Creep Rupture Mechanisms in Annealed and Overheated 7075 Al under Multiaxial Stress States

AHMADALI YOUSEFIANI, FARGHALLI A. MOHAMED, and JAMES C. EARTHMAN

The creep deformation and rupture behavior of annealed and overheated 7075 Al was investigated under uniaxial, biaxial, and triaxial stress states. Examinations of samples prior to and after testing using optical microscopy, scanning electron microscopy (SEM), and transmission electron microscopy (TEM) were also performed to develop a better understanding of the microstructural mechanisms governing this behavior. These observations combined with analyses of the test data indicate that annealed 7075 Al under present testing conditions exhibits characteristics of dislocation creep with a concomitant contribution from grain boundary sliding (GBS). By contrast, the results for overheated 7075 Al suggest that GBS is suppressed. This hypothesis is supported by observations of large particles at grain boundaries in the overheated microstructure and few or no particles at boundaries in the annealed microstructures. Rupture times for the different stress states were also compared with respect to four multiaxial stress parameters, each of which is linked to a particular physical mechanism that can facilitate creep rupture. It was found that creep rupture in annealed 7075 Al (regardless of sample orientation) is dominated by cavitation coupled with GBS. By contrast, the rupture behavior of overheated 7075 Al is consistent with a model that describes cavitation constrained by relatively uniform creep deformation in the matrix. Thus, the rupture findings also indicate that GBS is prevented in the overheated microstructure, while it gives rise to significant stress redistribution in the annealed microstructure.

UNIAXIAL stress conditions have commonly been
employed in comprehensive studies concerning failure of
engineering materials at high temperatures. These investiga-
tions have provided substantial information regarding the
 creep conditions, a fundamental power law exists, which simply relates the time to rupture, t_f , and the applied stress, and through analysis of experimental data, coefficients α or σ , as follows:

$$
t_f = M\sigma^{-\chi} \tag{1}
$$

I. INTRODUCTION t_f utilizing data from conventional uniaxial creep rupture tests. Table I describes several examples of such parameters

 ν are adjusted to obtain the optimum correlation between the rupture time data for different stress states.

where *M* and *X* are stress-independent constants for a given
material and testing condition.
In high-temperature applications, however, the majority
of the creep life of components. However, it should be consid-
of the toward obtaining a representative stress parameter, σ_{rep} and is used to forecast terms, these stress parameters are each associated with a specific set of physical mechanisms that control the creep rupture process. The four different mechanistic parameters AHMADALI YOUSEFIANI, formerly Graduate Student Researcher
with the Department of Chemical and Biochemical Engineering and Materi-
with the Department of Chemical and Biochemical Engineering and Materi-
stress, σ_i ; the als Science, University of California, Irvine, is now Senior Engineer/Scien-
is the constrained growth stress parameter,
tist with the Space and Communications Group, The Boeing Company,
 σ_C , are described briefly in Ta tist with the Space and Communications Group, The Boeing Company, σ_C , are described briefly in Table I. In addition to predicting Huntington Beach, CA 92647-2099. FARGHALLI A. MOHAMED and creep life of a component unde JAMES C. EARTHMAN, Professors, are with the Department of Chemical
and Biochemical Engineering and Materials Science, University of Califor-
nia, Irvine, Ca 92697-2575.
Manuscript submitted December 15, 1999.
Manuscript su ture process. Thus, the ability to identify representative

multiaxial stress parameters from a knowledge of underlying 7075 Al-L) or the long transverse (designated as 7075 Almechanisms offers considerable advantages, in addition to LT) rolling direction. Overheated samples were only eliminating the need for a large number of multiaxial experi- machined with their long axis parallel to the longitudinal ments that are both expensive and time consuming. rolling direction (designated as 7075 Al-OH).

Over the years, high-strength aluminum alloys have suc-

Three types of specimens, each corresponding to a differ-

cessfully been used as structural materials in aerospace,

ID.

Three types of specimens, each correspondi transportation, and many other industries. Their utilization Uniaxial stress state was obtained using conventional smooth for applications requiring elevated temperature performance cylindrical creep specimens ($\sigma_1 = \sigma$, $\sigma_2 = \sigma_3 = 0$). A depends strongly on the understanding of their creep and Bridgman circular notched tensile bar was app rupture characteristics under multiaxial loading conditions. a triaxial state of stress. Finite element calculations reported The present investigation examines the multiaxial creep rup- $\frac{1}{2}$ for this sample geometry under creep conditions^[9] indicate ture behavior of a representative high-strength Al alloy (7075 that the stresses across the notch become uniform very early Al), utilizing a general approach in which the validity of in the life of the specimen. Afterward, the stress state in the the aforementioned mechanistic criteria is determined using Bridgman notch geometry is essentially constant and can be stress state as a variable. Two microstructures were exam-
approximated by $\sigma_1 = 2.7 \sigma_{\text{nom}}$ and $\sigma_2 = \sigma_3 = \sigma_1/3$. The ined, one which results from a relatively standard annealing state of biaxial shear was achieved using a modified double treatment and one that results when the alloy is taken to a shear specimen.^[10] The generated shear stress is highest at substantially higher annealing temperature. The creep deformation and rupture results obtained for the two different microstructures were then used to characterize the mecha- **Table II. Specimen Geometry and Corresponding Stress** nisms controlling high-temperature rupture in the present **States**

The commercial 7075 Al alloy used in the present work **Uniaxial Tension** was received as 4.5-cm-thick rolled plates in the T6 condition. Samples were either annealed at 723 K or overheated $\tau_{\text{max}} \approx F/2A_{\text{min}}$ $\tau_{\text{max}} \approx F/2A_{\text{min}}$ at 823 K for 4 hours, air cooled to room temperature, and $\sigma_1 = \tau_{\text{max}}$
learnt there for at least 1 week before testing. Crean tests $\sigma_2 = 0$, $\sigma_3 = -\tau_{\text{max}}$ kept there for at least 1 week before testing. Creep tests **Biaxial Shear** were conducted in air under isothermal/isostress conditions, with the temperature maintained at 648 ± 2 K. Annealed
samples were machined along two orientations, with their
long axis parallel to either the longitudinal (designated as
 $\sigma_2 = \sigma_3 = 0.33 \sigma_1$

ent stress state, were utilized in this investigation (Table II). Bridgman circular notched tensile bar was applied to produce

the minimum cross-sectional area of the notches. Assuming no bending moment is present in the notches, and stress concentrations are relaxed early in the life of the specimen, the stress state for this specimen may be characterized by $\sigma_1 = -\sigma_3 \approx \tau_{\text{max}}$ and $\sigma_2 = 0$.^[10,11]

Microstructural examinations were conducted through the use of both optical and electron microscopy. Optical microscopy was utilized to reveal (1) grain size/morphology and particle size/distribution in 7075 Al (as-received and heattreated conditions) and (2) the extent/distribution of cavities in samples tested to failure. Samples were polished using conventional metallographic techniques and etched using Keller's reagent where necessary. Transmission electron microscopy (TEM) examinations, which were performed to investigate substructural features in the alloy, were conducted using a PHILIPS* CM20 transmission electron

PHILIPS is a trademark of Philips Electronic Instruments Corp., Mahwah, NJ.

microscope operating at 200 kV. Thin foils were mechanically polished to about 100 μ m and twin-jet polished using a 25 pct nitric acid-75 pct methanol solution, maintained below 240 K. Scanning electron microscopy (SEM) was used to provide information regarding (1) the morphological features of the fracture surfaces, (2) cavity nucleation sites, (3) chemical compositions of second-phase particles, and (4) grain structure and precipitate-free zones (PFZs) in the alloy. The SEM investigations were performed using a PHIL-IPS XL30 scanning electron microscope, with an energy dispersive spectroscopy (EDS) attachment.

