RESEARCH PAPER

Yet another parameter‑free shape optimization method

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Abstract

The use of node coordinates as design variables in shape optimization offers a larger design space than computer-aided design (CAD)-based shape parameterizations. It also allows for the optimization of legacy designs, i.e., a fnite element mesh from an existing design can be readily optimized to meet new performance requirements without involving a CAD model. However, it is well known that the node coordinate parameterization method is fraught with numerical difficulties, which makes it impractical to use. This has led to several of "parameter-free" shape optimization methods that seek the advantages and avoid the pitfalls of the naïve node coordinate parameterization method. These methods come in two main varieties: sensitivity fltering (or gradient smoothing) and consistent fltering. The latter is analogous to the density flter method used in topology optimization (TO). Herein, we use the PDE flter from TO and energy-based flters to implement consistent shape optimization filtering schemes easily and efficiently. Numerical experiments demonstrate that consistent methods are more robust than sensitivity fltering methods.

Keywords Shape optimization · Node coordinate parameterization · Consistent fltering

1 Introduction

Shape optimization morphs material boundaries to achieve optimal design performance. When compared to topology optimization, shape optimization exhibits less design freedom; however, it provides an explicit boundary/interface description. Thus, when simulations involve boundary or interface-dominant physics, e.g., stress computations for optimization metrics or solid–fuid interaction, shape optimization is often advantageous. Further, shape optimization results are easily communicated to manufacturing processes due to the explicit boundary and interface representations.

Shape optimization poses two key challenges. First, a parameterization must be defned that allows design fexibility while maintaining adequate design control. Unfortunately, these two desires are often contradictory. Second, a

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 \boxtimes Kenneth E. Swartz swartz10@llnl.gov conforming mesh of each iterate design must be generated for use in the finite element simulation. The difficulty of mesh generation depends on the complexity of the parameterization. To ease this process, we combine the parameterization with the fnite element mesh. Namely, we use the mesh nodal coordinates as the optimization variables. Users only generate a mesh via their favorite preprocessor for the initial design. Thereafter, the optimizer controls the mesh updates. This "parameter-free" approach reduces design limitations imposed by restrictive parameterizations.

To motivate our work, we optimize the shape of a hole in a biaxially loaded plate to minimize the compliance subject to a maximum volume constraint Fig. $1¹$ $1¹$ In our so-called parameterized optimization, we require the hole to be an ellipse and optimize the lengths of the axes, Fig. [2.](#page-1-1) The compliance of the optimized design is 79% of the initial design's compliance. The analytical solution, assuming the plate is infnite, is an ellipse with 2:1 axes ratio, so our design with 1.81 axes ratio is nearly optimal. Nonetheless, to improve the design, we no longer restrict it to the 2 parameter elliptical hole optimization. Rather, an "independent node" optimization is performed in which *all* of the fnite element node coordinates (except those that would afect the plate's outer

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¹ The full details of this optimization problem are provided in Sect. [3](#page-9-0).

Fig. 1 Plate with hole (left), symmetry domain with biaxial loading boundary conditions (center) and design boundary conditions (right)

(a) Initial design (b) CAD param- (c) Independent eterization, opti- node parametermized ization, after 3 iterations

Fig. 2 Finite element meshes for optimized plate with hole

boundary) serve as design parameters. However, as seen in Fig. [2,](#page-1-1) after 3 iterations, the mesh gets tangled so the optimization is terminated. This is expected since the compliance and volume are insensitive to internal node movements and hence the optimizer only moves the boundary nodes. To resolve this mesh tangling defciency, various "parameterfree" shape optimization methods have been proposed; we propose yet another.

We aim to develop a parameter-free shape optimization method that uses readily available nonlinear programming (NLP) solvers, e.g., IPOPT (Wächter and Biegler [2006\)](#page-26-0) and fmincon (Matlab [2017\)](#page-26-1). In this way, we can readily accommodate an arbitrary number of constraints, verify the Karush-Kuhn-Tucker (KKT) optimality conditions, and avoid writing-specialized NLP solvers. We seek a robust method that does not require extensive parameter tuning when solving a range of optimization problems. Finally, we aim to develop an efficient method that converges rapidly and utilizes high-performance computing (HPC).

To achieve our parameter-free shape optimization goals, we mimic the implicit PDE density filter developed by Lazarov and Sigmund [\(2011\)](#page-26-2) from topology optimization. The use of a consistent sensitivity analysis, i.e., we flter nodal coordinates and their sensitivities, provides robust convergence during the design optimization.

We aim to optimize the shape of three-dimensional continuum structures using the fnite element method, and hence, we do not review literature that concerns optimization of plate and shell structures and isogeometric analysis.^{[2](#page-1-2)} Notwithstanding this omission, the shape optimization literature is vast, please refer to Samareh ([2001](#page-26-3)); Martinelli and Jameson [\(2012\)](#page-26-4); Upadhyay et al. ([2021](#page-26-5)) for thorough reviews of the subject. Our discussion focuses on parameter-free methods. The use of node coordinates as design variables in shape optimization has the advantage of ofering a large design space and eliminating the reliance on computer-aided design (CAD)-based shape parameterizations (Braibant and Fleury [1984](#page-25-0); Chen and Tortorelli [1997](#page-25-1)) that are troublesome for industrially relevant optimization problems. This is because industrially relevant CAD models are created using numerous boolean, blending, rounding, and trimming operations. In the optimization, fnite element meshes must be created from the CAD models and derivatives of the node coordinates with respect to the optimization parameters, e.g., select dimensions must be computed. While this task is easily accomplished for the simple optimization problem illustrated in Figs. [1](#page-1-0) and [2](#page-1-1), it is often extremely challenging for industrially relevant problems. CAD models associated with legacy designs may also be unavailable. However, if fnite element models for such designs are available, they can be shape optimized via the parameter-free method to meet new design specifcations. That said, the node coordinate parameterizations may lead to an ill-posed optimization problems (Chenais [1975;](#page-25-2) Bängtsson et al. [2003](#page-25-3); Mohammadi and Pironneau [2004](#page-26-6); Harbrecht [2008;](#page-26-7) Mohammadi and Pironneau [2008](#page-26-8)), but most defnitely, it is well known that naively using the node coordinates as design variables produces designs with tangled meshes which are artifacts of

² The reason being that plate/shell studies and isogeometric-based formulations are less susceptible to the mesh distortion problems that plague the shape optimization of continuum bodies modeled with the fnite element (or similar) method.

the fnite element discretization (Braibant and Fleury [1984](#page-25-0); Le et al. [2011](#page-26-9)), cf. the simple example above.

We compute the derivatives of the optimization cost and constraint functions using the adjoint method. The derivatives can be expressed as a combination of volumetric and surface integrals in the so-called domain method or solely with surface integrals in the so-called boundary method. Sensitivities computed via the "domain" method agree, within machine precision, to those computed via fnite difference approximations (Choi and Seong [1986](#page-25-4); Yang and Botkin [1987\)](#page-26-10) because the diferentiate-then-discretize and discretize-then-differentiate operations commute (Yang and Botkin [1987;](#page-26-10) Berggren [2010\)](#page-25-5). However, sensitivities computed via the boundary method are error prone due to a lack of continuity in the gradient computations across the inter element boundaries (Berggren [2010](#page-25-5)). Fortunately, these errors diminish with mesh refnement and may also be reduced by using nonconventional C^1 interpolation.

Gradient smoothing is analogous to sensitivity fltering in topology optimization (Sigmund [1997](#page-26-11)) where it is well known that the fltered sensitivities are inconsistent with the NLP solver and hence the NLP convergence sufers (Matsumori et al. [2017\)](#page-26-12). Two types of shape optimization gradient smoothing methods exist, explicit and implicit. The explicit approach uses a convolution flter, much like how Sigmund [\(1997\)](#page-26-11) computes a "smooth" gradient from the "raw" gradient, e.g. (Stück and Rung [2011;](#page-26-13) Gerzen et al. [2012\)](#page-25-6). In the implicit approach, a careful distinction is made between the derivative and the gradient, e.g., Pironneau ([1984](#page-26-14)). We generally compute the derivative via adjoint methods as described below. But, the gradient, not the derivative, is often required by popular NLP algorithms, e.g., the method of steepest descent. The gradient belongs to an inner product space, and hence, it depends on the defnition of the inner product. This is the impetus of the $H¹$ smoothing methods (Pironneau [1984](#page-26-14); Jameson [2004](#page-26-15); Gournay [2006](#page-25-7); Nonogawa et al. [2020](#page-26-16); Azegami [2021;](#page-25-8) Allaire et al. [2021\)](#page-25-9) which roughly project the derivative onto the chosen inner product space. The projected gradient requires the solution of a PDE and hence the "implicit" flter terminology.

The traction method presented by Shimoda et al. ([1997\)](#page-26-17) can be viewed as a gradient smoothing method in which the *H*¹ inner product is replaced with the energy inner product from linear isotropic elasticity. The method applies the shape derivative as a traction to compute the smoothed gradient and hence its name. In another variant, Azegami and Takeuchi ([2006](#page-25-10)) treat the shape derivative as an "ambient" displacement in a Robin-like boundary condition to obtain additional smoothness of the gradient. This latter approach is more closely related to the H^1 smoothing methods. Riehl and Steinmann ([2014](#page-26-18)) extend the traction method by incorporating an adaptive mesh refnement step to ensure suitable meshes and accurate simulations.

As noted by Schwedes et al. [\(2017](#page-26-19)), the inconsistency introduced by only fltering the sensitivity in the implicit methods can be resolved by incorporating the chosen inner product into the NLP solver. Unfortunately, this is seldom the case as most NLP libraries work in ℝ*ⁿ* rather than the underlying function space in which the optimization problem is naturally formulated, e.g., *H*¹ . Perhaps this is why most of the shape optimization methods cited above do not use popular NLP libraries; rather, they rely on augmented Lagrangian methods that incorporate a single constraint combined with inefficient steepest descent methods (Gournay [2006;](#page-25-7) Nonogawa et al. [2020](#page-26-16); Ertl et al. [2019\)](#page-25-11). Nonetheless, gradient smoothing algorithms prove efective as seen below and demonstrated by Ertl et al. [\(2019\)](#page-25-11).

Riehl et al. (2014) (2014) and Bletzinger (2014) (2014) (2014) provide efforts to maintain consistency in the optimization formulation. Therein explicit flters are used to compute the gradient from the derivative and its transpose is used to smooth the design update. This is akin to the density flter (Bruns and Tortorelli [2001](#page-25-13)) from topology optimization where the element volume fraction vector \mathbf{v} is optimized, but the filtered element volume fraction vector $\tilde{v} = \mathbf{F} v$ obtained from the filter matrix \vec{F} is used to define the design over which the response is computed. Sensitivities $\partial\theta/\partial\tilde{v}$ of the cost and constraint functions $\widetilde{\theta}(\widetilde{v})$ are computed via adjoint methods and the chain-rule is used to evaluate the sensitivities that are input to the NLP algorithm $\partial \theta / \partial \mathbf{v} = \mathbf{F}^T \partial \theta / \partial \tilde{\mathbf{v}}$. Here, the smoothed and unsmoothed element volume fraction vectors \tilde{v} and v are replaced by the smoothed and unsmoothed nodal coordinate vectors \tilde{X} and X ; however, an inconsistency arises that does not appear in topology optimization. Namely, in the work by Riehl et al. [\(2014\)](#page-26-20) and Bletzinger ([2014](#page-25-12)), the mesh over which \vec{F} is computed is defined by the smoothed node coordinate vector \tilde{X} , which continues to evolve during the optimization and consequently *F* changes during the optimization. This observation has led to the development of adaptive flter schemes, Antonau et al. $(2022)^3$ $(2022)^3$ $(2022)^3$ $(2022)^3$

In the end, it is difficult to know precisely what optimization problem is solved. Notably, the flter is recomputed at each optimization iteration using the confguration corresponding to the previous design iterate and hence the optimized design depends on the "trajectory" of the NLP algorithm. One cannot obtain KKT conditions to a predefned optimization problem, without taking this "trajectory" into account.

In the natural design variable method, Belegundu and Rajan [\(1988\)](#page-25-15) apply "design loads" P_d to the initial design and use separate "shape morphing" analyses to deform the

³ This publication also provides an excellent overview of the socalled vertex morphing methods.

initial design configuration Ω_0 into the optimized configuration Ω. The load P_d serves as the design variable vector in the optimization, Weeber et al. [\(1992\)](#page-26-21); Tortorelli ([1993](#page-26-22)). The initial design with node coordinate vector X_0 is morphed into the optimized structure with node coordinate vector $X = X_0 + D$. Here, *D* is the solution to the discretized shape morphing elasticity equation $K_d D = P_d$. Mimicking the sensitivity filtering discussion above, we have $D = K_d^{-1} P_d$ and $\partial \theta / \partial P = K_d^{-T} \partial \tilde{\theta} / \partial X$, i.e., K_d^{-1} plays the role of the filter matrix *F*. This consistent method has a well-defned optimization problem that quickly converges.

The "fictitious energy method" is similar the natural design variable method; however, Scherer et al. [\(2010\)](#page-26-23) use prescribed displacements rather than loads to morph the initial design into the optimized design. Mesh quality is ensured by using a "special" nonlinear, isotropic, hyperelastic material that is unafected by rigid motions and dilations/ contractions in the morphing analysis. In this way, rigid motions and dilations/contractions of elements are allowed whereas shear and unequal triaxial extensions/compressions of the elements are minimized. To further limit mesh distortion, an additional constraint on the total strain energy of the morphing deformation is imposed.

The explicit consistent fltering approach to shape opti-mization (Le et al. [2011\)](#page-26-9) can be viewed as the shape optimization equivalent of the popular density-based topology optimization method (Bruns and Tortorelli [2001\)](#page-25-13). The latter uses a consistent flter to formulate as well-posed optimization problem and hasten convergence of the NLP algorithm. Herein, we build upon the work of Le et al. [\(2011\)](#page-26-9) by replacing the explicit flter with various implicit flters, another idea borrowed from topology optimization (Lazarov and Sigmund [2011\)](#page-26-2) and used by Matsumori et al. ([2017\)](#page-26-12) in their shape optimization study. These implicit flters are more suitable to HPC and, in our opinion, easier to implement, although they require the solutions of additional PDEs. Indeed, the solvers are implemented using efficient numerical techniques leveraging scalable parallel computer architectures. On the other hand, the explicit flters require interprocessor communication to evaluate the flter convolution integrals and HPC libraries that support such computations are not generally available. A comparison of consistent explicit and implicit flters is presented by Najian and Bletzinger ([2023\)](#page-26-24).

