**ORIGINAL ARTICLE**



# **A kinematic precision reliability evaluation method for rotor‑bearing systems considering multi‑source wear degradations and random errors**

**Hongwei Wang1 · Jiawei Xiang<sup>1</sup>  [·](http://orcid.org/0000-0003-4028-985X) Xufeng Zhao2 · Yulong Li3**

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## **Abstract**

Rotor-bearing systems play a vital role in machine tools, aero engines, and wind turbines. Generally, worn-induced degradation quantities and manufacturing errors of components are the main error sources that infuence the precision reliability of rotor-bearing systems. The current precision reliability evaluation models are focusing on several error sources in only a few key components without agreeable results. Therefore, a precision reliability evaluation model is proposed considering all time-variant error sources and random error sources. Firstly, time-variant wear models for commonly occurred degradation types in a rotor-bearing system are developed. Secondly, the constructed time-variant wear models are inserted into the precision model with all moving components in the rotor-bearing system using meta-action structural decomposition method. Finally, the time-variant stochastic process discretization method is employed to establish the precision reliability evaluation model, and solve the precision reliability of the rotor-bearing systems. Case investigations are carried out to verify the performance of the present model, which provides a more accurate precision reliability evaluation model to estimate the conditions of rotor-bearing systems during the service period.

**Keywords** Rotor-bearing systems · Precision reliability evaluation · Time-variant wear models · Random error sources · Meta-action

# **1 Introduction**

Rotor-bearing systems are widely used in the transmission process of mechanical movement and momentum. The performances of rotor-bearing systems need to meet diferent requirements in diferent application scenarios. Thereby, the performance design has always been the dominant issue in the stage of system development design. The randomness and time variability of kinematic precision can be expressed by a reliability index [\[1\]](#page-13-0). The precision reliability can be earmarked as an important constraint index in the research on tolerance allocation [[2\]](#page-13-1), and it is also an important basis for formulating maintenance strategy [[3](#page-13-2)]. Consequently, it is necessary to establish a kinematic precision reliability evaluation method with more accurate performance for rotor-bearing systems to serve the precision design.

An effective and correct precision expression model plays an important role in the research of kinematic precision reliability. In recent years, much attention has been paid to the aspect of the precision modeling theories for rotor-bearing systems. Tang et al. [\[4\]](#page-13-3) introduced the stream of variation theory into geometric error modeling for mechanical systems, regarded each moving component as a station, and calculated the accumulated geometric errors according to the transmission sequence from the upstream station to the downstream station. Yang and Ding [\[5](#page-13-4)] represented the error propagation process from one component to the other diferent component with a diferential motion matrix, and built the integrated precision model of the whole kinematics chain by homogeneous transformation matrices. Bozca [[6\]](#page-13-5) calculated the transmission precision of the gear system based on

 $\boxtimes$  Jiawei Xiang wxw8627@163.com

School of Mechanical and Electrical Engineering, Wenzhou University, Wenzhou 325035, People's Republic of China

<sup>&</sup>lt;sup>2</sup> College of Economics and Management, Nanjing University of Aeronautics and Astronautics, Nanjing 210016, People's Republic of China

<sup>&</sup>lt;sup>3</sup> School of Mechanical Engineering, Yangzhou University, Yangzhou 225127, People's Republic of China

a state-space model. Liu et al. [[7\]](#page-13-6) regarded the error accumulation between components as the screw motion of rigid bodies, and established the precision model of systems by screw theory and Jacobian function. Yang et al. [\[8](#page-13-7)] established a six-degree-of-freedom geometric motion errors model of the precision rotary stage by adopting the rigid body hypothesis and the spatial transformation matrix. Wang et al. [\[9](#page-13-8)] proposed a transmission error model of the ball screw by converting manufacturing errors and installation errors to the errors in normal and axial direction. However, the assembly precision caused by the geometric errors of components and the transmission precision caused by the relative motion relationship of kinematic pairs are not distinguished in the current precision modeling methods, which cannot refect the motion essence of mechanical systems. Moreover, there are various error sources for the complex system with many components which lead to the cumbersome calculation process and inaccurate results. The main cause of this problem is that the error sources are not reasonably classifed in the traditional structured decomposition method. From the perspective of motion modularization, meta-action theory [[10\]](#page-13-9) produces a reasonable balance in the subordinate relationship between components and systems. And some applications in reliability analysis have been carried out [[11](#page-13-10)]; the advantages in failure mode and effects analysis have also been proved [[12](#page-13-11), [13](#page-13-12)]. Mu et al. [\[14](#page-13-13)] introduced the application of meta-action theory in the remaining useful life prediction on account of its advantage in the refection of the interaction among the components. Chen et al. [[15\]](#page-13-14) applied meta-action theory for the reliability optimization design of mechanical systems. Yang et al. [[16\]](#page-13-15) presented a performance degradation prediction approach by combining meta-action theory with the digital twin technology. Xiao et al. [\[17](#page-13-16)] further explored the motion modularization strategy of the meta-action unit, which can refect the characteristics of the mechanical systems that the function determined by the motion. Therefore, meta-action theory also lay a good foundation for the precision modeling.

