

NUMERICAL DETERMINATION OF EFFECT OF EXPLOSIVE DENSITY ON  
PARAMETERS OF AIR SHOCK WAVES

V. V. Koren'kov and V. N. Otishin

UDC 534.222

The effect of explosive density on parameters of shock waves in air was first studied experimentally in [1] using spherical charges of TEN and lead azide. Numerical calculations of one-dimensional explosive waves in air for standard density were performed in [2-5]. The present study will consider the problem of detonation in air of a spherical TEN charge of variable density.

The one-dimensional system of gasdynamics equations in Lagrangian variables

$$\begin{aligned} \partial u / \partial t &= -1/\rho_0 \cdot (r/\lambda)^2 \partial p / \partial \lambda, \quad \partial r / \partial t = u, \\ \partial \varepsilon / \partial t &= -p \partial (1/\rho) / \partial t, \quad \rho = \rho_0 (\lambda/r)^2 (\partial \lambda / \partial r) \end{aligned} \quad (1)$$

was solved numerically. Here  $p$ ,  $\rho$ ,  $u$ ,  $\varepsilon$  are the pressure, density, velocity, and specific internal energy of the gas;  $t$  is time;  $r$ ,  $\lambda$  are the Euler and Lagrange coordinates, related by the expression

$$\lambda = \left( \lambda_0^3 + 3/\rho_0 \int_{r_0}^r \rho r^2 dr \right)^{1/3};$$

the subscript 0 indicates the initial air density and the coordinates of the charge surface.

System (1) is completed by the equation of state of the medium. For air, the equation of state of an ideal gas,

$$p = (k_e - 1)\rho\varepsilon \quad (2)$$

was used, with effective adiabatic index  $k_e$  in the form of an analytical approximation over a wide range of thermodynamic parameters:

$$k_e = \begin{cases} k_0 - 0.042 (\varepsilon/\varepsilon_k)^2 & \text{for } \varepsilon \leq \varepsilon_k, \\ a_k + (1.36 - a_k) \exp(0.223(1 - \varepsilon/\varepsilon_k)) & \text{for } \varepsilon > \varepsilon_k, \end{cases}$$

where  $a_k = 1 + 0.163/(1 - 0.0573 \ln(\rho/\rho_k))$ ;  $\rho_k = 1.2921 \text{ kg/m}^3$ ,  $\varepsilon_k = 1.116 \cdot 10^3 \text{ J/kg}$ ,  $k_0 = 1.402$ .

If required, the air temperature can be determined from the thermal equation of state

$$T = p\mu_e/\rho R, \quad (3)$$

where  $R$  is the universal gas constant and  $\mu_e$  is the effective molecular mass of the air, determined by

$$\mu_e = \begin{cases} 28.96 & \text{for } \varepsilon \leq \varepsilon_k, \\ 11.5 + 17.46 \exp(0.0445(1 - \varepsilon/\varepsilon_k)) & \text{for } \varepsilon > \varepsilon_k. \end{cases}$$

For the detonation products, the following binomial equation of state was used [7]:

TABLE 1

|                                 |        |        |        |        |
|---------------------------------|--------|--------|--------|--------|
| $\rho_{10}$ , kg/m <sup>3</sup> | 1600   | 1200   | 800    | 400    |
| $D$ , m/sec                     | 7750   | 6040   | 4450   | 3020   |
| $Q \cdot 10^{-6}$ , J/kg        | 5,85   | 5,65   | 5,45   | 5,25   |
| $k$                             | 2,953  | 2,80   | 2,51   | 1,94   |
| $r_0/r_e$                       | 0,0137 | 0,0153 | 0,0177 | 0,0226 |

TABLE 2

|                                 |        |       |       |       |
|---------------------------------|--------|-------|-------|-------|
| $\rho_{10}$ , kg/m <sup>3</sup> | 1600   | 1200  | 800   | 400   |
| $r_n/r_0$                       | 18,35  | 17,10 | 15,28 | 12,84 |
| $\eta$ , %                      | 87,30  | 82,95 | 81,50 | 79,30 |
| $T_1$ , °K                      | 11 500 | 9600  | 8350  | 6500  |
| $T_2$ , °K                      | 4 800  | 4060  | 3400  | 2900  |
| $R_e$                           | 1      | 1,065 | 1,110 | 1,165 |

$$p = A\rho^n + \gamma\rho\varepsilon, \quad (4)$$

where the constants  $A$ ,  $n$ ,  $\gamma$  are determined from the Chapman—Jouguet parameters for detonation of a standard density explosive (single crystal).