Our approach is very similar to the $H¹$ filter approach described by Matsumori et al. ([2017](#page-26-12)); however, we do not constrain a mesh quality energy or limit the extent of the flter to the boundary region. Similarities also exist between our approach and that presented by Liatsikouras et al. ([2022](#page-26-25)).

Therein the coordinates of duplicate "handle nodes" on the boundary serve as the design parameters. The initial mesh is morphed into the optimized design by minimizing

the morphing strain energy subject to the constraint that the "follower" surface nodes are close to the handle nodes. Because the constraint is enforced via the penalty method, the design smoothness can be controlled by varying the penalty value. To a lesser extent, our approach is simi-lar to the fictitious energy approach (Scherer et al. [2010\)](#page-26-23); however, we replace their morphing displacement boundary conditions with Robin boundary conditions so we are able to flter all of the node coordinate updates, i.e., both in the domain's interior and on its boundary and we do not constrain the fctitious energy. Tikhonov regularization is used by Bängtsson et al. ([2003](#page-25-3)) to obtain a smooth boundary; the idea is motivated by Gunzburger et al. ([2000](#page-25-16)). Their approach requires the solution of an ODE (for their two-dimensional study) to map the optimization's jagged boundary to a smooth boundary over which the optimization cost and constraint functions are computed. Interior nodes are relocated by solving an elasticity problem as done in the fctitious energy approach (Scherer et al. [2010](#page-26-23)); however, their problem is linear so the mesh motion is more limited. Gradients of the cost and constraint functions are computed in a consistent manner using a combination of direct and adjoint methods. Their thorough study compares many diferent problem formulations.

We experiment with diferent flters based on the vector Helmholtz PDE used by Matsumori et al. ([2017](#page-26-12)), a nonlinear flter based on the fctitious energy approach of Scherer et al. ([2010](#page-26-23)), and a linearized version of this nonlinear flter. To solve the optimization problems, we use popular NLP libraries IPOPT (Wächter and Biegler 2006) and fmincon (Matlab 2017) to leverage their efficient algorithms and their abilities to incorporate numerous constraints and compute KKT optimality conditions. Some of our computations utilize the MFEM library (Anderson et al. [2021\)](#page-25-17), and hence, we can readily invoke parallelism and employ a wide variety of discretizations, e.g., high-order polynomials. Derivatives for the algorithms are computed consistently via adjoint methods and accurately using domain, rather than boundary, sensitivity integrations. Summarizing the salient features in our work, we highlight

- 1. Domain shape derivative computations,
- 2. Consistency with the NLP algorithm,
- 3. PDE and energy-based flters,
- 4. Multiple flter functions,
- 5. Use of popular NLP libraries.

We compare our consistent approaches to the fictitious energy approach of Scherer et al. ([2010\)](#page-26-23) and the gradient smoothing approaches of Allaire et al. [\(2021](#page-25-9)) and Azegami ([2021](#page-25-8)). These three methods are also integrated with the

popular NLP libraries and utilize the accurate domain shape derivative computations.

2 Independent node coordinate formulation

In the independent node coordinate formulation, the node coordinates contained in *X* that defne the *physical* domain Ω serve as the design variable. A linear elastic finite element analysis is performed over Ω to compute the *physical* displacement feld *u*. The cost and constraint functions θ_i , *i* = 0, 1, 2, … and their shape sensitivities $D\theta_i / DX$ are then evaluated. The values of the coordinates X , cost and constraint functions θ_i , and sensitivities $D\theta_i / DX$ are input to the NLP algorithm to update the coordinates $X \leftarrow X + \Delta X$. Iterations continue until convergence, or more likely until the appearance of a tangled mesh which causes the algorithm to crash.

In the fnite element analysis, we fnd the physical displacement $u \in H(u^p)$ that satisfies the linear elastic residual equation

$$
0 = r(u, w)
$$

= $\int_{\Omega} \nabla w \cdot \mathbb{C}[\nabla u] dv - \int_{\Omega} w \cdot b dv$
- $\int_{A^t} w \cdot t^n da - \int_{A^p} w \cdot p \, n da$
= $a(u, w) - \ell(w)$ (1)

for all $w \in H(0)$. In the above $\mathbb C$ is the elasticity tensor which exhibits the usual symmetries, *b* is the prescribed body force per unit volume. $\partial\Omega$ is the boundary of Ω with outward unit normal vector n ; it is divided into three complementary regions A^u , A^t , and A^p over which the displacement u^p , traction t^p , and normal pressure p are prescribed. Finally, a is the bilinear form, ℓ is the load linear form, and the space of admissible functions is

$$
\mathcal{H}(u^p) = \{ u \in H^1 : u = u^p \,\forall \, x \in A^u \}. \tag{2}
$$

Without loss of generality, we assume $u^p = 0$ henceforth.

Upon evaluating the displacement feld, we compute the optimization cost and constraint functions θ_i

$$
\theta_i(\mathbf{x}) = \int_{\Omega} \Lambda(\nabla \mathbf{u}, \mathbf{u}, \mathbf{x}) \, d\mathbf{v} + \int_{A'} \tau(\mathbf{u}, \mathbf{x}) \, da + \int_{A'} \tau(\mathbf{u}, \mathbf{x}) \cdot \mathbf{n} \, da
$$
 (3)

where x denotes the location of material particles in Ω and Λ, *𝜏* , and *𝝉* are generic functions that can be used to compute, e.g., the volume of Ω by equating $Λ = 1, τ = 0$, and $\tau = 0$ or compliance by equating $\Lambda = \nabla u \cdot \mathbb{C}[\nabla u], \tau = 0$, Require: X Ensure: Θ_i , $D\Theta_i/DX$ Solve primal problem $KU = P$ for U for $i \in \{0, 1, 2, \dots\}$ do Compute Θ_i and the derivatives $\partial \Theta_i / \partial \mathbf{X}$ and $\partial \Theta_i / \partial \mathbf{U}$, cf. (34) and (35) Solve adjoint problem $\boldsymbol{K}\boldsymbol{W} = \left(\partial \Theta_i / \partial \boldsymbol{U}\right)^T$ for \boldsymbol{W} Compute pseudo load $\partial R/\partial X$, cf. (36) Compute sensitivity $D\Theta_i/D\bm{X}=\partial\Theta_i/\partial\bm{X}-\partial R/\partial\bm{X}$ end for return Θ_i , $D\Theta_i/DX$

Algorithm 1 Independent node coordinate iteration,Appendix 1 for the discretization

and $\tau = 0$. The shape sensitivity of the above is evaluated using the adjoint method (Haug and Choi [1986](#page-26-26)). Details of the sensitivity analysis and fnite element discretization are provided in Appendix.

In practice, many of the node coordinates are not moved during the optimization. To illustrate this, we optimize the shape of a hole in the center of a homogeneous isotropic 2×2 square plate that is subjected to biaxial traction, Fig. [1.](#page-1-0) To facilitate the plane stress analysis, we utilize domain symmetry whence the optimized hole geometry is necessarily symmetric with respect to reflections about the e_1 and e_2 axes. Since we are only optimizing the hole shape, the node coordinates are constrained such that the nodes on the vertical edges cannot move in the e_1 direction and the nodes on the horizontal edges cannot move in the e_2 direction.

One iteration for the independent node coordinate optimization algorithm is partially summarized in Algorithm 1. For each iteration of the optimization, we have a design Ω defned by the node coordinates *X* over which we compute the θ_i and their derivatives $D\theta_i / DX$. The optimizer updates the design, i.e., $X \leftarrow X + \Delta X$, whereafter another iteration commences if the convergence criterion has not been satisfed. Unfortunately, this independent node coordinate formulation invariably produces tangled meshes which cause the optimization to prematurely crash, Fig. [2.](#page-1-1)

2.1 Gradient smoothing algorithm

The gradient smoothing formulation is similar to the independent node coordinate formulation as we are given a body with material points identifed by their positions *x* in the initial design configuration Ω₀ and x ← $x + δx$ in the optimized design configuration $Ω$. The mesh tangling incurred by the independent node coordinate formulation is resolved using numerous variants of explicit (Stück and Rung [2011\)](#page-26-13) or implicit (Pironneau [1984;](#page-26-14) Jameson [2004;](#page-26-15) Gournay [2006](#page-25-7); Scherer et al. [2010;](#page-26-23) Allaire et al. [2021](#page-25-9); Müller et al. [2021](#page-26-27); Welker [2021](#page-26-28)) gradient smoothing algorithms. Mimicking the sensitivity flter in topology optimization (Sigmund [1997\)](#page-26-11), the explicit method computes a smooth shape sensitivity by fltering the raw gradient, i.e., shape sensitivity. Implicit smoothing uses the Riesz representation theorem to relate the derivative $\delta\theta$ to the gradient grad θ . In \mathbb{R}^n , under the standard Euclidean inner product, there is no diference in the representation of the derivative and the gradient. Thus, when the design, i.e., the node coordinate vector *X*, is viewed as an element of ℝ*ⁿ*, the design derivative and gradient are (numerically) interchangeable. However, this is no longer the case when one views the design as an element of a function space H , which is the case if we use a perturbation $\Delta x \in \mathcal{H}$ to deform the current design into the next design iterate, i.e., points originally located at *x* in the current design are displaced to locations $x = x + \Delta x$ in the next design iterate. Assuming the inner product on H is not the Euclidean inner product, the shape derivative is no longer (numerically) interchangeable with the gradient. Instead, assuming it exists, the derivative at Ω , $\delta\theta(\mathbf{x};\cdot) \in \text{Lin}(\mathcal{H} \to \mathbb{R})$, is a linear operator from \mathcal{H} into \mathbb{R} and hence by the Riesz representation theorem there exists a unique grad $\theta(x) \in \mathcal{H}$ such that $\delta\theta(x, \delta x) = a(\text{grad}\theta(x), \delta x)$ for all $\delta x \in H$. In this equality $a(\cdot, \cdot)$ is the inner product for H and this is where things get murky as there are multiple choices for the space H and inner product $a(\cdot, \cdot)$. For example if we choose $\mathcal{H} = L^2$ and the standard inner product

$$
a(\mathbf{u}, \mathbf{w}) = \int_{\Omega} \mathbf{u} \cdot \mathbf{w} \, d\mathbf{v} \tag{4}
$$

we obtain the "rough" gradient grad $\theta(x)$ via

$$
a(\text{grad}\theta(\mathbf{x}), \delta \mathbf{x}) = \int_{\Omega} \text{grad}\theta(\mathbf{x}) \cdot \delta \mathbf{x} \, d\mathbf{v} = \delta \theta(\mathbf{x}; \delta \mathbf{x}) \tag{5}
$$

^{[4](#page-5-0)} However, if we choose $\mathcal{H} = H^1$ and the nonstandard inner product

$$
\widetilde{a}(\mathbf{u}, \mathbf{w}) = \int_{\Omega} (\nabla \mathbf{u} \cdot \gamma \nabla \mathbf{w} + \mathbf{u} \cdot \mathbf{w}) dv
$$
\n(6)

with $\gamma > 0$, we obtain the "smooth" gradient grad $\theta(x)$ via

$$
\widetilde{a}(\widetilde{\text{grad}}\theta(\mathbf{x}), \delta \mathbf{x}) = \int_{\Omega} \left(\nabla \widetilde{\text{grad}}\theta(\mathbf{x}) \cdot \gamma \nabla \delta \mathbf{x} + \widetilde{\text{grad}}\theta(\mathbf{x}) \cdot \delta \mathbf{x} \right) d\mathbf{v}
$$

$$
= \delta \theta(\mathbf{x}; \delta \mathbf{x})
$$
(7)

In the above, we emphasize that $\text{grad} \theta(x)$ is a vector field in H^1 and $\nabla \widetilde{\text{grad}}\theta(x)$ is the spatial derivative of this field.

To see why we refer to grad $\theta(x)$ and grad $\theta(x)$ as the rough and smooth gradients, consider the following minimization problem

$$
\min_{\text{grad}\theta \in H^1} \qquad \int_{\Omega} |\widehat{\text{grad}\theta} - \text{grad}\theta|^2 \, dv
$$
\n
$$
\text{such that} \qquad \int_{\Omega_0} |\nabla \widehat{\text{grad}\theta}|^2 \, dv = 0 \tag{8}
$$

where grad $\theta \in L^2$ is the known rough gradient and we have dropped the arguments for conciseness. Solving the above via the penalty method results in the unconstrained problem

$$
\min_{\text{grad}\theta(x)\in H^1} \int_{\Omega} |\widehat{\text{grad}\theta}(x) - \text{grad}\theta(x)|^2 \, dv +
$$
\n
$$
\gamma \int_{\Omega_0} |\nabla \widehat{\text{grad}\theta}(x)|^2 \, dv \tag{9}
$$

where $\gamma > 0$ is the penalty parameter. The stationary condi-tion of ([9\)](#page-5-1) requires $\widetilde{\text{grad}}\theta(x) \in H^1$ to satisfy

$$
\widetilde{a}(\widetilde{\text{grad}}\theta(x), w) = a(\text{grad}\theta(x), w) = \delta\theta(x; w)
$$
\n(10)

for all $w \in H^1$. Larger values of γ increase the enforcement of the $\int_{\Omega_0} |\nabla \overrightarrow{\text{grad}} \theta(\mathbf{x})|^2 d\nu = 0$ constraint, i.e., lessen the spatial variations in $\widetilde{\text{grad}}(\mathbf{x})$. An excellent tutorial on the need for using grad $\theta(x)$ rather than grad $\theta(x)$ appears in Jameson ([2004\)](#page-26-15).

The H^1 smoothing algorithm uses the smooth gradient $\text{grad}\theta(\mathbf{x})$ to compute the design updates $\delta \mathbf{x}$. Smoother gradients result in smoother updates and hence less mesh tangling. Setting $\gamma = 1$ we see that \tilde{a} reduces to the standard inner product for H^1 and hence the H^1 terminology.

