

# Chapter 1c

## Metallic Biomaterials: Titanium and Titanium Alloys

H. Breme, V. Biehl, Nina Reger, and Ellen Gawalt

### 1c.1 Composition

**Table 1c.1** Comparison of international standards for titanium and titanium alloys (Refs. 1, 2<sup>a</sup>)

Alloy	Germany DIN	United Kingdom BS	France AIR	International Organization for Standardization ISO	United States ASTM	Japan JIS
Ti-1	17850	2TA1	9182 T-35	5832/II	F67/Grade 1	H4600
Ti-2	17850	2TA2	9182 T-35		F67/Grade 2	
Ti-3	17850	2TA6	9182 T-50		F67/Grade 3	
Ti-4	17850	2TA6, 2AT7 2AT8, 2AT9	9182 T-60		F67/Grade 4	
Ti6Al4V	17851	2TA10, 2TA11	9183T-A6V	5832/III	F136-13	H4607
		2TA13, 2TA28				
		TA56, TA59				
Ti5Al2.5Fe	17865	BS7252-10	–	5832/X	–	–
Ti6Al7Nb	17851		S94-081	5832/XI	FI295	–

<sup>a</sup>Ti-1–Ti-4 = commercially pure unalloyed titanium; others: direct chemical composition

H. Breme • V. Biehl

Lehrstuhl für Metallische Werkstoffe, Universität des Saarlandes, Saarbrücken, Germany

N. Reger • E. Gawalt (✉)

Department of Chemistry and Biochemistry, Bayer School of Natural and Environmental Sciences, Mellon 337, 600 Forbes Ave., Pittsburgh, PA 15282, USA

e-mail: [gawalte@duq.edu](mailto:gawalte@duq.edu)

**Table 1c.2** Chemical composition of commercially pure (cp)-titanium (wt%) (Refs. 1–3)

cp-Ti	Fe max.	O approx.	N max.	C max.	H max.	Ti
Grade 1	0.2	0.18	0.03	0.1	0.015	Balance
Grade 2	0.3	0.25	0.03	0.08	0.015	Balance
Grade 3	0.3	0.35	0.05	0.1	0.015	Balance
Grade 4	0.5	0.40	0.05	0.1	0.015	Balance

**Table 1c.3** Chemical composition of ( $\alpha+\beta$ )-titanium alloys (wt%) (Ref. 1)

Alloy	Al	V	Fe	Nb	Ta	O	N	C	H	Others		Ti
										Single	Sum	
Ti6Al4V	5.5–6.5	3.5–4.5	0.25	–	–	0.13	0.05	0.08	0.012	0.1	0.4	Balance
Ti5Al2.5Fe	4.5–5.5	–	2.0–3.0	–	–	0.2	0.05	0.08	0.015	0.1	0.4	Balance
Ti6Al7Nb	5.5–6.5	–	0.25	6.5–7.5	0.5	0.2	0.05	0.08	0.009	–	–	Balance

**Table 1c.4** Chemical composition (wt%) of  $\beta$  and near- $\beta$  titanium alloys (Refs. 4–7)

Alloy	Al	Mo	Zr	Ta	Nb	Sn	Ti
Ti15Mo5Zr3Al	3.8	15	5				77
Ti12Mo5Zr5Sn		12	5			4.5	Balance
Ti30Nb					30		70
Ti30Ta				29.6			Balance

## 1c.2 Physical Properties

**Table 1c.5** Physical properties of cp-Ti grade 1 (Refs. 8–11)

Young's modulus	102.7 GPa
Density	4.51 g/cm <sup>3</sup>
Melting point	1670 °C
Boiling point	3260 °C
Transformation temperature $\alpha$ - $\beta$	890 °C
Crystal structure	>882 °C $\beta$ bcc <882 °C $\alpha$ cph
Magnetic properties	Paramagnetic
Heat of transformation	67 kJ/kg
Thermal neutron-capture cross section	$5.8 \times 10^{-22}$ cm <sup>2</sup>
Specific heat at 15 °C	0.52 kJ/kg K
Heat of fusion	419 kJ/kg
Thermal conductivity at room temperature	17 W/mK
Thermal expansion coefficient between 0 and 315 °C	$9.6 \times 10^{-6}$ °C <sup>-1</sup>
Specific heat resistivity at 20 °C	0.5 $\mu\Omega$ m

**Table 1c.6** Physical properties of ( $\alpha + \beta$ )-titanium alloys (Refs. 4, 8–9, 12–14)

Property	Ti6Al4V	Ti5Al2.5Fe	Ti6Al7Nb
Young's modulus (GPa)	100–110	110	105
Density (g/cm <sup>3</sup> )	4.43	4.45	4.52
Microstructure	A1–A9	A1–A9	A1–A9
Transformation temperature (°C)	990 ± 15	950 ± 15	1010 ± 15
Thermal conductivity at room temperature (W/mK)	6.6	–	7
Coefficient of thermal expansion between 30 and 200 °C ( $\times 10^{-6} \text{ K}^{-1}$ )	9.5	9.3	–
Specific heat at 20 °C (kJ/kg K)	0.56	–	0.54
Specific electrical resistivity at 20 °C (52 mm <sup>2</sup> /m)	1.66	–	–