Equation of state (4) permits determination of detonation wave parameters in an explosive of arbitrary density with consideration of the change in heat of explosion [8]. In particular, for TEN the calculation results for parameters on the front are approximated well by the expressions

$$\rho_*/\rho_1 = 0.053 + 1.28(\rho_{10}/\rho_1), \quad D_*/D_1 = 0.225 + 0.775(\rho_{10}/\rho_1)^{1,2},$$

while for the flow region for a spherical wave we have

$$\begin{aligned} \rho_z = u_z = 1 - (1 - r_z^{1,15})^{0,5} & \text{ for } r_1 \leq r \leq r_*, \\ u = 0, \rho = \rho_1 & \text{ for } r < r_1, \end{aligned} \quad (5)$$

where  $\rho_z = (\rho - \rho_1)/(\rho_* - \rho_1)$ ;  $u_z = u/u_*$ ;  $r_z = (r - r_1)/(r_0 - r_1)$ ;  $D$  is the detonation rate; the subscripts  $*$ ,  $1$ ,  $10$ ,  $i$  refer to parameters on the detonation front, in the central stationary zone, initial values, and values for a standard density explosive. The radius of the stationary zone and the explosion product density therein are given by

$$r_1/r_* = 0.455 - 0.036(1 - \rho_{10}/\rho_1), \quad \rho_i/\rho_* = 0.616 - 0.348 \exp(-3.54(\rho_{10}/\rho_1)).$$

Remaining detonation wave parameters are determined from known relationships on the wave front, and from the detonation products deloading adiabat.

Equation (5) is satisfied by asymptotic solutions in the vicinity of the detonation wave front and weak discontinuity, and is used as initial conditions for numerical solution of the problem of explosion of a spherical TEN charge of variable density.

System (1)-(4) was integrated using an explicit completely conservative difference technique with approximation order  $O(h^2 + \tau)$  [9]. Homogeneity is ensured by oblique calculation of the contact discontinuity zone. Thus, to maintain the local approximation upon transition through the contact discontinuity in the detonation product region, a logarithmic grid was used, with cell size decreasing toward the contact discontinuity. Calculation of the shock wave was carried out using the characteristics of system (1) together with dynamic compatibility relationships, ensuring distinction of the strong discontinuity and improving the calculation accuracy. To "diffuse" the internal discontinuities an artificial viscosity was used, introduced by analysis of the differential properties of the solution [10]:

$$q = \delta\rho\Delta r \frac{\partial u}{\partial r} \left( S_1 a + S_2 \Delta r \left| \frac{\partial u}{\partial r} \right| \right),$$

where

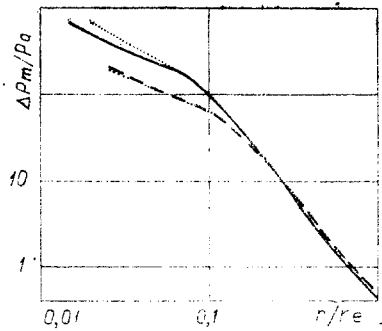


Fig. 1

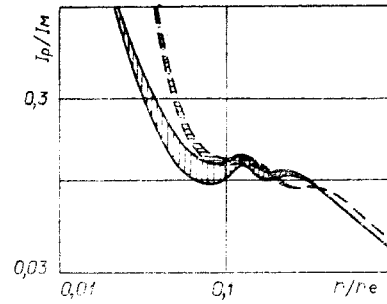


Fig. 2

$$\delta = \begin{cases} 0 & \text{for } \frac{\partial u}{\partial r} \geq 0 \\ \delta_1 & \text{for } \frac{\partial u}{\partial r} < 0 \end{cases}, \text{ if } \left| \frac{\partial^2 u}{\partial r^2} \right| dr - \delta_3 \left| \frac{\partial u}{\partial r} \right| \leq 0,$$

$$\delta = \begin{cases} \delta_2 & \text{for } \frac{\partial u}{\partial r} < 0 \end{cases}, \text{ if } \left| \frac{\partial^2 u}{\partial r^2} \right| dr - \delta_3 \left| \frac{\partial u}{\partial r} \right| > 0;$$

$a$  is the speed of sound;  $S_1, S_2$  are coefficients of the linear and quadratic components of the pseudoviscous pressure  $q$ ;  $\delta_1, \delta_2, \delta_3$  are coefficients of the differential analyzer.