The random geometric errors of components caused by the uncertainty of the manufacturing process are considered as the initial error sources afecting the precision reliability of rotor-bearing systems, and some meaningful research work on the analysis of such errors has been proposed. Huang et al. [\[18\]](#page-13-17) considered the original manufacturing errors of gears as the random variables afecting the transmission error of the gear mechanism, and presented a kinematic accuracy reliability analysis method using the saddle-point approximation technique. Cai et al. [\[19](#page-13-18)] established a geometric/ kinematic error model for mechanical systems using homogenous transformation matrices, regarded the geometric error values as random quantities within the tolerance range, and developed an accuracy reliability analysis model based on the high-order moment standardization technique. Wang et al. [[20\]](#page-13-19) estimated the probability distribution of the precision state function for mechanical systems with random geometric errors based on the statistical fourth moment method, and employed the Edgeworth series to analyze the dynamic precision reliability. Zhang et al. [[21\]](#page-13-20) conducted the importance sampling method to analyze the accuracy reliability of mechanical systems after establishing a comprehensive error model affected by the random geometric errors. Yang et al. [\[22](#page-13-21)] analyzed the error propagation between parts based on small displacement torsor and multi-tolerance coupling, presented the assembly precision model of mechanical rotating systems in accordance with the Jacobian-torsor method, and calculated the assembly precision reliability according to the probability distribution of the assembly error index. Ma et al. [\[23\]](#page-13-22) performed the effects of the stochastic rotor/stator gap caused by manufacturing on vibrational behaviors and reliability of the rotor systems. In addition, there are some timevariant error sources such as the wears on gears and bearings during the operation of the rotor-bearing systems. Wang et al. [\[24\]](#page-13-23) calculated the wear depths of the parts by the Archard wear model, considered the part geometric parameters in the wear formula as input random variables, and constructed the Kriging models to predict the time-variant reliability of mechanical systems. Chen et al. [[25\]](#page-13-24) decomposed the timevarying dynamic transmission error of the gear mechanism into many random variables according to the meshing period, thus realizing the transformation of a time-dependent reliability solution into a time-independent reliability solution, and estimated the reliability based on the saddle-point approximation method. Jiang et al. [[26\]](#page-13-25) calculated the cumulative wear of gears in the gear transmission system and the meshing force and stress distribution under the infuence of gear wear, and predicted the system dynamic reliability according to the gear strength degradation rule. Furthermore, stochastic models [[27\]](#page-13-26) can also be used to describe the process of performance degradation, such as the gamma process model [\[28](#page-13-27)] and the Wiener process model [[29](#page-13-28)], and the precision reliability evaluation using the PHI2 method [[30\]](#page-13-29), support vector machine [[31\]](#page-13-30), response surface method [[32\]](#page-13-31), and the Kriging method [[33\]](#page-13-32) have been reported in literature. Unfortunately, to the best of our current knowledge, there are few reports in literature which have combined the manufacturing error sources with the time-variant wear sources to evaluate the efects on the kinematic precision reliability of rotor-bearing systems, and that leads to many important error items being ignored and the inaccuracy model.

Motivated by the above-mentioned limitations, a novel kinematic precision reliability modeling method is proposed to improve the accuracy of reliability assessment for rotorbearing systems in this paper. Specifcally, based on the metaaction structural decomposition method, the precision analysis model containing each source related to the output precision of rotor-bearing systems is established. When the distribution

of random errors and the expression of time-variant wears are determined, the kinematic precision reliability can be calculated by the time-variant stochastic process discretization method. The remainder of this manuscript is organized as follows: Sect. [2](#page-2-0) provides the time-varying wear models for the probably worn components in rotor-bearing systems. In

Sect. [3,](#page-4-0) the output precision model based on the meta-action structural decomposition method is established. Section [4](#page-7-0) presents the precision reliability evaluation model considering multi-source wear degradations and random errors. The applicative case is reported in Sect. [5](#page-8-0) to validate the proposed method. The conclusions are fnally drawn in the fnal section.

## <span id="page-2-0"></span>**2 Time‑variant wear models**

The precision time variation and failure of rotor-bearing systems are primarily caused by the wearing of components, especially the wear occurring on gears and bearings. Therefore, these two types of wear are analyzed specifcally and respectively as follows.

#### **2.1 Tooth thickness wear**

Tooth thickness wear is the gear material continuous reduction process under stress. The wear quantity of the tooth profle surface at a certain time can be calculated by the Archard model. Let *V* represent the wear volume of the tooth surface and *s* the relative sliding distance; then

$$
\frac{V}{s} = k \frac{W}{H} \tag{1}
$$

where  $k$  is the wear coefficient,  $W$  the normal pressure on the contact surface, and *H* the hardness of the material.

By discretizing the contact area on the tooth surface into several contact points, the infnitesimal wear quantity of the arbitrary contact point at a certain time can be obtained and calculated as

$$
\delta h = \int_0^s \frac{k}{H} ds \tag{2}
$$

where *P* represents the contact stress at the contact point.

The wear coefficient  $k$  is a parameter comprehensively afected by factors such as working condition, surface roughness, lubrication characteristics, and material characteristics. According to the Janakiraman regression formula [[34\]](#page-14-0), it can be calculated as

$$
k = \frac{3.981 \times 10^{29}}{E_e} L^{1.219} G^{-7.377} S^{1.589}
$$
  
And  $L = \frac{W'}{E_e R_e}, G = \alpha_0 E_e, S = \frac{\sqrt{R_{a_1}^2 + R_{a_2}^2}}{R_e}$ . (3)

Where equivalent elastic modulus  $E_e = 1$  $\int (1-v_{(1)}^2)$  $\frac{-v_{(1)}^2}{E_{(1)}} + \frac{1-v_{(2)}^2}{E_{(2)}}$ *E*(2)  $\lambda$ and  $v_{(1)}$ ,  $v_{(2)}$  and  $E_{(1)}$ ,  $E_{(2)}$  are Poisson ratio and elastic modulus of the driving and driven wheel's material, respectively. *W*′ is the load per unit length. Equivalent radius of curvature  $R_e = 1 / (\frac{1}{r_e})$  $\frac{1}{r_{b(1)}} + \frac{1}{r_{b(2)}}$  $\left( \right)$ ,  $r_{b(1)}$ , and  $r_{b(2)}$  are the base circle radius of the driving and driven wheel, respectively.  $\alpha_0$  is the pressureviscosity coefficient of lubricating oil (assuming it is a constant not affected by pressure during the wearing process).  $R_{\alpha 1}$  and  $R_{\alpha 2}$ are the roughness of these two contact surfaces, respectively.

According to Hertz contact theory, the contact stress *P* can be solved by taking the tooth surface meshing problem of the driving and driven gear equivalent to the mutual extrusion problem of two cylinders. When these two tooth surfaces are in line contact under stress, a rectangular contact area is formed due to elastic deformation. The contact stress distribution of a certain cross section in the contact area at a certain time is shown in Fig. [1](#page-2-1).where  $P_{\text{max}}$  represents the maximum Hertz contact stress. Half width of Hertz contact area  $a = \sqrt{\frac{4W'R_e}{\pi E}}$  $\frac{d\mathbf{r}}{d\mathbf{r}E_e}$  and the variation range of the relative sliding distance *s* is [0, *a*].