**Table 1c.7** Influence of alloying elements and heat treatment on Young's modulus (Refs. 5, 15–18)

Alloy	Heat treatment	Young's modulus (GPa)
Ti30Ta	As rolled	69
	1 h/1000 °C/H <sub>2</sub> O	63
Ti30Nb	As rolled	80
Ti15Mo5Zr3Al ELI <sup>a</sup>	Solution heat treated at 840 °C	75
	Solution heat treated at 740 °C	88
	Solution heat treated at 740 °C	113
	+ aged at 600 °C	

<sup>a</sup>ELI=extra-low interstitial

### 1c.3 Processing of cp-Ti and Ti Alloys

Provided that the following characteristics of titanium are taken into consideration, almost all processing procedures are possible:

1. High affinity to oxygen, nitrogen, and hydrogen gases
2. High reactivity to all metals to produce intermetallic compounds
3. Relatively low Young's modulus and therefore backspringing
4. Relatively low thermal conductivity
5. Tendency to stick to tools

#### 1c.3.1 Hot Working and Heat Treatment

Titanium and titanium alloys are fabricated into semifinished products by conventional methods such as forging, rolling, pressing, and drawing. When Ti materials are heated, care must be taken to avoid an excessive adsorption of oxygen, nitrogen, and hydrogen. Therefore, heating and annealing should take place in a neutral or

slightly oxidizing atmosphere. During heating in a gas-fired furnace, direct contact with the flame must be avoided because of the risk of hydrogen pickup and local overheating. In a short heating period, oxygen pickup is restricted to the surface area. This surface zone must be removed by chemical or mechanical methods. Hydrogen is able to penetrate the matrix rapidly; therefore, a reducing atmosphere must be avoided. The hot working temperature depends on the alloy composition and should be selected to obtain the best mechanical properties and grain structure (A1–A9 according to ETTC2, Ref. 17). Table 1c.8 summarizes the deformation temperatures for the various Ti materials. Table 1c.9 lists the temperature ranges and recommended annealing times for stress relieving, soft annealing, and solution treating with age hardening. When the cross section is very small, annealing is favorably carried out in a high-vacuum furnace. Prior to this annealing treatment, the oxide film must be removed from the surface to avoid diffusion of oxygen into the material.

### 1c.3.2 Working of Sheet

At room temperature cp-Ti grades 1 and 2 can be worked very well, and grades 3 and 4 only moderately well. Titanium alloys, because of their high yield/tensile strength ratio, can be worked only under certain conditions.

**Table 1c.8** Deformation temperatures for various titanium materials (Refs. 19, 20)

Alloy	Deformation temperature (°C)
cp-Ti Grade 1	650–870
cp-Ti Grade 2	650–870
cp-Ti Grade 3	700–900
cp-Ti Grade 4	700–930
Ti6Al4V	700–1100

**Table 1c.9** Recommendation for the heat treatment of cp-Ti and Ti alloys (Refs. 2, 8)

Alloy	Stress relief	Soft annealing	Solution annealing + age hardening
Ti grade 1	15 min–4 h at 480–595 °C/air	6 min–2 h at 650–750 °C/air	–
Ti grade 2			
Ti grade 3			
Ti grade 4			
Ti6Al4V	3 min/mm 1–4 h at 480–650 °C/air	Min. 1 h max. 4 h at 700–850 °C air or furnace cool	15 min–1 h at 955–970 °C/H <sub>2</sub> O +2 h at 480–595 °C/H <sub>2</sub> O
Ti5Al2.5Fe	3 min/mm 15 min–4 h at 540–650 °C/air	Min. 10 min max. 4 h at 720–845 °C air or furnace cool	15–1 h at 800–920 °C/H <sub>2</sub> O +2–4 h at 480–600 °C/air

For deep drawing, special coatings in the form of polymer foils have proved to be effective. At high temperatures colloidal graphite and common hot press greases with graphite or molybdenum disulfide additives have been successful. The Fe-, Ni-, and Cr-contents should be limited to 0.05, 0.1, and 0.33 wt%, respectively, to allow during a short annealing treatment (1–5 min at 750–850 °C) a grain growth producing an average grain size of 50–70  $\mu\text{m}$ . Due to this grain growth a deformation by twinning, with a resulting increased deep drawability, occurs (Ref. 19).

Superplastic forming is a material-saving and cost-reducing process for manufacturing parts of a complicated shape because it can be carried out together with diffusion welding in a single operation. Fine-grained alloys like Ti6Al4V and Ti5Al2.5Fe can be used for superplastic deformation. Other special deformation processes such as stretch forming, spinning, or explosion forming are also possible.

### 1c.3.3 Descaling

The average thickness of the oxide layer on the surface of cp-Ti as a function of temperature can be found in Table 1c.10 and the composition of the oxide layer on cp-Ti and Ti alloys can be found in Table 1c.11. The oxide layer on the surface of thick-walled pieces generated during deformation and/or heat treatment is removed by sandblasting and/or pickling. The workpiece is treated in an aqueous solution of 20 wt%  $\text{HNO}_3$  and 2 wt% HF.

Thin-walled pieces are merely pickled in an electrolytic solution or salt bath. It is important that not only the surface layer of oxide but also the underlying oxygen enriched diffusion zone is removed. Otherwise, the machinability and the service life of turning and milling tools would be negatively affected.

