20\]](#page-10-18) and this investigation is the only one that studies ventilated supercavitation flow in an open circuit water tunnel. Also, most studies were conducted to obtain detail information about the cavity shape at a constant velocity [[2,](#page-10-1) [5,](#page-10-4) [9,](#page-10-8) [14\]](#page-10-13). In this study, however, important supercavitation parameters including the cavity shape, formation and the drag coeffcient of the cavitator are investigated experimentally in an open water tunnel along with a numerical analysis using the fnite element method. This experimental research is accompanied with the numerical modelling.

The paper is organized as follows: frst, a description of the physical problem is presented, followed by an experimental and numerical analysis. Then, the experimental and computational results are presented and discussed. Finally, a summary is given and the conclusions are drawn.

# **2 Experimental set‑up**

The experiments were conducted in a water tunnel located at the Marine Research Center of Iran. The tunnel is a



<span id="page-2-0"></span>Fig. 1 The schematic view of the water tunnel. The figure is not scale



<span id="page-2-1"></span>**Fig. 2** Test section of water tunnel

semi-open loop water tunnel which is capable of generating velocities up to 40 m/s. The water tunnel is equipped with a computer system, control system, high-speed data collection analyzer and high-speed photograph camera. A model with a nose and cylindrical body is selected without any fn, or other control surface being mounted on it. A schematic view of the components of the water tunnel is shown in Fig. [1](#page-2-0).

The cavitation generated on a symmetric cone is investigated based on experimental observations carried out in a water tunnel. Figure [2](#page-2-1) shows the test section, where a cone cavitator is placed. The test section is cylindrical with an internal diameter of D and a length of 5D (Fig. [3](#page-2-2)). The internal wall of test section is made of Plexiglas with a highly smooth surface. The end part of the test section is at the atmospheric pressure and the water is discharged into an open tank. This water tunnel includes a cylindrical tank which is flled with water injected by the pressurized air to



<span id="page-2-2"></span>**Fig. 3** Dimension of test section

a certain height. Depending on the speed, the air pressure injected for each test is variable. The installed model with a maximum diameter of d is attached to a dynamometer which is outside the test section by a cylindrical body. The cavitator geometry is shown in Fig. [2.](#page-2-1) The cavitator diameter is 10 mm, the cone length of  $y_0$  is determined based on the cone angle and the Froude number (Fr) varies from 33 to 53 relative to the free-stream velocity.

The water tunnel is equipped with seven pressure sensors to measure the fow pressure. Five of these sensors are installed on the cylindrical body of the model (Fig. [2](#page-2-1)) and the other two are mounted on the test section (P6) and the water tank (P7). The accuracy of sensors is  $2 \pm 0.01$  bar. Sensors output is converted into digital codes via an A/D convertor card, which is fnally recorded. It should be noted that the data frequency of ultimate vector in experiments is 1000 Hz for all sensors. Cavitation profle formed around the cavitator is recorded by a high-speed camera with a frame rate of 600 fps.

To measure drag force, a load cell of 5 kg f was mounted in the standard airfoil out of the test section. Also, load cell and pressure sensor wires were driven out of the water tunnel via this airfoil.

Before conducting the water tunnel tests, the results of cavity shape on a fat disk are compared with the work of Franc and Michel [\[21](#page-11-0)]. In Fig. [4](#page-3-0), the results of the study about the dimensionless cavity length (Lc/d) against the cavitation number of fow have been compared with the experimental data of France and Michel. As can be seen, the results are consistent with experimental data. The results indicate a maximum of 12 % difference between this experiment and Reichardt's theory and a maximum of 8 % difference with other studies. According to Eq. ([1\)](#page-2-3), the theoretical value of the blockage cavitation number is 0.211, whereas the minimum value of cavitation number at the maximum velocity of tests is about 0.25. Considering that  $\sigma_{blockage} < 0.25$  the effect of blockage on the measured drag coefficient and cavity shape is negligible in this experiment.

