

Experimental study of fibre laser microdrilling of aerospace superalloy by trepanning technique

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Abstract In the present study, fibre laser microdrilling of NIMONIC®263 sheet is investigated through an experimental testing campaign. Design of experiments (DOE) and analysis of variance (ANOVA) are applied with the aim to study the influence of the process parameters on the hole geometry and metallurgical properties. The results show that the laser allows drilling ~0.4-mm-thick NIMONIC®263 sheets with a trepanning speed up to 600 mm/min, obtaining good tolerance, low recast layer and absence of microcracks.

Keywords Nimonic · C263 · Recast layer · Effusion cooling system

1 Introduction

Nowadays, the development of higher-efficiency aero-engines involves the increasing of the flame temperature. The latter allows a reduction of the emission levels and the pressure losses. However, the increase of flame temperature is limited by the maximum temperature of the materials available for the engine construction [1]. To protect the combustor component

from the hot temperature, it is possible to form a cooling film of air on the surface of the component, using of a large number of small holes (typically less than 1.0 mm in diameter) [2, 3]. Thus, in the hot section of aero-engines, an effusion cooling system is placed inside the turbine blade, allowing reducing thermal stress and avoiding premature failure of the turbine blades [3]. The turbine blade effusion system consists of a thin sheet, with more than 200 of neighbouring holes; the latter allows obtaining a cooling film within the blade wall. The holes diameters vary in the range of 0.3–1.0 mm.

Currently, electrical discharge machining (EDM) technique is adopted to drill the holes for effusion cooling systems of turbine blades in the hot section of aero-engines to satisfy the hole requirements (tight tolerances, no cracks in the base material and low recast layer). On the other hand, EDM microdrilling needs long process time and adopts dielectric liquid in which the workpiece is immersed. This liquid is not environment-friendly and needs to be processed.

Laser beam machining (LBM) represents a possible solution [4–10]. Compared to traditional technologies, laser drilling offers several benefits (i.e. absence of mechanical contact and tool wear reduction of liquid pollutants, no need for complex fixtures, high productivity, process flexibility and the possibility to create complex shapes and accurate geometries with narrow kerfs) on almost all categories of materials including metals [11–16], non-metals [17], ceramics [18] and composites [19].

In this research work, laser microdrilling of NIMONIC®263 sheets, 0.38 mm in thickness, by a 100 W fibre laser, working in modulated wave mode, is investigated. Two experimental test series were carried out. At first, pre-tests were executed varying the speed in the range 500–1000 mm/min, in order to identify a speed compatible with the machine dynamic. On the basis of the tests' results, and on the results of previous studies [20, 21], the control factor and their levels were individuated. In the second

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Table 1 Main laser system characteristics (SPI-RedPower SP100C)

Parameters	Value	Unit
Wavelength	1090	[nm]
Maximum power	100	[W]
Mode operation	CW or modulated	–
Pulse frequency	1–18	[kHz] ^a
Pulse duration	1–0.01 ms	[ms] ^a
Beam diameter (1/e ²)	5.0 ± 0.5	[mm]
Full angle divergence	<0.4	[mrad]
Beam quality	TEM ₀₀ (M ² < 1.1) BPP 0.38	– [mm.mrad]
Focal length	50	[mm]
Beam diameter at the focal spot	≈48	[μm]
Nozzle diameter	0.5	[mm]

^a In modulated regime

experimental test series, drilling tests were performed by varying the speed, the pulse duration, the focus position and the tool path. The obtained holes were investigated by digital microscopy (KH-8700 by Hirox). The hole diameter and circularity were measured at the entrance and the exit of the laser beam.