**Table 1c.10** Thickness of oxide formed on the surface of cp-Ti as a function of temperature (Ref. 2)

Temperature (°C)	Measurable thickness (mm)
315	None
425	None
540	None
650	<0.005
705	0.005
760	0.008
815	<0.025
870	<0.025
925	<0.05
980	0.05
1040	0.10
1095	0.36

**Table 1c.11** Nature of oxides formed on cp-Ti and titanium alloys (Ref. 22)

Alloy	Oxide				
	TiO <sub>2</sub>	Al <sub>2</sub> O <sub>3</sub>	Nb <sub>2</sub> O <sub>5</sub>	MoO <sub>3</sub> /MoO <sub>2</sub>	ZrO <sub>2</sub>
cp-Ti	×				
Ti6Al4V	×	×			
Ti5Al2.5Fe	×	×			
Ti6Al7Nb	×	×	×		
Ti15Mo5Zr3Al	×	×		×	×

### 1c.3.4 Machining

The machining of titanium materials presents no difficulties provided that the following properties are taken into account:

1. Relatively low thermal conductivity which may cause high thermal stresses at the cutting edge of the tool
2. Low Young's modulus which applies pressure to the tool
3. Tendency to stick to the tool

Titanium materials must be machined at low cutting speeds, at a relatively high feed rate, and with an ample supply of coolant (sulfur-containing oil; mixture of tetrachloride carbon, molybdenum sulfide, and graphite; 5 % aqueous solution of sodium nitrite, 5–10 % aqueous solution of water-soluble oil or sulfurized chlorinated oil). The cutting tools should be sharp and mounted as rigidly as possible. Recommended parameters for turning and milling are given in Table 1c.12. Since titanium dust and chips can easily catch fire, safety precautions must be taken. Threads should be cut on a lathe, as thread-cutting discs are subject to seizure.

Sawing causes no difficulties if a high blade pressure is used and the coolant supply is sufficient. Coarse-toothed blades (4 teeth per inch) are recommended.

For grinding, aluminum oxide (5–10 m/s) and silicon carbide (20–30 m/s) can be used.

### 1c.3.5 Soldering and Brazing

Immediately before soldering and brazing, the oxide layer present on the surface of titanium material must be removed. For direct applications using a torch, aluminum–zinc and tin–zinc solders are suitable. The higher temperatures required for brazing present the difficulty of avoiding the formation of intermetallic phases. As with almost all metals, titanium forms brittle intermetallic phases in the fusion zone. The only exception is silver, so that this metal forms one of the main constituents of brazers. The sources of heat used for brazing and soldering are the acetylene torch, high-frequency induction coils, infrared radiation, an inert-gas-shielded arc with graphite or tungsten electrodes, furnaces with an argon atmosphere (min. 99.99 %

**Table 1c.12** Recommendations for the cutting and milling of cp-Ti and Ti alloys (Ref. 8)

Alloy	Cutting speed (m/min)	Feed (mm/rev)	Depth of cut (mm)	Cutting angle	Relief angle
<i>Rough cutting</i>					
cp-Ti TC	30–75	0.2–0.4		0–6°	6–8°
HSS	7.5–4.0	0.1–1.25		6–15°	5–7°
			>2.5		
Ti alloys TC	15–25	0.2–0.4		0–6°	6–8°
HSS	3–15	0.1–0.4		6–15°	5–7°
<i>Forecutting</i>					
cp-Ti TC	60–100	0.1–0.4		6–6°	6–8°
HSS	18–50	0.075–0.2		6–15°	5–7°
			0.75–2.5		
Ti alloys TC	20–50	0.1–0.4		6–6°	6–8°
HSS	5–15	0.075–0.2		6–15°	5–7°
<i>Finish cutting</i>					
cp-Ti TC	60–100	0.075–0.3		0–15°	6–8°
HSS	20–50	0.05–0.1		5–6°	5–7°
			0.1–0.75		
Ti alloys TC	20–70	0.075–0.3		0–15°	6–8°
HSS	9–15	0.05–0.1		5–6°	5–7°
<i>Milling</i>					
cp-Ti TC	25–30	0.07–0.15			
HSS	50–60	0.1–0.2	>1.25 face cutter	–	–
Ti alloys TC	7.5–20	0.07–0.2	>2.5 gear cutter		
HSS	15–30	0.1–0.2			

TC hard metal (tungsten carbide)

HSS high-speed steel

and/or a dew point below  $-50$  °C), and high-vacuum furnaces. If brazing is not performed under vacuum or in a controlled atmosphere, fluxes are necessary to dissolve the oxide layers and prevent a pickup of gases.