<span id="page-2-3"></span>
$$
\sigma_{blockage} = \frac{S_u^2}{S_d^2} - 1.0\tag{1}
$$



<span id="page-3-0"></span>**Fig. 4** Comparison of the present experimental results with those of Reichardt [\[21\]](#page-11-0)

where  $S_{u}$  and  $S_{d}$  stand for the cross-sectional areas of the upstream and downstream regions of the liquid flow respectively.

One of fundamental dimensionless parameters for natural supercavities is Froude number, Fr =  $U_{\infty}/(gD)^{1/2}$ . Froude number varies from 44.7 to 59 with a cavitation number between 0.25 and 0.36. Semenenko [[22\]](#page-11-1), citing Logvinovitch (1973), states that the effect of gravity can be significant if  $\sigma$ Fr < 2. The present application does not require the inclusion of gravity effects.

## **3 Mathematical models**

#### **3.1 Governing equations**

In this study, a mixture-type multiphase-fow model, based on the isotropic hypothesis for the fuid, along with the averaged Navier–Stokes equations are employed. The equations related to the mass and momentum conservation are solved to obtain the velocity and pressure felds. The mass, momentum, and volume fraction equations can be written as:

$$
\frac{\partial \rho_m}{\partial t} + \frac{\partial}{\partial x_i} (\rho_m u_i) = 0 \tag{2}
$$

$$
\frac{\partial}{\partial t}(\rho_m u_i) + \frac{\partial}{\partial x_j}(\rho_m u_i u_j) \n= -\frac{\partial P}{\partial x_i} + \frac{\partial}{\partial x_j} \left[ \mu_m \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \right] + \frac{\partial}{\partial x_j} (\tau_{ij})
$$
\n(3)

where *u* and *P* are velocity and pressure respectively. The Reynolds stress can be modeled by Boussinesq hypothesis [\[11](#page-10-10)] using the following equation:

$$
\tau_{ij} = \mu_t \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) - \frac{2}{3} \left( \rho k + \mu_t \frac{\partial u_k}{\partial x_k} \right) \delta_{ij},\tag{4}
$$

where  $\delta_{ij}$  is Kronecker symbol and  $\mu_t$  is the turbulent viscosity.

Among turbulence models, shear stress transport (SST) *k*–*ω* turbulence model is able to provide accurate predictions under adverse pressure gradients and separating fow [\[23](#page-11-2)]. Also, most researchers [[14,](#page-10-13) [18,](#page-10-17) [24\]](#page-11-3) have used turbulence model of SST to simulate cavitation. Therefore, this model is chosen as the turbulence model for the cavitation simulation in this paper.

The mixture density  $\rho_m$  and viscosity  $\mu_m$  are defined by:

$$
\rho_{\rm m} = \alpha_{\rm v} \rho_{\rm v} + \alpha_{\rm g} \rho_{\rm g} + \left(1 - \alpha_{\rm v} - \alpha_{\rm g}\right) \rho_l \tag{5}
$$

$$
\mu_{\rm m} = \alpha_{\rm v} \mu_{\rm v} + \alpha_{\rm g} \mu_{\rm g} + (1 - \alpha_{\rm v} - \alpha_{\rm g}) \mu_l \tag{6}
$$

where  $\alpha$  is the volume fraction of one component. The subscripts *v*, *g* and *l* refer to the components of the vapor, noncondensable gas and liquid respectively.

A cavitation process is governed by the mass transfer equations. Equation ([7\)](#page-3-1) gives the conservation equation of vapor volume fraction, and Eq. ([8\)](#page-3-2) shows the conservation equation of gas volume fraction.

<span id="page-3-1"></span>
$$
\frac{\partial}{\partial t}(\rho_v \alpha_v) + \frac{\partial}{\partial x_i}(\rho_v \alpha_v u_i) = \dot{m}^- - \dot{m}^+ \tag{7}
$$

<span id="page-3-2"></span>
$$
\frac{\partial}{\partial t} \left( \rho_g \alpha_g \right) + \frac{\partial}{\partial x_i} \left( \rho_g \alpha_g u_i \right) = 0 \tag{8}
$$

The domain of problem and boundary conditions are shown in Fig. [5](#page-4-0). The velocity components, volume fractions, turbulence intensity and length scale are specifed at the velocity inlet boundary and extrapolated at the pressure outlet boundaries.

