**ORIGINAL ARTICLE**



# **Powder stream thermal dynamics in directed energy deposition**  for high aggregation and thermal efficiency using computational fluid **dynamics modeling**

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## **Abstract**

Exactly converging the metal powder fux at a given substrate position is of great importance for additively manufacturing high-quality metallic products, and it essentially depends on the geometric structure and thermal dynamics of powder streams. In this study, we developed a 3D numerical model to elucidate the interactions between powder, gas, and laser beam in a coaxial nozzle during laser directed energy deposition (DED). The numerical simulation indicates that the waist-shaped powder stream converges approximately 11.2 mm below the nozzle outlet, where the minimum nominal radius is achieved. Within the laser irradiation area, the powder particle temperature increases dramatically and reaches the maximum average value at approximately 20 mm away from the nozzle outlet. Furthermore, optimal powder stream convergence and high thermal efficiency can be achieved when the powder stream driven by low inner gas flow rate converges 4 mm below the laser beam focal plane. The results of high-speed and infrared imaging analysis demonstrate that the proposed model has good predictability for the mean particle velocity and temperature, with average  $R^2$  of 0.91 and 0.96, respectively. This study provides an effective approach to achieve desired metal-powder stream convergence and high thermal efficiency for DED powder feeding.

**Keywords** Directed energy deposition · Computational fuid dynamics · Powder stream structure · Powder stream consolidation · Thermodynamics

# **1 Introduction**

Over the last two decades, the additive manufacturing (AM) technique has evolved rapidly from scientific research to industrial production  $[1-3]$  $[1-3]$  $[1-3]$ . As one of the most representative laser AM methods, direct energy deposition (DED) delivers a

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feedstock material in the form of metal powder or wire to the processing area accompanied by a coaxial laser beam to heat the feedstock and substrate material, while metal deposition assembly moves layer by layer with the CNC-controlled robot motion [\[4](#page-15-2)[–7](#page-15-3)]. Such a mechanical configuration makes the DED process highly fexible and versatile, and compared with the traditional manufacturing means, the near net shape DED manufacturing of metal parts is considered to be cost-efective and highly efficient  $[8-10]$  $[8-10]$  $[8-10]$ . With the state-of-the-art technique, a wide variety of metal materials, such as hard-to-process superalloys, high-performance aluminum alloys, and complexshaped titanium alloy parts, have been successfully manufactured and applied in many industrial felds [\[11,](#page-15-6) [12\]](#page-15-7).

Powder-based DED has a wide range of applications due to its high precision and design freedom. In the primary stage of DED, the metal particles supplied by the powder feeder are blown out of the nozzle by a high-speed inert gas. After leaving the nozzle, powder particles can absorb heat from the interaction with the laser beam, while dissipating part of the heat via convection and radiation.

During fight, some particles even melt and evaporate. Ultimately, all powders reach the deposition area but only part of them are captured and transfer mass and heat to the molten pool to form the cladding. The powder feeding process is very complex that significantly affects the cladding formation [[13,](#page-15-8) [14](#page-15-9)]. Therefore, the interactions between the powders, gas, and laser beam during DED are of great significance for both the deposition efficiency and deposited workpiece quality.

A large number of experimental and theoretical researches have been conducted to investigate the powder fow process in laser DED. Some use statistical methods to develop relevant analytical models to predict the powder distribution. Mouchard et al. [[15](#page-15-10)] employed an analytical model to reproduce the powder flow distribution from a coaxial annular nozzle to the powder focus position, characterizing the concentration of powder stream as a discrete sum of Gaussian distributions. By considering the powder fow as separate Gaussian distributions, Liu et al. [\[16](#page-15-11)] created an analytical model in which the simple arithmetic average of the powder fow from four-jet nozzles is calculated as the total concentration distribution. With a Gaussian distribution directly used to quickly determine the powder concentration variations, Pinkerton and Li [[17,](#page-15-12) [18\]](#page-15-13) proposed an analytical approach considering the powder fow as two separate cases before and after convergence point. Besides powder concentration, Tan et al. [[19\]](#page-15-14) developed an analytical method to characterize the temperature evolution of incident particles under diferent laser power and incident angles, based on image gray analysis. The analytical method analyzes a large amount of statistical data based on the priori assumption of Gaussian distribution which performs optimally within the Gaussian region. However, the distribution of the powder stream from the nozzle outlet to the substrate does not always follow a Gaussian pattern. And particles cannot be clearly recorded due to small size, high velocities, and opaque nozzle. Moreover, analytical models overlook the physical essence of certain complex phenomena, such as the interaction between powder and gas, and particle heating and melting during the powder feeding process [\[20](#page-15-15)].

Due to the limitations aforementioned, computational fluid dynamics (CFD) combined with experimental validation has been widely employed to build physical models about powder fow. Gao et al. [\[21](#page-15-16)] utilized CFD and discrete element methods to simulate the interactions between the powder particles and gas fows, in which the predefned laser irradiation rate and powder aggregation degree were used to assess the impact of nozzle geometries on powder fow. Pacheco and Cardoso [[22](#page-16-0)] concentrated on the thermomechanical behavior of discrete particles interacting with a laser source. They discussed the effects of variations in laser power, initial velocity, and surface energy on the

powder flow bonding process. Zhang et al. [[23](#page-16-1)] adopted a  $k$ - $\varepsilon$  based gas-solid two-phase flow theory to simulate the powder flow field and a separate model was developed in FLUENT to evaluate the powder concentration. Wen et al. [[24\]](#page-16-2) presented a 2D axis symmetrical numerical model to predict the coaxial powder fow. The thermodynamic behavior of multi-particles is completely modeled by solving the coupled momentum transfer equations between the particle and gas phase. Guan and Zhao [\[25\]](#page-16-3) developed a numerical model for powder stream dynamics, incorporating laser attenuation and two-way coupling between gas and particles. Reynolds averaged Navier–Stokes (RANS) was chosen to solve the turbulent continuum gas fow, while a discrete phase model (DPM) was applied to describe the powder dynamic behavior. Tabernero et al. [\[26\]](#page-16-4) proposed a turbulence  $k$ - $\varepsilon$  based numerical model to simulate the powder fux distribution in which the energy density attenuation was estimated.