We finish the derivation with the imposition of the "boundary conditions." To fx the location of portions of the boundary $\partial\Omega$ during the shape optimization, we replace H^1 in ([10\)](#page-5-2) with

$$
H_0^1 = \{ u \in H^1 | u \cdot e = 0 \text{ on } A^f \}
$$
 (11)

in which *A^f* ⊂ $\partial Ω$ ₀ are the surface regions which are fixed in the *e* direction during the optimization. E.g. in the Fig. [1](#page-1-0) example $\mathbf{u} \cdot \mathbf{e}_2 = 0$ on the horizontal edges and $\mathbf{u} \cdot \mathbf{e}_1 = 0$ on the vertical edges. Telling the nonlinear programming

⁴ We emphasize above that both grad $\theta(x) \in \mathcal{H}$ and $\delta x \in \mathcal{H}$ are functions of position, i.e., $a(\text{grad}\theta(\mathbf{x}), \delta \mathbf{x}) = \int_{\Omega} \text{grad}\theta(\mathbf{x})(\mathbf{y}) \cdot \delta \mathbf{x}(\mathbf{y}) d\mathbf{v}_{y} = \delta \theta(\mathbf{x}; \delta \mathbf{x}).$

Require: X Ensure: Θ_i , Grad Θ_i Solve primal problem $KU = P$ for U for $i \in \{0, 1, 2, \dots\}$ do Compute Θ_i and the derivatives $\partial \Theta_i / \partial \mathbf{X}$ and $\partial \Theta_i / \partial \mathbf{U}$, cf. (34) and (35) Solve adjoint problem $\boldsymbol{K}\boldsymbol{W} = \left(\partial \Theta_i / \partial \boldsymbol{U}\right)^T$ for \boldsymbol{W} Compute pseudo load $\partial R/\partial X$, cf. (36) Compute sensitivity $D\Theta_i/DX = \partial \Theta_i/\partial X - \partial R/\partial X$ Solve the smoothed gradient problem $(\widetilde{\mathbf{K}} + \widetilde{\mathbf{M}})$ Grad $\Theta_i = (D\Theta_i/D\mathbf{X})^T$ for Grad Θ_i end for return Θ_i , Grad Θ_i

Algorithm 2 H^1 smoothing algorithm iteration, Appendix 2 for the discretization

algorithm that $\widetilde{\text{grad}}\theta(\mathbf{x}) \cdot \mathbf{e} = 0$ on A^f ensures that these surfaces remain unchanged in these directions during the optimization.

The H^1 smoothing algorithm is analogous to the PDE filter in topology optimization (Lazarov and Sigmund [2011](#page-26-2)). However, it is not performed in a consistent manner, and hence, it is also analogous to the sensitivity fltering approach in topology optimization (Sigmund 1997).^{[5](#page-6-0)} As such, the optimization algorithm may converge slowly or not at all.

We note that the traction method (Shimoda et al. [1997\)](#page-26-17) is a variant of the H^1 smoothing algorithm in which the elasticity bilinear form *a* of ([1\)](#page-4-0) replaces \tilde{a} of [\(6](#page-5-3)), i.e., we use the energy inner product. As such, "infnitesimal rigid motions" characterized by skew $\nabla \text{grad} \theta(x)$ are not penalized, as these do not adversely afect mesh quality. Multiple variants of the traction method are described by Azegami ([2021\)](#page-25-8).

The H^1 smoothing algorithm is only a slight modification to the independent node coordinate formulation. For each iteration of the optimization, we are given the node coordinates *X* whereupon we complete Algorithm 2 to compute the Θ_i and their H^1 smoothed gradients $\widetilde{\text{Grad}} \Theta_i$. The optimizer processes these quantities and either updates the design, i.e., $X \leftarrow X + \Delta X$ or declares convergence.

2.2 Fictitious energy formulation

In the fictitious energy formulation of Scherer et al. (2010) (2010) , an initial design Ω_0 is deformed into the optimized design Ω such that points located at x_0 in the initial design Ω_0 are displaced to the locations $\mathbf{x} = \mathbf{x}_0 + \mathbf{d}$ in the optimized design. As such, the design displacement feld *d* serves as the control in the optimization. To be clear, *d* is only optimized on the boundary $\partial \Omega_0 \backslash A^f$ where A^f is defined in the discussion surrounding [\(11\)](#page-5-4). For the Fig. [1](#page-1-0) example, $d^p = d \cdot e_1$ and $d^p = d \cdot e_2$ are optimized over the circular arc, $d^p = d \cdot e_1$ is optimized over the horizontal edges and $d^p = d \cdot e_2$ is optimized over the vertical edges. Over the remainder of the boundary A^f , $\mathbf{d} \cdot \mathbf{e} = 0$. To obtain the displacement \mathbf{d} on the interior Ω_0 , a "fictitious" nonlinear elasticity analysis is performed. The analysis assumes an isotropic hyperelastic material that is unafected by rigid deformations and dilations/contractions.

The morphing analysis fnds the *d* that solves the minimization problem

$$
\min_{d \in \mathcal{H}_f(d^p)} \int_{\Omega_0} \Psi_f(\boldsymbol{F}_d) \, d\boldsymbol{\nu} \tag{12}
$$

where Ψ_f is the fictitious energy density defined in Appendix [3](#page-22-0), $\mathbf{F}_d = \mathbf{I} + \nabla \mathbf{d}$ is the fictitious deformation gradient,

$$
\mathcal{H}_f(d^p) = \{ \mathbf{d} \in H^1 | \mathbf{d} \cdot \mathbf{e} = 0 \text{ on } A^f \text{ and}
$$

$$
\mathbf{d} \cdot \mathbf{e} = d^p \text{ on } \partial \Omega_0 \setminus A^f \}
$$
(13)

is the space of admissible fictitious displacements, and d^p is the prescribed displacement that is optimized. Stationarity of [\(12](#page-6-1)) requires $d \in H_f(d^p)$ to satisfy the residual equation

$$
r_f(\mathbf{d}, \mathbf{w}) = \int_{\Omega_0} \nabla \mathbf{w} \cdot \mathbf{P}_f \, d\mathbf{v} = 0 \tag{14}
$$

for all $w \in H_f(0)$ where $P_f = \frac{\partial \Psi_f}{\partial F_d}$ is the fictitious First Piola-Kirchhoff stress. In the finite element method, we express $\mathbf{d} = \mathbf{d}^0 + \mathbf{d}^p$ where $\mathbf{d}^0 \in \mathcal{H}_f(0)$ and $\mathbf{d}^p \in \mathcal{H}_f(d^p)$ is a known extension function that satisfes the prescribed displacement boundary condition. In this way, we solve ([12\)](#page-6-1) by finding the $d^0 \in H_f(0)$ that satisfies

$$
r_f(\boldsymbol{d}^0, \boldsymbol{w}, \boldsymbol{d}^p) = \int_{\Omega_0} \nabla \boldsymbol{w} \cdot \frac{\partial \Psi_f}{\partial \boldsymbol{F}_d} \, d\boldsymbol{v} = 0 \tag{15}
$$

for all $w \in H_f(0)$. We then trivially compute $d = d^0 + d^p$. The solution of the above residual equation for d^0 is obtained via Newton's method, Appendix [3](#page-22-0) for details.

⁵ The lack of consistency in ℝ^{*n*} means $D\theta / DX \neq \widetilde{\text{grad}\theta}$ which "confuses" optimization algorithms developed in ℝ*ⁿ* .

As an added measure to prevent severe mesh distortion during the shape optimization, the fictitious energy approach places a constraint on the total fictitious energy, i.e.,

$$
\theta_f(d^p) = \int_{\Omega_0} \Psi_f(\mathbf{F}_d) \, dv - \overline{\Psi}_f \le 0 \tag{16}
$$

where $\overline{\Psi}_f$ is the maximum permissible total fictitious energy. The above is enforced in addition to the $\theta_i(d^p) \leq 0$ and $i = 1, 2, \dots$ constraints.

The fictitious energy approach is a consistent approach; thereby, it exhibits faster convergence verses the H^1 smoothing formulation. However, the consistency comes with a price with regard to the sensitivity analysis as two adjoint analyses are now required to evaluate the sensitivity of each cost and constraint function with respect the boundary displacement d^p , Appendix [3](#page-22-0) for details.

In each iteration of the fictitious energy formulation, we are given the initial node coordinates X_0 and the node design displacement vector D^p whereupon we complete Algorithm 3 to compute the Θ*ⁱ* and their derivatives $D\Theta_i / DD^p$, $i = f, 0, 1, 2, \dots$. The optimizer subsequently uses these quantities to either update the design, i.e., $D^p \leftarrow D^p + \Delta D^p$ or declare the convergence.

2.3 Consistent fltering formulation

The consistent filtering formulation is similar to the fictitious energy formulations as we again deform an initial design Ω_0 into the optimized design Ω and the design displacement feld *d* controls the design. Here, however, points at locations x_0 in the initial design Ω_0 are displaced to locations $x = x_0 + d$ in the optimized design Ω , where *d* is the *filtered* version of *d*. The method is analogous to PDE sensitivity fltering in topology optimization (Lazarov and Sigmund [2011\)](#page-26-2). As such, it eliminates the inconsistency of the *H*¹ smoothing formulation and flters the boundary updates which the fictitious energy formulation does not.

To obtain the fltered design in our "PDE flter" variant, we find $\tilde{d} \in \mathcal{H}_c$ such that

$$
0 = r_c(\tilde{d}, w, d)
$$

=
$$
\int_{\Omega_0} (\nabla w \cdot \gamma \nabla \tilde{d} + w \cdot \tilde{d}) dv
$$

-
$$
\int_{\Omega_0} w \cdot d dv
$$

=
$$
a_c(\tilde{d}, w) - \ell_c(w, d)
$$
 (17)

for all $w \in \mathcal{H}_c$ where

Require: X_0 , D^p Ensure: Θ_i , $D\Theta_i/DD^p$ while $|R_{\ell}^{0}| > \epsilon$ Solve mesh morphing problem do Solve Newton update problem $\mathbf{K}_{f}^{00} \Delta \mathbf{D}_{0}^{0} = -\mathbf{R}_{f}^{0}$ for $\Delta \mathbf{D}_{0}^{0}$ on the Ω_{0} mesh Update displacement $\boldsymbol{D}^0 \leftarrow \boldsymbol{D}^0 + \Delta \boldsymbol{D}^0$ Compute \mathbf{R}_{f}^{0} on the Ω_{0} mesh end while Compute Θ_f on the Ω_0 mesh Compute sensitivity $D\Theta_f/DD^p = \mathbb{R}^p$ Define the Ω mesh with node coordinates $\boldsymbol{X} = \boldsymbol{X}_0 + \boldsymbol{D}$ Solve primal problem $\boldsymbol{K}\boldsymbol{U} = \boldsymbol{P}$ for \boldsymbol{U} on the Ω mesh for $i \in \{0, 1, 2, \dots\}$ do Compute Θ_i and the derivatives $\partial \Theta_i / \partial X$ and $\partial \Theta_i / \partial U$ on the Ω mesh, cf. (34) and (35) Solve adjoint problem $\boldsymbol{K}\boldsymbol{W} = \left(\partial \Theta_i / \partial \boldsymbol{U}\right)^T$ for \boldsymbol{W} on the Ω mesh Compute pseudo load $\partial R/\partial X$ on the Ω mesh, cf. (36) Compute sensitivity $D\Theta_i/DX = \partial \Theta_i/\partial X - \partial R/\partial X$ Solve adjoint problem $\mathbf{K}_{\rm f}^{00} \mathbf{W}^0 = (D\Theta_i/D\mathbf{X}^0)^T$ for \mathbf{W}^0 Compute sensitivity $D\Theta_i/DD^p = D\Theta_i/DX^p - \boldsymbol{W^0}^T\ \boldsymbol{K}^{0p}_f$ end for return Θ_i , $D\Theta_i/DD^p$, $i = f, 0, 1, 2, \cdots$

Algorithm 3 Fictitious energy formulation, Appendix 3 for the discretization

$$
\mathcal{H}_c = \{ \widetilde{\boldsymbol{d}} \in H^1 | \widetilde{\boldsymbol{d}} \cdot \boldsymbol{e} = \boldsymbol{0} \text{ on } A^f \} \tag{18}
$$

$$
(11).
$$

Similar to the H^1 smoothing discussion, to obtain physical insight into the flter, we consider the minimization problem

$$
\min_{\widetilde{d}\in\mathcal{H}_c} \int_{\Omega_0} |\widetilde{d} - d|^2 dv
$$
\nsuch that

\n
$$
\int_{\Omega_0} |\nabla \widetilde{d}|^2 dv = 0
$$
\n(19)

and solve it via the penalty method, i.e., we solve

$$
\min_{\widetilde{d}\in\mathcal{H}_c} \int_{\Omega_0} |\widetilde{d} - d|^2 \, dv
$$
\n
$$
+ \gamma \int_{\Omega_0} |\nabla \widetilde{d}|^2 \, dv \tag{20}
$$

. The stationarity condition of [\(20\)](#page-8-0) requires $\widetilde{d} \in \mathcal{H}_c$ to satisfy [\(17\)](#page-7-0) for all $w \in H_c$. In this way, \tilde{d} is "close" to *d*, but $|\nabla \tilde{d}|$ will be "small," i.e., \tilde{d} is the smoothed version of d . The smoothness of \tilde{d} increases with increasing γ because the enforcement of the $\int_{\Omega_0} |\nabla \tilde{d}|^2 dv = 0$ constraint increases with increasing γ . Our filter is very similar to that used by Matsumori et al. (2017) (2017) (2017) ; however, they only integrate $|\tilde{d} - d|^2$ over regions that are close to the boundary $\partial \Omega_0$ so the interior nodes are less constrained, which is most likely fine.