The pressure and volume fractions on walls are extrapolated and the no-slip boundary condition or free-slip boundary is specifed. The mass fow rate boundary is defned at the blowhole (injection).The volume fraction of water is assumed one and other phase is considered zero as initial conditions. This means that the entire volume is flled with water.

The non-dimensional parameters of interest are defned as:

$$
\sigma_{v} = \frac{P_{\infty} - P_{v}}{\frac{1}{2}\rho U^{2}}
$$
\n(9)

$$
\sigma_c = \frac{P_{\infty} - P_C}{\frac{1}{2}\rho U^2} \tag{10}
$$

<span id="page-4-0"></span>**Fig. 5** Boundary condition and domain extent



$$
C_d = \frac{F_d}{1/2\rho U^2 A} \tag{11}
$$

 $\rho$  is water density, U<sub>∞</sub> is inflow velocity, P<sub>*v*</sub> is water vapor pressure at ambient temperature,  $P_{\infty}$  is flow pressure,  $P_c$  is the pressure of cavitation,  $\sigma_{v}$  is natural cavitation number,  $\sigma_c$  is ventilated cavitation number and C<sub>d</sub> is drag coefficient.

#### **3.2 Cavitation model**

In this study, a "two-phase mixture" approach is introduced by local vapor volume. It has the spatial and temporal variation of the vapor function, which is described by the transport equation and source terms for the mass transfer rate between two phases. The notable advantage of this model is rooted in the convective character of equation, which can signifcantly contribute to the reproduction of the physics of cavitating fows, such as cavity detachment and cavity closure, and therefore allows modeling the impact of inertial forces and drift of bubbles.

Numerical models of cavitation differ in terms of mass transfer term  $\dot{m}$ . The most common semi-analytical models are Singhal cavitation model [[25\]](#page-11-4), Merkle model [\[26](#page-11-5)], Schnerr and Sauer model [[27\]](#page-11-6) and Kunz model [\[28](#page-11-7)]. In the present study, the cavitation model developed by Singhal et al. has been adopted. Source terms included in the transport equation defne vapor generation (liquid evaporation) and vapor condensation, respectively. Source terms are a variable of local flow conditions (static pressure and velocity) and fuid properties (liquid and vapor phase densities, saturation pressure and liquid vapor surface tension). The source terms are derived from the Rayleigh-Plesset equation, and high-order terms and viscosity terms are adapted from Singhal et al.

They are given by:

$$
\dot{m}^+ = C_{evap} \frac{V_{ch}}{\sigma} \rho_l \rho_v \left[ \frac{2}{3} \frac{p_v - p}{\rho_l} \right]^{1/2} \frac{\rho_l \alpha_l}{\rho_m} \tag{12}
$$

$$
\dot{m}^+ = C_{cond} \frac{V_{ch}}{\sigma} \rho_l \rho_v \left[ \frac{2}{3} \frac{p - p_v}{\rho_l} \right]^{1/2} \frac{\rho_v \alpha_v}{\rho_m} \tag{13}
$$

where  $C_{evap} = 0.02$ ,  $C_{cond} = 0.01$  and  $V_{ch} = \sqrt{k}p_v$  and  $\sigma$ and k denotes the saturated pressure of liquid, surface tension and turbulence energy, respectively.

## **4 Numerical method**

The numerical simulation (Ansys CFX 14) can be used to detect cavitation in cavitation fow around cone cavitator. It is important to note that Ansys CFX uses the CV-FEM (control volume-fnite element method) method, and the CV-FEM method with the hexahedral mesh outperforms the tetrahedral model, which tends to degrade the computing efficiency.

With the inclusion of CV-FEM method in Ansys CFX 14, the linearized momentum and mass equations can be solved simultaneously with an algebraic multi-grid method based on the additive correction multi-grid strategy. The high-resolution scheme is adopted in space discretization to solve the differential equation, with the second-order space accuracy.

