# **Chapter 26 Modeling of Large-Strain Cyclic Plasticity Including Description of Anisotropy Evolution for Sheet Metals**

#### **Fusahito Yoshida, Takeshi Uemori and Hiroshi Hamasaki**

**Abstract** The present paper describes a framework for the constitutive modeling of a large-strain cyclic plasticity to describe the evolution of anisotropy and the Bauschinger effect of sheet metals that is based on the Yoshida-Uemori kinematic hardening model. In the model, the shapes of the yield and the bounding surfaces are assumed to change simultaneously with increasing plastic strain. An anisotropic yield function that varies continuously with the plastic strain is defined by a nonlinear interpolation function of the effective plastic strain using a limited number of yield functions determined at a few discrete points of plastic strain. With this modeling framework, any type of yield function can be used and the convexity of the yield surface is always guaranteed. A set of kinematic parameters can be identified experimentally independent of the anisotropic parameters.

**Keywords** Constitutive model · Yoshida-Uemori model · Cyclic plasticity · Anisotropy evolution · Sheet metal

### **26.1 Introduction**

The use of constitutive models that properly describe the elastic-plastic deformation behavior is essential for accurate numerical simulation of sheet metal forming. The anisotropy of sheets is of great concern to the forming industry because it strongly

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influences the formability of sheets. Thus, many types of anisotropic yield functions have been proposed in the past, e.g., Hi[l](#page-13-0)l [\(1948,](#page-13-0) [1979](#page-13-1), [1990](#page-13-2)); Goto[h](#page-13-3) [\(1977](#page-13-3)); Barlat and Lia[n](#page-12-0) [\(1989\)](#page-12-0); Barlat et al[.](#page-12-1) [\(1991,](#page-12-1) [2003](#page-12-2), [2005](#page-12-3)); Cazacu and Barla[t](#page-12-4) [\(2001](#page-12-4), [2003,](#page-12-5) [2004](#page-12-6)); Karafillis and Boyc[e](#page-13-4) [\(1993](#page-13-4)); Bron and Besso[n](#page-12-7) [\(2004\)](#page-12-7); Banabic et al[.](#page-12-8) [\(2005](#page-12-8)); H[u](#page-13-5) [\(2005,](#page-13-5) [2007](#page-13-6)); Leacoc[k](#page-13-7) [\(2006](#page-13-7)); Vegter and Boogaar[d](#page-14-0) [\(2006\)](#page-14-0); Comsa and Banabi[c](#page-12-9)  $(2008)$ ; Steglich et al[.](#page-14-1)  $(2011)$  $(2011)$ ; Desmorat and Maru[l](#page-13-8)l  $(2011)$ , etc.

Another important issue in material modeling is describing cyclic plasticity behavior. Descriptions of the Bauschinger effect and workhardening have been intensively investigated within the framework of a combined isotropic-kinematic hardening model for the past few decades, e.g., Armstrong and Frederic[k](#page-12-10) [\(1966](#page-12-10)); Mró[z](#page-13-9) [\(1967\)](#page-13-9); Krie[g](#page-13-10) [\(1975\)](#page-13-10); Dafalias and Popo[v](#page-13-11) [\(1976](#page-13-11)); Ohn[o](#page-13-12) [\(1982\)](#page-13-12); Chaboche and Rousselie[r](#page-12-11) [\(1983\)](#page-12-11); Ohno and Wan[g](#page-13-13) [\(1993\)](#page-13-13); MacDowel[l](#page-13-14) [\(1995](#page-13-14)); Geng and Wagone[r](#page-13-15) [\(2002](#page-13-15)); Yoshid[a](#page-14-2) [\(2000](#page-14-2)); Yoshida et al[.](#page-14-3) [\(2002](#page-14-3), [2013,](#page-14-4) [2015](#page-14-5)); Yoshida and Uemor[i](#page-14-6) [\(2002,](#page-14-6) [2003\)](#page-14-7); Haddadi et al[.](#page-13-16) [\(2006\)](#page-13-16); Tale[b](#page-14-8) [\(2013\)](#page-14-8); for more details, refer to reviews by Chaboch[e](#page-12-12) [\(2008](#page-12-12)); Ohn[o](#page-13-17) [\(2015\)](#page-13-17). Before 2000, most cyclic plasticity models were constructed within the theory of infinitesimal deformation without considering material anisotropy because they were applied mainly to structural analyses for predicting low-cycle fatigue life and ratcheting. In the early 2000s, some researchers pointed out that the Bauschinger effect of materials greatly affects springback behavior, especially for high-strength steel (HSS) sheets, and several cyclic plasticity models were proposed for springback simulation, e.g., Yoshida and Uemor[i](#page-14-6) [\(2002,](#page-14-6) [2003](#page-14-7)); Geng and Wagone[r](#page-13-15) [\(2002\)](#page-13-15). The present authors previously proposed a model of large-strain cyclic plasticity, so-called 'Yoshida-Uemori (Y-U) model' (Yoshida et al[.](#page-14-3) [2002](#page-14-3); Yoshida and Uemor[i](#page-14-6) [2002](#page-14-6), [2003\)](#page-14-7) to describe the following cyclic plasticity behaviors (see Fig. [26.1\)](#page-1-0) together with the anisotropy of materials:



<span id="page-1-0"></span>**Fig. 26.1** Elastic-plastic behavior in a reverse deformation: early re-yielding, transient Bauschinger effect, workhardening stagnation and permanent softening (Yoshida and Uemor[i](#page-14-7) [2003](#page-14-7))

- Two stages of the Bauschinger effect: (i) *the transient Bauschinger deformation* characterized by early re-yielding and smooth elastic-plastic transition with a rapid change in the workhardening rate; and (ii) *the permanent softening* observed in a region after the transient period;
- *Workhardening stagnation*, which appears at a certain range of reverse deformation;
- The strain-range and mean-strain dependency of cyclic hardening, i.e., the larger cyclic strain range induces the larger saturated stress amplitudes.

