# **Calculating the Inherent Strain in 3D Printed Part Based on the Heat Affected Zone**



**Ngoc-Hien Tran and Thai-Son Nguyen** 

**Abstract** Today, 3D printing technology has been applied in industrial applications. However, improving the quality of printed products is still an issue that needs to be studied for this technology to be used more widely. This research proposes the calculation of the inherent strain (IS) value based on the heat affected zone (HAZ). Then, the IS values are used for calculating the deformation of the printed part. HAZ depends on the printing process parameters and material properties. This paper presents method to calculate the HAZ in different plans according to the printing process parameters for printing Ti6Al4V with selective laser melting (SLM) printing method.

**Keywords** Additive manufacturing · Metal printing · Inherent strain · Heat affected zone · Deformation

## **1 Introduction**

3D printing is used widely in the manufacturing industry because of its advantage. It is applied in aerospace, biomedical, automotive, etc. The 3D printing process can be divided into two groups such as powder spreading and powder spraying method. Selective laser melting (SLM) is a powder spreading method using a laser source to melt a thin powder layer. SLM is developed in 1995 by German scientists [\[1](#page-6-0)].

N.-H. Tran  $(\boxtimes)$ 

Faculty of Mechanical Engineering, University of Transport and Communications, Hanoi, Vietnam

e-mail: [tranhien.tkm@utc.edu.vn](mailto:tranhien.tkm@utc.edu.vn) 

T.-S. Nguyen

Center for Transport Science and Technology, University of Transport and Communications, Hanoi, Vietnam

<span id="page-1-0"></span>

Figure [1](#page-1-0) shows the SLM process. During the SLM process, a product is formed by selectively melting layers of powder using a laser beam. The powder is heated and melted to form a liquid pool. Then the molten pool solidifies and cools down and starts to form the product. The building platform is lowered by the amount of layer thickness and a new layer is deposited. The process is repeated until the product is made. Because of its ability to heat and cool rapidly, the SLM is used commonly. However, several defects usually exist in an SLM part, such as part distortion and cracks. The printed parts are affected by key influence factors such as powder properties, printing process parameters, and machine characteristics.

To obtain the best quality in a printed part, key influence factors must be considered. However, during the SLM process, large thermal gradients and fast cooling exist that cause residual stress, which leads to a loss of part shape and other failures of the SLM part. The failure quality of parts results in manufacturing costs, as well as waste and scrap [[2\]](#page-6-1). Therefore, all requirements and predictions for the process need to be fulfilled at the design stage to reduce failure and increase accuracy. To predict the quality at the design stage, a computational simulation is the best choice.

According to the previous research on this, the current methods to reduce deformation include [[1,](#page-6-0) [2\]](#page-6-1):

- Experiment: Achieve printing experiments to resolve the relationship between the printing parameters and deformation.
- Simulation: Analysis to complete the residual stress and then predict the deformation.

The contribution of this research is to suggest the deformation calculation based on the inherent strain (IS) value. The IS values in x, y, z directions are determined by the heat-affected zone (HAZ). HAZ depends on the printing process parameters and material properties. This paper presents a method to calculate the HAZ in different plans according to various printing process parameters using Ti6Al4V alloy material with selective laser melting (SLM) printing method.

#### **2 Calculation of Inherent Strain**

Inherent strain resulting from welding is a combination of thermal strains  $\varepsilon_{thermal}$ , phase transfer strain ε*phase*, plastic strains ε*plastic* and creep strain ε*creep* [[3,](#page-6-2) [4\]](#page-6-3).

In SLM printing process, the total strain is calculated as follows:

$$
\varepsilon_{total} = \varepsilon_{elastic} + \varepsilon^* \tag{1}
$$

in which  $\varepsilon_{elastic}$  is the elastic strain;  $\varepsilon^*$  is the inherent strain.