As demonstrated in Fig. [1](#page-2-1), the maximum contact stress occurs in the center of the contact area and has a decaying trend to both sides. Contact stress at the arbitrary contact point is calculated as

$$
P(x) = \frac{2P_{\text{max}}}{\pi a^2} \sqrt{(a^2 - x^2)}
$$
 (4)

where the maximum contact stress  $P_{\text{max}} = T_{(1)}/B_{(1)}r_{b(1)}$ .  $T_{(1)}$ is the load transmitted by the gear pair;  $B_{(1)}$  is the tooth width of the driving gear. Therefore, the calculation equation of the single infnitesimal wear quantity can be rewritten as

$$
\delta h = \int_0^a \frac{2kT_{(1)}}{\pi a^2 B_{(1)} r_{b(1)} H} \sqrt{(a^2 - x^2)} dx
$$
 (5)

When the driving wheel rotates once, the contact point of the tooth surface completes single wear accumulation, while the driven wheel completes the corresponding number of times of wear accumulations due to the relationship of the



<span id="page-2-2"></span><span id="page-2-1"></span>**Fig. 1** Hertz contact stress distribution

transmission ratio. Therefore, the wear quantity of the driving gear tooth thickness can be calculated as

$$
\Delta e_{g(1)}(t) = \delta h_{g(1)} \cdot n_{g(1)} \cdot t \tag{6}
$$

The wear quantity of the driven gear tooth thickness can be calculated as

$$
\Delta e_{g(2)}(t) = \delta h_{g(2)} \cdot n_{g(2)} \cdot t \tag{7}
$$

#### **2.2 Bearing wear**

Rolling bearings are generally composed of rolling elements, bearing outer ring, locking inner ring, and bearing cage. The roller and the raceway move relative to each other in a point-contact manner. A single roller not only rotates around its own axis, but also rolls around the bearing axis on the raceway, and under the force of the bearing cage, the roller also slides on the raceway to a certain extent. Therefore, the internal friction of the rolling system is a compound friction including rolling and sliding friction.

To facilitate the analysis and calculation, it is assumed that the rolling elements are uniformly worn, that is, the wear loss of each roller at a certain time is the same. The rollers move independently, and there is no radial clearance between the rollers and the inner and outer rings. Similar to the calculation of tooth thickness wear, the continuous wear process of bearing is frstly discretized into several infnitesimal wearing moments. Assuming that the wear loss of each point on the contact line at each moment is the same, according to the Archard wear model, the infnitesimal wear loss of the contact point at a certain moment can be calculated as

$$
\delta h = \int_0^t k P v_s dt \tag{8}
$$

where  $k$  represents the material wear coefficient, and the calculation equation is the same as Eq. ([3\)](#page-2-2). *P* represents the contact stress at the contact point.  $v<sub>s</sub>$  represents the relative sliding speed. *dt* is the infnitesimal contact time.

Normally, the rollers contact with the outer raceway due to the centrifugal force  $F_c$ . The force analysis of the roller is shown in Fig. [2](#page-3-0). Considering that the rotation speed *ω* is small, the infuence of centrifugal force can be ignored. where  $N_i$  represents the normal pressure exerted by the inner raceway on the rollers.  $N<sub>o</sub>$  represents the support force of the outer raceway to the roller.  $F_k^1$  and  $F_k^2$  represent the force of the bearing cage.  $f_i$  and  $f_o$  represent the friction between the inner and outer raceways and the rollers, respectively.

According to Hertz contact stress distribution in Fig. [1,](#page-2-1) the half-width *a* of the contact area is discretized into *m* units with a width of  $\Delta a = a/m$ . Taking the contact between the inner raceway and the roller as an example, the calculation equation of the half width of the contact area  $a^{(i,g)}$  is

<span id="page-3-1"></span>

<span id="page-3-0"></span>**Fig. 2** Force analysis of the cylindrical roller in the load-bearing area

<span id="page-3-2"></span>
$$
a^{(i,g)} = \sqrt{\frac{4W^{(i,g)}}{\pi} \left( \frac{1 - v_i^2}{E_i} + \frac{1 - v_g^2}{E_g} \right) \left( \frac{r_i r_g}{r_i + r_g} \right)}
$$
(9)

where  $W^{(i,g)}$  is the contact load.  $v_i$ ,  $v_g$  and  $E_i$ ,  $E_g$  are Poisson ratio and elastic modulus of the inner ring and roller material, respectively.  $r_i$  and  $r_g$  are the contact curvature radius of the inner raceway and the roller, respectively.

Considering that the efective meshing width is much smaller than the meshing length, when the contact area is discretized into units, the slight diference in the equivalent curvature radius between contact units is ignored. Thus, the normal contact stress of the *k*-th unit within the half width of each contact area can be obtained as

$$
P_k^{(i,g)} = P_{\max} \times \sqrt{1 - \left(\frac{\Delta a^{(i,g)} \times k - \Delta a^{(i,g)}/2}{a^{(i,g)}}\right)}
$$
(10)

Let  $R_i$  be the diameter of the inner raceway,  $R_o$  the diameter of the contact point of the outer raceway,  $R_{\varphi}$  the diameter of the rollers,  $R_{or}$  the diameter of the bearing rollers' center circle,  $\theta = R_g / R_{or}$ , and *n* the rotational speed of the inner ring, and let the angular velocity of the inner ring and the worm gear be kept the same. Then, the revolution angular velocity of rollers is calculated as

$$
\omega_g^R = \frac{1}{2}\omega(1-\vartheta) \tag{11}
$$

The rotation angular velocity of rollers is calculated as

$$
\omega_g^r = \frac{R_{or}}{2R_g} \omega (1 + \vartheta)(1 - \vartheta)
$$
\n(12)

Since the inner ring rotates and the outer ring remains stationary during the motion of the rolling bearing, there are two pairs of friction pairs in the bearing, so the wear value of the bearing includes three parts: the outer ring, the inner ring, and the rollers. The wear values of the inner and outer rings are determined by the single wear value

and the number of rolling contact at a contact point in one revolution cycle. The bearing including *z* rollers, when the inner ring of the bearing makes one rotation on its axis,

$$
N_i = \frac{\omega - \omega_g^R}{\omega} \times z = \frac{z}{2}(1 + \theta)
$$
\n(13)