A recent topic of plasticity for sheet metals has been the modeling of anisotropic hardening. Conventional plasticity models assume that the shape of the yield surface does not change during a plastic deformation; consequently, the *r*-values and flow stress directionality calculated with these models remain constant throughout the deformation. However, some metallic sheets exhibit significant changes in *r*-value anisotropy and flow stress directionality (e.g., H[u](#page-13-6) [2007](#page-13-6); Stoughton and Yoo[n](#page-14-9) [2009](#page-14-9); An et al[.](#page-12-13) [2013](#page-12-13); Safaei et al[.](#page-14-10) [2014](#page-14-10)) and the shape of the yield surface (e.g., Tozaw[a](#page-14-11) [1978](#page-14-11); Kuwabara et al[.](#page-13-18) [1998;](#page-13-18) Yanaga et al[.](#page-14-12) [2014](#page-14-12); Yoon et al[.](#page-14-13) [2014](#page-14-13)) as plastic strain increases. Although there are some models of the anisotropic hardening (e.g., H[u](#page-13-6) [2007](#page-13-6); Plunkett et al[.](#page-14-14) [2008;](#page-14-14) Stoughton and Yoo[n](#page-14-9) [2009;](#page-14-9) An et al[.](#page-12-13) [2013](#page-12-13); Safaei et al[.](#page-14-10) [2014](#page-14-10); Yanaga et al[.](#page-14-12) [2014](#page-14-12); Yoon et al[.](#page-14-13) [2014\)](#page-14-13), most of them exclude a description of the Bauschinger effect. Distortion yield function modeling is another type of formulation used to represent the Bauschinger effect and stress-strain responses under non-proportional cyclic loading (e.g., Shiratori et al[.](#page-14-15) [1979](#page-14-15); Voyiadjis and Foroozes[h](#page-14-16) [1990](#page-14-16); Kurtyka and Życzkowsk[i](#page-13-19) [1996](#page-13-19); Francoi[s](#page-13-21) [2001](#page-13-20); Feigenbaum and Dafalias [2007](#page-13-21); Barlat et al[.](#page-12-14) [2011,](#page-12-14) [2013,](#page-12-15) [2014](#page-12-16)). However, to the best of the present authors' knowledge, only Barlat et al.'s homogeneous anisotropic hardening (HAH) model (Barlat et al[.](#page-12-14) [2011](#page-12-14), [2013](#page-12-15), [2014\)](#page-12-16) reproduces the Bauschinger effect well together with the anisotropy evolution of sheet metals.

The present paper proposes a model of large-strain cyclic plasticity that describes the evolution of anisotropy and the Bauschinger effect of sheet metals based on the Y-U kinematic hardening model. This modeling framework has great advantages over other models. It allows any type of yield function to be used, and the convexity of the yield surface is always guaranteed. A set of kinematic parameters can be identified from experimentally independent of anisotropic parameters.

## **26.2 Framework of Combined Anisotropic-Kinematic Hardening Model**

With the assumption of small elastic and large plastic deformation, the rate of deformation  $D$  is decomposed into its elastic and plastic parts,  $D^e$  and  $D^p$ , respectively, as follows:

$$
D = D^e + D^p \tag{26.1}
$$

The decomposition of the continuum spin *W* is given as follows:

$$
\mathbf{W} = \mathbf{\Omega} + \mathbf{W}^{\mathrm{p}},\tag{26.2}
$$

where  $W^p$  denotes the plastic spin and  $\Omega$  is the spin of substructures. The constitutive equation of elasticity is expressed as follows:

$$
\dot{\sigma} = \dot{\sigma} - \Omega \sigma + \sigma \Omega = C : D^e,
$$
 (26.3)

where  $\sigma$  and  $\mathring{\sigma}$  are the Cauchy stress and its objective rate, respectively,  $\Omega$  is the spin tensor; and *C* is the elasticity modulus tensor. The initial yield criterion is expressed by the following equation:

$$
f = \phi_0(\boldsymbol{\sigma}) - Y = \overline{\sigma}(\boldsymbol{\sigma}) - Y = 0, \qquad (26.4)
$$

where *Y* is the initial yield stress and  $\overline{\sigma}$  is the effective stress. To describe the Bauschinger effect, as well as the evolutionary change of anisotropy, the subsequent yielding is expressed by the following equat Bauschinger effect, as well as the evolutionary change of anisotropy, the subsequent yielding is expressed by the following equation: *f* is the initial yield stress and ger effect, as well as the evolutions expressed by the following  $f = \phi(\sigma - \alpha, \bar{\epsilon}) - Y = \bar{\sigma}(\tilde{\sigma})$ 

$$
f = \phi(\boldsymbol{\sigma} - \boldsymbol{\alpha}, \boldsymbol{\bar{\varepsilon}}) - Y = \overline{\sigma}(\widetilde{\boldsymbol{\sigma}}, \boldsymbol{\bar{\varepsilon}}) - Y = 0, \quad \widetilde{\boldsymbol{\sigma}} = \boldsymbol{\sigma} - \boldsymbol{\alpha}, \tag{26.5}
$$

where  $\alpha$  denotes the backstress. Based on the following definitions of the effective plastic strain and its rate ess. B<br>  $\dot{\vec{\epsilon}} = \tilde{\sigma}$ 

$$
\overline{\sigma}\dot{\overline{\boldsymbol{\varepsilon}}} = \widetilde{\boldsymbol{\sigma}} : \boldsymbol{D}^{\mathrm{p}}, \qquad \overline{\boldsymbol{\varepsilon}} = \int \dot{\overline{\boldsymbol{\varepsilon}}} \mathrm{d}t,\tag{26.6}
$$

the associated flow rule is written as follows:

en as follows:  
\n
$$
\mathbf{D}^{\mathbf{p}} = \frac{\partial f}{\partial \widetilde{\boldsymbol{\sigma}}} \dot{\lambda} = \frac{\partial \phi}{\partial \widetilde{\boldsymbol{\sigma}}} \dot{\lambda}
$$
\n(26.7)

where  $\lambda = \overline{\varepsilon}$ .