A comparison of the fltering approaches is worthy of discussion. The fictitious energy residual Eq. (15) (15) for d^0 and consistent flter residual Eq. [\(17](#page-7-0)) for *̃d* are solved over the initial design configuration Ω_0 whereas the H^1 smoothing Eq. [\(10](#page-5-2)) for $\widetilde{\text{grad}\theta_i}$ is solved over the current design configuration $Ω$. One difference between the fictitious energy and consistent approaches lies in the defnitions of the spaces of the admissible displacements \mathcal{H}_f and \mathcal{H}_c , ([13\)](#page-6-3) and ([18](#page-7-1)). In the fictitious energy method, $\mathbf{d} = d^p \mathbf{e}$ is prescribed by the optimizer over $\partial \Omega_0 \backslash A^f$ whereas in the consistent approach, *d* is not prescribed over this surface. Because of this, no smoothing is performed on this boundary in the fictitious energy method. Another diference between these methods is that the fctitious energy method requires the solution of a nonlinear PDE; albeit in the second variant of our consistent approach described next, we must also solve a nonlinear PDE.

The $\nabla \mathbf{w} \cdot \gamma \nabla \tilde{\mathbf{d}}$ term in ([17\)](#page-7-0) penalizes $|\nabla \tilde{\mathbf{d}}|$. As such, rigid transformations or dilations/contractions are penalized even though they do not adversely afect mesh quality. Motivated by this observation and by Scherer et al. [\(2010](#page-26-23)), in our second variant "energy flter" approach, we solve a nonlinear PDE by replacing $∫_{Ω_0} ∇ w ⋅ γ ∇ d v$ with $\int_{\Omega_0} \nabla \mathbf{w} \cdot \gamma \, \partial \Psi_f(\widetilde{\mathbf{F}}_d) / \partial \widetilde{\mathbf{F}}_d \, d\nu$ where $\widetilde{\mathbf{F}}_d = \mathbf{I} + \nabla \widetilde{\mathbf{d}}$. This has the effect of weakly enforcing the constraint $\int_{\Omega_0} \Psi_f(\vec{F}_d) dv = 0$ rather than $\int_{\Omega_0} |\nabla \tilde{\mathbf{d}}|^2 dv = 0$. And in our third variant, the "linearized energy flter," we alternatively penalize the linearized energy by replacing $\int_{\Omega_0} |\nabla \tilde{d}|^2 dv$ with $1/2 \int_{\Omega_0} \nabla \widetilde{\mathbf{d}} \cdot \mathbf{A}_f[\nabla \widetilde{\mathbf{d}}] dv = 0$ where $\mathbf{A}_f = \frac{\partial^2 \Psi_f(\mathbf{I})}{\partial \widetilde{\mathbf{F}}_d^2}$ provides the linearized energy, Appendix [3](#page-22-0) for details.

In these consistent approaches, we maintain a clear distinction between the filtered field \tilde{d} that is used to define Ω and the design displacement feld *d* that we optimize. So like the fictitious energy approach, this has ramification in Require: X_0 , D Ensure: Θ_i , $D\Theta_i/DD$ Solve mesh morphing problem $(K_c+M_c) D = M_c D$ for D on the Ω_0 mesh Define the Ω mesh with node coordinates $\bm{X}=\bm{X}_0+\widetilde{\bm{D}}$ Solve primal problem $\boldsymbol{K}\boldsymbol{U} = \boldsymbol{P}$ for \boldsymbol{U} on the Ω mesh for $i \in \{0, 1, 2, \dots\}$ do Compute Θ_i and the derivatives $\partial \Theta_i / \partial X$ and $\partial \Theta_i / \partial U$ on the Ω mesh, cf. (34) and (35) Solve adjoint problem $\boldsymbol{K}\boldsymbol{W} = \left(\partial \Theta_i / \partial \boldsymbol{U}\right)^T$ for \boldsymbol{W} Compute pseudo load $\partial R/\partial X$ on the Ω mesh, cf. (36) Compute sensitivity $D\Theta_i/D\mathbf{X} = \partial \Theta_i/\partial \mathbf{X} - \partial R/\partial \mathbf{X}$ Solve adjoint problem $(K_c+M_c)W = (D\Theta_i/DX)^T$ for W Compute sensitivity $D\Theta_i/DD = M_c W$ end for return Θ_i , $D\Theta_i/DD$

Algorithm 4 Consistent fltering formulation, Appendix 4 for the discretization

the sensitivity analysis, namely we again have to solve two adjoint problems, Appendix [4](#page-24-0) for details.

In each iteration of the consistent PDE fltering formulation, we are given the initial node coordinates X_0 and the design displacement *D* whereupon we complete Algorithm 4 to compute the Θ_i and their derivatives $D\Theta_i/DD$. The optimizer subsequently uses these quantities to either update the design, i.e., $D \leftarrow D + \Delta D$ or declare the convergence.

As expected, fltering *d* reduces numerical anomalies due to the discretization and hastens convergence of the optimization problem by reducing the effective design space. Indeed, the many design parameters in *d* are "mollifed" into a fewer number of infuential parameters. To see this, we use the spectral decomposition theorem to $\exp\left(\frac{\mathbf{\Phi}^T(\mathbf{K}_c + \mathbf{M}_c)}{\mathbf{\Phi}^T(\mathbf{K}_c + \mathbf{M}_c)}\right)$ = diag($\lambda_1, \lambda_2, \dots$) where (λ_i, Φ_i) are the eigenpairs of $K_c + M_c$ that are arranged such that $\lambda_i \leq \lambda_{i+1}$, and the eigenvectors are mass normalized such that $\mathbf{\Phi}_i^T \mathbf{M}_c \mathbf{\Phi}_j = \delta_{ij}$ and $\mathbf{\Phi} = [\mathbf{\Phi}_1 \ \mathbf{\Phi}_2 \ \cdots]$. In this way, with $\widetilde{D} = \Phi \widetilde{D}^*$ and $D = \Phi D^*$, we have

⁶ Appendix [4](#page-24-0) for details of this discretization.

$$
\widetilde{\boldsymbol{D}}^* = \begin{Bmatrix} \frac{1}{\lambda_1} D_1^* \\ \frac{1}{\lambda_2} D_2^* \\ \vdots \end{Bmatrix}
$$
 (21)

We see that the "modes" associated with the smaller eigenvalues contribute more to *D̃* than those associated with the larger eigenvalues. And the mode shapes associated with smaller eigenvalues are smoother than those associated with larger eigenvalues. Hence, \vec{D} is smoother than D , i.e., by replacing \vec{D} with \tilde{D} , we have mollified the space of possible design configurations $Ω$.

3 Filter comparison

To illustrate the parameter-free methods described in Sect. [2,](#page-4-1) we return to the problem discussed in Sect. [1](#page-0-1) and optimize the shape of a hole in a plane stress plate, initially a circle with radius $r = 1/(2\sqrt{\pi})$, to minimize the compliance (or maximum stress) subject to a constraint that the area be less than $A = 0.75 V$ where *A* is the area of the plate without the hole, Fig. [1](#page-1-0). The 2×2 plate is comprised of an isotropic linear elastic material with Young's modulus $E = 10.0$ and Poisson ratio $v = 0.3$ and subjected to a biaxial tensile load with prescribed traction $t^p = 2e_1$ and $t^p = e_2$ over the vertical and horizontal boundaries. Quarter symmetry is utilized to reduce the computational burden. For the $H¹$ and traction methods with $\Delta x = d$, and consistent method, we solve

$$
\min_{d \in \mathcal{H}_H} \theta_0
$$
\nsuch that

\n
$$
\theta_1 = \int_{\Omega} da - \overline{A} = 0
$$
\n
$$
\underline{d} \leq \mathbf{d} \cdot \mathbf{e}_i \leq \overline{d} \quad \text{for} \quad i = 1, 2
$$
\n
$$
(22)
$$

. And for the fctitious energy method, we solve

$$
\min_{d^p} \theta_0
$$
\nsuch that

\n
$$
\theta_1 = \int_{\Omega} da - \overline{A} = 0
$$
\n
$$
\theta_f = \int_{\Omega_0} \Psi_f(\mathbf{F}_f) da - \overline{\Psi}_f \leq 0
$$
\n
$$
d \leq d^p \leq \overline{d}
$$
\n(23)

In our reduced space optimization, we use the adjoint sensitivity analyses to account for the implicit dependence of the response, e.g., *u* with respect to the design. For the $H¹$, traction, and fictitious energy approaches, it is easy to place box constraints on the design because *d* and d^p directly affect the shape as $x = x_0 + d$; for all of these cases $d, d = \pm 0.25$. On the other hand, in the consistent

approaches, $x = x_0 + \tilde{d}$ where \tilde{d} is implicitly defined by the design *d*. So while we do provide box constraints on *d*, their range is large in comparison with those in the other approaches. As a guide, we choose $\underline{d}, \overline{d} \approx \pm \lambda_i \underline{\tilde{d}}$ where $\underline{\tilde{d}}$ is the bound we desire on \tilde{d} and λ_i is the *i* − *th* smallest eigenvalue of $K_c + M_c$ where $i \approx 4, 5, 6$. For all of these cases, $d, \overline{d} = \pm 20 \approx \pm 0.25 \cdot \lambda_4 = 18.25$. In all cases, these box constraints are necessary to obtain well-posed problems. Indeed, removing them could result in designs with fne scale shape oscillations in the continuum setting and tangled fnite element meshes in the discrete setting. All of the above functions θ_i are scaled so that the sensitivities of each function has a unit norm for the initial design, e.g., $|D\theta_i/DD| = 1$ for $D = 0$. An exception for this is the maximum energy constraint in the fictitious energy approach which is normalized for $d^p = 0.02$ as $D\theta_f/Dd^p = 0$ for $d^p = 0$. To keep things relatively equal, the H^1 smoothing and consistent PDE filter use $\gamma = 10$, the traction uses $E = 10$ and $v = 0.3$, and the fictitious energy and consistent approaches with the linearized energy function use $\gamma = 5$. On the other hand, the fictitious energy and consistent approaches with the nonlinear energy function use $\gamma = 1$. We conjecture that the lower γ value is needed due to the stiffening effect of the nonlinear energy function Ψ_f , i.e., a higher nonlinear energy is required to achieve the same level of mesh morphing as compared to the linearized energy. Loosely based on the $\overline{\Psi}_f = .006\overline{A} = 0.0045$ suggested by Scherer et al. (2010) (2010) (2010) (which does not include the γ factor), we assign $\overline{\Psi}_f = \gamma 0.030 \overline{A} = 0.1125$ for the maximum energy for the fictitious energy method with the linearized energy. We decreased the maximum to $\overline{\Psi}_f = 4 \gamma \Psi_f(0.02) = 0.0285$, i.e., 4γ times the "normalization" energy for the nonlinear energy case. The selection of these maximum energy bounds defnitely requires some "experimentation."

The optimization cost function θ_0 is either the compliance

$$
\Theta = \int_{\Omega} \nabla \boldsymbol{u} \cdot \mathbb{C}[\nabla \boldsymbol{u}] d\mathbf{a} \tag{24}
$$

or the *p*-norm aggregation of the von Mises stress σ_{VM}

$$
\Upsilon = \left(\int_{\Omega} \sigma_{VM}^p \, da\right)^{\frac{1}{p}} \tag{25}
$$

in which $p = 6$ throughout Sect. [3](#page-9-0). We use Θ and Υ to denote compliance and *p*-norm aggregation of the von Mises stress, respectively, throughout, and the superscript $⁰$ to denote the</sup> metric value of the initial design; e.g., Θ^0 refers to the compliance of the initial design.

The interior-point method of Matlab's fmincon with a limit of 100 function evaluations is used for the optimizations. The accuracies of the user supplied derivatives are verifed for the fctitious energy and consistent approaches. **Table 1** Compliance minimization results

Fig. 3 Strain energy distributions of compliance optimized designs

None of the methods produce the expected elliptical hole with 2:1 axis ratio due to the finite domain size; however, they all tend in that direction.

3.1 Compliance minimization

We first minimize the compliance $\theta_0 = \Theta$ of [\(24](#page-9-1)). This is an easy problem so all of the methods should and do perform admirably. Results from the various algorithms are presented in Table [1](#page-10-0).

The fictitious energy values for the optimized designs obtained via the linear and nonlinear energy functions are 0.117 and 0.0271; hence, the maximum energy constraints are, respectively, violated and relatively active. The nonlinear energies for the fictitious energy and consistent approaches required some special care in that the Newton solver did not always converge. To resolve this, the load was incrementally applied over numerous load steps as needed; as such, the morphing analysis dominated the computational expense. Figure [3](#page-10-1) illustrates the designs and their strain energy distributions. The contour plots qualitatively compare the methods with a consistent coloring scheme whereas the tables provide a quantitative comparison. The designs exhibit subtle diferences which are seen through their energy distributions. The consistent methods with linearized energy filter $1/2 \gamma \nabla d \cdot A_f[\nabla d]$ and nonlinear energy filter $\gamma \Psi_f$ produce the best designs, although all of the designs are reasonable. The consistent method with nonlinear energy filter $\gamma \Psi_f$ produces the best mesh based on the "eyeball" norm. The inconsistent $H¹$ and traction methods converge in the fewest number of iterations because the change in the design between iterations is too small, presumably because the optimizer gets "confused" by the inconsistent sensitivity analysis, as evidenced by greater compliance values.

3.2 Stress minimization

We next minimize the *p*-norm of the von Mises stress, i.e., in the optimization, we replace [\(24](#page-9-1)) with ([25\)](#page-9-2) so that $\theta_0 = \Upsilon$. This is a more difficult problem as we replace the compliance which is a global quantity with an approximation of the maximum stress which is a local quantity. All of the

Fig. 4 Von Mises stress distributions of stress optimized designs

Table 2 Stress minimization results

other parameters are the same as those described above. Results from the various algorithms are denoted in Table [2](#page-11-0) and Fig. [4](#page-11-1). Descriptions of the table and fgures are the same as those for the compliance minimization. The fctitious energy values for the optimized designs obtained via the linear and nonlinear energy functions are 0.115 and 0.0246; hence, the maximum energy constraints are again, respectively, violated and relatively active. The design obtained by the fictitious energy method with nonlinear energy is quite good; however, its nonlinear morphing analysis dominates the computational cost. All of the designs obtained by the consistent methods appear to be reasonable. And again the consistent method with nonlinear energy filter $\gamma \Psi_f$ produces the best mesh based on the "eyeball" norm. Anomalies exist in the designs obtained with the remaining methods. Parameter tuning could be used to improve these designs, but the point of this exercise is to fnd methods that are amenable to common nonlinear programming algorithms that do not require excessive parameter tuning.

