**ORIGINAL ARTICLE**



# **Microstructure‑based fnite element modeling of a martensitic stainless steel during hot forging**

Simin Dourandish<sup>1</sup> · Henri Champliaud<sup>1</sup> · Jean-Benoit Morin<sup>2</sup> · Mohammad Jahazi<sup>1</sup>

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### **Abstract**

This study aims to analyze material fow behavior and microstructure evolution during the hot deformation of a martensitic stainless steel with the view to simulating the open die forging process. For this purpose, hot compression tests were conducted using a Gleeble-3800 thermo-mechanical. A material model, based on the Arrhenius equation, was developed and implemented into the fnite element simulation code Forge NxT 3.0® using a special user subroutine. Microstructure maps for the entire volume of the specimen were determined, and the occurrences of diferent softening phenomena were predicted as a function of hot deformation parameters. Good correlations were obtained between the simulation predictions and experimental results. The fow behavior of the material was also analyzed using the dynamic material model, and the unstable regions were identifed. The obtained results were compared with the experimental and numerical simulation fndings. The approach proposed in this paper, which integrates microstructure-based fnite element simulation combined with a three-dimensional processing map, can be used for rapid and accurate optimization of the hot deformation process of martensitic stainless steels.

**Keywords** Martensitic stainless steels · Hot deformation · Constitutive equation · Microstructure-based fnite element  $simulation \cdot Three-dimensional (3D) thermal processing \cdot Instability coefficient$ 

# **1 Introduction**

Hot forging of large size ingots made of martensitic stainless steels (MSS) is a challenging processing step during the manufacturing of structural components that require high strength, high toughness, and good corrosion resistance such as valves, shafts, bearings, rotors, and bolting of gas turbine blades  $[1-3]$  $[1-3]$ . The main concern during the forging of these steels is the occurrence of large variations in the microstructure (e.g., austenite grain size) between the center and the surface of the forged part [\[4](#page-17-1)]. Such variations are at the source of cracking during the hot forging operation or

 $\boxtimes$  Simin Dourandish simin.dourandish.1@ens.etsmtl.ca

> Mohammad Jahazi mohammad.Jahazi@etsmtl.ca

<sup>1</sup> Department of Mechanical Engineering, École de Technologie Supériere, 1100 Notre Dame West, Montreal, Quebec H3C 1K3, Canada

<sup>2</sup> Finkl Steel-Sorel Inc., 100 McCarthy, Saint-Joseph-de-Sorel, Quebec J3R 3M8, Canada

can lead to unacceptable variations in properties in diferent regions of the fnal product.

The manufacturing process consists of alloy production in electric arc furnaces, followed by ingot casting and open die forging. During the latter step, the heterogeneous cast structure is broken down into a more refned microstructure and improved chemical homogeneity through the thickness of the large-size forged block  $[5-8]$  $[5-8]$  $[5-8]$ . The flow characteristic of a hot forging process consists of competing mechanisms of strain hardening, also called work hardening (WH), and softening phenomena such as dynamic recovery (DRV) and dynamic recrystallization (DRX). These mechanisms are afected by hot working parameters like temperature, strain, and strain rate  $[9-12]$  $[9-12]$ . So, optimizing these working parameters, related to metal forming, is of critical importance for the heavy forging industries, and therefore, the hot forging process should be designed carefully to obtain both the right shape and microstructure.

To accurately predict microstructure evolution and fow stress behavior during the ingot breakdown process at high temperatures, the development of reliable constitutive models is essential [[13](#page-17-6)[–15](#page-17-7)]. In this regard, isothermal hot compression tests are typically used to simulate the material response to thermomechanical processes, and then constitutive material models are developed to describe the stress–strain curves at diferent temperatures and strain rates [[16,](#page-17-8) [17\]](#page-17-9). Different models have been developed to assess the individual and mutual infuences of hot working process parameters on the fow stress evolution of stainless steel. Ying Han et al. [[18\]](#page-17-10) studied the hot deformation behavior of 904L superaustenitic stainless steels using the Arrheniustype constitutive model. They reported that processing variables including strain, strain rate, and deformation temperature had a significant effect on the occurrence of DRX which was accelerated by increasing temperature and decreasing strain rate. Samantaray et al. [\[19](#page-17-11)] analyzed the high-temperature flow behavior of various grades of austenitic stainless steels, such as 304L, 304, 304 (as-cast), 316L, and alloy D9, using the modifed Zerilli− Armstrong (MZA) model and found that the developed model predicted well the elevated temperature fow behavior over the entire ranges of strain rate, temperature, and strain. However, while a large number of eforts have been invested into the hot deformation behavior of austenitic stainless steels, little data is available on the martensitic grades.

In order to develop microstructure-based predictive tools for optimum thermomechanical processing, fnite element simulation is used to predict strain, strain rate, and temperature all over the volume of the material as a function of processing conditions [[20](#page-17-12)]. The implementation of a constitutive model would therefore allow correlating microstructure evolution to the local variations in strain, strain rate, and temperature. The constitutive models, which can predict the material fow behavior, are integrated into the FEM software. However, a very limited number of microstructure-based models have been integrated with the FEM software, and no report is available on the implementation of such models to study the hot workability of martensitic stainless steels.