<span id="page-3-0"></span>Most kinematic hardening models assume the following form of the evolution equation of the back stress:

hardening models assume the following form of the evolution  
\n
$$
\dot{\mathbf{\alpha}} = \left\{ \frac{A}{Y} (\boldsymbol{\sigma} - \boldsymbol{\alpha}) - \mathbf{x} \right\} \dot{\overline{\boldsymbol{\varepsilon}}} = \left( \frac{A}{Y} \widetilde{\boldsymbol{\sigma}} - \mathbf{x} \right) \dot{\overline{\boldsymbol{\varepsilon}}},
$$
\n(26.8)

Here  $(\ldots)$  denotes the objective rate. For example, in the linear kinematic hardening model:

<sup>α</sup>˚ <sup>=</sup> *<sup>H</sup>*-*L K <sup>Y</sup>* (σ <sup>−</sup> α)˙ <sup>ε</sup> <sup>=</sup> *<sup>H</sup>*-*L K Y* σ-˙ ε, *A* = *H*-*L K* , *x* = 0 (26.9)

In the Armstrong-Frederick model (Armstrong and Frederic[k](#page-12-10) [1966](#page-12-10)):

Modeling of Large-Strain Cyclic Plasticity Including Description ...  
\n
$$
\hat{\boldsymbol{\alpha}} = \left\{ \frac{\gamma_1}{Y} (\boldsymbol{\sigma} - \boldsymbol{\alpha}) - \gamma_2 \boldsymbol{\alpha} \right\} \dot{\bar{\boldsymbol{\varepsilon}}} = \left( \frac{\gamma_1}{Y} \tilde{\boldsymbol{\sigma}} - \gamma_2 \boldsymbol{\alpha} \right) \dot{\bar{\boldsymbol{\varepsilon}}}, \qquad A = \gamma_1, \boldsymbol{x} = \gamma_2 \boldsymbol{\alpha} \qquad (26.10)
$$

The Y-U kinematic hardening law has the same form (for details, see the following ⎧section). The constitutive equation is given by the following form:

$$
\mathbf{\dot{\sigma}} = \mathbf{C}^{\text{ep}} : \mathbf{D},\tag{26.11}
$$

$$
\mathbf{\hat{\sigma}} = \mathbf{C}^{\text{ep}} : \mathbf{D}, \qquad (26.11)
$$
\n
$$
\mathbf{C}^{\text{ep}} = \begin{cases}\n\mathbf{C} & \text{if } \lambda = 0 \\
\mathbf{C} - \frac{\mathbf{C} : \frac{\partial \phi}{\partial \mathbf{\hat{\sigma}}} \otimes \mathbf{C} : \frac{\partial \phi}{\partial \mathbf{\hat{\sigma}}}}{\frac{\partial \phi}{\partial \mathbf{\hat{\sigma}}} + H' - \frac{\partial \phi}{\partial \overline{\varepsilon}}} & \text{if } \lambda > 0\n\end{cases} \qquad (26.12)
$$

<span id="page-4-1"></span>where

$$
\frac{\partial \varphi}{\partial \widetilde{\sigma}} + H' - \frac{\partial \varphi}{\partial \overline{\varepsilon}}
$$
  

$$
H' = A - \frac{\partial \varphi}{\partial \widetilde{\sigma}} : \mathbf{x}
$$
 (26.13)

# **26.3 Cyclic Plasticity Model to Describe the Bauschinger Effect and Workhardening Stagnation: Yoshida-Uemori Model**

The Y-U model was constructed within the framework of two-surface modeling (Krie[g](#page-13-10) [1975](#page-13-10)), wherein the yield surface moves kinematically within a bounding surface, as schematically illustrated in Fig. [26.2.](#page-4-0) To describe anisotropic hardening (i.e., expansion of the surface with shape change) and also kinematic hardening, the bounding surface  $F$  is expressed by the equation:

<span id="page-4-0"></span>

$$
F = \phi(\boldsymbol{\sigma} - \boldsymbol{\beta}, \overline{\varepsilon}) - (B + R) = 0,
$$
\n(26.14)

where  $\beta$  denotes the center of the bounding surface, and *B* and *R* are the initial size of the surface and its workhardening component, respectively. To include the description of anisotropic hardening in the model, it is assumed that the shapes of both the yield and bounding surfaces vary simultaneously.