The previously mentioned numerical simulation extensively used the  $k$ - $\varepsilon$  method for gas turbulence modeling. Actually, the *k-ε* model excels in fully developed turbulence but is not well-suited for the DED powder feeding process. Numerous studies [[27](#page-16-5), [28\]](#page-16-6) have proved that the *k-ω* model outperforms *k-ɛ* in powder feeding process, due to its capability for addressing the low Reynolds number turbulence and near wall internal flow. Furthermore, there are no normalized metric features to quantitatively characterize the concentration distribution of the powder stream. How to achieve a powder stream with improved convergence and thermal efficiency also remains underexplored. To bridge this gap, we proposed a systematic approach to investigate the thermal dynamics of the powder stream in DED using CFD modeling.

In this study, a three-dimensional two-phase flow numerical model was developed to investigate the complex thermodynamic behavior of the powder feeding process which considers the particle size, powder-gas interaction, laser irradiation, and beam divergence. A more comprehensive *k-ω* model was utilized to solve the low Reynolds number turbulence of the gas phase. High-speed imaging and infrared camera observations were also carried out to validate the theoretical model. Moreover, based on two-phase fow simulation, the matching relationship between the powder flow and the laser beam was discussed to achieve a higher thermal efficiency which is commonly difficult to be determined experimentally. The theoretical powder heating efficiency shows a clear improvement from 39.7 to 55.9%. The proposed method provides an efective way for enhancing the convergence and thermal efficiency of the DED powder stream, and the simulation prediction can be used to further build the high-fdelity model of the entire DED process.

### **2 Methodology**

The CFD numerical modeling of powder feeding process is described in Sects. [2.1](#page-2-0) and [2.2.](#page-4-0) Experimental validation is described in Sects. [2.3](#page-4-1) and [2.4.](#page-6-0)

#### <span id="page-2-0"></span>**2.1 Numerical modeling of powder feeding**

The powder gas flow is considered as a dispersed multiphase flow, which regards the gas and the powder as the continuous phase and discrete phase, respectively. In this paper, the Euler–Lagrange [[29\]](#page-16-7) method is utilized to model the gas-powder flow. A one-way coupling strategy is used to solve the momentum transfer between these two phases due to the small powder size with dilution fraction less than 10% in gas. To save computational costs, while maintaining the representability of simulation, some appropriate assumptions are made as follows:

- (1) The continuous gas phase is in a steady state, as the incompressible Newtonian fuid.
- (2) All powder particles are spherical, and the particle diameter conforms to a Gaussian distribution with a mean diameter of 80  $\mu$ m and a variance of 10  $\mu$ m.
- (3) The interaction between the powder particles and nozzle walls is completely elastic.
- (4) The model only accounts for gravity, drag force, and inertia.
- (5) The laser beam intensity conforms to a Gaussian distribution.
- (6) The evaporation and radiative heat dissipation from particles are neglected.
- (7) The temperature inside the particle is uniform.

#### **2.1.1 Gas fow governing equations**

The continuous gas flow can be described by the incompressible, Newtonian continuity equation and the momentum conservation equation in fuid mechanics. Also called the Navier–Stokes (NS) equation as follows:

Conservation equation of mass:

$$
\nabla \cdot \mathbf{u} = 0 \tag{1}
$$

Conservation equation of momentum:

$$
\frac{\partial \rho u}{\partial t} + \nabla \cdot (\rho u u) = -\nabla p + \nabla \cdot \tau + \rho g \tag{2}
$$

$$
\tau = \mu(\nabla u + \nabla u^T) - \frac{2}{3}\mu I(\nabla \cdot u)
$$
\n(3)

where the bold formatted quantities represent vectors;  $\rho$  is the density of the gas phase which is a constant due to the low Mach number and relatively low pressure drop. ∇ is the Hamiltonian; *t* and *p* are the time and pressure, respectively;  $u$  is the velocity vector;  $g$  is the gravitational acceleration of 9.8 m/s<sup>2</sup>;  $\tau$  is the viscous stress tensor described by Eq. ([3\)](#page-2-1), in which  $\mu$  is the dynamic viscosity. The flow state of a fuid is determined by the Reynolds number and defned as:

<span id="page-2-2"></span>
$$
Re = \frac{\rho u L}{\mu} \tag{4}
$$

In Eq. ([4](#page-2-2)), the *Re* of the nozzle gas flow can be estimated to be approximately 4056, the gas density  $\rho$  is 1.634 kg/m<sup>3</sup>, the averaged gas velocity is 4 m/s, the nozzle exit diameter is 14 mm, and the dynamic viscosity is obtained as 22.56 µPa∙s. The calculated *Re* is higher than the critical value of 2300, indicating the dominance of inertial effects over the viscous effects in gas flow. Therefore, the velocity oscillations and violent mixed transport phenomena in the continuous phase fow will occur, making the solution of the pure NS equation computationally intensive.