Table 2 Chemical composition and properties of NIMONIC@263

Element	Ni	Co	Cr	Mo	Ti	C
Min [%]	Bal.	19	19	5.6	1.9	0.04
Max [%]		21	21	6.1	2.4	0.08
Element	Mn	Al	Si	Cu	Fe	S
Min [%]	–	–	–	–	–	–
Max [%]	0.60	0.60	0.4	0.20	0.07	0.007
Element	B	Pb	Ag	Bi		
Min [%]	–	–	–	–		
Max [%]	0.005	0.002	0.0005	0.0001		
Physical properties	Value		Units			
Density	8.36		[g/cm ³]			
Melting Range						
Liquidus temperature	1355		[°C]			
Solidus temperature	1300					
Specific heat	461		[J/kg, °C]			
Thermal conductivity	11.72 ^a		[W/m°C]			
Linear thermal expansion	11.0		[10 ⁻⁶ /°C]			
Mechanical properties ^a	Value		Units			
Tensile strength	1004 ^b		[MPa]			
Yield strength (at 0.2%)	585 ^b		[MPa]			
Elongation	45 ^b		[%]			
Reduction of area	41 ^b		[%]			
Young's modulus	221 ^b		[GPa]			
Torsional modulus	86 ^b		[GPa]			

^a Heat treatment: 2 h/1150 °C/WQ + 8 h/800 °C/AC

^b At 20 °C

Table 3 Control factors and setting levels

Control factor	Low (-1)	Middle (0)	High (+1)	Unit
Pulse duration, <i>D</i>	0.1	0.2	0.4	[ms]
Speed, <i>S</i>	200	400	600	[mm/min]
Focus position, <i>F</i>	0	–	0.2	[mm]
Tool path, <i>Tp</i>	TP1	–	TP2	–

Analysis of variance (ANOVA) was adopted to study the influence of the process parameters on the hole characteristics.

2 Equipment, material and experimental procedures

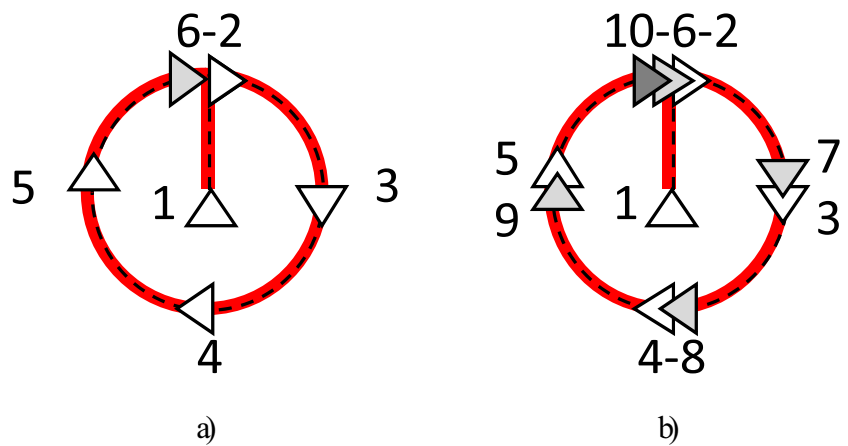
A fibre laser (RedPower SP100C by SPI), working at the wavelength, $\lambda = 1090$ nm, was adopted for the drilling tests. In this device, the laser radiation is transferred via an optical fibre to the laser head (by HAAS LTI), that is mounted in a 3 + 1 axis CNC system, finecut (Y 340 M by ROFIN). An external laser controller (MCA LCT3001) controls the power (from 10 to 100% maximum nominal power) and the regime (continuous wave or modulated). In modulated regime, the pulse frequency and the pulse duration are settable by the controller. The CNC system controls the laser source power, the geometric patterns and the beam speed. Table 1 shows the main laser system characteristics.

The material used in the present investigation is NIMONIC@263, under form of rolled sheets, 0.38 mm in thickness, (UNS N07263/W. Nr. 2.4650). The NIMONIC® alloy 263 has high properties in terms of creep strength and stress. Table 2 shows the chemical composition and the main properties of NIMONIC@263 alloy.

The tests companion was developed adopting a systematic approach to design experiments, as proposed in [22] and successfully applied in [19–21, 23–26]. Pre-design sheets were developed on the basis of main bibliography [5–15], relevant background [20, 21] and pre-tests.

On the basis of bibliographic data [6], two laser-drilling techniques were considered: percussion drilling and trepanning drilling. Although percussion drilling has the potential for faster drilling times, trepanning generally gives better hole quality compared to the first one. On the other hand, trepanning drilling requires high quality beam (i.e. narrow beam with low M² factor). Consequently, it was chosen to adopt the latter technique together to a high brilliance laser source. The experimental campaign was developed in two steps. First, in order to find out a good level range for machining speed in trepanning drilling, pre-tests were performed. During this phase, the other process parameters were fixed according to

Fig. 1 Tool path for trepanning. **a** TP1. **b** TP2. In the figure, the arrows and the numbers indicate the path sequences



the value reported in Table 3. The speed values were identified on the basis of previous research [20, 21].