The kinematic hardening of the yield surface describes the transient Bauschinger deformation, which is characterized by early re-yielding and subsequent rapid change in workhardening rate. The relative kinematic motion of the yield surface with respect to the bounding surface is expressed by the following equation:

$$
\boldsymbol{\alpha}_{*} = \boldsymbol{\sigma} - \boldsymbol{\beta} \tag{26.15}
$$

The evolution of  $\alpha_*$  is given by the following equation:

$$
\mathbf{\alpha}_{*} = \mathbf{\sigma} - \mathbf{\rho}
$$
\nsolution of  $\mathbf{\alpha}_{*}$  is given by the following equation:

\n
$$
\mathbf{\dot{\alpha}}_{*} = C \left\{ \left( \frac{a}{Y} \right) (\mathbf{\sigma} - \mathbf{\alpha}) - \sqrt{\frac{a}{\overline{\alpha}_{*}}} \right\} \dot{\overline{\varepsilon}} = \left\{ \left( \frac{Ca}{Y} \right) \widetilde{\mathbf{\sigma}} - C \sqrt{\frac{a}{\overline{\alpha}_{*}}} \right\} \dot{\overline{\varepsilon}},\qquad(26.16)
$$

 $\overline{\alpha}_* = \phi(\alpha_*)$ ,  $a = B + R - Y$  (26.17)

An Armstrong-Frederick-type evolution equation is used to express the kinematic hardening of  $\beta$ 

Frederick-type evolution equation is used to express the kinematic  
\n
$$
\mathring{\boldsymbol{\beta}}_{*} = k \left\{ \left( \frac{b}{Y} \right) (\boldsymbol{\sigma} - \boldsymbol{\alpha}) - \boldsymbol{\beta} \right\} \dot{\overline{\varepsilon}} = \left( \frac{kb}{Y} \tilde{\boldsymbol{\sigma}} - k \boldsymbol{\beta} \right) \dot{\overline{\varepsilon}}
$$
\n(26.18)

Thus, in Eq. [\(26.8\)](#page-3-0),

$$
A = Ca + kb, \qquad \mathbf{x} = C \sqrt{\frac{a}{\overline{\alpha}_*}} \mathbf{\alpha} - \left( C \sqrt{\frac{a}{\overline{\alpha}_*}} - k \right) \mathbf{\beta} \tag{26.19}
$$

and in Eq. [\(26.13\)](#page-4-1).

With respect to the expansion of the bounding surface, i.e., the evolution of *R*, in the first version of the Y-U model (Yoshida and Uemor[i](#page-14-6) [2002,](#page-14-6) [2003\)](#page-14-7), the following equation based on the Voce hardening law (Voc[e](#page-14-17) [1948](#page-14-17)) was proposed:

$$
R = R_{\text{Voce}} = R_{\text{sat}} \{ 1 - \exp(-k\overline{\varepsilon}) \},\tag{26.20}
$$

written as

$$
\dot{R} = \dot{R}_{\text{Voce}} = k(R_{\text{sat}} - R_{\text{Voce}})\dot{\bar{\varepsilon}} \tag{26.21}
$$

<span id="page-5-0"></span>However, it is not necessary to use the Voce-type formulation. For example, based on the Swift law (Swif[t](#page-14-18) [1952\)](#page-14-18):

$$
R = R_{\text{Swift}} = K\{(\varepsilon_0 + \overline{\varepsilon})^n - \varepsilon_0^n\},\tag{26.22}
$$

<span id="page-6-0"></span>the following evolution equation can be obtained

$$
\dot{R} = \dot{R}_{\text{Swift}} = nK^{1/n} (R_{\text{Swift}} + K \varepsilon_0^n)^{(n-1)/n} \dot{\overline{\varepsilon}} \tag{26.23}
$$

Furthermore, a combination of the above two hardening laws, Eqs. [\(26.22\)](#page-5-0) and [\(26.23\)](#page-6-0), is also possible and can be expressed as follows:

$$
\dot{R} = \omega \dot{R}_{\text{Swift}} + (1 - \omega) \dot{R}_{\text{Voce}}, \qquad 0 \le \omega \le 1 \tag{26.24}
$$

where  $\omega$  is a weighting coefficient. This model has high flexibility in describing various levels of workhardening at large strain levels.

One of the features of the Yoshida-Uemori model is that it is able to describe the *workhardening stagnation* that appears in a reverse stress-strain curve for a certain range of reverse deformation (see Hasegawa and Yako[u](#page-13-22) [1975](#page-13-22); Christodoulou et al[.](#page-12-17) [1986](#page-12-17)). This phenomenon is closely related to the strain-range and mean-strain dependency of cyclic hardening. Specifically, the larger the cyclic strain range is, the larger the saturated stress amplitudes are. This dependency is expressed by the stagnation of the expansion of the bounding surface for a certain range of reverse deformation. The states of hardening ( $\dot{R} > 0$ ) and non-hardening ( $\dot{R} = 0$ ) of the bounding surface are determined for a so-called non-IH (isotropic hardening) surface,  $g_{\sigma}$ , defined in the stress space as follows and schematically illustrated in Fig. [26.3a](#page-6-1), b:

$$
g_{\sigma} = \phi(\sigma - q, r) - r = 0, \qquad (26.25)
$$

where  $q$  and  $r$  denote the center and size of the non-IH surface, respectively. It is assumed that the center of the bounding surface *q* exists either on or inside of the surface  $g_{\sigma}$ . The expansion of the bounding surface takes place only when the center point of the bounding surface,  $q$ , lies on the surface  $g_{\sigma}$  (see Fig. [26.3b](#page-6-1)), i.e., when



<span id="page-6-1"></span>**Fig. 26.3** Schematic illustration of the non-IH surface defined in the stress space, when expansion of the bounding surface **a** stops, and **b** takes place (Yoshida and Uemor[i](#page-14-7) [2003](#page-14-7))

$$
\dot{R} > 0: g_{\sigma}(\boldsymbol{\beta} - \boldsymbol{q}, r) = \phi(\boldsymbol{\beta} - \boldsymbol{q}, r) - r = 0 \qquad (26.26)
$$

and

$$
\dot{R} = 0: \Gamma = \frac{\partial g_{\sigma}(\boldsymbol{\beta} - \boldsymbol{q}, r)}{\partial \boldsymbol{\beta}} : \dot{\boldsymbol{\beta}} > 0 \tag{26.27}
$$

otherwise. In an analysis of some experimental data, the plastic strain region of workhardening stagnation was found to increase with the accumulated plastic strain. To describe this phenomenon, it was assumed that the surface  $g_{\sigma}$  moves kinematically as it expands. The governing equations of the kinematic motion and expansion of the surface are given by Eqs. [\(26.28\)](#page-7-0) and [\(26.29\)](#page-7-1), respectively.