4 Test problems

The advantages of consistent fltering methods were demonstrated in Sect. [3](#page-9-0). Here, we further showcase the three consistent fltering approaches when used to design both 2D and 3D structures. Specifcally, tradeofs between ease of implementation, computational burden, and mesh quality are explored. A summary of relevant parameters used throughout this section is presented in Table [3](#page-12-0). All results in this section used the MFEM library (Anderson et al. [2021\)](#page-25-17) to solve the PDEs and the IPOPT library (Wächter and Biegler [2006](#page-26-0)) to solve the optimization problems. The Livermore Design Optimization (LiDO) code was utilized to compute the required sensitivities, and the Targeted Matrix

Table 3 Parameters used throughout Sect. [4](#page-11-2)

Fig. 5 2D plate test problem, design/analysis symmetry cell outlined by dashed line

Optimization Paradigm (TMOP) (Knupp [2012;](#page-26-29) Dobrev et al. [2019](#page-25-18); Barrera et al. [2022](#page-25-19)) is used to implement the consistent nonlinear energy filter.

4.1 2D plate design

To compare the fltering methods, we revisit the design of the plane stress plate, but under biaxial or shear loading and with an initially square 1×1 hole rather than a circular hole, Fig. [5.](#page-12-2) Morphing the square into the optimal circle (biaxial loading) or diamond (shear loading) is challenging since it requires a right angle to be morphed into an arc or straight line, respectively. All optimizations presented in this section solve the stress minimization problem of Equations [\(22](#page-9-3)) and (25) . We chose to minimize the maximum stress since it is generally more challenging than compliance, which could hide deficiencies in the methods.

(a) Initial design

Fig. 6 Design of 2D plate under biaxial loading with the PDE flter

4.1.1 Consistent method with the PDE flter

We first consider the consistent PDE filter, which has one parameter γ that controls the tradeoff between how "smooth" the mesh is and how closely \tilde{d} matches d , as well as a set of parameter bounds d, \overline{d} . A smaller γ requires smaller, i.e., more restrictive, parameter bounds to avoid mesh entanglement, while a larger γ permits more liberal parameter bounds.

Figure [6](#page-12-3) illustrates how γ , d , and \overline{d} affect the optimized design. The two optimal designs boast similar performance with Y/Y^0 values of 0.38 and 0.39. The first takeaway is that we can use values of γ that are two orders of magnitude apart and obtain satisfactory results by tuning the *d* and *d* bounds. However, both of our optimized designs tend to "smash" elements along the symmetry planes, and further, this effect worsens for the larger value of γ . The advantage of this approach is the smooth boundary, which comes at the expense of element quality. In Fig. [7,](#page-13-0) we present the optimal design for shear loading, using the same $\gamma = 10^{-1}$ value that produced the higher quality mesh under biaxial loading. We again see a smooth boundary but notice many highly distorted elements near the symmetry planes. The stress contour plots, which use consistent color schemes for each load case, are provided to qualitatively compare designs, whereas Tables [4](#page-13-1) and [5](#page-13-2) quantitatively compare them.

To summarize, this PDE filter is simple to implement, especially for those familiar with the PDE flter from topology optimization. Indeed, repeated use of a

 7 In our implementation of TMOP (Barrera et al. [2022\)](#page-25-19), their objec-tive function of ([8\)](#page-5-5) $F(\mathbf{x}) = \sum_{E' \in M} \int_{E'} \mu(T) d\nu + w_{\sigma} \sum_{s \in S} \sigma^2(X^s)$ is minimized with respect to the node coordinates of the physical mesh $X = X_0 + \overline{D}$. In this expression, the so-called target elements *E^t* comprise all of the elements in the Ω_0 mesh, $\mu = \Psi_f$ is the quality metric and $T = \widetilde{F_d}$. TMOP constrains the nodes on A^f such that $X \cdot e = X_0 \cdot e$ and weakly enforces the $\widetilde{X} = X_0 + D$ "constraint" via the penalty function $\sigma^2(X^s) = |\widetilde{D}^s - D^s|^2$ for all nodes $s \in \Omega_0 \setminus A^f$
which the problem value of A^f . The TMOD matrix of with the small penalty value $w_{\sigma} = 1/\gamma$. The TMOP metrics μ_2 and μ_{303} Dobrev et al. [\(2019](#page-25-18)) equal Ψ_f of [\(40\)](#page-22-1) for 2 and 3 dimensions.

Table 4 Summary of 2D plate under biaxial loading optimization results

Table 5 Summary of 2D plate under shear loading optimization results

Fig. 7 Design of 2D plate under shear loading with the PDE flter **Fig. 8** Optimal 2D plate designs obtained via the linearized energy

scalar-valued PDE flter could be used to compute each component of \tilde{d} , although a vector-valued implementation was used here. This method is computationally inexpensive since it only requires linear PDE solves. Optimizations using this flter generally produce smooth boundaries at the expense of poor element quality and consequently reduced simulation accuracy.

4.1.2 Consistent method with the linearized energy flter

The linearized energy flter promotes higher element quality than the PDE flter as seen in the designs for biaxial and

filter

shear loading illustrated in Fig. [8.](#page-13-3) Specifcally, the elements are nearly squareshaped throughout the domain. Notably, the high mesh quality near the symmetry planes is in stark contrast to the results in Sect. [4.1.1;](#page-12-4) however, the tradeoff is that the optimizer struggles to remove the "corner" in the initial design's square hole. Thus, the maximum von Mises ratios in Fig. [8](#page-13-3) are slightly higher than their counterparts in Sect. [4.1.1](#page-12-4), i.e., 0.40 and 0.36 for the designs in this section, compared with 0.38 and 0.34 for the designs in Sect. [4.1.1.](#page-12-4)

When compared to the PDE flter, we see that the linearized energy flter ensures higher mesh quality at the expense of design freedom. It is also simple to implement,

especially since a linear elasticity solver can be re-purposed for this flter with proper selection of the Lamé parameters, i.e., $\lambda = -\frac{1}{3}$ and $\mu = \frac{1}{2}$ (see Appendix [3](#page-22-0) for details), and the computational expense is therefore equivalent to a single linear elasticity PDE solve.

4.1.3 Consistent method with the nonlinear energy flter

Finally, we consider the nonlinear energy filter (with quality metric μ_2 from Dobrev et al. [\(2019](#page-25-18)) which is the same as the $Ψ_f$ function of [\(40](#page-22-1))). Although this filter is more computationally expensive, it proves to be more capable than the linear flters presented in Sects. [4.1.1](#page-12-4) and [4.1.2.](#page-13-4) The robustness of the formulation is demonstrated by the mesh refnement study for the biaxial loaded design illustrated in Fig. [9.](#page-14-0) Indeed, the same parameters, $\gamma = 10^{-1}$, *d*, $\overline{d} \mp 0.1$, were used to obtain each result, despite their difering element size. Additionally, we see very similar optimized designs, demonstrating the mesh independence of this technique. Similar observations hold for the mesh refnement study performed on the shear loaded designs, Fig. [10.](#page-15-1) Notably, the same values of γ , *d*, and \overline{d} were used for both loadings, and again, we see consistent designs for the diferent mesh resolutions. When compared to the consistent PDE filter in Sect. [4.1.1,](#page-12-4) we notice higher element quality throughout Figs. [9](#page-14-0) and [10](#page-15-1), and when compared to the linearized energy flter in Sect. [4.1.2](#page-13-4), we notice a smoother boundary. The nonlinear energy flter produces the lowest maximum stress in the optimal design when comparing equivalent meshes across the three consistent flters, as listed in Tables [4](#page-13-1) and [5](#page-13-2). Thus, the nonlinear flter exhibits the advantages of both linear flters, i.e., high mesh quality and smooth boundaries, at the cost of additional computational burden.

As an alternative to increasing the number of elements, i.e., *h*-refnement, simulation accuracy can also be improved by increasing the polynomial order of the mesh, i.e., *p*-refnement. Figures [11](#page-15-0) and [12](#page-16-0) present optimal designs corresponding to linear, quadratic, and cubic elements. The *p*-refnement study used the same values of γ , d , and d as the *h*-refinement study. The sharp corner is much improved when using quadratic elements, and nearly entirely removed when using cubic elements. Further, the element quality is high throughout the domain. The designs obtained using the cubic meshes and the nonlinear energy flter appear to be the best when considering mesh quality, design performance, and number of degrees-of-freedom.

The consistent nonlinear energy flter can be summarized as an extremely efective, but computationally expensive alternative to the consistent linear flters. Indeed morphing computations may dominate the physics computations, especially if the latter are linear. Additionally, the implementation of a nonlinear filter is more difficult than the linear

Fig. 9 Effect of *h*-refinement on the design of 2D plate under biaxial loading with the nonlinear energy flter. Left column is initial (init) designs, and right column is optimized (opt) designs

flters. However, the improved optimization performance may warrant the added effort. Further, the combination of a smooth boundary representation and high element quality throughout the domain, especially when using meshes with higher-order elements, is unmatched by the other techniques. As design problems require more complex physics, the improved element quality obtained via the nonlinear energy flter may become even more important for accurate response predictions. Moreover, the computational cost will

Fig. 10 Efect of *h*-refnement on the design of 2D plate under shear loading with the nonlinear energy flter. Left column is initial (init) designs, and right column is optimized (opt) designs

be less signifcant when compared to the cost of the complex nonlinear physics simulations.

4.2 3D dam design

We use the design of a 3D dam subject to a design-dependent hydrostatic pressure load to further challenge the optimization scheme. Indeed, the direction of the applied load changes as the mesh morphs. The problem is summarized in Fig. [13](#page-16-1) by overlaying the pressure load and mechanical

Fig. 11 Effect of *p*-refinement on the design of 2D plate under biaxial loading with the nonlinear energy flter. Left column is initial (init) designs, and right column is optimized (opt) designs. *p* denotes the order of the solution basis functions

displacement boundary conditions on the mesh of the initial design. As pictured, we enforce homogeneous essential boundary conditions, i.e., $\mathbf{u} \cdot \mathbf{n} = 0$, on the surfaces with normals $n = -e_2$ and $n = \pm e_3$. We also enforce design constraints by fixing the normal displacement $\mathbf{d} \cdot \mathbf{n} = 0$ over the surfaces with normals $n = \pm e_2$ and $n = \pm e_3$. The initial design has dimensions [200, 700, 1200] in the e_1 , e_2 , and e_3 directions, respectively. We consider both compliance and maximum stress minimization.

4.2.1 Consistent method with the PDE flter

The PDE flter achieves satisfactory results for the 3D dam design problem. The minimum compliance design pictured in Fig. [14](#page-16-2) has 50% of the initial design's compliance. Element quality is difficult to qualitatively compare for 3D structures, but we note that there are highly distorted elements near the bottom corners of the design.

Fig. 12 Effect of *p*-refinement on the design of 2D plate under shear loading with the nonlinear energy flter. Left column is initial (init) designs, and right column is optimized (opt) designs. *p* denotes the order of the solution basis functions

Fig. 13 Dam design problem loads and boundary conditions applied to the initial design

Fig. 14 Compliance min. with the PDE flter

(a) Initial stress, front

(b) Initial stress, back

Fig. 15 Initial von Mises stress distribution

(a) Optimal stress, front (b) Optimal stress, back

Fig. 16 Maximum von Mises stress minimization with the PDE flter

A minimization of the maximum von Mises stress produces a design that resembles the compliance minimization design. The initial and optimized von Mises stress distributions are presented in Figs. [15](#page-16-3) and [16,](#page-16-4) respectively. The optimal design reduces the peak von Mises stress by nearly half; however, this flter provides the worst performance of the three consistent flters, Table [7](#page-18-0). Generally, the PDE flter is not able to displace the initial mesh as far as the energy-based flters before mesh entanglement occurs. Since this flter simply "smooths" the node coordinates without regard for element quality, the elements become distorted quite easily and therefore the optimal performance is limited by how far the mesh is able to distort before it tangles.

 (a) 33,495 design vari- (b) 33,495 design variables, front ables, back

 (c) 247,779 design vari- (d) 247,779 design variables, front ables, back

(e) $1,904,163$ design vari- (f) $1,904,163$ design variables, front ables, back

(a) Optimal stress, front (b) Optimal stress, back

Fig. 18 Maximum von Mises stress minimization with the linearized energy flter

4.2.2 Consistent method with linearized energy

The linearized energy filter was effective in the dam optimization. An *h*-refinement study of the compliance

(a) Initial stress, front

(b) Initial stress, back

Fig. 19 Initial Drucker-Prager stress distribution

(a) Optimal stress, front (b) Optimal stress, back

Fig. 20 Maximum Drucker-Prager stress minimization with the linearized energy flter

minimization problem was performed to investigate the mesh independence of the shape optimization framework, Fig. [17.](#page-17-0) Two important behaviors were observed: the same flter and bound parameters were efective for all three meshes, and the same performance, i.e., $\Theta/\Theta_0^0 = 0.42$ was obtained for all three designs. The fnal mesh contained 1,904,163 design variables, demonstrating the scalability of the method to large design problems. *The robustness provided by meshindependent flter parameters allows designers to tune optimization parameters on coarser meshes before using them on refned meshes.*

The linearized energy flter produced a design with less compliance than the PDE flter, Table [6.](#page-18-1) This trend was consistent with the minimum von Mises stress designs. Inspection of Fig. [18](#page-17-1) reveals a superior design to that in Fig. [16](#page-16-4); the linearized energy flter allows the mesh to morph further while keeping elements from tangling and therefore allows the fnal design to mitigate stress more efectively.