Another predictive tool to determine the hot workability of a material based on changing the working parameters including temperature, strain, and strain rate is the processing maps proposed by Prasad et al. [[21\]](#page-17-13). The maps are developed based on the dynamic material model (DMM), which provides processing conditions for a defect-free fnal product through optimizing the hot working parameters. The DMM technique provides the guidelines for avoiding the fow instability domains where inhomogeneous deformation and localized fow could take place. It also allows determining and locating the domains where a fne and homogenous microstructure develops during hot deformation. The diferent domains are obtained by the superimposition of a power dissipation map and an instability map for diferent temperatures, strains, and strain rates [[22\]](#page-17-14). A large number of data have been reported on the application of the DMM method to various alloys, including stainless steels [[23–](#page-17-15)[25\]](#page-17-16), but very few are on martensitic stainless steels [\[26](#page-17-17), [27\]](#page-17-18). Furthermore, very little or no data is available on microstructure-based FEM modeling of hot deformation of stainless steels and its combination with processing map in general and martensitic ones, in particular [[28\]](#page-17-19).

In the present study, an accurate constitutive model for a modifed AISI410 MSS which describes the fow stress in terms of hot working variables including strain, strain rate, and deformation temperature is developed. The derivation of actual stress from measured one caused by deformation heating and friction effect was corrected. A comparison is made between the experimental flow stress data and the one calculated by the established constitutive equations. The constitutive model that best predicts the fow curves was then implemented into the FEM code Forge NxT 3.0® software through the development of an original user subroutine (UMAT). The simulation results were frst compared to the experimental ones for validation purposes and then were further utilized to analyze the efect of hot working parameters on microstructure evolution during the hot deformation process under diferent conditions. The DMM method was used to generate 3D processing maps which were correlated with the FEM and experimental results to discuss the hot deformation mechanisms and to determine the optimum hot working conditions.

# **2 Material and experimental procedures**

The material used for the current investigation was supplied by Finkl Steel-Sorel Forge, Quebec, Canada. The production cycle starts with melting using a 45-ton electric arc furnace followed by ladle metallurgy degassing and refning processes along with tight control of the chemical composition. After the ingot-casting step, the solidifed ingot is taken to the forge furnace and heated up to the forging temperature (1200–1260 °C). The hot-forged ingot undergoes heattreatment steps including the quench and tempering cycle. The samples for the isothermal compression test were cut from the ingot after hot forging. Figure [1](#page-2-0) shows the position where the compression samples were cut from the large-size block and Table [1](#page-2-1) displays the nominal chemical composition of the X12Cr13 used in this investigation.

The hot-compression tests were performed on cylindrical specimens with a diameter of 10 mm and a height of 15 mm based on the ASTM E209 standards with Gleeble 3800® thermomechanical simulator (Fig. [2a](#page-2-2)) at four diferent temperatures, 1050 ºC, 1100 ºC, 1150 ºC, and 1200 ºC and four strain rates,  $0.001 s^{-1}$ ,  $0.01 s^{-1}$ ,  $0.1 s^{-1}$ , and  $1 s^{-1}$ . The selected thermomechanical processing parameters are representative of the actual industrial forging process. The specimens were heated up to the test temperature at a heating



**Fig. 1** An illustration depicting the position of the Gleeble test samples in the industrial-sized ingot

<span id="page-2-1"></span><span id="page-2-0"></span>

<span id="page-2-2"></span>**Fig. 2 a** Hot compression setup in the Gleeble machine; **b** three thermocouples placed at diferent locations of the sample for precise measurement of holding time; **c** sample before and after hot compression test

composition of (%wt.)



rate of 2 ºC/s, held for 15 min, and then subjected to compressive deformation at the selected strain rates. To ensure maximum uniformity and stability of temperature distribution over the entire sample and obtain a similar grain size, as the initial microstructure, three sets of thermocouples, one at the center and another two at the edges, as shown in Fig. [2b,](#page-2-2) were used to determine the optimum holding time of the target temperature. Figure [2c](#page-2-2) shows a sample before and after the hot compression test.

The temperature readings of all three thermocouples (TC) were recorded (Fig. [3a](#page-3-0)), and once the target temperature of 1230 ºC was reached, time was calculated until all three thermocouples gave the same reading. This confrmed the best time for uniform heat distribution in the specimen. On this basis, a 15-min holding time was determined for the temperature homogenization and used in all experiments. Figure [3b](#page-3-0) displays the schematic of the hot compression experiments, where it can be seen that after the 15-min holding time, the deformation is applied. Tantalum sheets of 0.1 mm thickness were used as a lubricant between the sample and the deformation anvils which are made of pure tungsten.

After cooling to room temperature, the deformed samples were cut parallel to the compression axis by a precision cutter machine for microscopic examinations. The specimens



<span id="page-3-0"></span>**Fig. 3 a** Infuence of holding time at high temperature on temperature homogenization; **b** schematic illustration of a hot compression test

were mechanically polished down to 0.5 μm, and to reveal the microstructure, they were etched with Villela solution composed of 1 g  $(O_2N)_3C_6H_2OH$ , 5 ml HCL, and 100 ml  $C<sub>2</sub>H<sub>5</sub>OH$  for approximately 25 s. For microstructural characterizations, an Olympus LEXT OLS4100 laser confocal microscope was used.