### 1c.3.6 Welding

The inert-gas-shielded arc processes (TIG and MIG) are mainly used for fusion welding. In special cases resistance, ultrasonic, electron beam, diffusion, and laser welding are applied. With the cp-Ti grades the weld attains mechanical properties approximating those of the base metal. A slight decrease in ductility may occur with high tensile grades. Under passivating conditions, titanium welds have the same corrosion resistance as the base metal. On the contrary, in reducing media the weld may be subjected to a more severe corrosive attack than the base metal. During the welding operation, the weld, the heat-affected zone, and the underside of the weld are shielded from the atmosphere. The filler rod used is an uncoated wire of the same grade or of a grade

with a lower hardness than the base metal. Careful preparation of the joint is necessary; that is, surface impurities must be removed by grinding or pickling in order to avoid porosity. Even fingermarks can produce a hardening of the weld. A single layer can weld sheets up to 2.5 mm thick. In order to avoid local oxygen concentrations oxidation products, such as those found at the tip of the electrode, must be cut off. The effectiveness of the inert gas is responsible for the welding rate. The optimum argon flow rate has proved to be about 6–8 l/mm. After welding, the appearance of a dark blue or gray oxide layer indicates an insufficient inert gas shielding and an embrittlement of the weld due to oxygen and/or nitrogen pickup. The hardness of a good weld may exceed that of the fully recrystallized base metal by a maximum of 50 VHN. If, after a slight grinding of the surface, a hardness test should give a higher value, the weld must be completely removed because of embrittlement.

Electron beam welding is particularly suitable for titanium materials. It offers many advantages such as very narrow seams and small heat-affected zones, weldability of thick diameters, high welding speed, and reproducibility of even complex welds.

Titanium materials can be spot welded without any particular preparation under similar conditions to those of stainless steel. Using flat-tipped electrodes, spot welding can be performed without inert gas. A hardening of the zone by up to 50 VHN compared with the base metal is regarded as normal and does not diminish the strength of the joint. Seam and flashbutt welding are also possible if an argon atmosphere is used.

Diffusion welding is of particular importance for titanium materials because these materials are more amenable to a homogeneous band in the solid state than other metals. After welding, the joint zone shows a higher temperature under high vacuum or, in an inert atmosphere, a microstructure very similar to that of the base metal.

## 1c.4 Mechanical Properties

**Table 1c.13** Mechanical properties of commercially pure titanium (Ref. 3, 21)

Titanium <sup>a</sup>	Tensile yield strength (0.2 %) (MPa)	Ultimate tensile strength (MPa)	Ratio yield/tensile strength	Elongation at fracture <sup>b</sup> (%)	Reduction of area <sup>b</sup> (%)	Brinell hardness	Bend radius (105°) for sheet thickness (T)	
							<1.8 mm	>1.8–4.75 mm
Grade 1	170	240	0.71	24	30	110–170	3T	4T
Grade 2	275	345	0.80	20	30	140–200	4T	5T
Grade 3	380	450	0.84	18	30	140–200	4T	5T
Grade 4	483	550	0.88	15	25	200–275	5T	6T

<sup>a</sup>Condition: Sheet, as rolled

<sup>b</sup>Minimum values



**Table 1c.14** Influence of a cold deformation on the mechanical properties of commercially pure titanium (Ref. 23)

cp-Titanium	Condition (%)	Tensile yield strength (0.2 %) (MPa)	Ultimate tensile strength (MPa)	Ratio yield/tensile strength	Elongation at fracture (%)
Ti grade 1	30	555	635	0.87	18
	40	560	645	0.87	16
	55	605	710	0.85	15
	60	620	725	0.86	14
	65	640	730	0.88	14.5
Ti grade 2	30	605	680	0.89	18
	40	645	740	0.87	17
	50	680	780	0.87	16
	60	685	795	0.86	16
	65	692	810	0.85	16.5

**Table 1c.15** Mechanical properties of  $\beta$  and near- $\beta$ -titanium alloys (experimental alloys) (Refs. 6, 15, 24–26)

Alloy	Tensile yield strength (0.2 %) (MPa)	Ultimate tensile strength (MPa)	Ratio yield/tensile strength	Elongation at fracture (%)	Reduction of area (%)
Ti5Mo5Zr3Al	838	852	0.98	25	48
Ti30Nb	500	700	0.71	20	60
Ti30Ta	590	740	0.80	28	58

**Table 1c.16** Influence of heat treatment on the mechanical properties of  $\beta$ - and near- $\beta$ -titanium alloys (Refs. 6, 15)

Alloy	Condition	Tensile yield strength (0.2 %) (MPa)	Ultimate tensile strength (MPa)	Ratio yield/tensile strength	Elongation at fracture (%)	Reduction of area (%)
Ti5Mo5Zr3Al	SHT at 840 °C	870	882	0.99	20	83.2
	SHT at 740 °C	968	975	0.99	16.9	64.5
	SHT at 740 °C	1087	1099	0.99	15.3	57.5
	+ aged at 600 °C					
	45 % cold worked	1284	1312	0.98	11.3	43.8
	+ aged at 600 °C					
Ti30Ta	Annealed at 1100 °C	650	800	0.81	8	42
	Annealed at 1200 °C	660	800	0.83	8	38
	Annealed at 1300 °C	665	800	0.83	8	30
	Annealed at 1400 °C	680	800	0.83	6	18

SHT solution heat treatment



**Table 1c.18** Influence of a solution treating and ageing on the mechanical properties of Ti6Al4V

Condition	Tensile yield strength (0.2 %) (MPa)	Ultimate tensile strength (MPa)	Ratio yield/ tensile strength	Elongation at fracture (%)	Reduction of area (%)
Sheet $\leq$ 12.5 mm 15–60 min at 800–920 °C/H <sub>2</sub> O + 2–4 h 480–600 °C/air	1070	1140	0.94	8	20