III. EXPERIMENTAL RESULTS

A. *Microstructure*

Figure 1 illustrates and compares the microstructural features observed in 7075 Al in the as-received (top row), annealed (middle row), and overheated (bottom row) conditions. The triplanar optical micrographs show the etched (left column) and polished (right column) microstructures in the longitudinal and transverse directions.

ished surface of the alloy in the as-received condition, reveals the size and distribution of the insoluble second-phase con- $(FeMnCr)_{3}SiAl_{12}.$ ^[12,13]
The two types of heat treatments conducted prior to testing

resulted in remarkably different microstructures (Figures ure $1(f)$) and a dense population of large precipitates at the $1(c)$ through (f)). While both treatments (annealing and over-grain boundaries. $1(c)$ through (f)). While both treatments (annealing and overheating) served the common purposes of relieving strain-
Resolving grain boundaries in the annealed microstructure

Partially recrystallized grains, elongated in the rolling Fig. 1—Triplanar optical micrographs showing microstructural features direction, are typical of the alloy in the T6 (as-received) condition (*a*) etched and (*b*)

stituent particles. During prior working operations, particles and the elimination of dislocation substructures. Figure 1(c) break down and roughly align as stringers in the rolling shows the etched microstructure after annealing. Due to the direction. The EDS analysis indicated that the majority of dense precipitation that resulted from annealing, no grain these particles, with sizes ranging between 2 and 10 μ m, contrast could be produced through etching. Figure 1(d) are Fe-bearing phases; however, Si-bearing particles were indicates that annealing results in the dissolution of existing also identified. Due to its very low solid solubility, most of phases and subsequent precipitation of slightly coarser (3 the Fe present within the Al matrix appears as intermetal- to $15 \mu m$), more homogeneously distributed second-phase lic second-phase particles (in combination with other ele- constituents compared to those found in the as-received conments). Insoluble phases that have been observed in 7075 dition. By contrast, overheating 7075 Al produces a fully Al are mainly A_1 , Cu_2Fe , A_2 , $CuFe$, Al_6Fe , Mg_2Si , and recrystallized, coarse-grained microstructure (Figure 1(e)), with a completely heterogeneous distribution of coarse (5) to 25 μ m) second-phase constituents within the grains (Fig-

hardening effects, inducing recrystallization, and stabilizing was not possible using optical microscopy. However, the the microstructure, overheating also resulted in grain growth presence of relatively wide PFZs and constituent particles

itates, and PFZs in (*a*) 7075 Al-L, $\varepsilon = 0.137$, $\sigma = 10.5$ MPa; and (*b*) 7075 appreciable difference in the (sub)grain structure was Al-OH, $\varepsilon = 0.140$, $\sigma = 10.3$ MPa. Representative samples shown in this observed T

ble to observe grain size and morphology by SEM examina-
tion prior to creep deformation^[14] or by dynamic/continuous
tion in the backscatter mode following creep deformation
recrystallization occurring during the early tion in the backscatter mode following creep deformation recrystallization occurring during the early stages of creep
(Figure 2). Measurements on representative annealed sam-
deformation.^[15,16] The exact determination (Figure 2). Measurements on representative annealed sam-
ples, deformation of the process
ples, deformed to approximately 15 pct at different stress
that has led to the development of a fine grain structure in the ples, deformed to approximately 15 pct at different stress that has led to the development of a fine grain structure in the levels, revealed a recrystallized, fairly equiaxed grain mor-
annealed samples is beyond the scope phology, with an average spatial grain size d ($d = 1.74L$, However, it seems that continuous recrystallization takes where \overline{L} is the mean linear intercept) of $10 \pm 1 \mu$ m. The place dynamically in the very early s where *L* is the mean linear intercept) of 10 ± 1 μ m. The place dynamically in the very early stages of creep deforma-
size of the constituent particles ranged from about 0.1 to tion based on the observations made in 0.3 μ m, and the average width of the PFZs was determined probably the subgrains (with low angle boundaries) present to be 0.5 \pm 0.1 μ m. The overheated samples exhibited an in the alloy coarsen as creep deformatio average equiaxed grain size of 80 \pm 5 μ m, PFZs with an similar to observations made elsewhere,^[15,16] the subgrains average width of 3 ± 0.5 μ m, and platelets of constituent increase in misorientation, due to concurrent slip and/or particles decorating the grain boundaries and varying in size grain boundary sliding (GBS), until th from 0.8 to 6 μ m along the long axis. \qquad angle boundaries is complete.

The TEM micrographs included in Figure 3 illustrate and compare the substructural features observed in the annealed

(top row) and overheated (bottom row) conditions. Figures B. *High-Temperature Deformation* 3(a) and (b) indicate that annealed 7075 Al contains the Due to the complexities introduced by the notches, analyfollowing: (1) a well-defined relatively elongated subgrain sis of creep deformation in the biaxial and triaxial specimens structure; (2) small amounts of Cr-bearing dispersoids, most has not been included in the present investigation. However, likely $Al_{18}Mg_3Cr_2$;^[12] and (3) rodlike MgZn₂, which resulted the standard data obtained under uniaxial conditions have from the coarsening (during annealing) of the Guinier-
been used to study and analyze creep be Preston (G–P) zones present in the T6 condition. In contrast, Typical examples of creep curves obtained under uniaxial Figures 3(c) and (d) reveal the absence of subgrains and conditions are shown in Figure 4, where the true tensile the presence of much coarser precipitates in the overheated strain, ε , is plotted against time, *t*. The applied stress ranged

condition. Dispersoids, such as those observed in the annealed condition, are present intentionally to control the grain structure. They are virtually insoluble and resistant to coarsening under normal heating conditions. However, the overheating treatment, which takes place at 823 K, is well above conventional solutionizing temperatures and into the region of incipient melting (805 K). This is expected to have a pronounced effect on the microstructure of the alloy and may have led either to the dissolution or to significant coarsening of the Cr-bearing dispersoids, which is most likely the reason for both the absence of dislocation substructures and the significant grain growth for the overheated condition. The morphological differences in the constituent particles observed using SEM (Figure 2) show similar characteristics. The particles distributed along the boundaries in the annealed condition (Figure 2(a)) are relatively small and well rounded (representing the Cr-bearing dispersoids) on some boundaries and nonexistent on others. Constituent particles decorating the grain boundaries in the overheated condition (Figure 2(b)) are much coarser and plateletlike (corresponding to coarsened $MgZn₂$ precipitates) and appear to be uniformly present on all boundaries.