The number of rollers in contact with the outer raceway is

$$
N_o = \frac{\omega_s^R}{\omega} \times z = \frac{z}{2}(1 - \theta)
$$
\n(14)

Therefore, the calculation equation of the wear quantity of the contact point on the inner raceway is obtained as

$$
\delta h_i = a^{(i,g)} \cdot z \cdot (1+\vartheta) \cdot n \cdot t \cdot \sum_{k}^{m} \left[ k_k^{(i,g)} \left( P_k^{(i,g)} \right) \cdot P_k^{(i,g)} \right] \tag{15}
$$

The calculation equation of the wear quantity of the contact point on the outer raceway is obtained as

$$
\delta h_o = a^{(o,g)} \cdot z \cdot (1 - \vartheta) \cdot n \cdot t \cdot \sum_{k}^{m} \left[ k_k^{(o,g)} \left( P_k^{(o,g)} \right) \cdot P_k^{(o,g)} \right]
$$
\n(16)

For the roller, its revolution speed is  $n_g^R = \frac{1}{2}n(1 - \theta)$ . Since the roller is in contact with both the inner and outer raceways during one rotation, the calculation equation of the wear quantity of the contact point on the roller can be obtained as



<span id="page-4-1"></span>**Fig. 3** Rotation meta-action unit

imprecise model expression due to the large number of components, and cannot consider all the error terms that afect the motion. The meta-action unit (MAU) is a structural unit composed of the moving part and other related components that ensure its correct action. Meta-action theory fnds a reasonable balance between the dependencies of components and systems. Most of the components that compose the meta-action unit are assembled in a fxed connection, and its movement is mainly accomplished by plane or cylindrical contact. To solve the relationship between indicators and geometric error items, the accumulation geometric error of intra-MAU and transmission kinematic precision of inter-MAU are calculated separately.

#### **3.1 Meta‑action basic concepts**

The meta-action unit refers to the most basic form of motion in mechanical products which can accomplish motion and

$$
\delta h_g = a^{(i,g)} \cdot \frac{n_g^r}{n_g^R} \cdot n \cdot t \cdot \sum_{k}^{m} \left[ k_k^{(i,g)} \left( P_k^{(i,g)} \right) \cdot P_k^{(i,g)} \right] + a^{(o,g)} \cdot \frac{n_g^r}{n_g^R} \cdot n \cdot t \cdot \sum_{k}^{m} \left[ k_k^{(o,g)} \left( P_k^{(o,g)} \right) \cdot P_k^{(o,g)} \right]
$$
\n
$$
= \frac{R_{or}}{R_g} \cdot (1+\vartheta) \cdot n \cdot t \cdot \left\{ a^{(i,g)} \cdot \sum_{k}^{m} \left[ k_k^{(i,g)} \left( P_k^{(i,g)} \right) \cdot P_k^{(i,g)} \right] + a^{(o,g)} \cdot \sum_{k}^{m} \left[ k_k^{(o,g)} \left( P_k^{(o,g)} \right) \cdot P_k^{(o,g)} \right] \right\}
$$
\n
$$
(17)
$$

Finally, the total wear volume of the bearing  $\Delta e_r(t)$  is calculated as

<span id="page-4-2"></span>power transmission. The most common forms of motion in mechanical systems are the rotary motion around the

$$
\Delta e_r(t) = \delta h_i + \delta h_o + \delta h_g = a^{(i,g)} \cdot \left(z + \frac{R_{or}}{R_g}\right) \cdot (1 + \vartheta) \cdot n \cdot t \cdot \sum_{k}^{m} \left[k_k^{(i,g)} \left(P_k^{(i,g)}\right) \cdot P_k^{(i,g)}\right] + a^{(o,g)} \cdot \left[z \cdot (1 - \vartheta) + \frac{R_{or}}{R_g} \cdot (1 + \vartheta)\right] \cdot n \cdot t \cdot \sum_{k}^{m} \left[k_k^{(o,g)} \left(P_k^{(o,g)}\right) \cdot P_k^{(o,g)}\right]
$$
\n(18)

## <span id="page-4-0"></span>**3 Output kinematic precision model based on meta‑action structural decomposition**

The performance of rotor-bearing systems is often presented as multiple indicators, and the essential factors that form and afect these indicators are the geometric characteristics of the components. Traditional precision modeling often leads to fxed axis and the linear motion along the mating surface. Therefore, these two motions are defned as rotation and translation MAU, respectively. Rotor-bearing systems are composed of rotation MAUs.

*M* represents the transmission torque,  $\theta$  the output angular displacement, *ω* the angular velocity, and *α* the angular acceleration, as shown in Fig. [3](#page-4-1).

<span id="page-5-0"></span>

MAU is the smallest granularity to accomplish an action, and the most basic structural unit that transmits motion and power in mechanical products. As is shown in Fig. [4](#page-5-0), MAU consists of a power input part, middle parts, fasteners, support parts, and a power output part. The power input part provides the power source or receives the output power of the upper MAU, such as gears and screws. The role of the middle parts is to transmit motion and ensure the correct position of the power input and output parts, such as connecting keys, rotating shafts, locating pins, bearings, and bushings. Fasteners play the role of connection, tightening, anti-loosening, and sealing, such as screws, springs, sealing rings, and end caps. The boxes and bases are generally taken as the support parts. The power output part is used as the end movement or power output.

## **3.2 Spatial pose of the power output part in MAU**

The essence of MAU is to ensure that the power output part completes the designed motion. In order to achieve the expected motion trajectory, the position and orientation errors of the power output part in the *X*, *Y*, *Z*,  $\alpha$ ,  $\beta$ , and  $\gamma$  directions in the space coordinate system need to be within the specifed error range, and the geometric errors of the power input and

output parts. The spatial error of the power output part is the accumulation result of machining errors, assembly errors, and wear of the components in MAU. During the assembly process of MAU, the actual pose of the power output part is offset relative to the ideal pose in the *X*, *Y*, *Z*,  $\alpha$ ,  $\beta$ , and  $\gamma$  directions in the space coordinate system, as shown in Fig. [5](#page-5-1).