To accurately and efectively model the DED gas fow turbulence, the RANS approach is usually applied to decompose the fow to the average and fuctuating component. The Boussinesq eddy viscosity approximation [\[28](#page-16-6)] computed the Reynolds stress tensor as follows:

$$
-\rho \overline{\mathbf{u} / \mathbf{u}'} = \mu_T (\nabla \overline{\mathbf{u}} + \nabla \overline{\mathbf{u}}^T) \tag{5}
$$

where  $u'$  is the corresponding pulsation velocity;  $\bar{u}$  is the averaged velocity;  $\mu_T$  represents the eddy viscosity which needs to be modeled further. In this study, the *k-ω* turbulence model [[30\]](#page-16-8) is selected due to its better performance and compatibility for wall bounded and low-*Re* turbulence of coaxial nozzle gas fow. The *k-ω* model is capable to transform the pulsating velocity into an instantaneous average velocity by solving the transport equations for turbulence kinetic energy (*k*) and energy dissipation rate (*ω*). The detailed information can be seen in Ref. [[31\]](#page-16-9).

#### <span id="page-2-3"></span>**2.1.2 Powder fow governing equations**

<span id="page-2-1"></span>As mentioned above, the momentum transfer between the gas and particles is described in terms of one-way coupling strategy. The trajectory, velocity, and forces of the particles are solved using the COMSOL [[32\]](#page-16-10) particle tracking interface in which the drag force, gravity, particle collisions, and inertial forces are taken into account for particle dynamics. The governing equations of the powder particle dynamics are given as follows:

$$
\frac{d\mathbf{v}}{dt} = \frac{1}{\tau_p}(\mathbf{u} - \mathbf{v}) + \frac{\rho_p - \rho}{\rho_p}\mathbf{g} + \frac{4E}{3(1 - \text{pois}^2)}\sqrt{\frac{d_p\delta^3}{2}}\tag{6}
$$

$$
\tau_p = \frac{4\rho_p d_p^2}{3\mu C_D Re_r} \tag{7}
$$

where the bold formatted quantities represent vectors; *u* and *v* are the velocity vectors; the second term on the right-hand side of Eq. ([6\)](#page-3-0) is the particle gravity which considers the effect of buoyancy, while the first term describes the drag force to obtain the turbulence efect on particles. And the third term is the particle collision forces based on classical Hertz theory [[33\]](#page-16-11) where *E* is Young's modulus, *pois* is Poisson's ratio, and  $\delta$  is the particle deformation.  $\tau_p$  is the velocity response time, which needs to be further modeled according to the relative Reynolds number *Rer*:

$$
Re_r = \frac{\rho |u - v| d_p}{\mu} \tag{8}
$$

The particle shapes are not all perfectly spherical in DED, thus the Haider-Levenspiel [[34\]](#page-16-12) drag coefficient was used by considering the sphericity of the particles. This coefficient is given by:

$$
C_D = \frac{24}{Re_r}(1 + A * Re_r^B) + \frac{C}{1 + D/Re_r}
$$
(9)

where *A*, *B*, *C*, and *D* are all dimensionless empirical correlations of the particle sphericity defned as follows:

$$
A = \exp(2.3288 - 6.4581S + 2.4486S^2)
$$
  
\n
$$
B = 0.0964 + 0.5565S
$$
  
\n
$$
C = \exp(4.905 - 13.8944S + 18.4222S^2 - 10.2599S^3)
$$
  
\n
$$
D = \exp(1.4681 + 12.2584S - 20.7322S^2 + 15.8855S^3)
$$

In this study, the sphericity *S* is set to be 0.5, which defnes the surface ratio of a volume equivalent sphere to the considered non-spherical particle.

Besides, the turbulence difusion by instantaneous fow fuctuations is the main mechanism for particle dispersion and deposition. Hence, a stochastic discrete random walk (DRW) [[35\]](#page-16-13) approach is used to describe the particles turbulent dispersion in which the drag force is computed by frst adding a random velocity perturbation to the mean fow feld based on the local value of the kinetic energy *k*.

To characterize the convergence of the powder stream, a user-defned function is chosen to calculate the concentration distribution of particles from the Lagrange trajectory using the particle source in cell (*PSI-C*) [[36\]](#page-16-14):

$$
C_j = \frac{\dot{M}\sum_{i=1}^n dt_{(i,j)}}{V_j} \tag{10}
$$

<span id="page-3-0"></span>where  $\dot{M}$  is the number flow rate of each trajectory,  $V_j$  is the volume of a computational cell, *dt* is the particle residence time, and the subscript (*i,j*) refers to the *i*th trajectory and the *j*th cell.

#### **2.1.3 Powder heating**

Additionally, the particles are heated by laser irradiation during fight and even undergo phase changes such as melting, while dissipating heat in the forms of convection and radiation. During the fight of particles, the energy conversation equations are given as follows:

<span id="page-3-1"></span>
$$
m_p C_p \frac{\partial T}{\partial t} = \alpha I A_p - h S_p (T - T_0) - m_p L \frac{df_l}{dt}
$$
\n(11)

where  $m_p$  is the particle mass;  $C_p$  is the specific heat of powder particle;  $T$  and  $T_0$  are the temperatures of particles and gas environment, respectively;  $\alpha$  is the laser absorption coefficient of particles; *I* is the laser intensity;  $A_n$  is the effective irradiated area of powder particles on laser beam;  $S_n$  is the surface area of powder particle; *h* is the convective heat transfer coefficient; *L* is the latent heat of fusion; and  $f_l$  is the liquid phase fraction.