# **3 Results and discussion**

### **3.1 True stress–strain curves**

The true stress–strain curves acquired from hot compression tests at diferent deformation temperatures and strain rates are shown in Fig. [4.](#page-4-0) It is seen that stress–strain curves are signifcantly afected by changing the temperature and strain rate. The fow curves decrease markedly with increasing temperature while for a fxed temperature, the fow stress increases with increasing the strain rate.

From the stress–strain curves of Fig. [4,](#page-4-0) three diferent stress changes can be observed with increasing stress. At the beginning of deformation (I), the stress increases signifcantly due to work hardening (WH). In the second stage (II), flow stress shows a continuous reduction with increasing stress until a peak point or an infection of work-hardening rate. This shows that thermal softening, due to DRV and DRX, becomes more and more predominant until it exceeds WH. At the third stage (III), the stress curve shows three diferent patterns with the increasing strain: (i) gradual decrease to a steady state with DRV/DRX softening. This is the case for all deformation temperatures and strain rates between 0.001 and 0.1  $s^{-1}$  except those at 1050 °C and 1100 ºC; (ii) higher stress levels without signifcant softening and work-hardening at 1050 ºC and 1100 ºC and strain rate of

 $0.1$  s<sup>-1</sup>; and (iii) continuous increase with significant work hardening (all deformation temperatures and strain rate of  $1 s<sup>-1</sup>$  [[24](#page-17-20)]. Therefore, it can be concluded that the softening due to DRX, characterized by a flow curve with a single peak followed by a steady-state fow, takes place at high temperatures and low strain rates. In contrast, at higher strain rates and lower temperatures, the higher work hardening rate slows down the rate of softening due to DRX, and therefore, both the peak stress and the onset of steady-state flow are shifted to higher strain levels. In fact, the drop observed in stress is because of dynamic recrystallization occurrence at all temperatures and strain rates of  $0.001-0.1$  s<sup>-1</sup> in Fig. [4](#page-4-0) [\[25\]](#page-17-16). Figure [5](#page-4-1) shows the microstructure evolution at the end of deformation characterized by a signifcant grain refnement due to DRX (Fig. [5b](#page-4-1)) in comparison with the initial microstructure (Fig. [5a\)](#page-4-1). However, at the strain rate of  $1 \text{ s}^{-1}$ , the fow curves are characterized by a continuous increase (without any peak), which is generally considered an indication of material undergoing DRV (Fig. [5c\)](#page-4-1).

# **3.2 Corrections for deformation heating and friction efects**

### **3.2.1 Deformation heating correction**

During the hot compression test, part of the accumulated deformation energy can be transformed into heat resulting in a higher temperature than the nominal one [\[29](#page-17-21)]. This difference, directly, related to the applied strain rate [[30\]](#page-17-22) can cause some errors in measuring stress values by the equipment and needs to be corrected. At a low strain rate, the duration of deformation can provide enough time for the generated heat to dissipate into the surrounding environment, and therefore, the fow behavior is considered to be





<span id="page-4-0"></span>**Fig. 4** True stress-true strain curves of the investigated steel at diferent deformation conditions



<span id="page-4-1"></span>**Fig. 5** Microstructure of the **a** as-received steel; **b** deformed sample at 1150 °C and 0.1 s<sup>-1</sup>; **c** deformed sample at 1050 °C and 0.1 s<sup>-1</sup>

under isothermal conditions. While at higher strain rates, the deformation heating cannot be fully removed from the sample, and consequently, the sample temperature rises. In this case, the process will be adiabatic. The following expression proposed by Goetz, R. Le is often used to correct the adi-abatic heating effect [[31](#page-17-23)]:

<span id="page-4-2"></span>
$$
\Delta T = \frac{0.95\eta}{\rho C_p} \int_{0}^{\epsilon} \sigma d\varepsilon \tag{1}
$$

where  $\rho$  is the material density,  $C_p$  is the specific heat,  $\sigma$  is the flow stress,  $\varepsilon$  is the strain, and  $\eta$  is the thermal efficiency being calculated as follows:

$$
\eta = \begin{cases}\n0 & \hat{\epsilon} \le 10^{-3} s^{-1} \\
0.95 + (0.316) \log \hat{\epsilon} & 10^{-3} s^{-1} \le \hat{\epsilon} \le 1.0 s^{-1} \\
0.95 & \hat{\epsilon} \ge 1.0 s^{-1}\n\end{cases}
$$
\n(2)

The values of  $\rho$  and  $C_p$  in the case of the investigated steel were calculated using JmatPro simulation software ([https://](https://www.sentesoftware.co.uk/jmatpro) [www.sentesoftware.co.uk/jmatpro](https://www.sentesoftware.co.uk/jmatpro)) for diferent temperatures and are listed in Table [2.](#page-5-0)

The temperature increments were calculated using Eqs. [\(1\)](#page-4-2) and ([2\)](#page-5-1) for strain values in the range of 0–0.8. Table [3](#page-5-2) shows the variation of  $\Delta T$  with  $\dot{\epsilon}$  for samples deformed at diferent strain rates and temperatures.