**Table 1c.19** Influence of a plasma nitriding (PVD) on the mechanical properties of Ti6Al4V (Ref. 30)

Treatment	Tensile yield strength (0.2 %) (MPa)	Ultimate tensile strength (MPa)	Ratio yield/ tensile strength	Elongation at fracture (%)
Untreated	809	894	0.90	20
Vacuum annealed 20 h/850 °C	815	924	0.88	21
Plasma nitrided <sup>a</sup> 20 h/850 °C/N <sub>2</sub>	805	914	0.88	20
Plasma nitrided <sup>a</sup> 20 h/850 °C/NH <sub>3</sub>	880	984	0.89	20

<sup>a</sup>Plasma nitriding at 20–40 kW

**Table 1c.20** Fracture toughness of Ti-alloys (Refs. 6, 31)

Alloy	Condition	Fracture toughness $K_{IC}$ (N/mm <sup>3/2</sup> )
Ti6Al4V	Annealed	1740
	Solution treated + annealed	2020
Ti5Al2.5Fe	Annealed (2 h/700 °C/air)	1225
	Solution treated + annealed (1 h/900 °C/H <sub>2</sub> O/2 h/700 °C/air)	1785
Ti5Mo5Zr3Al	Solution treated at 740 °C	4580
	Solution treated at 740 °C + annealed at 600 °C	2430
	40 % cold worked + annealed at 600 °C	980

## 1c.5 Fatigue

**Table 1c.21** High cycle fatigue strength  $\sigma_B$  and rotating fatigue strength  $\sigma_R$  of pure titanium and titanium alloys (Wöhler curves) (Refs. 15, 32–37)

Alloy	$R$	$\sigma_B$ (MPa)	$R$	$\sigma_R$ (MPa)
cp-Ti	-1	230–280	-1	200
Ti6Al4V	-1	610–625	-1	500–660
Ti5Al12.5Fe	-1	580	-1	450–550
Ti6Al7Nb	-1	500–600	-1	450–600
cp-Nb	-1	270	-1	150
cp-Ta	-1	410	-1	200
Ti30Ta	-1	–	-1	400

**Table 1c.22** Rotating bending fatigue tests of unnotched and notched titanium alloys (Ref. 31)

Alloy	$R$	Condition	Stress concentration factor	Fatigue strength for alternating tensile stresses ( $>10^7$ cycles) (MPa)
Ti6Al4V	-1		1.0	725
Ti5Al12.5Fe	-1	Wrought + annealed	1.0	725
	-1	Wrought + solution	3.6	300
	-1	Treated + annealed		
	-1	Cast + HIP	3.6	300
	-1	Cast + HIP	1.0	450

**Table 1c.23** High cycle fatigue strength of hip endoprostheses of titanium alloys, measured in 0.9 % NaCl solution at 37 °C. Testing conditions similar to DIN 58840 (simulated loosened shaft, 50 mm) (Refs. 34, 36, 38)

Alloy	Maximum load in $2 \times 10^7$ cycles (kN) 0.9 % NaCl ( $f=2$ Hz)
Hot wrought Ti6Al4V	6.5–8.0
Wrought Ti5Al12.5Fe	8
Ti6Al7Nb	3.5–6.0

**Table 1c.24** Influence of the mean stress  $S_m$  on the fatigue strength of Ti6Al4V (Ref. 39)

$S_m$ (MPa)	$R$	Notch factor $K_f$	Fatigue strength (MPa) at $10^7$ cycles
0	-1	1	400
		2.82	250
250	-0.1	1	300
	0.33	2.82	125
500	0.33	1	250
	0.66	2.82	100
750	0.7	1	125
	0.81	2.82	80

**Table 1c.25** Influence of the notch factor on the fatigue strength of Ti6Al4V (Refs. 40–43)

Notch factor $K_f$	Stress ratio $R$	Fatigue strength (MPa)
1	-1	400
1.7	-1	300
3.7	-1	150
6.0	-1	100

**Table 1c.26** Influence of interstitial elements on the rotating bending strength of Ti6Al4V (Ref. 44)

Chemical composition of the base alloy (wt%)							
Al	V	Fe	C	N	O	H	Ti
6.03	3.96	0.10	0.02	0.009	0.1	0.005	balance
Composition				Notch factor		Fatigue strength	
Heat treatment		$K_f$	$R$	(MPa)			
Base alloy		1 h 815 °C/furnace	1	-1	610		
		600 °C air	3	-1	210		
Base alloy		1 h 855 °C/furnace	1	-1	510		
+0.07 % N		600 °C air	3	-1	180		
Base alloy		1 h 870 °C/furnace	1	-1	550		
+0.2 % O		600 °C air	3	-1	210		
Base alloy		1 h 840 °C/furnace	1	-1	560		
+9.2 % C		600 °C air	3	-1	230		
Base alloy		1 h 930 °C/furnace	1	-1	580		
+0.07 % N		600 °C air	3	-1	240		
+0.2 % O							
+0.2 % C							

**Table 1c.27** Influence of texture and test directions on the rotating bending fatigue strength of Ti6Al4V (fine equiaxed microstructure in rolled plates) (Ref. 45)

Type of texture and method of production	$R$	Text direction	Fatigue strength (MPa) test direction at $10^7$ cycles
Basal (0002 tilted out of the rolling plane by about 30°) cross rolling in lower ( $\alpha + \beta$ )-field	-1	Rolling direction	625
Transverse (0002 aligned parallel to the rolling direction) unidirectional in the higher ( $\alpha + \beta$ )-field, 950 °C	-1	Rolling direction Transverse direction	630 590
Basal/transverse (both are present) Unidirectional roll at about 930 °C	-1	Rolling direction Transverse direction	720 690