The dimensionless number *Bi* describes the distribution pattern of the temperature feld inside an object under nonstationary thermal conductivity, which is defned as:

$$
Bi = \frac{h}{k_p} L_C \tag{12}
$$

where  $L_c$  is the characteristic length, typically the ratio of particle volume to surface area, and  $k_p$  is the thermal conductivity. Note that the temperature inside the particle can be assumed as uniform and calculated as a single number rather than a temperature distribution due to a small value of *Bi*. The convective heat transfer coefficient  $h$  is specified by the Nusselt number (*Nu*) [\[37](#page-16-15)], which is written as:

$$
Nu = \frac{hd_p}{k_g} 2 + 0.6Re^{0.5} Pr^{0.33}
$$
\n(13)

where  $k_g$  is the thermal conductivity of gas;  $Pr$  is the Prandtl number, written as:

$$
Pr = \frac{C_{p,g}\mu}{k_g} \tag{14}
$$

where  $C_{p,g}$  is the specific heat of gas.

Some particles may undergo phase transitions under laser irradiation during fight. To accurately model the temperature evolution of particles, the liquid fraction method is utilized to describe the phase change heat transfer. Such method adds the latent heat of phase transition as a source term into Eq.  $(11)$ . A linear function is used to describe

 $\overline{a}$ 

$$
f_l(T) = \begin{cases} 0, & T < T_s \\ \frac{T - T_s}{T_l - T_s}, & T_s \le T \le T_l \\ 1, & T > T_l \end{cases} \tag{15}
$$

where  $T_s$  and  $T_l$  are the solidus and liquidus temperatures, respectively. The derivative of  $f_l$  with respect to the time  $t$ can be expressed as:

$$
\frac{df_l}{dt} = \begin{cases}\n0 & T < T_s, T > T_l \\
\frac{1}{T_l - T_s} \frac{dT}{dt} & T_s < T < T_l\n\end{cases}
$$
\n(16)

The intensity of laser beam conforms to a Gaussian distribution at the focal plane and the planes which are parallel to the focal plane. It can be calculated from Eq. ([17\)](#page-4-2):

$$
I = \frac{3P}{\pi R_b^2(z)} \exp(-\frac{3(x^2 + y^2)}{R_b^2(z)})
$$
(17)

where *P* is the laser power, and *x* and *y* are the Cartesian coordinates. In general, the convergence and divergence of the laser beam have a crucial efect on the powder stream heating efficiency. Hence, the laser beam effective radius  $R_b(z)$  is considered as a piecewise function with respect to *z*. In this study, the laser beam with a radius of 1 mm at the focal plane is firstly focused at  $z = -9$  mm with a half-angle  $\varphi$ =3.6°, followed by a depth of field (DOF) of 4 mm, and then diverged with an identical half-angle.  $R_b(z)$  can be calculated by Eq. ([18\)](#page-4-2):

$$
R_b(z) = \begin{cases} (z+9) * \tan(\varphi) + 1 \ z \in (-9, +\infty) \\ 1 \qquad z \in [-13, -9] \\ (-13 - z) * \tan(\varphi) + 1 \ z \in (-\infty, -13) \end{cases} (18)
$$

The calculated laser beam intensity profle is depicted in Fig. [1](#page-4-3), where the intensity distributes more reasonably than that produced by using a conventional heat source.

#### <span id="page-4-0"></span>**2.2 Boundary conditions**

A symmetrical 3D numerical model of the powder feeding process with annular nozzle is developed through COMSOL [\[32\]](#page-16-10) software. The 3D geometry of the model and some of the boundary conditions are shown in Fig. [2a](#page-5-0). The nozzle consists of an inner wall and an outer wall with two inlet planes, which are the transport channels of the carrier gas, powder, and protective gas. To investigate the kinetic behavior of the powder flow below the nozzle, a sufficiently large cylindrical free space is provided. Only half of the model is calculated to save computational cost due to the geometric



<span id="page-4-3"></span><span id="page-4-2"></span>**Fig. 1** Laser beam intensity profle with a radius of 1 mm at the focal plane, a half-angle of 3.6°, and a laser power of 800 W

symmetry of the nozzle. The corresponding boundary conditions are summarized as follows:

- (1) Set velocity inlet and fully developed boundary conditions on both inlet planes according to the corresponding gas volume flow rates.
- (2) Set non-slip boundary conditions on the nozzle walls.
- (3) Pressure exit boundary conditions are set on the exit planes.
- (4) Set a symmetry boundary condition on the symmetry plane.

The detailed input parameters of the boundary conditions and material physical properties are shown in Tables [1](#page-5-1) and [2](#page-5-2), respectively. About 5000 powder particles are randomly released from the inlet plane to fully investigate the particle movement and powder stream concentration.

To accurately and efficiently capture the flow field, temperature feld, and powder concentration distribution, a highquality 3D structured tetrahedral mesh is generated, and the nozzle walls and the z-axial parallel region are appropriately local refned. Totally 130,000 cells with a minimum cell mass of 0.1682 are generated and presented in Fig. [2](#page-5-0)b.