The results show a maximum temperature correction of about 16 ºC at the temperature of 1050 ºC and a strain rate of 1 s<sup> $-1$ </sup> that needs to be taken into consideration. The isothermal flow stresses (corrected for deformation heating),  $\sigma_c$ , were then determined using the following relationship [[32](#page-17-24)]:

$$
\sigma_c(\varepsilon, \dot{\varepsilon}, T_0) = \sigma_{wc}(\varepsilon, \dot{\varepsilon}, T_0) - \frac{\partial \sigma^H(T_0)}{\partial T_0} \bigg| \varepsilon \dot{\varepsilon} \Delta T \tag{3}
$$

where  $\sigma_{wc}$  is the flow stress uncorrected for deformation heating and  $T_0$  is the initial temperature. Figure [6](#page-6-0) shows the correction of the adiabatic heating effect on true-strain curves of X12Cr13. The diference between corrected and uncorrected curves is increasing with increasing strain rate and decreasing temperature.

### **3.2.2 Friction correction**

The presence of friction at the anvil-specimen interface leads to the formation of the dead zone and inhomogeneous deformation and therefore introduces errors in measuring

<span id="page-5-0"></span>**Table 2** Density and specifc heat for diferent temperatures

Temperature $(^{\circ}C)$	1050	1100	1150	1200	
Specific heat $(J/g K)$	0.64	0.65	0.66	0.67	
Density $(g/cm^3)$	7.5	7.48	7.45	7.43	

the flow stress during hot compression tests. Although tantalum sheets were used to lessen the friction, the impact of friction between anvil and specimen increases with increasing strain [\[33\]](#page-17-25).

<span id="page-5-1"></span>Figure [7a](#page-6-1) shows a schematic representation of the solid compression test where  $H_0$  and  $R_0$  are the initial height and radius of the cylinder, respectively. *H* is the height of the cylinder after deformation, and  $R_M$  and  $R_T$  are the maximum and top radius of the cylinder after deformation, respectively. To correct the flow stress, the friction coefficient  $\mu$  was initially calculated using the equation "Friction coefficient" reported in Fig. [7b](#page-6-1) [[28](#page-17-19)]. In this figure, b is the barreling factor,  $\Delta R$  is the difference between the maximum and top radius ( $\Delta R = R_M - R_T$ ), and  $\Delta H$  is the difference between the initial and final height. After calculation of constants, the flow stress can be corrected by the equation "corrected stress" shown in Fig. [7b](#page-6-1), where  $\sigma$  is the measured flow stress,  $\sigma_f$  is the corrected flow stress, and *ε* is the measured deformation degree. The corrected flow stresses for both adiabatic heating and friction effects are shown in Fig. [8](#page-7-0). The experimental stress curves are higher than the corrected ones which suggest that the effect of friction was greater than adiabatic heating on flow stress. The corrected flow stress curves were then used in the establishment of the constitutive equation.

# **3.3 Constitutive modeling**

#### **3.3.1 Hansel‑Spittel model**

The Hansel-Spittel constitutive model [\[34\]](#page-17-26) is currently implemented in Forge NxT 3.0® simulation software. This model relates flow stress to strain, strain rate, and temperature through the following equation:

$$
\sigma = Ae^{m_1T} \varepsilon^{m_2} \varepsilon^{m_3} e^{\frac{m_4}{\varepsilon}} (1+\varepsilon)^{m_5T} e^{m_7\varepsilon} \varepsilon^{m_8T} T^{m_9}
$$
(4)

Material constants, A and  $m_1$  to  $m_9$ , can be derived from the stress–strain data obtained from hot compression tests [[35\]](#page-18-0). In an initial stage, the above model was applied to the investigated steel, and the diferent material constants were calculated and are provided in Table [4:](#page-7-1)

<span id="page-5-2"></span>

te



<span id="page-6-0"></span>Fig. 6 Flow stress curves corrected for the effect of adiabatic heating

<span id="page-6-1"></span>

 $\sqrt{a}$  $\overline{(b)}$  $b=4\frac{\Delta R}{R}\cdot\frac{H}{\Delta H}$  $\overline{R}$ Barreling factor:  $R = R_0 \sqrt{\left(\frac{H_0}{H}\right)}$  $\overline{H}$ Before hot compression Radius (mm):  $\mu = \frac{\left(\frac{R}{H}\right)b}{\left(\frac{4}{\sqrt{2}}\right)-\left(\frac{2b}{3}\sqrt{3}\right)}$ R. Friction coefficient:  $R_{\rm r}$ After hot compression  $\overline{F}$  $\sigma_f = \frac{\sigma}{1 + (\frac{2}{3\sqrt{3}})\mu(\frac{R_0}{H_0})\exp(\frac{3\varepsilon}{2})}$ Corrected stress (MPa):

Figure [9](#page-8-0) shows the comparison between the experiment and calculated flow stress based on the Hansel-Spittel model. It can be seen that the difference between experimental and predicted stress is significant, especially at higher strain rates. The results also show that this model is not able to predict the softening behavior of the studied alloy for any of the investigated deformation conditions. Therefore, another constitutive model is needed to more accurately predict the flow stress behavior of the investigated steel.