The chosen tool path (TP1) for the present tests was as follow: first the laser generates an initial hole (piercing), smaller than intended, at the centre of required location. After piercing, the laser beam moves along the radius and then to an orbit describing a circular path, to create the final hole (see Fig. 1a).

To study the influence of the control factors (i.e. laser parameters) on the hole characteristics, a second set of tests was performed, starting from the information came out from the preliminary tests. In this second phase, following to the design of experiments (DOE) methodology, a factorial plan was adopted.

The adopted control factors were pulse duration D , speed S focus position F , and tool path, Tp . The chosen tool paths for second test campaign were as follow: first, piercing in the middle of the area where the hole is to be produced, then the laser beam moves one time (TP1) or two time (TP2) through an orbit describing a circular path to create the final hole (see Fig. 1a, b).

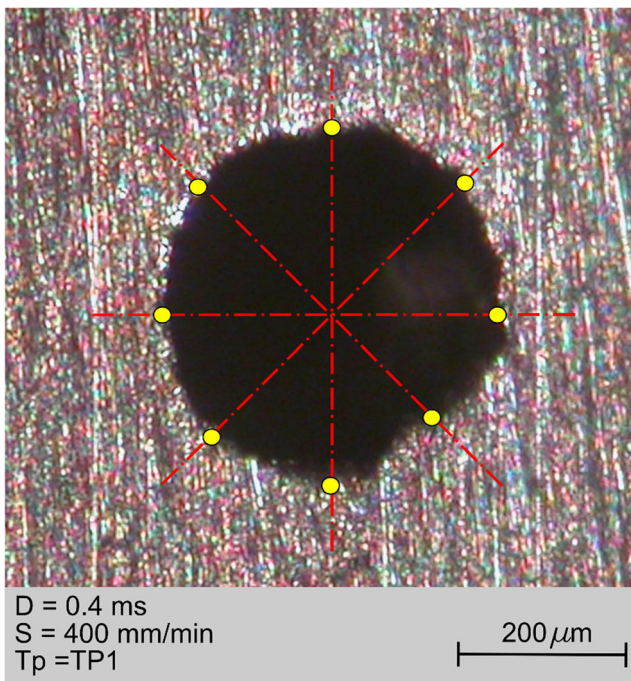


Fig. 2 Strategy for measuring with eight sectors

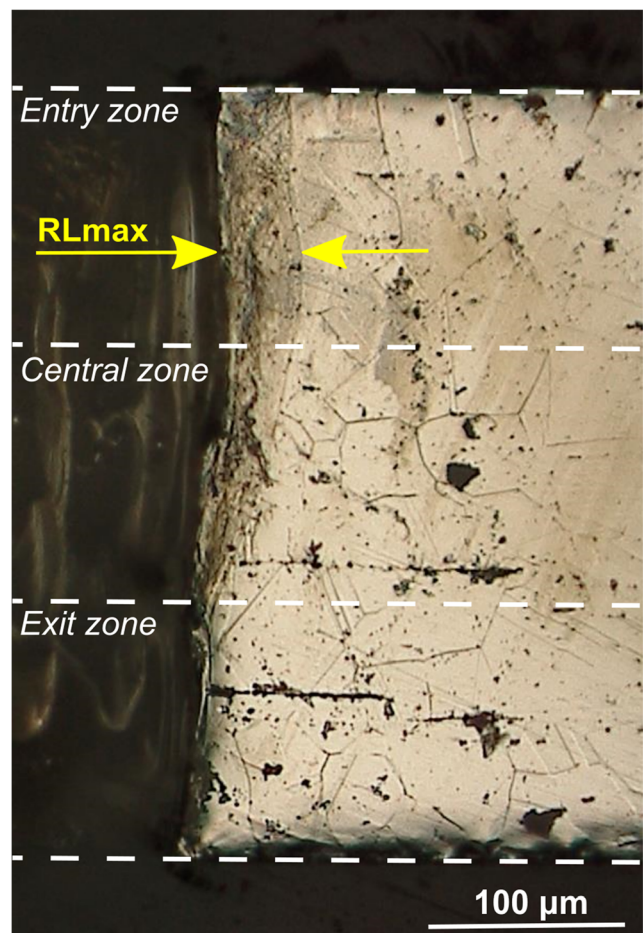


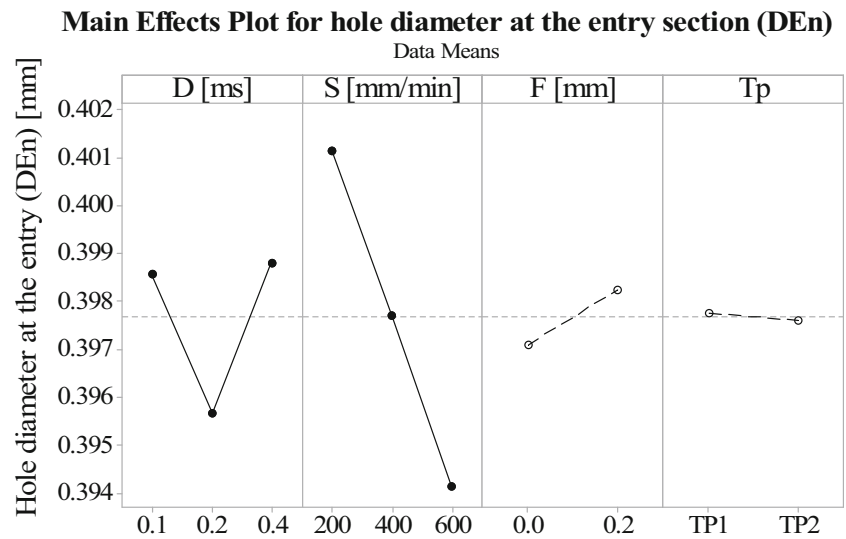
Fig. 3 Schematic of recast layer (RLmax) measurement

Table 4 Result of ANOVA in terms of p value

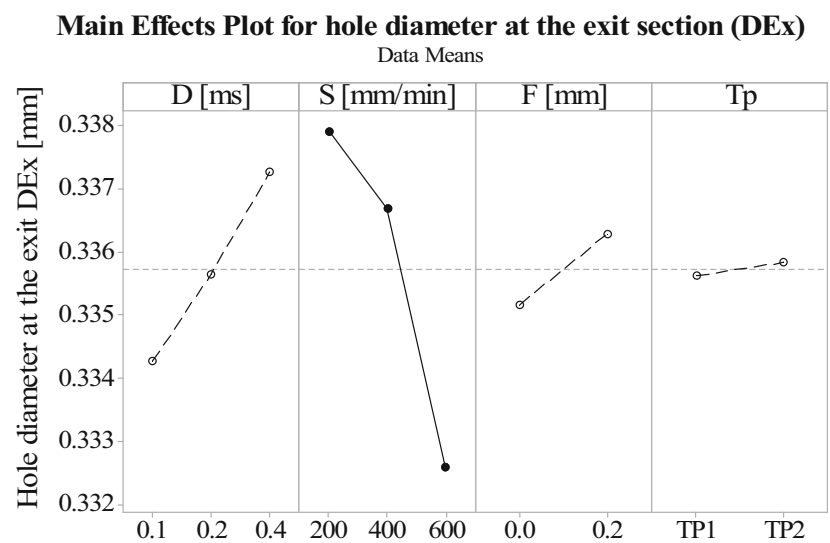
Source	Hole diameter at the entry (D _{En})	Hole diameter at the exit section (D _{Ex})	Hole circularity at the entry section (C _{En})	Hole circularity at the exit section (C _{Ex})	Recast layer (RL _{max})	Taper angle (T _a)
<i>D</i> (pulse duration)	<i>0.012</i>	0.141	0.687	0.843	0.592	0.474
<i>S</i> (speed)	<i>0.000</i>	<i>0.001</i>	0.315	0.272	0.453	0.214
<i>F</i> (focus position)	0.174	0.361	0.938	0.649	0.281	0.783
<i>TP</i> (Tool path)	0.994	0.866	0.334	0.504	0.589	0.746
D*S	0.119	0.267	<i>0.031</i>	<i>0.014</i>	0.196	0.120
D*F	0.193	0.393	0.449	0.368	0.756	0.068
D*TP	0.452	0.923	0.864	0.732	0.806	0.732
S*F	0.330	0.358	0.216	0.202	0.802	0.777
S*TP	0.552	0.910	0.072	0.631	0.426	0.180
F*TP	0.161	0.813	0.529	0.888	0.417	0.465