**Table 1c.28** Influence of heat treatment (annealing and precipitation hardening, respectively) on the fatigue strength of Ti6Al4V (Ref. 43, 46)

Condition	Yield strength (MPa)			Tensile strength (MPa)
As annealed	900			955
As hardened	1100			1195
				Fatigue strength (MPa)
Condition	$s_m$	$K_f$	$R$	at $10^7$ cycles
As annealed	0	1	-1	510
	0	3.3	-1	300
As hardened	0	1	-1	600
	0	3.3	-1	280
As annealed	200	1	-0.3	425
	200	3.3	-0.01	200
As hardened	200	1	-0.5	600
	200	3.3	0	200
As annealed	300	1	0.14	400
	300	3.3	0.23	165
As hardened	300	1	-0.22	550
	300	3.3	-0.2	190

**Table 1c.29** Influence of the beta field heat treatment on the fatigue strength of Ti6Al4V (Ref. 47)

Heat treatment	$R$	Fatigue strength (MPa) at $10^7$ cycles
0.5 h 1010 °C/AC+2 h 700 °C/AC	0	525
5 h 1000 °C/AC+2 h 700 °C/AC	0	620
0.5 h 1010 °C/H <sub>2</sub> O+2 h 700 °C/AC	0	750
5 h 1010 °C/H <sub>2</sub> O+2 h 700 °C/AC <sub>2</sub>	0	650

**Table 1c.30** Influence of the surface treatment on the rotating bending fatigue (fine lamellar microstructure, produced by annealing in 15 min/1050 °C/H<sub>2</sub>O+1 h/800 °C/H<sub>2</sub>O) (Ref. 48)

Surface treatment	$R$	Fatigue strength (MPa) at $10^7$ cycles
Electrically polished	-1	680
Mechanically polished (7 µm)	-1	750
Mechanically polished (80 µm)	-1	605
Mechanically polished (80 µm)+1 h 500 °C	-1	550
Mechanically polished (180 µm)	-1	600
Mechanically polished (80 µm)+1 h 800 °C	-1	450

**Table 1c.31** Influence of the surface treatment on the rotating bending fatigue of Ti6Al4V (fine equiaxed microstructure produced by rolling at 800 °C/H<sub>2</sub>O + 1 h/800 °C (HP)) (Ref. 48)

Surface treatment	<i>R</i>	Fatigue strength (MPa)
Electrically polished	-1	610
Shot peened	-1	710
Shot peened + 1 h 500 °C	-1	390
Shot peened + 1 h 500 °C 20 µm surface removed	-1	800
Shot peened + 20 µm surface removed	-1	820

**Table 1c.32** Influence of surface working on the rotating bending of Ti6Al4V (Ref. 40, 49)

Surface working	Notch factor ( <i>K<sub>t</sub></i> )	<i>R</i>	Fatigue strength (MPa)
Mechanically polished	1	-1	620
Mechanically polished + cold roll bent	1	-1	660
Ground	1	-1	540
Mechanically polished	2.02	-1	330

**Table 1c.33** Influence of plasma nitriding (PVD) on the rotating bending fatigue of Ti6Al4V (Ref. 30)

Treatment	<i>R</i>	Maximum bending stress at 10 <sup>7</sup> cycles (MPa)
Untreated	-1	600
Vacuum annealed 20 h/850 °C	-1	370
Plasma nitrided <sup>a</sup> 20 h/850 °C/N <sub>2</sub>	-1	470
Plasma nitrided <sup>a</sup> 20 h/700 °C/NH <sub>3</sub>	-1	550
Solution treated 1 h/940 °C/vac. + Ar cooled	-1	530
Solution treated 1 h/940 °C/vac. + Ar cooled + plasma nitrided <sup>a</sup> at 20 h/770 °C/N <sub>2</sub>	-1	500

<sup>a</sup>Plasma nitriding at 20–40 kW

## 1c.6 Corrosion and Wear

**Table 1c.34** Electrochemical data for titanium and titanium alloys in 0.1 M NaCl under different conditions (Refs. 24, 50–54)

Alloy	Corrosion potential	Passive current density breakdown potential	
	$E_{\text{corr}}$ (mV)	$I_p$ ( $\mu\text{A}/\text{cm}^2$ )	$E_b$ (mV)
cp-Ti			
pH 7	–628	2.5	
pH 2	–459	5–10	>1500
Ti5Al2.5Fe			
pH 7	–529	0.68	
pH 2	–567	0.71	>1500
Ti6Al4V			
pH 7	–510	0.92	
pH 2	–699	0.69	>1500
Ti6Al7Nb			
pH 7	–368	0.53	>1000
Ti30Ta			
pH 7	–419	0.3	>1500

**Table 1c.35** Polarization current ( $i$ ) and polarization resistance ( $R_c$ ) of titanium and titanium alloys in pure saline at 37 °C (Ref. 50) and in 0.9 % NaCl with a stable redox system  $[\text{Fe}(\text{CN})_6]^{4-}/\text{Fe}(\text{CN})_6^{3-}$  (Ref. 55)