The preceding results illustrate the microstructural similarities and differences observed in 7075 Al under various heat treatments. However, it is important to further examine the fine grain structure developed as a result of annealing. As noted earlier, revealing the grain structure of the annealed samples requires slight creep deformation. The TEM observations indicate that annealing alone has very little influence on the as-received microstructure. Its effect seems to be limited to the precipitation of dissolved constituents, and Fig. 2—SEM micrographs revealing grain structure, grain boundary precip- even though recrystallization may have taken place, no Al-OH, $\varepsilon = 0.140$, $\sigma = 10.3$ MPa. Representative samples shown in this observed. Therefore, the evolution of the fine-grained struction of the fine-grained struction of the fine-grained struction of the fine-grained st result of static recrystallization. Following appropriate thermomechanical treatments, it is possible to produce a fine located preferentially on the grain boundaries made it possi-
ble to observe grain size and morphology by SEM examina-
tion prior to creep deformation^[14] or by dynamic/continuous annealed samples is beyond the scope of this investigation. tion based on the observations made in this study. Most in the alloy coarsen as creep deformation proceeds, and grain boundary sliding (GBS), until the development of high

been used to study and analyze creep behavior in 7075 Al.

Fig. 3—TEM micrographs revealing the substructural features observed in (*a*) annealed 7075 Al, (*b*) annealed 7075 Al at high magnification, (*c*) overheated 7075 Al, and (*d*) overheated 7075 Al at high magnification. Arrows indicate particle-free boundaries present in the annealed microstructure.

Fig. 4—Typical creep curves showing true tensile strain, ε , as a function of time, *t*. Samples were tested under uniaxial conditions.

from 8 to 30 MPa, covering close to four orders of magnitude high stresses. Strain rate, $\dot{\epsilon}$, is plotted against true tensile strain, ϵ . of strain rate. It should be noted that the instantaneous strain, which generally increased with increasing applied stress, was subtracted from the total strain. Therefore, the creep Figure 5), where applicable, minimum creep rate has been and represent only the strain due to creep. Figure 5, in samples tested at low and relatively high stresses, providing in general, all creep curves exhibit a normal primary stage state behavior at the highest stress levels tested. $(d\dot{\epsilon})dt < 0$), followed by a very limited secondary stage of The differences in creep ductility become apparent in steady-state creep rate $(d\dot{\epsilon}/dt = 0)$, and an extended tertiary Figure 6, where the rupture strain, ϵ_f , stage ($d\dot{\varepsilon}/dt > 0$) that accounts for most of the sample life.

Fig. 5—Results of uniaxial tests conducted on samples at low and relatively

curves shown in Figure 4 start at the origin of the coordinates used instead of steady state. Examination of the creep curves and represent only the strain due to creep. Figure 5, in in Figure 5, as well as those not shown which ε is plotted against strain rate, $\dot{\varepsilon}$, includes results from of a very sharp creep-rate minimum in 7075 Al-OH at all samples tested at low and relatively high stresses, providing stress levels. While 7075 A additional means of interpreting and comparing the creep similar, but less accentuated, creep-rate minimum at the behavior under different conditions. Regardless of the type lowest stress levels, they generally exhibit a smooth secondof heat treatment or sample orientation, results indicate that, ary stage transition and approach an almost ideal steady-

Figure 6, where the rupture strain, ε_f , is plotted against the applied stress. While creep ductility remains relatively Due to the limited extent of the secondary stage (evident in constant with increasing stress levels (increasing slightly at

Fig. 6—Rupture strain, ε_f , as a function of the applied stress, σ , for samples tested to failure under uniaxial conditions.

7075 Al-L and 7075 Al-LT show an increase in ductility, L and 7075 Al-LT, which show identical stress dependencies. which reaches a maximum and ultimately decreases at high In order to investigate the occurrence of dislocation activstresses. The overall creep ductility is highest in 7075 Al-

Let us during creep deformation, additional uniaxial tests were

conducted at intermediate stress levels. The tests were termi-

for practical purposes can be described in the following ment. Thin foils were prepared parallel to the tensile axis simple form (Norton's law):
from regions located in the most highly deformed sections

$$
\dot{\varepsilon}_{\min} = A \sigma^n \tag{8}
$$

stant, and *n* is the stress exponent for creep, normally ranging location jogs, loops, and dipoles in the interior of the subfrom 3 to 8, depending on the type of alloy and the governing grains. Figure 8(b), at very high magnification, reveals arrays creep mechanism. Figure 7, in which ε_{\min} has been plotted of dislocations forming simple sub-boundaries.

Fig. 8—TEM micrographs representing typical substructural observations made in 7075 Al. (*a*) Dislocation networks, and dislocation jogs, loops, and dipoles in the interior of the subgrains. (*b*) Arrays of dislocations forming simple sub-boundaries. The sample was tested under uniaxial tension ($\varepsilon \approx 0.2$ and $\sigma = 12.5$ MPa).

against σ on a double logarithmic scale, can be utilized to investigate the creep behavior in 7075 Al. The straight fit through the data obtained at 648 K indicates that creep behavior in the alloy, over the range of strain rates investigated, obeys Eq. [8]. Regardless of heat treatment or sample Fig. 7—The stress dependence of minimum creep rate for 7075 Al tested gated, oveys Eq. [8]. Regardless of heat treatment or sample orientation, results show that the value of the stress exponent at 648 K. $n = (\partial \ln \varepsilon_{\text{min}}/\partial \ln \sigma)_T$ for 7075 Al is approximately 5.3. Results also indicate that creep strength in 7075 Al-OH is the highest stress) for 7075 Al-OH, it is evident that both almost an order of magnitude higher than that of 7075 Al-

followed by 7075 Al-LT and 7075 Al-OH. conducted at intermediate stress levels. The tests were termi-
The stress dependence of minimum (or steady-state) creep nated at $\varepsilon \approx 0.2$ and the representative samples were coole The stress dependence of minimum (or steady-state) creep nated at $\varepsilon \approx 0.2$, and the representative samples were cooled rate can often be well approximated by a power law, which rapidly under load to preserve any substr rapidly under load to preserve any substructural developfrom regions located in the most highly deformed sections of the specimens. The TEM micrographs shown in Figure 8 represent the typical observations made in the alloy. Figure where *A* is a temperature- and microstructure-dependent con- 8(a) indicates the presence of dislocation networks, and dis-

for samples tested under uniaxial tension.