The significant control factors (p value < 0.05) are italicized

Fig. 4 Main effects plots for diameters at the **a** entry section (D_{En}) and **b** exit section (D_{Ex})



a)



b)

Table 3 summarizes the control factors and their settings levels, adopted in the experimental phase. A full factorial experimental design was performed.

Each treatment (i.e. process condition) was repeated four times (four replications). Thus, a total of 144 experimental runs were performed. The replications of each treatment were executed in order to provide more consistent response repeatability during the first experimental study.

The order of trials was completely randomized (treatments and replications) to reduce the disturbance of any unconsidered noise factor.

The response variables considered for this study are the diameters of the entry (D_{En}) and exit (D_{Ex}) sections of the hole; the circularity of the hole for the entry (C_{En}) and exit (C_{Ex}) sections; recast layers (RL_{max}); and taper angle (Ta). Moreover, the presence of cracks and oxides were also checked.

According with ISO 12181-1:2011 (Geometrical product specifications—GPS—Roundness, which define the diameter as a parameter associated with a circle), the measuring was developed through an internal algorithm to detection system Quadracheck200 in combination to the microscope Stemi 2000CS. LSBF: Fit determined by minimizing the sum of the squared point deviation from the form fit.

The measuring procedure requires a minimum of three points to measure a circle, and a maximum of 100 points can be probed and will be processed by a fit algorithm to define the circle. So, in order to standardize the measurement procedure, removing the uncertainty connected to the operator, and reduce the measure time, a subdivision into eight sectors was used, as shown in Fig. 2. The eight points, in the intersection between the hole border and the sector pattern, were used for the hole fitting. The use of eight points was considered a good compromise in term of accuracy and measure time. Finally, the circularity at the entry and exit section was calculated as the ratio between the minimum inscribed and the maximum circumscribed diameters.

After the hole characteristic measurement, the specimens were embedded with an epoxy resin and then polished, using abrasive paper up to a grit size of P2500 (Standard ISO 6344). The recast layer extension was measured in three zones: at the entry, in the central, and at the exit section. The greatest value (worst condition) was adopted for the analysis. Figure 3 depicts a schematic of recast layer measurement.

3 Experimental results and discussion

All the 144 samples have never presented cracks, neither in the base material nor in the recast layer. This result is due to two reasons: in comparison with other superalloys (e.g. Inconel 718), the investigated material is less prone to crack formation. Moreover, the adopted average power and pulse