#### <span id="page-4-1"></span>**2.3 Powder feeding experiment setup**

In this study, a powder delivery observation experiment was set up for model validation. Commercially



<span id="page-5-0"></span>**Fig. 2** Modeling set-up: **a** geometry and boundary conditions and **b** mesh of the computational domain

<span id="page-5-1"></span>**Table 1** Boundary conditions for the simulation of DED powder feeding in coaxial nozzle

Boundary planes	Boundary conditions
Carrier gas inlet	Gas flow rate: $6, 8, 10$ L/min
Protective gas inlet	Gas flow rate: 10, 15 L/min
Particles inlet	Powder flow rate: 6.54 g/min
Outlet	Pressure (relative): 0 Pa

<span id="page-5-2"></span>**Table 2** Thermo-physical properties of Inconel 718 and argon used in the simulation



gas-atomized Inconel 718 powders were used as the feeding materials whose stated diameter is within the range from 53 to 150 µm. The particles are spherically shaped with the size of  $D10 = 49.1$  µm,  $D50 = 73.5$  µm, and  $D90 = 124.5$  µm according to actual measurement. The chemical compositions of Inconel 718 powders are shown in Table [3.](#page-5-3) Prior to experiments, the Inconel 718 powders were dried in a vacuum oven at  $150 \pm 10$  °C for 2 h.

To systematically observe the thermal dynamic behavior of the powder flow, a high-speed imaging experiment system was established to record the powder stream in coaxial nozzle. As shown in Fig. [3,](#page-6-1) the DED experimental system used in this study is built around a 2000-W fber laser (IPG, YLS-2000-CT) with a wavelength of 1070 nm as the heat source. The designed DED paths were performed by a 6-axis robot (KUKA, KR30HA). A coaxial nozzle (PRECITIEC, YC52) and a powder feeder with three hoppers (RAYCHAM, RC-PGF-D) were used for powder delivery as it has high conveying efficiency and uniformity. Argon gas of 99.99% high-purity was utilized to deliver metal powders and provide a protective atmosphere.

Besides, a high-speed camera (Photron FASTCAM SA-X2) with a 200-mm fxed-focus macro lens was used

<span id="page-5-3"></span>



<span id="page-6-1"></span>**Fig. 3** Schematic of powder stream observation experiment

<span id="page-6-2"></span>**Table 4** Parameters of the high-speed camera used in this study

Resolution	Full resolu-	Recording	Dynamic	Exposure
	tion	rate	range	time
$1024 \times 1024$ 1000 fps		$1$ GB/s	10 bits	$300 \text{ ns} \sim 5 \text{ s}$

to capture the images and videos of the powder stream. During the powder feeding experiment, the high-speed camera was placed along the x-axis with the focal point located approximately in the powder convergence region to keep the powder stream in the center feld of view. The axis of the nozzle is perpendicular to the XOY horizontal plane and a uniform blue light source illuminates the powder stream along the y-axis. Meanwhile, a dark screen was placed opposite to the camera to make powder stream more visible. The camera parameters are shown in Table [4.](#page-6-2) Furthermore, an infrared camera (InfRec NS9500, temperature measurement range 0–2000 °C) was used to detect the powder stream temperature.

To validate the numerical model, four levels of fow rates 8, 10, 12, and 15 L/min were chosen for the carrier gas and protective gas to fully investigate the efect of gas fow rates on the powder stream convergence. The powder feeding rate was fxed at 6.54 g/min which equivalents to 1 r/min of the powder feeder rotation speed. The laser beam has a radius of 1 mm, a half angle of 3.6°, and a focal length of 9 mm. The laser power was set from 600 to 1200 W in steps of 200

W for this study. During the photography, the powder feeding process was photographed for 1 s at a shutter speed of 0.001 s once it already reached a steady state.

## <span id="page-6-0"></span>**2.4 Image processing‑based powder stream measurement**

The clear and sharp images are captured by the high-speed camera in powder feeding experiment. Image processingbased methods were used to characterize the powder stream convergence and concentration distribution. According to the Mie theory [[38](#page-16-16)], the powder stream concentration is proportional to the gray value of the image under constant powder feeding parameters and photography conditions. To reduce the effect of background impurities and light source inhomogeneities, the acquired powder stream images are treated by the background correction using Eq. ([19](#page-6-3)), where *I* is the original image, *B* is the background image without powder feeding, and *I*′ is the preprocessed image.

<span id="page-6-3"></span>
$$
I'(i,j) = I(i,j) - B(i,j)
$$
\n(19)

In addition, the highly random and discrete distribution of powder particles in the original image is not accurate for analyzing the powder stream convergence. Thus, the random particle trajectories are considered as probability distributions, which will reach a steady state after a long enough time. In this study, nearly 500 consecutive images are averaged to weaken the random efect of particle trajectories.

Moreover, the powder cluster nominal radius  $R_p$  is defined to characterize the powder stream aggregation. And *p* is the ratio of the powder amount within the efective central area to the whole powder spot, and it ranges from 0 to 1. A smaller  $R_p$  means the better powder stream convergence when  $p$  is constant. As shown in Fig. [4](#page-7-0), the powder stream concentration distribution on a x–y horizontal section is divided into  $V_0$  and  $V_1$  by the red grid. Here, the volume  $V_0$ represents the quantity of powder within the efective central area. Thus, the ratio *p* can be expressed as:

$$
p = \frac{V_0}{V_1 + V_0} = \frac{\int_0^{R_p} 2\pi \rho(r) dr}{\int_0^{+\infty} 2\pi \rho(r) dr}
$$
(20)



<span id="page-7-0"></span>

where the denominator is the powder amount of the whole spot, and  $\rho(r)$  is the gray value at *r* distance from the center point. In this study, the ratio *p* is set to be 0.7, and to obtain the  $R_p$  in radial direction, an iterative calculation is performed until  $p$  is greater than 0.7. During calculation, the initial value and increment of  $R_p$  are set as 0 and image pixel size, respectively. The distance between the minimum  $R_p$ position and the nozzle outlet is defned as the consolidation distance of the powder stream.