<span id="page-7-1"></span><span id="page-7-0"></span>
$$
\mathring{\mathbf{q}} = \frac{(1-h)\Gamma}{r}(\mathbf{\beta} - \mathbf{q}),\tag{26.28}
$$

$$
\dot{r} = h\Gamma \tag{26.29}
$$

Here, *h* is a parameter that controls the strength of the workhardening stagnation characteristic. A larger value of *h* corresponds to a larger strain region within which workhardening stagnation occurs, and as a result, a larger value of *h* leads to weaker cyclic hardening of a material. We may assume that the shape of the surface  $g_{\sigma}$ , is fixed  $\phi = \phi_0$ , or even  $\phi$  = von Mises type, throughout the deformation, because the shape of  $g_{\sigma}$  has not been measured experimentally yet, and its effect on the stress-strain calculation would be rather minor.

Models of workhardening stagnation were recently reviewed by Ohn[o](#page-13-17) [\(2015](#page-13-17)). It should be noted that Ohno's model of non-isotropic-hardening, where the nonhardening region is expressed in the plastic strain space, is identical to the infinitesimal-strain Yoshida-Uemori model when assuming a linear kinematic hardening of the bounding surface.

In the proposed model, the size of the yield surface is held constant. However, if we carefully observe the stress-strain response during unloading after plastic deformation, we find that the stress-strain curve is no longer linear but rather is slightly curved due to very early re-yielding and the Bauschinger effect. To describe this phenomenon, in the model, the following equation for *plastic-strain-dependent Young's modulus* is introduced (Yoshida et al[.](#page-14-3) [2002\)](#page-14-3):

$$
E = E_0 - (E_0 - E_\alpha)\{1 - \exp(-\xi \overline{\varepsilon})\},\tag{26.30}
$$

where  $E_0$  and  $E_\alpha$  are Young's modulus for virgin and infinitely large pre-strained materials, respectively, and  $\xi$  is a material constant.

Figure [26.4](#page-8-0) shows stress-strain responses of 780MPa high strength steel sheets under cyclic straining and uniaxial tension, calculated by the Y-U model, together with the corresponding experimental results.

<span id="page-8-0"></span>

### **26.4 Description of Evolution of Anisotropy**

The evolution of anisotropy is expressed by the anisotropic hardening of the yield surface, as follows (refer to Yoshida et al[.](#page-14-5) [2015](#page-14-5)): evolution of anisotropy is expressed by th<br>
ice, as follows (refer to Yoshida et al. 2015<br>  $\phi(\tilde{\sigma}, \bar{\varepsilon}) = \mu(\bar{\varepsilon})\phi_A(\tilde{\sigma}) + (1 - \mu(\bar{\varepsilon}))\phi_B(\tilde{\sigma})$ surface, as follows (refer to Yoshida et al. 2015):<br>  $\phi(\tilde{\sigma}, \bar{\varepsilon}) = \mu(\bar{\varepsilon})\phi_A(\tilde{\sigma}) + (1 - \mu(\bar{\varepsilon}))\phi_B(\tilde{\sigma})$  for  $\bar{\varepsilon}_A \leq \bar{\varepsilon} \leq \bar{\varepsilon}_B$  (26.31)<br>
Here,  $\phi_A(\tilde{\sigma})$  and  $\phi_B(\tilde{\sigma})$  are two different yield functi

$$
\phi(\widetilde{\boldsymbol{\sigma}}, \overline{\varepsilon}) = \mu(\overline{\varepsilon})\phi_A(\widetilde{\boldsymbol{\sigma}}) + (1 - \mu(\overline{\varepsilon}))\phi_B(\widetilde{\boldsymbol{\sigma}}) \quad \text{for} \quad \overline{\varepsilon}_A \le \overline{\varepsilon} \le \overline{\varepsilon}_B \qquad (26.31)
$$

<span id="page-8-1"></span> $\phi(\tilde{\sigma}, \bar{\varepsilon}) = \mu(\bar{\varepsilon})\phi_A(\tilde{\sigma}) + (1 - \mu(\bar{\varepsilon}))\phi_B(\tilde{\sigma})$  for  $\bar{\varepsilon}_A \leq \bar{\varepsilon} \leq \bar{\varepsilon}_B$  (2)<br>Here,  $\phi_A(\tilde{\sigma})$  and  $\phi_B(\tilde{\sigma})$  are two different yield functions defined at the effective<br>tic strains  $\bar{\varepsilon}_A$  and  $\$ tic strains  $\overline{\varepsilon}_A$  and  $\overline{\varepsilon}_B$ , respectively, i.e.,  $\phi_A(\widetilde{\sigma}) = \phi(\widetilde{\sigma}, \overline{\varepsilon}_A)$  and  $\phi_B(\widetilde{\sigma}) = \phi(\widetilde{\sigma}, \overline{\varepsilon}_B)$ , and  $\mu(\bar{\varepsilon})$  an interpolation function of the effective plastic strain, where<br>  $1 = \mu(\bar{\varepsilon}_A) \ge \mu(\bar{\varepsilon}) \ge \mu(\bar{\varepsilon}_B) = 0$ <br>
Note that the types of these two yield functions,  $\phi_A(\tilde{\sigma})$  and  $\phi_B(\tilde{\sigma})$ , do n

$$
1 = \mu(\overline{\varepsilon}_A) \ge \mu(\overline{\varepsilon}) \ge \mu(\overline{\varepsilon}_B) = 0 \tag{26.32}
$$