The astute civil engineer might notice, however, that our optimal dams are to this point concave with respect to the hydrostatic pressure load, whereas real-world dams are often convex with respect to the water they are retaining. The main driver of this inconsistency is the von Mises stress metric, which assumes materials are equally strong in tension and compression. Thus, a better stress metric to design for is the Drucker-Prager criterion (Drucker and Prager [1952](#page-25-20)) since re

re

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Table 6 Summary of 3D dam compliance minimization results	Filter	Init fig	Opt fig	Elements	Order	Dof	γ	$d, \overline{d} \mp$	Optimal Θ/Θ^0
	PDE	13	14	9520	1	33,495	10^{5}	700	0.50
	Lin energy	13	17a, 17b	9520	1	33,495	10^{5}	1400	0.42
	Lin energy	n/a	17c, 17d	76,160	1	247,779	10^{5}	1400	0.42
	Lin energy	n/a	17e, 17f	609,280	1	1,904,163	10^{5}	1400	0.42
	Nonlin energy	13	21	9520	1	33,495	10^{5}	14×50	0.39
Table 7 Summary of 3D dam von Mises stress optimization results	Filter	Init fig	Opt fig	Elements	Order	Dof	γ	d, \overline{d}	Optimal Υ/Υ^0
	PDE	15	16	9520	1	33,495	10^{5}	500	0.57
	Lin energy	15	18	9520	1	33,495	10^{5}	1000	0.45
	Nonlin energy	15	22	9520	1	33,495	10^{5}	13×50	0.37
Table 8 Summary of 3D dam Drucker-Prager stress optimization results	Filter	Init fig	Opt fig	Elements	Order	Dof	γ	d, \overline{d}	Optimal Υ/Υ^0
	Lin energy	19	20	9520	1	33,495	10^{5}	800	0.33
	Nonlin energy	19	23	9520	1	33,495	10^{5}	13×50	0.19

Fig. 21 Compliance minimization with nonlinear energy flter

most dams are made of concrete, which is strong in compression and weak in tension. If we assume that our constituent material is 10 times stronger in compression than in tension and use the Drucker-Prager stress criterion, we obtain the initial stress feld pictured in Fig. [19](#page-17-2). Designing for the Drucker-Prager criterion has a fascinating efect on the optimal geometry; the top surface of the dam switches from concave to convex with respect to the hydrostatic pressure load, which is a trait of arch dams, and two reinforcement columns appear on the front face, which is a trait of buttress dams, Fig. [20.](#page-17-3) In summary, when we account for the type of materials typically used to build dams, the optimized design resembles dams seen in the real world.

4.2.3 Consistent method with nonlinear energy

We attempted to use the nonlinear energy filter (with quality metric μ_{303} from Dobrev et al. [\(2019\)](#page-25-18) which equals Ψ_f from (40) (40)) in the same manner as in Sect. [4.1.](#page-12-5) A value was selected for γ and then the parameter bounds d , d were tuned accordingly. Through trial and error, it was determined that this technique was not efective when using the nonlinear energy flter for the 3D dam optimization problem. Specifcally, the optimal dam design required a much larger mesh distortion than the optimal plate design, which caused convergence issues during the fltering operation. Thus, a slightly diferent technique was developed for this problem. Rather than selecting a large bound range [*d*, *d*], we follow Scherer et al. [\(2010](#page-26-23)) and solve a series of optimizations with smaller ranges, wherein the initial design for each subsequent optimization was the optimized design from the previous optimization. In this way, the design was allowed to distort incrementally through many, more constrained, design problems. The key diference between solving the design problem in a single pass versus breaking it into numerous steps, as outlined here, is that the reference element is incrementally updated in the latter technique which introduces inconsistencies with respect to the initial design optimization problem.

Figure [21](#page-18-2) depicts the optimized design for compliance minimization using the nonlinear energy flter. The fnal design is superior to those obtained with the linear flters, Table [6](#page-18-1). Similarly, the best designs with respect to von Mises and Drucker-Prager metrics were obtained using the nonlinear energy filter with the incremental design process, Tables [7](#page-18-0) and [8.](#page-18-3) The fnal designs are displayed in Figs. [22](#page-19-0) and [23](#page-19-1), for von Mises and Drucker-Prager metrics, respectively. The design stress feld histories for the Drucker-Prager minimization are presented in Fig. [24](#page-19-2) to

(b) Optimal stress, back (a) Optimal stress, front

Fig. 22 Maximum von Mises stress minimization with nonlinear energy flter

Fig. 23 Maximum Drucker-Prager stress minimization with nonlinear energy flter

demonstrate the sequential optimization process described above. We emphasize that although 14 optimizations were required to obtain optimal designs, each of the optimizations converged quickly due to the smaller range [*d*, *d*]. In fact, the total number of design iterations for all of the design steps combined was similar to that required for the single-pass optimizations in Sect. [4.2.2.](#page-17-4) That said, the overall computational cost remains higher for the nonlinear flter since each design iteration requires the solution of the nonlinear flter PDE.

Again, we emphasize the tradeoff between computational cost and optimization performance must be considered. For example, the design problems using the highly refned meshes in Figs. [17e](#page-17-0) and [17f](#page-17-0) would have an increased execution time, perhaps an order of magnitude more, if using the nonlinear energy flter compared to the linearized energy flter to obtain only marginal compliance improvement, i.e., Θ/Θ^0 of 0.42 vs. 0.39. However, the design for Drucker-Prager stress using the nonlinear energy flter was signifcantly more efective than that obtained using the linear filter, i.e., Θ/Θ^0 of 0.33 vs. 0.19, possibly justifying the increased computational cost.

(b) Back views

Fig. 24 History of Drucker-Prager stress minimization designs with $d, \overline{d} \equiv 50$

5 Conclusions

Shape optimization provides an attractive option over topology optimization, especially when the boundary phenomena dominate the design, e.g., in stress-based design. Shape optimizations typically use CAD-based parameterizations which move the positions of the finite elements through globally supported basis functions. This type of design description can be difficult to implement for industry applications and limits the design to these predefined parameterizations.

Independent node parameterization has been extensively used, but numerical difficulties, e.g., mesh entanglement, still pose a challenge. Signifcant progress has been made to alleviate this challenge and here we add to this body of work. Specifcally, we demonstrate that consistent fltering, i.e., fltering both updated nodal positions and their sensitivities, is advantageous over fltering only sensitivities. We unify a technique for parameterizing locally supported fnite element quantities, i.e., element "densities" in topology optimization and nodal coordinates in shape optimization, to minimize human intervention in the design process.

We demonstrate consistent fltering by designing a 2D plate under biaxial and shear loading and a 3D dam subject to a hydrostatic pressure load. We show that the consistent PDE, linearized energy, and nonlinear energy flters all produce satisfactory designs, although each has pros and cons relative to the others. In summary, the PDE flter maintains design regularity, is computationally inexpensive and simple to implement, but sufers from poor element quality. The linearized energy flter is computationally inexpensive and simple to implement and produces higher element quality at the expensive of less design freedom. Finally, the nonlinear energy flter exhibits superior performance by simultaneously ensuring design freedom and high element quality, but incurs a steep computational cost. The choice of an appropriate flter is therefore problem dependent; the tradeof between design freedom, ease of implementation, and computational cost must be analyzed on a case-by-case basis.

Appendix 1: Independent node coordinate derivations

The variation of θ at x with respect to the shape perturbation δx is expressed as the sum of an *explicit* variation $\delta \theta^E$ and *implicit* variation $\delta\theta^I$, i.e., $\delta\theta = \delta\theta^E + \delta\theta^I$. The former is expressed

$$
\delta \theta^{E}(\mathbf{x}; \delta \mathbf{x})
$$
\n
$$
= \int_{\Omega} \left((-\nabla^{T} \mathbf{u} \frac{\partial \Lambda}{\partial \nabla \mathbf{u}} + \Lambda \mathbf{I}) \cdot \nabla \delta \mathbf{x} + \frac{\partial \Lambda}{\partial \mathbf{x}} \cdot \delta \mathbf{x} \right) d\nu
$$
\n
$$
+ \int_{A^{I}} \left(\frac{\partial \tau}{\partial \mathbf{x}} \cdot \delta \mathbf{x} + \tau (\mathbf{I} - \mathbf{n} \otimes \mathbf{n}) \cdot \nabla \delta \mathbf{x} \right) d\mathbf{a}
$$
\n
$$
+ \int_{A^{P}} \left(\left(\frac{\partial \tau}{\partial \mathbf{x}} \right)^{T} \mathbf{n} \cdot \delta \mathbf{x} + ((\tau \cdot \mathbf{n}) \mathbf{I} - (\mathbf{n} \otimes \tau)) \cdot \nabla \delta \mathbf{x} \right) d\mathbf{a}
$$
\n(26)

and the latter as

$$
\delta\theta^{I}(\mathbf{x};\stackrel{\star}{u}) = \int_{\Omega} \left(\frac{\partial \Lambda}{\partial \nabla u} \cdot \nabla \stackrel{\star}{u} + \frac{\partial \Lambda}{\partial u} \cdot \stackrel{\star}{u} \right) dv + \int_{A^{i}} \stackrel{\star}{u} \cdot \frac{\partial \tau}{\partial u} da + \int_{A^{p}} \stackrel{\star}{u} \cdot \left(\frac{\partial \tau}{\partial u} \right)^{T} n da
$$
 (27)

To obtain the above, we defne the *material derivative* with respect the *design velocity* δx as $\phi(x) = \delta u(x; \delta x) + \nabla u(x) \delta x$; it is implicitly defined by (1) (1) (1) .^{[8](#page-20-0)} We also use the commuter identity $\frac{\star}{\nabla u} = -\nabla u \nabla \delta x + \nabla \dot{u}.$

In the adjoint sensitivity method, we annihilate the implicit variation $\delta\theta^I$. To do this, we take the variation of the residual Eq. ([1](#page-4-0)) and, as above, express it as the sum of an explicit variation δr^E and implicit variation δr^I , i.e., $\delta r = \delta r^E + \delta r^I$. The explicit and implicit variations are

$$
\delta r^{E}(\mathbf{u}, \mathbf{w}, \mathbf{x}; \delta \mathbf{x}) =
$$
\n
$$
\int_{\Omega} \left(\left(-\nabla^{T} \mathbf{w} \mathbb{C}[\nabla \mathbf{u}] - \nabla^{T} \mathbf{u} \mathbb{C}[\nabla \mathbf{w}] \right) \right. \\
\left. + (\nabla \mathbf{w} \cdot \mathbb{C}[\nabla \mathbf{u}] - \mathbf{w} \cdot \mathbf{b}) \mathbf{I} \right) \cdot \nabla \delta \mathbf{x} \\
+ \left(\left(\frac{\partial \mathbb{C}[\nabla \mathbf{u}]}{\partial \mathbf{x}} \right)^{T} [\nabla \mathbf{w}] - \left(\frac{\partial \mathbf{b}}{\partial \mathbf{x}} \right)^{T} \mathbf{w} \right) \cdot \delta \mathbf{x} \right) d\mathbf{v} \\
- \int_{A'} \left(\left(\frac{\partial t^{p}}{\partial \mathbf{x}} \right)^{T} \mathbf{w} \cdot \delta \mathbf{x} \right. \\
\left. + (\mathbf{w} \cdot t^{p}) (\mathbf{I} - \mathbf{n} \otimes \mathbf{n}) \cdot \nabla \delta \mathbf{x} \right) da \\
- \int_{A^{p}} \left(\left(\frac{\partial p}{\partial \mathbf{x}} \otimes \mathbf{w} \right) \mathbf{n} \cdot \delta \mathbf{x} + p ((\mathbf{w} \cdot \mathbf{n}) \mathbf{I} \right. \\
\left. - (\mathbf{n} \otimes \mathbf{w})) \cdot \nabla \delta \mathbf{x} \right) da
$$

and

¹ \overrightarrow{B} In the material derivative context, $\overrightarrow{u}(x, t) = \partial u(x, t) / \partial t + \nabla u(x, t) v(t)$ where *v* is the *design velocity feld* and *t* is the pseudo time. The $\partial u(x, t)/\partial t$ term accounts for the change in *u* due to the shape perturbation for a fixed *location* in space *x* whereas the $\nabla u(x, t) v(t)$ term accounts for the movement of the material point.

$$
\delta r^{I}(\boldsymbol{u}, \boldsymbol{w}, \boldsymbol{u}; \boldsymbol{\dot{u}}) = \int_{\Omega} \nabla \boldsymbol{w} \cdot \mathbb{C}[\nabla \boldsymbol{\dot{u}}] dv
$$

= $a(\boldsymbol{\dot{u}}, \boldsymbol{w})$ (29)

Now we subtract the zero $\delta r = \delta r^E + \delta r^I = 0$ from $\delta\theta = \delta\theta^E + \delta\theta^I$ and rearrange to obtain

$$
\delta_i \theta(\mathbf{x}, \delta \mathbf{x}) = \delta \theta^E(\mathbf{x}, \delta \mathbf{x}) \n- \delta r^E(\mathbf{u}, \mathbf{w}, \mathbf{x}; \delta \mathbf{x}) \n+ \left(\delta \theta^I(\mathbf{x}, \mathbf{u}) - \delta r^I(\mathbf{u}, \mathbf{w}, \mathbf{u}; \mathbf{u}) \right)
$$
\n(30)

To eliminate the implicit variations $\delta \theta^I(x, \mathbf{x}) - \delta r^I(\mathbf{u}, \mathbf{w}, \mathbf{u}; \mathbf{x})$, we solve the following adjoint equation for the heretofore arbitrary $w \in \mathcal{H}(0)$

$$
0 = a(\mathbf{u}, \mathbf{w})
$$

\n
$$
- \int_{\Omega} \left(\nabla \mathbf{u} \cdot \frac{\partial \Lambda}{\partial \nabla u} + \mathbf{u} \cdot \frac{\partial \Lambda}{\partial u} \right) dv
$$

\n
$$
- \int_{A'} \mathbf{u} \cdot \frac{\partial \tau}{\partial u} da
$$

\n
$$
- \int_{A'} \mathbf{u} \cdot \left(\frac{\partial \tau}{\partial u} \right)^T \mathbf{n} da
$$
\n(31)

for all $\vec{u} \in H(0)$. As such, the sensitivity reduces to

$$
\delta\theta(\mathbf{x};\delta\mathbf{x}) = \delta\theta^{E}(\mathbf{x};\delta\mathbf{x}) - \delta r^{E}(\mathbf{u}, \mathbf{w}, \mathbf{x};\delta\mathbf{x}).
$$
\n(32)

Note that $\delta\theta$ can be expressed as a boundary integral by applying the divergence method, Haug and Choi [\(1986](#page-26-26)), but it is well known that this introduces error in the fnite element computations as mentioned in Sect. [1.](#page-0-1) Consequently, we use the "domain" expression ([32](#page-21-0)) in *all* of our computations.^{[9](#page-21-1)}

We now discuss the fnite element discretization. The Equilibrium Equation ([1\)](#page-4-0) becomes

$$
\boldsymbol{W}^T(\boldsymbol{K}\,\boldsymbol{U}-\boldsymbol{P})=0\tag{33}
$$

where *U* and *W* are the degrees-of-freedom vectors corresponding to u and w , K is the symmetric stiffness matrix defined such that $W^T K U = a(u, w) = \int_{\Omega} \nabla w \cdot \mathbb{C}[\nabla u] dv$ and *P* is the load vector defined such that $W^T P = \ell(w) = \int_{\Omega} w \cdot b \, dv + \int_{A^t} w \cdot t^n \, da + \int_{A^p} w \cdot p \, n \, da$. The arbitrariness of *W* yields the linear equation $K U = P$ that we solve for *U*.