#### **3.3.2 Arrhenius equation**

The Arrhenius-type equation initially proposed by Sellars and McTegart relates the strain rate, temperature, and the activation energy at a constant strain value, through the following equation  $[36]$  $[36]$  $[36]$ :

<span id="page-6-2"></span>
$$
Z = \dot{\varepsilon} \exp\left(\frac{Q}{RT}\right) = A[\sinh(\alpha\sigma)]^n \tag{5}
$$



<span id="page-7-0"></span>**Fig. 8** Corrected fow stress curves due to the friction and adiabatic heating efects for tested temperatures and strain rates

where *Z* is the Zener-Hollomon parameter, *Q* is the activation energy required to overcome deformation barriers,  $A, \alpha$  $(MPa<sup>-1</sup>)$ , and *n* are material constants, *R* the universal gas constant,  $T$  the temperature in Kelvin,  $\dot{\varepsilon}$  is the strain rate, and  $\sigma$  is the applied stress.

According to Eq. ([5\)](#page-6-2), the fow stress of the material, at a given strain, can be expressed as follows:

$$
\sigma = \frac{1}{\alpha} \left[ \sinh^{-1} \left( \frac{Z}{A} \right) \right]^{\frac{1}{n}} \tag{6}
$$

After some algebraic operations, the following expression is used to calculate the predicted stress:

$$
\sigma = \frac{1}{\alpha} ln \left\{ \left( \frac{Z}{A} \right)^{\frac{1}{n}} + \sqrt{\left( \frac{Z}{A} \right)^{\frac{2}{n}} + 1} \right\}
$$
(7)

The Arrhenius equation has the advantage to be physicsbased by including the activation energy term and therefore is more sensitive to changes in the microstructure. The strain rate during high-temperature deformation is given by [\[37](#page-18-2), [38](#page-18-3)]:

$$
\dot{\varepsilon} = AF(\sigma)exp(-\frac{Q}{RT})
$$
\n(8)

where  $F(\sigma)$  is in the form of power function:

$$
F(\sigma) = \begin{cases} \sigma^{n1} & \alpha \sigma < 0.8\\ exp(\beta \sigma) & \alpha \sigma > 1.2\\ [sinh(\alpha \sigma)]^{n} & \text{(for all } \sigma) \end{cases}
$$
(9)

The constants  $n_1$  and  $\beta$  are the slopes of the curves *lni* vs. *lno* and *lni* vs.  $\sigma$ , respectively, as indicated in Eq. ([10\)](#page-7-2) and illustrated in Fig. [10.](#page-9-0)

<span id="page-7-3"></span>
$$
n_1 = \frac{\partial \ln \dot{\varepsilon}}{\partial \ln \sigma}, \beta = \frac{\partial \ln \dot{\varepsilon}}{\partial \sigma}
$$
 (10)

<span id="page-7-2"></span>Thus:

<span id="page-7-1"></span>



<span id="page-8-0"></span>**Fig. 9** Comparison between the experimental and calculated fow stress developed by Hansel-Spittel model at four strain rates and temperatures of **a** 1200 ºC, **b** 1150 ºC, **c** 1100 ºC, **d** 1050 ºC

$$
\alpha = \frac{\beta}{n_1} \tag{11}
$$

Similarly, the constant  $n$  is the slope of the curve  $ln\epsilon$  vs.  $ln[sinh(\alpha\sigma)]$  as indicated in Eq. [\(12](#page-8-1)) and reported in Fig. [10d](#page-9-0) [\[39](#page-18-4)]:

$$
n = \frac{\partial \ln \dot{\varepsilon}}{\partial \ln[\sinh(\alpha \sigma)]} \tag{12}
$$

For determining constant *A*, the activation energy, *Q*, should be calculated frst. For a specifc strain rate, the slope of  $ln[sinh(\alpha\sigma)]$  vs.  $T^{-1}$  gives the values of  $\frac{Q}{nR}$  by linear regression of each curve. Thus,  $Q$  is obtained by substituting  $R = 8.314$  J.mol<sup>-1</sup> K<sup>-1</sup> and calculating *n*:

$$
Q = Rn \frac{\partial \ln[\sinh(\alpha \sigma)]}{\partial(\frac{1}{T})}
$$
\n(13)

Finally, "*A*" is the intercept of the curve *lnZ* vs.  $ln[sinh(\alpha\sigma)]$ . The constant values are listed in Table [5,](#page-9-1) and the plots used to obtain all constants are shown in Fig. [10.](#page-9-0)

As indicated above, the Arrhenius equation is based on constant strain condition. Moreover, as shown in Fig. [8,](#page-7-0)

the impact of strain on the fow stress is signifcant; thus, it is necessary to take the strain term into account for a more accurate constitutive equation. The evolution of the above material constants with strain could be described through polynomial functions of strain, as also reported in the literature  $[40-42]$  $[40-42]$  $[40-42]$ . As illustrated in Fig. [11,](#page-10-0) in the present study, it is possible to obtain an accurate ftting of the diferent constants with the applied strain using a sixth-order polynomial.