The screw theory is used to calculate the accumulation error between the parts in MAU. When the arbitrary point in the space coordinate system rotates around a unit vector  $\omega$  by an angle  $\theta$ , its rotation matrix can be calculated according to Rodrigues' rotation equation as

$$
e^{\theta \widehat{\omega}} = \mathbf{E} + \widehat{\boldsymbol{\omega}} \sin \theta + \widehat{\boldsymbol{\omega}}^2 (1 - \cos \theta)
$$
 (19)

where the antisymmetric matrix

<span id="page-5-2"></span>
$$
\widehat{\omega} = \begin{pmatrix} 0 & -\omega_z & \omega_y \\ \omega_z & 0 & -\omega_x \\ -\omega_y & \omega_x & 0 \end{pmatrix}.
$$

The motion of the arbitrary rigid body in the space coordinate system can be regarded as a spiral motion around a certain vector axis. The screw-exponential product form of its coordinate change is expressed as

$$
\mathbf{g} = \begin{pmatrix} e^{\theta \widehat{\mathbf{\omega}}} \left( \mathbf{E} - e^{\theta \widehat{\mathbf{\omega}}} \right) (\boldsymbol{\omega} \times \mathbf{v}) + \theta \boldsymbol{\omega} \boldsymbol{\omega}^T \mathbf{v} \\ 0 & 1 \end{pmatrix}
$$
 (20)



<span id="page-5-1"></span>**Fig. 5** Space pose formation process of the power output part



<span id="page-6-0"></span>**Fig. 6** Two ft forms in rotor-bearing systems

where  $\nu$  represents the displacement vector moving along the vector axis.

The pose change of the power output part in the space coordinate system can be decomposed into the spiral motion around the *X*, *Y*, and *Z* directions, and further decomposed into the movement  $\delta_i$  ( $i = x, y, z$ ) and rotation  $\varepsilon_j$  ( $j = x, y, z$ ) around the *X*, *Y*, and *Z* axes. Considering that the error amount is infnitesimal relative to the geometric size of the part, let  $\sin \epsilon_i = \epsilon_i$ ,  $\cos \epsilon_i = 1$  and ignore the high-order infinitesimal. Then, according to Eq.  $(20)$  $(20)$ , the pose change screw expression of the power output part is obtained as

$$
\mathbf{g}^o = \prod_{\substack{i = x, y, z}} \mathbf{g}_{\delta_i} \mathbf{g}_{\epsilon_j} = \begin{pmatrix} 1 & -\epsilon_z & \epsilon_y & \delta_x \\ \epsilon_z & 1 & -\epsilon_x & \delta_y \\ -\epsilon_y & \epsilon_x & 1 & \delta_z \\ 0 & 0 & 0 & 1 \end{pmatrix}
$$
(21)

The basic ft forms in rotor-bearing systems include plane ft and cylindrical ft, which are shown in Fig. [6](#page-6-0).

As shown in Fig. [6](#page-6-0)a, the plane ft constrains three degrees of freedom in the *X*,  $\beta$ , and  $\gamma$  directions, so errors in these three directions are also generated. Assuming that all parts in MAU are rigid bodies, and their contact is a rigid ft, the spatial errors of plane ft during the assembly process are represented by homogeneous coordinate transformation as

$$
\mathbf{g}_p^o = \begin{pmatrix} 1 & -\varepsilon_z & \varepsilon_y & \delta_x \\ \varepsilon_z & 1 & 0 & 0 \\ -\varepsilon_y & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}
$$
(22)

where  $\delta_x \approx \frac{E}{2}$  $\frac{E}{2}$ ,  $\varepsilon_y \approx \frac{E}{D}$ , and  $\varepsilon_z \approx \frac{E}{D}$ . *D* represents the diameter of the rotating shaft and *E* represents the error value.

the MAU chain

As is shown in Fig. [6](#page-6-0)b, the plane fit constrains four degrees of freedom in the *Y*, *Z*, *β*, and *γ* directions, so errors in these four directions are also generated. The spatial errors of cylindrical ft during the assembly process are represented by homogeneous coordinate transformation as

$$
\mathbf{g}_c^o = \begin{pmatrix} 1 & -\varepsilon_z & \varepsilon_y & 0 \\ \varepsilon_z & 1 & 0 & \delta_y \\ -\varepsilon_y & 0 & 1 & \delta_z \\ 0 & 0 & 0 & 1 \end{pmatrix}
$$
 (23)

where  $\delta_y \approx \frac{E}{2}$  $\frac{E}{2}$ ,  $\delta_z \approx \frac{E}{2}$  $\frac{E}{2}$ ,  $\varepsilon_y \approx \frac{E}{L}$ , and  $\varepsilon_z \approx \frac{E}{L}$ . *L* represent the length of the rotating shaft and *E* represents the error value.

The spatial pose of the power output part is determined by the geometric errors  $\mathbf{g}^o_i(i=1, 2, ..., n)$ . The error expression model of the power output part after assembly is

<span id="page-6-3"></span>
$$
\mathbf{g}_m = \prod_i^n \mathbf{g}_i^o \tag{24}
$$

#### **3.3 Output precision of MAU chain**

According to the principle of freedom degree, the function of a MAU chain is generally accomplished by series structures. The MAU chain transmits power from the input end to the output end in the motion transmission direction (Fig. [7](#page-6-1)). Few rotor systems with redundant structures or dual driving structures are beyond the scope of this paper.

There are several motion precision indicators, such as indexing precision, repositioning precision, and radial runout, which are used to characterize the performance of the motion transmitted by the MAU chain. These indicators are obtained through the accumulation of the error components in each direction, which are caused by the spatial pose of the power output part and the geometric characteristics of the power input and output part. In this paper, the indexing precision reliability is selected as the performance evaluation index. Indexing precision is determined by the tooth profle error, tooth thickness error, and eccentricity errors of the power input and output parts. And the eccentricity errors include the assembly error component of the power output part in the direction perpendicular to the rotation axis.

<span id="page-6-2"></span>The tooth profile error refers to the normal distance between the two tooth profle lines with the smallest distance within the tooth width effective range on a given section (Fig. [8\)](#page-7-1). The tooth profle error is approximately equivalent

<span id="page-6-1"></span>

to a harmonic function of the tooth-to-tooth and total tangential composite errors.