The  $\varepsilon^*$  value is calculated by considering other strains as follows:

$$
\varepsilon^* = \varepsilon_{thermal} + \varepsilon_{plastic} + \varepsilon_{phase} + \varepsilon_{creep}
$$
 (2)

During heating and cooling process, at the plastic stage the IS value in x, y, z directions can be defined as follows [[5\]](#page-6-4):

$$
\varepsilon_x^* = \frac{W_x}{F_x}; \varepsilon_y^* = \frac{W_y}{F_y}; \varepsilon_z^* = \frac{W_z}{F_z}
$$
 (3)

In which  $F_x$ ,  $F_y$ , and  $F_z$  are the heat effected zone (HAZ) areas where  $W_x$ ,  $W_y$ , and  $W_z$ , respectively, are distributed.  $W_x$ ,  $W_y$ , and  $W_z$  are calculated as follows:

$$
W_x = \xi. q_v; W_y = W_z = K. q_v \tag{4}
$$

In which  $q_v$  is the linear energy density (J/mm) [[6\]](#page-6-5):

$$
q_{v} = \frac{Q_{heatsource}}{u}
$$
 (5)

ξ and K (mm/J) values are used from the welding process due to the same heating and cooling process.

#### **3 Numerical Model for SLM Process**

For determining the temperature distribution during SLM process, the mathematical model of the heat transfer is as follows [[7\]](#page-6-6):

$$
\rho C \frac{\partial T}{\partial t} + \rho C \mathbf{u} \nabla T = \nabla (k \nabla T) + Q_{heatsource}
$$
 (6)

in which *T* is temperature;  $\rho$ , *C*, *k*, and *u* are density, thermal capacity, thermal conductivity factor and laser velocity, respectively. *Q* is defined as follows:

<span id="page-3-0"></span>



 $Q_{heatsource} = Q_{requiredpower} + Q_{conduction}$  (7)

$$
Q_{\textit{requiredpower}} = Q_{\textit{deliveredlaser}} - Q_{\textit{convection}} - Q_{\textit{radiation}} - Q_{\textit{conduction}}
$$

$$
Q_{deliveredlaser} = Q \frac{6\sqrt{3}}{ab\pi\sqrt{\pi}} e^{\left[\frac{-3z^2}{a^2} + \frac{-3y^2}{b^2}\right]} \left[\frac{f_f}{C_f} e^{\frac{-3x^2}{C_f^2}} + \frac{f_r}{C_r} e^{\frac{-3x^2}{C_r^2}}\right]
$$

The power distribution is given by the moving Goldak's double-ellipsoid heat source model as shown in Fig. [2](#page-3-0) [\[8](#page-6-7)]. The model parameters are explained in Table [1.](#page-4-0)

#### **4 Results and Discussion**

To determine the HAZ during the SLM process, we use the data from Table [1](#page-4-0) as the input data using Comsol tool. Figures [3,](#page-4-1) [4](#page-4-2) and [5](#page-5-0) show the method to calculate the HAZ in Fx, Fy, Fz plans.

$$
F_x = (x+e).d - e.c]
$$
\n<sup>(8)</sup>

With  $d = 0.153$  mm;  $c = 0.09$  mm;  $e = 1.97$  mm; and  $x = 2.34$  mm, we have  $F_x$  $= 0.482$  mm<sup>2</sup>. With  $\xi = 1.57 \cdot 10^{-3}$  mm<sup>3</sup>/J;  $q_v/h^2 = 104.52$  J/mm<sup>3</sup> for Ti6Al4V and  $h = 0.045$  mm, we have:

$$
\varepsilon_{inh-x} = -0.000937
$$
  

$$
F_y = 2(x + e).(a + b) - 2a.e]
$$
 (9)

Name	Description	Value
$x_0$	X-start coordinate	$0 \text{ (mm)}$
$y_0$	Y-start coordinate	$0 \text{ (mm)}$
$\mathcal Q$	Laser power from machine	140(W)
$\boldsymbol{a}$	Ellipsoid length in y axis	$0.4205$ (mm)
b	Ellipsoid length in z axis	$0.1682$ (mm)
$C\!f$	Front ellipsoid length in x axis	$0.4205$ (mm)
$\cal Cr$	Rear ellipsoid length in x axis	$0.84$ (mm)
$f_{\it f}$	Coefficient of heat source in the front ellipsoid	2/3
$f_r$	Coefficient of heat source in the rear ellipsoid	4/3
$\boldsymbol{u}$	Printing speed	$1300$ (mm/s)
$\mathcal{C}_{0}^{0}$	Thermal capacity	$500 \times 10^{-3}$ (J/(g.K))
λ	Thermal conductivity factor	$19.4 \times 10^{-3}$ [W/(mm.K)]
$T_a$	Ambient temperature	293(K)
$h_c$	Convective and (W/mm <sup>2</sup> .K)	186.3 [W/(mm <sup>2</sup> .K)]
$h_r$	Radiant heat transfer coefficient	4.4 [W/(mm <sup>2</sup> .K)]
$\boldsymbol{h}$	Layer thickness	$0.045$ (mm)
Ha	Hatch distance	$0.07$ (mm)

<span id="page-4-0"></span>**Table 1** SLM process parameters for Ti6Al4V material

<span id="page-4-1"></span>



<span id="page-4-2"></span>**Fig. 4** Calculation of the HAZ in Fy plane



With a = 0.19 mm; b = 0.102 mm; e = 1.97 mm; and x = 2.34 mm, we have  $F_v$  $= 1.275$  mm<sup>2</sup>.