The creep-damage tolerance, $\lambda = \varepsilon_f/(\varepsilon_{\text{min}} t_f)$, is a parameter used to measure the tolerance of a creeping material with used to measure the tolerance of a creeping material with Neglecting σ_0 , which is reasonable due to the relatively large respect to strain concentrations.^[2] It has been shown that cavity sizes observed in 7075 Al. respect to strain concentrations.^[2] It has been shown that cavity sizes observed in 7075 Al, and substituting for the failure occurs by creep cavitation for $1 < \lambda < 2.5$, and effective strain rate gives the growth rate microstructural degradation becomes the dominant damage Assuming only a small fraction of t_f is involved in the final mechanism. At intermediate levels, the likelihood of an inter-
stages of failure (microcrack interlin mechanism. At intermediate levels, the likelihood of an inter-
action between the two mechanisms exists. Such a situation the rupture time will be governed by the cavity growth rate. has been investigated in detail,^[17] using a two-state-variable damage approach. In this way, λ may be used to identify mechanisms controlling creep rupture. Uniaxial creep rup-
ture data obtained for 7075 Al are used in Figure 9 to plot
derived for situations where cavitation is coupled with highly ture data obtained for 7075 Al are used in Figure 9 to plot the damage tolerance parameter against the applied stress. localized deformation processes, such as GBS.^[20] During It can be seen in this figure that the data for 7075 Al-L and sliding, shear stresses on inclined bounda It can be seen in this figure that the data for 7075 Al-L and sliding, shear stresses on inclined boundaries are relieved, 7075 Al-LT fall within the $1 < \lambda < 2.5$ (shaded) region. giving rise to a redistribution of stresse 7075 Al-LT fall within the $1 < \lambda < 2.5$ (shaded) region. giving rise to a redistribution of stresses to transverse bound-
The λ values for 7075 Al-OH, on the other hand, range from aries that cavitate leading to rupture The λ values for 7075 Al-OH, on the other hand, range from aries that cavitate leading to rupture. The average tensile a minimum of 2.6 at lower stresses to an average of 4.2 at stress on grain boundary facets perpendi intermediate and high stresses. Hence, λ values indicate that fore amplified, and it is suggested that this enhanced stress 7075 Al-L and 7075 Al-LT predominantly fail by creep level drives the rupture process. cavitation, while other modes of microstructural degrada-

High-temperature rupture data for the alloy are presented

in Figure 10. The plots compare each of the representative tion, such as particle coarsening, have a significant role in in Figure 10. The plots compare each of the representative the rupture of 7075 Al-OH.

opposed to σ_e . On the other hand, the better representative

pct C steel, and a Mg alloy tested at 473, 723, and 293 K, respectively. The samples in this case exhibit practically no cavitation other than that observed in the close vicinity of the rupture surface. However, it has been observed over time that the most accurate representative stress parameters depend on both σ_1 and σ_e .^[4,6–8] Nevertheless, these parameters symbolize extremes of material behavior and were consequently considered in the present investigation.

The constrained growth stress parameter, which is based on a cavity growth model developed originally by Rice,^[19] and later modified by Riedel, $[1]$ can be utilized for conditions where grain boundary cavitation is constrained by the continuum creep rate of the surroundings. In the constrained limit, the model estimates the cavity growth rate by

$$
\dot{R} = 0.13 \dot{\varepsilon}_e a \left(\frac{b}{R}\right)^2 \frac{\sigma_1 - (1 - \omega)\sigma_0}{\sigma_e} \tag{9}
$$

where R is the radius of the cavities (assumed to have all nucleated upon loading) with a uniform spacing equal to *b*, Fig. 9—Damage tolerance parameter, λ , as a function of the applied stress *a* is the diameter of a cavitating grain boundary facet, ω = $(2R/b)^2$ is the area fraction of cavitated grain boundary, $\dot{\epsilon}_e = A \sigma_e^n$ is the effective strain rate in the remote creeping material, *A* is a temperature- and microstructure-dependent C. *High-Temperature Rupture*

The creep-damage tolerance, $\lambda = \varepsilon_f/(\varepsilon_{\text{min}}t_f)$, is a parameter since energy and ψ is the sintering stress (γ_s is the surface surface stress (γ_s is the surface stress). failure occurs by creep cavitation for $1 < \lambda < 2.5$, and effective strain rate gives the growth rate in the following when λ takes on larger values (commonly greater than 5), simplified form: $\dot{R} = B\sigma_1 \sigma_e^{n-1}$, where $B = 0.13Aa(b/R)^2$. the rupture time will be governed by the cavity growth rate. Therefore, it follows that $t_f \propto (\sigma_1 \sigma_e^{n-1})^{-1}$ under conditions of constrained cavity growth, and $\sigma_C = \sigma_1 \sigma_e^{n-1}$ (Table I).
The principal facet stress parameter, σ_F (Table I), was

stress on grain boundary facets perpendicular to σ_1 is there-

the rupture of 7075 Al-OH.
Results of early multiaxial creep rupture studies^[18] con-
logarithmic scale. Using the general form of Eq. [1], a regreslogarithmic scale. Using the general form of Eq. [1], a regresducted on a variety of metals and alloys suggested that creep sion analysis was applied to calculate the least-squares fit life under multiaxial stress states can generally be predicted through the rupture data fields for all three stress states. The by either σ_1 or σ_e . For example, copper, NIMONIC* 75, best fit and the corresponding correlation coefficients, as shown in the figure, can be used to accurately determine *NIMONIC is a trademark of INCO Alloys International, Inc., Hunting- the validity of each representative stress parameter. The ton, WV. parameter that brings the data closest to a single line would and Mo steel were tested under multiaxial stresses at 523, therefore be best for predicting lifetime under multiaxial 923, and 823 K, respectively. It was found that the specimens stress states. The superior correlation of the principal facet progressively develop a homogeneous distribution of cavit-
stress parameter for 7075 Al-L $(R^2 = 0.96)$ and 7075 Alies at a level microscopically visible from the onset of testing. LT $(R^2 = 0.95)$ is clearly evident in Figure 10. However, it The rupture times for these samples were correlated better can be seen that $\sigma_C (R^2 = 0.96)$ correlates the rupture time by σ_1 as the representative stress parameter in Eq. [1], as best in the case of 7075 Al-OH, brin by σ_1 as the representative stress parameter in Eq. [1], as best in the case of 7075 Al-OH, bringing all of the rupture opposed to σ_e . On the other hand, the better representative data onto a single curve. It is im parameter was determined to be σ_e for Al alloy RR59, 0.2 the slope of the fit through the data is approximately -1 in

Fig. 10—High-temperature multiaxial rupture data for 7075 Al. The plots compare each of the representative stress parameters with respect to ruptur e time on a double logarithmic scale. The best fit and the corresponding correlation coefficients are also included.