Let the gear meshing frequency be  $\omega_m$  and the gear rotation frequency be  $\omega$ ; then, the tooth profile error is calculated as

$$
\Delta e_s = 0.5 \Delta f'_{ic} \sin(2\pi \omega_m t) + 0.5 \Delta F'_{ic} \sin(2\pi \omega_s t)
$$
 (25)

Let the initial tooth thickness error of the power input part in MAU be  $\Delta e_{ig}^I$ , the tooth thickness wear be  $\Delta e_g^I(t)$ , the tooth profile deviation be  $\Delta e^I_s$ , and the eccentricity error caused by the accumulated assembly error be  $\Delta e_{ip}^I$ . Angular velocity is  $\omega_m^I$ , pressure angle is  $\alpha$ . And initial phase angle  $\beta^I$  are random variables that obey uniform distribution in the interval  $[0, 2\pi]$ .

Let the initial tooth thickness error of the power output part in MAU be  $\Delta e_{ig}^O$ , the tooth thickness wear be  $\Delta e_g^O(t)$ , the tooth profile deviation be  $\Delta e_s^O$ , the eccentricity error caused by the accumulated assembly error be  $\Delta e_{ip}^O$ , and the eccentricity error caused by wear be  $\Delta e_p^O(t)$ . Angular velocity is  $\omega_m^O$ , pressure angle is *α*. And initial phase angle  $β^O$  are random variables that obey uniform distribution in the interval [0, 2*π*].

Then the indexing precision of MAU is obtained as

$$
g(X(t), Y, t) = \sum_{i}^{n} \Delta e_i(t) - \Delta e^*
$$
\n(28)

This function includes both the initial random error variables and the time-variant wear quantity. Stochastic process discretization is adopted to solve the time-variant reliability question. Firstly, the time-variant process in a certain time interval is discretized into many micro time units. When the discretization degree reaches a certain level and the infuence of time variables on the output value is ignored, the error in a micro time unit approximately obeys a Gaussian distribution. In this way, the time-variant reliability problem is transformed into the random variable problem in traditional reliability research.

Assuming that the running time of a rotor-bearing system is [0, *T*], the time interval is discretized into *m* equal micro time units with step length  $\Delta t = \frac{T}{m}$ . Then, the random process vector  $X(t) = (X_1(t), X_2(t), ..., X_n(t))$  with *n* time-variant wear parameters is discretized as

$$
X_i = [X_1(t_i), X_2(t_i), ..., X_n(t_i)]^T, i = 1, 2, ..., m
$$
 (29)

<span id="page-7-2"></span>(26)

$$
\Delta e_i(t) = \Delta e_{ig}^l + \Delta e_g^l(t) + \Delta e_s^l + \Delta e_{ip}^l \sin(2\pi \omega_m^l t + \beta^l + \alpha) + \Delta e_{ig}^O + \Delta e_g^O(t) +
$$
  

$$
\Delta e_s^O + \left(\Delta e_{ip}^O + \Delta e_p^O(t)\right) \sin(2\pi \omega_m^O t + \beta^O + \alpha)
$$

Assuming that the rotor-bearing system is composed of *n* MAUs, then the indexing precision of the rotor-bearing system can be obtained as

$$
\Delta e(t) = \sum_{i}^{n} \Delta e_i(t) \tag{27}
$$

## <span id="page-7-0"></span>**4 The construction of kinematic precision reliability evaluation model**

Assuming that the allowable indexing accuracy of rotor-bearing systems is Δ*e*<sup>∗</sup> , the precision reliability function of rotor-bearing systems is built as



<span id="page-7-1"></span>**Fig. 8** One-tooth meshing error of the worm wheel

Thus, the kinematic precision reliability of rotor-bearing systems is expressed as [\[35](#page-14-1)]

$$
P_{s}(T) = P\left\{\bigcap_{i=1}^{m} \left[g(X_{i}, Y, t_{i}) > 0, t_{i} = \left(i - \frac{1}{2}\right)\Delta t, \Delta t = \frac{T}{m}\right]\right\}
$$
\n(30)

where  $g(X_i, Y, t_i)$  is the random performance function discretized from the time-variant performance function  $g(X(t), Y, t)$ , and *Y* is an *l*-dimensional random vector independent of  $X(t)$ .

Considering the correlation between the discrete random vectors, the parameters are converted into independent standard normal distributions before solving. Based on Nataf transform [\[36](#page-14-2)], the random vectors  $X_i$  and  $Y$  in Eq. ([30](#page-7-2)) are converted into standard normal distribution vectors  $U_i$  and *V* as follows:

$$
U_{i,k} = \Phi^{-1}\Big(F_{X_{i,k}}(X_{i,k})\Big), i = 1, 2, ..., m; k = 1, 2, ..., n \tag{31}
$$

$$
V_p = \Phi^{-1}\Big(F_{Y_p}(Y_p)\Big), p = 1, 2, ..., l
$$
\n(32)

where  $\Phi^{-1}(\cdot)$  is the cumulative distribution inverse function of the standard normal distribution, and  $F_{X_{i,k}}(\cdot)$  and  $F_{Y_p}(\cdot)$  are the cumulative distribution functions of random vectors *X<sup>i</sup>* and *Y*, respectively.

Let  $g^*(U_i, V, t_i)$  represent the transformed performance function of the random vectors  $U_i$  and  $V$ ; then, the most probable point of the function is solved as

$$
\begin{cases} \min \left\| \left( \boldsymbol{U}_i^T, \boldsymbol{V}^T \right)^T \right\| \\ s.t. g^* \left( \boldsymbol{U}_i, \boldsymbol{V}, t_i \right) = 0 \end{cases}
$$
\n(33)

Thus, the calculation equation of  $\beta_i$ , the *i*-th index value in the reliability index vector  $\beta$  corresponding to  $g^*(U_i, V, t_i)$ , is obtained as

$$
\beta_i = \left\| \left( \overline{U}_i, \overline{V} \right) \right\| \tag{34}
$$

where  $(U_i, V)$  is the most probable point of the *i*-th performance function  $g^*(U_i, V, t_i)$ .