After evaluating U , we can interpolate u and ∇u within the elements as $u(x) = N(x) U$ and $\nabla u(x) = B(x) U$ where *N* contains the element shape functions, $B = \partial N / \partial x$ and the gradient $∇$ *u* is expressed in the usual vector representation formed by stacking its columns. These interpolations allow us to evaluate the cost and constraint functions θ of ([3\)](#page-4-2) by summing over the elements Ω*e* and performing the usual quadratures. The same interpolation is used to express $x = N(x)X, \delta x = N(x) \delta X$, and $\nabla \delta x = B(x) \delta X$. In this way, $\delta \theta^{E}(\mathbf{x}; \delta \mathbf{x}) = \delta \mathbf{X}^{T} \partial \Theta / \partial \mathbf{X}$ where

$$
\frac{\partial \Theta}{\partial X} = \mathbb{A}_{e=1}^{n_{\text{elem}}} \left[\int_{\Omega_e} \left(\mathbf{B}^T \left(-\nabla^T \mathbf{u} \frac{\partial \Lambda}{\partial \nabla \mathbf{u}} + \Lambda \mathbf{I} \right)^T + \mathbf{N}^T \frac{\partial \Lambda}{\partial \mathbf{x}} \right) d\nu \right. \left. + \int_{A_e'} \left(\mathbf{N}^T \frac{\partial \tau}{\partial \mathbf{x}} + \mathbf{B}^T \tau (\mathbf{I} - \mathbf{n} \otimes \mathbf{n}) \right) d\mathbf{a} \right. \left. + \int_{A_e'} \left(\mathbf{N}^T \left(\frac{\partial \tau}{\partial \mathbf{x}} \right)^T \mathbf{N} \right. \left. + \mathbf{B}^T ((\tau \cdot \mathbf{n}) \mathbf{I} - (\mathbf{n} \otimes \tau)) \right) d\mathbf{a} \right]
$$
\n(34)

in which A is the usual finite element assembly operator and all tensors, e.g., $-\nabla^T u \frac{\partial \Lambda}{\partial \nabla u} + \Lambda I$ are expressed in their vector representations. We similarly have $\delta\theta^I(x;\vec{u}) = \vec{U}\partial\Theta/\partial U$ where

$$
\frac{\partial \Theta}{\partial U} = A_{e=1}^{n_{\text{elem}}} \left[\int_{\Omega_e} \left(\boldsymbol{B}^T \frac{\partial \Lambda}{\partial \nabla \boldsymbol{u}} + \boldsymbol{N}^T \frac{\partial \Lambda}{\partial \boldsymbol{u}} \right) d\boldsymbol{v} \right]
$$

$$
+ \int_{A_e'} \boldsymbol{N}^T \frac{\partial \boldsymbol{\tau}}{\partial \boldsymbol{u}} d\boldsymbol{a} + \int_{A_e'} \boldsymbol{N}^T \left(\frac{\partial \boldsymbol{\tau}}{\partial \boldsymbol{u}} \right)^T \boldsymbol{N} d\boldsymbol{a} \right]
$$
(35)

Further, $\delta r^{E}(\boldsymbol{u}, \boldsymbol{w}, \boldsymbol{x}; \delta \boldsymbol{x}) = \delta \boldsymbol{X}^{T} \partial R / \partial X$ where

$$
\frac{\partial R}{\partial X} = A_{e=1}^{n_{\text{elem}}}\left[\int_{\Omega_e} \left(B^T \left(-\nabla^T w \mathbb{C}[\nabla u]\right) - \nabla^T u \mathbb{C}[\nabla w] + (\nabla w \cdot \mathbb{C}[\nabla u] - w \cdot b)I\right)\right] \n+ N^T \left(\left(\frac{\partial \mathbb{C}[\nabla u]}{\partial x}\right)^T [\nabla w] - \left(\frac{\partial b}{\partial x}\right)^T w\right)\right) dv \n- \int_{A_e'} \left(N^T \left(\frac{\partial t^p}{\partial x}\right)^T w \n+ B^T (w \cdot t^p) (I - n \otimes n)\right) da \n- \int_{A_e'} \left(N^T \left(\frac{\partial p}{\partial x} \otimes w\right) n \n+ B^T p ((w \cdot n)I - (n \otimes w)) \right) da
$$

in which *w* and ∇*w* are interpolated similarly to *u* and ∇*u*. Finally

⁹ The gradient smoothing and fictitious energy algorithms cited in Sect. [1](#page-0-1) use the less accurate boundary expression. We implement those algorithms here using the more accurate domain expression.

$$
\delta r^{I}(\boldsymbol{u}, \boldsymbol{w}, \boldsymbol{x}; \stackrel{\star}{\boldsymbol{u}}) = a(\boldsymbol{w}, \stackrel{\star}{\boldsymbol{u}}) = \stackrel{\star}{\boldsymbol{U}}^{T} \boldsymbol{K} \boldsymbol{W} \tag{37}
$$

so that

$$
\delta \theta^{I}(\mathbf{x}, \mathbf{u}) - \delta r^{I}(\mathbf{u}, \mathbf{w}, \mathbf{x}; \mathbf{u})
$$

=
$$
\mathbf{U}^{T} (\partial \Theta / \partial U)^{T} - \mathbf{K} \mathbf{W}).
$$
 (38)

To annihilate this implicit term, we equate \boldsymbol{U}^T ($\partial \boldsymbol{\Theta}/\partial \boldsymbol{U}^T$ – $\boldsymbol{K} \boldsymbol{W}$) = 0 and use the arbitrariness of *U* to define the adjoint problem $\boldsymbol{K} \boldsymbol{W} = (\partial \Theta / \partial U)^T$ that we solve for *W*. Finally, we evaluate the sensitivity $D\Theta/DX = \frac{\partial \Theta}{\partial X} - \frac{\partial R}{\partial X}$.

Appendix 2: Gradient smoothing derivations

The finite element implementation of (10) (10) gives the sensitivity flter equation

$$
W^{T}\left((\widetilde{K} + \widetilde{M})\widetilde{\text{Grad}\Theta} - \left(\frac{D\Theta}{DX}\right)^{T}\right) = 0
$$
\n(39)

where we define Grad_Θ as the degrees-of-freedom vector corresponding to grad θ , \tilde{K} as the symmetric smoothing stiffness matrix that satisfies $W^T \tilde{K} U = \int_{\Omega} \nabla w \cdot \gamma \nabla u \, dv$ and *M̃* as the symmetric smoothing mass matrix that satisfes $W^T \widetilde{M} U = \int_{\Omega} w \cdot u \, dv$. The arbitrariness of *W* gives the linear equation $(\widetilde{\mathbf{K}} + \widetilde{\mathbf{M}})$ $\widetilde{\text{Grad}\Theta} = (D\theta / DX)^T$ that we solve for $Grad$ Θ.

Appendix 3: Fictitious energy derivations

Fictitious energy function

We desire a fictitious energy density functional Ψ_f that is unafected by rigid motions and dilations/contractions as we do not want to penalize an element that is subjected to such a shape preserving transformation. To these ends, we use pose the fctitious energy function

$$
\Psi_f(\boldsymbol{F}_d) = \frac{\text{tr}\boldsymbol{C}_d}{(\det \boldsymbol{C}_d)^{\frac{1}{n}}} - n
$$
\n(40)

where $C_d = F_d^T F_d$ is the fictitious symmetric right-Cauchy-Green strain and $n = 2(3)$ for two(three)-dimensional domains. It is readily verified that for $F_d = \alpha R_d$, i.e., a combined rotation \mathbf{R}_d and dilation/contraction $\alpha \mathbf{I}$ that $Ψ_f(F_d) = 0.$

To obtain the Eq. ([40\)](#page-22-1) energy density functional Ψ_f , we note that

$$
\boldsymbol{F}_d = \boldsymbol{R}_d \, \boldsymbol{U}_d = \boldsymbol{R}_d \, (\det \boldsymbol{F}_d)^{1/n} \, \overline{\boldsymbol{U}}_d \tag{41}
$$

where \mathbf{R}_d is a rotation and \mathbf{U}_d is symmetric positive definite. The frst equality is a result of the polar decomposition theorem and the second uses the definition $\overline{U}_d = (\det F_d)^{-1/n} U_d$ so that det $U_d = 1$. As seen above, the fictitious deformation gradient consists of an isochoric distortion/contraction U_d , followed by a dilation $(\det F_d)^{1/n} I$ followed by a rotation *Rd*. Based on this decomposition, we pose the Neo-Hookean like fictitious energy function that does not depend on the dilation/contraction det F_d *I* nor the rotation R_d

$$
\Psi_f(\mathbf{F}_d) = \text{tr}\overline{\mathbf{U}}_d^2 - n \tag{42}
$$

which is equivalent to ([40\)](#page-22-1). Using the spectral decomposition theorem, we can express the above for two dimensions as

$$
\Psi_f(\mathbf{F}_d) = \lambda^2 + \frac{1}{\lambda^2} - 2 \tag{43}
$$

where λ and $1/\lambda$ are the positive eigenvalues of \overline{U}_d and for three dimensions as

$$
\Psi_f(\mathbf{F}_d) = \lambda_1^2 + \lambda_2^2 + \frac{1}{\lambda_1^2 \lambda_2^2} - 3 \tag{44}
$$

where λ_1 , λ_2 and $1/(\lambda_1 \lambda_2)$ are the positive eigenvalues of U_d . A simple derivation subsequently indicates that the minimum energy is obtained for $\lambda = 1$ in two dimensions and for $\lambda_1 = \lambda_2 = 1$ in three dimensions which implies $U_d = I$, i.e., there is no distortion. This further confirms that $\mathbf{F}_d = \alpha \mathbf{R}_d$ is an energy minimizer for which $\Psi_f(\alpha \mathbf{R}_d) = 0$.

Newton's method

To solve the residual Eq. [\(15\)](#page-6-2), we use Newton's method and iterate by finding the $\Delta d \in \mathcal{H}_f(0)$ such that

$$
\delta_{d^0} r_f(\boldsymbol{d}^0, \boldsymbol{w}, \boldsymbol{d}^p; \Delta \boldsymbol{d}^0) = \int_{\Omega_0} \nabla \boldsymbol{w} \cdot \frac{\partial^2 \Psi_f}{\partial \boldsymbol{F}_d^2} [\nabla \Delta \boldsymbol{d}^0] \, d\boldsymbol{v}
$$
\n
$$
= -r_f(\boldsymbol{d}, \boldsymbol{w}, \boldsymbol{d}^p) \tag{45}
$$

for all $w \in H_f(0)$. Upon evaluating Δd^0 we update $d^0 \leftarrow d^0 + \Delta d^0$ and continue iterating until $r(d, w, d^p) \leq \epsilon$ for some suitable convergence tolerance ϵ . Note that the fictitious energy residual Eq. (14) (14) for d^0 is solved over the initial configuration $Ω_0$ whereas the *H*¹ smoothing Eq. ([10\)](#page-5-2) for grad θ is solved over the current design configuration Ω .

To evaluate the residual and tangent, we need

$$
\frac{\partial \Psi_f}{\partial F_d} = 2 F_d \frac{\partial \Psi_f}{\partial C_d}
$$
\n
$$
\frac{\partial^2 \Psi_f}{\partial F_d^2} = 2 I \boxtimes \frac{\partial \Psi_f}{\partial C_d}
$$
\n
$$
+ 4(F \boxtimes I) \frac{\partial^2 \Psi_f}{\partial C_d^2} (F \boxtimes I)^T
$$
\n(46)

where, for any 2-tensors *A* and *B*, the conjugate product $A \boxtimes B$ is the 4-tensor that satisfies $(A \boxtimes B)[C] = A C B^T$ for any 2-tensor C . For the energy function of (40) (40) , we have

$$
\frac{\partial \Psi_f}{\partial C_d} = (\det C_d)^{-\frac{1}{n}} \left(I - \frac{1}{n} \operatorname{tr} C_d C_d^{-1} \right)
$$

$$
\frac{\partial^2 \Psi_f}{\partial C_d^2} = \frac{1}{n} (\det C_d)^{-\frac{1}{n}} \left(-I \otimes C_d^{-1} - C_d^{-1} \otimes I \right)
$$

$$
+ \frac{1}{n} \operatorname{tr} C_d C_d^{-1} \otimes C_d^{-1} + \operatorname{tr} C_d C_d^{-1} \otimes C_d^{-1} \right)
$$
(47)

where the dyadic product $A \otimes B$ is the 4-tensor that satisfies $(A \otimes B)[C] = (B \cdot C)A$ for any 2-tensor *C*.