tions, illustrates and compares the cavitational features similar stress levels. Similar observations and trends can be and 7075 Al-OH (bottom row). The influence of stress level compare 7075 Al-L (left column) and 7075 Al-OH (right on cavitation behavior of the alloy may also be investigated column) tested under triaxial tension (top row) and biaxial by comparing samples tested at high stresses (left column) shear (bottom row). Samples chosen for Figure 12 were with those tested at low stresses (right column). Each image tested at intermediate levels (within the range of applied is a montage of several optical micrographs, covering a large stresses for each state) and represent the general observaarea of the polished cross section in the vicinity of the rupture tions. It should be noted that the change in cavity morphology surface. With this perspective, both the deformation profile observed might be assumed to have originated from a transiand cavitating regions away from the tip can be studied tion in the dominant high-temperature rupture mechanism. and compared. Examination of Figure 11 reveals that (1) However, the results shown in Figure 10 indicate that such widespread cavitation occurs in all samples and is not limited an assumption is unfounded, since each of the corresponding to regions near the rupture surface; (2) the extent of cavita-
tion decreases with increasing stress levels; (3) cavity distri-
over the entire stress range, depending on the sample clearly visible; (4) cavity morphology changes at higher being dominant for a certain fraction of the creep life.^[2] stresses, appearing more elongated in the direction of the While a certain mechanism, governed by the correlating tensile axis; and (5) necking becomes more prominent with parameter, may predominate most of the rupture time, the

all conditions using σ_C , the parameter fails to satisfactorily increasing stress level. It is also evident that the overall correlate the data for 7075 Al-L and 7075 Al-LT. extent of cavitation damage in 7075 Al-OH (Fi extent of cavitation damage in 7075 Al-OH (Figures 11(e) Figure 11, which shows the polished longitudinal cross and (f)) is significantly greater than that in 7075 Al-L and sections of specimens tested to rupture under uniaxial condi- 7075 Al-LT (Figures 11(a) through (d)) when compared at observed in 7075 Al-L (top row), 7075 Al-LT (middle row), noted for the optical micrographs shown in Figure 12, which over the entire stress range, depending on the sample bution is more homogeneous at lower stresses, with blunt (annealed or overheated). During creep rupture, different microcracks that developed due to cavity coalescence being mechanisms typically occur in a coupled manner, with each

Fig. 11—Polished longitudinal cross sections of specimens tested to rupture under uniaxial conditions. (*a*) 7075 Al-L, σ = 19.25 MPa; (*b*) 7075 Al-L, $\sigma = 8.86$ MPa; (*c*) 7075 Al-LT, $\sigma = 21.60$ MPa; (*d*) 7075 Al-LT, $\sigma = 8.87$ MPa; (*e*) 7075 AL-OH, $\sigma = 21.70$ MPa; and (*f*) 7075 Al-OH, $\sigma = 10.54$ MPa. Each image is a montage of several optical micrographs, covering a large area near the rupture surface.

strain to failure and/or the final rupture mode may be facili- and (b), 7075 Al-OH in Figure 13(c), and 7075 Al-LT in tated by a different process. In most cases of extensive Figure 13(d)) indicate that the majority of cavities form at cavitation, void growth by creep deformation only becomes large second-phase constituent particles. The representative

dominant in the final stages of rupture.

SEM micrographs included in Figures 13(a) through (d)

Detailed examination of the polished longitudinal cross

Show samples tested at intermediate stress levels, with Detailed examination of the polished longitudinal cross show samples tested at intermediate stress levels, with sections of specimens tested to rupture were conducted at regions of examination located at least 1.5 mm away regions of examination located at least 1.5 mm away from relatively high magnification using SEM. Typical observa- the rupture surface. These figures illustrate the following tions made in different samples (7075 Al-L in Figures 13(a) general observations: (1) aligned cavities, which had formed

Fig. 12—Polished longitudinal cross sections of specimens tested to rupture under biaxial and triaxial conditions. (*a*) 7075 Al-L, triaxial tension, $\sigma_{\text{nom}} =$ 6.89 MPa; (*b*) 7075 Al-OH, triaxial tension, $\sigma_{\text{nom}} = 6.86$ MPa; (*c*) 7075 Al-L, $\tau_{\text{max}} = 7.99$ MPa; and (*d*) 7075 Al-OH, $\tau_{\text{max}} = 7.99$ MPa. Each image is a montage of several optical micrographs, covering a large area near the rupture surface.

along particle stringers, were frequently observed in all sam-
tested to failure at $\sigma = 10.5$ MPa is shown in Figure 15. ples (Figure 13(a)); (2) regardless of the type of heat treat- Particles located inside a dimple are clearly visible in this ment, a limited amount of cavities were found to have SEM micrograph. nucleated at cracks most likely formed in particles during primary processing (Figure 13(b)); (3) the majority of the cavities observed specifically in 7075 Al-OH had nucleated **IV. DISCUSSION** as a result of particle/matrix debonding (Figure 13(c)); and A. *High-Temperature Deformation* (4) most cavities observed in 7075 Al-L and 7075 Al-LT were found to have nucleated at particle/matrix interfaces The results obtained in the present investigation imply

the aforementioned cavity-nucleating particles. The presence either Al₇Cu₂Fe or Al₂₃CuFe₄, and those represented by the the mechanism controlling creep deformation in the alloy.

Results obtained in numerous investigations^[22,23,24] indi-

(bottom row), intermediate (middle row), and high (top row) from an elevated testing temperature ($T > 0.5$ T_m , where applied stresses, illustrates and compares the features T_m is the melting point), the main requirement for the occurobserved in 7075 Al-L (left column), 7075 Al-LT (middle rence of superplasticity (the ability of materials to exhibit column), and 7075 Al-OH (right column). The figure shows extensive neck-free elongations) is a stable grain size, usuthat regardless of heat treatment or sample orientation (1) ally less than 10 μ m. On this basis, samples tested in the dimpled nature of the rupture surfaces; (2) the average dimple size increases with decreasing stress level; and (3) the characteristic of creep deformation in superplastic alloys is increasing stress level. When compared at equivalent stress between the applied stress and creep rate, which is distin-OH is significantly greater than that for 7075 Al-L and 7075 region I (the low-stress region, with *n* typically greater than

(Figure $13(d)$). It is important to note that cavities were also that, under existing testing conditions and regardless of heat located at the shoulders of samples tested to failure. treatment or sample orientation, 7075 Al seems to exhibit The EDS analysis was conducted on a large number of the characteristics of dislocation creep.^[21] This is demon-
eaforementioned cavity-nucleating particles. The presence strated by the following observations, which are of Fe was the common feature in spectrums obtained from of all three sample types studied: (1) the presence of both an all particles, and generally, regardless of the type of heat instantaneous strain upon loading and a relatively extensive treatment, three types of spectrums were prominent (Figures decelerating primary stage, (2) a creep stress exponent of 13(e) through (g)). The majority of the particles in all sam- approximately 5.3, and (3) the dislocations tending to form ples were represented by the spectrum shown in Figure 13(e). simple sub-boundaries and dislocation arrays during creep. Based on available literature, $^{[12,13]}$ it appears that the particles Despite these correlations, there are issues that require furrepresented by spectrums shown in Figures 13(e) and (f) are ther examination before a conclusion can be drawn regarding