In this paper, several basic hypothesis have been assumed as follows: (1) buoyancy of bodies in water is assumed negligible, (2) gravity is assumed negligible, (3) fluid is considered as incompressible, (4) slip velocity between gas phase and liquid phase is assumed negligible, (5) vapor cavitation due to lower velocity feld is assumed negligible.

### **5 Grid independency study**

To demonstrate the grid independency of the results, grids similar to that of Fig. [6](#page-5-0) are used. Then, length and diameter cavity fractions are calculated and plotted in Fig. [7](#page-5-1) at four different nodes (N = 420,000, N = 920,000, N = 1,180,000



<span id="page-5-0"></span>**Fig. 6** Mesh near the cavitator



<span id="page-5-1"></span>**Fig. 7** Variations of length and diameter of cavity fractions at four different nodes

and  $N = 1,652,000$  nodes). Since the variation between the two grids is small, a grid with  $N = 1,180,000$  nodes is chosen for the present study. In total domain, a structural grid with 1,180,000 nodes is formed.

## **6 Result and discussion**

In Fig. [8](#page-5-2), the change of Froude number with cavitation number is shown. Changes of cavitation number versus Froude number are linear. The results indicate that the cavitation number drops as Froude number rises Increases. In this paper, the ventilated cavitation experiment is conducted on a 30° cavitator and the cavity shape and drag coefficient of tests are presented.



<span id="page-5-2"></span>**Fig. 8** Variation of cavitation number with Froude number

#### **6.1 Shape of cavity**

Figure [9](#page-6-0) shows the effect of velocity variations on cavity shape in a 30° cavitator. The fow velocities are in the range of 28–37 m/s. For air ventilation, four ports are used around the slender body at 0.5 d from the leading edge of cavitator. In this study, the ventilation fow rate is 0.0851 l/s with the results indicating the significant effect of velocity on the cavity length.

In Fig.  $10$ , the streamlines around the  $30^\circ$  cavitator and the volume fraction contours are shown for the cavitation number of 0.23. As can be seen, the cavity pressure is constant but where the cavity is closed, the pressure increases to the maximum level, and then declines gradually. In this way, the length of cavity can be estimated. For instance, when the flow velocity is 30 m/s, the cavity length will be 41 mm (Fig. [11](#page-7-0)).

Figures [12](#page-7-1) and [13](#page-7-2) show the dimensionless supercavity length and diameter, which compare the results of the numerical method with experimental tests. In both fgures, a good agreement was observed between the numerical results and those of experiments with a maximum of 6 % difference. The results show that at low velocities, the effect of ventilation on length and diameter of cavity is insignifcant.

The experimental and numerical results indicate that with an increase in the Froude number, the maximum cavity diameter rises initially and then remains at a nearly constant level (Fig. [13\)](#page-7-2).

At  $v < 29$  m/s, the end of the cavity is closed. In the ventilated cavitation after cavity closure, the length and diameter of cavity are less dependent on fow velocity. In this state, the rate of ventilated air is especially effective.

<span id="page-6-0"></span>**Fig. 9** Ventilated cavity results on a 30° cavitator in the water tunnel at different velocities



 $\overline{4}$ 

 $\overline{2}$ 

 $6<sub>n</sub>$  D

<span id="page-6-1"></span>**Fig. 10** Streamlines and air volume fraction contours

Given the practical importance of circular disks, experimental data and theoretical methods have been the subject of extensive studies. To evaluate the present results, the Reichardt's semi-empirical relations are selected as nondimensional characteristics of the cavity. The related formulas for these characteristics are given in Eqs. ([14\)](#page-6-2)–[\(16](#page-6-3)). The cavitation number is the main factor in these formulas [\[21](#page-11-0)].