Material	$i$ ( $\mu\text{A}/\text{cm}^2$ )	$R_c$ ( $\text{k}\Omega \text{cm}^2$ )	
		Pure saline	Saline + redox
cp-Ti	0.010	1000	714
Ti6Al4V	0.008	1250	455

**Table 1c.36** Repassivation time in 0.9 % NaCl and breakdown potential in Hanks' solution of cp-Ti and Ti alloys (Refs. 24, 32, 56, 57)

Alloy	Breakdown potential (mV) vs. calomel electrode	Repassivation –500 mV	Time (ms) +500 mV
cp-Ti	2400	43	44.4
cp-Ta	2250	–	–
Ti30Ta	>1500	41.7	47.5
Ti40Nb	>1500	44.6	43.4
Ti6Al4V	>2000	37	41
Ti5Al2.5Fe		110–130	120–160



**Table 1c.37** Electrochemical data for anodic titanium and Ti6Al4V at 37 °C in different solutions (de-aired) versus standard calomel electrode (SCE) (Ref. 58)

Alloy	Corrosion potential vs. SCE $E_p$ (mV)	Passive current density $I_b$ ( $\mu\text{A}/\text{cm}^2$ )	Breakdown potential $E$ (mV)	Solution <sup>a</sup>
cp-Ti	-440 to 490	1.0–3.0	1300	1
	-94 to 140	0–1.0	1750	2
	-94	5.0–9.0	1950	3
Ti6Al4V	-200 to 250	0.9–1.0	1155–1240	1
	-240 to 300	0.8–1.5	1900	2
	-180 to 250	0.9–2.0	1550	3

<sup>a</sup>1=Ringer's solution; 2=Hanks' solution; 3=0.17 M NaCl solution

**Table 1c.38** Electrochemical data for cp-Ti and Ti alloys after 7 days in artificial saliva (Ref. 59)

Alloy	Corrosion potential (mV)	Current densities	
		$I_{\text{corr}}$ (nA/cm <sup>2</sup> )	$I_{\text{pass}}$ ( $\mu\text{A}/\text{cm}^2$ )
cp-Ti	-260	21	9.5
Ti6Al4V	-230	31	10
Ti6Al7Nb	-220	32	9
Ti5Al2.5Fe	-180	30	6

**Table 1c.39** Repassivation time of titanium and titanium alloys in contact with different metallic materials (Ref. 60)

NaCl pH=7.4 (shortcut alloy)	HCl pH=3 (activated alloy)	Repassivation time $t_c$ (ms)
cp-Ti	cp-Ti	–
	Ti30Ta	37.7
	Ti6Al4V	41.8
	cp-Nb	480.0
	Co30Cr6.5Mo	38.7
	Co28Cr5Mo	38.4
	X3CrNiMo1812	1000.0
Ti30Ta	cp-Ti	43.0
	Ti30Ta	48.6
	Ti6Al4V	39.0
	cp-Nb	1080.0
	Co30Cr6.5Mo	44.1
	Co28Cr5Mo	(4200)
	X3CrNiMo1812	1000.0
Ti6Al4V	cp-Ti	44.0
	Ti30Ta	34.2

**Table 1c.40** Influence of the surface treatment on the fretting corrosion behavior of Ti6Al4V (Ref. 61)

Material combination	Total weight loss ( $\mu\text{g}$ )	Ti ( $\mu\text{g}$ )	V ( $\mu\text{g}$ )
Untreated–untreated	2423	3925	78.5
Untreated–nitrogen ion implanted	1295	1260	31.2
Untreated–PVD coated with TiN	1002	902	15.0
Untreated–plasma ion nitrided	807	716	6.4
PVD–PVD	713	470	8.5
Plasma ion nitrided–plasma ion nitrided	273	87	0.5

Testing conditions: plate screw system (micromotion=100), 14 days in calf serum solution (1 Hz for 1,200,000 cycles)

**Table 1c.41** Influence of the surface treatment on the wear behavior of Ti6Al7Nb as a result of a pin-on-disk test (Ref. 62)

Property	PVD coated with 3 $\mu\text{m}$ TiN layer	Oxygen diffusion hardened (ODH) (30 $\mu\text{m}$ hardened surface)
Wear factor ( $10^{-7}$ $\text{mm}^3/\text{Nm}$ )	2.111	1.353
Coefficient of friction	0.078	0.051
Surface roughness $R_z$ ( $\mu\text{m}$ )	0.159	0.330
Wetting angle ( $^\circ$ )	47	49

**Table 1c.42** Volumetric wear rate of Ti6Al4V and Ti6Al7Nb under different sliding speeds and normal load (Ref. 63)

Sliding speed	Load (N)	Volumetric wear ( $\text{mm}^3 \text{N}/\text{mm}$ )	
		Ti6Al4V	Ti6Al7Nb
1 mm/s	3	$4.45 \times 10^{-3}$	$5.48 \times 10^{-3}$
	6	$4.24 \times 10^{-3}$	$9.64 \times 10^{-3}$
	10	$11.08 \times 10^{-3}$	$13.12 \times 10^{-3}$
15 mm/s	3	$20.35 \times 10^{-3}$	$22.06 \times 10^{-3}$
	6	$37.98 \times 10^{-3}$	$38.10 \times 10^{-3}$
	10	$51.55 \times 10^{-3}$	$57.35 \times 10^{-3}$
25 mm/s	3	$27.32 \times 10^{-3}$	$31.38 \times 10^{-3}$
	6	$42.15 \times 10^{-3}$	$45.28 \times 10^{-3}$
	10	$54.21 \times 10^{-3}$	$57.62 \times 10^{-3}$