ectrum shown in Figure 13(g) are (FeMnCr)₃SiAl₁₂. Results obtained in numerous investigations^[22,23,24] indi-
Figure 14, which shows SEM micrographs of the rupture cate that suitably processed 7000 series Al alloys cate that suitably processed 7000 series Al alloys possess a surfaces of uniaxial specimens tested to failure under low degree of superplasticity during creep deformation. Apart all alloys fail in a ductile manner, as characterized by the annealed condition ($T = 648 \text{ K} > 0.5 \text{ T}_m$ and $d = 10 \text{ }\mu\text{m}$) dimpled nature of the rupture surfaces; (2) the average dim-
would be expected to behave sup depth and ridge sharpness of the dimples increases with the experimental observation of a sigmoidal relationship levels, it is evident that the average dimple size in 7075 Al-
guished by the presence of three regions of behavior:^[25,26] Al-LT. Also, it is important to add that closer examination 3), region II (the intermediate-stress or superplastic region of the interior of many dimples (at higher magnification) where maximum ductility is observed; with $n < 2.5$), and revealed the second-phase constituent particles responsible region III (the high-stress region, with $n > 3$). The absence for cavity nucleation. An example of a 7075 Al-OH sample of such a sigmoidal trend is apparent in 7075 Al under

Fig. 13—Representative SEM micrographs showing cavity nucleation at
second-phase constituent particles in (a) 7075 Al-L, (b) 7075 Al-L, (c)
7075 Al-OH, and (d) 7075 Al-LT. Samples were tested at intermediate
81 stress elev

present experimental conditions ($n \sim 5.3$ over the entire stress range (Figure 7)). This is further demonstrated in Figure 16, where the creep data obtained elsewhere for 7075 $\Delta I(d = 12 \ \mu m)^{[24]}$ and 7475 Al ($d = 11 \ \mu m$)^[22] are compared with those obtained for annealed 7075 Al in the present investigation. Examination of Figure 16 reveals a pattern, which can be utilized to explain the behavior observed in annealed 7075 Al. It is clearly evident that the slope of region II increases with decreasing testing temperature, and apparently, region III is extended to lower stress levels at the expense of region II. Ultimately, at the lowest testing temperature, only region III behavior is observed. Unlike deformation in region II, where creep characteristics appear to be consistent with the predictions of models that are based on GBS accommodated by some form of diffusion process,[27–31] deformation in region III is largely facilitated by dislocation creep mechanisms.^[32] Therefore, the dislocation creep characteristics observed in the present study may be interpreted on this basis. However, it should be noted that the results of GBS measurements^[32] conducted on various superplastic materials indicate that GBS still contributes almost 30 pct of the total strain in region III. Thus, GBS still has an active role in affecting creep deformation when $n > 3$.

The preceding discussion suggests that the deformation behavior of 7075 Al is consistent with the characteristics of region III of the sigmoidal relation between applied stress and creep rate. Further support for this suggestion is provided by the following correlation. It has been demonstrated that for both superplastic Pb-62 wt pct Sn and Zn-22 wt pct Al alloys, the transition from the superplastic region II to the high stress region III occurs when the grain size of the material, *d*, is equal to the subgrain size δ ^[33] As reported elsewhere, $[21,34]$ the subgrain size is dependent on the applied stress through an experimental relationship of the following form:

$$
\frac{\sigma}{G} = 20 \left(\frac{\delta}{\mathbf{b}} \right)^{-1} \tag{10}
$$

where *G* is the shear modulus and **b** is the burgers vector. Using $\delta = d = 10 \ \mu \text{m}$, $\mathbf{b} = 2.86 \times 10^{-8} \text{ cm}$, and the value of *G* for Al at 648 K leads to $\sigma = 11$ MPa, a value which is very close to the lowest value of applied stress used in the present investigation on 7075 Al.

Based on the present microstructural observations (Figure 2), it is likely that the dense population of coarse and plateletlike precipitates, which uniformly decorate the grain boundaries in 7075 Al-OH, would inhibit boundary sliding at the relatively low testing temperatures utilized in the present investigation. The influence of GBS on creep deformation may be characterized by the stress enhancement factor, *f*. In the presence of GBS, the creep rate of the bulk material can be expressed as

$$
\dot{\varepsilon}_{\min} = A(f\sigma)^n \tag{11}
$$

from the rupture surface. (*e*) to (*g*) EDS spectrums for cavity nucleating structure where sliding does not occur. In the absence of particles in 7075 Al. (**GBS** (7075 Al-OH), the bulk material deforms uniformly GBS (7075 Al-OH), the bulk material deforms uniformly $(f = 1)$ and the minimum creep rate is given by Eq. [7]. The results in Figure 7 indicate that $f = 1.48$ for annealed

Fig. 14—SEM micrographs of the rupture surfaces of uniaxial specimens tested to failure under low, intermediate, and high applied stresses. (*a*) 7075 Al-L, $\sigma = 8.86$ MPa; (*b*) 7075 Al-LT, $\sigma = 8.87$ MPa; (*c*) 7075 Al-OH, $\sigma = 8.89$ MPa; (*d*) 7075 Al-L, $\sigma = 12.21$ MPa; (*e*) 7075 Al-LT, $\sigma = 13.64$ MPa; (*f*) 7075 Al-OH, $\sigma = 12.24$ MPa; (*g*) 7075 Al-L, $\sigma = 29.7$ MPa; (*h*) 7075 Al-LT, $\sigma = 29.7$ MPa; and (*i*) 7075 Al-OH, $\sigma = 25.5$ MPa.

7075 Al assuming that GBS is prevented in the overheated microstructure as a result of the observed particles on the boundaries.

Several researchers have studied the effects of GBS on overall creep rates using finite element methods.[35,36,37] For example, Gharemani^[35] developed a two-dimensional (2-D) model of hexagonal grains deforming by steady-state creep and GBS. This model predicts a value of *f* equal to 1.2 for $n = 5$. Dib and Rodin^[36] developed a three-dimensional (3-D) steady-state creep model that incorporates Wigner–Seitz cells as grains. Their results also predict a stress enhancement factor of 1.2 for a statistical array of grains and $3 \le n \le$ 8. Similarly, Chakraborty and Earthman^[37] also calculated $f = 1.2$ for $n = 5$ with a hexagonal grain model of both primary and secondary creep deformation in conjunction with GBS. Thus, $f \approx 1.2$ is uniformly predicted by several
models of dislocation creep coupled with GBS for both 2-
D and 3-D grain configurations.
with GBS for both 2-
micrograph at higher magnification revealing the in

The present experimental results shown in Figure 7 imply a value of *f* equal to approximately 1.5 if GBS is suppressed in the overheated microstructure. Therefore, it appears that and overheated microstructures cannot be completely the difference in minimum creep rate between the annealed accounted for by the effect of GBS, as predicted by finite

Al $(d = 12 \ \mu m)^{[24]}$ and 7475 Al $(d = 11 \ \mu m)^{[22]}$ and those obtained for annealed 7075 Al in the present investigation.