Using the frst-order Taylor series to expand the functional function  $g^*(U_i, V, t_i)$  at the most probable point, the reliability calculation equation of the approximate performance function is obtained as

$$
P_s(T) = P\left\{\bigcap_{i=1}^m \beta_i + \alpha_{U,i} U_i^T + \alpha_{V,i} V^T > 0\right\}
$$
(35)

where  $\alpha_{U,i} = \frac{\overline{U}_i}{\beta_i}$  and  $\alpha_{V,i} = \frac{\overline{V}}{\beta_i}$  are the gradient vectors of the random vector  $U_i$  and  $V$ , respectively.

Let  $L_i = \beta_i + \alpha_{U,i} U_i^T + \alpha_{V,i} V^T$ . Since the transformed random vectors  $U_i$  and  $V$  are independent standard normal distribution vectors,  $L_i$  obeys a normal distribution with the expected value  $\mu_i = \beta_i$  and the standard deviation  $\sigma_i = 1$ . Considering that the time-variant wear quantity function is equivalent to a Gaussian process, the correlation coefficient between random variables is obtained as

$$
\rho_{i,j} = \frac{Cov(L_i, L_j)}{\sigma_{L_i} \sigma_{L_j}} = Cov(L_i, L_j) = \sum_{k=1}^{m} \alpha_{U,i,k} \alpha_{U,j,k} C_k(t_i, t_j) + \alpha_{V,i}^{T} \alpha_{V,j}
$$
\n(36)

where  $C_k(\cdot)$  represents the autocorrelation function of timevariant parameter  $X_k(t)$ .

Let the correlation coefficient matrix  $\rho = [\rho_{ij}]$ . Thus, using the frst-order second-moment method to solve Eq. ([30\)](#page-7-2), the precision reliability of the rotor-bearing system is obtained as

$$
P_s(T) = \Phi_m(\beta, \rho) \tag{37}
$$

# <span id="page-8-0"></span>**5 Case investigations**

In this section, a numerical control rotary table is taken as an example to elaborate the proposed method in detail, as shown in Fig. [9.](#page-8-1) The rotary table is composed of the gear MAU, the worm shaft MAU, and the worktable MAU. The gear MAU provides power input, the worm shaft MAU transmits motion and power, and the worktable MAU realizes the function of indexing and positioning.

As shown in Fig. [10,](#page-9-0) the wear of the worktable MAU mainly occurs from the worm wheel and bearing. The structural parameters of the worm wheel are listed in Table [1;](#page-9-1) the structural parameters of the bearing are listed in Table [2.](#page-10-0)

According to Eqs.  $(3)$ – $(6)$  $(6)$ , the time-variant wear (μm/h) of the worm wheel is calculated as

$$
\Delta e_g^{(w)I}(t) = 0.034707t
$$

According to Eqs.  $(9)$  $(9)$ – $(18)$  $(18)$  $(18)$ , the time-variant wear (μm/h) of the bearing is calculated as

$$
\Delta e_p^{(w)O}(t) = 0.0005063t
$$

Generally, the initial random errors of the components are subject to normal distribution within the tolerance range. The distribution information of the random errors affecting spatial pose errors of the power output part during the worktable MAU assembly process are listed in Table [3.](#page-10-1)

The values of each random error are obtained by Monte Carlo simulation method [\[37](#page-14-3)]. Substituting these values into



<span id="page-8-1"></span>**Fig. 9** The numerical control rotary table

Item Value Item Value Poisson ratio  $0.33$  Axial module 2.25 Elastic modulus  $1.19 \times 10^{11}$  N/m.<sup>2</sup> Surface roughness 1.6  $\mu$ m

Torque 180 N·m Face width 0.036 m Speed 11.1 r/min Pressure angle 15°

<span id="page-9-1"></span>**Table 1** Structural parameters of the worm wheel

 $1.6819 \times 10^{-8}$  $m<sup>2</sup>/N$ 



<span id="page-9-0"></span>**Fig. 10** The structure of the worktable MAU

Eqs. ([22](#page-6-2)[–24\)](#page-6-3), the spatial pose errors of the power input part and the power output part are calculated. The ftting joint distribution of error components in each degree of freedom can be obtained with enough sampling times. The combined distribution of the eccentric error of the worm wheel and worktable is as follows:

$$
\Delta e_{ip}^{(w)I} \sim N(4.25, 2.25^2), \Delta e_{ip}^{(w)O} \sim N(28.25, 6.80^2)
$$

The geometric characteristic error distribution of the worm wheel is listed in Table [4:](#page-10-2)

According to the model in Sect. [3](#page-4-0), the indexing precision of the worktable MAU is calculated as

$$
\Delta e^{(w)}(t) = \Delta e_{ig}^{(w)I} + 0.5 \Delta f_{ic}^{(w)I} \sin(2\pi \omega^{(w)}t) + 0.5 \Delta F_{ic}^{(w)I} \sin(2\pi \omega^{(w)}t) +
$$
  
0.034707t +  $\Delta e_{ip}^{(w)I} \sin(2\pi \omega^{(w)}t + \beta^{(w)I} + \alpha^{(w)}) +$   

$$
\Delta e_{ip}^{(w)O} + 0.0005063t \sin(2\pi \omega^{(w)}t + \beta^{(w)O} + \alpha)
$$

Similarly, the indexing precision of the worm shaft MAU is calculated as

when the NC rotary table has been operating for 930 h, its kinematic precision reliability begins to decrease. The confdence interval is generally 95%; when the operation time reaches 1255 h, the kinematic precision reliability is reduced to 0.95. One thousand two hundred ffty-fve hours is considered as the precision life limit of the rotary table, and maintenance is required. In comparison with the expected

The probability distributions of random variables in gear

Base radius 0.10125 m

The indexing precision of the rotary table is calculated as

Considering that the output precision of the servo motor is generally fxed, the allowable indexing precision value after subtracting the servo motor precision is  $\Delta e^* = 80 \mu m$ . The numerical analysis is carried out according to the method in Sect. [4.](#page-7-0) According to the simulation results shown in Fig. [11,](#page-11-1)

MAU are listed in Table [6](#page-11-0).