**Table 1c.43** Influence of ion implantation of nitrogen on the wear properties of commercial cp-Ti and Ti6Al4V (Ref. 64)

Material	Friction couple	Total wear (mg)	Friction coefficient
cp-Ti <sup>a</sup>	Untreated–untreated	632.3	0.48
	Nitrided–nitrided	54.3	0.10
	4 h/940 °C/N <sub>2</sub> :H <sub>2</sub> =2:1		
Ti6Al4V <sup>b</sup>	Untreated–untreated	600.0	0.46
	Nitrided–nitrided	40.1	0.10
	4 h/940 °C/N <sub>2</sub> : H <sub>2</sub> =2:1		
	Nitrided–nitrided	92.3	0.12
	6 h/800 °C/N <sub>2</sub> :H <sub>2</sub> =1:1		

<sup>a</sup>Friction distance = 1257 m<sup>b</sup>Friction distance = 1885 m**Table 1c.44** Rate of formation of corrosion products for cp-Ti in Hanks' solution during current-time-tests (Ref. 65)

Polishing method	Metal converted into compound (ng/cm <sup>2</sup> h)
Mechanically polished	4.1
Chemically polished	3.5

## 1c.7 Biological Properties

**Table 1c.45** Biocompatibility of cp-Ti and Ti alloys, survival rate of L132 cells incubated with powders (Ref. 66)

Alloy	Powder concentration (mg/L)	Survival rate of cells
cp-Ti	>400	
Ti6Al4V	>400	
Ti5Al2.5Fe	>400	>80 %
Ti30Ta	>400	
Ti30Nb	>400	

**Table 1c.46** Influence of the implantation time (in vivo) on the surface roughness and peak-to-valley (P–V) height of Ti6Al4V femoral heads (Ref. 67)

Implantation time								
Position	Before implantation		85 months		110 months		124 months	
	R <sub>a</sub> (nm)	P–V	R <sub>a</sub> (nm)	P–V	R <sub>a</sub> (nm)	P–V	R <sub>a</sub> (nm)	P–V
Anterior	43±10	370±72	58±50	746±509	250±147	2044±1178	86±81	812±763
Posterior	41±6	591±333	150±125	2281±1842	114±96	1175±778	142±131	1045±890
Medial	51±14	411±159	44±29	649±259	118±69	1224±731	412±11	401±125
Lateral	52±9	364±68	71±55	722±474	117±106	1195±1009	40±16	527±156

## 1c.8 Nitinol: Shape Memory

**Table 1c.47** Properties of Nitinol (shape memory, Ni<sub>45</sub>Ti) alloy (Refs. 50, 68–71)

	Above (=austenitic)		Below (= martensitic)
	Transition temperature (–200 to 110 °C)		
Density (g/cm <sup>3</sup> )		6.45	
Melting point (°C)		1310	
Young's modulus (GPa)	83		28–41
Tensile yield strength (0.2 %) (MPa)	195–690		70–140
Ultimate tensile strength (MPa)	640–1380		103–862
Ratio yield/tensile strength	0.75–0.68		0.33–0.16
Elongation at fracture (%)	1–15		up to 60
Thermal expansion coefficient (°C <sup>-1</sup> )	1.1 × 10 <sup>-7</sup>		6.6 × 10 <sup>-6</sup>
Specific heat (Cal/g °C)		0.20	
Electrical conductivity (S/m)		1–1.5	
Corrosion potential (mV)			
in 0.1 M NaCl pH 7		–431	
pH 2		–518	
Passive current density (A/cm <sup>2</sup> )			
in 0.1 M NaCl pH 7		0.44	
pH 2		0.61	
Breakdown potential (mV)			
in 0.1 M NaCl pH 7		890	
pH 2		960	

**Table 1c.48** Resulting oxide layer thickness on Nitinol stents using different preparation techniques (Ref. 72)

Surface treatment	Oxide layer thickness (Å)
Electropolished	34 ± 14
Passivated in 10 % HNO <sub>3</sub>	30 ± 1
Air aged at 450 °C	240 ± 70
Heat treated in a NO <sub>2</sub> <sup>-</sup> /NO <sub>3</sub> <sup>-</sup> solution 500 °C	911 ± 270

**Table 1c.49** Ion release from Nitinol incubated with L132 cell culture (Ref. 73)

	Ni (ppm)		Ti (ppm)	
	3 days	6 days	3 days	6 days
hp-Ni	6.599 ± 0.037	11.364 ± 0.034	n.m.	n.m.
cp-Ti	n.m.	n.m.	0.001	0.002
NiTi	0.081 ± 0.006	0.176 ± 0.008	0.004	0.006

*hp-Ni* high-purity nickel, *n.m.* not measured

**Table 1c.50** Survival rate of L132 cells incubated with Nitinol powder (Ref. 74)

Metal	Powder concentration ( $\mu\text{g/mL}$ )	Survival rate of cells
Nitinol	400	71.2 %

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