element techniques. However, it should be noted that none LT clearly fall within the $1 < y < 2.5$ region, and σ_1 should of the aforementioned models allow for any cavities at the sused to predict multiaxial runture. Sec of the aforementioned models allow for any cavities at the be used to predict multiaxial rupture. Second, the γ values triple junctions constraining GBS. Chakraborty and Earth-
for 7075 ALOH lie above the cavitation-co triple junctions constraining GBS. Chakraborty and Earth-
man^[38] have developed finite element creep models that but are loss than 5, which represents the typical limit above

mot play a significant role in 7075 Al-OH. Bearing this in lates the rupture lates the rupture times for annealed 7075 Al to a correlation is now focused on high-temperature rupture lates the rupture times for annealed 707 coefficient of 0.95, bringing the data for three stress states behavior in 7075 Al.

present investigation indicate that cavitation damage is the tation in the alloy (Figure 11 through 14) also lends strong dominant failure mechanism in 7075 Al. Optical microscopy support for the validity of these paramete dominant failure mechanism in 7075 Al. Optical microscopy support for the validity of these parameters. The success of (Figures 11 and 12) revealed that, regardless of heat treat- σ_F implies that cavitation in 7075 Al-L (Figures 11 and 12) revealed that, regardless of heat treatment, sample orientation, stress level, or stress state, exten- coupled with either GBS or other localized deformation sive cavitation occurred in all samples and was not limited processes along inclined directions such as rapid creep

to the vicinity of the fracture surfaces. Scanning electron microscopy, which was utilized primarily to identify cavity nucleation sites (Figures 13 and 15), also revealed cavities located at the shoulders of samples tested to failure (experiencing \sim 25 pct of the applied stress), providing further evidence for the dominance of cavitation damage. The dimpled nature of the fracture surfaces (Figure 14) also indicates that failure resulted from extensive cavitation.

Early studies^[18] suggested that creep life under multiaxial stress states can be predicted using either σ_1 , when cavities nucleate readily and distribute homogeneously on all grain boundaries, or σ_e , when failure is primarily due to changes in the internal dislocation structure. Accordingly, the creepdamage tolerance parameter (Section III–C) was thought to provide a simple means of predicting multiaxial creep rupture characteristics.^[39] For example, when $1 < \lambda < 2.5$, it was proposed that creep rupture should be controlled by diffusive cavitation making σ_1 the appropriate representative stress parameter. Furthermore, substructural softening is thought to become dominant when λ takes on larger values Fig. 16—A comparison between creep data obtained elsewhere for 7075 ($\lambda > 5$), and σ_e should be considered as the correlating Al ($d = 12 \text{ mm}^{[24]}$ and 7475 Al ($d = 11 \text{ mm}^{[22]}$) and those obtained for stress paramete shown in Figure 9 would lead to the following expectations. First, creep rupture in annealed 7075 Al is due to diffusioncontrolled cavitation, since data for 7075 Al-L and 7075 Alman⁵⁸¹ have developed finite element creep models that

but are less than 5, which represents the year all mot

incorporate cavitation as well as GBS. Their results in comparison to their serior element and point

incor

onto a single curve (Figure 10). On the other hand, it was B. *High-Temperature Rupture* and the same alloy but in an overheated condition (Figure 10). As dis-The detailed microstructural observations made in the cussed earlier, microstructural evidence of widespread cavi-

within PFZs. By contrast, the success of σ_C suggests that under complex loading conditions and the governing creep cavitation in 7075 Al-OH is constrained by relatively uni- rupture mechanism in the alloy under investigation. form creep deformation, and the driving force for cavity growth is therefore determined primarily by the rate of dislo-

cation creep within the matrix of the grains.
The principal facet stress was found to provide the best **V. CONCLUSIONS** correlation of multiaxial rupture data for a wide range of High-temperature deformation and rupture behavior of
engineering alloys in earlier studies.^[20,40,42] The rationale for commercial 7075 Al was investigated at 6 the success of this parameter has been the operation of at

labiaxial, and triaxial stress states. Annealed and over-

least one localized deformation mechanism, such as GBS

or rapid creep within PFZs that facilitate stre

controlled components, becomes constrained if it is faster boundaries containing few or no particles on boundaries in than the creep rate of the surroundings.^[43] In coarse-grained the annealed microstructure. 7075 Al-OH, the large variations in strength between the Rupture times for different stress states have been com-
wide PFZs and the surrounding material may significantly pared with respect to different physically based mu enhance the plasticity-controlled component, leading to con- stress parameters, each of which is linked to a particular strained cavity growth rates. Consequently, in a manner that physical mechanism known to control the creep rupture prois envisaged in the derivation of σ_C , grain boundary cavita-
tion in 7075 Al-OH becomes constrained by creep deforma-
nor the von Mises effective stress could satisfactorily corretion of the surroundings. late the rupture data for 7075 Al. While the rupture time

the large majority of the cavities observed in 7075 Al-OH successfully correlated by the principal facet stress parameform at large second-phase constituent particles (due to parti-
ter, it was found that σ_c , based on the constrained cavity cle/matrix debonding). As expected, the heterogeneous dis-
tribution of cavities in 7075 Al-OH (Figures 11(e) and (f)) alloy tested in the overheated condition. was also found to correspond directly with that of coarse (5 The success of these parameters has two major implica-
to 25 km) second phase constituents (Figure 1(f)) Begions tions. First, creep rupture in annealed 7075 Al

bility of the approach utilized in the present investigation. The principal advantage of this approach lies in the fact that
the geometry of the samples used to generate multiaxial
ACKNOWLEDGMENTS rupture data (Table II) is simple, and tests are performed
using conventional creep testing apparatus. Results can suc-
Department of Energy under Grant No. DE-FG03cessfully be applied to identify both the suitable representa- 90ER45420 and, in part, by the National Science Foundation tive stress parameter for predicting creep life of components under Grant No. DMR-9810422.

 σ_C , rather than σ_F , provides better correlation for 7075 Also stage, (2) a creep stress exponent of almost 5.3, and (3)
OH. This failure of σ_F may be related to the absence of
GBS in the 7075 Al-OH, which contain

pared with respect to different physically based multiaxial nor the von Mises effective stress could satisfactorily corre-Microstructural observations (Section III–C) indicate that for annealed 7075 Al (regardless of sample orientation) was

The preceeding discussion highlights the general applica-
The preceding discussion highlights the particle spectra in preceding discussion highlights the particle spectra in the particle spectra in the particle with GBS. S