# **3 Results and discussion**

#### **3.1 Prediction of powder stream convergence**

Figure [5](#page-7-1) presents the computed particle trajectories during DED powder feeding process, which display a waist-like appearance. Given the symmetrical nature of the annular nozzle, only half of the geometry is considered in computations. The powder particles frstly enter the gap between the inner and outer nozzles from the top inlet, and then collide with the nozzle walls. These collisions cause the particles to be frequently refected and redirected, making the particle trajectories below the nozzle outlet unpredictable based solely on the nozzle geometry. As the particles advance, they are accelerated by the drag force of the high-speed argon gas and gravity, resulting in a continuous momentum transfer. Upon ejection from the nozzle outlet, the powder stream begins to converge, and at a certain in-fight distance, a high-concentration powder cluster forms. Beyond **Fig. 4** Schematic of defning nominal powder stream radius this region, the powder stream starts to diverge. The particle



<span id="page-7-1"></span>**Fig. 5** Simulated particle trajectories colored by velocity with an inner gas fow rate of 15 L/min and diferent carrier gas fow rates: **a** 8 L/min, **b** 10 L/min

trajectories reveal a consistent distribution pattern at diferent gas fow rates. The maximum velocity of the powder particles increases from 2.39 to 2.82 m/s with increasing the carrier gas flow rate from 8 to 10 L/min. This indicates that the carrier gas serves as the primary driving force for the powder feeding process.

Figure [6](#page-8-0) illustrates the particle concentration distribution at various XY cross-sections by using the method in Sect. [2.1.2](#page-2-3). The cross-sectional distribution of powder concentrations aligns with the form of particle trajectories depicted in Fig. [5.](#page-7-1) The powder stream below the nozzle outlet can be categorized into three stages: pre-convergence, convergence, and divergence. In the initial stage, the powder concentration distribution on the XY section presents a ring-like shape, with the ring radius continuously shrinking due to the confnement of the inner and outer nozzle walls. Consequently, the powder particles gravitate towards the convergence region upon exiting the nozzle. In the second stage, the powder flow commences converge at  $z = -8$  mm and attains a peak concentration of 69.8 kg/m<sup>3</sup> at *z* of approximately − 11 mm. The concentration of the powder stream aggregates at a single point, indicating the distinct characteristics of powder flow convergence, as shown in Fig. [6b](#page-8-0). In the fnal stage, as the



<span id="page-8-0"></span>**Fig.** 6 Powder concentration distribution on different x–y cross sections:  $a z = -5$  mm,  $b z = -10$  mm,  $c z = -15$  mm, and  $d z = -20$  mm



<span id="page-9-0"></span>**Fig. 7** Powder concentration distribution on diferent x–y cross sections at diferent carrier gas fow rates: **a**, **c** 8 L/min, **b**, **d** 10 L/min

powder flow progresses, the powder concentration swiftly reduces to around  $14.3 \text{ kg/m}^3$ . Figure [6d](#page-8-0) confirms that the apparent divergence of cross-sectional powder concentration appears at  $z = -20$  mm.

The powder stream concentration distributions exhibit similar convergence and divergence stages across diferent carrier gas fow rates. As illustrated in Fig. [7,](#page-9-0) when the carrier gas fow rate increases from 8 to 12 L/min, the maximum powder concentration within the cross-section at an equivalent distance from the nozzle outlet diminishes, i.e., the maximum concentration of  $15.73 \text{ kg/m}^3$  in Fig. [7a](#page-9-0) exceeds that of  $12.78 \text{ kg/m}^3$  in Fig. [7b](#page-9-0). This variance is signifcant between Fig. [7](#page-9-0)c and d. The increased carrier gas fow rates intensify the collision between the

powder particles and nozzle walls, resulting in a more uniform concentration distribution of the powder stream.

#### **3.2 Prediction of powder stream thermal history**

Figure [8](#page-10-0) illustrates the trajectories of powder particles under two diferent laser powers, which are colored by the magnitude of temperature. These particles begin to enter the laser irradiation area at approximately 8 mm below the nozzle exit, as indicated by a signifcant elevation in the temperature. When they pass through the DOF of the laser beam, they are rapidly heated to high temperatures. As the particles continue to move forward, the heated particles disperse in a free space. As shown in Fig. [8](#page-10-0)b, the overall temperature of



<span id="page-10-0"></span>**Fig. 8** Powder stream temperature distribution at diferent laser powers: **a** 800 W, **b** 1000 W



<span id="page-10-1"></span>**Fig. 9** The average temperature of the powder stream along the z-axis under diferent laser powers

the particle trajectories at a laser power of 1000 W is signifcantly higher than that at a laser power of 800 W presented in Fig. [8a](#page-10-0). Under the same interaction duration within laser irradiation area, the laser intensity is signifcant for the laser heating effect. Figure [9](#page-10-1) presents the variation curve of the average particle temperature along the z-axis under diferent laser powers. A large gradient in temperature is noticeable around the laser focal plane, followed by a gradual heating zone until the maximum temperature is achieved. Beyond this region, the particle temperature tends to be steady. It is evident that particles exhibit a similar thermal history trend across diferent laser powers, with a higher laser power leading to a higher maximum particle temperature.