Best results were obtained with values  $S_1 = 0.25$ ;  $S_2 = 1$ ;  $\delta_1 = \delta_2 = 2$ ;  $\delta_3 = 3$ . Calculations were performed on a BESM-6 computer with a grid containing 60 nodes in the detonation product zone and 140 nodes in the air shock wave region, at which levels the law of conservation of energy is satisfied within the problem within 0.5%. Calculation time for one variant to a shock wave front radius of  $\sim 120r_0$  at a Courant number of 0.25 was of the order of 5 min.

Calculations were performed for air at a temperature of  $T = 15^\circ\text{C}$  and TEN charges with densities  $\rho_{10} = 1600, 1200, 800, 400 \text{ kg/m}^3$ , with detonation characteristics presented in Table 1 ( $Q$  is the specific heat of explosion,  $k$  is the adiabatic index on the detonation wave front).

Some calculation results are presented in Figs. 1-4, which show the dependence of excess pressure on the front  $\Delta P_m$ , excess pressure impulse  $I_p = \int_0^t \Delta p dt$ , velocity head impulse  $I_u = \int \frac{\rho u^2}{2} dt$ , and duration of compression phase  $\tau$  on distance  $r$  for charges with  $\rho_{10} = 1600 \text{ kg/m}^3$  (solid lines) and  $\rho_{10} = 400 \text{ kg/m}^3$  (dashed lines). Experimental values of  $\Delta p_m$  [1] are shown by the dotted line of Fig. 1. All results are shown in dimensionless form using the scale factors:  $p_0 = 102,325 \text{ Pa}$ ;  $\rho_0 = 1.2249 \text{ kg/m}^3$ ;  $u_m = (p_0/\rho_0)^{1/2}$ ;  $r_e = (mQ/p_0)^{1/3}$ ;  $t_m = r_e/u_m$ ;  $I_m = p_0 t_m$ . Values of the dimensionless charge radius are presented in the last column of Table 1.

The results for excess pressure on the wave front (Fig. 1) agree well with experimental data. At distances of  $r/r_e < 0.3-0.4$ , but also all other shock wave parameters increase with decreasing charge density. Thus, upon change from  $\rho_{10} = 1600 \text{ kg/m}^3$  to  $\rho_{10} = 400 \text{ kg/m}^3$  in the far zone the values of  $\Delta p_m, I_u$ , and  $I_p$  increases by 10, 12, and 15%, respectively. The nonmonotonic change of  $I_p$  and  $\tau$  (Figs. 2 and 4) in the range  $0.05 < r/r_e < 0.2$  reflects the complex wave character of the flow of the medium, related to formation and propagation of a secondary shock wave. Increase in  $I_p$  in the second shock wave is indicated in Fig. 2 by the shaded region.

For the variants calculated, Table 2 presents values of maximum explosion product bubble radius  $r_b$ , explosion efficiency  $\eta$  (fraction of energy liberated which is radiated into the air in the first pulsation of the explosion products), and air temperature at the boundary with the explosion products at the initial time  $T_1$  and at the time the bubble reaches its maximum radius  $T_2$ .