A linearization about $F_d = I$, which is accurate when ∇d is "small," provides

$$
\Psi_f = \frac{1}{2} \nabla d \cdot A_f[\nabla d]
$$

\n
$$
\frac{\partial \Psi_f}{\partial F_d} = A_f[\nabla d]
$$

\n
$$
\frac{\partial^2 \Psi_f}{\partial F_d^2} = A_f
$$
\n(48)

where

$$
\mathcal{A}_f = \frac{\partial^2 \Psi_f}{\partial \mathbf{F}_d^2} \bigg|_{\mathbf{F}_d = \mathbf{I}} = 4 \bigg(\mathbb{I}_{\text{sym}} - \frac{1}{n} \mathbf{I} \otimes \mathbf{I} \bigg)
$$
(49)

wherein the symmetrizer \mathbb{I}_{sym} is the 4-tensor that satisfies $\mathbb{I}_{sym}[C] = \frac{1}{2}(C + C^{T})$ for any 2-tensor *C*. Note that $\mathbb{A}_f[C] = 4 \left(\mathbb{I}_{sym}[C] - \text{tr}CI \right)$ is deviatoric and symmetric and hence as expected $\mathbb{A}_f[\nabla d] = \mathbf{0}$ for $\nabla d = \alpha \mathbf{I}$ an infinitesimal dilation/contraction, for $\nabla d = W$ where *W* is skew, i.e., an infinitesimal rotation and for their combination $\nabla d = \alpha W$. The stiffness tensor $\mathbb C$ for a linear elastic isotropic material can be expressed as $\mathbb{C} = 2 \mu \mathbb{I}_{sym-\text{dev}} + 3 \kappa \mathbb{I}_{sph}$ where $\mathbb{I}_{sym\text{-dev}} = \mathbb{I}_{sym} - \frac{1}{n}I \otimes I$ and $\mathbb{I}_{sph} = \frac{1}{n}I \otimes I$ are the projections onto the spaces of symmetric deviatoric tensors and spherical tensors, respectively. And hence, the "fctitious" material can be thought of as an elastic material with shear stifness $\mu = 1/2$ and bulk modulus $\kappa = 0$ which also implies it has first Lamé parameter $\lambda = -1/3$, Poisson ratio $\nu = -1$, and Young's modulus $E = 0$.^{[10](#page-23-0)[11](#page-23-1)}

Sensitivity analysis

Noting that $\delta x = \delta d$ allows us to write

$$
\delta\theta(\mathbf{d};\delta\mathbf{d}) = \delta\theta^{E}(\mathbf{x};\delta\mathbf{x}) - \delta r^{E}(\mathbf{u},\mathbf{w},\mathbf{x};\delta\mathbf{x})
$$
\n(50)

from ([32](#page-21-0)). The problem arises because *d* is an implicit function of d^p as seen through (15) (15) (15) , i.e., we have $d = d^0(d^p) + d^p(d^p)$ and hence, with a slight abuse of notation, we have

$$
\theta = \theta(d^p) = \theta(d(d^p)) = \theta(d^0(d^p), d^p(d^p)).
$$
\n(51)

We annihilate the implicit variation $\delta d^0(d^p; \delta d^p)$ in ([50](#page-23-2)) by again applying the adjoint method. To do this, the total vari-ation of [\(15](#page-6-2)) with respect to d^p in the direction δd^p , which equals zero, is subtracted from (50) (50) to give

$$
\delta\theta(d^p;\delta d^p) = \delta_{d^0}\theta(d^0(d^p), d^p(d^p);\delta d^0(d^p;\delta d^p))
$$

+
$$
\delta_{d^p}\theta(d^0(d^p), d^p(d^p);\delta d^p(d^p;\delta d^p))
$$

-
$$
\{\delta_{d^0}r_f(d^0(d^p), w, d^p(d^p);\delta d^0(d^p;\delta d^p))
$$

+
$$
\delta_{d^p}r_f(d^0(d^p), w, d^p(d^p);\delta d^p(d^p;\delta d^p))\}
$$

(52)

where $\delta_{d^0} r_f$ is the tangent operator of ([45](#page-22-2)) and the term in braces equals zero. Upon requiring the heretofore arbitrary $w \in H_f(0)$ to satisfy

$$
\delta_{d^0} r_f(d^0(d^p), w, d^p(d^p); \delta d^0(d^p; \delta d^p))
$$

=
$$
\delta \theta(d(d^p); \delta d^0(d^p; \delta d^p))
$$
 (53)

for all $\delta d^0 \in \mathcal{H}_f(0)$, ([62\)](#page-24-1) reduces to

$$
\delta\theta(d^p;\delta d^p) = \delta_{d^p}\theta(d^0(d^p), d^p(d^p);\delta d^p(d^p;\delta d^p))
$$

-
$$
\delta_{d^p}r_f(d^0(d^p), w, d^p(d^p);\delta d^p(d^p;\delta d^p))
$$
 (54)

which is the desired explicit sensitivity expression. The fact that r_f and θ are defined over different domains Ω_0 and Ω is not a concern as the change-of-variable theorem can always be used to express θ over Ω_0 . Indeed, the change-of-variable theorem is often used to obtain the shape sensitivity expressions $\delta \theta_i^E$ and δr^E of [\(26](#page-20-1)) and [\(28](#page-20-2)).

A simplifcation arises for the sensitivity analysis of the Equation (16) (16) (16) constraint on the total fictitious energy. Indeed, using the adjoint sensitivity analysis, one can show that

¹⁰ For the $n = 2$ case, $\mathbb C$ is obtained by using the plane stress assumption.

¹¹ A constraint on the linearized energy, i.e., mesh quality metric, $\nabla d \cdot \mathbb{I}_{sym-dev}[\nabla d]$ is used by Matsumori et al. ([2017\)](#page-26-12) to limit mesh distortion.

$$
\delta\theta_f(d^p; \delta d^p) = \int_{\Omega_0} \frac{\partial \Psi_f(\mathbf{F}_d)}{\partial \mathbf{F}_d} \cdot \nabla \delta d_p(d^p; \delta d^p) dv \n- \delta_{d^p} r_f(\mathbf{d}^0(d^p), \mathbf{w}, \mathbf{d}^p(d^p); \delta \mathbf{d}^p(d^p; \delta d^p))
$$
\n
$$
= r_f(\mathbf{d}^0(d^p), \delta \mathbf{d}^p(d^p; \delta d^p), \mathbf{d}^p(d^p))
$$
\n(55)

where we use ([15](#page-6-2)) with $w = \delta d^p (d^p; \delta d^p)^{12}$ and the fact that the adjoint response $w = 0$. The latter claim is true since, for all $\delta d^0 \in \mathcal{H}_f(0)$, $w \in \mathcal{H}_f(0)$ must satisfy

$$
\delta_{d^0} r_f(d^0(d^p), w, d^p(d^p); \delta d^0(d^p; \delta d^p))
$$
\n
$$
= \int_{\Omega_0} \frac{\partial \Psi_f(\mathbf{F}_d)}{\partial \mathbf{F}_d} \cdot \nabla \delta d^0(d^p; \delta d^p) dv
$$
\n
$$
= r_f(d^0(d^p), \delta d^0(d^p; \delta d^p), d^p(d^p))
$$
\n(56)

where we used ([15\)](#page-6-2) with $w = \delta d^0(d^p; \delta d^p) \in \mathcal{H}_f(0)$ to obtain the last equality. We trivially obtain $w = 0$ since the load linear term on right-hand side equals zero.

Discretization

In the finite element discretization, we partition *D* the degrees-of-freedom vector corresponding to $d = d^0 + d^p$, into the degrees-of-freedom vectors D^0 and D^p that correspond to d^0 and d^p ; the discretization and partition of w are similarly performed. The residual Equation [\(15\)](#page-6-2) is then expressed as

$$
\boldsymbol{W}^{0^T} \boldsymbol{R}_f^0(\boldsymbol{D}^0, \boldsymbol{D}^p) = 0 \tag{57}
$$

where R_f is the residual vector defined such that $W^T R_f = \int_{\Omega_0} \nabla w \cdot \frac{\partial \Psi_f}{\partial F_d} dv$ and R_f^0 is the partition of R_f that corresponds to W^0 . Notably, we only solve the residual equation $r_f = 0$ for the degrees-of-freedom that corresponds to the unknown \boldsymbol{D}^0 . The discretization of the Newton update equation ([45\)](#page-22-2) yields

$$
\boldsymbol{W}^{0^T} \left(\boldsymbol{K}_f^{00} \Delta \boldsymbol{D}^0 + \boldsymbol{R}_f^0 \right) = 0 \tag{58}
$$

where K_f is the symmetric tangent stiffness matrix defined such that $W^T K_f \Delta D = \int_{\Omega_0} \nabla w \cdot \frac{\partial^2 \Psi_f}{\partial F_d^2} [\nabla \Delta d] dv$ and K_f^{00} is the partition of K_f that corresponds to W^0 and ΔD^0 . The arbitrariness of W^0 renders the linear update equation K_f^{00} $\Delta D^0 = -R_f^0$ that we solve for ΔD^0 . Updates $\overline{D}^0 \leftarrow \overline{D}^0 + \Delta \overline{D}^0$ continue until $|R_f^0| \leq \epsilon$ for suitably small tolerance ϵ .

The adjoint Eq. (53) (53) is discretized as

$$
\delta \boldsymbol{D}^{0^T} \left(\boldsymbol{K}_f^{00} \boldsymbol{W}^0 - \left(\frac{D\theta}{D\boldsymbol{X}^0} \right)^T \right) = 0 \tag{59}
$$

where $D\theta / DX^0$ is the partition of $D\theta / DX$ that is associated with the D^0 degrees-of-freedom. The arbitrariness of δD^0 renders the linear adjoint equation $K_f^{00} W^0 = (D\theta / D X^0)^T$ that we solve for W^0 . Finally, the sensitivity Eq. ([54](#page-23-4)) is discretized such that

$$
\frac{D\theta}{DD^p} = \frac{D\theta}{DX^p} - W^{0^T} K_f^{0p}
$$
\n(60)

where $D\theta$ / DX^p is the partition of $D\theta$ / DX that is associated with D^p and K_f^{0p} is the partition of K_f that is associated with W^0 and ΔD^p . The sensitivity of the energy constraint ([16\)](#page-7-2) is obtained directly by evaluating $D\Theta_f/DD^p = \partial \Theta_f/\partial D^p$. The computation of $\partial \Theta_f / \partial D^p = R_f^p$ follows from ([15](#page-6-2)) and ([55\)](#page-24-3) where \mathbf{R}_f^p is the partition of \mathbf{R}_f that corresponds to the D^p degrees-of-freedom.

Appendix 4: Consistent fltering derivations

Noting that $\delta x = \delta \tilde{d}$ allows us to trivially obtain

$$
\delta\theta(\tilde{d}; \delta\tilde{d}) = \delta\theta^{E}(x; \delta x) - \delta r^{E}(u, w, x; \delta x)
$$
 (61)

from ([32](#page-21-0)). The problem arises because \tilde{d} is an implicit function of d as seen through (17) (17) (17) , i.e., we have $\tilde{d} = d(d)$ and hence, using our abusive notation, we have $\theta = \theta(d) = \theta(\tilde{d}(d))$. We annihilate the implicit variation $\delta d(d; \delta d)$ in ([61\)](#page-24-4) by again applying the adjoint method. The variation of [\(17](#page-7-0)) with respect to d in the direction δd , which equals zero, is subtracted from (61) (61) to give^{[13](#page-24-5)}

$$
\delta\theta(\mathbf{d};\delta\mathbf{d}) = \delta\theta(\widetilde{\mathbf{d}};\delta\widetilde{\mathbf{d}})
$$

=
- $a_c(\delta\widetilde{\mathbf{d}}, \mathbf{w}) + \ell_c(\mathbf{w}, \delta\mathbf{d})$ (62)

Upon requiring the heretofore arbitrary $w \in \mathcal{H}_c$ to satisfy

$$
a_c(\delta \tilde{\mathbf{d}}, \mathbf{w}) = \delta \theta(\tilde{\mathbf{d}}; \delta \tilde{\mathbf{d}})
$$
\n(63)

¹² The fact that $\delta d^p(d^p; \delta d^p) \notin \mathcal{H}_f(0)$ is not an issue here.

¹³ The bilinear form a_c in ([17](#page-7-0)) uses $\int_{\Omega_0} \nabla \mathbf{w} \cdot \gamma \nabla \delta \vec{d} d\gamma$ for the PDE filter. For the nonlinear energy filter replace this integral with $\int_{\Omega_0} \nabla \mathbf{w} \cdot \partial \Psi_f / \partial \widetilde{F}_d[\nabla \delta \widetilde{d}] dv$ and for the linearized energy filter replace it with $\int_{\Omega_0} \nabla \mathbf{w} \cdot \mathbf{A}_f[\nabla \delta \tilde{\mathbf{d}}] d\mathbf{v}$.

for all $\delta \tilde{d} \in \mathcal{H}_c$, [\(62](#page-24-1)) reduces to the desired explicit sensitivity expression

$$
\delta\theta(\mathbf{d};\delta\mathbf{d}) = \mathcal{C}_c(\mathbf{w}, \delta\mathbf{d})\tag{64}
$$

As discussed above, the fact that a_c , ℓ_c and $\delta\theta$ are defined over diferent domains in ([62](#page-24-1)) - ([64](#page-25-21)) is not a concern.

The finite element discretization of ([17](#page-7-0)) yields the equation

$$
\boldsymbol{W}^T((\boldsymbol{K}_c + \boldsymbol{M}_c)\widetilde{\boldsymbol{D}} - \boldsymbol{M}_c \boldsymbol{D}) = 0 \tag{65}
$$

where D, \widetilde{D} , and *W* are the degrees-of-freedom vectors corresponding to d , \tilde{d} , and w , K_c is the symmetric filter stiffness matrix defined such that^{[14](#page-25-22)} $W^T K_c \widetilde{D} = \int_{\Omega_0} \nabla w \cdot \gamma \nabla \widetilde{d} dv$ and M_c is the symmetric filter mass matrix defined such that $W^T M_c D = \int_{\Omega_0} w \cdot d \, dv$. Using the symmetries of K_c and M_c , the adjoint problem (63) (63) is discretized as

$$
\delta \widetilde{\boldsymbol{D}}^T (\boldsymbol{K}_c + \boldsymbol{M}_c) \boldsymbol{W} = \delta \widetilde{\boldsymbol{D}}^T \left(\frac{D\theta}{D\boldsymbol{X}} \right)^T \tag{66}
$$

. The arbitrariness of δD renders the linear adjoint equation $(K_c + M_c)$ $W = (D\Theta / DX)^T$ that we solve for *W*. Finally, the derivative of ([64](#page-25-21)) is computed from $D\Theta/DD = M_c W$.

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Declarations

Conflict of interest On behalf of all authors, the corresponding author states that there is no confict of interest.

Replication of results The source code used to generate the presented results cannot at this time be released as it is proprietary to Lawrence Livermore National Laboratory. However, the authors believe that suffcient information has been provided to allow readers to replicate the presented results.

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¹⁴ Footnote 13 for the necessary modifications to K_c required when using the energy flter and linearized energy flter.

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