To comprehensively understand the heating process involving multiple particles, temperature distributions with respect to the radial distance colored by particle velocities are presented in Fig. [10,](#page-11-0) where the particle temperature and velocity data are extracted from four different cross sections along z-axis. As depicted in Fig. [10](#page-11-0)a, the particles are located at the periphery of the laser beam and thus remain at room temperature. As the particles advance to the *z* of − 10 mm, the particle temperature distribution presents a dromedary pattern, with the highest temperature at the center and a gradual decay along the radial direction, as shown in Fig. [10](#page-11-0)b. Notably, the particle temperature in the central area sharply rises to approximately 1300 K, which is a clear indication of the laser heating effect.

Owing to the divergence of the powder stream, a subset of particles is observed at positions that are distant from the z-axis, as shown in Fig. [10c](#page-11-0). Some of these particles traverse the high laser intensity region, interacting with the laser beam for an extended duration. Therefore, their temperatures are higher than those of particles exposed to a shorter irradiation time in the low laser intensity region. This phenomenon is still present in Fig. [10d](#page-11-0). Furthermore, a portion of particles undergoes melting phase transformation as their temperature exceeds the solidus temperature of 1528 K. The overall particle temperature continues to increase due to the intense laser irradiation, and the heated powder stream is in a state of complete divergence. The radial temperature distribution of the particles corresponds to the thermodynamics illustrated in Figs. [6](#page-8-0) and [7.](#page-9-0) In terms of the radial distribution, the distribution of the powder stream signifcantly infuences the thermal profle of the particles.



<span id="page-11-0"></span>**Fig. 10** Particle temperature scatters colored by velocity on different cross sections at a laser power of 800 W:  $a z = -5$  mm,  $b z = -10$  mm, **c** *z*= −15 mm, and **d** *z*= −20 mm

# **3.3 Efect of laser‑powder ofset on thermal efficiency**

The discrepancy between the powder consolidation distance and the laser focal length signifcantly afects the thermal interaction between the powder stream and the laser beam. In this study, we flexibly adjusted this offset to investigate the DED powder heating efficiency that is commonly difficult to be determined experimentally. We defne the powder heating efficiency as the proportion of particles heated to the melting temperature within a space of 0.5 mm in height and 1.5 mm in radius near the consolidation plane.

Figure [11](#page-11-1) illustrates the variation of particle heating efficiency against ofset. A negative ofset indicates that the laser beam focuses ahead of the powder stream. With a negative offset, the powder particles have long interaction time with the laser beam before reaching the substrate, owing to the existence of the DOF. Conversely, at a positive offset, the laser beam has not yet begun to focus when the powder stream reaches the convergence region.



<span id="page-11-1"></span>Fig. 11 Variations of powder heating efficiency against the offset between laser beam and powder stream

The maximum powder heating efficiency, 55.9%, appears at an ofset of−4 mm, demonstrating a signifcant improvement of 16.2% compared to the mode of simultaneous focusing. A better alignment between the powder consolidation distance and the laser beam can enhance the powder stream heating efficiency, thereby improving the powder capture efficiency.

## **3.4 Experimental determination of powder stream convergence**

Figure [12a](#page-12-0) depicts the particles as white bright spots after the threshold segmentation binarization process. Upon exiting the nozzle, the particles have a mean velocity of

1.5 m/s and are further accelerated by trailing and gravitational forces to a maximum velocity of 2.8 m/s. These values are calculated based on the length of trajectories. In Fig. [12](#page-12-0)b, the averaged image is relatively smooth, and the powder concentration displays improved distribution characteristics compared to that in Fig. [12](#page-12-0)a. This suggests that the preprocessing operations of background correction and homogenization efectively mitigate the impacts of particle trajectory randomness. Figure [12c](#page-12-0) presents a contour plot of the powder concentration distribution, which clarifes the efective area of the powder stream. The highest powder concentration is observed at the consolidation position. Also, it is reasonable that the convergence of powder streams can be well described by the nominal powder size  $R_p$ .



<span id="page-12-0"></span>**Fig. 12** Powder stream image preprocessing: **a** an original particle image, **b** an averaged particle image, and **c** the powder stream concentration contour



<span id="page-12-1"></span>**Fig. 13** Powder stream concentration profles: **a** diferent distances below the nozzle exit and **b** normalized profle comparison at an in-fight distance of 10 mm

Given a carrier gas volume flow rate of 10 L/min and a shielding gas volume flow rate of 15 L/min, five profiles of the gray value at varying distances from the nozzle outlet are extracted to quantitatively illustrate the powder stream distribution. As shown in Fig. [13a](#page-12-1), at a distance of 5 mm from the nozzle outlet, the powder concentration profle exhibits a bimodal shape with low gray values. As the in-fight distance increases, the distribution curve exhibits a peak with a high gray value, indicating that the powder particles gradually aggregate to a single point.

Obviously, the consolidation distance of the powder stream lies within the range of 10–12 mm below the nozzle exit. Beyond this region, the powder stream diverges, which will not contribute to the formation of the clad. As shown in Fig. [13](#page-12-1)b, the comparison of the normalized concentration profle reveals a good agreement between the experimental and simulated results.