Note that the types of these two yield functions,  $\phi_A(\tilde{\sigma})$  and  $\phi_B(\tilde{\sigma})$ , do not need to be the same. An advantage of this modeling framework is that, if the two yield functions  $1 = \mu(\bar{\varepsilon}_A) \ge \mu(\bar{\varepsilon}) \ge \mu(\bar{\varepsilon}_B) = 0$  (26.32)<br>
Note that the types of these two yield functions, φ<sub>*A*</sub>(**σ**) and φ<sub>*B*</sub>(**σ**), do not need to be<br>
the same. An advantage of this modeling framework is that, if the two are expressed as follows:

$$
\frac{\partial \phi}{\partial \widetilde{\sigma}} = \mu(\overline{\varepsilon}) \frac{\partial \phi_A(\widetilde{\sigma})}{\partial \widetilde{\sigma}} + (1 - \mu(\overline{\varepsilon})) \frac{\partial \phi_B(\widetilde{\sigma})}{\partial \widetilde{\sigma}}, \qquad \frac{\partial \phi}{\partial \overline{\varepsilon}} = (\phi_A(\widetilde{\sigma}) - \phi_B(\widetilde{\sigma})) \frac{\partial \mu(\overline{\varepsilon})}{\partial \overline{\varepsilon}} \tag{26.33}
$$

Several linear and nonlinear functions can be used for the interpolation function  $\mu(\bar{\varepsilon})$ . Assuming that  $\bar{\varepsilon}_A = 0$  (at initial yielding) and  $\bar{\varepsilon}_A = \infty$  (at infinitely large  $\frac{\partial \varphi}{\partial \vec{\sigma}} = \mu(\overline{\varepsilon}) \frac{\partial \varphi_A(\sigma)}{\partial \vec{\sigma}} + (1 - \mu(\overline{\varepsilon})) \frac{\partial \varphi_B(\sigma)}{\partial \vec{\sigma}}, \qquad \frac{\partial \varphi}{\partial \overline{\varepsilon}} = (\phi_A(\overline{\varepsilon}))$ <br>Several linear and nonlinear functions can be used for the i<br> $\mu(\overline{\varepsilon})$ . Assuming that  $\overline{\varepsilon}_A = 0$  (at i ), Eq.  $(26.31)$  reduces to the following suming that  $\overline{\varepsilon}_A = 0$  (at initial yielding) a<br>d that  $\phi_A(\widetilde{\sigma}) = \phi_0(\widetilde{\sigma}) = \phi_0(\sigma)$  and  $\phi_B(\widetilde{\sigma})$ <br>lowing<br> $\phi(\widetilde{\sigma}, \overline{\varepsilon}) = \mu(\overline{\varepsilon})\phi_0(\widetilde{\sigma}) + (1 - \mu(\overline{\varepsilon}))\phi_\infty(\widetilde{\sigma})$ 

$$
\phi(\widetilde{\sigma}, \overline{\varepsilon}) = \mu(\overline{\varepsilon})\phi_0(\widetilde{\sigma}) + (1 - \mu(\overline{\varepsilon}))\phi_\infty(\widetilde{\sigma}) \qquad 0 \le \overline{\varepsilon} \le 0 \tag{26.34}
$$

Some examples of forms of interpolation functions are as follows:

$$
\mu(\overline{\varepsilon}) = \exp(-\lambda \overline{\varepsilon}),\tag{26.35}
$$

$$
\mu(\overline{\varepsilon}) = a \exp(-\lambda_1 \overline{\varepsilon}) + (1 - a) \exp(-\lambda_2 \overline{\varepsilon}), \tag{26.36}
$$

where  $\lambda$ ,  $a$ ,  $\lambda_1$  and  $\lambda_2$  are material constants.

If we have M sets of experimental data ( $\sigma_0$ ,  $\sigma_{45}$ ,  $\sigma_{90}$ ,  $\sigma_b$ ,  $r_0$ ,  $r_{45}$ ,  $r_{90}$ , etc.) for material parameter identification corresponding to *M* discrete plastic strain points,  $\overline{\varepsilon}_1(= 0), \overline{\varepsilon}_2, \ldots, \overline{\varepsilon}_i, \overline{\varepsilon}_{i+1}, \ldots, \overline{\varepsilon}_M$ , we can determine *M* sets of yield functions where  $\lambda$ ,  $a$ ,  $\lambda_1$  and  $\lambda_2$  are material constants.<br>If we have M sets of experimental data ( $\sigma_0$ ,  $\sigma_4$ ,  $\sigma_{90}$ ,  $\sigma_b$ ,  $r_0$ ,  $r_{45}$ ,  $r_{90}$ , etc.) for<br>material parameter identification corresponding to M d If we have M sets of ex<br>material parameter identifica<br> $\overline{\varepsilon}_1 (= 0), \overline{\varepsilon}_2, ..., \overline{\varepsilon}_i, \overline{\varepsilon}_{i+1}, ..., \phi_1(\widetilde{\sigma}), \phi_2(\widetilde{\sigma}), ..., \phi_i(\widetilde{\sigma}), \phi_i$ <br> $\mu(\overline{\varepsilon})$ , the yield function  $\phi(\widetilde{\sigma})$  $\mu(\overline{\varepsilon})$ , the yield function  $\phi(\tilde{\sigma}, \overline{\varepsilon})$  can be defined by the following equation: = 0),  $\bar{\varepsilon}_2, ..., \bar{\varepsilon}_i, \bar{\varepsilon}_{i+1}, ..., \bar{\varepsilon}_M$ , we can d<br>  $\tilde{\sigma}$ ),  $\phi_2(\tilde{\sigma}), ..., \phi_i(\tilde{\sigma}), \phi_{i+1}(\tilde{\sigma}), ..., \phi_M(\tilde{\sigma})$ <br>  $\tilde{\varepsilon}$ ), the yield function  $\phi(\tilde{\sigma}, \bar{\varepsilon})$  can be defined<br>  $\phi(\tilde{\sigma}, \bar{\varepsilon}) = \mu(\bar{\varepsilon})\phi_i(\tilde{\sigma}) + (1$ 