Despite the fact that the relative size of the gas bubble decreases with decreasing charge density, its absolute size increases, and at  $\rho_{10} = 400 \text{ kg/m}^3$  reaches a value of 20.5 times the radius of a charge with density  $\rho_{10} = 1600 \text{ kg/m}^3$ . With decrease in explosive density the explosion efficiency also falls, so that the increase in shock wave parameters

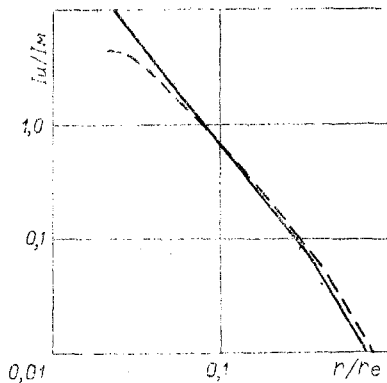


Fig. 3

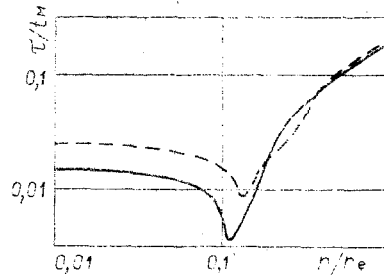


Fig. 4

at large distances can only be explained by a decrease in irreversible losses on the shock wave front. The air temperature on the explosion product contact surface decreases by a factor of 2.2-2.4 during the process of gas bubble expansion.

The falloff in air shock wave parameters in the region  $r/r_e > 0.55$  ( $\Delta p_m < 2p_0$ ) is practically the same for any charge density, making it possible to introduce an energy equivalent relative to a standard density charge  $K_e$ , values of which are presented in Table 2. The relationship of  $K_e$  to explosive density in the range  $400 \text{ kg/m}^3 \leq \rho_{10} \leq 1600 \text{ kg/m}^3$  is approximated well by the linear expression

$$K_e = 1 + 0.1376(1.6 - \rho_{10}/1000). \quad (6)$$

In the range  $2 < \Delta p_m/p_0 \leq 10$  the value of  $K_e$  depends not only on explosive density, but also on shock-wave intensity, obeying the law

$$K_e = 1 + 0.0172(10 - \Delta p_m/p_0)(1.6 - \rho_{10}/1000). \quad (7)$$

Introduction of the energy equivalent  $K_e$  permits use of the same expressions for air shock-wave parameters over the range  $\Delta p_m < 10p_0$  for charges of various densities. Calculations for other solid explosives indicate that Eqs. (6), (7) are applicable within  $\sim 1\%$ .

Thus, the binomial equation of state for detonation products and the simplest approximation for the thermodynamic properties of air permit satisfactory consideration of explosive density in calculating the parameters of air shock waves.

#### LITERATURE CITED

1. B. D. Khristoforov, "Shock-wave front parameters in air upon explosion of TEN and lead azide charges of various densities," Zh. Prikl. Mekh. Tekh. Fiz., No. 6 (1961).
2. H. L. Brode, "Blast wave from a spherical charge," Phys. Fluids, 2, No. 2 (1959).
3. A. S. Fonarev and S. Yu. Chernyavskii, "Shock wave calculation in explosion of spherical charges in air," Izv. Akad. Nauk SSSR, Mekh. Zhidk. Gaza, No. 5 (1968).
4. V. T. Markin, N. I. Nosenko, and N. N. Sysoev, "Flow field behind a nonstationary shock wave formed by explosion of a spherical charge," Uch. Zap. Tsentr. Aero-gidrodin. Inst., 10, No. 2 (1979).
5. S. A. Zhdan, "Calculation of the dynamic load acting on the wall of an explosion chamber," Fiz. Goreniya Vzryva, No. 2 (1981).
6. F. A. Baum, L. P. Orlenko, K. P. Stanyukovich, V. P. Chelyshev, and B. I. Shekhter, Physics of Explosions [in Russian], 2nd ed., Nauka, Moscow (1975).
7. A. V. Kashirskii, L. P. Orlenko, and V. N. Okhitin, "Effect of the equation of state on expansion of explosion products," Zh. Prikl. Mekh. Tekh. Fiz., No. 2 (1973).
8. V. N. Okhitin, "Effect of explosive density on detonation parameter," Trudy MVTU No. 358. Questions of Explosion and Collision Physics [in Russian], 3rd ed., MVTU, Moscow (1981).
9. A. A. Samarskii and Yu. P. Popov, Difference Networks in Gasdynamics [in Russian], Nauka, Moscow (1975).
10. A. I. Ivandaev, "One method for introducing pseudoviscosity and its use in refining difference solutions of the hydrodynamics equations," Zh. Vychisl. Mat. Mat. Fiz., 15, No. 2 (1975).