Department of Energy under Grant No. DE-FG03-

-
- 2. M.F. Ashby and B.F. Dyson: *Proc. 6th Int. Conf. on Fracture ICG6*, 22. N.E. Paton, C.H. Hamilton, J. Wert, and M. Mahoney: *J. Met.*, 1982, S.R. Valluri, D.M.R. Taplin, P. Rama Rao, J.F. Knott, and R. Dubey, vol. 34, pp. 21-27.
eds., Pergamon, New Delhi, India, 1984, vol. 1, pp. 3-30. 23 C.H. Hamilton C.C.
- *Materials and Structures*, B. Wilshire and D.R. Owen, eds., Pineridge AIME, San Diego, CA, 1982, pp. 173-89.
Press, Swansea, United Kingdom, 1982, pp. 1-52. 24. P. Malek: *Mater. Sci. Eng. A.* 1991, vol. Press, Swansea, United Kingdom, 1982, pp. 1-52. 24. P. Malek: *Mater. Sci. Eng. A*, 1991, vol. A137, pp. 21-26.
4. D.R. Hayhurst: *J. Mech. Phys. Solids*, 1972, vol. 20, pp. 381-90. 25. F.A. Mohamed and T.G. Langdon: *Acta*
-
- 5. D.R. Hayhurst and F.A. Leckie: *Proc. 4th Int. Conf. on the Mechanical* 117-24. Press, Stockholm, 1984, vol. 2, pp. 1195-1211. (697-709, 6. R.J. Browne, D. Lonsdale, and P.E.J. Flewitt: Trans. ASME, J. Eng. 27. A. Ball and P.E.J. Flewitt: Trans. ASME, J. Eng. 27. A. Ball and P.E.J. Flewitt: Trans. ASM
- 6. R.J. Browne, D. Lonsdale, and P.E.J. Flewitt: *Trans. ASME, J. Eng.* 27. A. Ball and M.M. Hutchinson: *Met. Sci. J.*, 1969, vol. 3, pp. 1-7. *Mater. Technol.*, 1982, vol. 104, pp. 291-96. 28. A.K. Mukherjee: *Mater. Sci. Eng.*, 1971, vol. 8, pp. 83-89.
29. R.C. Gifkins: *Metall. Trans. A*, 1976, vol. 7A, pp. 1225-327. 29. R.C. Gifkins: *Metall. Trans. A*, 1976,
-
- vol. 107, pp. 421-29.

9. D.R. Hayhurst, F.A. Leckie, and J.T. Henderson: *Int. J. Mech. Sci.*, 31. A. Arieli an
- 1977, vol. 19, pp. 147-59.
10. H.-K. Kim, F.A. Mohamed, and J.C. Earthman: *J. Testing Eval.*, 1991, 22. T.G. L
- 10. H.-K. Kim, F.A. Mohamed, and J.C. Earthman: *J. Testing Eval.*, 1991, 32. T.G. Langdon: *Metall. Mater. Trans. A*, 1982, vol. 13A, pp. 689-701.
- 11. N.L. Johnson and J.C. Earthman: *J. Testing Eval.*, 1994, vol. 22, pp. 759-62.
- 12. R. Ayer, J.Y. Koo, J.W. Steeds, and B.K. Park: *Metall. Trans. A*, 1985, 2007-12. vol. 16A, pp. 1925-36. 35. F. Ghahremani: *Int. J. Solids Struct.*, 1980, vol. 16, pp. 847-62.
- 29A, pp. 1145-51. 725-47.
- 14. J.A. Wert, N.E. Paton, C.H. Hamilton, and M.W. Mahoney: *Metall.* 37. A. Chakraborty and J.C. Earthman: *Metall. Mater. Trans. A*, 1997, *Trans. A*, 1981, vol. 12A, pp. 1267-76. vol. 28A, pp. 979-89.
- 15. X. Yang, H. Miura, and T. Sakai: *Mater. Trans. JIM*, 1996, vol. 37, 38. A. Chakraborty and J.C. Earthman: *Acta Mater.*, 1997, vol. 45, pp. pp. 1379-87. 4615-26.
- 61. 40. H.-K. Kim, F.A. Mohamed, and J.C. Earthman: *Metall. Trans. A*, 1991,
- 17. B.F. Dyson and T.B. Gibbons: *Acta Metall.*, 1987, vol. 35, pp. vol. 22A, pp. 2629-36. 2355-69. 41. F.A. Leckie: *Eng. Fract. Mech.*, 1986, vol. 25, pp. 505-21.
- *Relaxation and Fracture of Metallic Alloys*, HMSO, Edinburgh, 1962. 1996, vol. 27A, pp. 891-900.
- 19. J.R. Rice: *Acta Metall.*, 1981, vol. 29, p. 81. 43. B.F. Dyson: *Met. Sci.*, 1976, vol. 10, pp. 349-53.
- **REFERENCES** 20. W.D. Nix, J.C. Earthman, G. Eggeler, and B. Ilschner: *Acta Metall.*, 1989, vol. 37, pp. 1067-77.
- 21. J.E. Bird, A.K. Mukherjee, and J.E. Dorn: in *Quantitative Relation* 1. H. Riedel: in *Fracture at High Temperatures*, B. Ilschner and N. Grant, *between Microstructure and Properties*, D.G. Brandon and A. Rosen, eds., MRE Springer-Verlag, New York, NY, 1986. eds., Israel Universities Press, Haifa, 1969, pp. 255-342.
	-
- eds., Pergamon, New Delhi, India, 1984, vol. 1, pp. 3-30. 23. C.H. Hamilton, C.C. Bampton, and N.E. Paton: in *Superplastic Form*ing of Structural Alloys, N.E. Paton and C.H. Hamilton, eds., TMS-
	-
	- 25. F.A. Mohamed and T.G. Langdon: *Acta Metall.*, 1975, vol. 23, pp.
	- 26. F.A. Mohamed and T.G. Langdon: *Phil. Mag.*, 1975, vol. 32, pp.
	-
	-
- 7. B.J. Cane: *Acta Metall.*, 1981, vol. 29, pp. 1581-91. 29. R.C. Gifkins: *Metall. Trans. A*, 1976, vol. 7A, pp. 1225-32.
	- 8. R.L. Huddleston: *Trans. ASME, J. Pressure Vessel Technol.*, 1985, 30. J.H. Gittus: *Trans. ASME, H, J. Eng. Mater. Technol.*, 1977, vol. 99,
	- 31. A. Arieli and A.K. Mukherjee: *Mater. Sci. Eng.*, 1980, vol. 45, pp.
	-
	- 33. F.A. Mohamed and T.G. Langdon: *Scripta Metall.*, 1976, vol. 10, pp.
	- 34. T.J. Ginter and F.A. Mohamed: *J. Mater. Sci.*, 1982, vol. 17, pp.
	-
	- 13. M. Gao, C.R. Feng, and R.P. Wei: *Metall. Mater, Trans. A*, 1998, vol. 36. M.W. Dib and G.J. Rodin: *J. Mech. Phys. Solids*, 1993, vol. 41, pp.
	-
	-
- 16. J. Liu and D.J. Chakrabarti: *Acta Mater.*, 1996, vol. 44, pp. 4647- 39. F.A. Leckie: *Phil. Trans. R. Soc. London A*, 1995, vol. 351, pp. 611-23.
	-
	-
- 18. A.E. Johnson, J. Henderson, and B. Khan: *Complex Stress Creep,* 42. H. Yu-Hsian, Z. Hongyan, and G.S. Daehn: *Metall. Mater. Trans. A*,
	-