# **3.5 Efect of gas fow rates on powder consolidation distance**

The role of auxiliary gas in infuencing the consolidation distance of the powder stream is signifcant. In this study, three diferent gas fow rates (8, 10, and 12 L/min) are selected to examine the interplay between the inner and carrier gas. The consolidation distances, both simulated and experimental, were determined using the method in Sect. [2.4](#page-6-0).

Figure [14](#page-13-0) illustrates the variation of the powder stream consolidation distance in relation to the inner and outer gas flow rates in the coaxial nozzle. As shown in Fig. [14a](#page-13-0), an increase in the inner gas fow rate results in a further movement of approximately 0.2 mm the consolidation distance from the nozzle outlet. This displacement helps to prevent the premature convergence of the powder flow due to the inner gas's blowing efect. On the other hand, an increase in the carrier gas fow rate, especially when it exceeds the inner gas fow rate, gradually reduces the consolidation distance of the powder stream, thereby enhancing the quality of powder stream convergence, as depicted in Fig. [14](#page-13-0)b. The minor discrepancy depicted in the fgure remains within 5%, indicating a good agreement between the simulated and experimental results. The constant offset between the two is justifiable, considering that the simulation has not yet accounted for the energy loss from particle–wall collisions.

## **3.6 Comparison between the experimental and simulated data**

The simulated results are validated by the experiment described in Sect. [2.3.](#page-4-1) As seen from Fig. [15](#page-14-0)a, both the experimental and simulated results demonstrate a fact that as the in-fight distance extends from 10 to 40 mm, the powder temperature increases from 1000 to 1800 K, and then decreases from 1800 to 1600 K. This pattern refects the heating of the powder particles in the laser irradiation region, followed by a cooling stage where heat is dissipated through convection and radiation during free flight. The  $R^2$ value of 0.96 for the average temperature prediction demonstrates a good agreement between the experimental and simulated data. Furthermore, Fig. [15b](#page-14-0) depicts a consistent increase in the mean velocity of the powder particles with the expansion of the in-flight distance. An  $R^2$  value of 0.91 for the velocity prediction also indicates a good consistency of experimental and simulated results.



<span id="page-13-0"></span>**Fig. 14** Infuences of gas fow rates on the powder stream consolidation distance: **a** a carrier gas fow rate of 10 L/min and **b** an inner gas fow rate of 10 L/min



<span id="page-14-0"></span>**Fig. 15** Comparison between the experimental and simulated results: **a** averaged particle temperatures and **b** averaged particle velocities



<span id="page-14-1"></span>**Fig. 16** Comparison of powder stream nominal radius ( $R_p$ ,  $p=0.7$ ) along z-axis

In addition, the method outlined in Sect. [2.4](#page-6-0) was employed to assess the convergence of the powder stream. Figure [16](#page-14-1) depicts the variation of the nominal radius  $(R_p)$  of the powder stream along the z-axis, with the two curves representing the experimental and simulated results. Initially, the powder stream has a large  $R_p$  value, which is indicative of a low powder concentration. As the in-fight distance expands, the  $R_p$  value gradually diminishes until reaching a minimum of 2.8 mm, implying the consolidation of the powder stream at approximately 11.2 mm. After this position, the  $R_p$  value increases due to the divergence of the powder stream. Both the simulated and experimental results display a U-shaped curve along the in-fight direction. This reveals the three stages of powder stream evolution, i.e., the pre-convergence, convergence, and divergence. The comparison verifes a good agreement between the simulated and experimental results, particularly in the region of powder stream convergence. During the experiment, some particles situated in the defocused area of the camera were not captured, whereas the simulation accurately accounted for all particles. Thus, the difference of the  $R_p$  in pre-convergence and divergence region resents reasonably.

## **4 Conclusions**

In this study, a 3D numerical model is developed to systematically explore the intricate interactions among the powder, gas, and laser beam. A quantitative analysis is conducted on the structure, movement, and heating behavior of the powder stream. The experimental observation based on image processing is also carried out to validate the model. The conclusions are summarized as follows:

- 1) The simulation results identify the three stages of the powder flow, i.e., pre-convergence, convergence, and divergence. The maximum powder concentration is found to be 68.9 kg/m<sup>3</sup>, and the minimum  $R_p$  is 2.8 mm at an in-fight distance of 11.2 mm.
- 2) The average velocity and temperature of the powder stream at specifc distances below the nozzle outlet have a good agreement between the experimental and simulated results, with  $R^2$  values of 0.91 and 0.96, respectively.
- 3) The gas fow rate has a certain efect on the powder stream convergence. The carrier gas facilitates better consolidation of the powder stream, whereas the blowing

efect of the inner gas tends to shift the consolidation distance away from the nozzle outlet.

4) The simulation reveals that a laser heating efficiency of 55.9% could be achieved when the laser beam is focused at 4 mm ahead of the powder stream. The efficiency is signifcantly improved of 16.2% compared to the case of zero offset.

This study provides an efective approach to comprehend and improve the convergence and heating efficiency of the DED powder feeding process by numerical simulation. In the future, the current model's capabilities of accurate predictions of the powder velocity and temperature will be utilized as inputs to further investigate the thermodynamic efects of particles on the molten pool.

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**Data availability** The data that support the fndings of this study are available from the corresponding author upon reasonable request.

#### **Declarations**

**Ethical approval** Not applicable.

**Consent to participate** The manuscript has been read and approved by all authors.

**Competing interests** The authors declare no competing interests.

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