$$
\phi(\widetilde{\boldsymbol{\sigma}}, \overline{\varepsilon}) = \mu(\overline{\varepsilon})\phi_i(\widetilde{\boldsymbol{\sigma}}) + (1 - \mu(\overline{\varepsilon}))\phi_{i+1}(\widetilde{\boldsymbol{\sigma}}) \quad \text{for} \quad \overline{\varepsilon}_i \le \overline{\varepsilon} \le \overline{\varepsilon}_{i+1} \quad (26.37)
$$

<span id="page-9-1"></span>The following nonlinear equation is proposed for use as the interpolation function:

$$
\mu(\overline{\varepsilon}) = 1 - \left(\frac{\overline{\varepsilon} - \overline{\varepsilon}_i}{\overline{\varepsilon}_{i+1} - \overline{\varepsilon}_1}\right)^{p_i} \qquad \overline{\varepsilon}_i \le \overline{\varepsilon} \le \overline{\varepsilon}_{i+1},\tag{26.38}
$$

where  $p_i(i = 1, 2, \ldots, M - 1)$  are material constants.

Among the various types of anisotropic yield functions available, stress polynomial-type models (e.g., Hil[l](#page-13-0) [1948;](#page-13-0) Goto[h](#page-13-3) [1977;](#page-13-3) Soare et al[.](#page-14-19) [2008;](#page-14-19) Yoshida et al[.](#page-14-4) [2013](#page-14-4)) are suitable for use in modeling anisotropy evolution. A polynomial-type yield criterion is given by the following equation:

$$
f = \phi^{(m)}(\sigma) - Y^m = \overline{\sigma}^m - Y^m = 0,
$$
\n(26.39)

where  $\phi^{(m)}(\sigma)$  denotes the *m*th order stress polynomial-type yield function. For example, when  $m = 6$  (Yoshida et al[.](#page-14-4) [2013](#page-14-4)) under plane stress condition,

$$
\phi^{(6)} = C_1 \sigma_x^6 - 3C_2 \sigma_x^5 \sigma_y + 6C_3 \sigma_x^4 \sigma_y^2 - 7C_4 \sigma_x^3 \sigma_y^3 + 6C_3 \sigma_x^2 \sigma_y^4 - 3C_6 \sigma_x \sigma_y^5
$$
  
+  $C_7 \sigma_y^6 + 9(C_8 \sigma_x^4 - 2C_9 \sigma_x^3 \sigma_y + 3C_{10} \sigma_x^2 \sigma_y^2 - 2C_{11} \sigma_x \sigma_y^3 + 2C_{12} \sigma_y^4) \tau_{xy}^2$   
+  $27(C_{13} \sigma_x^2 - C_{14} \sigma_x \sigma_y + C_{15} \sigma_y^2) \tau_{xy}^4 + 27C_{16} \tau_{xy}^6$   
(26.40)  
the same manner as Eq. (26.31), when the following equation is assumed  

$$
\phi^{(m)}(\tilde{\sigma}, \bar{\varepsilon}) = \mu(\bar{\varepsilon}) \phi_A^{(m)}(\tilde{\sigma}) + (1 - \mu(\bar{\varepsilon})) \phi_B^{(m)}(\tilde{\sigma}),
$$
 (26.41)

In the same manner as Eq.  $(26.31)$ , when the following equation is assumed

$$
\phi_x - C_1 4 \sigma_x \sigma_y + C_1 5 \sigma_y \mu_{xy} + 2 \mathcal{L} C_1 6 \mu_{xy}
$$
\n(26.40)  
\n
$$
\phi^{(m)}(\tilde{\sigma}, \bar{\varepsilon}) = \mu(\bar{\varepsilon}) \phi_A^{(m)}(\tilde{\sigma}) + (1 - \mu(\bar{\varepsilon})) \phi_B^{(m)}(\tilde{\sigma}),
$$
\n(26.41)

<span id="page-9-0"></span>it reduces to an interpolation for material parameters  $C_k$ ,  $k = 1, 2, ..., N$ , as follows

$$
C_k = \mu(\overline{\varepsilon})C_{k(A)} + (1 - \mu(\overline{\varepsilon}))C_{k(B)} \tag{26.42}
$$

Here,  $C_{k(A)}$  and  $C_{k(B)}$  are material parameters determined at the effective plastic strains,  $\bar{\varepsilon}_A$  and  $\bar{\varepsilon}_B$ , respectively. Assuming that  $\bar{\varepsilon}_A = 0$  (at initial yielding) and  $\overline{\varepsilon}_B$  =  $\infty$  (at infinitely large strain) and that  $C_{k(A)} = C_{k(0)}$  and  $C_{k(B)} = C_{k(\infty)}$ ,

Eq. [\(26.42\)](#page-9-0) reduces to the following:

$$
C_k = \mu(\overline{\varepsilon})C_{k(0)} + (1 - \mu(\overline{\varepsilon}))C_{k(\infty)}, \qquad 0 \le \overline{\varepsilon} \le \infty \tag{26.43}
$$

<span id="page-10-0"></span>In discretization form:

$$
C_k = \mu(\overline{\varepsilon})C_{k(i)} + (1 - \mu(\overline{\varepsilon}))C_{k(i+1)}, \qquad i = 1, 2, ..., M - 1 \tag{26.44}
$$