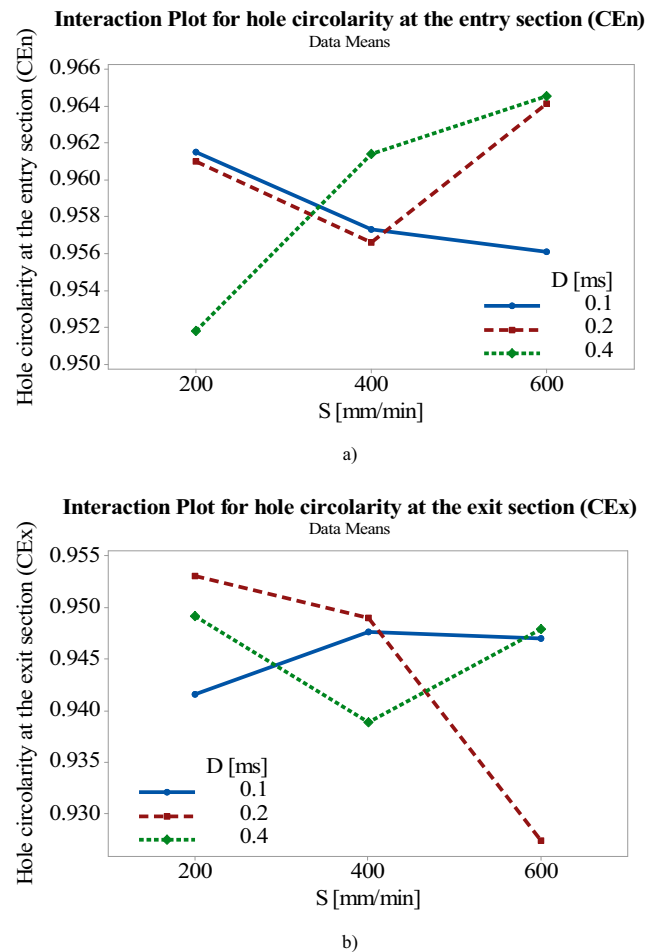


Fig. 5 Interaction plots for circularity of the hole at a the entry section (C_{En}) and b exit section (C_{Ex})

power are very limited (80 and 100 W); thus, no large thermal shock occurs. This represents a fundamental result for the experimentation. In fact, the aeronautical restrictions require total absence for the cracks in the base material, since, as well known, they represent points of weakness in the final component.

In addition, the metallurgical analysis gave a total absence of oxides. This result was possible thanks to the use of an inert assistant gas (Nitrogen) during processing. However, it is necessary to specify that the use of such gas was effective because of the little thickness of the machined workpiece.

Table 5 Optimal parameter set

Control factor	Best set	Unit
Pulse duration, <i>D</i>	0.4	[ms]
Speed, <i>S</i>	600	[mm/min]
Focus position, <i>F</i>	0.2	[mm]
Tool path, <i>T_p</i>	TP1	–

Regarding the recast layer, it was found that about 65% of the treatments results into the aeronautical specification (thickness of recast layer less than 0.040 mm).

The ANOVA was used to check the statistical significance of the factor effects for each response variable. Before the analysis, the assumptions that the observations are normally and independently distributed were successfully checked via analysis of residuals, in agreement with [27]. Table 4 shows the ANOVA results, in terms of p values, for the studied response variables. Assuming a 95% confidence level ($\alpha = 0.05$), a control factor, or a combination of control factors, is considered significant where the p value is less than 0.05.

From Table 4, the DEN is affected by pulse duration (D) and speed (S); the DEx is affected by S , and the circularity (for both entry, Cen, and exit sections, CEx) is affected by interaction between pulse D and S . The ANOVA does not show any significant effect for RLmax and Ta. Figure 4a, b shows the main effect plots for the diameters of the entry and exit sections of the hole, respectively. The significant effects are drawn by continues lines. Figure 5 shows the significant interaction plots of circularity for the exit section.

Figure 4a depicts the main effects plot for DEN; this diameter has a minimum when the middle value of D is adopted ($D = 0.2$ ms). This phenomenon was not expected. Generally speaking, when higher pulse duration is adopted, a longer interaction time occurs; therefore, an increase of the diameter is expected, as reported in [28]. It is worth noting that the adopted laser source works in modulated regime; thus, to ensure a constant average power (80 W), when the pulse duration is low ($D = 0.1$ ms), the pulse frequency is high ($F = 8$ kHz). The opposite occurs when the duration is high ($D = 0.4$ m and $F = 2$ kHz). Therefore, it is possible to assume that the behaviour of the DEN is the result of two competing effects: the pulse width (which increases diameter) and the frequency (which decreases diameter).

Regarding the effect of the S on DEN, the behaviour shown in Fig. 4a is in accordance with the literature [10, 16, 29–31]: for a fixed average power, an increase in speed reduces the

amount of energy released per unit length and thus the amount of removed material.

Similar remarks can be made on DEx, it decreases when the S increases (Fig. 4b).

Generally speaking, high speed values ensure low processing time, but results in low hole quality, either for the spot overlap decreasing and either for the increase of the vibrations on the laser machine. On the other hand, a too low speed gives a longer interaction between laser beam and workpiece, increasing the recast layer thickness inside the hole.