<span id="page-8-1"></span>The sixth-order polynomial ft results are obtained as given in Eq.  $(14)$  $(14)$  $(14)$ :

<span id="page-8-2"></span>
$$
n = 4.776 + 11.181\varepsilon - 140.46\varepsilon^2 + 517.92\varepsilon^3 - 901.73\varepsilon^4 + 766.97\varepsilon^5 - 256.54\varepsilon^6
$$
  
\n
$$
\alpha = 0.0307 - 0.2058\varepsilon + 1.2646\varepsilon^2 - 3.8842\varepsilon^3 + 6.3855\varepsilon^4 - 5.3762\varepsilon^5 + 1.8174\varepsilon^6
$$
  
\n
$$
Q = 485.31 + 29.685\varepsilon - 1353.9\varepsilon^2 - 4448.6\varepsilon^3 + 29723\varepsilon^4 - 45492\varepsilon^5 + 22050\varepsilon^6
$$
  
\n
$$
lnA = 37.311 + 11.868\varepsilon - 213.31\varepsilon^2 + 30.281\varepsilon^3 + 1724.8\varepsilon^4 - 3112\varepsilon^5 + 1598.6\varepsilon^6
$$
 (14)

After developing the models of *n*, *α*, *Q*, and *lnA* by considering the efect of strain, the fow stress can be predicted using Eq. ([7](#page-7-3)) which relates the stress to the Zener-Hollomon parameter, strain rate, deformation temperature, and the material constants as a function of strain [[43](#page-18-7)].



<span id="page-9-0"></span>**Fig. 10** Plots for determination of the constants **a**  $n_1$ , **b**  $\beta$ , **c**  $n$ , **d**  $Q$  at a deformation temperature of 1200 °C, 1150 °C, 1100 °C, and 1050 °C. The constants represent the slope of the respective curves determined using linear regression

# **3.4 Verifcation of the developed constitutive equations**

To evaluate the accuracy of the developed constitutive models in predicting the hot deformation behavior of the martensitic stainless steel, the predicted value should be compared with the experimental one. As shown in Fig. [12,](#page-11-0) a good agreement can be observed between the experimental data and the predicted values for all of the experimental conditions used in this work.

The accuracy and the reliability of the Arrhenius model were compared in terms of correlation coefficient (*R*) and the average absolute relative error (ARRE  $\Delta$ ):

<span id="page-9-1"></span>**Table 5** Material constants in the Arrhenius model

$\beta$ (MPa <sup>-1</sup> ) $n_1$		$\alpha$ (MPa <sup>-1</sup> ) $n$	$Q$ (KJ mol <sup>-1</sup> ) LnA	
0.097021	5.3010 0.0181		3.95 412.948	31.463

$$
R = \frac{\sum_{i=1}^{N} (\sigma_E^i - \overline{\sigma}_E)(\sigma_P^i - \overline{\sigma}_P)}{\sqrt{\sum_{i=1}^{N} (\sigma_E^i - \overline{\sigma}_E)^2} \sum_{i=1}^{N} (\sigma_P^i - \overline{\sigma}_P)^2}
$$
(15)

The absolute average error  $(\Delta)$  is expressed as:

$$
\Delta = \frac{1}{N} \sum_{i=1}^{N} \left| \frac{\sigma_E^i - \sigma_P^i}{\sigma_E^i} \right| \times 100 \tag{16}
$$

where  $\sigma_F$  is the experimental data, and  $\sigma_P$  is the calculated value based on the proposed constitutive equations.  $\overline{\sigma}_P$  and  $\overline{\sigma}_E$  are the mean values of  $\sigma_P$  and  $\sigma_E$ , respectively, and *N* is the number of data points. The calculated  *coefficient is* shown in Fig. [13b](#page-11-1), where a good correlation can be observed between the measured and calculated data.

It must be noted that the value of the *R* coefficient indicates the intensity of a linear relationship between the predicted and the experimental values. However, a higher *R*-value is not always indicative of the higher accuracy of the model because the model is often biased towards higher or lower values. Therefore, an unbiased statistical parameter,



<span id="page-10-0"></span>**Fig. 11** The polynomial ft of order 6 of variation of; **a** *Q*, **b** *lnA*, **c** *α*, **d** *n*. The blue square denotes experimental data and the black line denotes polynomial models

Δ, is used to evaluate the accuracy of the Arrhenius model in prediction [[4\]](#page-17-1). The lower is the absolute average error, the higher is the predictability of the model. The values of  $\Delta$  at diferent temperatures and strain rates are listed in Fig. [13b.](#page-11-1) It can be seen that the maximum value is 5.04% at the strain rate of 0.01 s<sup> $-1$ </sup> and the deformation temperature of 1200 °C. The minimum value is  $0.81\%$  at the strain rate of  $0.1 \text{ s}^{-1}$ and the deformation temperature of 1050 ºC, and the mean value of  $\Delta$  for all the deformation conditions is 3.29% which is a very small error. Thus, it could be concluded that the established constitutive equations can well describe the relationship between fow stress, strain rate, temperature, and strain. However, the developed constitutive equation should be implemented in Forge NxT 3.0® software in order to predict temperature, strain, and strain rate maps in diferent locations of the specimen and then validate the predictions through microstructure examination.