<span id="page-6-2"></span>
$$
\frac{L}{D} = \frac{\sigma_v + 0.008}{\sigma_v (1.7\sigma + 0.066)} \left(\frac{D}{d}\right)
$$
\n(14)

$$
\frac{D}{d} = \left[ \frac{C_D}{\sigma_v \left( 1 - 0.132 \sigma_v^{0.5} \right)} \right]^{0.5}
$$
\n(15)

<span id="page-6-3"></span>
$$
C_D = C_{Do}(1 + \sigma_v) \tag{16}
$$

 $6<sub>10</sub>$ 

 $\overline{4}$ 



<span id="page-7-0"></span>**Fig. 11** distribution pressure behind cavitator at  $v = 30$  m/s



<span id="page-7-1"></span>**Fig. 12** Dimensionless cavity length versus Froude number

where d is the diameter of the cavitator and D and L are the maximum diameter and length of the cavity respectively. In Eq. ([16\)](#page-6-3),  $C_D$  is a parameter which depends on the cavitator geometry with a recommended value of 0.84 for disks and 0.26 for cones [[29\]](#page-11-8). It should be mentioned that Reichardt's formulations are limited to  $\sigma$  < 0.12 and natural cavitation, with the above equations being extrapolated to higher cavitation numbers in the present study.

Also, according to the application of Matched Asymptotic Expansions Method (MAEM), the empirical formulas for the vapor cavity diameter under small cavitation number were obtained [\[30](#page-11-9)]:



<span id="page-7-2"></span>**Fig. 13** Dimensionless cavity diameter versus Froude number



<span id="page-7-3"></span>**Fig. 14** The effect of cavitation numbers on dimensionless cavity length

$$
\frac{D}{d} = \sqrt{\frac{C_{D_o}(1 + \sigma_v)}{\sigma_v}}
$$
\n(17)

Figures [14](#page-7-3) and [15](#page-8-0) draw a comparison between the cavity length and diameter and that of the experimental data, numerical results and analytical relations reported by Reichardt [[21\]](#page-11-0). The results show a maximum of 12 % difference between present results and those of Reichardt's theory reported for the cavity length. As discussed earlier, the reduced cavitation number leads to an increase in the length and diameter of the cavity.



<span id="page-8-0"></span>**Fig. 15** The effect of cavitation numbers on dimensionless cavity diameter

The shape and dimensions of both natural and ventilated supercavitations are identical when the cavitation number is insignifcant (Figs. [14,](#page-7-3) [15\)](#page-8-0). According to Fig. [15,](#page-8-0) the maximum difference between present results and those of Reichardt and Ingber [\[30](#page-11-9)] for a cavity diameter is 10 and 8 % respectively.

According to the numerical predictions, cavity length can be increased by 150 % for low cavitation number, while a cavity length increase of 170 % is attained in higher cavitation condition (Table [1](#page-8-1)). According to the Fig. [16,](#page-8-2) the changes rate of cavity length is similar for both ventilated cavitation and natural cavitation. The length of ventilated cavity is almost equal to that of natural cavity for the same cavitation number, and the difference does not exceed 10 %. However, the variation rate of cavitation number is different for both ventilated cavitation and natural cavitation at various velocities.

6.5  $\overline{a}$ 6 **Ventilated cavitation** 5.5 **Natural cavitaiton (Eq.14)** 5 4.5 4 **L d** 3.5 3 2.5 2 1.5 0.15 0.2 0.25 0.3 0.35 0.4 0.45 <sup>1</sup> *c*  $\sigma$   $(\sigma_c \text{ or } \sigma_u)$ 

<span id="page-8-2"></span>**Fig. 16** Comparisons of cavity length with and without ventilation

#### **6.2 Drag coeffcient**

In this section, the drag force and drag coefficient are investigated for a 30° cavitator. Figure [17](#page-9-0) shows the drag force of 30° cavitator for  $0.15 < \sigma < 0.24$ . Changes of drag force versus cavitation number are linear. It is observed that the drag coeffcient declines steadily with an increase in the cavitation number. Figure [20](#page-9-1) indicates the experimental and numerical drag coefficients versus Froude number. The Figs. [17](#page-9-0) and [18](#page-9-2) show a good qualitative agreement between the experimental and the numerical results derived from the finite element method. As shown in Fig. [18](#page-9-2), the drag coefficient reduces as the Froude number rises.

According to Fig. [18,](#page-9-2) the drag coefficient increases initially and then remains at a constant level when Froude number drops, which is mainly due to the open end of cavity and the direct ejection of the ventilated air at  $v > 29$  m/s.