Regarding circularity, ANOVA does not indicate as significant any main effect, while marks as significant the interactions between speed and pulse duration ($D*S$). The first phenomenon can be explained taking into account the high levels of spot overlap: for all combinations of the control factors, the overlap is always higher than 90%. For this reason, the double orbitation (TP2) of the laser beam does not lead to an improvement of the circularity. The single orbitation (TP1) already guarantees an excellent finish in the contouring.

The statistically significant interaction plots are reported in Fig. 5. Figure 5a shows that the increasing of the S can produce opposite behaviours: when low level of pulse duration ($D = 0.1$ ms) is adopted, the circularity decreases; for high value of pulse duration ($D = 0.4$ ms), the circularity increases. This because, at low value of speed ($S = 200$ mm/min), high duration generates high molten material, that is hard to remove and so a highly irregular hole with low circularity is obtained.

On the contrary, low pulse D generates a lower quantity of molten that can be easily expelled, allowing obtaining a more regular hole. When the speed increases ($S = 600$ mm/min), the opposite occurs: the increase of speed creates a benefit when the pulse duration is high, because less molten material is generated for the reductions of the interaction time, and so a more regular hole is obtained.

Figure 5b shows the interactions between $D*S$ for CEx. The statistical significance of this interaction is marked because of the strong decrease of CEx when the intermediate value of pulse duration ($D = 0.2$) is adopted. Such behaviour could be in line with the pulse on and pulse off mechanism:

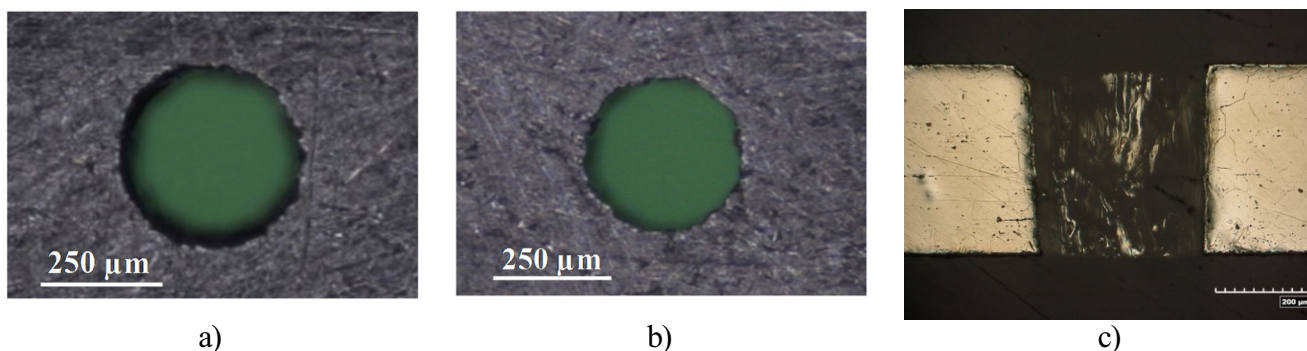


Fig. 6 Hole aspect at **a** entry, **b** exit, and **c** cross section obtained with the optimal parameter set

when the pulse D is set at 0.2 ms, the molten material is not so hot, and therefore, an increase of the speed can only worsen the contouring mechanism, giving a lower circularity in exit.

As a result of the information came out from the analysis of both geometric and metallurgical data, it was possible to identify an optimal parameter set for the microdrilling process. The best parameter set are detailed in Table 5. Through this parameter set, the supposed time to complete the process for the whole component is around 10 min instead of about 120 min for the correspondent process by EDM. Figure 6 shows a hole obtained with the optimal parameter set.

4 Conclusion

In this study, laser microdrilling of NIMONIC®263 sheet was performed by modulated fibre laser, to study how the process parameters affect the quality of the holes. On the basis of the discussed results, the main conclusions can be summarized as follow:

- fibre laser technology represents a valid alternative to EDM technology in the processing of the effusion cooling systems, ensuring a good quality and lower processing time;
- all the holes are free from cracks, neither in the base material nor in the recast layer;
- thanks to the use of an inert assistant gas, the samples show total absence of oxides;
- about 65% of the treatments results into the aeronautical specification (thickness of recast layer less than 0.040 mm);
- this work increases understanding of fibre laser trepanning and represents an excellent starting point for future developments.

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