## **3.5 Microstructure‑based FEM simulation**

A specifc user subroutine was developed to implement the constitutive equation in the FE code. For the analysis, threedimensional fnite element method was performed to simulate the hot compression process. The boundary conditions for the FE simulations were similar to the experimental tests (i.e., temperature, strain rate, strain, and thermal exchange between piece and anvils). The friction factor was selected as 0.35, and the convergence tests were performed by changing the element size to ensure the accuracy. The numerical simulations were performed on a FEM model discretized with four-node tetrahedral elements. Figure [14a](#page-12-0) and [b](#page-12-0) show the meshed fnite element models before and after deformation respectively, and Fig. [14c](#page-12-0) shows a comparison between the predicted and experimental data of force versus time where good and acceptable predictability of the model is demonstrated.

According to simulation results, the strain distribution can be illustrated as a contour plot aided by color series at the temperature of 1050–1200 ºC, shown in Fig. [15a–d.](#page-13-0) In this fgure, the non-uniformity of deformation distribution is seen. The specimen is roughly divided into several deformation zones (non-uniformity) according to the severity of deformation, which leads to diferent fows of material during hot deformation and diferent microstructure evolution. Figure [16](#page-14-0) shows the microstructure evolution during an upset operation at 1200 ºC and a strain rate of 0.1 s<sup>-1</sup>. The specimen was cut parallel to



<span id="page-11-0"></span>**Fig. 12** Comparison between the experimental and calculated fow stress developed by Arrhenius model at 4 strain rates and temperatures of **a** 1200 ºC, **b** 1150 ºC, **c** 1100 ºC, **d** 1050 ºC

<span id="page-11-1"></span>**Fig. 13 a** Correlation between the experimental and calculated flow stresses by the developed constitutive equation; **b** value of Δ at diferent temperatures and strain rates





the compression axis by a precision cutter machine for carrying out a metallographic examination. The circled numbers in Fig. [16a,](#page-14-0) represent strain levels that the material experienced during hot deformation, based on the map of strain distribution, and the yellow dotted lines show the dead zones. In Fig. [16b–e,](#page-14-0) the microstructure of the corresponding four zones is shown.

As shown in Fig. [16b](#page-14-0), for material regions that experienced a strain of 0.2, the microstructure remained unchanged due to a low degree of deformation. By increasing strain, Fig. [16c,](#page-14-0)



<span id="page-12-0"></span>**Fig. 14** FE model **a** before deformation; **b** FE model after deformation; **c** force versus time plot of experimental and predicted for all deformation conditions

the nucleation of new grains around the initial grain boundaries can be seen indicating that the workability of the material increased through recrystallization. The microstructure at strain of 0.4 is heterogeneous with a mix of deformed and undeformed grains which makes the material susceptible to the occurrence of defects and wedge cracking, thereby deteriorating the mechanical properties. However, the material is fully recrystallized at the strain of 0.6 and 0.8, as shown in Fig. [16d](#page-14-0) and [e,](#page-14-0) respectively.

# **3.6 Processing map**

The part with fewer defects in the final production process signifies the ideal processing conditions and higher



<span id="page-13-0"></span>**Fig. 15** FE results of effective strain distribution in the sample deformed to a strain of 0.8 and a strain rate of 0.1 s<sup>-1</sup> **a** 1200 °C, **b** 1150 °C, **c** 1100 ºC, and **d** 1050 ºC

workability of metal, which means the higher plastic deformation ability that a metal can be deformed easily without fracture during the forming process. In 1997, Prasad and colleagues proposed the deformation mechanism model (DMM) which produces the flow instability domains [[21\]](#page-17-13). This model is an analytical method used to determine optimum deformation temperature and strain rate ranges in hot forming operations. Based on the DMM, the workpiece undergoing hot deformation goes through a process of power dissipation (P), defined by Eq. [\(17](#page-13-1)). The total input power is dissipated through plastic work that is converted into heat (*G*) and changes in the microstructure (*J*), such as flow localization, DRV, DRX, phase transition, and shear band formation [[44](#page-18-8)].

$$
P = G + J = \int_{0}^{\dot{\epsilon}} \sigma d\dot{\epsilon} + \int_{0}^{\sigma} \dot{\epsilon} d\sigma = \sigma \dot{\epsilon}
$$
 (17)

The flow stress, at constant strain and deformation temperature, is given by [[45](#page-18-9)]:

$$
\sigma = K \dot{\varepsilon}^m \tag{18}
$$

where K is a material constant, and *m* is the strain rate sensitivity and can be represented as follows:

$$
m = \frac{\partial(\ln \sigma)}{\partial(\ln \dot{\epsilon})}
$$
(19)

By comparison of *J* co-content with the maximum possible dissipation, the dimensionless parameter,  $\eta$ , efficiency of power dissipation is defned as

$$
\eta = \frac{J}{J_{\text{max}}} = \frac{2m}{m+1} \tag{20}
$$

η represents the ability of the workpiece to change its microstructure upon deformation. By plotting contours of η as a function of temperature and strain rate, a power dissipation map is obtained.

<span id="page-13-1"></span>Softening mechanisms such as DRX and DRV are associated with η. However, in the case of void formation or wedge cracking, the power dissipation map alone is not sufficient to fully characterize the hot working behavior of an alloy. In order to achieve this and, in particular, to optimize the processing parameters, it is important to distinguish the undesirable regions characterized by the unstable fow. The instability criteria are defned in order to identify such regions and

<span id="page-14-0"></span>**Fig. 16** Optical microscope views of **a** deformed sample at 1200 ºC and a strain rate of 0.1 s −1 to strains of **b** 0.2, **c** 0.4, **d** 0.6, and **e** 0.8



build an instability map that, once superimposed on the iso efficiency contours, provides the final processing map.