<span id="page-8-1"></span>**Table 1** Comparison of Cavity length for natural and ventilated cavitation



<span id="page-9-0"></span>**Fig. 17** Drag force variation with cavitation number



<span id="page-9-2"></span>Fig. 18 Drag coefficient variation with Froude number

For this reason, at  $v < 29$  m/s, the drag coefficient variation is negligible.

The drag coefficient variation is also significantly influenced by the ventilated cavitation numbers, as it is less than 0.22. As shown in Fig. [19](#page-9-3), the drag coefficient increases initially at a relatively fxed rate and it is then consolidated. This change of drag coefficient is consistent with the experimental results of Phillip [\[31](#page-11-10)] and the numerical result of Li-ping et al. [\[10](#page-10-9)]. Furthermore, with a reduction in the cavitation number, the pressure drag coeffcient increases gradually and friction drag coefficient drops.



<span id="page-9-3"></span>**Fig. 19** The relationship between ventilated cavitation number and drag coefficient



<span id="page-9-1"></span>Fig. 20 Comparison of estimated drag coefficient with experimental data and Reichardt's theory

For a 30° cavitator, the total drag coefficient can be reduced to a minimum value with a cavity thickness to cavitator diameter ratio of about 0.016 or the ventilated cavitation number of 0.15.

In Fig. [19,](#page-9-3) the numerical drag coefficient is compared with the experimental data. A comparison of Figs. [19](#page-9-3) and [20](#page-9-1) shows that the variation of drag coefficient versus ventilated cavitation number is non-linear although the change of drag coefficient versus natural cavitation number is linear.

Figure [20](#page-9-1) depicts the variation of drag coefficient with natural cavitation number. In this figure, the drag

coefficient is compared with the experimental data and the theoretical results provided by Reichardt [[21\]](#page-11-0) (Eq. [16\)](#page-6-3). As shown in Fig.  $20$ , the results indicate a maximum of 6 % difference between numerical and experimental results. The difference between present results and the theoretical data (mean  $= 80\%$ ) is due to the fact that Reichardt's theory is valid for small cavitation numbers and natural cavitation.

For ventilated cavitation, the drag coefficient variation corresponds to the result of natural cavitation (Reichardt's theory). Thus, for the ventilated cavitation, drag coefficient variation with ventilated cavitation number is linear. At  $0.15 < \sigma_c < 0.25$  with an increase in the cavitation number, the drag coefficient rises to 50  $\%$ .

## **7 Conclusion**

In this paper, a numerical and experimental analysis of the cavitating flow around a  $30^{\circ}$  cavitator was conducted. The experimental tests were carried out in a loop-water tunnel. The effects of flow velocity and cavitation number on cavity shape and drag coefficient were investigated.

The following conclusions can be drawn from the results of this study:

- 1. For the ventilated cavitation simulation, there is a strong agreement between the predicted cavitation characteristics such as cavity length, cavity diameter and cavity shape and the analytic and experimental results.
- 2. When the ventilated cavitation number is 0.15 or the ratio of cavity thickness to cavitator diameter is near 0.016, the drag coefficient of cavitator may be reduced to a minimum value.
- 3. According to the numerical predictions, cavity length can be increased by 150 % for the lower cavitation number, while this value can reach as high as 170 % for higher cavitation condition.
- 4. In other words, in higher cavitation numbers, ventilated cavitation has greater effect on increasing cavity length.
- 5. Flow velocity has a less signifcant effect on the diameter of cavity. Thus, with an increase in the fow velocity, the cavity length and diameter increased by 200 and 18 % respectively.
- 6. According to the results, the drag coeffcient reduction in the ventilated cavitation (over  $65\%$ ) was greater than the natural cavitation.
- 7. At constant rate of ventilated air, with an increase of the cavitation number from 0.15 to 0.25, the drag force is decreased by 62 %.
- 8. In ventilated cavitation and after cavity closure, the length and diameter of cavity was less dependent on the fow velocity, although the rate of ventilated air had a signifcant effect on increasing the cavity length.

9. Changes rate of cavity length is similar for both ventilated cavitation and natural cavitation. The length of ventilated cavity is almost equal to that of natural cavity at same cavitation number.

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