To validate the model, calculated stress-strain responses were compared with the corresponding experimental data for AA6022-T43 aluminum sheet (Stoughton and Yoo[n](#page-14-9) [2009](#page-14-9)). As for the yield function, sixth-order polynomial model is employed. Toon 2009). As for the yield function, stati-order potynomial model is employed.<br>
One of advantages of this model is that, flow stresses  $\sigma_0$ ,  $\sigma_{45}$ ,  $\sigma_{90}$ ,  $\sigma_b$  and *r*-values<br> *r*<sub>0</sub>, *r*<sub>45</sub>, *r*<sub>90</sub> are calc  $r_0$ ,  $r_{45}$ ,  $r_{90}$  are calculated by using the material parameters  $C_1 \sim C_{16}$  explicitly. Thus the material parameters are easily identified.

$$
\sigma_{90} = \left(\frac{C_1}{C_7}\right)^{\frac{1}{6}} \sigma_0, \quad \sigma_{45} = \left(\frac{C_1}{S + 9T + 27U + 27C_{16}}\right)^{\frac{1}{6}} \sigma_0, \quad \sigma_b = \left(\frac{C_1}{S}\right)^{\frac{1}{6}} \sigma_0, \tag{26.45}
$$

$$
r_0 = \frac{C_2}{2C_1 - C_2}, \quad r_{45} = \frac{-S - 3T + 9U + C_{16}}{2S + 12T + 18U}, \quad r_{90} = \frac{C_6}{2C_7 + C_6} \tag{26.46}
$$

$$
S = C_1 - 3C_2 + 6C_3 - 7C_4 + 6C_5 - 3C_6 + C_7,
$$
  
\n
$$
T = C_8 - 2C_9 + 3C_{10} - 2C_{11} + C_{12},
$$
  
\n
$$
U = C_{13} - C_{14} + 3C_{15}
$$

On AA6022-T43 aluminum sheet, *r*-value planar anisotropy remains fixed throughout the plastic deformation. In this calculation the kinematic hardening was excluded since in monotonic loading the stress-strain calculation is not affected by the kinematic hardening. The results of flow stresses,  $\sigma_0$ ,  $\sigma_4$ <sub>5</sub>,  $\sigma_{90}$ ,  $\sigma_b$  calculated using Eqs. [\(26.44\)](#page-10-0) and [\(26.38\)](#page-9-1), with  $M = 3$ , are compared to the experimental data, as shown in Fig. [26.5.](#page-11-0) Here, three discrete plastic strain points,  $\bar{\varepsilon} = 0, 0.1$  and 0.5 are selected to define  $\phi_1(\sigma)$ ,  $\phi_2(\sigma)$  and  $\phi_3(\sigma)$ . The calculated results agree well overall with the experimental results for the stresses.

The model was also validated by comparing the calculated results of stress-strain responses with experimental data on *r*-value and stress-directionality changes in an aluminum sheet (H[u](#page-13-6) [2007\)](#page-13-6) and a stainless steel sheet (Stoughton and Yoo[n](#page-14-9) [2009](#page-14-9)), as well as the variation of the yield surface of an aluminum sheet (Yanaga et al[.](#page-14-12) [2014](#page-14-12)). Furthermore, anisotropic cyclic behavior was examined by performing experiments of uniaxial tension and cyclic straining in three sheet directions on a 780MPa advanced high-strength steel sheet. For details of the results, refer to Yoshida et al[.](#page-14-5) [\(2015\)](#page-14-5).



<span id="page-11-0"></span>

## **26.5 Concluding Remarks**

The present paper describes a framework for the constitutive modeling of large-strain cyclic plasticity that describes the evolution of anisotropy and the Bauschinger effect of sheet metals based on the Y-U kinematic hardening model. The Y-U model predicts the springback much more accurately than the classical isotropic hardening model (e.g., refer to Yosh[i](#page-14-6)da and Uemori [2002;](#page-14-6) Eggertse[n](#page-13-23) and Mattiasson [2009](#page-13-23), [2010](#page-13-24); Ghaei et al[.](#page-13-25) [2010;](#page-13-25) Wagoner et al[.](#page-14-20) [2013](#page-14-20); Huh et al[.](#page-13-26) [2011\)](#page-13-26). It has gained popularity in the sheet metal forming industry because it has already been implemented into several FE commercial codes (e.g., PAM-STAMP, LS-DYNA, StamPack) and is widely used for springback simulation. The highlights of this modeling are summarized as follows.

- The Y-U model is highly capable of describing various cyclic plasticity characteristics such as the Bauschinger effect, the workhardening stagnation, strain-range-dependent cyclic workhardening, and the degradation of unloading stress-strain slope with increasing plastic strain. Furthermore, any type of anisotropic yield function can be used.
- It requires a limited number of material parameters (seven or eight plasticity parameters and three elasticity parameters including Young's modulus). The scheme for material parameter identification and testing have been clearly presented, see Yoshida and Uemor[i](#page-14-6) [\(2002\)](#page-14-6).
- The evolution of the anisotropy can be described by incorporating the proposed anisotropic hardening model in the Y-U model. In this modeling framework a set of kinematic hardening parameters can be identified experimentally independent of anisotropic hardening parameters, and their values remain fixed throughout the plastic deformation
- An anisotropic yield function that varies continuously with the plastic strain is defined by a nonlinear interpolation function of the effective plastic strain using a

limited number of yield functions determined at a few discrete points of the plastic strain. In this modeling framework, it is possible to use any type of yield function, and the convexity of the yield surface is always guaranteed.

• This approach, which requires only one interpolation equation, offers a great advantage over other approaches in that it involves fewer material parameters.

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