Instability maps are developed based on Ziegler's plasticity theory to the dissipation functions of the dynamic materials model. A dimensionless parameter for microstructural instability based on Kumar-Prasad and Murty-Rao criteria, [[46](#page-18-10)], respectively, is given by

$$
\xi(\dot{\varepsilon}) = \frac{\partial \ln(\frac{\mathrm{m}}{\mathrm{m}+1})}{\partial \ln \dot{\varepsilon}} + \mathrm{m} < 0 \tag{21}
$$

As reported by Ziegler et al. [[47\]](#page-18-11), mixed microstructure, flow localization, or adiabatic shear bands are susceptible to occur in regions with negative *ξ* values.

<span id="page-15-0"></span>



### **3.6.1 The 2D processing maps at various strains**

2D processing maps can be obtained through the superimposition of the power dissipation and fow instability 2D contour maps at a given strain. Figure [17](#page-15-0) shows the 2D processing maps of the X12Cr13 investigated steel at various strains with an interval of 0.2. It can be seen from the fgure that, by increasing the strain, the maximum power efficiency increases while the fow instability region frst increases up to 0.4 strain and then decreases. Moreover, the fow instability occurred for the highest strain rate ( $\epsilon = 0.182 \text{ s}^{-1} - 1 \text{ s}^{-1}$ ). The occurrence of such flow instability could be related to the



<span id="page-15-1"></span>Fig. 18 The 3D power efficiency map as a function of **a** strain and **b** temperature

limited time for dislocation movement due to the high strain rate and the subsequent inhibition of softening processes such as DRV and DRX. Under these conditions, strain accumulation could be restricted to specifc regions and the formation of wedge crack, voids, and shear bands is promoted, as also reported by Momeni et al. [[22\]](#page-17-14). Therefore, hot working parameters should be selected carefully to avoid regions with a negative value for the fow instability parameter which corresponds to the gray-shaded regions in Fig. [17.](#page-15-0)

#### **3.6.2 Continuous 3D processing maps**

In the power dissipation maps of Fig. [17,](#page-15-0) the color of the grid is related to the value of the power dissipation efficiency parameter η. As shown in Fig. [18](#page-15-1), η value increases by increasing strain and temperature. This means that higher η leads to higher workability and increased potential for microstructural changes. Based on the above fndings, because signifcant microstructural evolution via DRX occurs in the region with the highest η-value, the deformation should be carried out in this region.

The processing map has been used to identify unstable and risky situations through the determination of two factors: power dissipation and fow instability domains. The results obtained in the present investigation show a good agreement between the DMM results and those obtained with FEM simulation and microstructure observations. These results are interesting as it is the frst time such comparison is made between the predictions of the two methods and supported by experimental evidence. Specifcally, the non-uniformity of strain distribution during the deformation, predicted by the FEM analysis, was confrmed by microstructure examination, as reported in Fig. [16](#page-14-0). The processing map analysis shows that regions that experienced a lower strain have negative instability values and low power efficiency which means that they have low workability. In Fig. [16c](#page-14-0), the microstructure at strain of 0.4 is heterogeneous and is expected to have inconsistent mechanical properties. This fnding agrees with the negative instability criteria reported in Fig. [17b.](#page-15-0) On the other hand, as reported in Fig. [16d](#page-14-0) and [e,](#page-14-0) the material is fully recrystallized at the strains of 0.6 and 0.8, respectively. The DMM analysis also showed that under these conditions, the instability criteria have a positive value (not gray-shaded regions) which means higher workability as shown in Fig. [17c](#page-15-0) and [d](#page-15-0).

# **4 Conclusion**

The hot workability of a martensitic stainless steel was studied using microstructure-based numerical simulation. The following conclusions can be drawn from the present investigation:

- 1. The fow stress curves of the compressed X12Cr13 steels in the range of 1050–1200 ºC and the strain rate range of 0.001–1  $s^{-1}$  display a significant sensitivity of flow stress to strain rate, temperature, and strain.
- 2. A constitutive equation based on the Arrhenius model was developed in the present work and its validity was confrmed showing more accurate prediction than the Hansel-Spittel model.
- 3. The developed constitutive model was integrated into the Forge NxT 3.0® software and showed good correlations with the predictions of fow instability domains obtained using the dynamic material model.

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**Author contribution** Simin Dourandish: conducting the experimental tests and numerical analysis, methodology, writing, reviewing, and editing the paper

Henri Champliaud: fnite element modeling, reviewing, and editing

Mohammad Jahazi: idea of the project, defnition of the project, review of the methodology and experimental part, supervision, reviewing, and editing

Jean-Benoit: providing material, technical support, industrial application, reviewing, and editing

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**Data availability** The datasets used in the current study are available from the corresponding author on reasonable request.

**Code availability** Not applicable.

### **Declarations**

**Ethics approval** Not applicable.

**Consent to participate** Not applicable.

**Consent for publication** Not applicable.

**Consent for publication** All the authors have read and agreed to the published version of the manuscript.

**Competing interests** The authors declare no competing interests.

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