With  $K = 0.58 \times 10^{-3}$  mm<sup>3</sup>/J;  $q_v/h^2 = 104.52 \text{ J/mm}^3$  for Ti6Al4V and h = 0.045 mm, we have:

<span id="page-5-0"></span>



$$
\varepsilon_{inh-y} = -0.0000963
$$
  

$$
F_z = 2[(a+b).d - a.c]
$$
 (10)

 $a = 0.19$  mm;  $b = 0.106$  mm;  $d = 0.153$  mm; and  $c = 0.09$  mm, we have  $F_Z =$ 0.0564 mm<sup>2</sup>. With K = 0.58 10<sup>-3</sup> mm<sup>3</sup>/J;  $q_v/h^2 = 104.52$  J/mm<sup>3</sup> for Ti6Al4V and h  $= 0.045$  mm, we have:

$$
\varepsilon_{inh-z}=-0.00217
$$

With the same processing parameters such as the beam width, velocity, power, and material, we used Simufact<sup>™</sup> tool for predicting the inherent strain. The predicted results are shown in Fig. [6.](#page-5-1) In comparison with the proposed method, the biggest IS value is in the Z direction, this is correct for both simulation and experiment [[2,](#page-6-1) [7\]](#page-6-6).

<span id="page-5-1"></span>

### **5 Conclusions**

In this study, we propose to calculate IS value based on printing process parameters and HAZ area. To calculate the HAZ area during SLM printing process, the heat transfer equation was established. Then,  $Comsol^{\mathbb{N}}$  was used to determine the temperature distribution in the printed part. From the temperature distribution, the HAZ surfaces were calculated. The IS values are used to calculate the deformation of the printed part in all directions. From that, the reasonable printing process parameters are determined to reduce distortion when carrying out the actual printing process. The proposed IS calculation results are reliable when compared with the IS values calculated by Simufact™ software under the same conditions and printing method.

#### **References**

- <span id="page-6-0"></span>1. Schoinochoritis B, Chantzis D, Salonitis K (2015) Simulation of metallic powder bed additive manufacturing processes with the finite element method: a critical review. Proc IMechE Part B: J Eng Manuf 231:96–117
- <span id="page-6-1"></span>2. Setien I, Chiumenti M, Veen S, Sebastian MS, Garciandía F, Echeverría A (2019) Empirical methodology to determine inherent strains in additive manufacturing. Comput Math Appl 78:2282–2295
- <span id="page-6-2"></span>3. Shokrieh MM, Jalili SM, Kamangar MA (2018) An eigen-strain approach on the estimation of non-uniform residual stress distribution using incremental hole-drilling and slitting techniques. Int J Mech Sci 148:383–392
- <span id="page-6-3"></span>4. Wang J, Ma N, Murakawa H (2015) An efficient FE computation for predicting welding induced buckling in production of ship panel structure. Mar Struct 41:20–52
- <span id="page-6-4"></span>5. Chen J, Lu H, Wang J, Chen W, Hao D (2000) Prediction of welding deformation with inherent strain method based on FEM, Shanghai Huizhong Automotive Manufacturing Co. LTD report
- <span id="page-6-5"></span>6. Donik Č, Kraner J, Paulin I, Godec M (2020) Influence of the energy density for selective laser melting on the microstructure and mechanical properties of stainless steel. Metals 10:919
- <span id="page-6-6"></span>7. Roberts IA (2012) Investigation of residual stresses in the laser melting of metal powders in additive layer manufacturing. Dissertation, University of Wolverhampton
- <span id="page-6-7"></span>8. Samad Z, Nor NM, Fauzi ERI (2019) Thermo-mechanical simulation of temperature distribution and prediction of heat-affected zone size in MIG welding process on aluminium alloy EN AW 6082-T6. IOP Conf Ser: Mater Sci Eng 530