Viscosity-pressure coefficient of lubricant

 $\Delta e(t) = \Delta e^{(w)}(t) + \Delta e^{(s)}(t) + \Delta e^{(g)}(t)$ 

1200-h precision life, it can be determined that the precision design of the NC turntable is reasonable and the proposed evaluation model is effective. As demonstrated in the figure, the curve is gentle at the initial stage of kinematic precision reliability decline, and the precision reliability decreases signifcantly when reaching the precision limit. This is also identical with the actual operation situation. Consequently, it shows that this method can accurately predict the precision life of the rotary table in the design stage.

$$
\Delta e^{(s)}(t) = \Delta e_{ig}^{(s)I} + 2.9264 \times 10^{-5}t + 0.5\Delta f_{ic}^{(s)I}\sin(2\pi\omega^{(s)}t) + 0.5\Delta F_{ic}^{(s)I}\sin(2\pi\omega^{(s)}t) +
$$
  
\n
$$
\Delta e_{ip}^{(s)I}\sin(2\pi\omega^{(s)}t + \beta^{(s)I} + \alpha^{(s)I}) + \Delta e_{ig}^{(s)O} + \Delta e_{is}^{(s)O} +
$$
  
\n
$$
\Delta e_{ip}^{(s)O} + 0.0167t \sin(2\pi\omega^{(s)}t + \beta^{(s)O} + \alpha^{(s)O})
$$

The probability distributions of random variables in worm shaft MAU are listed in Table [5.](#page-10-3)

The indexing precision of the gear MAU is calculated as

Further, we investigate the infuence of diferent wear resistances of component materials on the kinematic precision life of the rotary table. From the calculation results, it

$$
\Delta e^{(g)}(t) = \Delta e_{ig}^{(g)O} + 5.8528 \times 10^{-5} t + 0.5 \Delta f_{ic}^{(g)O} \sin(2\pi \omega^{(g)} t) + 0.5 \Delta F_{ic}^{(g)O} \sin(2\pi \omega^{(g)} t) +
$$
  
+  $\Delta e_{ip}^{(g)O} \sin(2\pi \omega^{(g)} t + \beta^{(g)O} + \alpha^{(g)})$ 

## <span id="page-10-0"></span>**Table 2** Structural parameters of the bearing



# <span id="page-10-1"></span>**Table 3** Distribution information of each random error/μm



<span id="page-10-2"></span>

Error	Distribution type	Distribution parameters
Tooth-to-tooth tangential composite error $\Delta f_{ic}^{'(w)I}/\mu$ m	Rayleigh distribution	R(1.2)
Total tangential composite error $\Delta F_{ic}^{(w)I}$ /µm	Rayleigh distribution	R(2.6)
Tooth thickness error caused by manufacturing $\Delta e_{i\varrho}^{(w)I}/\mu$ m	Normal distribution	$N(0,5.72.^2)$
Initial phase angle of eccentricity error $\beta^{(w)}$ /rad	Uniform distribution	$U(0,2\pi)$

<span id="page-10-3"></span>**Table 5** The probability distributions of random variables in worm shaft MAU



<span id="page-11-0"></span>**Table 6** The probability distributions of random variables in gear MAU



<span id="page-11-1"></span>



<span id="page-11-2"></span>**Fig. 12** Efects of the material type on the kinematic precision reliability



<span id="page-12-0"></span>



indicates that the main wears locate on the worm wheel of the NC rotary table. Therefore, three diferent grades of copper are selected for simulation. Based on the results shown in Fig. [12,](#page-11-2) when ZCuSn10P1 is selected as the worm wheel material, the precision life of the rotary table is 806 h. When ZCuAl10Fe3Mn2 is used, the precision life increases by 449 h. When the worm wheel is made of the ZCuAl9Fe4Ni4Mn2, the precision life is further improved to 1682 h. Combined with the conversion cost of each material and the maintenance cost of the rotary table, the material selection in the design stage can be realized.

Diferent tolerance ranges of components also have an impact on the kinematic precision reliability. In order to reveal the infuence, the assembly errors of the worm wheel are taken as an example; diferent distribution ranges of the worm wheel eccentric error are selected for simulation. According to the results shown in Fig. [13](#page-12-0), when  $\Delta e^{(w)O}_{ip} \sim N(19.78, 6.80^2)$ , the manufacturing error of the components is smaller, and the precision life of the rotary table increases to 1300 h. When  $\Delta e^{(w)O}_{ip} \sim N(36.73, 6.80^2)$ , this indicates that the manufacturing quality is relatively poor and leads to a reduction in precision life to 1170 h. Thus, the tolerance optimization design can be realized according to the precision demand and cost.

## **6 Conclusions**

In this paper, a practical and efective approach for kinematic precision reliability evaluation of rotor-bearing systems is proposed innovatively. This method has advantages in the quantity of error terms and the accuracy of assessment results, and can be provided with great application potential in precision design and maintenance strategy for rotorbearing systems. The study contains the following:

- 1. A comprehensive kinematic precision model describing the mapping relationship from component geometric errors to the system output precision is proposed based on the meta-action theory and screw theory, and it can overcome the limitations of the traditional methods in the number of error sources.
- 2. Both the initial errors conforming to specifed random distributions and the time-dependent wears are contained in the precision reliability function of rotor-bearing systems; a kinematic precision reliability evaluation model with multiple error sources for rotor-bearing systems is provided by introducing the stochastic process discretization method.
- 3. An example for a numerical control rotary table is investigated to demonstrate the efectiveness of this approach. The simulation results obtained by changing the parameters of error sources show that the proposed method is practical and helpful to increase the precision reliability of rotor-bearing systems in the design phase.

However, this method only calculates the wear errors among the time-variant error sources; some random errors afected by changeable environmental factors during the operation of rotor-bearing systems also need to be seriously considered, such as thermal errors, deformation errors caused by residual stress release, and runout errors caused by fastener loosening. In addition, diferent lubrication states will afect the wear rate, and the fatigue wear is also an important constituent part in the wear of gears. The infuence of the above-mentioned factors on the kinematic precision reliability of the systems will be comprehensively taken into consideration in our future investigations.

**Author contribution** All authors contributed to the study conception and design. Material preparation, data collection, and analysis were performed by Hongwei Wang, Jiawei Xiang, and Yulong Li. The frst draft of the manuscript was written by Hongwei Wang, and all authors

commented on previous versions of the manuscript. All authors read and approved the fnal manuscript.

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## **Declarations**

**Competing interests** The authors declare no competing interests.

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